# Advances in Entry System Technologies – Continuing the Ames' Innovation Heritage

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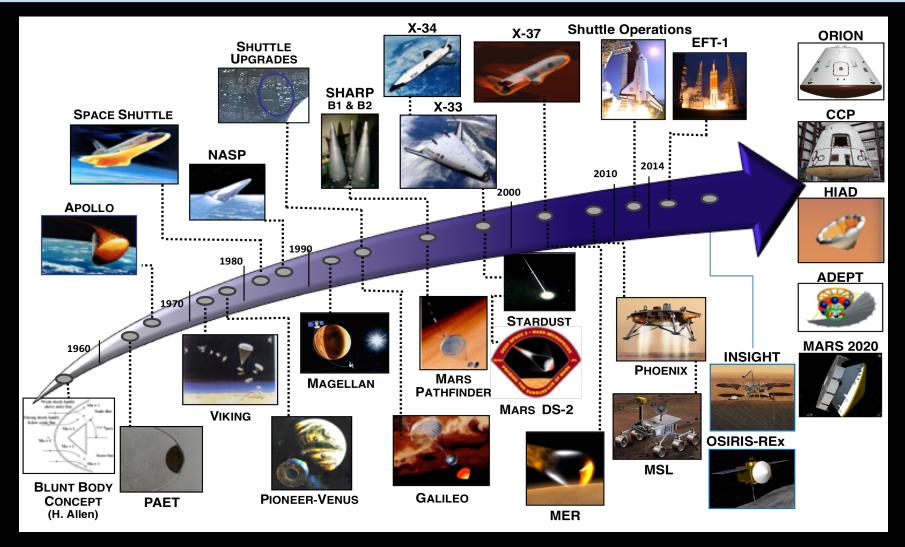
#### What is this talk about?



- Ames has a long history of innovations and contributions towards enabling NASA missions
  - Entry Systems and Technology Division at Ames, leveraging the past successes, is continuing this tradition.
- This talk highlights some of the recent innovations that are enabling current missions and laying the ground work for future scientific and human exploration "as only Ames can".

## Ames' Heritage and Continuing Contributions



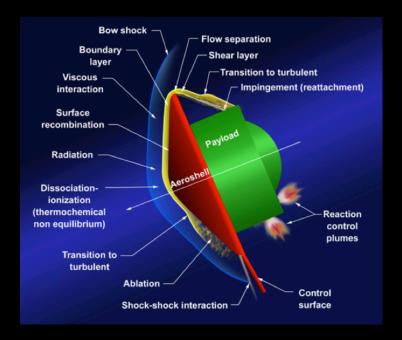


<sup>\*</sup> H. Julian Allen and Al Eggers, "A Study of the Motion and Aerodynamic Heating of Ballistic Missiles Entering the Earth's Atmosphere at High Supersonic Speeds," NACA-RM-A53D28, 1953 / NACA-TR-1381, 1958.



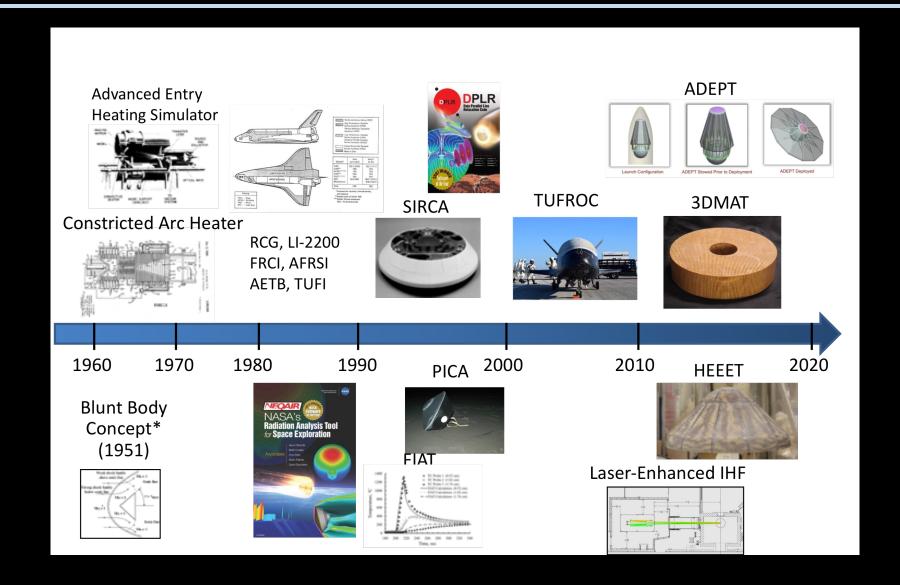
## **Hypervelocity Entry is a Hard Problem!**

