Performance Enhancements for the Lattice-Boltzmann Solver in the LAVA Framework
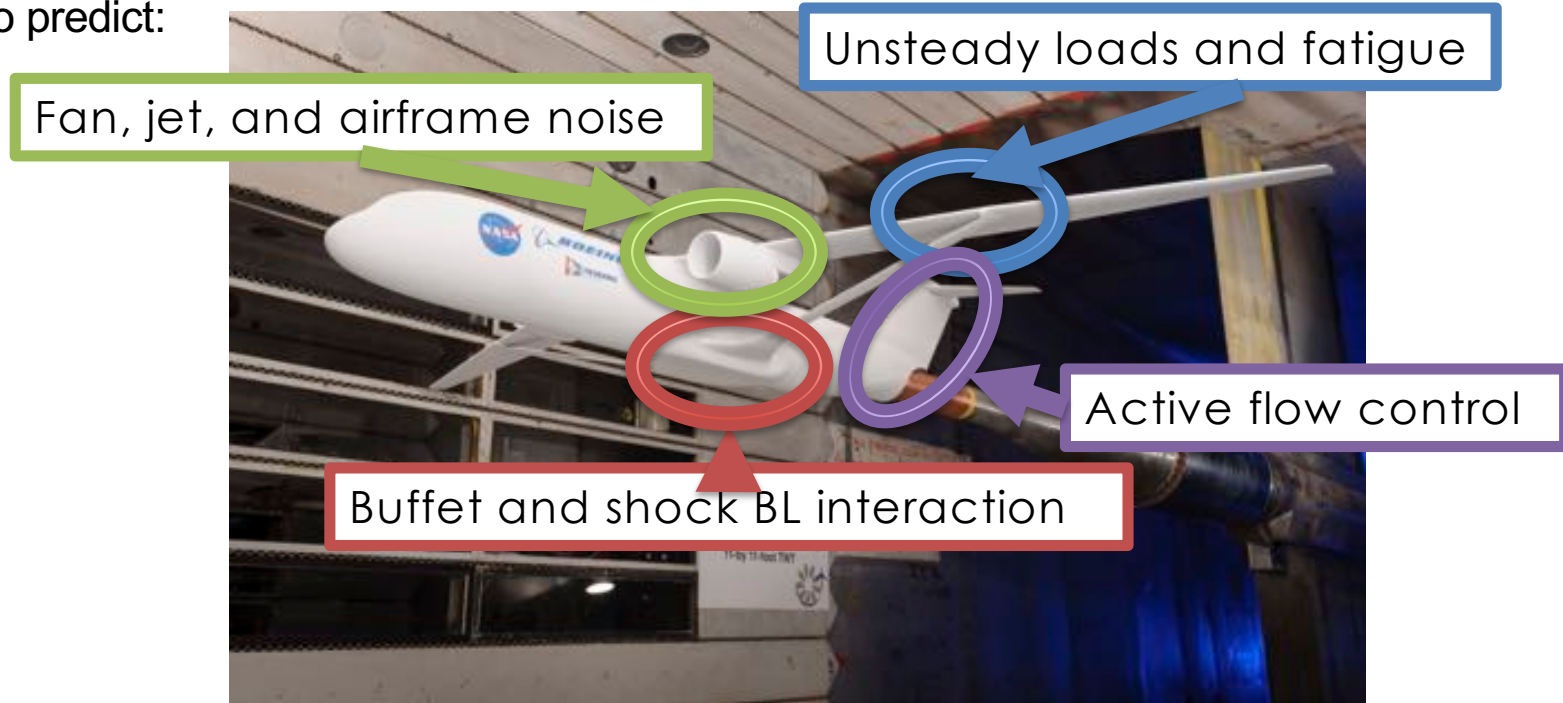
ICCFD10-2018-101

Michael Barad, Joseph Kocheemoolayil,
Gerrit Stich, and Cetin Kiris
Computational Aerosciences Branch
NASA Ames Research Center

ICCFD10 2018
July 9-13, Barcelona, Spain

Motivation

✓ **Increase predictive use of computational aerosciences capabilities for next generation aviation and space vehicle concepts.**
  - The next frontier is to use wall modeled and/or wall resolved large-eddy simulation (LES) to predict:
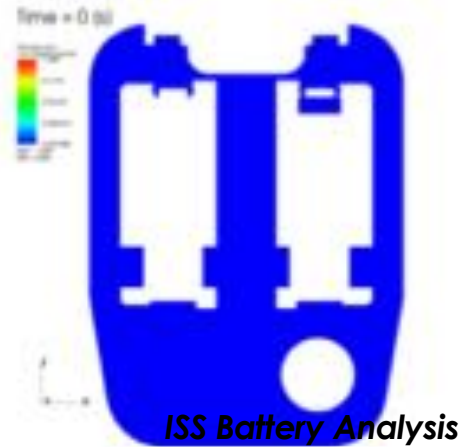

Fan, jet, and airframe noise

Unsteady loads and fatigue

Active flow control

Buffet and shock BL interaction

✓ **Need novel techniques for reducing the computational resources consumed by current high-fidelity CAA**
  - Need routine acoustic analysis of aircraft components at full-scale Reynolds number from first principles
  - **Need an order of magnitude or more reduction in wall time to solution!**
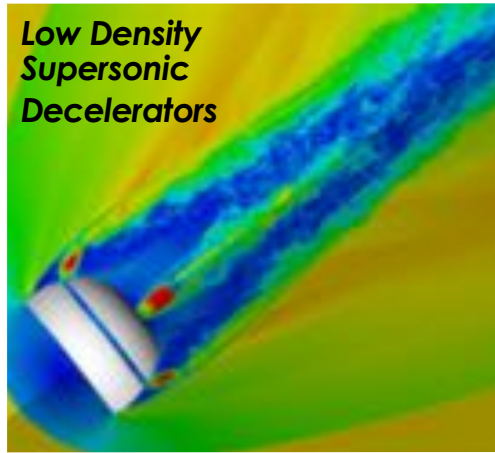
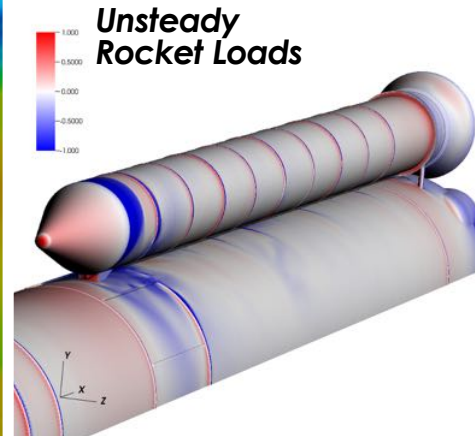Many successful applications of High-Performance High-Fidelity Cartesian methods to NASA Mission critical applications
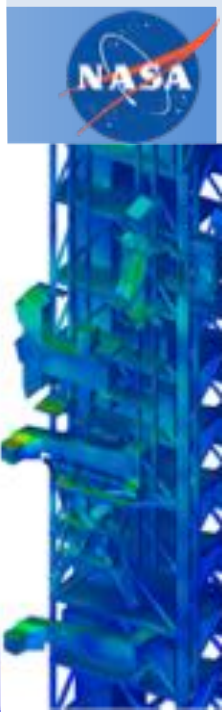
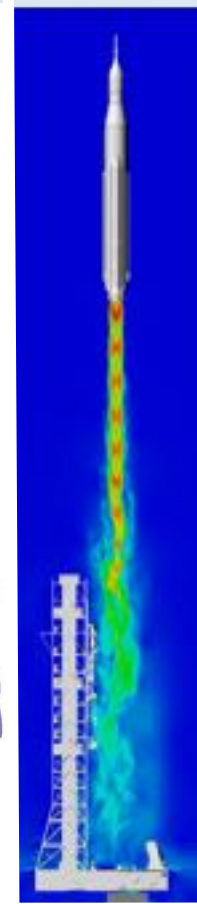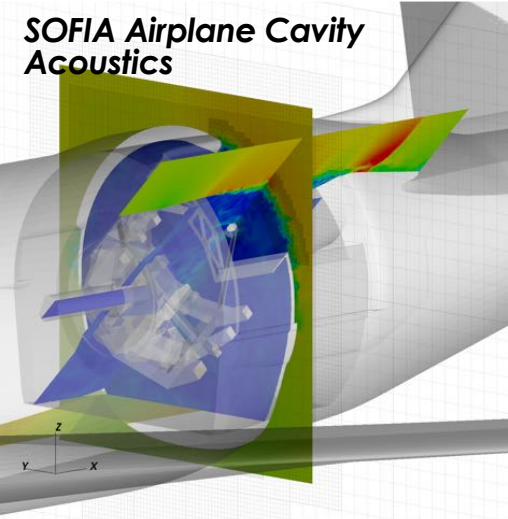*Launch Abort System Analysis for Orion*

