



A relocatable lander to explore Titan's
prebiotic chemistry and habitability

The Dragonfly Entry and Descent System

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Introduction



- **As of June 27, we are going to Titan!**
 - launch 2026, arrival 2034
- **This talk will focus on the Entry and Descent system that will deliver the rotorcraft safely to its release point**
 - see talks by Ken Hibbard (Tuesday) and Doug Adams (tomorrow morning) at this workshop for more details how we get from there to the ground!
- **Dragonfly EDL team combines expertise from NASA Ames/Langley and Lockheed Martin**
 - Aeroshell hardware provider (LM) has extensive hardware expertise for all relevant EDL missions since Pathfinder
- **Titan EDL is well understood with robust performance margins across the subsystems**
 - Extensive heritage exists that can be adapted to this mission application
 - Prior work for Huygens risk review (2004) and In-Space Propulsion Program lay the foundation for current EDL models
 - Titan EDL is more straightforward than Mars in several important ways

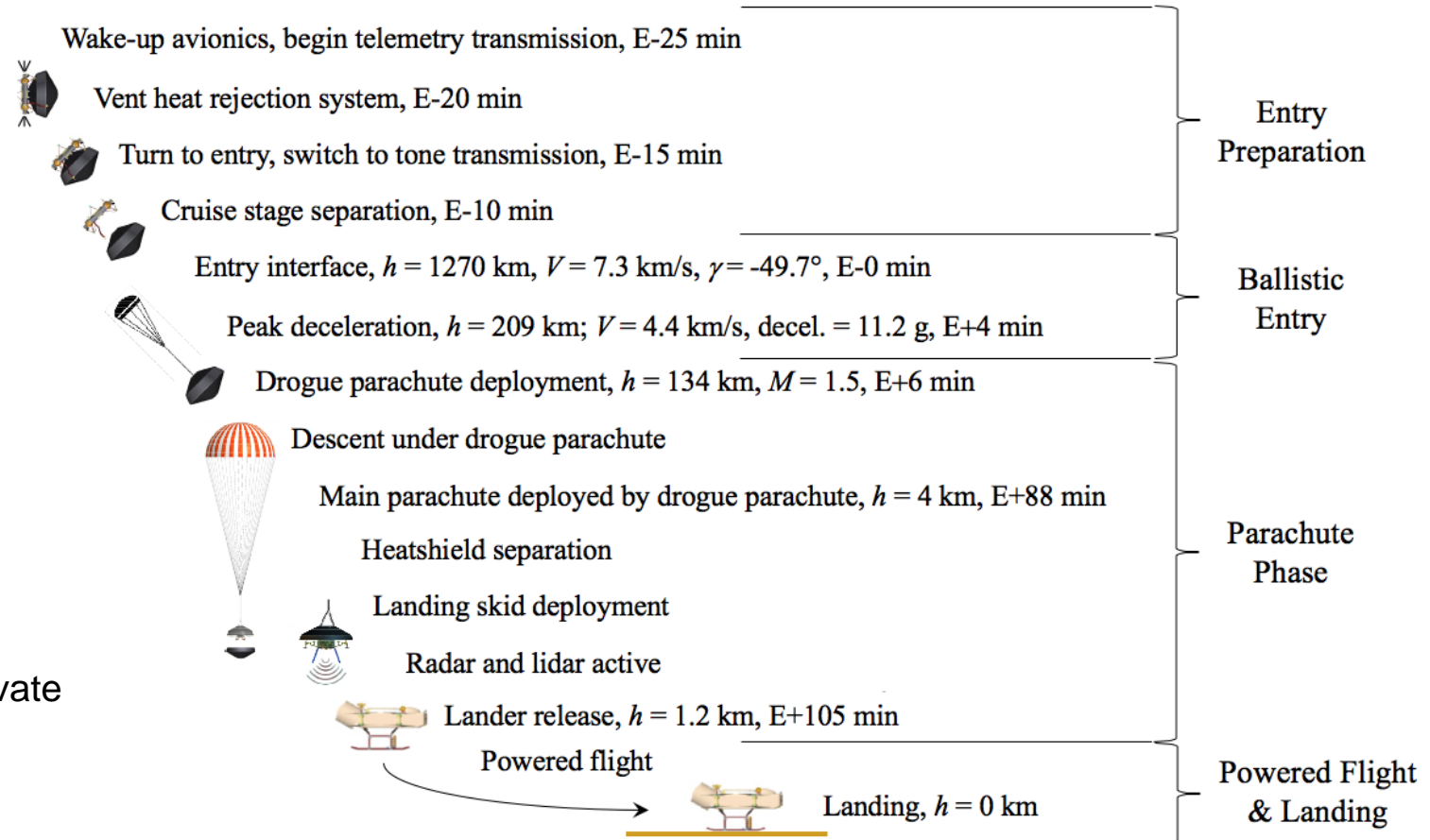


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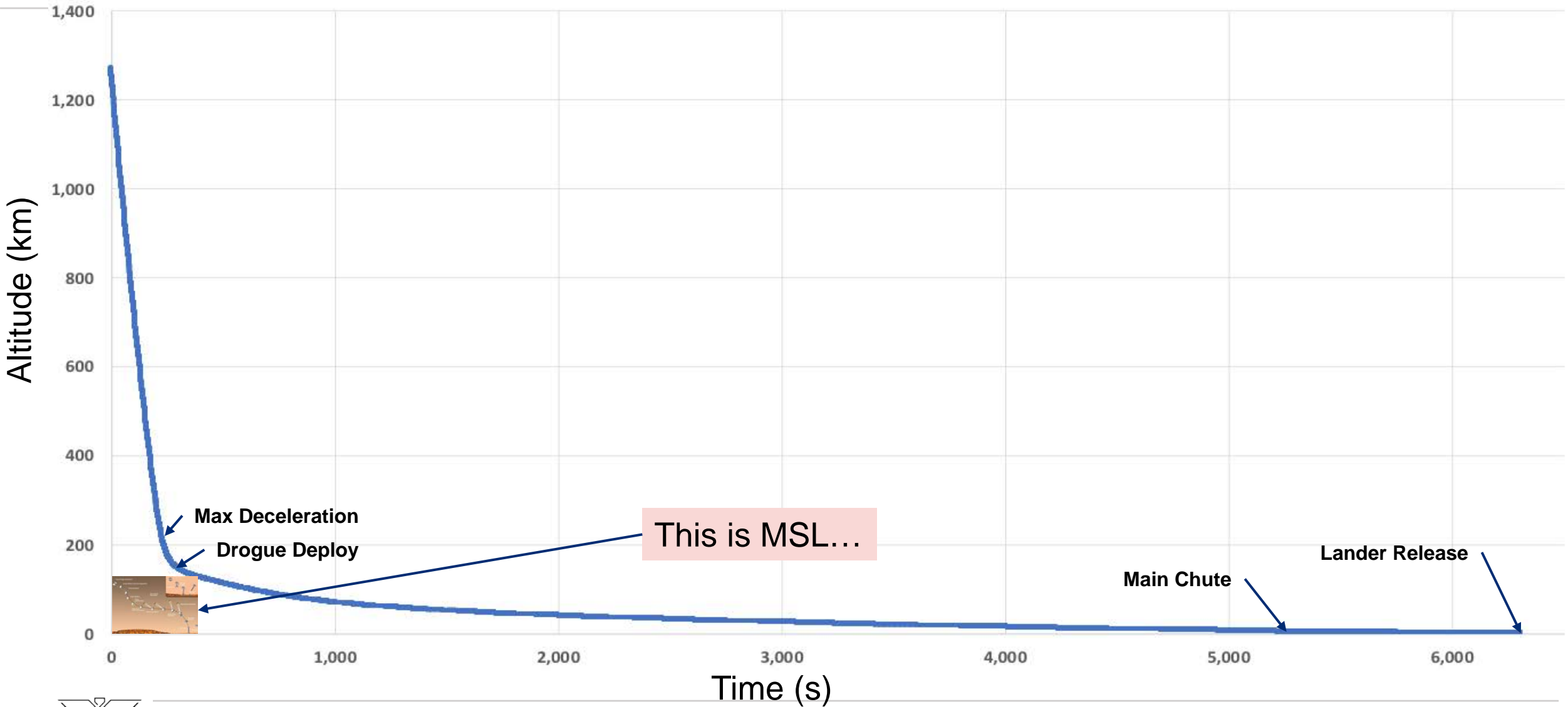
EDL Concept of Operations