Use of atmospheric drag is the most efficient way to slow down. Protection from the entry heating demands comprehensive understanding of the hypervelocity, reacting flow (aero-thermodynamics), and selection, design, testing and verification of the integrated entry system, especially thermal protection system.



Entry and Thermal Protection System have to be fail-safe, mass efficient and robust.

## NASA Ames Entry Systems Inventions



#### **Recent Innovations**

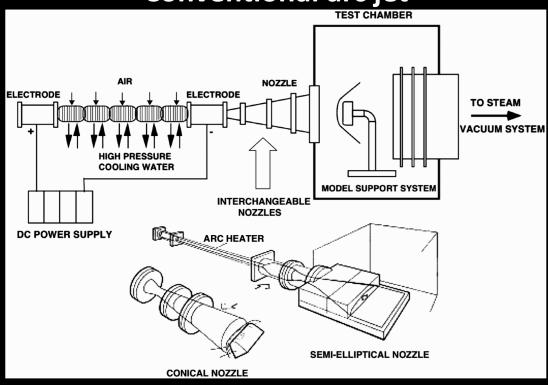


- Laser Enhanced Arc Jet Testing
- 3-D MAT
- HEEET
- ADEPT



## Laser Enhanced Arcjet Facility (LEAF)



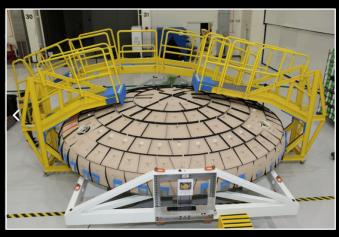


- Why Laser Enhanced arc jet?
- LEAF system at Ames
- Exploratory tests with Orion heat-shield material

## Why do we need a Laser Enhanced Arcjet Facility (LEAF)?



- Orion Heat shield design has to withstand both shock layer radiation and convective heating
- Heat Shield System Certification Challenges
  - Tiled System with gap-filler
  - Compression-pad region

