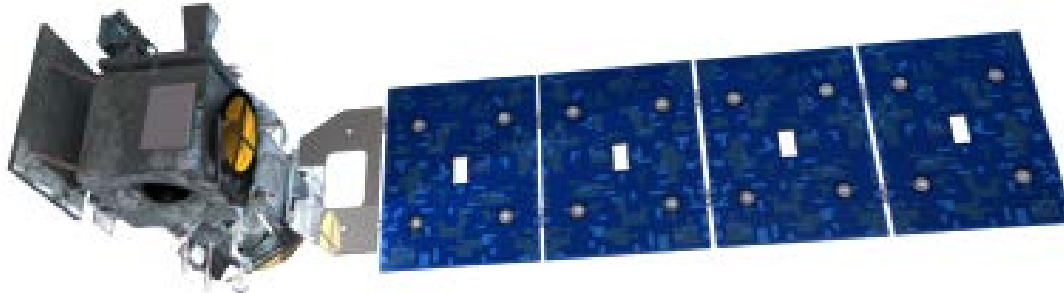




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***The Advanced Topographic Laser Altimeter System (ATLAS)
for NASA's Ice Cloud and Land Elevation Satellite-2 (ICESat-2)***

Design, Development, and On-Orbit Performance

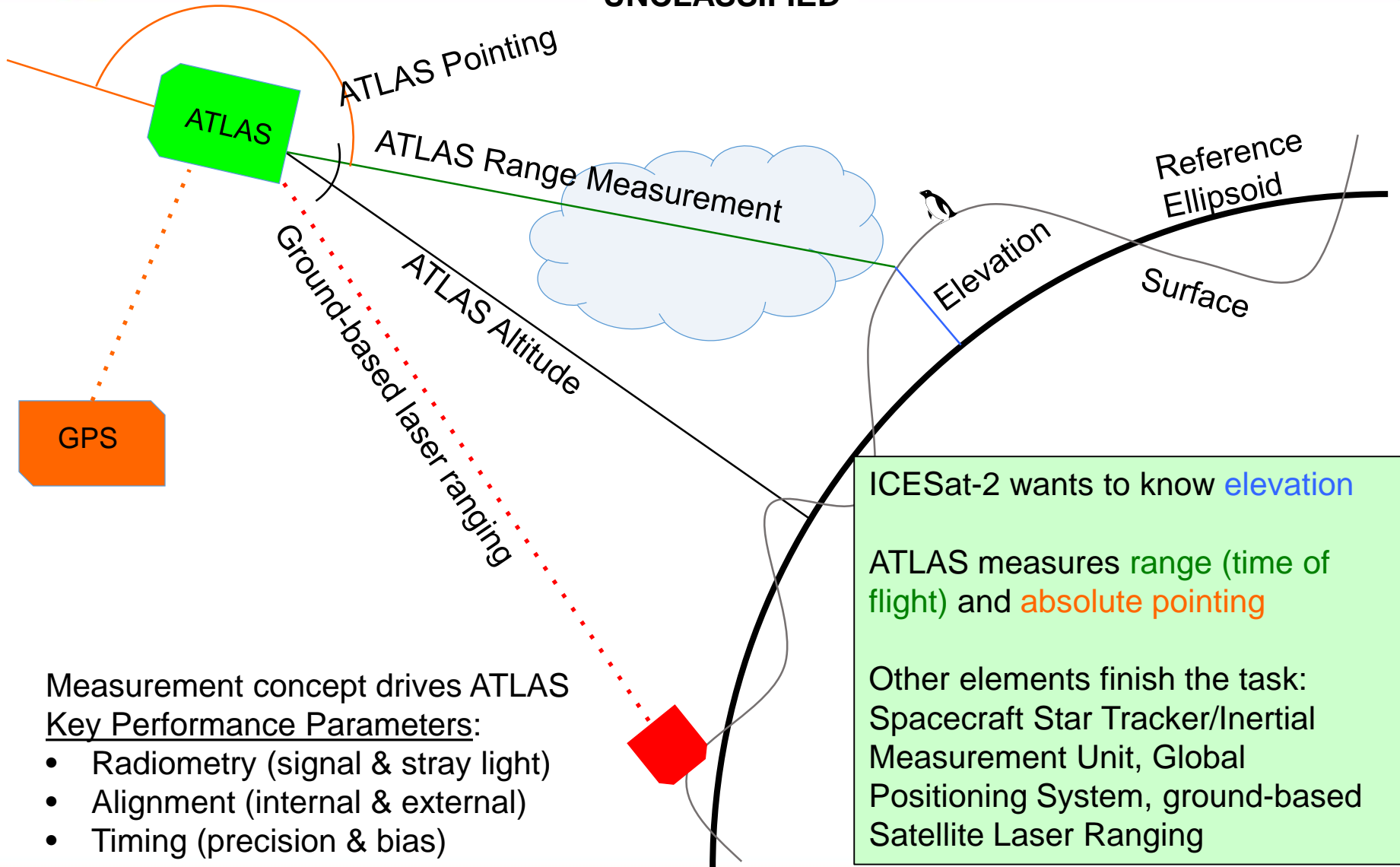
***John Cavanaugh
NASA Goddard Space Flight Center
Instrument/Payload Systems Engineering Branch
Code 592***

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ATLAS Measurement Concept

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Measurement concept drives ATLAS
Key Performance Parameters:

- Radiometry (signal & stray light)
- Alignment (internal & external)
- Timing (precision & bias)

ICESat-2 wants to know **elevation**

ATLAS measures **range** (time of flight) and **absolute pointing**

Other elements finish the task:
 Spacecraft Star Tracker/Inertial Measurement Unit, Global Positioning System, ground-based Satellite Laser Ranging

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ICESat Background



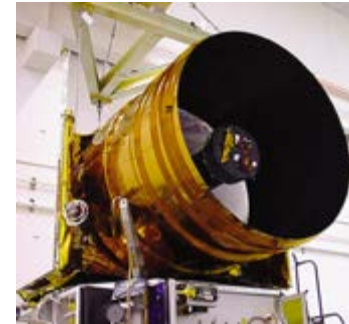
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ICESat

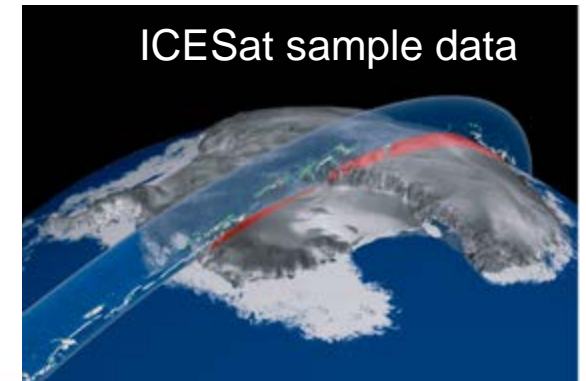
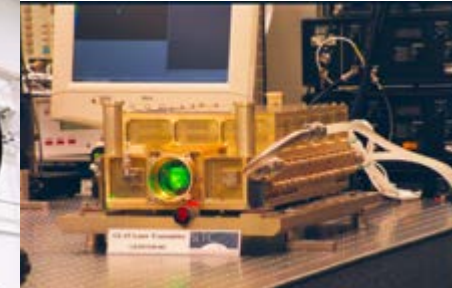
- Launch : January 13, 2003
 - Decommissioned August 17, 2010
- **Instrument : Geoscience Laser Altimeter System (GLAS)**
 - Diode Pumped-Nd:YAG Solid-state laser transmitter
 - 4-ns pulse
 - 40-Hz Pulse Repetition Frequency
 - 1064-nm altimeter
 - 70-m footprint
 - 532-nm atmospheric sounder
 - 1-m telescope receiver
 - Silicon –Avalanche Photodiode (Si-APD) altimeter detector
 - Analog waveform capture
 - Single Photon Counting Modules (Geiger-mode Si-APDs) for atmospheric detectors
 - Photon-counting atmospheric profiles
- Despite laser issues, ICESat collected a valuable dataset used to evaluate global ice sheets
 - This set the stage for ICESat-2



GLAS Instrument



GLAS Laser



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• Transmitter

- Solid-state Nd:YVO₄ Master Oscillator Power Amplifier (MOPA) Architecture
- 10-kHz pulse rate
- 1.2-ns output pulses
- 532.292-nm wavelength, tunable ± 15 picometers
- Diffractive beam splitter
 - 6 signal beams : 3 strong, 3 weak

• Receiver

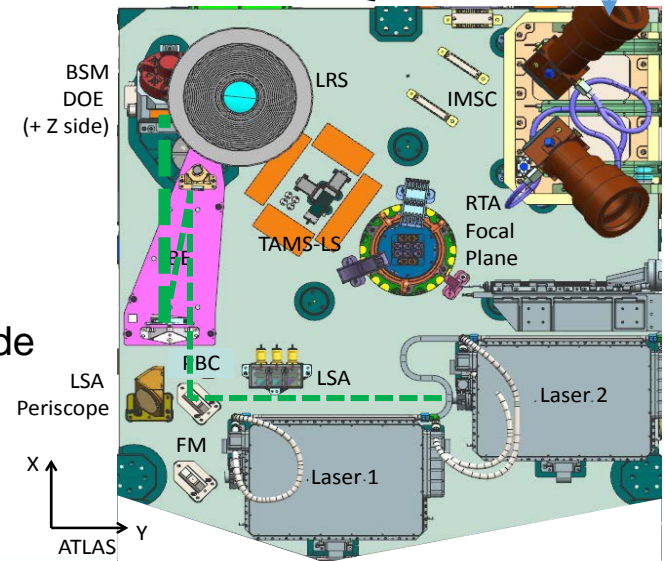
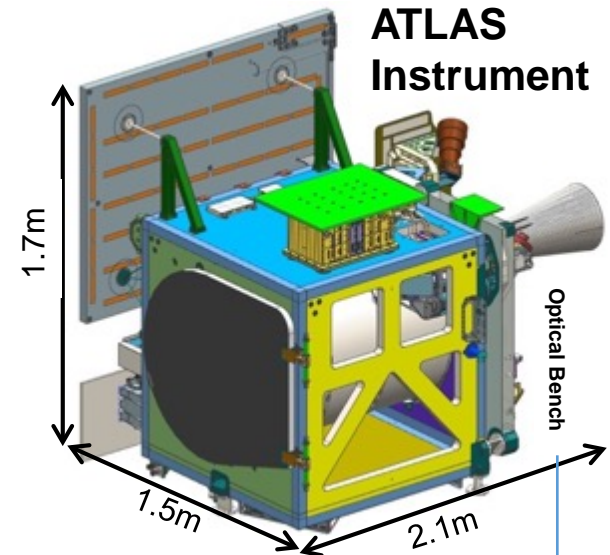
- 80-cm reflector telescope
- 30-pm bandpass filter
 - Temperature-tuned to match transmitter
- Single photon detection w/ 16-pixel photomultipliers

