Calibration Program for the Ocean Color Instrument (OCI) on the Plankton, Aerosol, and Cloud ocean Ecosystem (PACE) mission

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OCI calibration overview

- The basic product measured by OCI is the top-of atmosphere (TOA) radiance at different wavelengths
- Three types of calibration/characterization are necessary for ocean color processing:
 - Prelaunch calibration/characterization (absolute/spectral calibration and image artifacts)
 - On-orbit calibration (solar diffuser and lunar measurements)
 - Vicarious calibration (in-situ measurements of water-leaving radiance)

OCI calibration overview

Artifact	Measured	Measured	Applied during L1	Applied during L2
	Prelaunch	Postlaunch	processing	processing
Absolute gain	Instrument level,	Solar calibration	Yes (calibration	
(K1)	IVAC	and vicarious calibration ¹	equation)	
Temporal	Instrument level,	Solar and lunar	Yes (calibration	
response (K2)	reduced accuracy	calibration	equation)	
Temperature	Instrument level,	Solar and lunar	Yes (calibration	
correction (K3)	TVAC	calibration ²	equation)	
Response vs. scan	Instrument level,	Verification with	Yes (calibration	
angle (RVS) (K4)	ambient	ocean color	equation)	
		products		
Linearity (K5)	Instrument level,	Solar calibration ³	Yes (calibration	
	TVAC		equation)	
Tilt angle (K6)	Spacecraft level	Verification with	N/A	
	(verification only)	ocean color		
		products		
Polarization	Instrument level,	Verification with		Yes (atmosphere
sensitivity	ambient	ocean color		polarization)
		products		
Stray light	Instrument level,	Verification with		Yes
sensitivity	ambient	lunar cal.		
Crosstalk	Instrument level,	Verification with	Maybe	Yes
	ambient	lunar cal.		
Relative spectral	Instrument level,	Verification with	N/A (part of K1	Yes(atmospheric
response	TVAC	solar calibration	calculation)	correction)
		(Fraunhofer and		
		atm. abs. lines)		
Offset (DN0)	Every scan	Every scan	Yes	

¹Vic. Cal.: Visible and some NIR bands only
²If seasonal variations are observed in K2
³New technique developed for OCI (dim diffuser)

Ground support equipment: 20inch integrating sphere (8inch exit aperture)

Integrating Sphere

Halogen External Source

Turning Mirror and VAD

(HES) Lamp 150W

Attachments:

Various light sources (halogen, plasma, EQ-400, attenuators)

and sensors (SR4500, Ocean Optics, FRMS, Filter wheel (Si and

InGaAs)

Plasma Internal Lamp 250W Mercury With Argon Manual Pen-ray Atten HES 150W FRM EQ-400 S -HES 35W Motorized Filter Wheels HES HES 150W 75W Motorized IGA Detector SI Detector Filter Wheels 80/20 Integrating Red labels indicate sphere ports Sphere Rack SR-4500 **Ocean Optics** (320 to 340nm Only) Α

All light sources will be calibrated by NASA GSFC Code 618 Calibration Facility



OCI Solar Calibration Assembly

Preliminary design drawings

3 diffuser surfaces: 2 QVD (daily/monthly gain tracking) and 1 dim diffuser (linearity)





Linearity: hyperspectral CCDs accumulate several readout cycles (dim solar diffuser reflectance is lowest radiance, upper limit is saturation)

Calibration Equation for each channel

 $L_{m}=\ K_{1}\ ^{*}\ K_{2}(t)\ ^{*}\ (1-K_{3}\ ^{*}(T-T_{ref}))\ ^{*}\ K_{4}(\theta)\ ^{*}\ K_{5}(dn,\ T)\ ^{*}\ K_{6}(\omega)\ ^{*}\ dn$

 L_m = radiance measured in a hyperspectral or SWIR band

 K_1 = absolute gain factor

 $K_2(t)$ = relative gain factor as a function of time t

 K_3 = temperature correction factor

T = Instrument temperature measured at relevant location (electronics? housing? T.b.d.)