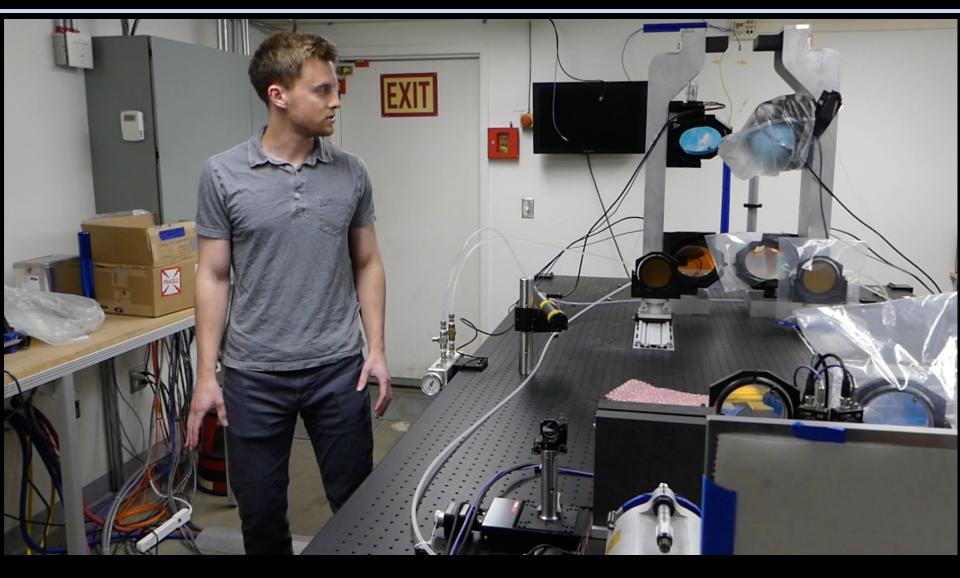






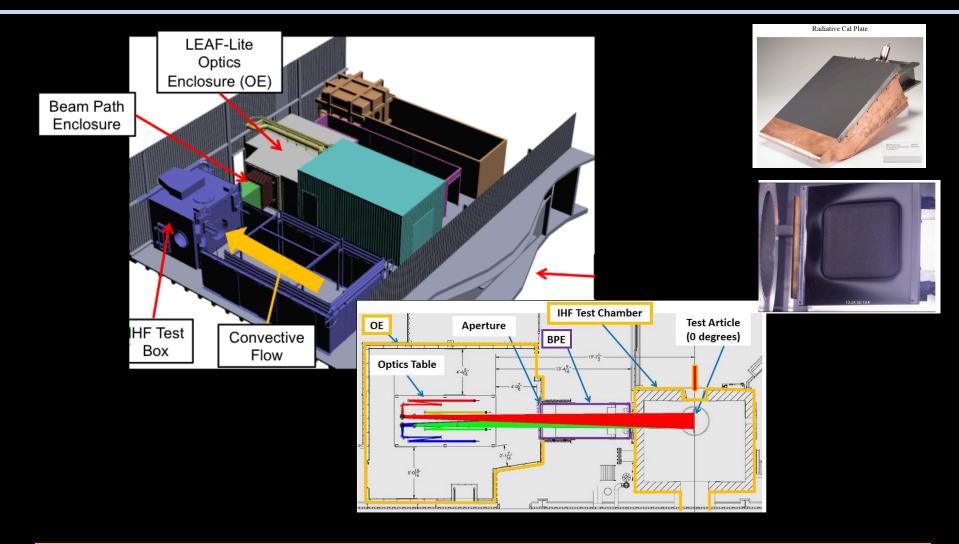


## LEAF Explained (by the Lead Engineer)



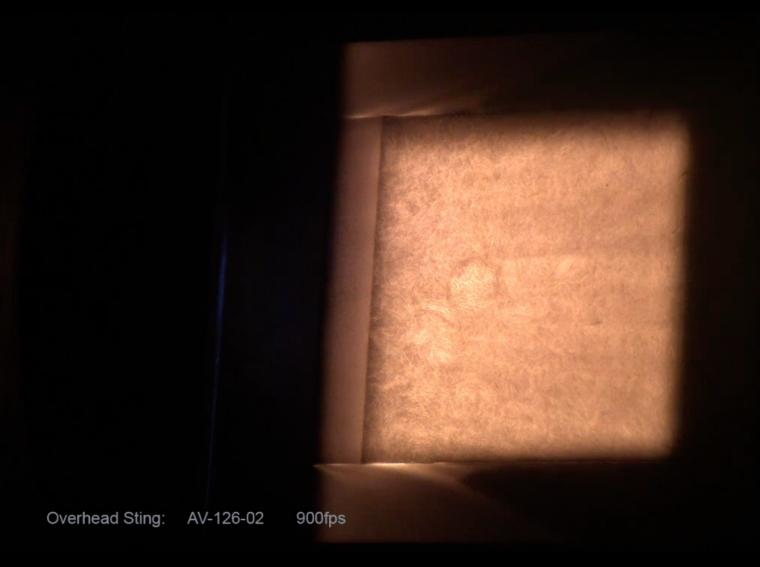


## Integrated IHF and LEAF Setup



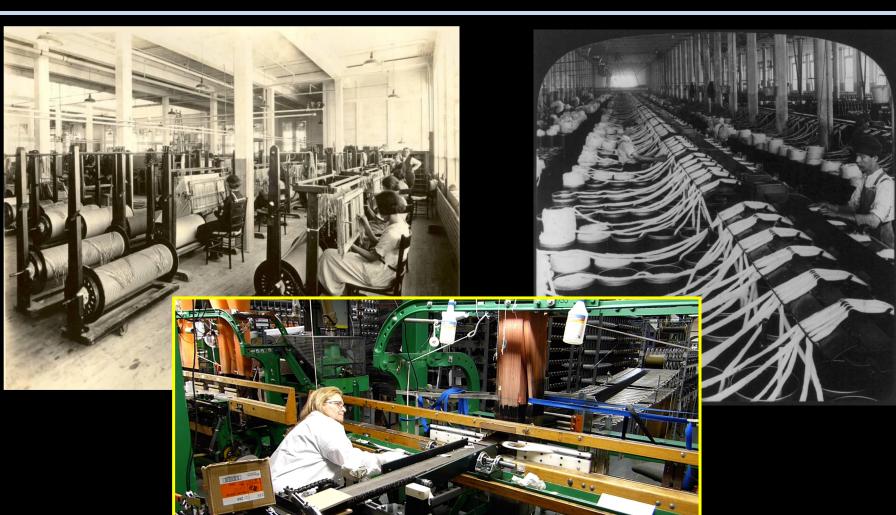
Up to 200 kW laser power with 60 MW arc jet provides unique test capability

### Avcoat: Arc jet flow on – Laser on Combined Heating





## **Weaving Technology**



## **3-D Weaving**



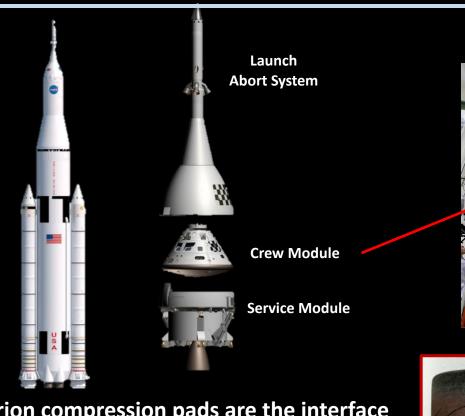




# 3-D Multi-functional Ablative TPS (3-D MAT) Enabling Orion

#### **Background: Orion Compression Pad**





Orion EFT-1 Vehicle

Compression Pad Locations (6)