• Timing

- 100-MHz Ultra Stable Oscillator w/ gate-delay Field Programmable Gate Array (FPGA)
- Range tracking algorithm utilizing on-board Digital Elevation Model/Digital Relief Model and position/attitude updates from spacecraft

• Alignment

- Active Transmit/Receive co-boresight alignment
 - Alignment Monitoring and Control System (AMCS)



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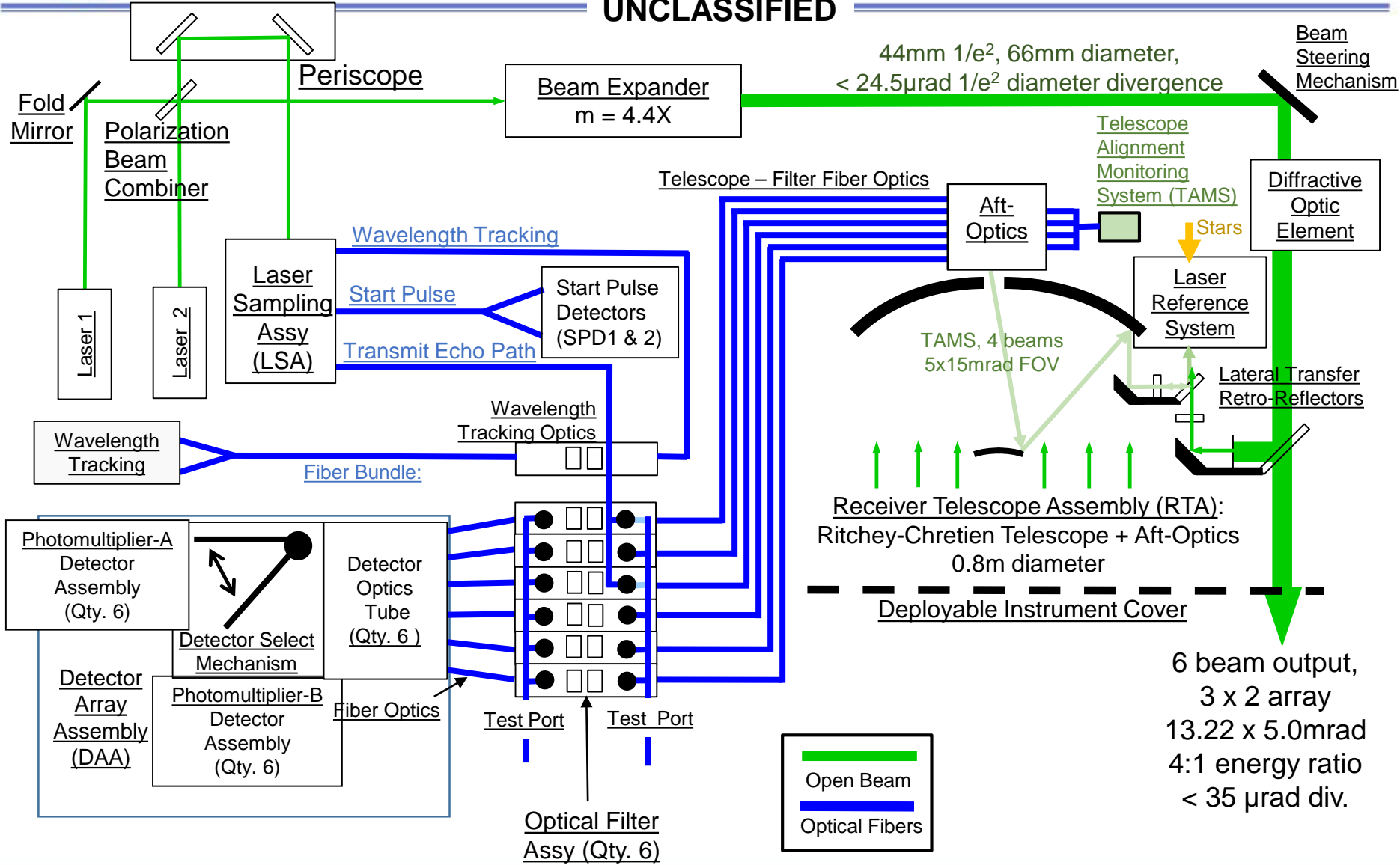
ATLAS Optical Bench



ATLAS Optical Block Diagram



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ATLAS Transmit Path

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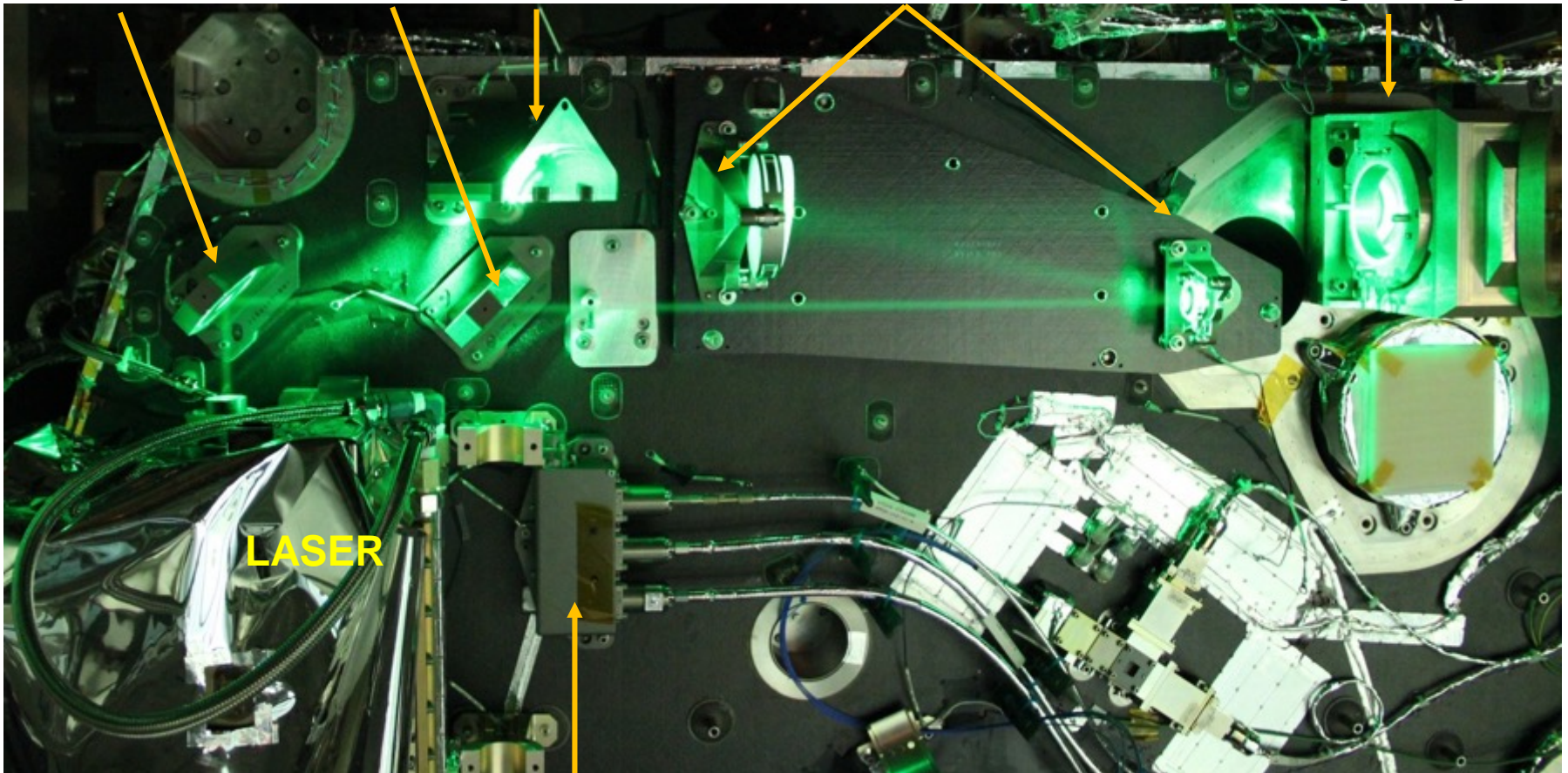
FOLD
MIRROR

