



3D unsteady model of arc heater plasma flow using the ARC Heater Simulator (ARChES)

Abstract

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Magnus Haw and Nagi N. Mansour



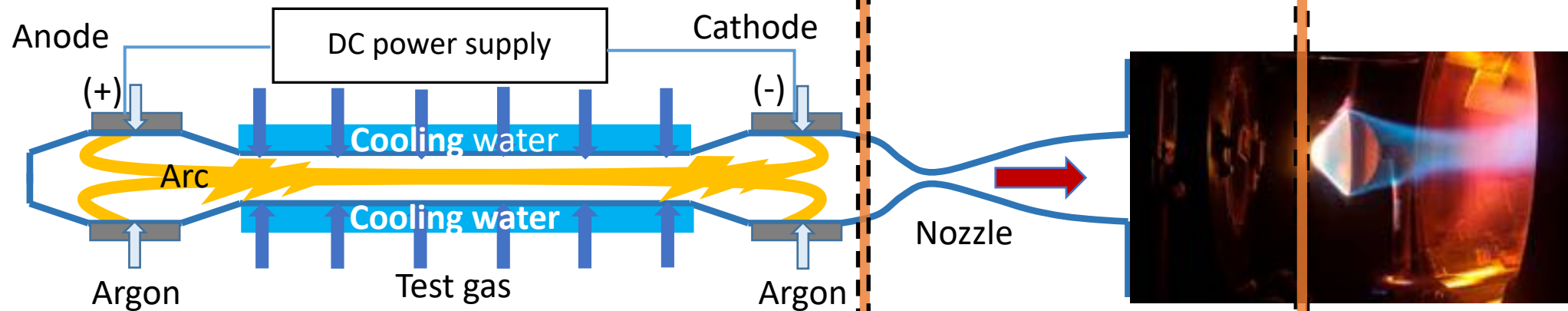
Arc jet facility



MHD + 3D radiative transfer

*Aerothermal
& Hypersonic*

*Material
Response*



ARChES is a platform being developed to...

- provide inlet conditions needed by aerothermal models that yield flow properties and conditions at the test article,
- provide understanding of facility operation,
- guide setting up test conditions,
- tailor conditions to improve uniformity,
- inform electrode maintenance schedule,
- inform (V&V) upgrade designs,
- optimize current operational capability...



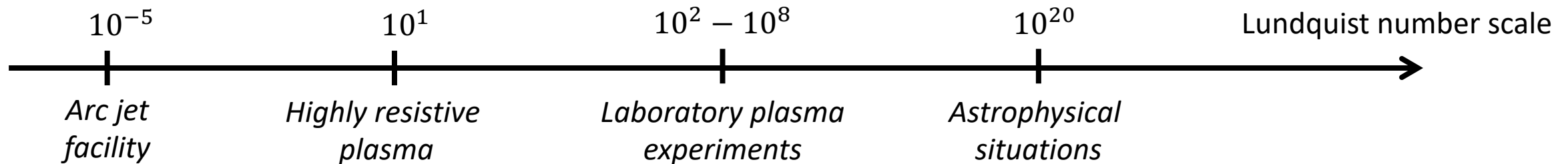
Plasma fields



FLOW CONDITION	$U_{\text{avg}} \approx 400 \text{ m/s}$	$P_{\text{avg}} \approx 13 \text{ atm}$	$T_{\text{avg}} \approx 8000 \text{ K}$
ELECTROMAG.	$I_{\text{avg}} \approx 1200 \text{ A}$	$B_{\text{avg}} \approx 0.2 \text{ T}$	$\sigma_{\text{avg}} \approx 0.1 \text{ S/m}$

$$S = \frac{L v_A}{\eta} = \frac{L \frac{B}{\sqrt{\rho \mu_0}}}{\eta} = 10^{-5}$$

High Lundquist numbers indicate **highly conducting** plasmas.
 Low Lundquist numbers indicate **more resistive** plasmas.





Assumptions



$$S = \frac{L v_A}{\eta} = 10^{-5} \longrightarrow \text{The magnetic convection is negligible}$$

$$\mathbf{J} = \sigma \mathbf{E} \longrightarrow \text{The Ohm's law is simplified}$$

$$\partial_{\mathbf{x}} \times \mathbf{B} = \mu_0 \mathbf{J} \longrightarrow \text{The displacement current is ignored } (V_0 \ll c)$$

$$\mathbf{B} = \partial_{\mathbf{x}} \times \mathbf{A} \longrightarrow \text{The vector potential formulation ensures zero divergence of the } \mathbf{B} \text{ field}$$

$$\partial_{\mathbf{x}} \cdot \mathbf{J} = 0 \longrightarrow \text{Diverge of Ampere's law gives the continuity of current}$$

$$-\partial_{\mathbf{x}}^2 \mathbf{A} = \mu_0 \mathbf{J} \longrightarrow \text{Rotational of Ampere's law simplifies the equation}$$

$$\begin{cases} \partial_{\mathbf{x}} \cdot (-\sigma \partial_{\mathbf{x}} \phi_i) = 0 \\ \partial_{\mathbf{x}}^2 \mathbf{A}_i - \mu_0 \sigma \partial_{\mathbf{x}} \phi_i = 0 \end{cases} \longrightarrow \text{System of electromagnetic equations solved in ARChES}$$



Physical model



MASS	$\partial_t \rho + \partial_x \cdot (\rho \mathbf{u}) = 0$
MOMENTUM	$\partial_t (\rho \mathbf{u}) + \partial_x \cdot (\rho \mathbf{u} \mathbf{u}) = -\partial_x p + \partial_x \cdot \bar{\boldsymbol{\tau}}$
ENERGY	$\partial_t (\rho E_0) + \partial_x \cdot (\rho H_0 \mathbf{u}) = \partial_x \cdot (\bar{\boldsymbol{\tau}} \cdot \mathbf{u} + \mathbf{q}^{\text{cond}})$



