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# Systems Health Monitoring and Prognostics using Model based Approach

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The technical slides represent work done by PCoE members who have contributed to this presentation. All details presented here are in the public domain and used for information purposes only.

<http://prognostics.nasa.gov>

# Acknowledgments

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# Agenda

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- Introduction to Prognostics
- Introduction to Model-based Prognostics
- Research Approach
- Architecture
- Accelerated Aging as a Prognostics Research Tool
- Case Study I: Prognostics of Electrolytic Capacitors
  - Model-based approach example
- Case Study II: Prognostics of Power Transistors
  - Precursors of Failure example
- Case Study III: Physics-based Prognostics of Capacitors
  - Degradation modeling example
- Case Study IV: Prognostics of Li-Ion Batteries
  - Degradation/Aging example
- Closing Remarks

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# **INTRODUCTION TO PROGNOSTICS**

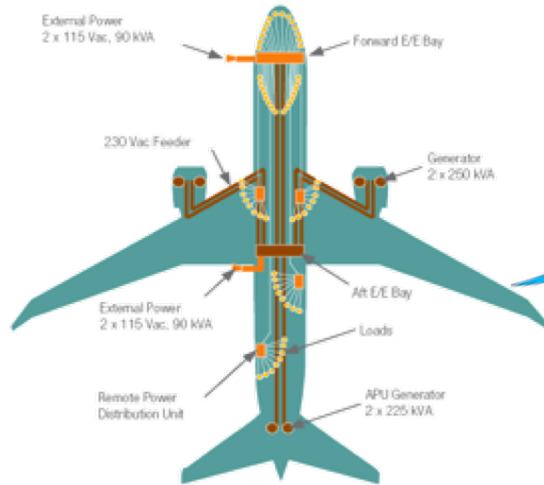
# Motivation (1/2)

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- Future aircraft systems will rely more on electrical and electronic components
- UAV's with all electric powertrain are increasingly being used for long missions
- Electrical and Electronic components have increasingly critical role in on-board, autonomous functions for
  - Vehicle controls, communications, navigation, radar systems
  - Power electronic devices such as power MOSFETs and IGBTs are frequently used in high-power switching circuits
  - Batteries are the sole energy storage
  - The integrated navigation (INAV) module combines output of the GPS model and inertial measurement unit.
- Assumption of new functionality increases number of faults with perhaps unanticipated fault modes
- We need understanding of behavior of deteriorated components to develop capability to anticipate failures/predict remaining RUL

# Motivation (2/2)

Images courtesy : Boeing



**Line Replaceable Unit:  
Power Controller**



**Component:  
Power Transistor**



Ref: [www.nasa.gov](http://www.nasa.gov)



# Definitions

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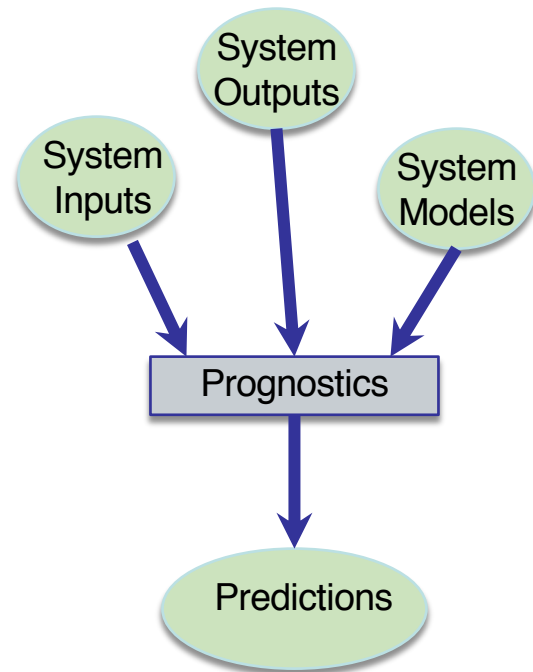
## So what is “Prognostics” anyway?

