

OPTICAL TELESCOPE ASSEMBLY COST ESTIMATING MODEL

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INTRODUCTION

Parametric cost models can be used by designers and project managers to compare cost between major architectural cost drivers and allow high-level design trades; enable cost-benefit analysis for technology development investment; and, provide a basis for estimating total project cost between related concepts. The NASA Marshall Space Flight Center has developed a 5parameter cost model that explains 93% (Adjusted R2) of the cost variation in a database of 46 total ground and space telescope assemblies. This model can be used to estimate the most probably cost for the Habitable Exoplanet Telescope Assembly.

COST MODEL

OTA\$ = \$20M 30^(S/G) D^(1.7) $\lambda^{(-0.5)}$ T^(-0.25) e^(-0.027) (Y-1960)

Fit value 19.0 -0.47 -0.027430.4 1.69 -0.24 SE 6.0 0.09 0.03 0.006 1.6 0.07 5E-7 2E-21 1E-20 p-value 2E-18 0.002 4E-5 Model explains 93% (Adjusted R2) of the cost variation in the OTA database

where:

- = 1 for Space Missions and 0 for Ground Telescopes. S/G
- = Aperture Diameter D
 - = Wavelength of Diffraction Limited Performance (UV is more expensive)
 - = Operating Temperature (Cryo is more expensive)

DISCLAIMERS

Parametric cost models cannot predict the cost of a future system. They are backward looking. They describe how historical system costs vary as a function of cost estimating relations (CERs) - the most important factors that drive cost. The only forward prediction that a cost model can make is to provide guidance as to how the cost of a potential future system might scale relative to an existing historical system. Furthermore, a parametric cost model is only as good as its database. The fundamental challenge of cost modeling is developing a parametric model that includes the most important CERs. To do this requires a database with sufficient samples and data diversity to yield statistically meaningful results, and engineering judgment to interpret the results. Finally, cost models are statistical. They only provide an estimate of the most likely or 50% probable cost +/- an uncertainty.

DEFINITIONS

The MSFC multivariable model estimates the most likely cost for ONLY an Optical Telescope Assembly (OTA). Where an OTA is defined as the subsystem which collects electromagnetic radiation and focuses it (focal) or concentrates it (afocal) into the science instruments. An OTA consists of the primary mirror, secondary mirror, auxiliary optics and support structure (such as optical bench or truss structure, primary support structure, secondary support structure or spiders, straylight baffles, mechanisms for adjusting the optical components, electronics or power systems for operating these mechanisms, etc.). An OTA does not include science instruments or spacecraft. = Year of Development (Cost reduces by 50% approx. every 25 years)

NOTE:

- Model predicts ONLY the MOST PROBABLE or 50% cost.
- The prediction uncertainty is 45%.
- Thus to get the 84% most probably cost, multiply by 1.45X.

Residual Analysis

The below graphic illustrates the residual error of the model. Each column shows cost versus CER. The top row is the 'raw' database date. Row two is after normalizing cost by diameter. Row three is after normalizing cost by wavelength. Row four is after including temperature and YOD normalization. And, row five is after invoking ground to space multiplier.



DATABASE

The MSFC OTA database contains information on 46 different cost, programmatic and engineering parameters for 35 space missions with normal incidence optical telescopes or antenna; and 26 ground telescopes or radio antenna. The database consists of both conventional imaging telescopes and non-imaging systems such as spectroscopic missions, LIDAR or radio antenna.

The cost model is developed by regressing over 18 combinations of 8 cost estimating relations (CERs) for 46 OTAs (26 Space and 20 Ground). Four CERs were identified as key: Aperture Diameter, Wavelength of Diffraction Limited Performance, Operating Temperature and Year of Development.