Physical model



MASS	$\partial_t \rho + \partial_x \cdot (\rho \mathbf{u}) = 0$
MOMENTUM	$\partial_t (\rho \mathbf{u}) + \partial_x \cdot (\rho \mathbf{u} \mathbf{u}) = -\partial_x p + \partial_x \cdot \bar{\boldsymbol{\tau}} + \mathbf{J} \times \mathbf{B}$
ENERGY	$\partial_t (\rho E_0) + \partial_x \cdot (\rho H_0 \mathbf{u}) = \partial_x \cdot (\bar{\boldsymbol{\tau}} \cdot \mathbf{u} + \mathbf{q}^{\text{cond}}) + \sigma \mathbf{E} ^2 + \mathbf{u} \cdot (\mathbf{J} \times \mathbf{B})$
IMPOSED CURRENT	$\partial_x \cdot (-\sigma \partial_x \phi_i) = 0 \quad \mathbf{E}_i = -\partial_x \phi_i \quad \mathbf{J}_i = \sigma \mathbf{E}_i$
IMPOSED MAGNETIC	$\partial_x^2 \mathbf{A}_i - \mu_0 \sigma \partial_x \phi_i = 0 \quad \mathbf{B}_i = \partial_x \times \mathbf{A}_i$
EXTERNAL MAGNETIC	$\mathbf{A}_e = \frac{\mu_0 I_e}{4\pi} \oint \frac{d\mathbf{l}}{ \mathbf{r} - \mathbf{r}' } \quad \mathbf{B}_e = \partial_x \times \mathbf{A}_e$
TOTAL FIELD	$\mathbf{B} = \mathbf{B}_i + \mathbf{B}_e \quad \mathbf{E} = \mathbf{E}_i \quad \mathbf{J} = \mathbf{J}_i$



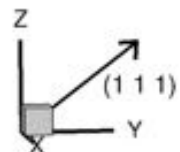
Physical model



MASS	$\partial_t \rho + \partial_x \cdot (\rho \mathbf{u}) = 0$
MOMENTUM	$\partial_t (\rho \mathbf{u}) + \partial_x \cdot (\rho \mathbf{u} \mathbf{u}) = -\partial_x p + \partial_x \cdot \bar{\boldsymbol{\tau}} + \mathbf{J} \times \mathbf{B}$
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IMPOSED CURRENT	$\partial_x \cdot (-\sigma \partial_x \phi_i) = 0 \quad \mathbf{E}_i = -\partial_x \phi_i \quad \mathbf{J}_i = \sigma \mathbf{E}_i$
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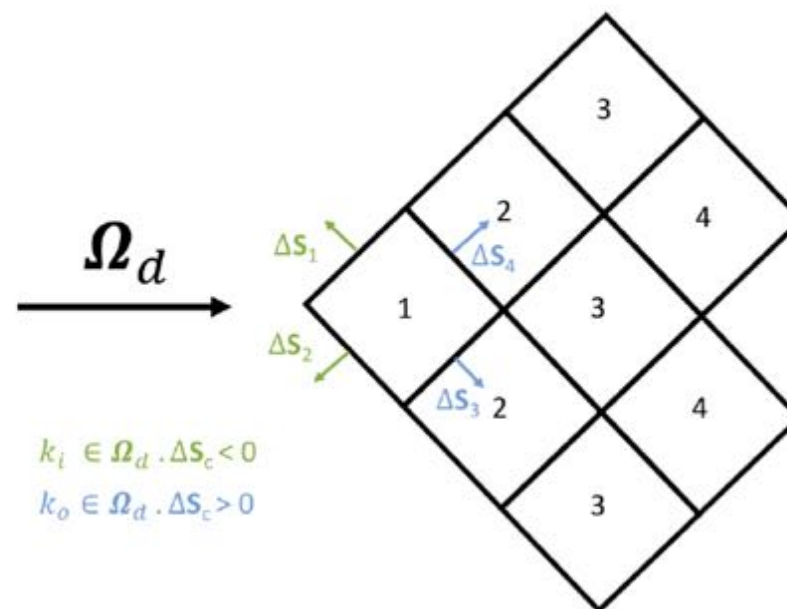


Physical model



$$I_{v,c}^d = \frac{\sum_{k_i}^{N_{k_i}} I_{v,k_i}^d \Omega_d \cdot \Delta S_c + \kappa_{v,c} B_{v,c} V_c}{\sum_{k_o}^{N_{k_o}} I_{v,k_o}^d \Omega_d \cdot \Delta S_c + \kappa_{v,c} V_c}$$

$$\partial_x \cdot q^{\text{rad}} = \sum_v^{N_b} \sum_d^{N_d} \frac{a_d I_{v,c}^d \Omega_d \cdot \Delta S_c}{V_c}$$

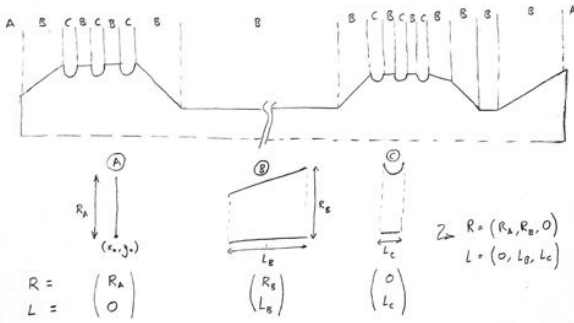




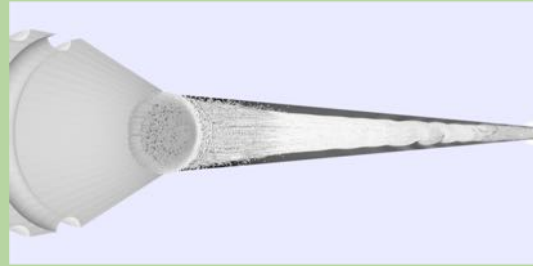
ARC Heater Simulator



Geometry concept



OpenFOAM



Finite Volume

I/O management

Massive MPI

Moving geometry

Basic mesh gen.