- Entry Interface 1270 km
 - Spin stabilized to 2 RPM
- Entry heat pulse: 250 sec.
 - Peak heat flux 250 W/cm² margined
- Drogue deploy E+6 min, ~Mach 1.5.
 - More than 80 minutes spent on drogue
- Main chute deploy E+88 min.
 - Low velocity & opening load
- Lander Release E+105 min.
 - Plenty of time to stage heatshield separation, activate radar & lidar, deploy landing legs



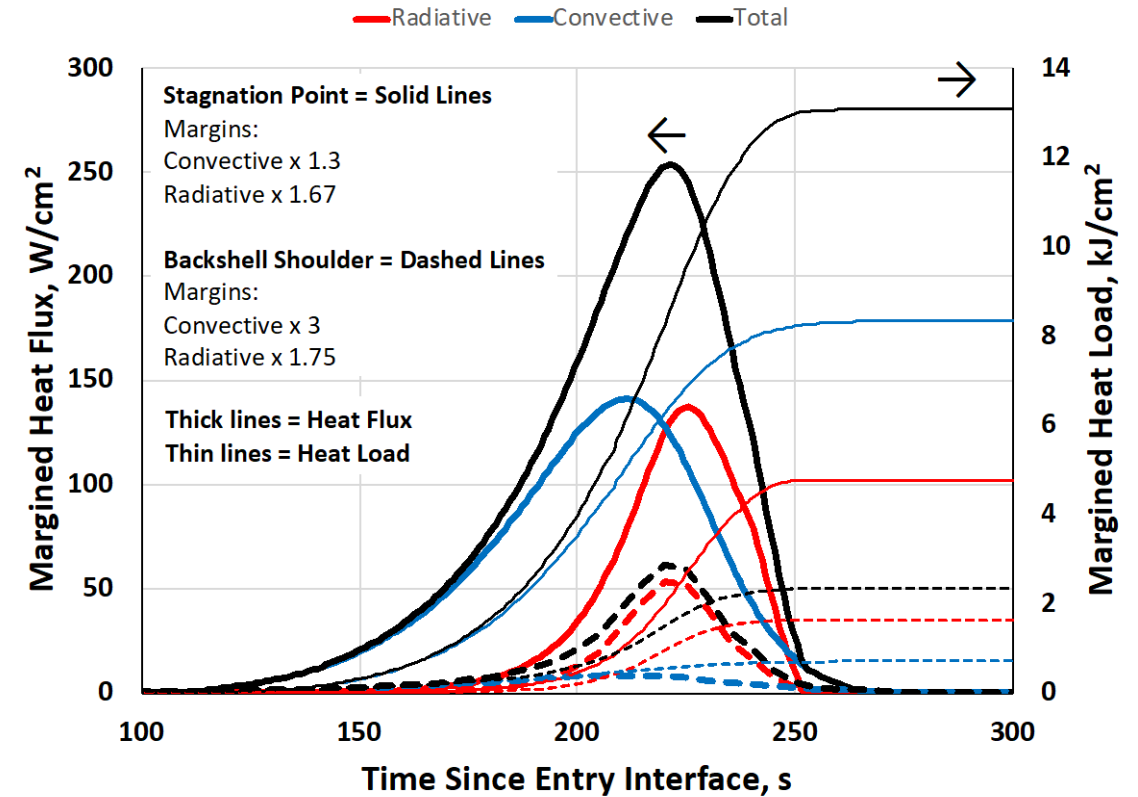
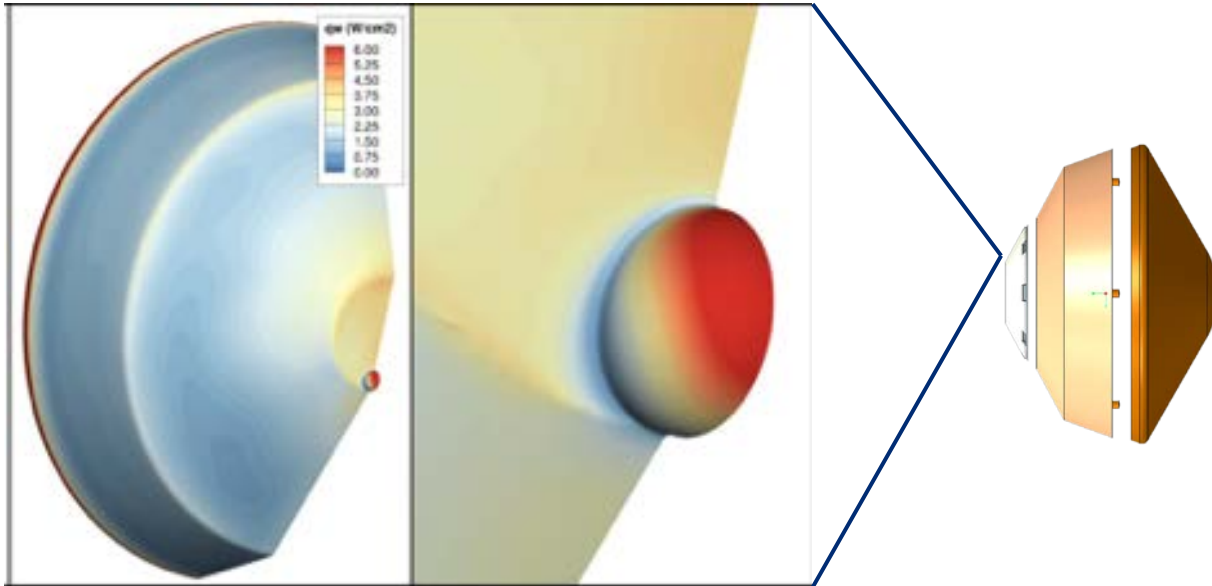
EDL Timeline Comparison...