SPACE TELESCOPES				GROU	GROUND TELESCOPES				
rev. 11.01.18	Effective Diameter [m]	Diffraction Limit λ [μ]	Operating Temperature [K]	Year of Development [year]	rev. 11.01.2018	Effective Diameter [m]	Diffraction Limit λ [μ]	Operating Temperature [K]	Year of Development [year]
Imaging					ЈКТ	1.00	1.00	270.00	1977
AFTA	2.40	0.78	284	1992	Commercial 1.00		0.50	300.00	2013
COM_0.7	0.70	0.50	283	1996	Starfire	3.50	0.53	273.00	1989
COM_1.1	1.10	0.65	283	2007	WIYN	3.50	0.42	263.00	1988
Herschel	3.50	80.00	80	2001	AEOS	3.67	0.85	273.00	1991
HST	2.40	0.50	294	1977	UKIRT	3.80	2.20	273.00	1974
IRAS	0.57	8.00	4	1977	SOAR	4.20	1.00	263.00	1997
JWST	6.20	2.00	50	2006	WHT	4.20	6.10	270.00	1981
Kepler	1.40	1.00	213	2001	DKIST	4.20	0.90	300.00	2011
MO / MOC	0.35	0.53	283	1986	MMT 6.5m replacement	6.50	1.60	262.00	1992
MRO / HiRISE	0.50	0.40	293	2001	Magellan 1	6.50	1.00	280.00	1994
OAO-2 / CEP	0.61	1.50	300	1962	Gemini 1	8.10	0.80	270.00	1994
OAO-3 / PEP	0.80	2.40	288.5	1963	Subaru	8.30	0.60	273.00	1988
Planck	1.70	300.00	40	2001	KECK 1	10.00	1.00	273.00	1986
Proprietary	2.40	0.60	300	2012	LBT	11.88	0.65	273.00	1997
Spitzer	0.85	6.50	5.5	1995	KECK-I&II	14.14	1.00	273.00	1986
WIRE	0.30	24.00	12	1995	HET	9.20	20.00	264.00	1994
WISE	0.40	2.75	17	2002	Commercial Radio	5.00	210000.00	300.00	2012
WMAP	2.10	1300.00	60	1996	SubMM Array Dish	6.00	300.00	300.00	1998
Non-Imaging					Green Bank Radio	100.00	6500.00	300.00	1991
ACTS	3.97	1950.00	263	1984					
Cloudsat	1.85	1300.00	250	2000					
GALEX	0.50	8.00	273	1998					
ICESat	1.00	8.00	283	1998					
IUE	0.45	3.50	273	1973					
MO / MOLA	0.50	15.00	283	1986					
OAO-B / GEP	0.97	5.00	289	1964					
SWAS	0.68	286.00	170	1993					

OTA COST ESTIMATION EXAMPLE

OTA cost can be estimated via two methods: using the model directly or using model to compare relative cost with other OTAs (i.e. Hubble & JWST).

For an OTA with:

- 4 m diameter
- 0.4 µm diffraction limit
- 270K operating temperature
- 2025 year of development

Direct Model Method

OTA(50% probable) = \$430M

 $= \$20M \times 30 \times (4)^{(1.7)} \times (.4)^{(-0.5)} \times (270)^{(-0.25)} \times e^{(-0.027)(2025-1960)}$

OTA\$ (84% probable) = \$620M

Diameter [meter] 0.46 6.35 WDLP [micrometer] 2.24 Temperature [K] -0.25 270 0.66 exp(YOD) -0.027 2025 2000 0.60 \$552 50% Predicted Cost [FY17 \$M] 0.40 85% Predicted Cost [FY17 \$M] \$801 HST HabEx Ratio \$530 Total Cost [FY17 \$M]

Relative Comparison Method

Total Cost [FY17 \$M]

JWST HabEx Ratio

\$1,380

Diameter [meter]	1.7	2.4	4	2.38
WDLP [micrometer]	-0.5	0.5	0.4	1.12
Temperature [K]	-0.25	294	270	1.02
exp(YOD)	-0.027	1973	2025	0.25
50% Predicted Cost [FY17 \$M]			\$354	0.67
85% Predicted Cost [FY17 \$M]			\$514	

Average Mission WBS Distribution



Average cost breakdown for 14 missions is: payload ~33%, spacecraft ~33%, labor ~25%; (Instruments + Spacecraft ~ 50%).

Analysis based on Cost Analysis Data Requirements (CADRe) reports for 14 missions: CALIPSO, CLOUDSAT, GALEX, ICESAT, JWST, Kepler, LANDSAT-7, Spitzer, STEREO, SWAS, TRACE WIRE, WISE and WMAP.

