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Functional Fault Model Development Process to Support Design Analysis and Operational Assessment

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Presentation Outline

- Introduction
 - Purpose
 - Motivation
 - What is a Functional Fault Model
- FFM Development Process
 - Phase 1: Knowledge Acquisition
 - Phase 2: Conceptual Design
 - Phase 3: Implementation & Verification
 - Phase 4: Application
- Concluding Remarks



Purpose of paper

- To characterize and document the current process used by NASA to develop functional fault models (FFMs)
- To identify new technologies and capabilities that contribute to an improved process.

Motivation for the paper

- Process has evolved over past 10 years with push to support development of new NASA human-rated space systems
- Modeling guidelines, best practices, and software tools have been developed to substantially improve:
 - The efficiency of the FFM development and verification process
 - The utility and impact of FFM applications
- Benchmark for future FFM development efforts as the process continues to evolve



What is a Functional Fault Model?

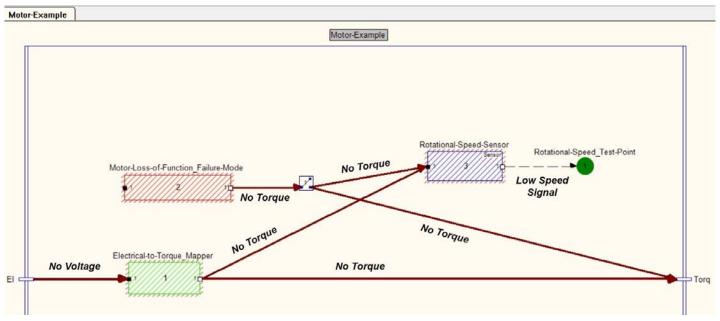
- Directed graph representation of failure effect propagation paths within the system architecture
- Developed to address limitations of traditional methods
- Initial models can be qualitative supporting requirements verification early in system design process

 Early models may be evolved to support real-time failure mode detection and isolation

 TEAMS Designer® software currently Mechanical Mechanical Sensor used for FFM development Element 1 Element **Effects Electrical** Fluid Electrical Fluid Component Effects Effects Component FM₂ FM₃ Mechanical Mechanical Sensor Element 2 Effects Element



Functional Fault Models in TEAMS

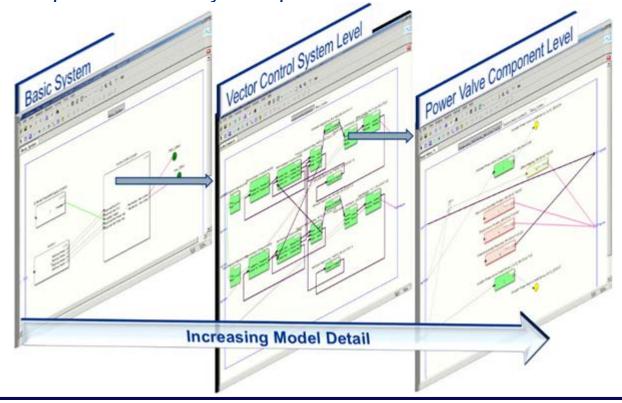


- Modules represent systems, assemblies, components
- Failure Mode Modules contain the qualitative failure information
- Mapper Modules represent the nominal transition of the failure effects being propagated
- Test-Points represent the observation points of the system (typically associated with sensors)
- Tests detect specific failure effects