POL. BEAM
COMBINER

LSA
PERISCOPE

BEAM
EXPANDER

BEAM
STEERING
MECHANISM



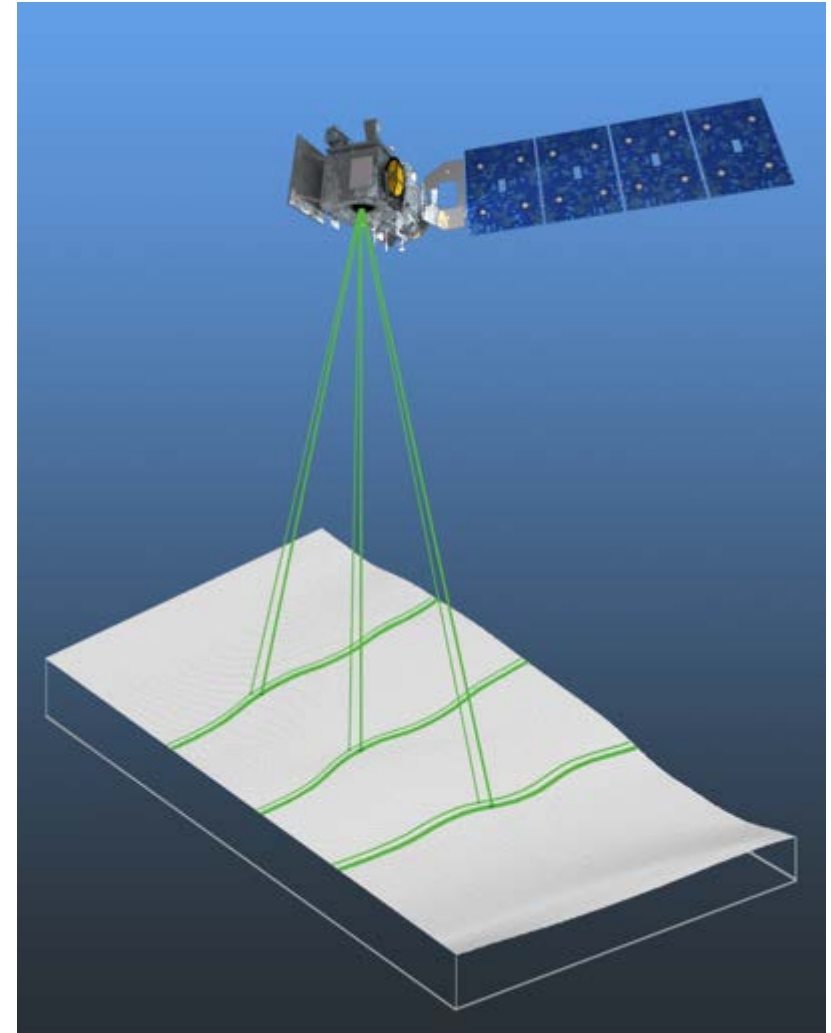
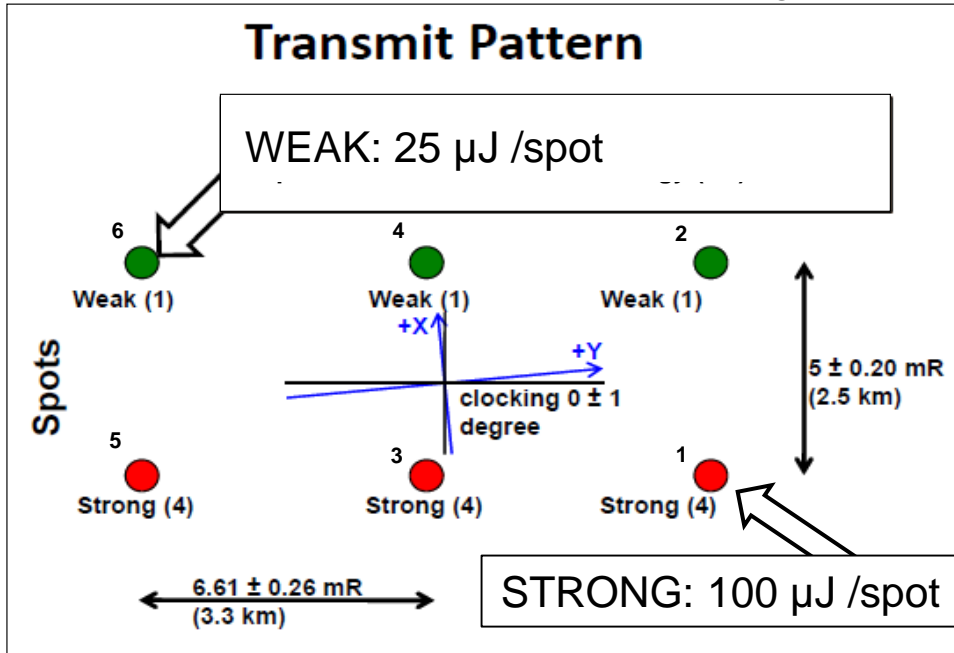
LASER

LASER SAMPLING ASSEMBLY

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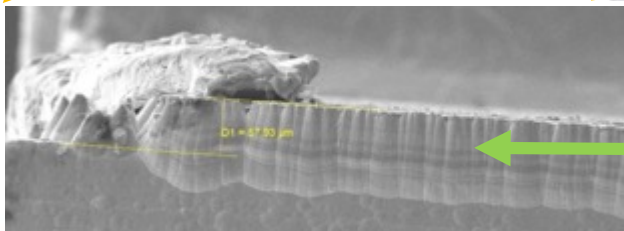
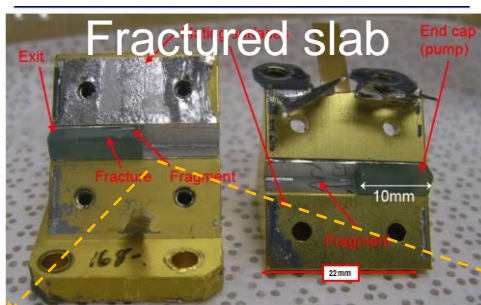
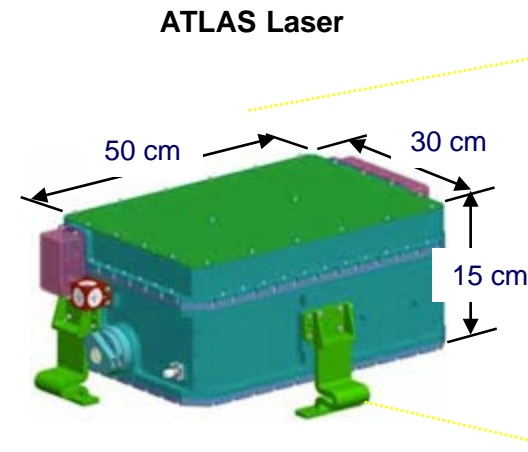
- Diffractive Optical Element (DOE) splits single input laser beam into 6 beams
- 3 x 2 array – 3 "Strong", 3 "Weak"
 - 4:1 energy ratio (Strong:Weak)
- 13.22 x 5.0 mrad angular separation
- Each spot <math><25\mu\text{rad}</math> divergence (same as input beam)
 - Each spot is ~12 m diameter on ground



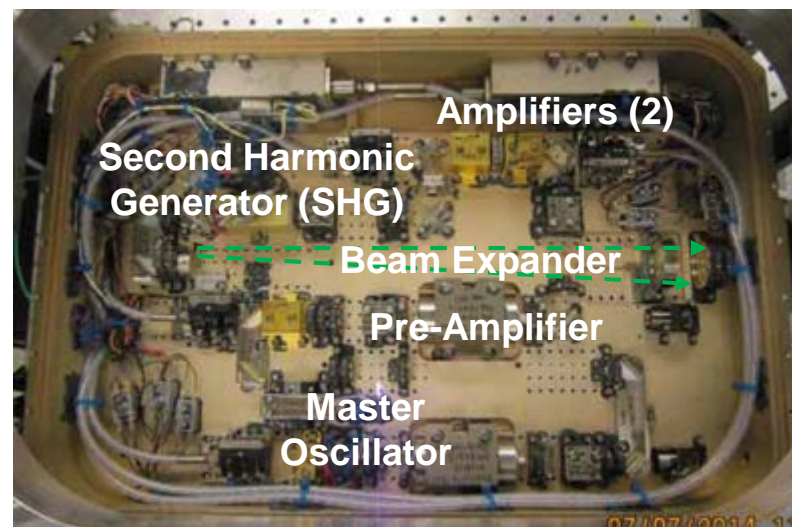
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- Pulse energy: $\sim 250 \mu\text{J} - 900 \mu\text{J}/\text{pulse}$, programmable
- Master Oscillator – Power Amplifier
 - Diode laser end-pumped Nd:YVO_4
- Oscillator :
 - Volume Bragg Grating output coupler for wavelength control
 - Active Q-switch
- Nd:YVO_4 slab fractured during test in 2016, prompting re-design
 - Design for flight:
 - Eliminated gold plating
 - Minimized mount induced stresses on slab
 - Selected crystals with the fewest flaws

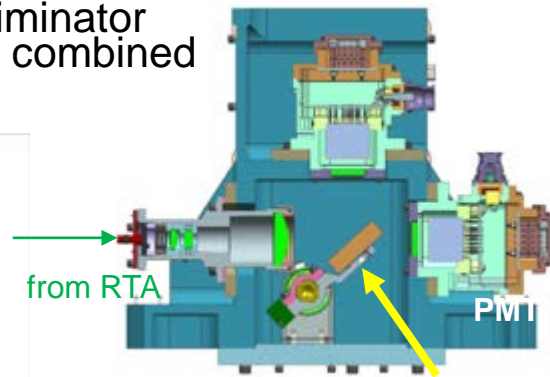
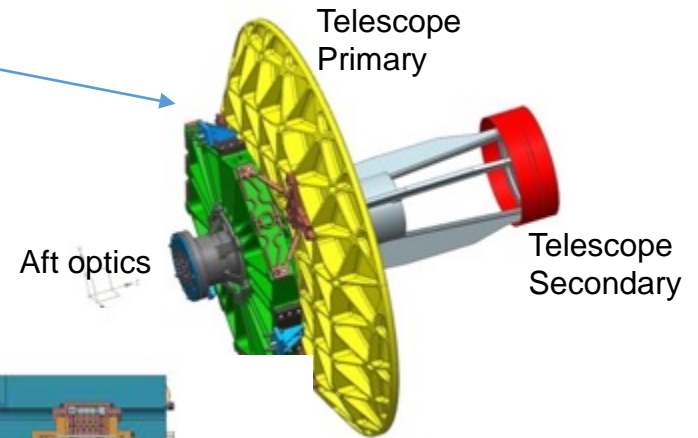