Aerothermal Environments



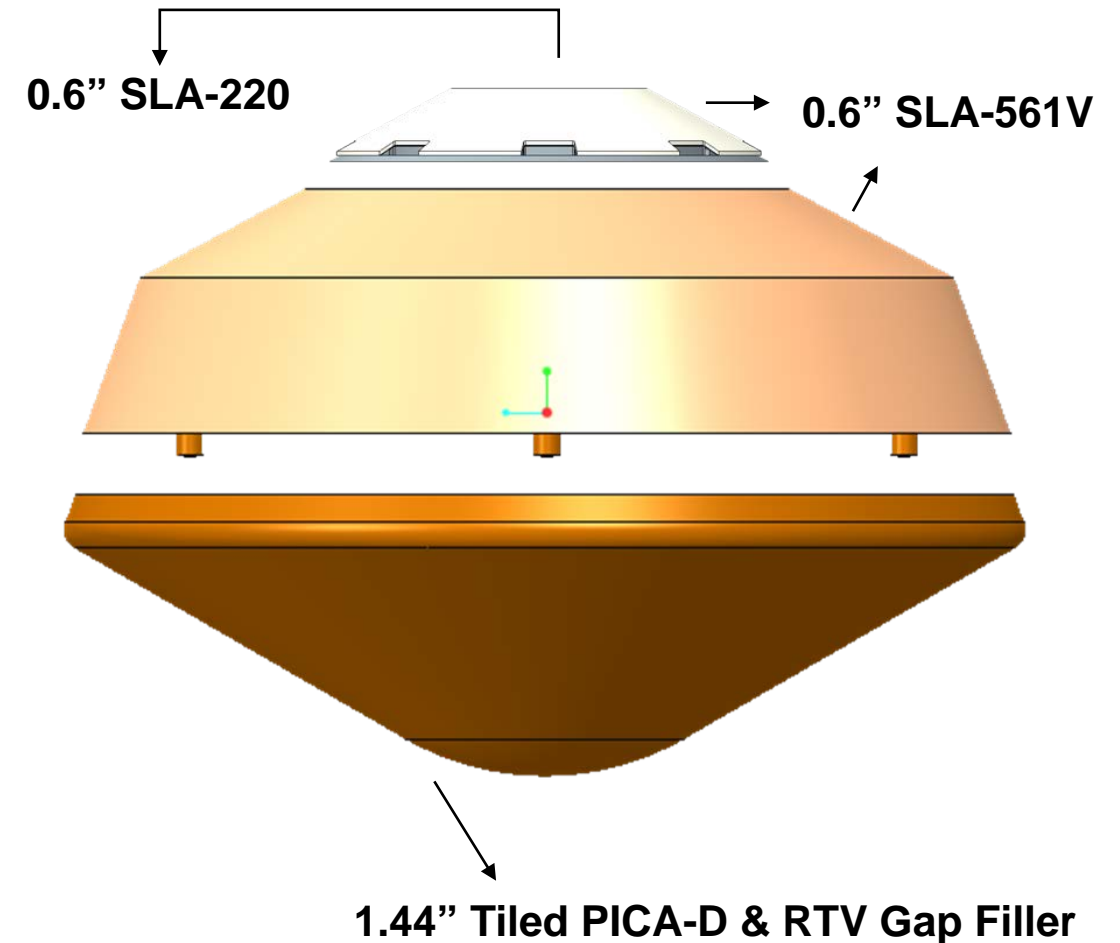
- NASA standard aerothermal (convective + radiative) models based tuned for Titan entry
- Total heating calculated over entire aeroshell
 - Conservative margins applied (Orion/MSL methodology)
 - Margined values well within tested limits for chosen TPS
 - Detailed simulations underway ahead of normal design cycle