ISS Battery Analysis
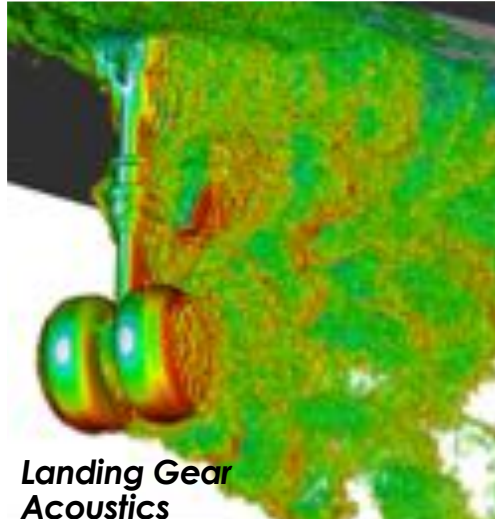
Low Density Supersonic Decelerators

Unsteady Rocket Loads

Launch Pad Design

SOFIA Airplane Cavity Acoustics

Landing Gear Acoustics

Contra-Rotating Open Rotor Propulsion

Challenges in Computational Aero-Acoustics

✓ **Computational Requirements**

- Space-time resolution requirements for acoustics problems are demanding.
- Resources used for recent Cartesian Navier-Stokes simulations:
  - Launch Environment: ~200 million cells, ~7 days (1000 cores)
  - Parachute: 200 million cells, 3 days (2000 cores)
  - Contra-Rotating Open Rotor: 360 million cells, 14 days (1400 cores)
  - Launch Abort System: 400 million cells, 28 days (2000 cores)
  - Landing Gear: 298 million cells, 20 days (3000 cores)

Lattice-Boltzmann Method (LBM)

$$\underbrace{f_i(\vec{x} + c\vec{e}_i\Delta t, t + \Delta t) - f_i(\vec{x}, t)}_{\text{Streaming}} = \underbrace{\frac{1}{\tau}(f_i(\vec{x}, t) - f_i^{eq}(\vec{x}, t))}_{\text{Collision}}$$

✓ **Physics:**

- Governs space time evolution of Density Distribution Functions
- Equilibrium distribution functions are truncated Maxwell-Boltzmann distributions
- Relaxation time related to kinematic viscosity
- Pressure related to density through the isothermal ideal gas law
- Lattice Boltzmann Equations (LBE) recover the Navier-Stokes equations in the low Mach number limit

✓ **Numerics:**

- Extremely efficient 'collide at nodes and stream along links' discrete analog to the Boltzmann equation
- Particles bound to a regularly spaced lattice collide at nodes relaxing towards the local equilibrium (RHS)
- Post-collision distribution functions hop on to neighboring nodes along the lattice links (LHS) – Exact, dissipation-free advection from simple 'copy' operation
- Macroscopic quantities such as density and momentum are moments of the density distribution functions in the discrete velocity space

Lattice-Boltzmann Method (LBM)

✓ **LBM Benefits:**

- Ultra high performance: excellent data locality, vectorizable, scalable.

- Minimal numerical dissipation that is critical for computational aeroacoustics, and ideal for Large Eddy Simulations.

- Simulation of arbitrarily complex geometry with high performance structured adaptive mesh refinement is straight forward, bypassing manual and/or expensive meshing bottlenecks.

✓ **NASA's LAVA-LBM**

- **Progress to Date:**
  - LAVA Cartesian infrastructure has been re-factored into Navier-Stokes (NS) and LBM. Existing LAVA Cartesian data structures and algorithms are utilized.
  - Parallel Structured Adaptive Mesh Refinement (SAMR) meshing, robust collision models, second-order boundary conditions, all implemented.
  - Verification & validation: Taylor-Green vortex, flow past a cylinder, and nose landing gear.
  - A 12 to 15 times speedup compared to LAVA-Cart-NS was demonstrated for landing gear.
- **Current Efforts:**
  - Performance
    - Enhanced Accuracy at Coarse/Fine interface ⎤
    - Parallel Efficiency and Scaling   ⎦ **Focus on these for this paper**
  - Moving Geometry ⎤ In testing phase
  - Wall Modeling ⎦
  - High Mach formulation ← Initial stages of development

"Lattice Boltzmann and Navier-Stokes Cartesian CFD Approaches for Airframe Noise Predictions", Barad, Kocheemoolayil, Kiris, AIAA 2017-4404