Mixture tables

Chemistry

Thermo/Transp.

Turbulence

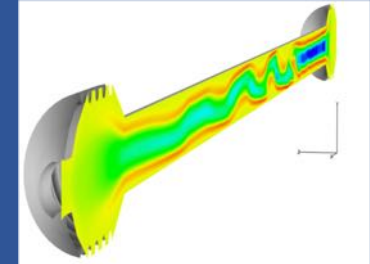
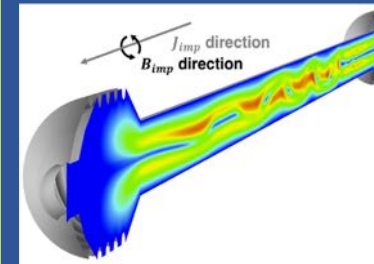
Compressible flow

Radiation

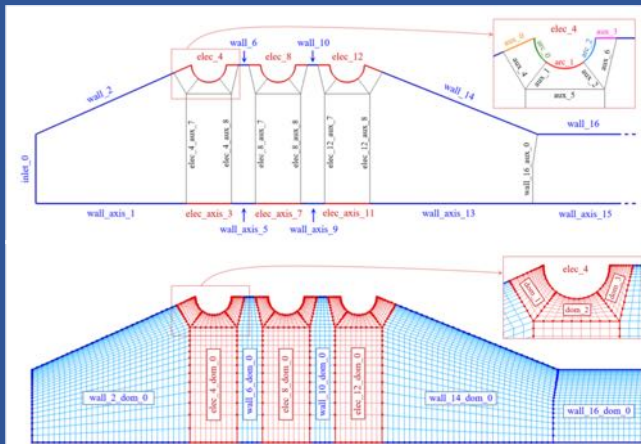
ARChES

MHD + ballast BC

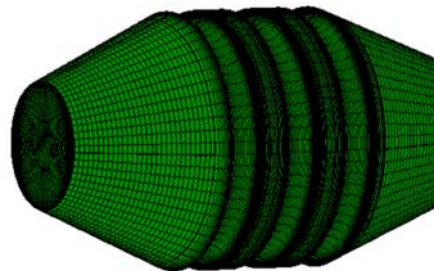
3D radiative transfer



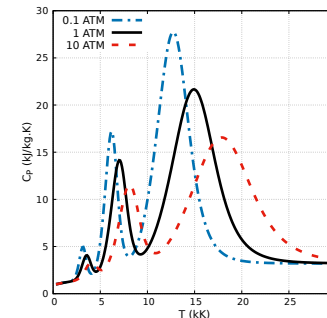
Automatic mesh inputs