Gold-Indide growth at slab-mount interface

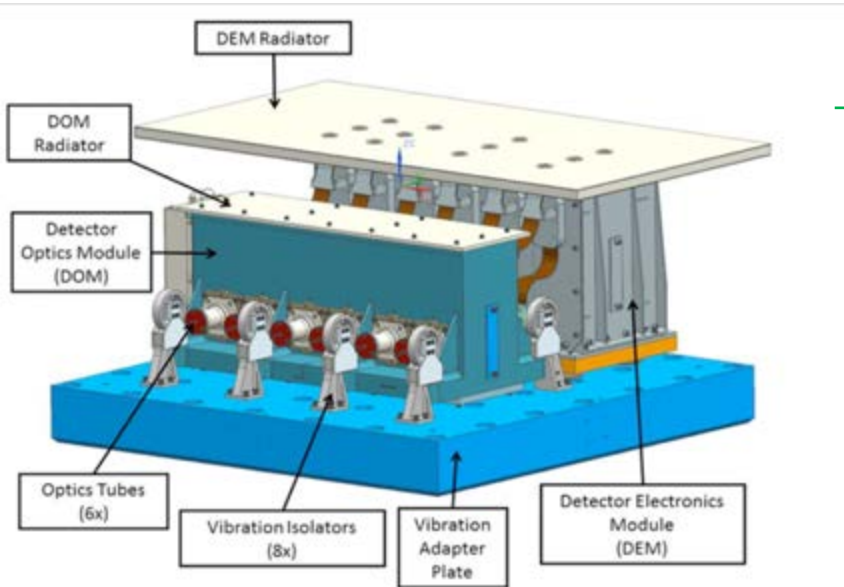
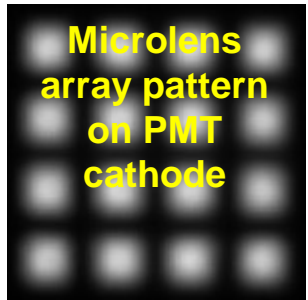


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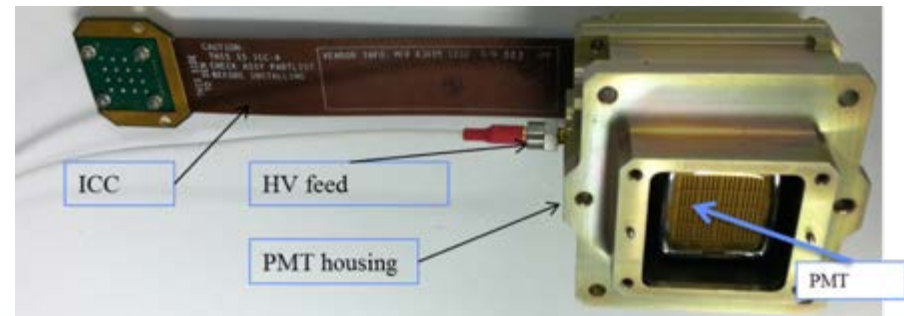
- Telescope : 80-cm Ritchey-Chretien Cassegrain
 - Fiber optics at focal plane couple received signals through temperature-controlled optical filters (30-pm bandwidth) to detector array
- Detectors : Hamamatsu R7600-300-M16 Green Extended Cathode Photomultipliers (PMT, qty. 6):
 - 16 channels Constant Fraction Discriminator (CFD) design for strong spots
 - 4 channels Constant Fraction Discriminator design for weak spots - 4 pixels are combined as one CFD channel.



Detector Select Mechanism



ATLAS Detector Array Assembly



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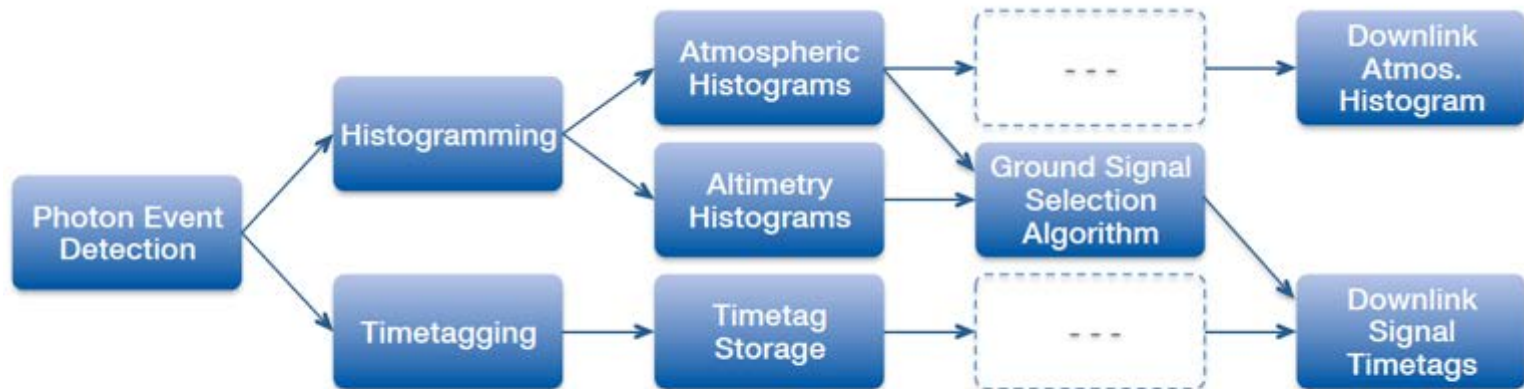


ATLAS Timing and Algorithms



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- Photon Counting Electronics
 - Time-tag each detected photon event from the Detector Electronics Module
 - Range gate determined by an on-board algorithm using
 - Digital Elevation/Relief maps (World Geodetic System 1984 ellipsoid)
 - Real-time spacecraft position and attitude (1-Hz, interpolated)
 - 100-MHz oscillator referenced to GPS 1-pps provides coarse time count
 - Gate delay FPGA provides fine time count with ~150-ps resolution
 - Time tags are histogrammed and likely ground signal selected for downlink by evaluating background, signal strength over time, and cloud cover



Data Processing Pipeline

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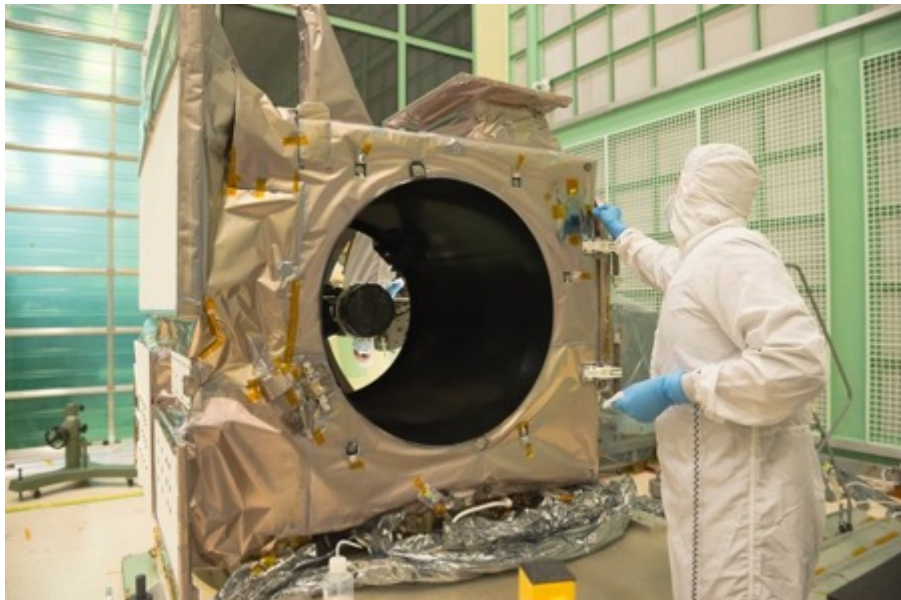


ATLAS Instrument Overview



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ATLAS in GSFC clean room



ATLAS in DELTA-II Fairing



MASS	470 kg
POWER	420 W
DAILY DATA VOLUME	~500 Gbits

On orbit laser pulse energy: 450 μ J

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ATLAS Transmit Optics Lifetime

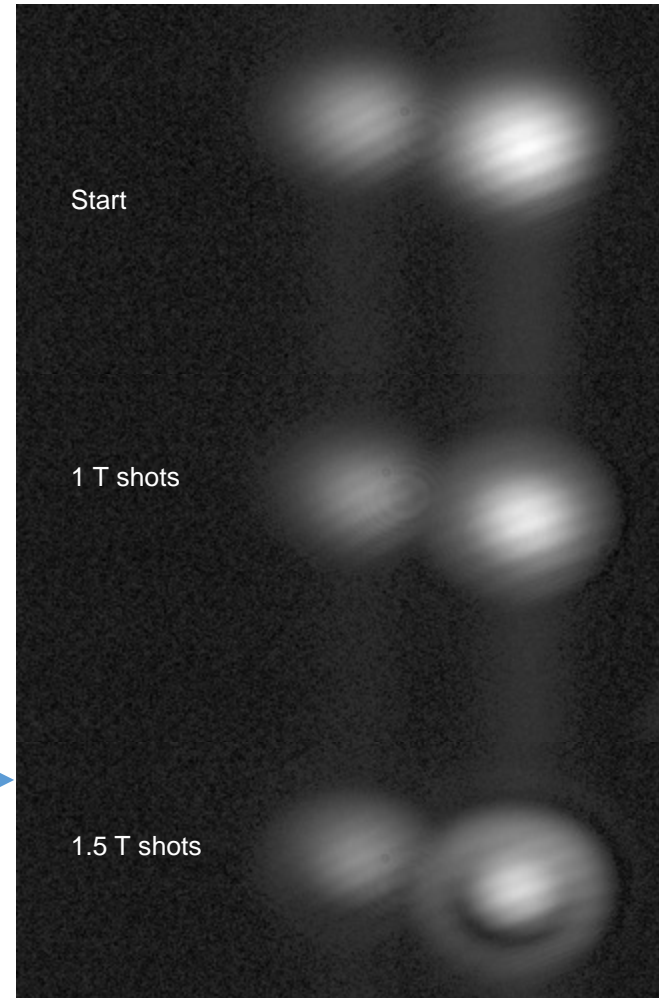


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• Transmit Optics Lifetime

- Optical coatings exposed to laser beams in vacuum tend to damage/degrade
- ATLAS lifetime requirement = 1T shots
- Mitigations:
 - Maintain 1 atm pressure inside laser
 - Expand laser beam inside pressurized laser before impinging on transmit optics in vacuum
 - Perform Laser Induced Damage Threshold testing to screen all optics
 - House transmit optics in heated enclosure to maintain temperature at 20°C
 - Optical Coatings Life Test (OCLT) →
 - Exposed coated optics to laser energy with varying levels of expected contaminants in vacuum.