Orion compression pads are the interface between Crew Module (CM) & Service Module (SM) Required to withstand:

- Launch & ascent structural loads
- Pyro-shock (CM/SM separation event)
- Earth re-entry (high heating, ablation)

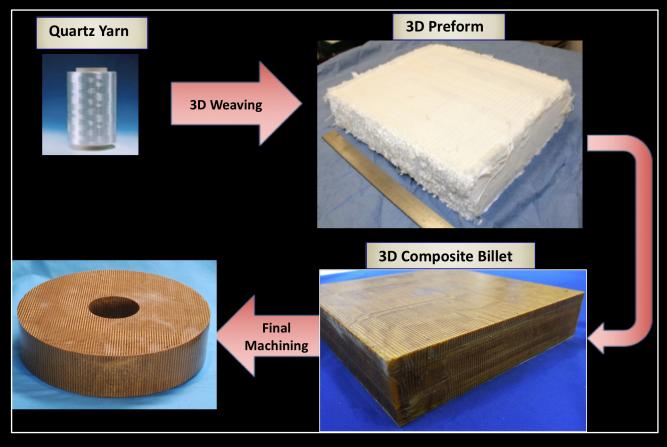




All EFT-1 carbon phenolic pads contained inter-laminar cracks post-flight



## Two Technical Challenges Need to be Addressed – Weaving and Resin Infusion



Bigger Challenge: Establishing partnership with industry (weaving and resining infusion), experimenting, testing, design assessment and demonstrating the capability for mission adoption in less than 36 months and under \$3M.