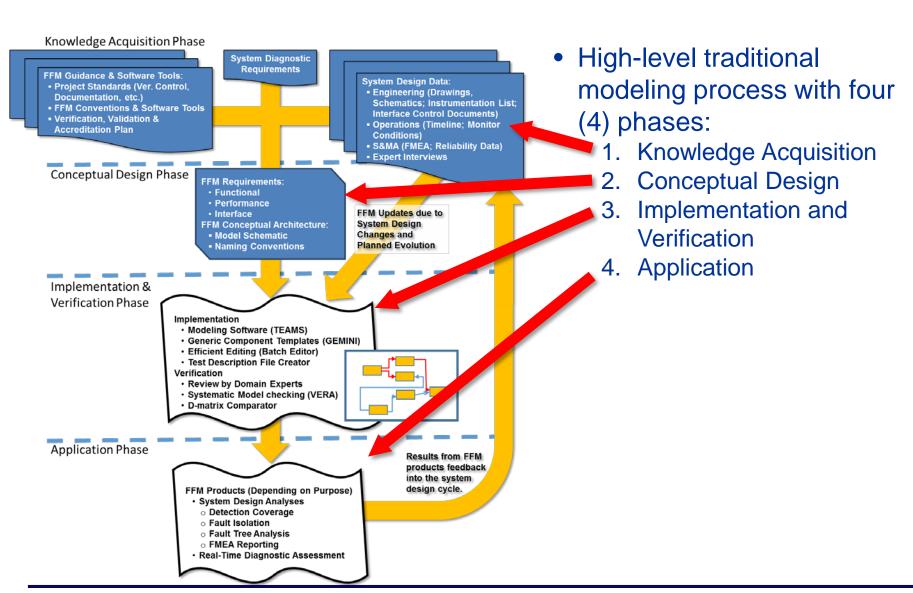
Functional Fault Models in TEAMS

- Hierarchical modeling capability:
 - Supports a model structure that reflects a hierarchical decomposition of system hardware & software
 - Facilitates portioning of large complex models into smaller models for implementation by multiple individual modelers



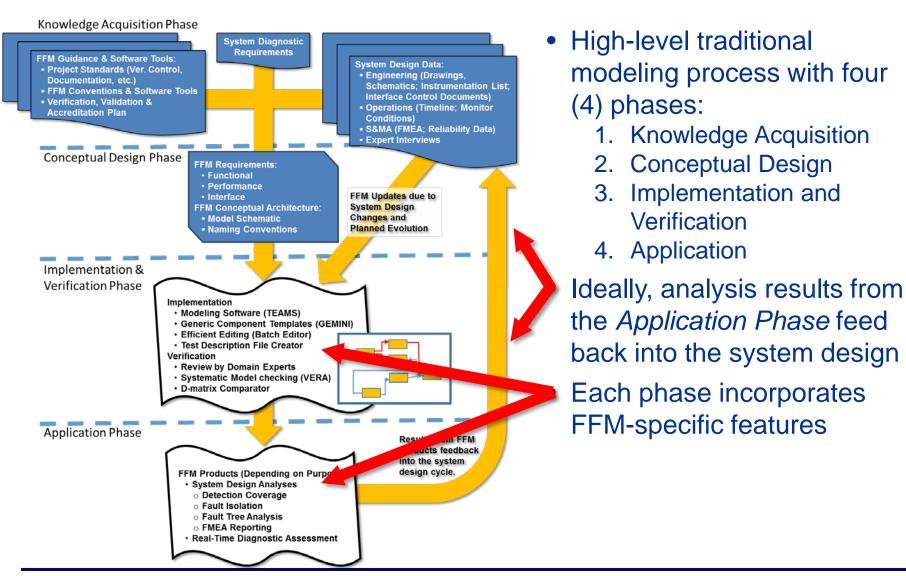


FFM Development Process: Overview



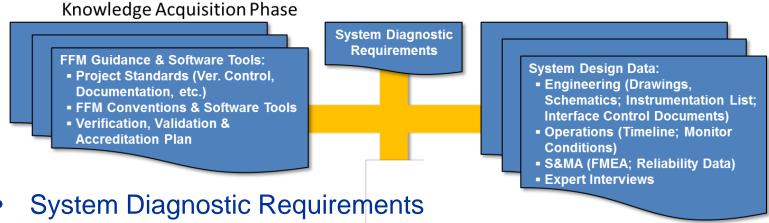


FFM Development Process: Overview





Phase 1: Knowledge Acquisition



- System-level requirements impact all phases of the FFM dev. process
- Examples: Abort conditions, launch commit criteria, line replaceable units
- System Design Data
 - Information that defines the system design and operation
 - Examples: Engineering drawings and reports, concept of operations, Failure Modes and Effects Analysis (FMEA)
- FFM Guidance & Software Tools
 - Information needed to implement a model that informs the Conceptual Design, Implementation & Verification, and Application phases
 - Examples: modeling conventions, model VV&A plan
- Establish System Breakdown Structure (SRS) & other databases



Phase 2: FFM Conceptual Design

Conceptual Design Phase

FFM Requirements:
 • Functional
 • Performance
 • Interface
FFM Conceptual Architecture:
 • Model Schematic
 • Naming Conventions

- FFM Requirements
 - Flowed down from system diagnostic requirements, FFM conventions and practices
 - Functional: Failure modes, test points, test logic
 - Performance: Time to detect/isolate failures, False positive/negative rates

- Interface: FFM-to-FFM, system to FFM to key decision makers on ground or vehicle
- FFM Conceptual Architecture
 - System Operational Profile
 - Model Schematic/Structure
 - Naming Conventions