Typical Beam Quality Degradation



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ATLAS GSE Bench Checkout Equipment

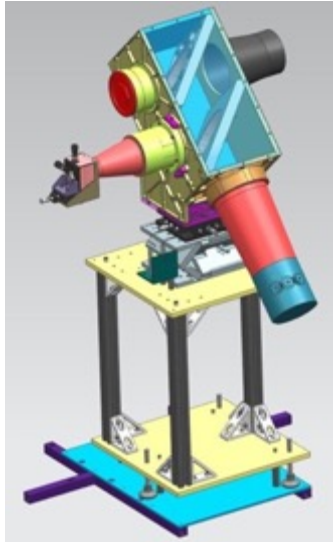


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LOAS

Laser Optical Attenuator and Sampler

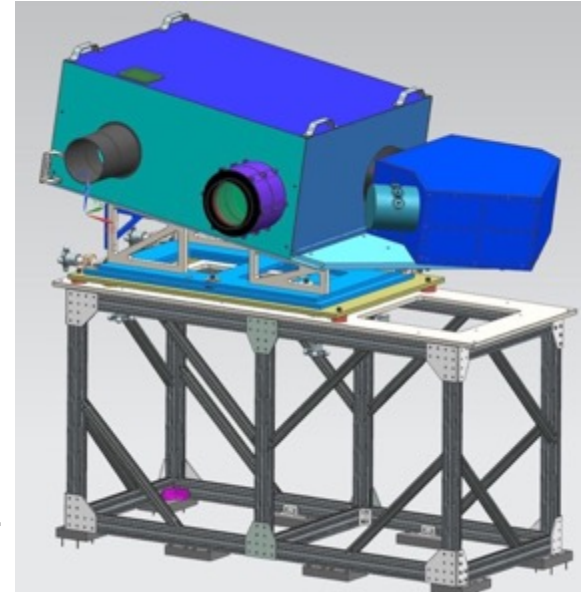
- Attenuates/stops and samples ATLAS laser beam
- Neutral Density filter stack
- Single mode fiber pick-off
- Integrating Sphere



MAAT

Main Alignment & Altimetry Target

- Retro-reflector & beam steering
- Laser Beam Diagnostics
 - Camera, Integrating Sphere, Fiber pickoff



Star Target

Fixed star pattern for LRS



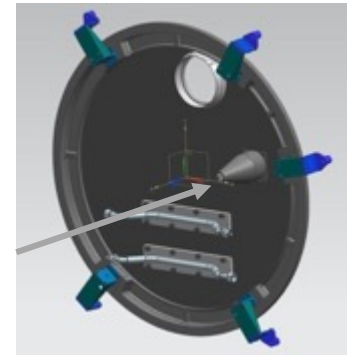
OTPSS

Optical Test Port Stimulator System

Diode laser sources, delay generators fiber coupled to ATLAS

Receiver Telescope Aperture Cover

- Path for aperture signal sources
- Showerhead
- Thermal target
- Stray light block



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Launch on the last Delta-II



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[Delta II](#) 7420-10C



15 September 2018, 13:02

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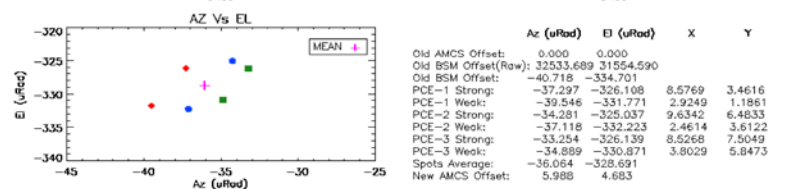
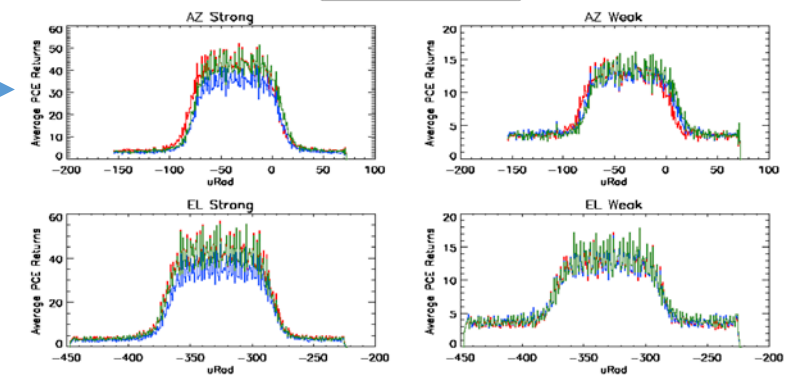
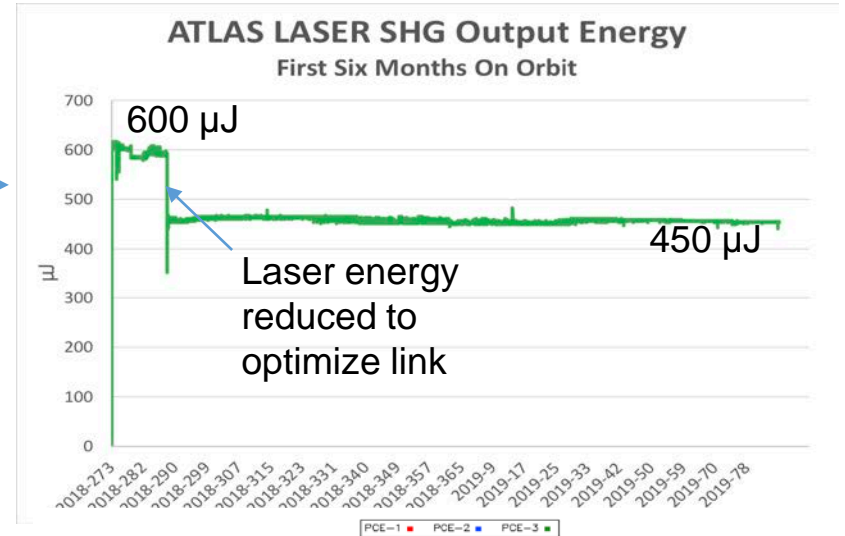


ATLAS Instrument Performance Transmitter



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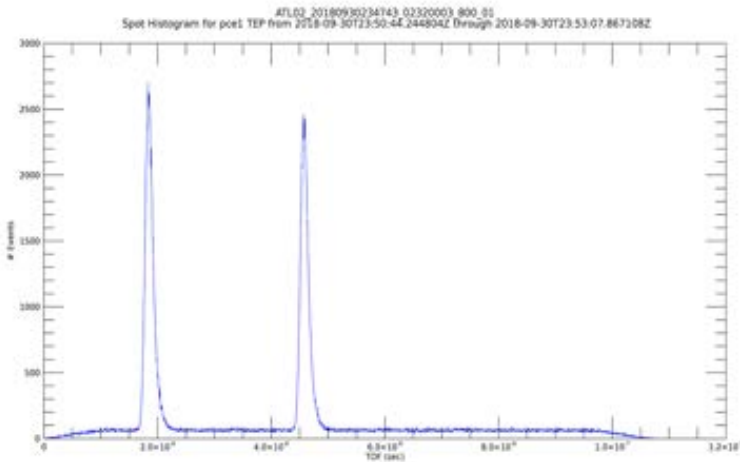
- Laser energy
 - Stable to within $\pm 2\%$ of ground test values
- Beam Quality
 - Nominal as indicated by slope of Alignment Monitoring and Control System (AMCS) sweeps
 - Plots show received beam convolved with RTA field of view
- Pointing
 - No change post-launch
 - Orbit variation $\sim 30 \mu\text{rad}$ p-p



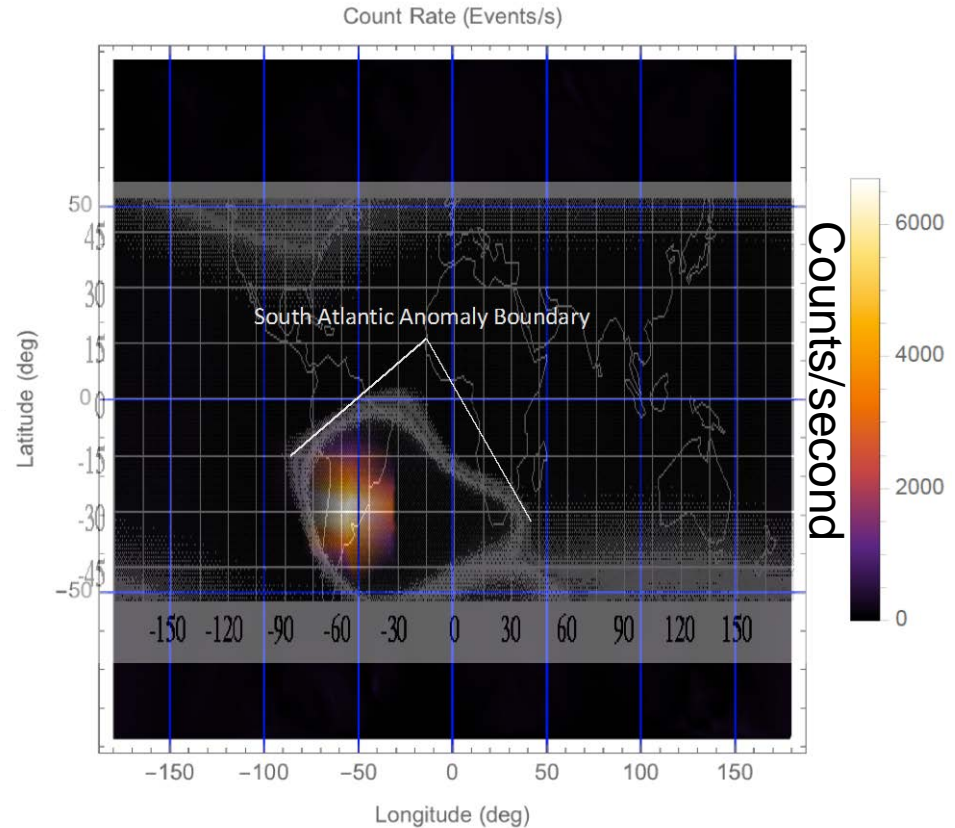
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- Receiver performance nominal
 - Impulse response as expected (Transmit Echo Pulse)
 - Solar background noise as expected



