Abstract—The Space Exploration Synthetic Aperture Radar (SESAR) is an advanced P-band beamforming radar instrument concept to enable a new class of observations suitable to meet multiple Decadal Survey science goals for planetary exploration. The radar is capable of providing unprecedented surface and near-subsurface measurements at full polarimetry and fine (meter scale) resolution, and achieves beam agility through programmable waveform generation and digital beamforming. The radar’s highly flexible modular architecture employs a novel low power, lightweight design approach to meet stringent planetary instrument requirements, all while minimizing cost and development time.

Keywords—P-band SAR; Digital Beamforming; MIMO radar.

I. INTRODUCTION

The Space Exploration Synthetic Aperture Radar (SESAR) is a “next generation” P-band (70 cm wavelength) radar instrument with capabilities beyond current planetary instruments [1] and ideally suited to meet a number of Decadal Survey Science Goals [2], [3]. The instrument’s operation, characterized by full polarimetry, high resolution (< 6 m), and programmable beams, have the potential to provide unprecedented planetary surface and near-subsurface measurements. The instrument design is based on a modular, low power, lightweight architecture that allows the instrument to be configured and optimized for a specific planetary mission, such as a Mars or a Moon mission. Advanced features, such as multiple RF channels, programmable waveform generation, and digital beamforming provides SESAR with enhanced capabilities to help answer key questions in planetary science.

Multiple decadal survey science goals require fine resolution views of subsurface stratigraphy, and the ability to expose bedrock and search for buried features that hold clues about the geologic history. For example, locating habitable regions, finding water, and determining the hydrology and cryosphere evolution is a primary goal of the Mars exploration.

SESAR’s long wavelength signals would penetrate through meters of material, image buried surfaces at fine spatial resolution and full polarimetry, and provide information to identify signatures of buried ice and water [2],[4].

On the Moon, SESAR would be able to image through meters of surface-covered regolith and provide information to characterize the near-surface stratigraphy and geology, crucial in the understanding of lunar processes and in the identification of landing sites for future lunar missions. SESAR’s fine resolution mapping and polarimetry would also provide details about the volcanic processes that built the lunar mare and to locate and track lava tubes [5], important for both science and future exploration purposes.

II. INSTRUMENT ARCHITECTURE

SESAR employs a multiple-input multiple-output (MIMO) and modular approach that distributes the radar systems into instrument panels composed of “smart” active subarrays, as illustrated in Figs 2 and 3. The radar architecture is fully programmable and capable of multi-mode radar operation.
including polarimetric SAR imaging, nadir SAR altimetry, and scatterometry. Some of other advanced programmable features include single, dual, or full polarimetry; multi-look angle data collection; selectable resolution and swath width; digital beam steering (no moving parts); and beam pattern control; among others. This radar design is based on the successful P-band EcoSAR and L-band DBSAR airborne radars that were developed at the NASA Goddard Space Flight Center [6], [7]. Table 1 lists some of SESAR’s main characteristics.

SESAR will use a distributed digital and RF electronics architecture that implements advanced waveform modulation techniques to provide the full beam steering agility while significantly reducing the system power consumption. Under the technology-development program awarded by NASA’s Planetary Instrument Concepts for the Advancement of Solar System Observations program (PICASSO), the SESAR team is developing an innovative Frequency Domain Multiplexing (FDM) techniques to reduce the mass and power consumption of the Radar Digital Unit (RDU). Through other R&D efforts, the SESAR team is also developing innovative techniques to reduce the mass of the antenna, and increase the efficiency of the RF electronics.

SESAR’s highly innovative architecture will allow synthesis of multiple antenna beams, simultaneously or interleaved, enabling the implementation of non-conventional imaging that can overcome fundamental limitations of conventional radar systems [8], [9], [10]. Some of its benefits include an increase in the measurement swath without reducing the received antenna gain, and the suppression of ambiguities or localized interference in the receiver signal by appropriate null-steering of the antenna pattern. The antenna gain, beam pointing angle, and sidelobe structure can be programmed to be controlled in real-time for specific tasks. Furthermore, multiple beams can be synthesized on both sides of the flight-track, as well as nadir, using a single nadir-looking antenna, thus increasing the coverage area.