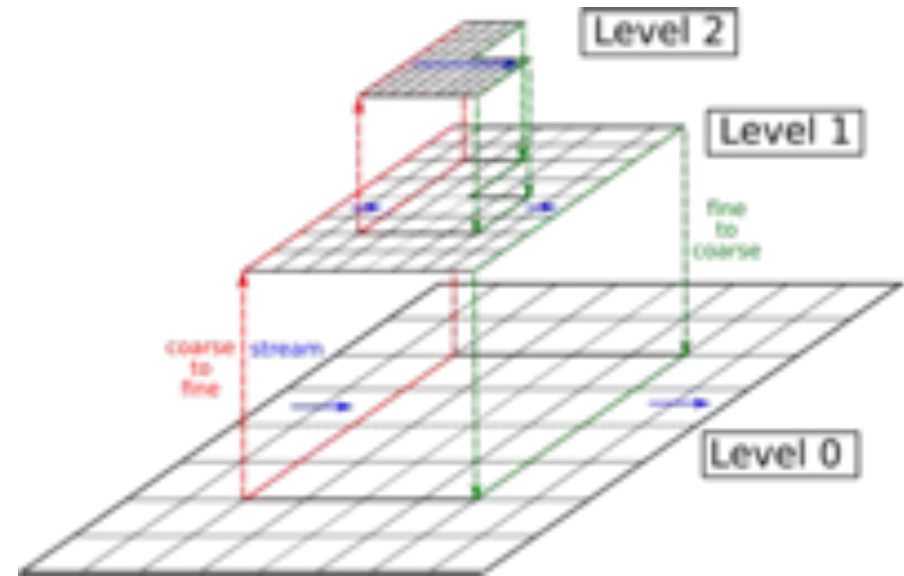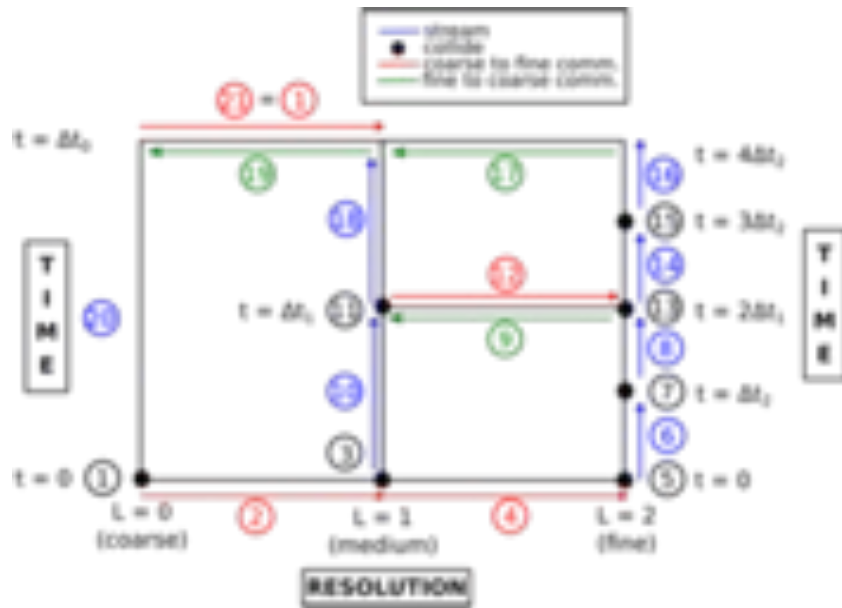LBM @ 1.6 billion – Velocity Magnitude at Centerline

Lessons from LAVA LBM Landing Gear Simulations

- Previously demonstrated the LBM approach on the AIAA BANC III Workshop Landing Gear problem IV.
  - Computed results compare well with the experimental data
  - 12-15 times speed-up was observed between LBM and NS calculations.

- After completing the LG study, we knew that the code can be even faster!
  - Node usage not optimal with pure MPI programming model → go to **hybrid MPI/OpenMP**
  - Not enough parallelism for modern hardware → add concurrency with **tiling**
  - Moving geometry applications introduce many complexities:
    - Load balancing, points to **bigger boxes, fewer MPI ranks per node, and dynamic thread scheduling within boxes**
    - Geometry kernels are expensive, CPU vendor supplied ray-tracing libraries work best with **hybrid MPI/OpenMP**
  - Exciting **new hardware is coming to HPC**…codes need to be:
    - Ready for extremely **high concurrency**,
    - Using **memory bandwidth** efficiently
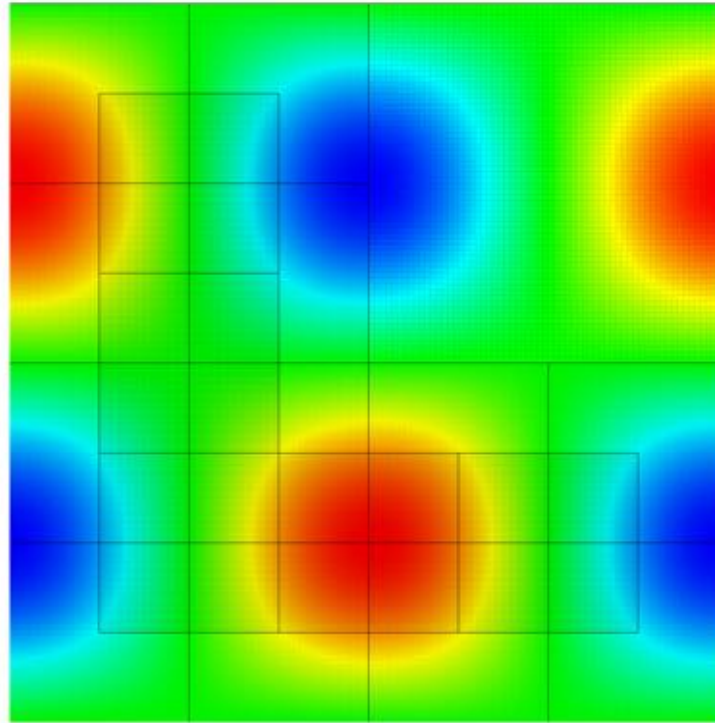
Conservative Coarse/Fine Interface

Sketch of conservative recursive sub-cycling algorithm.
- Block structured AMR showing 3 levels of refinement by factor 2.
- Arrows indicate direction of information propagation:
    - streaming (blue),
    - coarse-to-fine communication (red),
    - fine-to-coarse communication (green).

Refs: Schornbaum and Rude 2016, Rohde et al 2006, Chen et al 2006.

Conservative Coarse/Fine Interface

2D Conservation Test:

Step	Mass	X-Momentum	Y-Momentum
0	3.4842903112786844e-03	-1.3321944431884503e-07	1.3321944427329724e-07
50	3.4842903112787802e-03	-1.3321944436875571e-07	1.3321944451058669e-07
Error:	9.58434720477186e-17	4.99106752631308e-17	2.37289450970155e-16

Conservative Coarse/Fine Interface

3D Conservation Test:

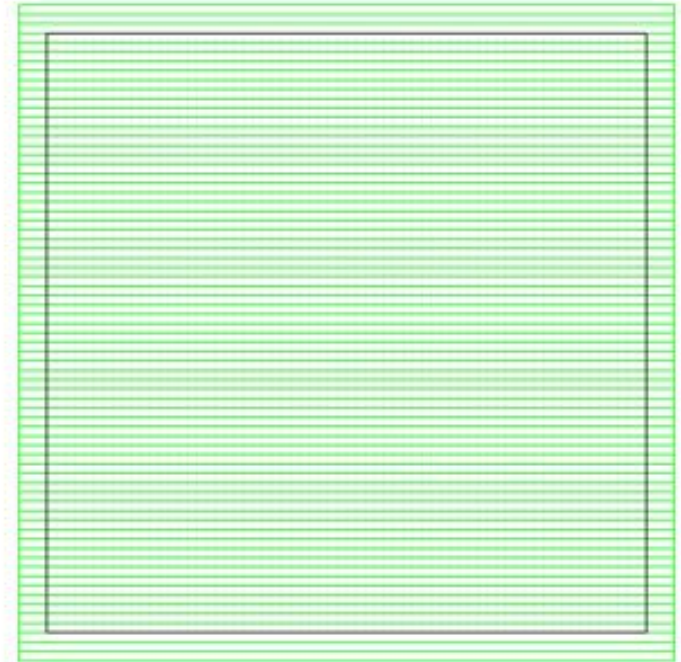| Step	Mass	X-Momentum	Y-Momentum	Z-Momentum
0	3.4842903112065815e-03	8.99705362799347e-08	-8.99705362994044e-08	1.47753463427836e-20
50	3.4842903112065789e-03	8.99705363602753e-08	-8.99705362695844e-08	-3.39618771662516e-19
Error:	2.602085213965e-18	8.034059859103e-17	2.982000666632e-17	3.543941180053e-19

More Parallelism: Tiling

(a) Regular Tiles

(b) Pencil Tiles

Different tile types for a single box:

    (a) regular tiles (8^D), including inner (blue) and outer (red); and

    (b) pencil tiles (green) for contiguous memory accesses

The box shown has 64^D cells, plus 3 ghost layers. 3D tiles are conceptually similar.