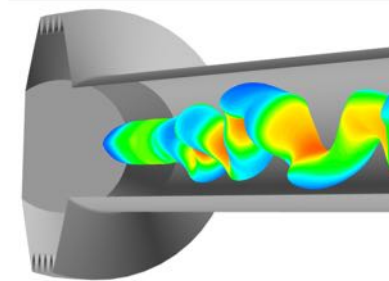
Complex mesh generation



Thermo/Transport/Chemistry

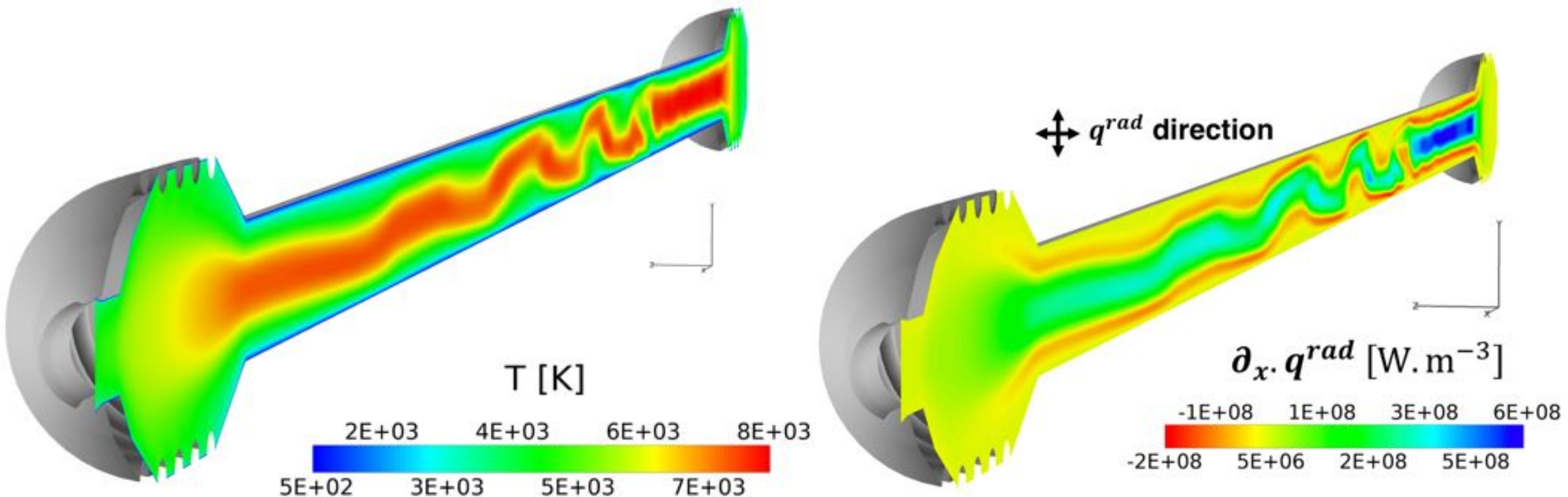


Post-processing 1D/2D/3D





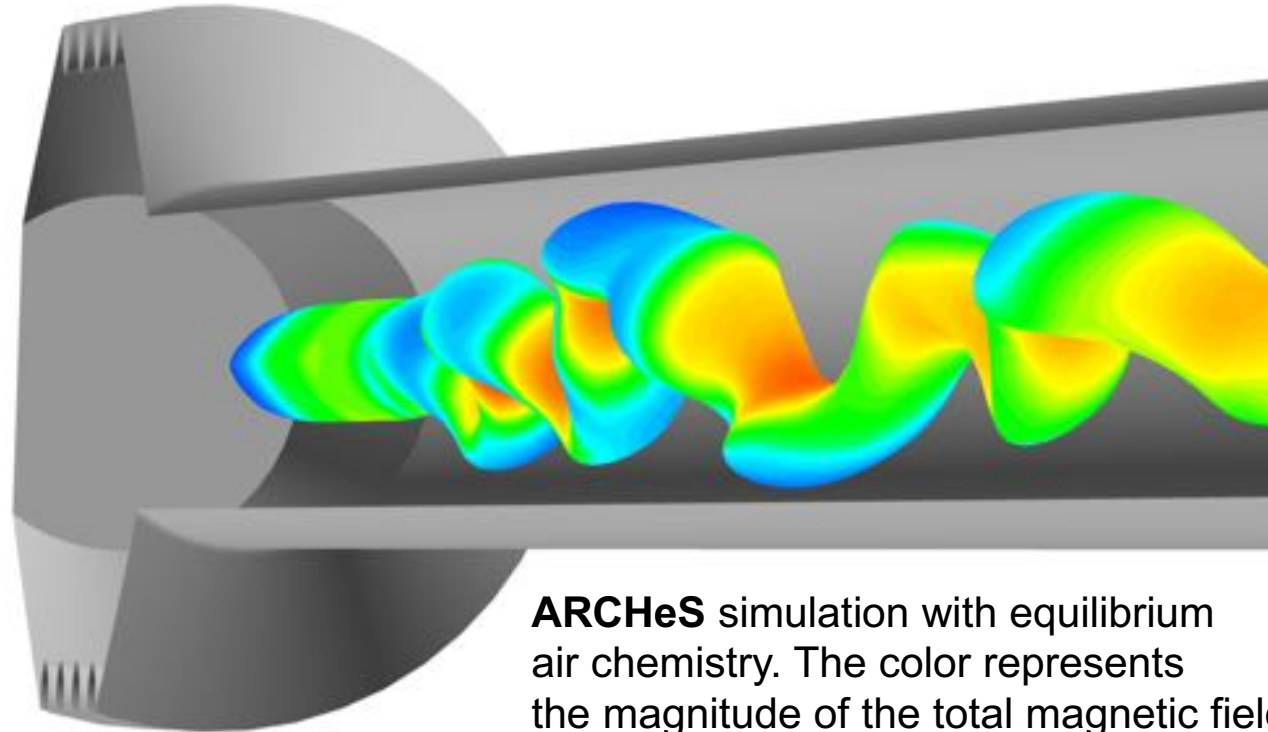
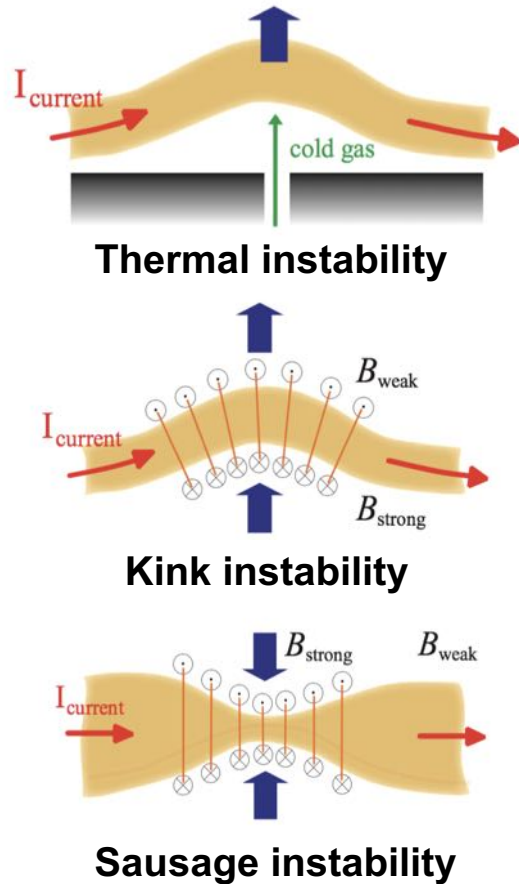
Cutting plane



Hot electric arc core cools down and the surroundings warm up.
Importance of the 3D radiative transfer.