Transmit Echo Pulses



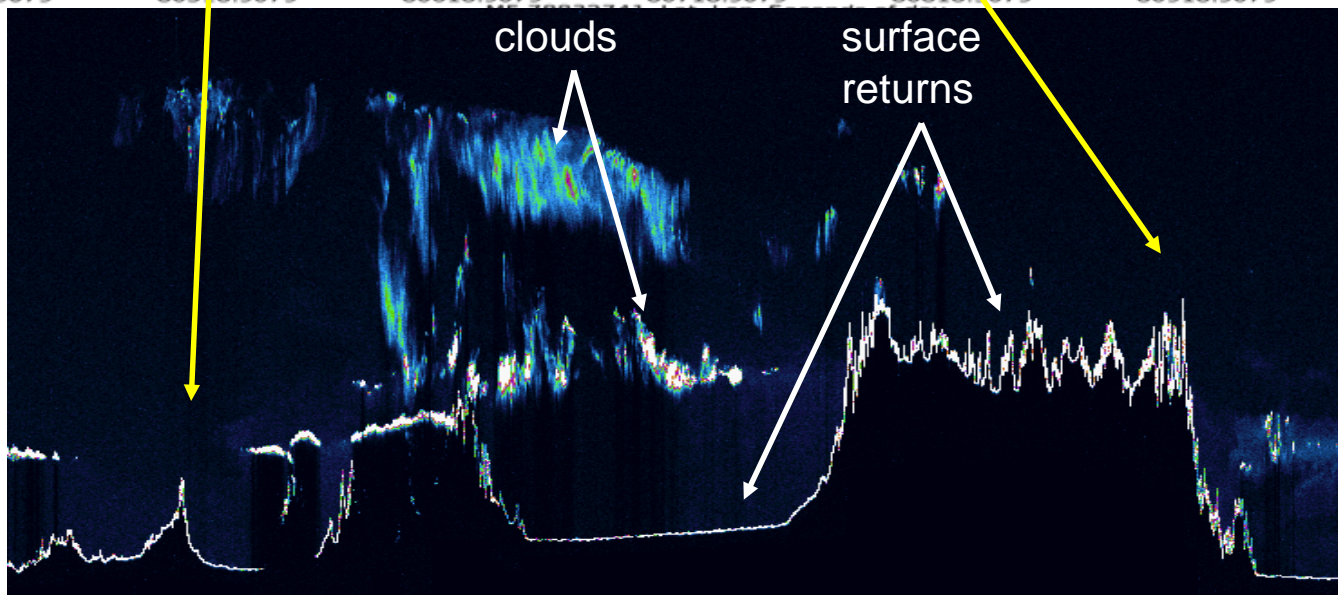
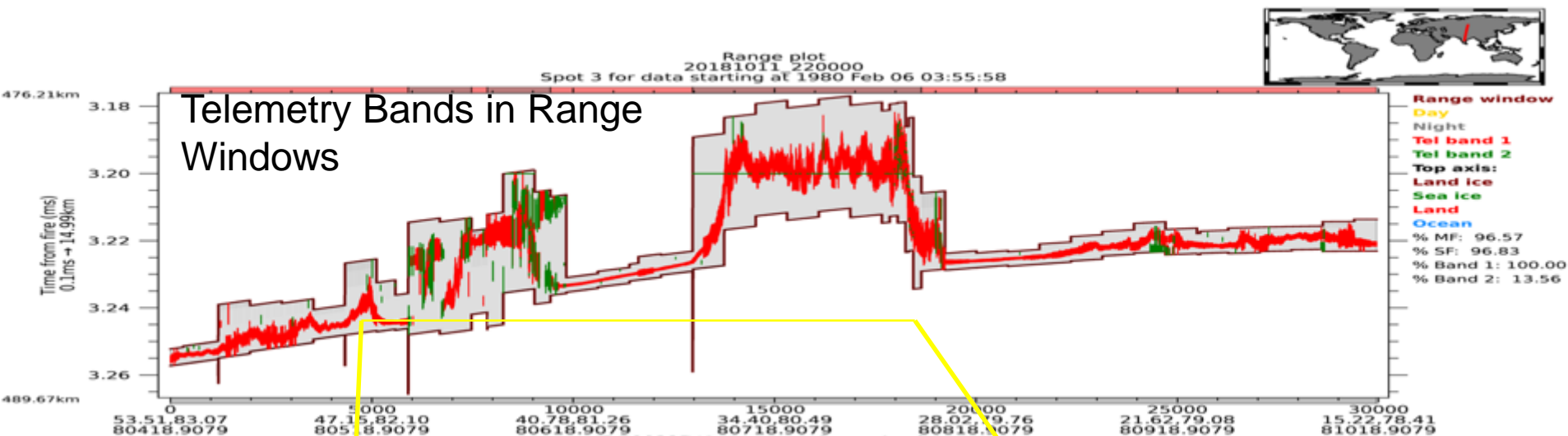
Photomultiplier responding to radiation over South Atlantic Anomaly prior to opening of aperture door. Counts are much lower than solar background rates



Receiver Algorithms On-Orbit Example Track over the Himalayas



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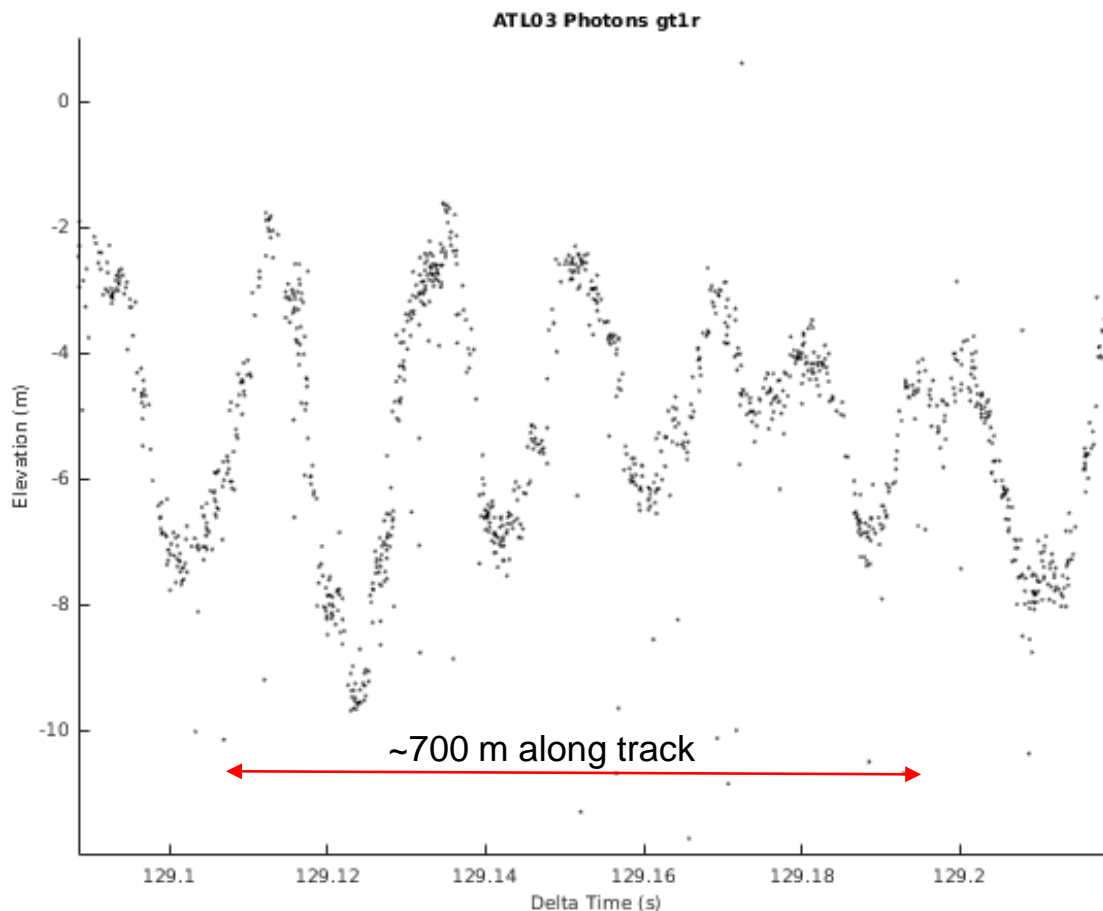
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ATLAS Observations Ocean surface waves