T_{ref} = Reference Temperature (used during TVAC prelaunch characterization, close to expected on-orbit temperature)

 K_4 = response versus scan

 θ = scan angle (usually replaced by science pixel number per scan)

 $K_5 =$ nonlinearity factor

 $K_6(\omega)$ = correction for tilt position ω (+/- 20°)

DN = digital number measured at a certain θ

 DN_0 = average of the digital numbers measured during dark current collection (average of ~40 numbers, once per scan)

 $dn = DN - DN_0$

Note: out-of-band, polarization and straylight/crosstalk correction are handled later in the processing stage (need other information, such as surrounding radiances for straylight, amount of rayleigh/aerosol/glint for polarization)

Absolute calibration K1: 3 uncertainties

$L_{m} = K_{1} * K_{2}(t) * (1-K_{3}*(T-T_{ref})) * K_{4}(\theta) * K_{5}(dn, T) * K_{6}(\omega) * dn$

•K1 is a single number per band and mirror side, with units [radiance/dn]

- •Prelaunch: GLAMR will provide absolute calibration, better than 0.5% accuracy
- •Initial on-orbit calibration: solar diffuser will provide absolute
- calibration with <2% uncertainty
- •Vicarious calibration will provide absolute calibration for most
- bands with 0.1% uncertainty after sufficient number of
- matchups have been acquired

Temporal calibration K2

 $L_{m} = K_{1} * K_{2}(t) * (1-K_{3}*(T-T_{ref})) * K_{4}(\theta) * K_{5}(dn, T) * K_{6}(\omega) * dn$

•Daily solar diffuser measurements will provide temporal trending

•A function of time (e.g. exponential, polynomial) will be fitted to the daily measurements

A monthly solar diffuser (limited exposure) will provide correction to degradation of reflectance of daily solar diffuser
After more than 2 years, lunar measurements will be used for temporal trending

•K2 uncertainty achieved with SeaWiFS lunar measurements:

0.13% (Eplee et al., Applied Optics, Vol. 51, Issue 36, 2012)

• K2 uncertainty allocation for OCI: 0.17%

Linearity correction K5

 $L_{m} = K_{1} * K_{2}(t) * (1 - K_{3} * (T - T_{ref})) * K_{4}(\theta) * K_{5}(dn, T) * K_{6}(\omega) * dn$

A monthly solar diffuser (dim target with reflectance of about 2%) will provide linearity correction via special OCI mode
OCI can hold charge for several cycles, testing the linearity of the electronics (not the detectors)

•Linearity will be evaluated at multiples of 2% in reflectance (2%, 4%, 6%,...100%)

Solar Diffuser Reflectance Degradation

•The monthly solar diffuser (limited exposure) will provide correction to degradation of reflectance of daily solar diffuser, but it will degrade as well

•The degradation pattern of the daily solar diffuser will be used to model the degradation of the monthly solar diffuser (heritage: MERIS, ozone instruments (OMI))

If the degradation of the monthly solar diffuser is smaller than 0.6% over the mission life (or 2 years), an uncertainty of 0.1% can be achieved with the solar diffuser measurements alone (Meister, Onorbit trending of solar diffuser reflectance, PACE memo, 2017)
Expected degradation for Quartz-Volume Diffuser: 0.15% (worst wavelength (350nm), based on on-orbit data from OMI/Aura)
Daily solar diffuser will be used to monitor short term changes.
Expected degradation at 350nm: 6.7%

Lunar Calibration

Lunar calibration background:

- Stable exo-atmospheric radiometric source with light levels comparable to TOA Earth observations.
- Moon used as reference by SeaWiFS, MODIS (2), and VIIRS.
- Observations require geometric correction for instrument-Moon and Sun-Moon distances, phase angle, libration angles.
- Frequency of observation: Twice per month (before and after full phase) over a limited range of phase angles (7deg +/- 0.5deg).
- Limitations:
 - Will require image oversampling correction.
 - Inherent scatter in observations (1-2%).
 - Multi-year time series required to identify radiometric trends.

Geometric corrections:

- Complication: Heterogeneous albedo distribution over the lunar surface
- Corrections provided by USGS ROLO Lunar Photometric Model
- ROLO Model used as reference by most Earth-observing instruments

Summary

- OCI will start a rigorous calibration program in June 2019 with the ETU (Flight Unit: summer 2020)
- Goal is to minimize uncertainties due to image artifacts described here in order to achieve overall radiometric uncertainty of 0.5% (excluding absolute calibration)
- OCI will provide excellent temporal stability over mission life time (2 solar diffusers, lunar measurements twice a month)
- Linearity will be verified on-orbit with dim solar diffuser (new approach)