Electric arc instabilities

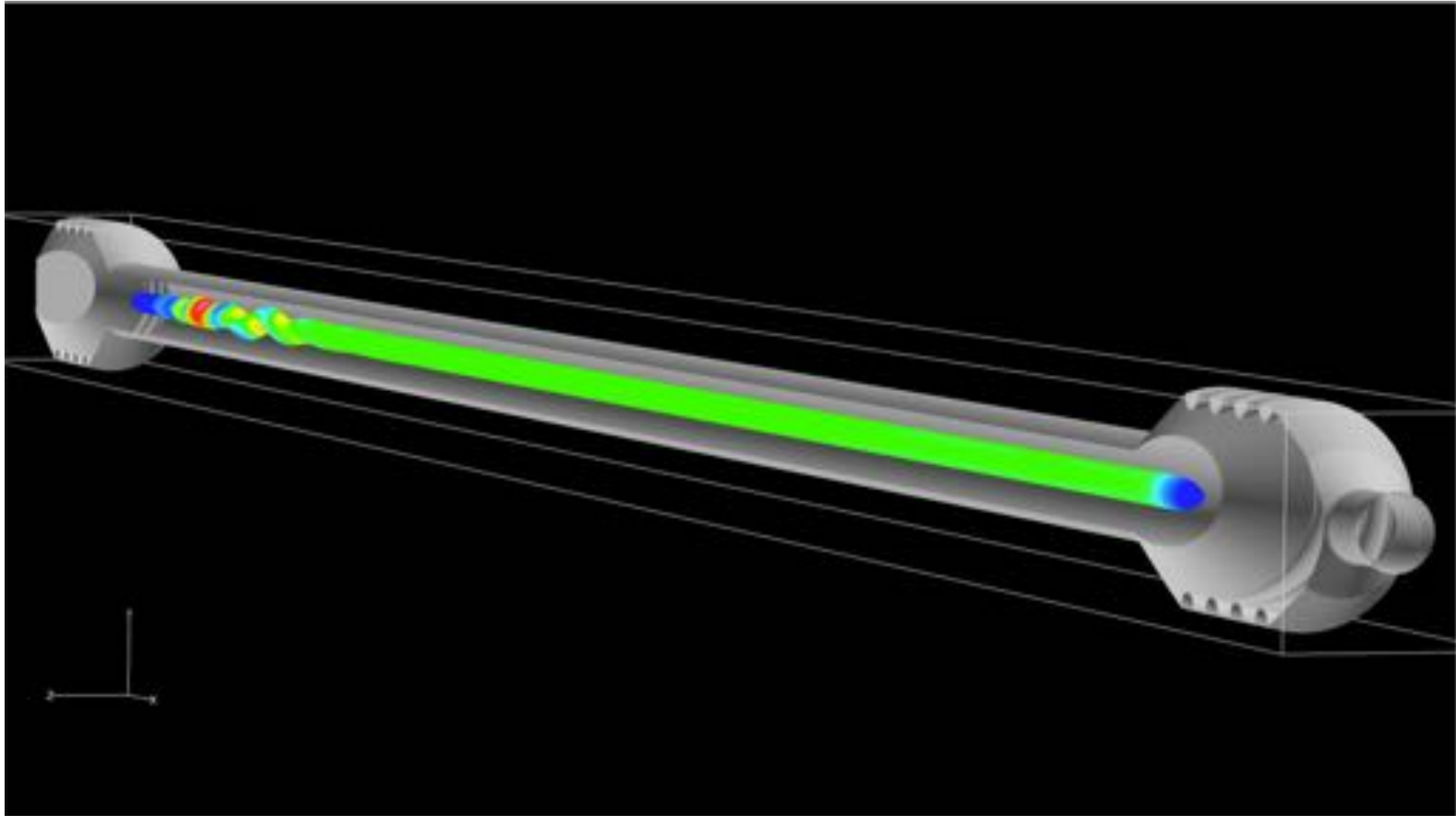


ARChES simulation with equilibrium air chemistry. The color represents the magnitude of the total magnetic field. Iso-surface of the current density of 1 MA/m².

Stable arc next to the electrode chambers.
Instabilities arise in the constrictor.