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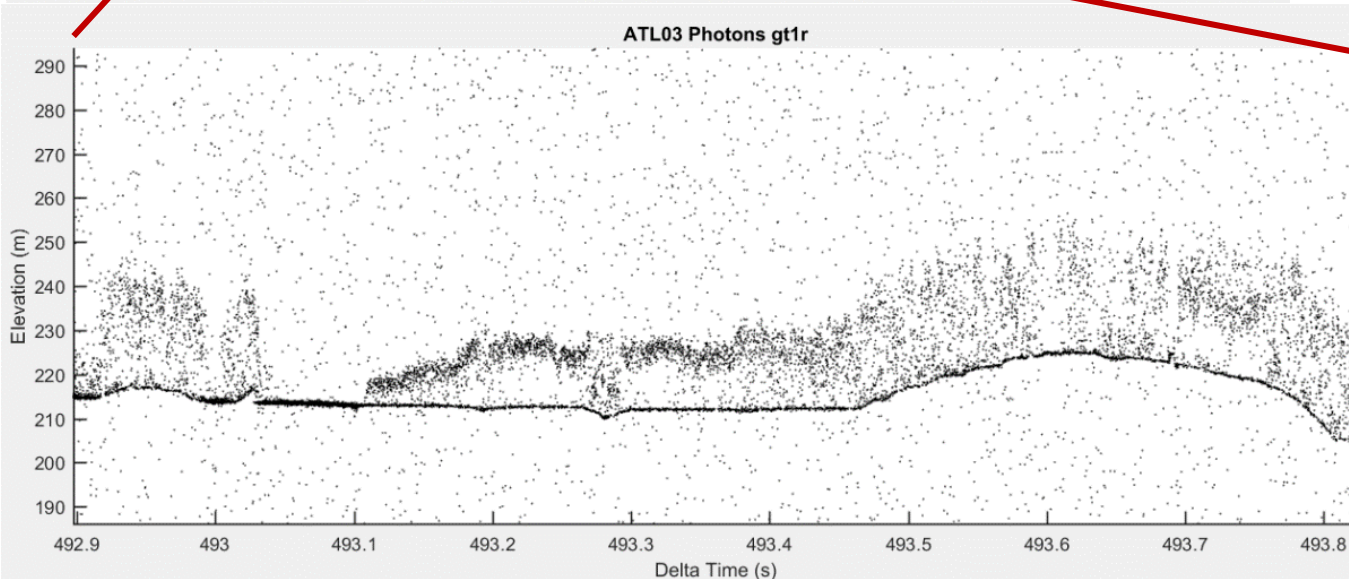
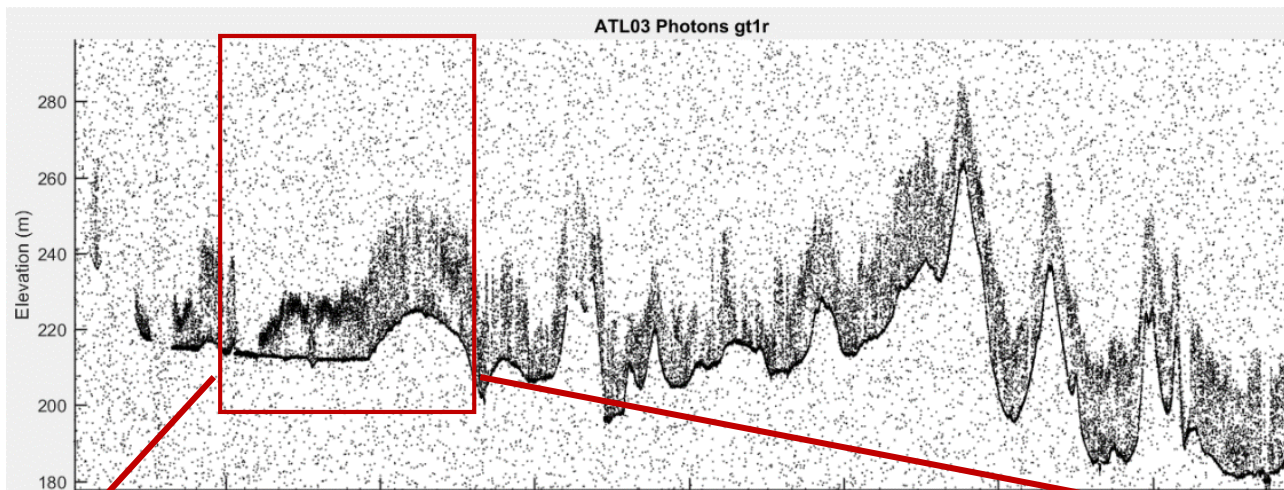


- With the strong beam, there is a well defined wave structure
- This example depicts wavelengths on the order of 140 m



ATLAS Observations Trees in Russia, daytime acquisition

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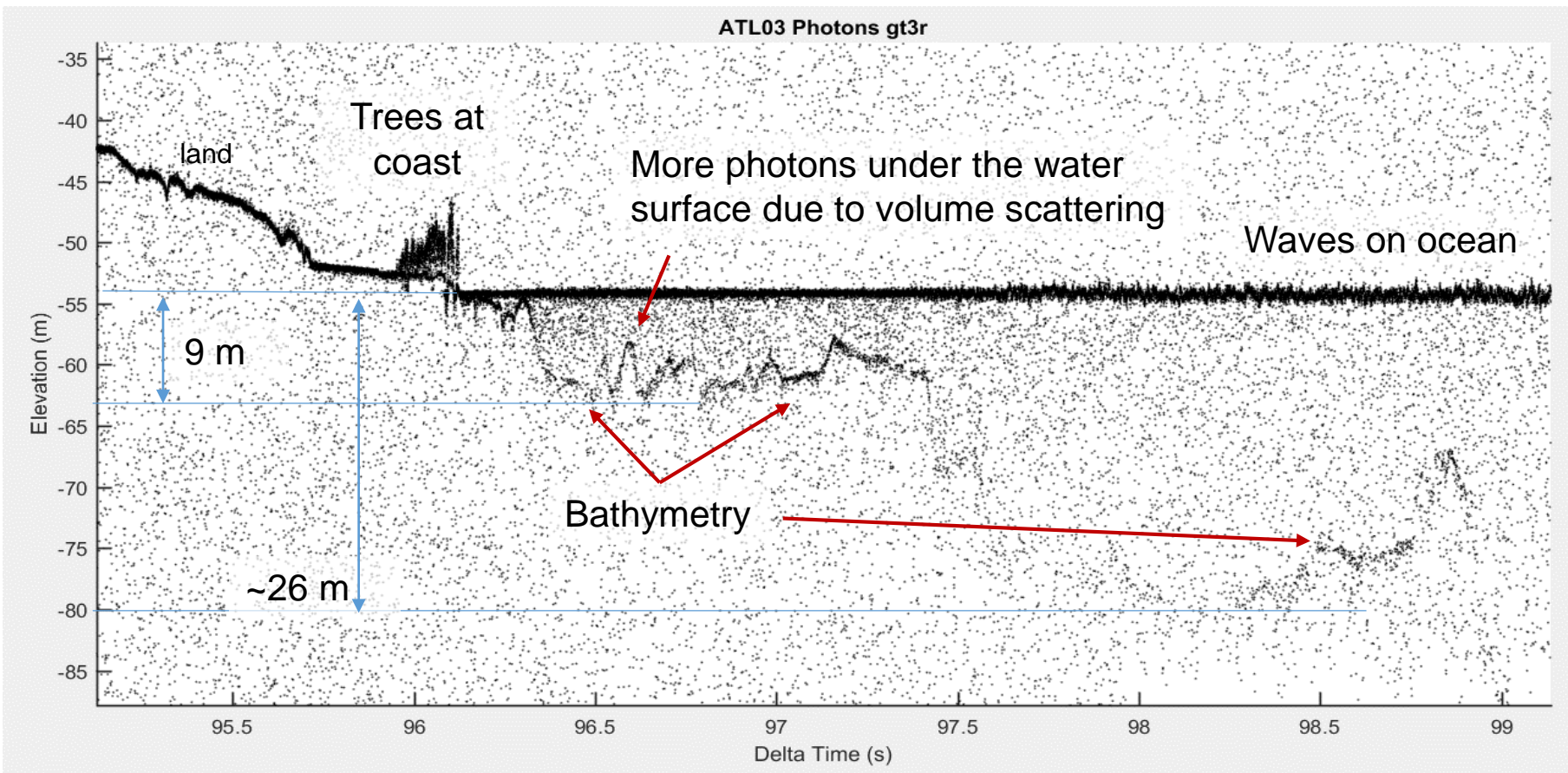




ATLAS Observations Australian Coast (NW) ATL03 Transect



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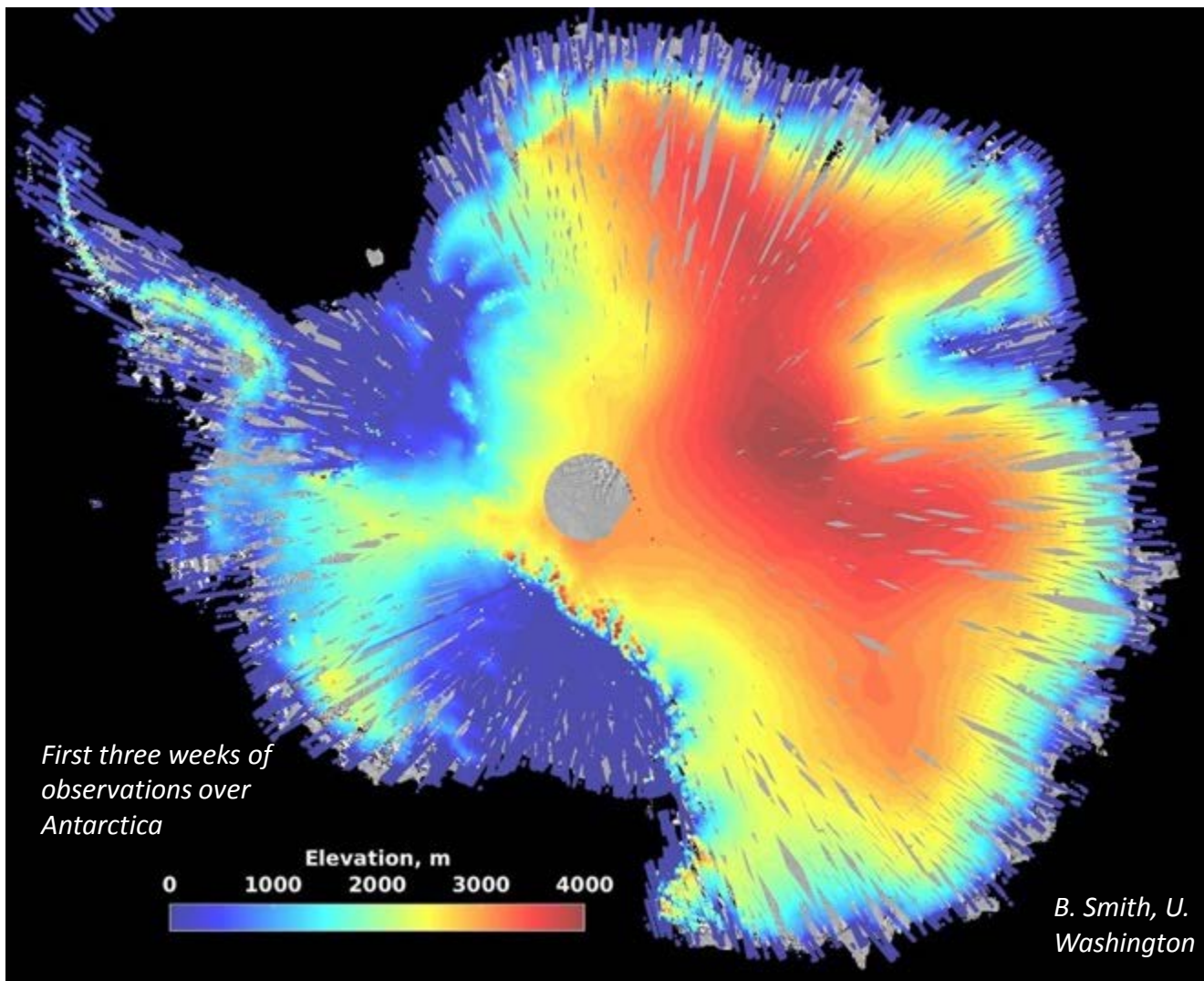
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ATLAS Observations Ice Sheet Elevation - Antarctica