Thermal Protection System



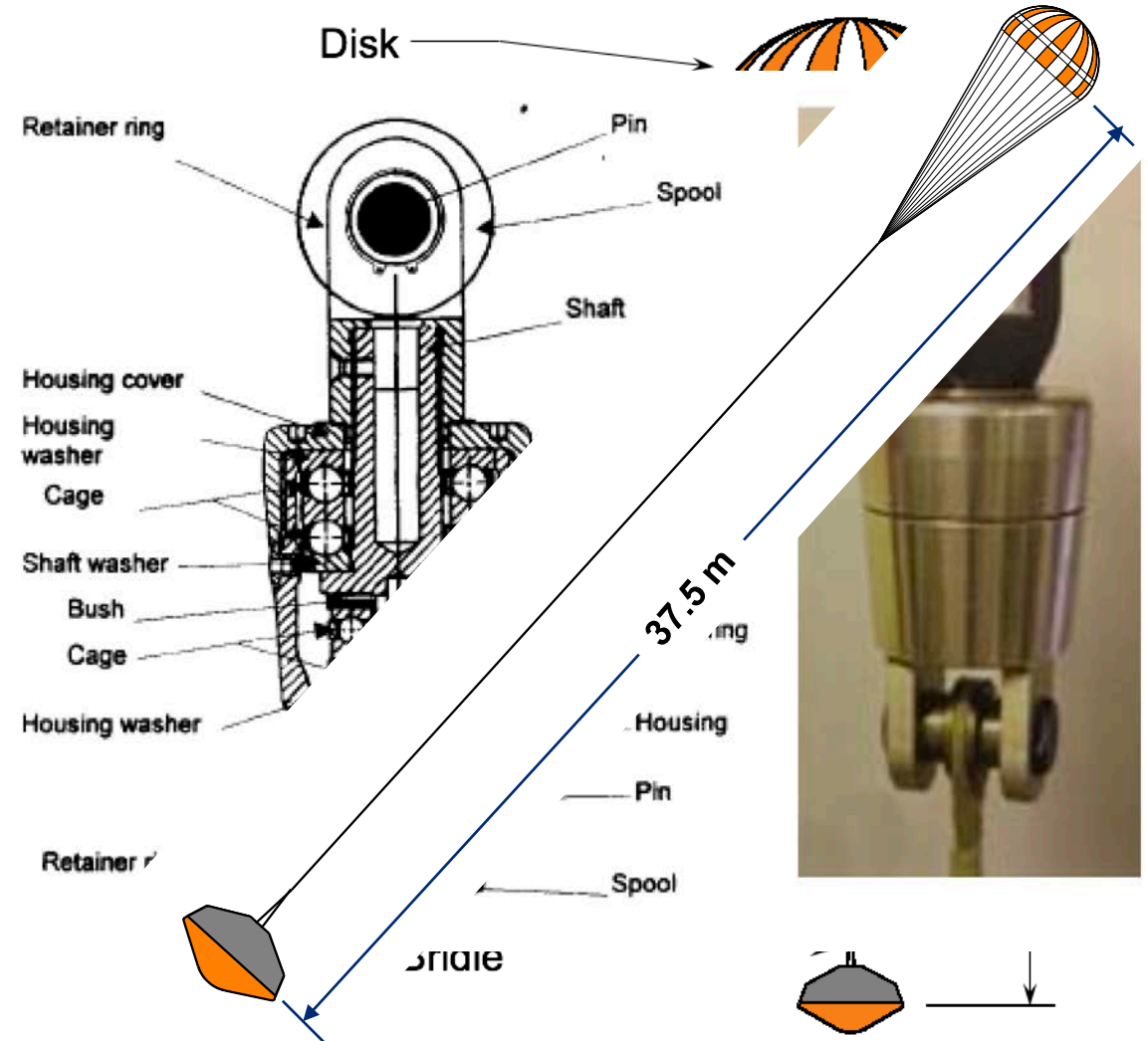
- Three TPS materials:
 - High TRL and used well within tested limits
 - Arc jet testing confirms performance
 - SMD funded PICA-D will be a drop-in replacement for heritage PICA. Ready by the end of 2019
- TPS sizing analysis uses mature processes developed during MSL/Orion
 - Design thicknesses carry unallocated margin
- TPS manufacture, testing, qualification and assembly follow standard procedures developed at LM and NASA ARC



Parachute Deceleration System



- Drogue Parachute (4.5m DGB)
 - Mortar deploy via trigger at Mach 1.5
 - Functions: stabilize capsule, decelerate through atmosphere, extract main chute
- Main Parachute (13.44m DGB)
 - Low speed subsonic deploy
 - Includes inversion netting
- Huygens heritage swivels prevent line twisting



Engineering Science Investigation



- **Dragonfly is the first NASA competed mission subject to the ESI requirement**