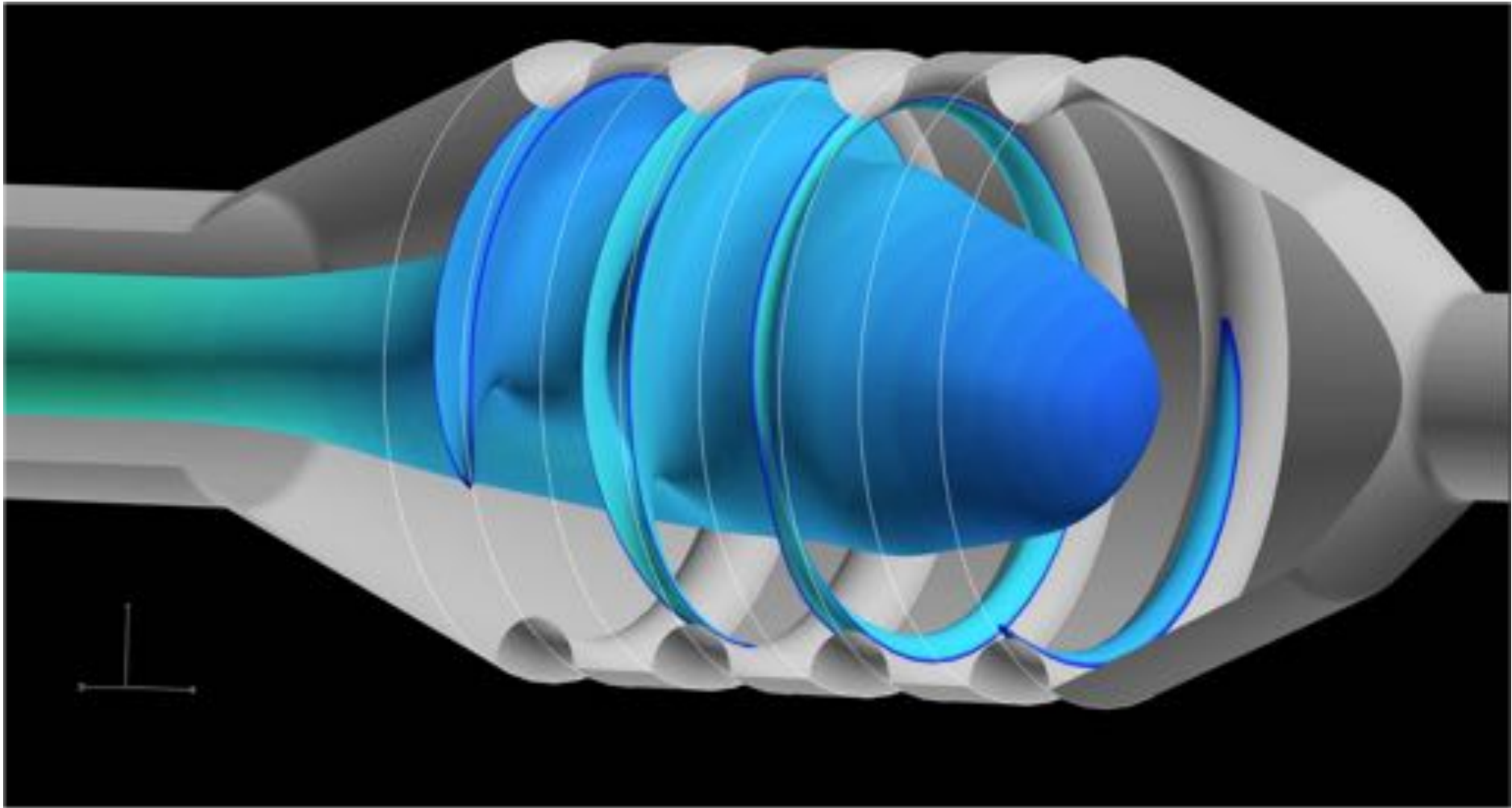


Air mixture





Electric arc attachment





Cross technology impact

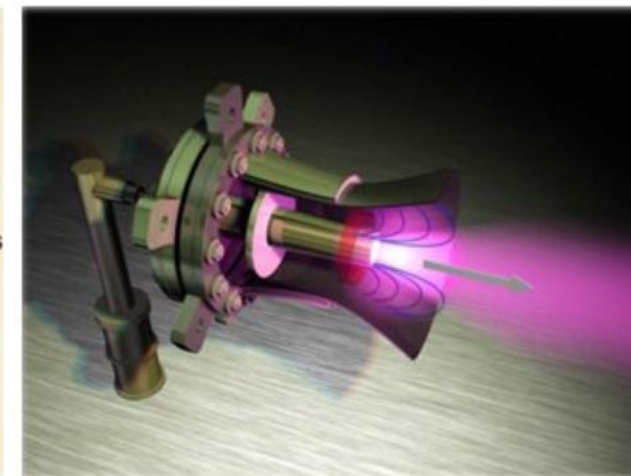
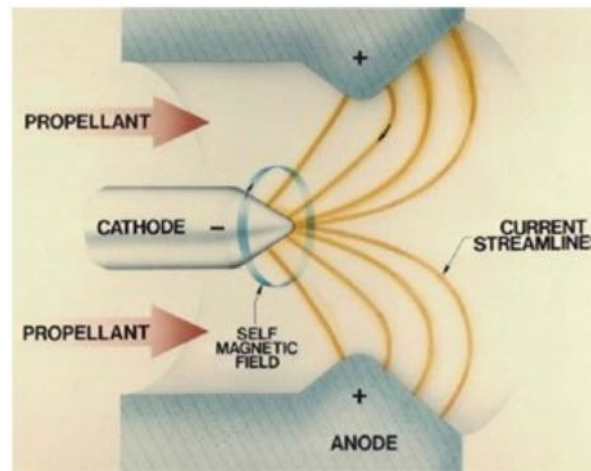


ARChES

Aerothermodynamics
Plasmadynamics

Electromagnetism

Radiative transfer



Magnetohydrodynamic Thruster

<https://www.nasa.gov/centers/glenn/about/fs22grc.html>, 08/10/2018

With **minimal investment** the toolkit can provide modern simulation toolkit for optimizing **in-space electric propulsion systems** such as:

- Magnetohydrodynamic Thruster
- Pulsed Plasma Thruster
- Hall Effect Thruster
- Helicon Double Layer Thruster
- Variable Specific Impulse Magnetoplasma Rocket



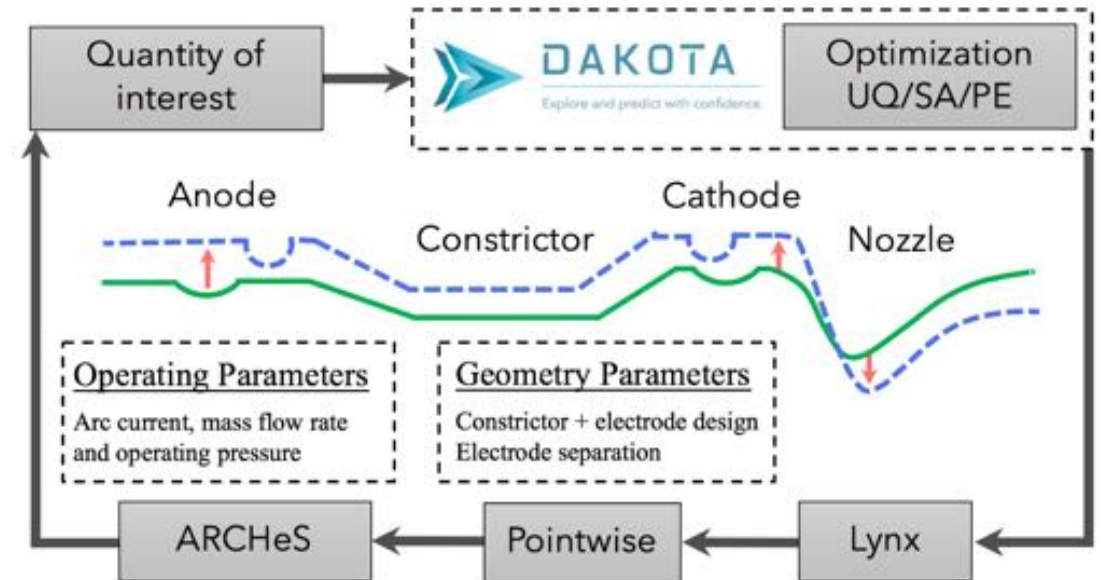
Future work



ARChES improvement

- **MHD physical models**
Magnetic flux equations with regions of zero conductivity
- **2 temperature formulation**
Very thin layer where the electron temperature and heavy separate due to strong magnetic field
- **Thermo/Transport/Chemistry**
Elemental conservation and finite-rate
- **Radiation**
Opacity tables with variable elemental composition
- **Electrodes surface response**
Melt & evaporation of copper formulation
Coupling to PATO
- **Validation experiments**