NASA

Phase 2: Conceptual Design

FFM Rqmts: Modeling Conventions & Practices

- Approved by NASA's SLS, Orion, and Ground Systems FFM communities.
- Documents FFM best practices of all three communities.
- Benefits:
 - Model elements and sub-models have consistent look and feel
 - Improves human understanding
 - Enables more efficient integration of independently developed FFMs
 - Facilitates development of the interfaces needed for integration of FFMs with realtime systems
 - Improves traceability of model features back to source documents



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ADVANCED GROUND SYSTEMS MAINTENANCE (AGSM) PROJECT

NASA GROUND SYSTEMS AND LAUNCH VEHICLES

TESTABILITY ENGINEERING AND MAINTENANCE

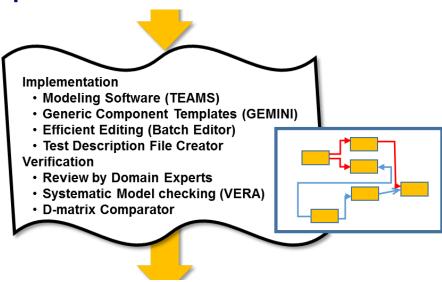
SYSTEM (TEAMS)

MODELING CONVENTIONS AND PRACTICES



Phase 3: Implementation & Verification

Implementation & Verification Phase



Start Modeling

Create/Revise FFM

- TEAMS Designer software
- NASA-developed software tools
 - Batch Editor
 - GEneric Model INstantlator (GEMINI) software
 - Test Description File Creator

Verify FFM

- Review by domain experts
- NASA-developed software tools
 - VERification Analysis (VERA) Tool
 - D-matrix Comparator

Tightly Coupled Lower-Level Processes



Phase 3: Implementation & Verification

NASA-Developed Tools

Batch Editor

- Includes model query and update commands to efficiently make broad systematic changes to FFMs (not available in TEAMS)
- Commands cover a wide variety of TEAMS modules & features
- Graphical or command line user interfaces
- Used by several other NASA-developed FFM tools to extract model information
- GEneric Model INstantItation (GEMINI) Tool
 - Supports the use of generic model libraries
 - Generates component-specific FFMs by adding user-provided component data to generic component models
- Test Description File Creator
 - Aligns real-time system data to FFM tests (mode dependent)
 - Defines thresholds—results in quanitative diagnostic assessment



Phase 3: Implementation & Verification

NASA-Developed Tools

- VERification Analysis (VERA) Tool
 - Checks model for adherence to NASA FFM conventions & practices
 - Reads model information into MS Excel Workbook
 - Analyzes model in four areas:
 - Technical
 - Practices & Conventions
 - Cosmetic
 - Informational content
 - Generates detailed reports that identify non-compliant FFM features
 - Provides scores to support accreditation of the model for operational use.
- D-Matrix Comparator
 - Reports differences between D-matrices from two different FFMs
 - Useful for regression testing to ensure minor model checks reflected in results



Phase 4: Application

Application Phase



FFM Analysis Products

- Failure Detectability Report
 - Analyzes FFM for detected / undetected failure modes
 - Verifies detection coverage rqmts.
- Test Utilization Report
 - Analyzes FFM for used/not used tests (sensors)
 - Supports sensor selection/buy-in
- Fault Isolation Report
 - Analyzes failure mode uniqueness/ ambiguity
 - Verifies rqmts for algorithms used to detect failure effects

- Component Isolation Report
 - Analyzes isolation of failure modes to user-defined components
 - Verifies requirements for line replaceable units
- FMEA Report
 - Uses data embodied in FFM to generate a report containing failure mode description data and detection capabilities from FFM



Phase 4: Application

Application Phase



Moving FFM from analytical use to real-time:

- Interface policies & software for generating FFM input from the real-time data
 - Handling dynamic data
 - Loss of data
 - Align FFM tests with software that processes real-time data

Real-time Diagnostic Assessment:

- Provide a list of failure modes, components, and sensors that align to the latest test detection
- Textual information traceable to design and FFM documentation
- Used by decision makers in flight and on ground.



Concluding Remarks

- This paper presented an iterative, four (4) phase process to support the development of FFMs.
- Special emphasis was placed on key approaches, capabilities, and tools that are unique to FFMs.
- The process has proved beneficial to recent systems engineering assessments under NASA's Ares I, Space Launch System, and Ground Systems Development and Operations Programs.
- Continued evolution of the process is anticipated as:
 - Current capabilities mature,
 - Additional capabilities are developed,
 - All capabilities are demonstrated in future flight and ground systems.



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