- **ESI Goals:**

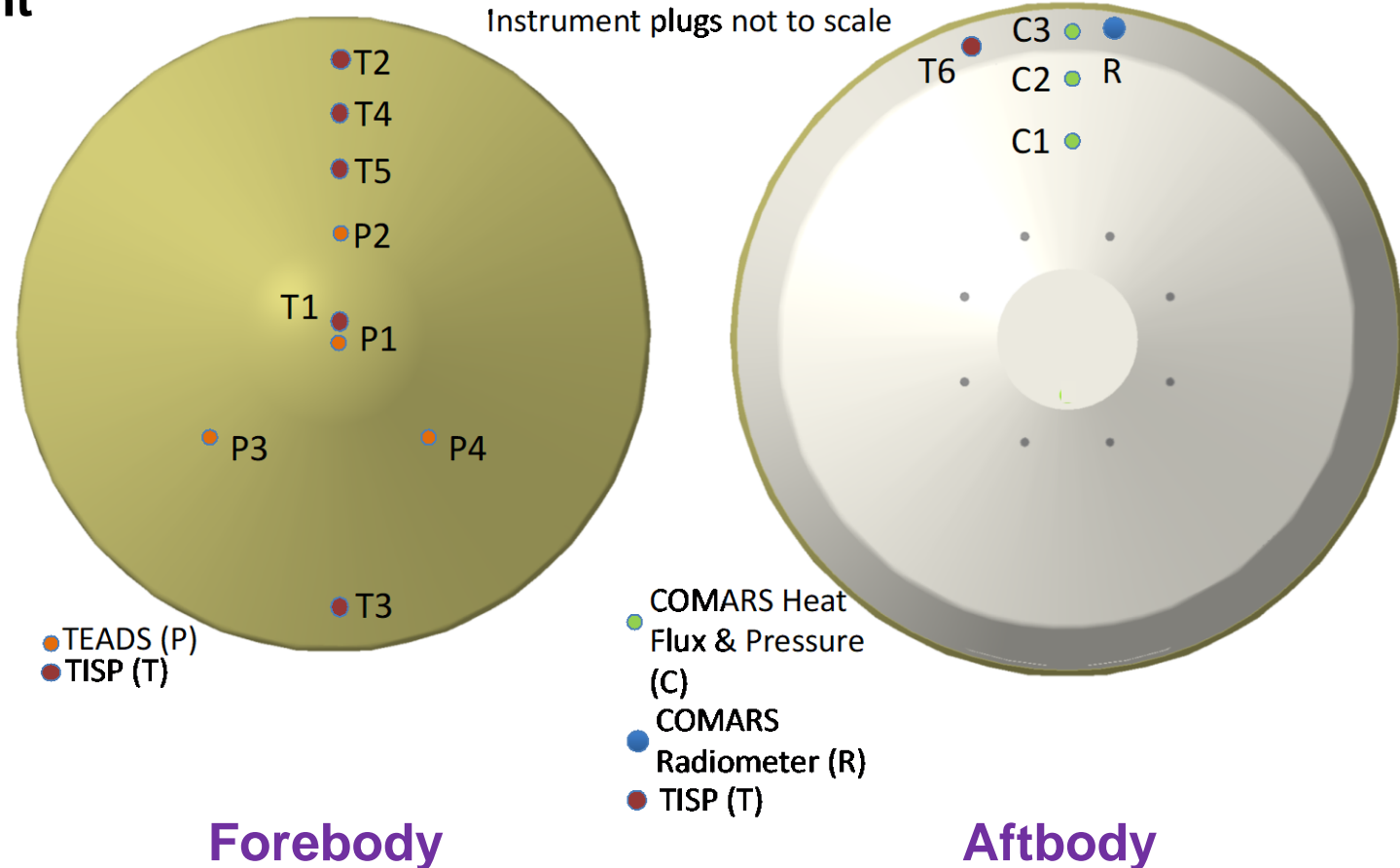
- Near surface and in-depth TPS temperature
- Surface pressure
- Total and radiative heat flux

- **ESI Implementation:**

- Partnership with German Space Agency (DLR)
- NASA provided MEDLI-style TC plugs and pressure ports on the heatshield and afterbody
- DLR provided COMARS sensor suite on the afterbody
- DLR provided data acquisition system for all instrumentation

Phase-A Preliminary Layout

Instrument plugs not to scale



Conclusions



- **Dragonfly EDL system uses mature, high TRL, flight proven components**
 - 3.75m Genesis derived aeroshell, flight proven TPS solutions, Viking derived parachute system
 - Aeroshell provider (LM) has extensive hardware expertise in all relevant EDL missions since Pathfinder
- **EDL is well understood; with robust performance margins across the subsystems**
 - Titan EDL is more straightforward than Mars in several important ways
 - Timeline is not a significant constraint in EDL design
 - Experienced analysis team is using NASA-standard processes and tools with appropriate modifications for Titan
 - A lot of analysis has been already done to reduce risk heading into Phase-B
- **ESI will collect the most flight data ever from any destination other than Earth or Mars**
 - Enabled by strong partnership with DLR

We are ready for Titan!



EDL Monte Carlo Analysis



- Full analysis in POST2 from EI to lander release
- Delivery accuracy ~149x72 km at release
 - Easily affords lander sufficient accuracy to navigate to selected landing zone
- Dispersions sources
 - Latitude: navigation errors
 - Longitude: on-chute winds