Coupling to optimization library



ARChES will be coupled to DAKOTA to enable optimization of operating parameters to achieve quantity of interest



Experimental validation plan

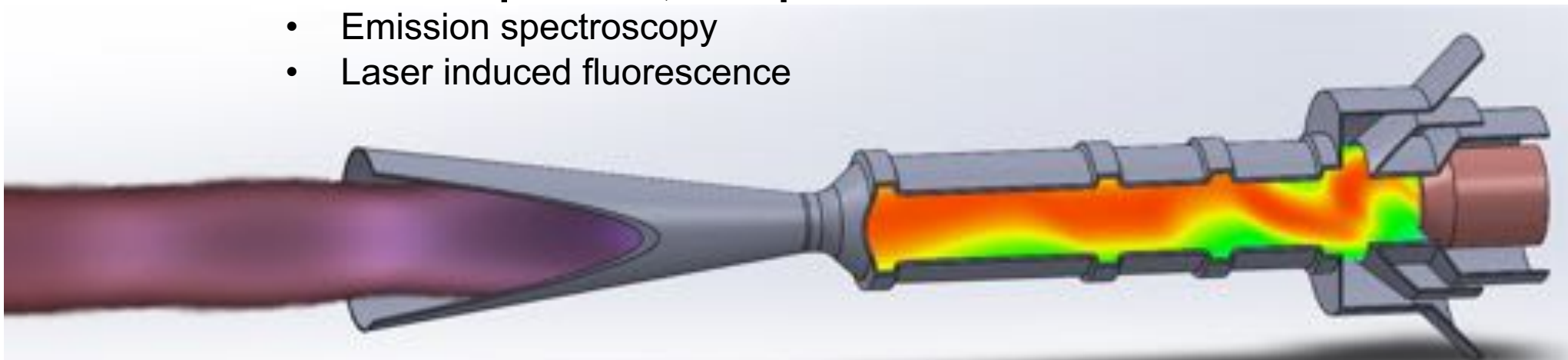


- **Electrode arc detachment events**
 - High frequency electrode current probe
 - High frequency electrode voltage probe
- **Thermal and electromagnetic boundary conditions**
 - Electrostatic (Langmuir) probes
 - External magnetic probes
 - Laser induced fluorescence
- **Internal temperature, flow profiles**
 - Emission spectroscopy
 - Laser induced fluorescence