## Arcjet testing of 3-D MAT



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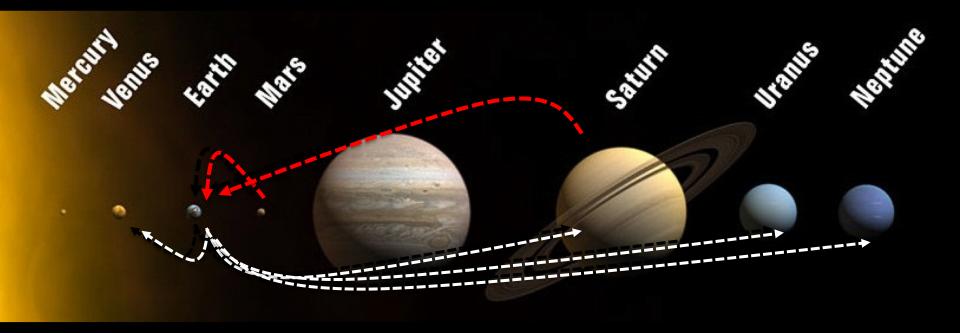
## 3-D Multi-functional Ablative TPS (3-D MAT)





## Heatshield for Extreme Entry Environment Technology (HEEET)

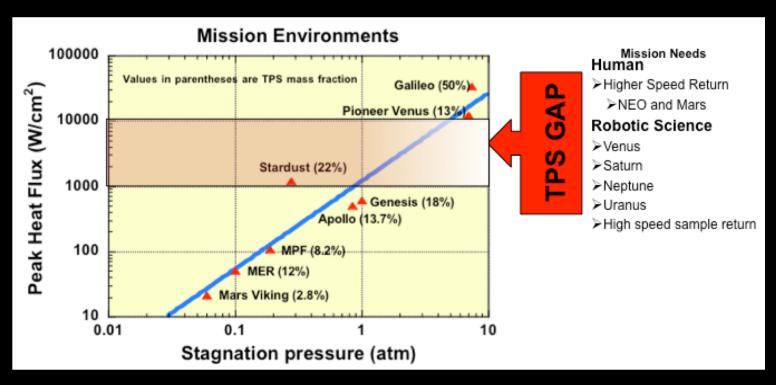




## NASA

## **Enabling Missions in the Coming Decades**

- Decadal Survey Recommended Missions (NF) require:
  - Revival of heritage carbon phenolic (HCP), or
  - Development of advanced TPS that is superior in performance to HCP, as robust as HCP and sustainable in the longer term

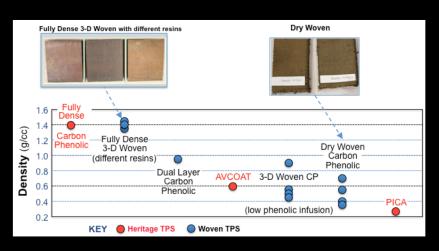


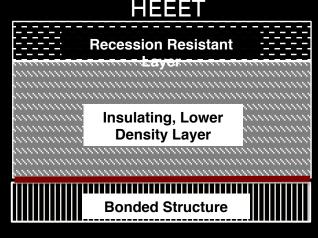
**Current ablative TPS State of the Art** 

## NASA

#### **3-D Woven TPS and HEEET**

- 3-D Woven TPS, not a single system but a family:
  - Woven TPS: An approach to the design and manufacturing of ablative TPS by the combination of 3-D weaving that allows precise placement of fibers in an optimized manner and resin infusion if needed