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*First three weeks of
observations over
Antarctica*

Elevation, m
0 1000 2000 3000 4000

*B. Smith, U.
Washington*

21

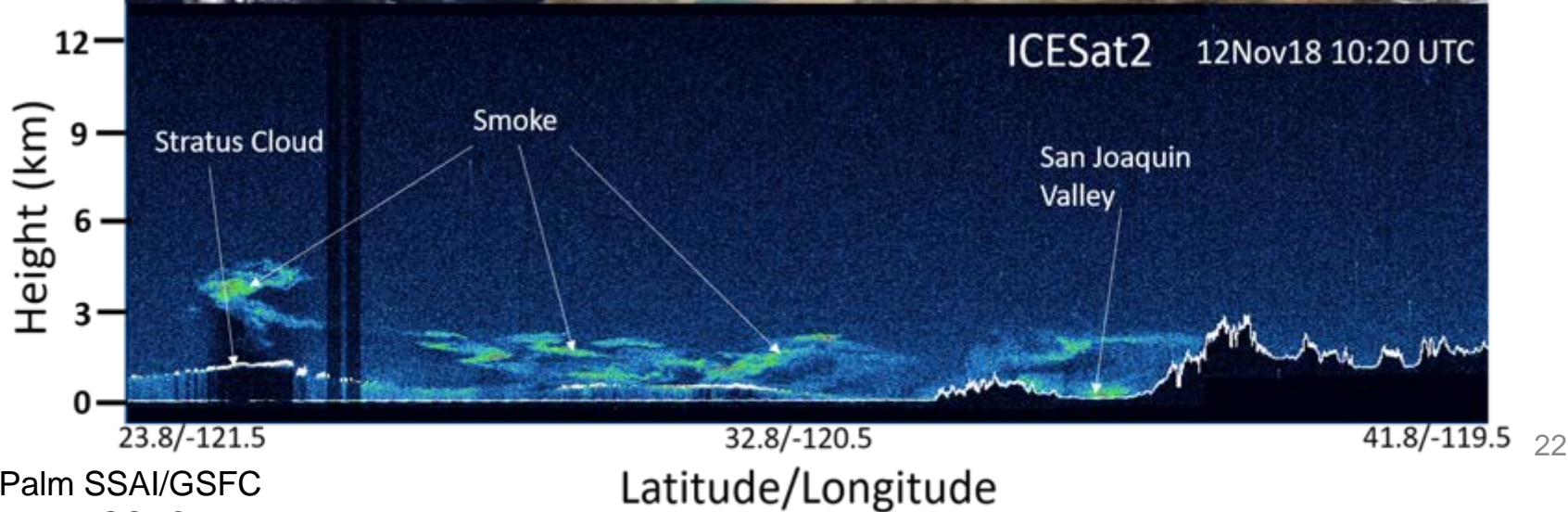
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Atmospheric Observations Smoke from California Fires



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Steve Palm SSAI/GSFC
Yuekui Yang GSFC

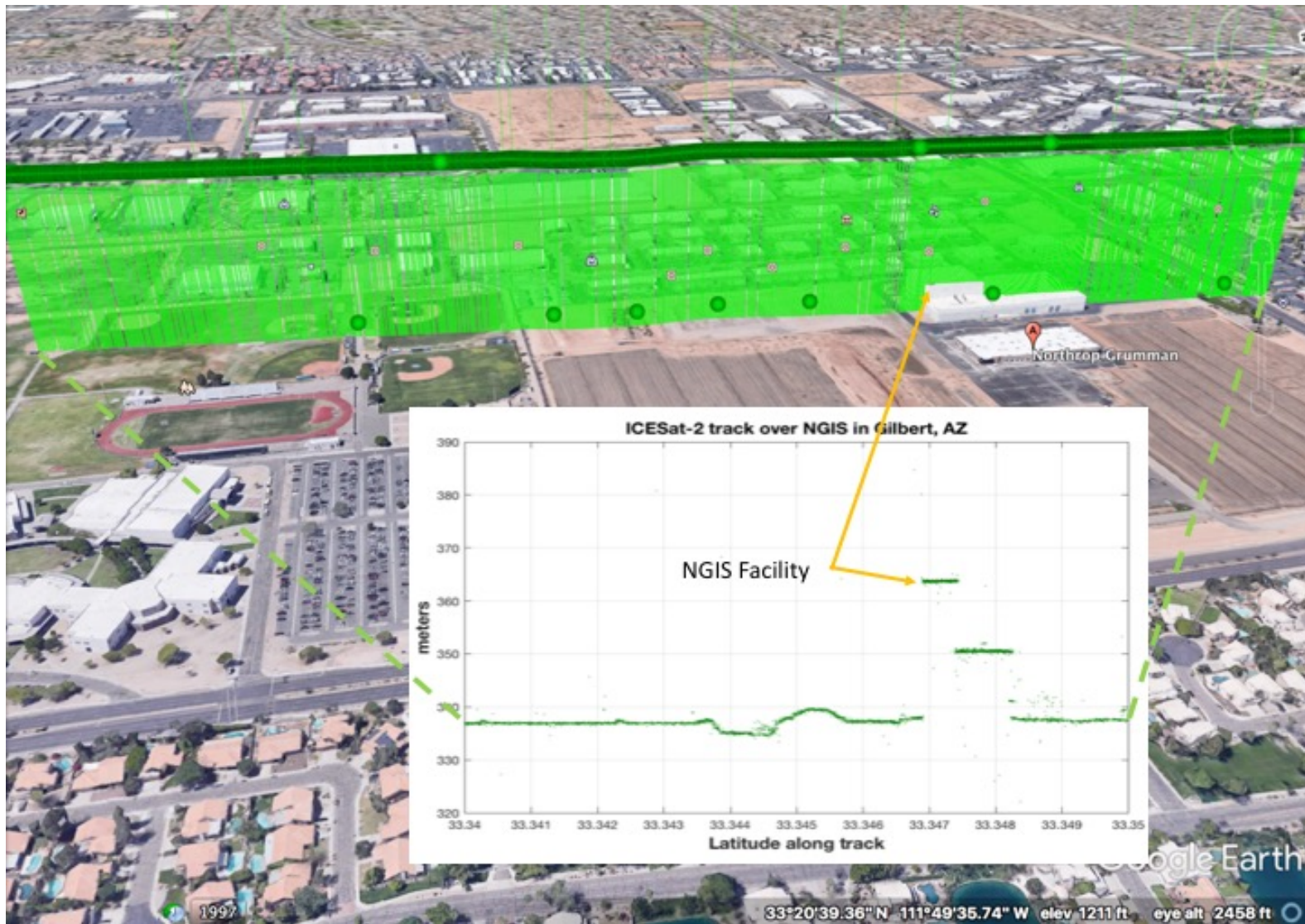
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Northrop Grumman Gilbert Facility ICESat-2 Spacecraft Integration & Test Location



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Acknowledgements



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Donya Douglas-Bradshaw/ NASA – Instrument Mgr.

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Tony Martino/NASA – Instrument Scientist

Tom Neumann/NASA – Project Scientist

Mike Bay/Bay Engineering – Instrument Systems

Megan Bock/NASA – Instrument Systems

Dann Brown/NASA – Safety & Mission Assurance

Jim Buchheit/NASA – Instrument Systems

Claudia Carabajal/Sigma Space- Algorithms

John Chambers/NASA - Optics

Tom Correll/NASA – Laser Reference System

Pete Gonzales/Bay Engineering – Instrument Systems

Jake Hageman/NASA – Instrument Systems

Pete Liiva/Sigma Space – Bench Checkout Equipment

Manuel Maldonado/NASA – Instrument Flight Software

Jan MacGarry/NASA – Algorithms Lead

Deepak Patel/NASA – Thermal Engineer

Luis Ramos-Izquierdo/NASA – Optics

Nick Sawruk/Fibertek - Laser

Oren Sheinman/NASA – Mechanical Systems

Kathy Strickler/Alcyon Tech. Services – I&T Manager

Joseph-Paul Swinski/NASA – Instrument Software

Elisavet Troupaki/NASA - Laser

Bonnieblue Valentin-Santiago/SGT - Contamination

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Acronyms



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AMCS Alignment Monitoring and Control Subsystem

ATLAS Advanced Topographic Laser Altimeter System

AZ Azimuth

BCE Bench Check-Out Equipment

BSM Beam Steering Mechanism

CCHP Constant Conductance Heat Pipe

CFD Constant Fraction Discriminator

DAA Detector Array Assembly

DEM Detector Electronics Module

DOE Diffractive Optical Element

DOM Detector Optics Module

DRM Digital Relief Model

EL Elevation

FM Fold Mirror

HV High Voltage

ICESat-2 Ice Cloud and Land Elevation Satellite - 2

IMSC Instrument-Mounted Spacecraft Components

LHP Loop Heat Pipe

LRS Laser Reference System

LSA Laser Sampling Assembly

MOPA Master Oscillator - Power Amplifier

Nd:YAG Neodymium-doped Yttrium Aluminum Garnet

Nd:YVO4 Neodymium-doped Yttrium Orthovanadate

PBC Polarization Beam Combiner

PMT Photomultiplier

RTA Receiver Telescope Assembly

SHG Second Harmonic Generator

Si-APD Silicon Avalanche Photodiode

SPCM Single Photon Counting Module

TAMS-LS Telescope Alignment Monitoring System Light Source

USO Ultra Stable Oscillator

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