Table 1. SESAR’s main characteristics

<table>
<thead>
<tr>
<th>Parameter</th>
<th>Value</th>
</tr>
</thead>
<tbody>
<tr>
<td>Frequency</td>
<td>435 MHz (P-band)</td>
</tr>
<tr>
<td>Wavelength</td>
<td>0.69 m</td>
</tr>
<tr>
<td>Max Bandwidth</td>
<td>100 MHz</td>
</tr>
<tr>
<td>Pulse Width</td>
<td>40 μs</td>
</tr>
<tr>
<td>Polarization</td>
<td>HH, VV, VH, HV</td>
</tr>
<tr>
<td>Noise Equivalent σ₀</td>
<td>&lt; -29 dB</td>
</tr>
<tr>
<td>Slant Range Resolution</td>
<td>1.5 m</td>
</tr>
<tr>
<td>Max. Transmit Power</td>
<td>300 W</td>
</tr>
<tr>
<td>Beam Steering Range</td>
<td>± 45 degrees</td>
</tr>
<tr>
<td>Antenna Panel Size</td>
<td>3.5 m x 2.1 m</td>
</tr>
<tr>
<td>Number of Antenna Panels</td>
<td>Configurable (e.g., 3 for Mars)</td>
</tr>
</tbody>
</table>

III. System Characteristics

SESAR concept permits the configuration of the radar instrument to meet specific mission requirements. This is possible by the modularity of its four main subsystems: A) the Antenna arrays, B) the RF Transceiver Modules (RTMs), C) the Radar Digital Units (RDUs), and D) the Frequency Domain Multiplexing electronics (FDM), as illustrated in Fig. 3. Each subsystem is designed to be reusable and to minimize mass, size, and power consumption.

A. Antenna Arrays:

SESAR employs arrays of antenna elements distributed over antenna panels, as shown in Figs 2, 3 and 4. The panels stow around the spacecraft during launch, and once deployed form a large array. The number of panels employed depend on the mission’s target and measurement requirements.

![SESAR architecture](image)
Fig. 4. Illustration of SESAR antenna panel architecture. The elements in each antenna form subarrays that enable cross track scanning. The linearly polarized antenna elements (H and V pol) are based on a proven stack-patch design.

After deployment, the antenna is oriented towards nadir, as illustrated in Fig. 2, permitting beamforming across the track over a large field of view (>45°). The antenna elements are based on a stacked-patch element design (see Fig 4) that permits horizontal (H) and vertical (V) polarization measurements over a 100 MHz bandwidth, and cross-polarization isolation greater than 30 dB [11]. The antenna elements are grouped as subarrays aligned in the flight direction, permit cross-track beam steering over a range of ±45 degrees.

B. Radar Transceiver Modules (RTMs):

SESAR’s RTMs build on the heritage of Goddard’s Digital Beamforming Synthetic Aperture Radar (DBSAR) and EcoSAR which miniaturized radar transceivers while improving system performance [6] [12]. SESAR’s RTMs employ surface mount components on printed circuit board (PCB) which yields a compact and robust design (see Fig. 5). The RTMs were designed for high efficiency and light weight. Ongoing efforts with the University of Oklahoma seek to increase the solid state power amplifiers (SSPA) to RF power while improving the efficiency to reduce the DC power.

The primary function of the transceivers is to amplify the transmit signal to the necessary power level to feed the antenna subarrays and to amplify the very low power RADAR return signal for processing. In addition, the transceivers provide accurate means to calibrate the transmitter and receiver independently, and to reject out-of-band interference to the receiver.

C. Radar Digital Units (RDUs):

Each subarray in the radar is controlled by an RDU, as shown in Fig. 6. The RDUs are capable of arbitrary waveform generation, data acquisition, and onboard processing (see fig. 5). These units are designed with Xilinx Kintex UltraScale FPGAs, low power digital-to-analog converters (D/As) and analog-to-digital converters (A/Ds). They synthesize based band signals with the appropriate phase and amplitude weights for transmit beam steering and side lobe control. Beamforming on receive is performed by coherently acquiring the raw complex data and processing them on board. They are also responsible for system synchronization, radar timing, and data transfers.

D. Frequency Domain Multiplexing (FDM) Electronics.

SESAR achieves further reduction in power consumption by employing a novel FDM technique [13]. The FDM approach reduces the number of DACs and ADCs and enables centralized waveform generation and data acquisition with reduced power and mass (see Fig. 5). This approach preserves full array steering capabilities while reducing the number of DACs and ADCs by a factor of 5 or higher, resulting in a significant reduction of SESAR's power consumption.

Fig. 5. SESAR’s compact and highly efficient Radar Transceiver Modules are distributed over the antenna panels driving each of the antenna elements.

Fig. 6. SESAR’s RDUs designed for low power consumption, employ Kintex 7 FPGAs, and low power digital-to-analog converters (D/As) and analog-to-digital converters (A/Ds). Significant power reduction is achieved by employing a Frequency Domain Multiplexing (FDM) Technique, which reduces the number of DACs and ADCs in the system.
IV. CONCLUSION

The SESAR instrument approach would be a first in planetary exploration. SESAR’s agile radar operation, modularity, and multimode operation, while using technology that can be optimized to produce the best possible data set for the individual science goals, will help pave the way for the next generation planetary radar systems. SESAR’s innovative approach to lower mass and power consumption will make these future missions feasible.

REFERENCES