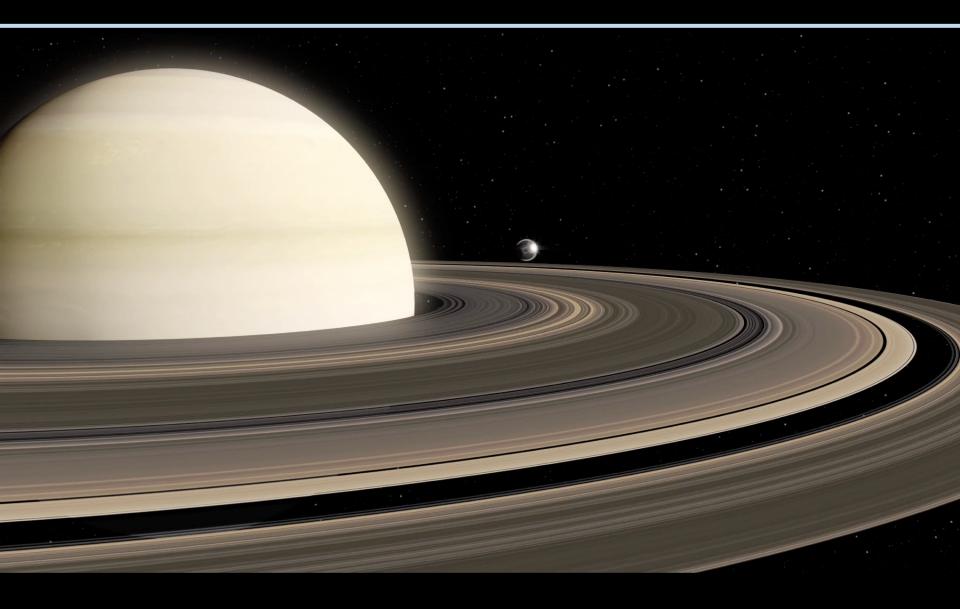


- System/Manufacturing
  - Molding flat panels
  - Seams
  - Resin Infusion at scale
- Integration with aeroshell sub-structure