More Parallelism: Tiling + OpenMP

Adding another level of parallelism has many benefits:

- **Loop collapse:** OpenMP over boxes on a proc & tiles in each box

```
#pragma omp for schedule(dynamic) collapse(2)
for (int ibox = 0; ibox < nbox; ++ibox)
  {
    for (int itile = 0;itile < ntile; ++itile)
      {
        work(ibox,itile);
      }
  }
```
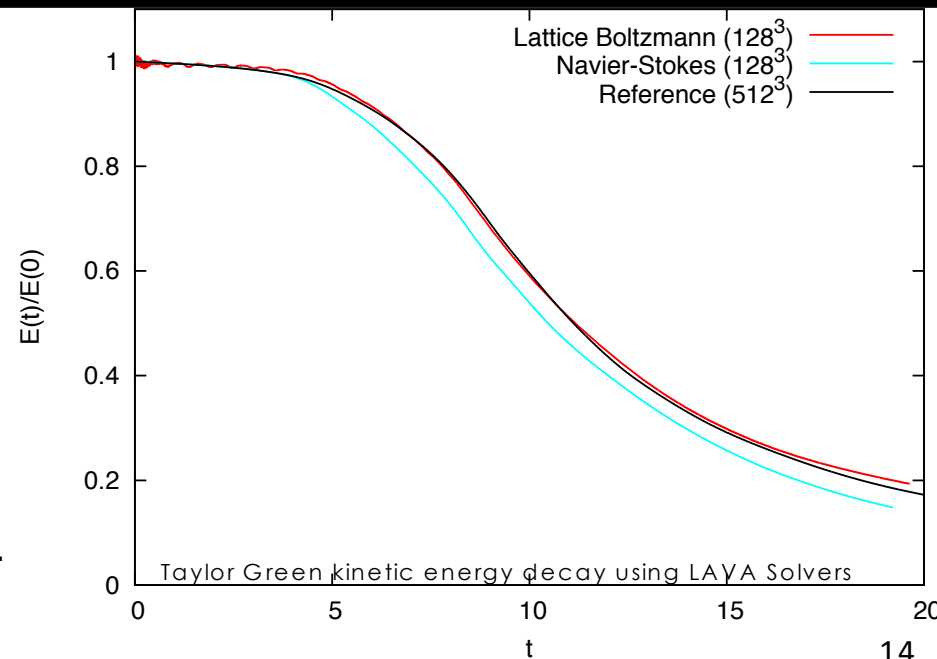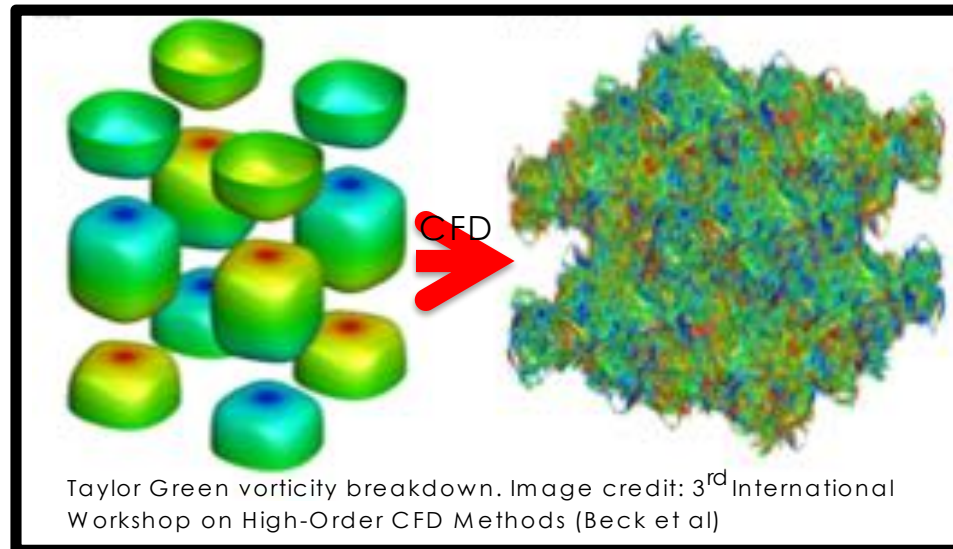
- **Improved load balancing** for irregularities:
  - Complex geometry
  - AMR
- **Bigger boxes** are possible which improves surface/volume ratios and reduces MPI expense
- **Asynchronous communication** is enabled:
  - Outer tiles are computed first, then non-blocking MPI sends
  - Inner tiles then computed
  - Finish MPI comms

LAVA LBM: Verification and Validation

TURBULENT TAYLOR GREEN VORTEX BREAKDOWN TEST CASE:

- **Motivation**:
  - Simple low speed workshop case for testing high-order solvers
  - Illustrates ability of solver to simulate turbulent energy cascade
  - Periodic boundary conditions
- **Setup**:
  - Analytic initial condition
    - Mach = 0.1
    - Reynolds Number = 1600
  - Triply periodic flow in a box
- **Comparisons:**
  - LAVA's Lattice Boltzmann (LB) solver captures the turbulent kinetic energy cascade from large scales to small scales extremely well.
  - Performance compared to LAVA's Cartesian grid Navier-Stokes WENO solver showed a factor of 50 speedup.

CFD

Taylor Green vorticity breakdown. Image credit: 3rd International Workshop on High-Order CFD Methods (Beck et al)

Taylor Green kinetic energy decay using LAVA Solvers

Taylor-Green Vortex (TGV) test case:

- 256^3 cells per node problem size, unless noted otherwise

- Single static level (i.e. no AMR issues)
- No geometry
- 64 time-steps performed, time to solution measured
- All simulations conducted on Skylake nodes on NASA's Pleiades supercomputer (1 node has 2 sockets, 20 physical cores per socket)
- Focused on 3 versions of the code:
  - Baseline (no tiling)
  - Tiling with data copies to tiles   See paper for these results
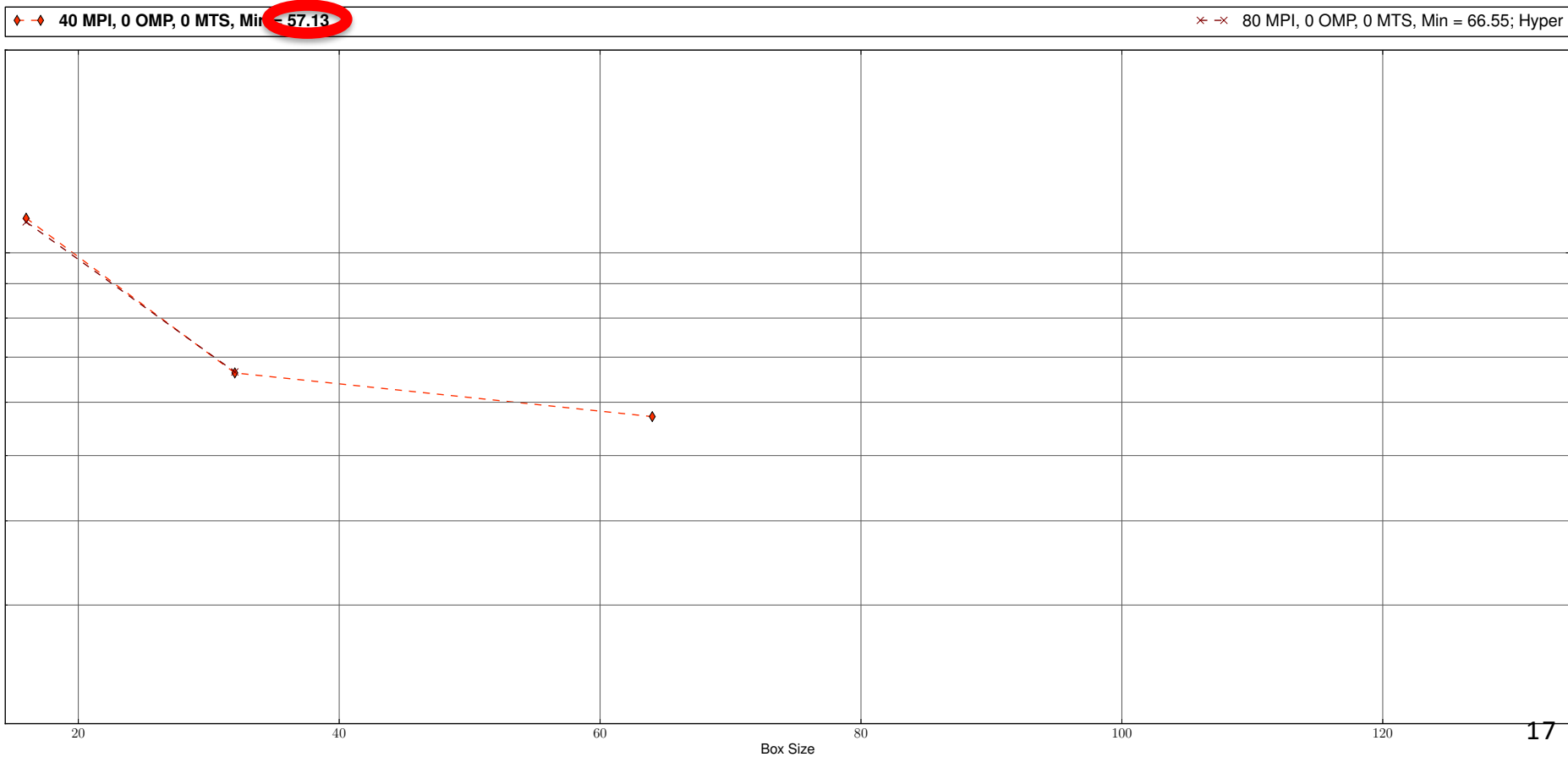 - Tiling without data copies to tiles

- **Parameter Space and Terminology:**
  - MBS: Max Box Size (i.e. box size) [16,32,64,128,256]
  - MTS: Max Tile Size (i.e. tile size) [0,4,6,8,10,12,16,32,64,128]
  - MPI: Number of ranks / node [1,2,4,8,12,16,20,24,28,32,36,40,50,60,70,80]
  - OMP: Number of OpenMP threads [0,1,2,4,8,12,16,20,24,28,32,36,40,50,60,70,80]
  - Hyper: Hyperthreading (i.e. over-subscribing cores) [no/yes]
  - Nodes: [1,8,64,512]

- **Three profiling analyses were performed:**
  1. Single packed-node parameters study
     → investigate MBS vs MTS vs MPI vs OMP parameter space
  2. Single-node strong scaling study
     → investigate parallel scaling on a single node
  3. Multi-node weak scaling study
     → investigate parallel scaling across nodes, keeping work per node fixed
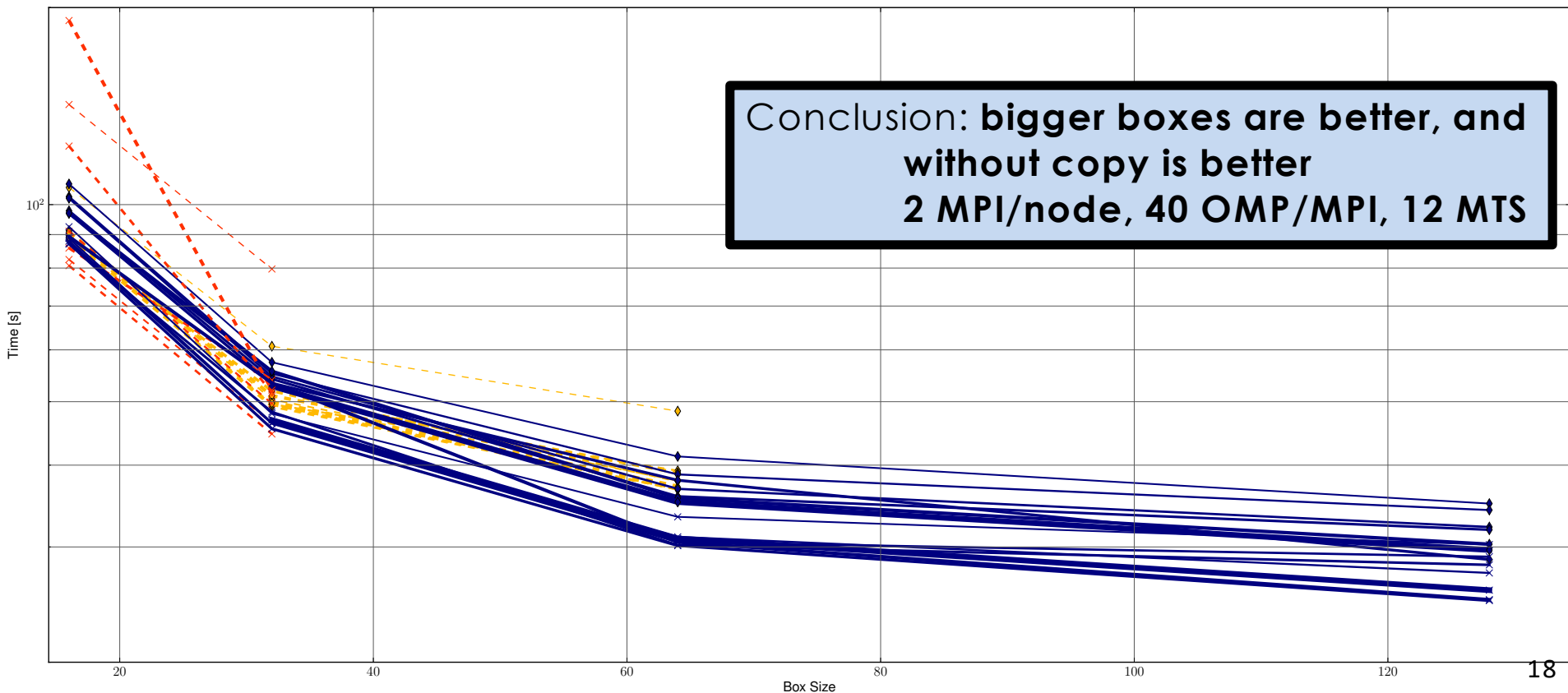
Baseline (no tiling), sensitivity to box size (MBS):