mARC 2.0

TS division

NASA Ames Research Center

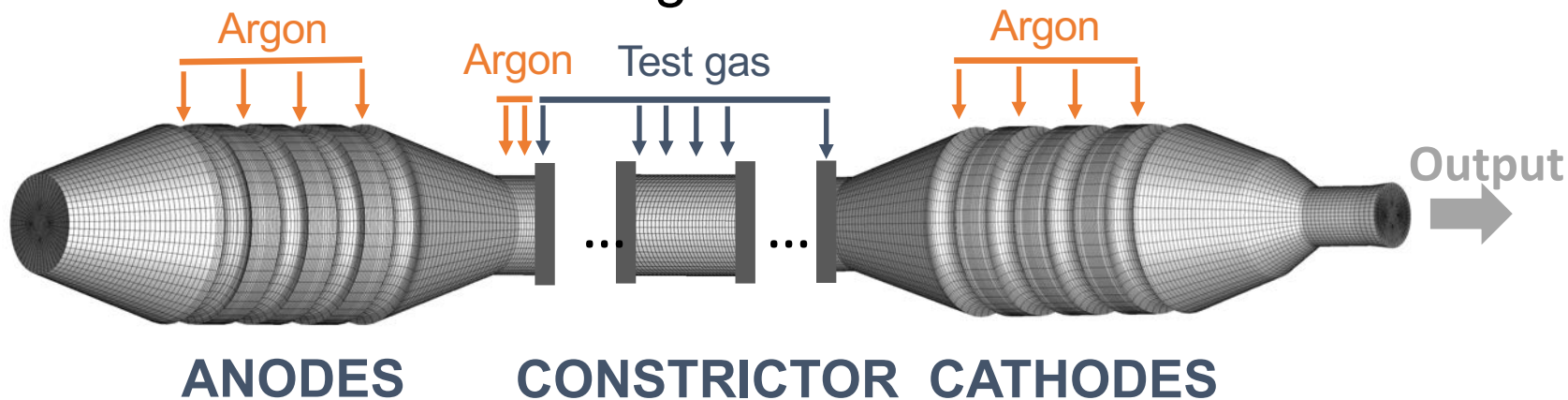




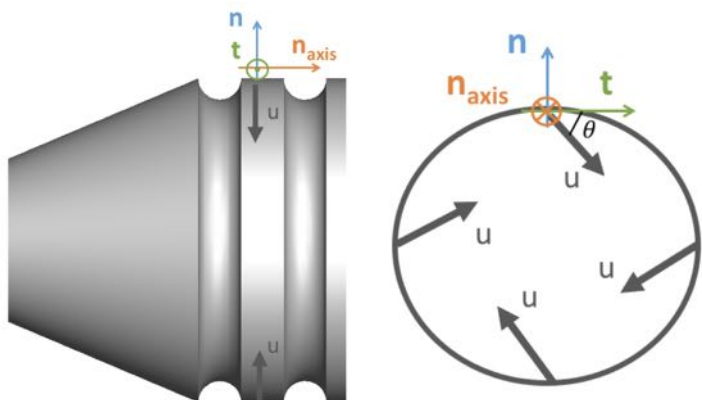
Important features



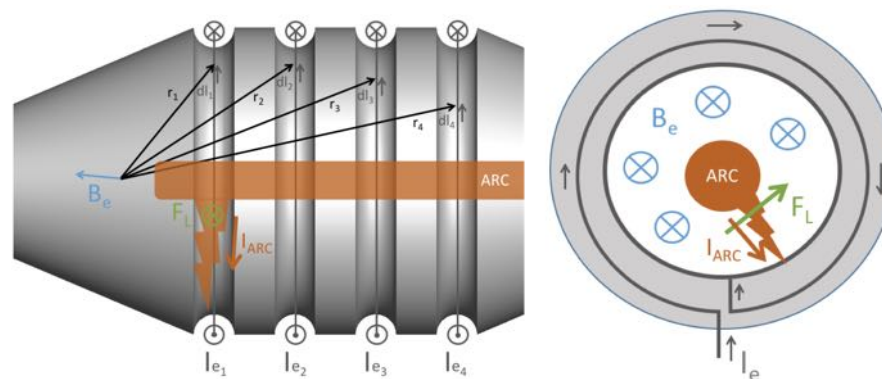
Variable gas mixture



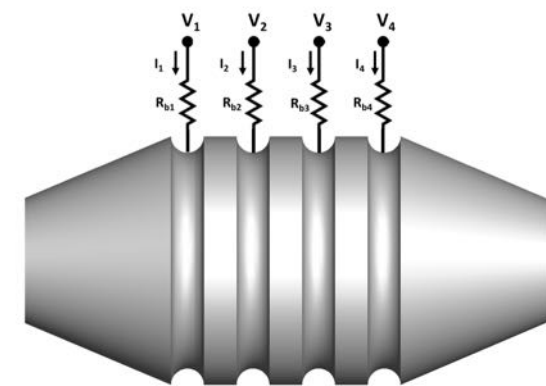
Swirl



Internal magnetic drive

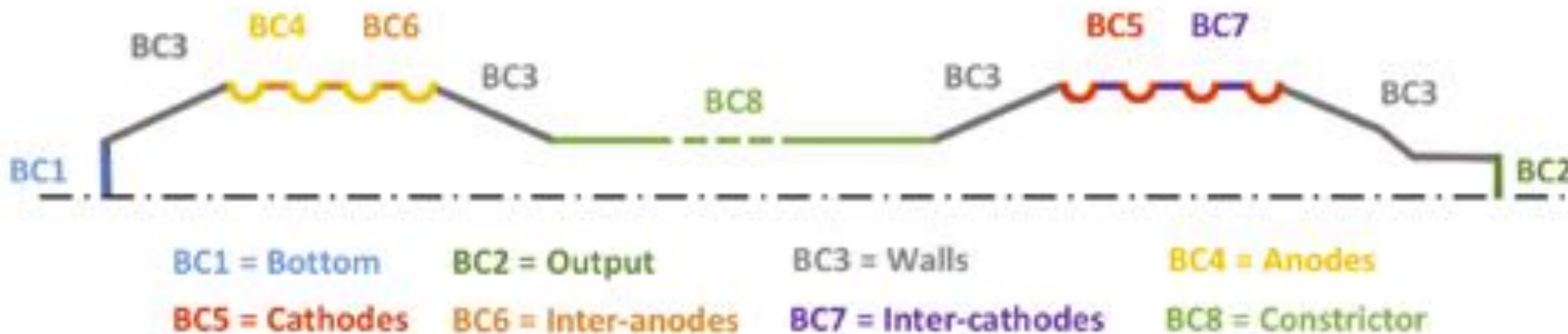


Electrical ballast





Boundary conditions



	p [Pa]	T [K]	u [m/s]
$BC1$	$\partial_x p \cdot n = 0$	T_w	$u = 0$
$BC2$	$\partial_x p \cdot n = 0$	$\partial_x T \cdot n = 0$	$\partial_x u \cdot n = 0$
$BC3$	$\partial_x p \cdot n = 0$	T_w	$u = 0$
$BC4$	$\partial_x p \cdot n = 0$	$\partial_x T \cdot n = 0$	$u = 0$
$BC5$	$\partial_x p \cdot n = 0$	$\partial_x T \cdot n = 0$	$u = 0$
$BC6$	$\partial_x p \cdot n = 0$	T_w	\dot{m}_{BC6}
$BC7$	$\partial_x p \cdot n = 0$	T_w	$u = 0$
$BC8$	$\partial_x p \cdot n = 0$	T_w	\dot{m}_{BC8}

BC of the Navier-Stokes equations

	ϕ_{imp} [V]	A_{imp} [T · m]	A_e [T · m]
$BC1$	$\partial_x \phi_{imp} \cdot n = 0$	$A_{imp} = 0$	$A_e = 0$
$BC2$	$\partial_x \phi_{imp} \cdot n = 0$	$A_{imp} = 0$	$A_e = 0$
$BC3$	$\partial_x \phi_{imp} \cdot n = 0$	$A_{imp} = 0$	$A_e = 0$
$BC4$	$I_{imp,tot}$	$A_{imp} = 0$	$A_e = 0$
$BC5$	$\phi_{imp,c_i}^n = \phi_G - R_b I_{imp,c_i}^{n-1}$	$A_{imp} = 0$	$A_e = 0$
$BC6$	$\partial_x \phi_{imp} \cdot n = 0$	$A_{imp} = 0$	$A_e = 0$
$BC7$	$\partial_x \phi_{imp} \cdot n = 0$	$A_{imp} = 0$	$A_e = 0$
$BC8$	$\partial_x \phi_{imp} \cdot n = 0$	$A_{imp} = 0$	$A_e = 0$

BC of the Maxwell equations