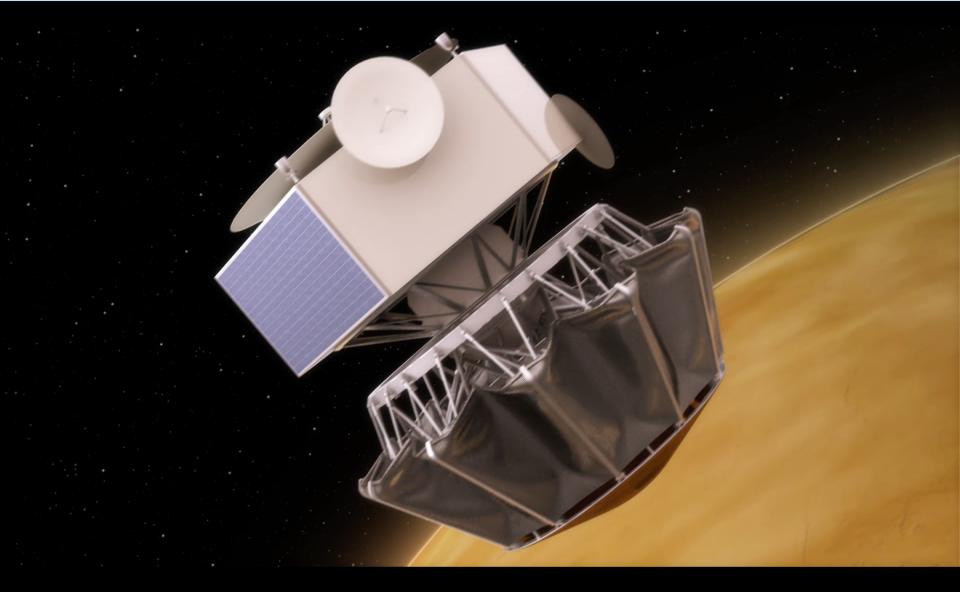
## HEEET





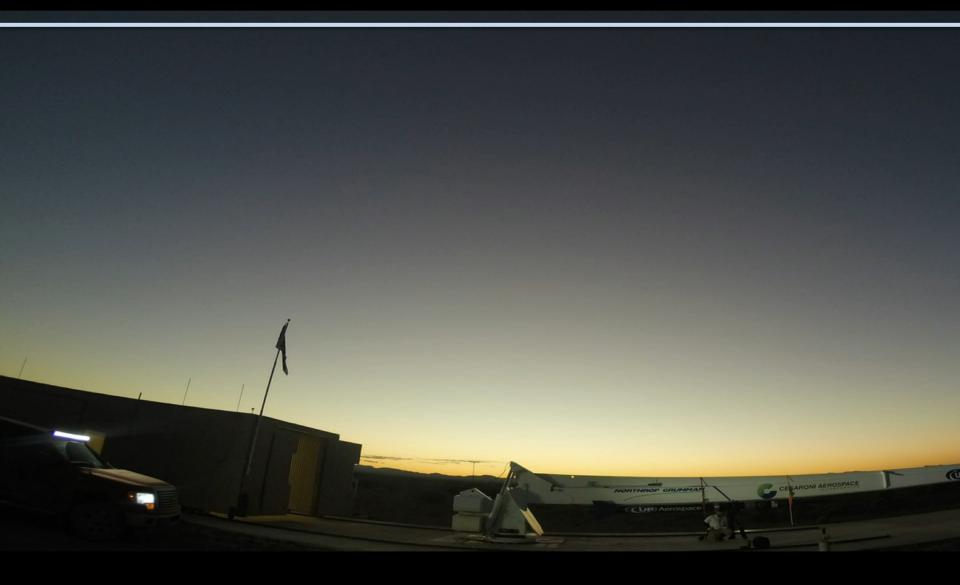
## **ADEPT**





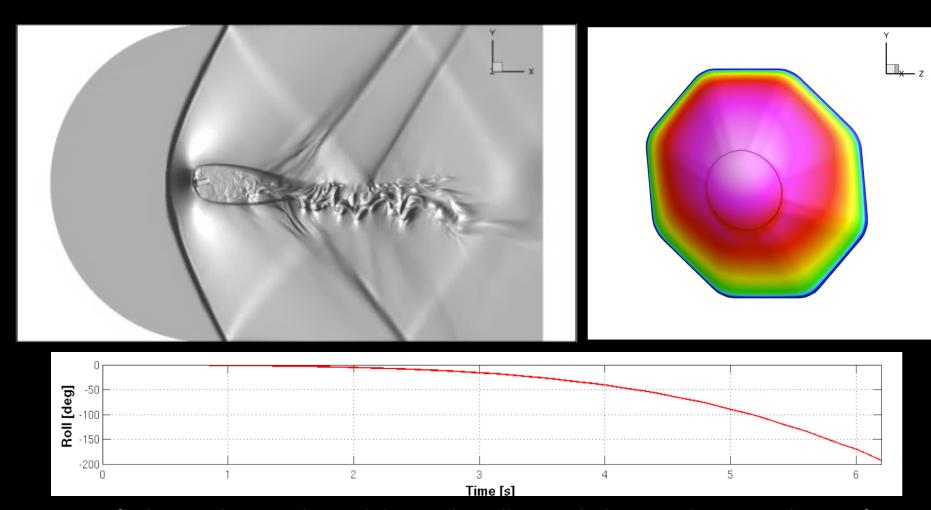


#### **ADEPT Flight Test**



## NASA

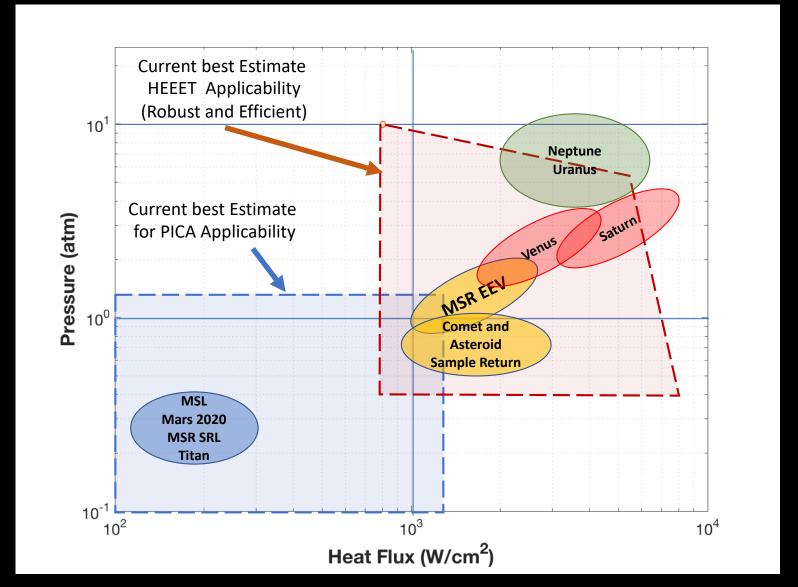
#### **SR-1 Flight Trajectory Simulations - Free Flight CFD**



- Free-flight simulations showed damped oscillations below ~15deg in total a.o.a. for high-altitude portion of trajectory
- Undamped oscillations up to 20deg in total a.o.a. for lower portion of trajectory

# NASA Ames Invented PICA and HEEET Adequate to Enable all Future Science Missions







## Thank you for the Privilege of your time

**Questions?**