Optimized, **without copy** into tiles, sensitivity to box size (MBS):

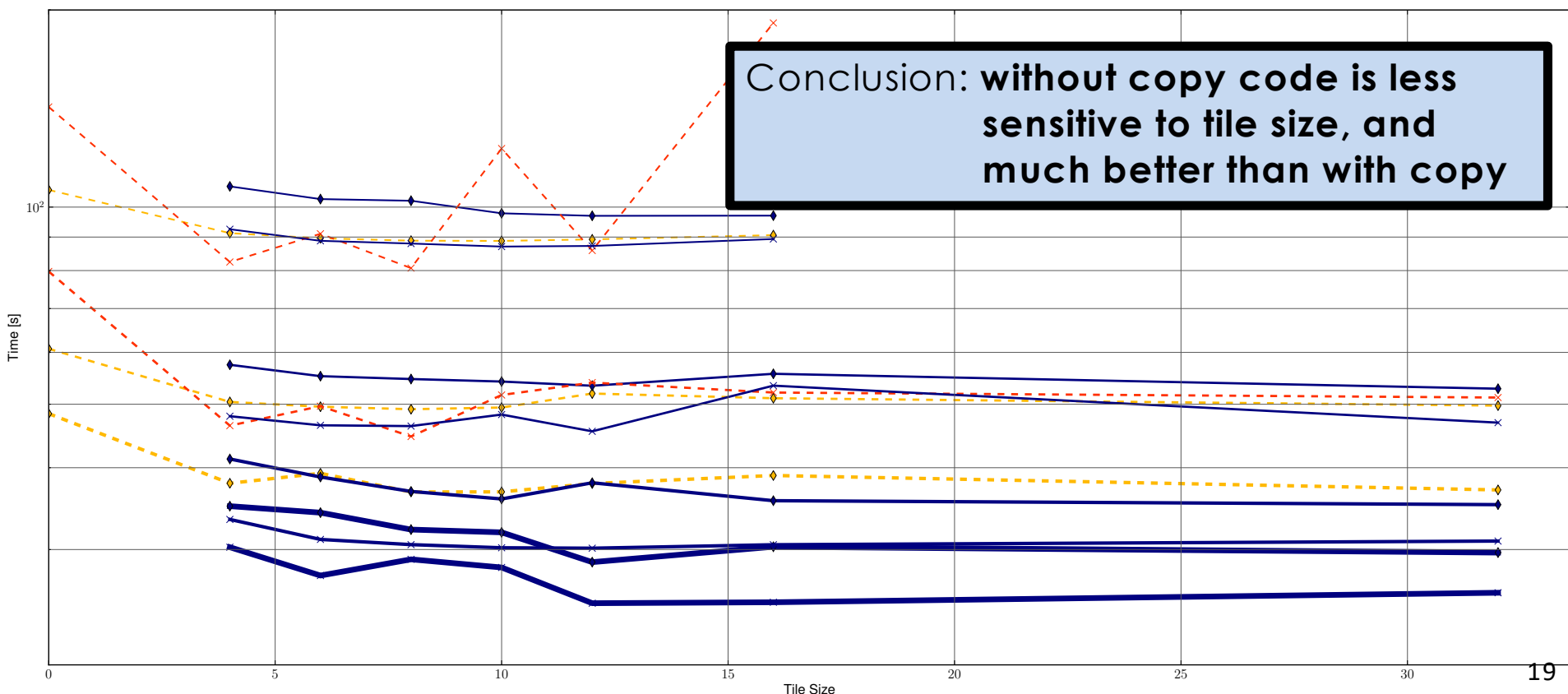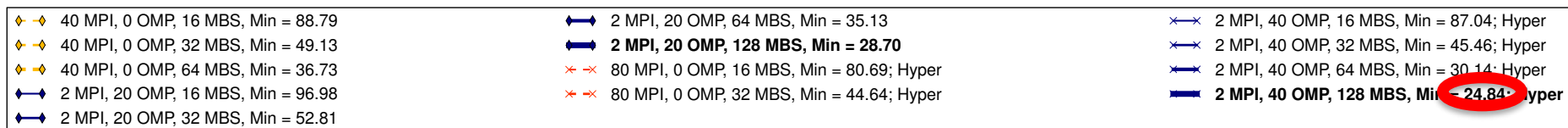|	
40 MPI, 0 OMP, 0 MTS, Min = 48.37	2 MPI, 20 OMP, 8 MTS, Min = 32.18
40 MPI, 0 OMP, 4 MTS, Min = 37.87	2 MPI, 20 OMP, 10 MTS, Min = 31.87
40 MPI, 0 OMP, 6 MTS, Min = 39.22	**2 MPI, 20 OMP, 12 MTS, Min = 28.70**
40 MPI, 0 OMP, 8 MTS, Min = 36.73	2 MPI, 20 OMP, 16 MTS, Min = 30.29
40 MPI, 0 OMP, 10 MTS, Min = 36.74	**2 MPI, 20 OMP, 32 MTS, Min = 29.68**
40 MPI, 0 OMP, 12 MTS, Min = 37.84	80 MPI, 0 OMP, 0 MTS, Min = 79.74; Hyper
40 MPI, 0 OMP, 16 MTS, Min = 38.92	80 MPI, 0 OMP, 4 MTS, Min = 46.37; Hyper
40 MPI, 0 OMP, 32 MTS, Min = 36.99	80 MPI, 0 OMP, 6 MTS, Min = 49.67; Hyper
2 MPI, 20 OMP, 4 MTS, Min = 34.94	80 MPI, 0 OMP, 8 MTS, Min = 44.64; Hyper
2 MPI, 20 OMP, 6 MTS, Min = 34.15	80 MPI, 0 OMP, 10 MTS, Min = 51.65; Hyper

80 MPI, 0 OMP, 12 MTS, Min = 53.96; Hyper
80 MPI, 0 OMP, 16 MTS, Min = 52.10; Hyper
80 MPI, 0 OMP, 32 MTS, Min = 51.19; Hyper
2 MPI, 40 OMP, 4 MTS, Min = 30.25; Hyper
2 MPI, 40 OMP, 6 MTS, Min = 27.38; Hyper
2 MPI, 40 OMP, 8 MTS, Min = 28.98; Hyper
2 MPI, 40 OMP, 10 MTS, Min = 28.17; Hyper
2 MPI, 40 OMP, 12 MTS, Min = 24.84; Hyper
2 MPI, 40 OMP, 16 MTS, Min = 24.92; Hyper
2 MPI, 40 OMP, 32 MTS, Min = 25.77; Hyper

Conclusion: **bigger boxes are better, and without copy is better**
2 MPI/node, 40 OMP/MPI, 12 MTS

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Optimized, **without copy** into tiles, sensitivity to tile size (MTS):

Legend
40 MPI, 0 OMP, 16 MBS, Min = 88.79
40 MPI, 0 OMP, 32 MBS, Min = 49.13
40 MPI, 0 OMP, 64 MBS, Min = 36.73
2 MPI, 20 OMP, 16 MBS, Min = 96.98
2 MPI, 20 OMP, 32 MBS, Min = 52.81
2 MPI, 20 OMP, 64 MBS, Min = 35.13
2 MPI, 20 OMP, 128 MBS, Min = 28.70
80 MPI, 0 OMP, 16 MBS, Min = 80.69; Hyper
80 MPI, 0 OMP, 32 MBS, Min = 44.64; Hyper
2 MPI, 40 OMP, 16 MBS, Min = 87.04; Hyper
2 MPI, 40 OMP, 32 MBS, Min = 45.46; Hyper
2 MPI, 40 OMP, 64 MBS, Min = 30.14; Hyper
2 MPI, 40 OMP, 128 MBS, Min = 24.84; Hyper

Conclusion: without copy code is less sensitive to tile size, and much better than with copy

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Optimized, **without copy** into tiles:

Hyperthreading region is marked with gray shading

1 MPI, 128 MBS, 12 MTS	4 MPI, 128 MBS, 12 MTS
2 MPI, 128 MBS, 12 MTS	8 MPI, 128 MBS, 12 MTS

-- Linear Scaling

Time [s]

Total Number of Threads [MPI*OMP]

Conclusion: **2 MPI per node (i.e. 1 per socket) has best performance**

Baseline (no tiling):

Optimized, **without copy** into tiles:

Optimized, **without copy** into tiles:

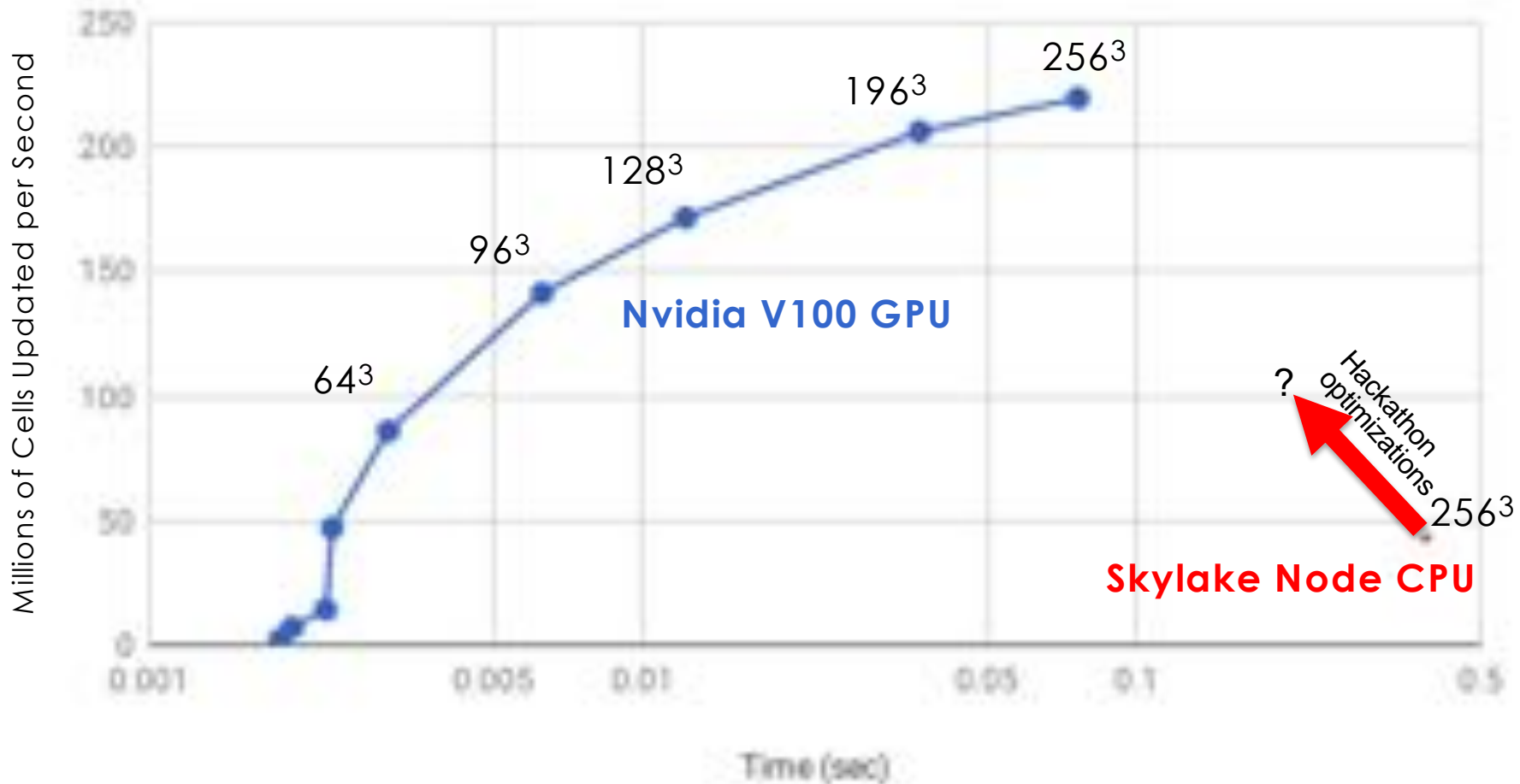| MPI per Node	OMP	MBS	MTS	HT	Nodes=1 N=256 [s]	Nodes=8 N=512 [s]	Nodes=64 N=1024 [s]	Nodes=512 N=2048 [s]
40	0	16	12	No	89.27	89.39	113.08	-
40	0	32	0	No	60.77	62.01	78.59	99.97
40	0	32	12	No	51.92	50.9	64.08	66.86
40	0	64	0	No	48.37	53.36	80.48	124.03
40	0	64	12	No	37.84	39.81	43.95	70.79
1	40	32	12	No	79.67	94.66	95.73	101.52
1	40	64	12	No	52.91	62.54	63.57	75.6
1	40	128	12	No	43.27	52.64	52.92	56.85
80	0	16	12	Yes	85.85	83.32	103.67	-
80	0	32	0	Yes	79.74	61.48	72.8	93.38
80	0	32	12	Yes	53.96	47.63	52.25	61.79
2	40	128	12	Yes	**24.84**	**31.77**	**32.54**	**34.92**

Best practice

~10 billion cells

Bonus: GPU Hackathon 2018

The LAVA team participated in a "GPU Hackathon" in Boulder, CO (06/2018)
Focused on a highly simplified LBM-mini app (single level TGV)

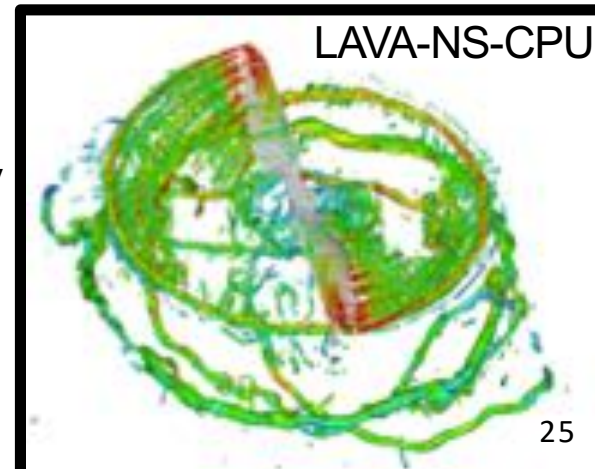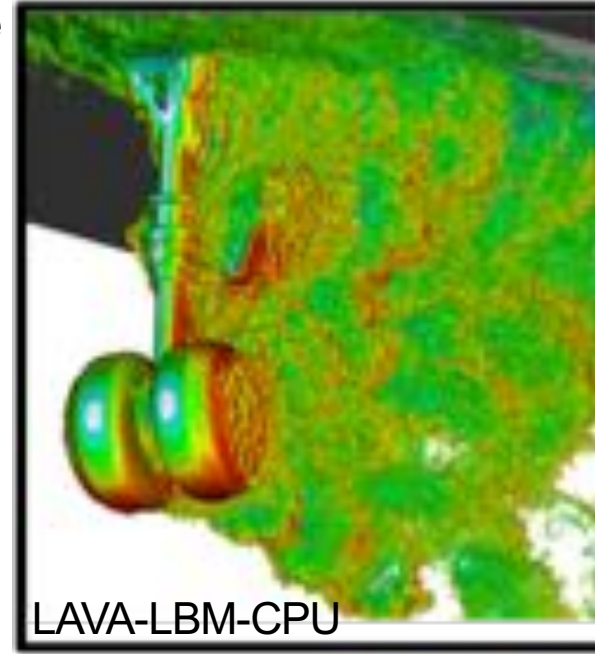Strong scaling with varying problem size:

Remaining Challenges for LAVA-LBM-GPU

The following key operations are implemented efficiently on the CPU, but not yet addressed during the hackathon for GPU:

- MPI parallel

- AMR operators

- Immersed boundaries

  - Fixed geometry

    - Introduces load imbalances at both simulation startup and during time-stepping
    - Treated using structured looping in LAVA -> should map to GPU with some effort

  - Moving geometry

    - Major cost / load imbalances are introduced at every timestep (re-computing geometry intersections, etc).

    - Expense on CPU treated using highly optimized vendor supplied ray-tracing kernels (Embree). Enabling technology for CPU calcs.

    - On CPU this is currently roughly a 1.2-1.5x hit in performance, not sure how this will be addressed on the GPU. Try using NVIDIA OptiX.

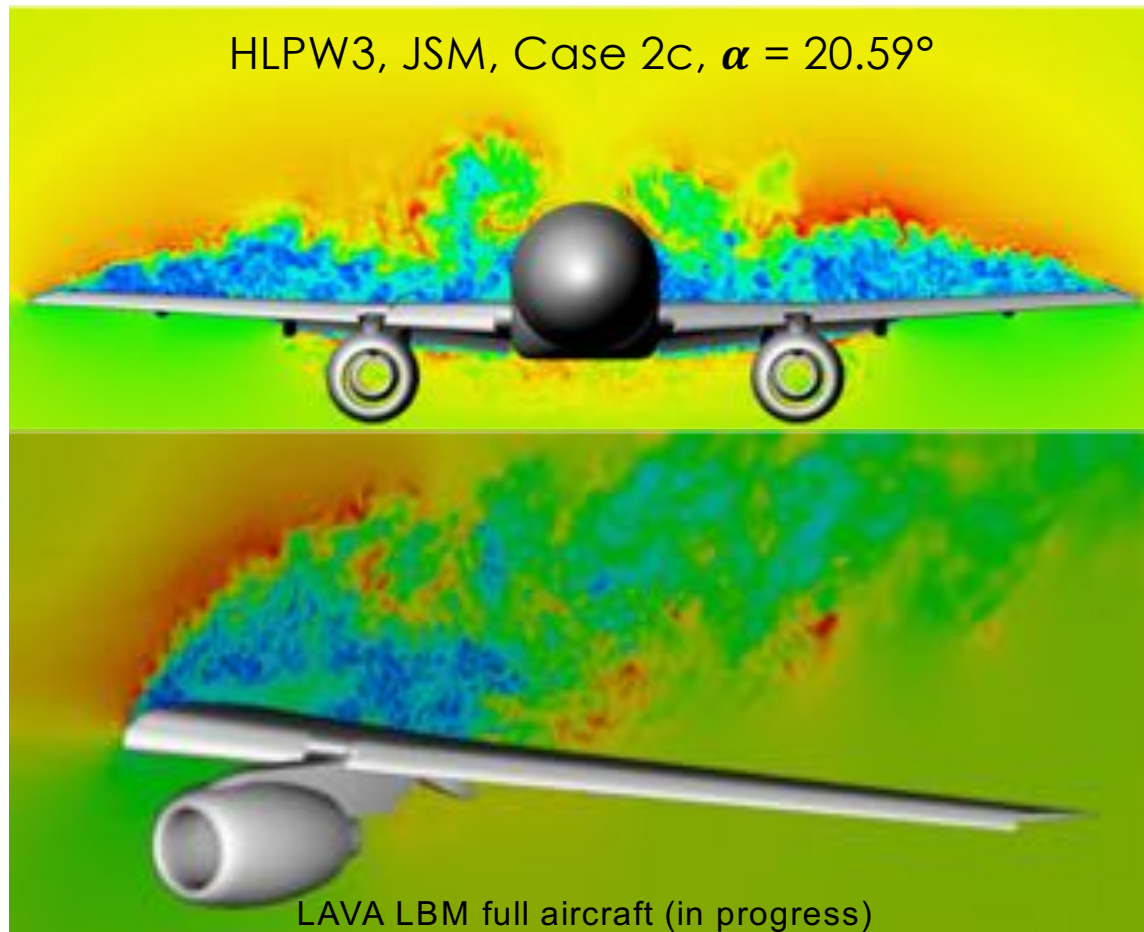LAVA-LBM-CPU

LAVA-NS-CPU

TGV Profiling: Summary

For the simple Taylor-Green Vortex problem:

- Found that copying into small tile sized memory is slower than just using the box based memory layout. Not enough re-use in LBM for cache-blocking.

- **Developed best practices**:
  - Larger boxes are better
  - Tile sizes of 8-12 are superior than smaller or larger
  - Hyperthreading yields a small improvement (~1.16x speedup)
  - 1 MPI per socket, 40 OMP threads per socket (i.e. hyperthreaded)

- Achieved a **2.3x speedup over the baseline** code for a single Skylake-SP CPU node containing 40 physical cores,

- Achieved a 2.14x speedup over the baseline code for 64 Skylake-SP nodes containing 2560 cores

- **Scaled the code almost perfectly to 20480 physical cores** where the problem size was **~10 billion cells**

- LAVA-LBM-GPU mini-app on Nvidia V100 yielded **11.5x speedup vs CPU baseline**. Could result in O(100)x speedup for full-app vs LAVA-NS-CPU.

Next Steps

- Further code optimizations for:
  - Moving geometry and
  - Adaptive meshing
- Improve wall modeling for arbitrarily complex geometry at high Reynolds numbers
- Extend Mach number range to transonic and high speed flows

HLPW3, JSM, Case 2c, α = 20.59°

LAVA LBM full aircraft (in progress)

Acknowledgments

Questions ?