- prog-nos-tic
  - M-W.com – “Something that foretells”
  - PHM Community – “Estimation of the *Remaining Useful Life* of a component”
- Remaining Useful Life (RUL) – The amount of time a component can be expected to continue operating within its stated specifications.
  - Dependent on future operating conditions
    - Input commands
    - Environment
    - Loads

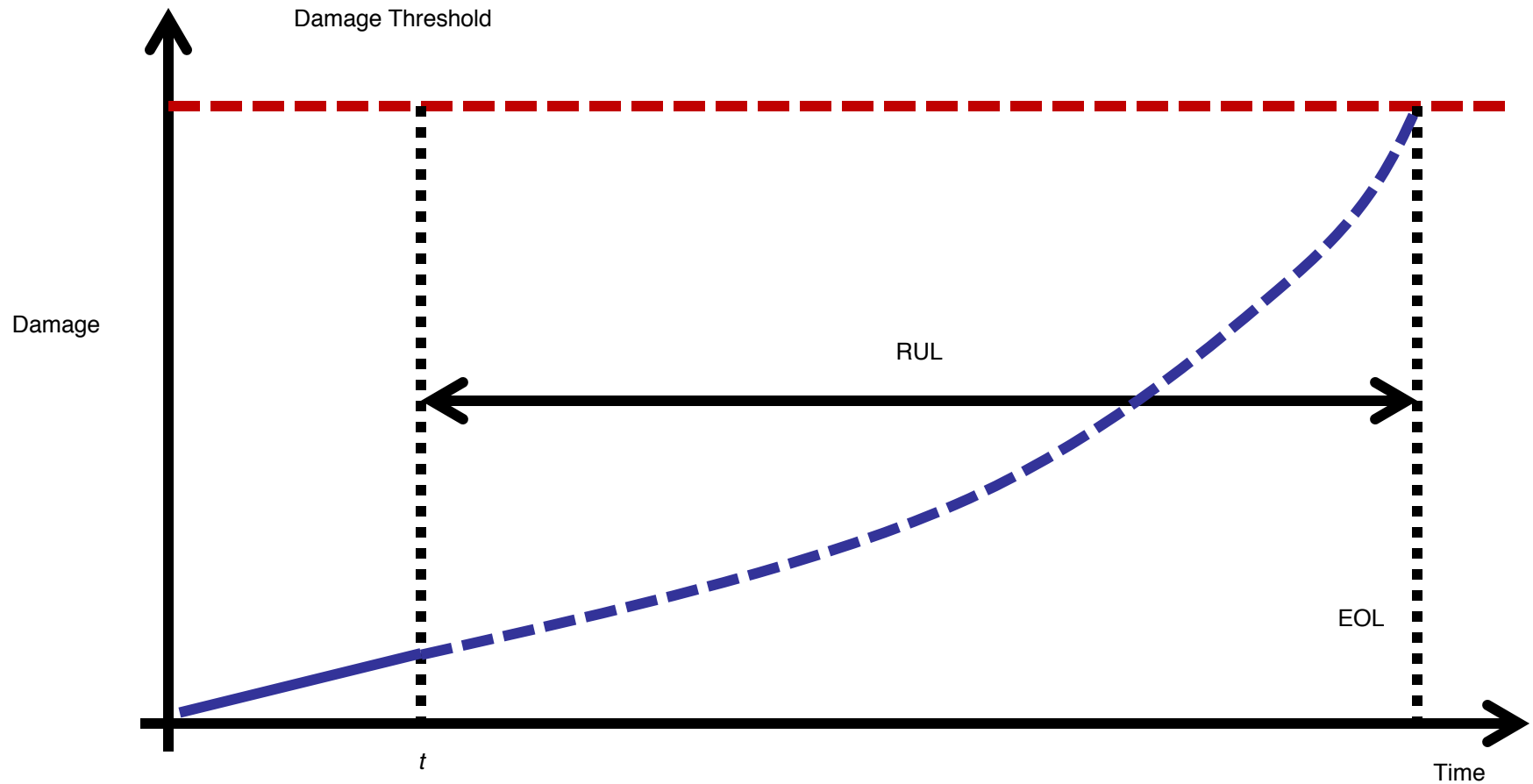
# Why Model-Based Prognostics?

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- With model-based algorithms, models are inputs
  - This means that, given a new problem, we use the same general algorithms
  - Only the models should change
- Model-based prognostics approaches are applicable to a large class of systems, given a model
- Approach can be formulated mathematically, clearly and precisely



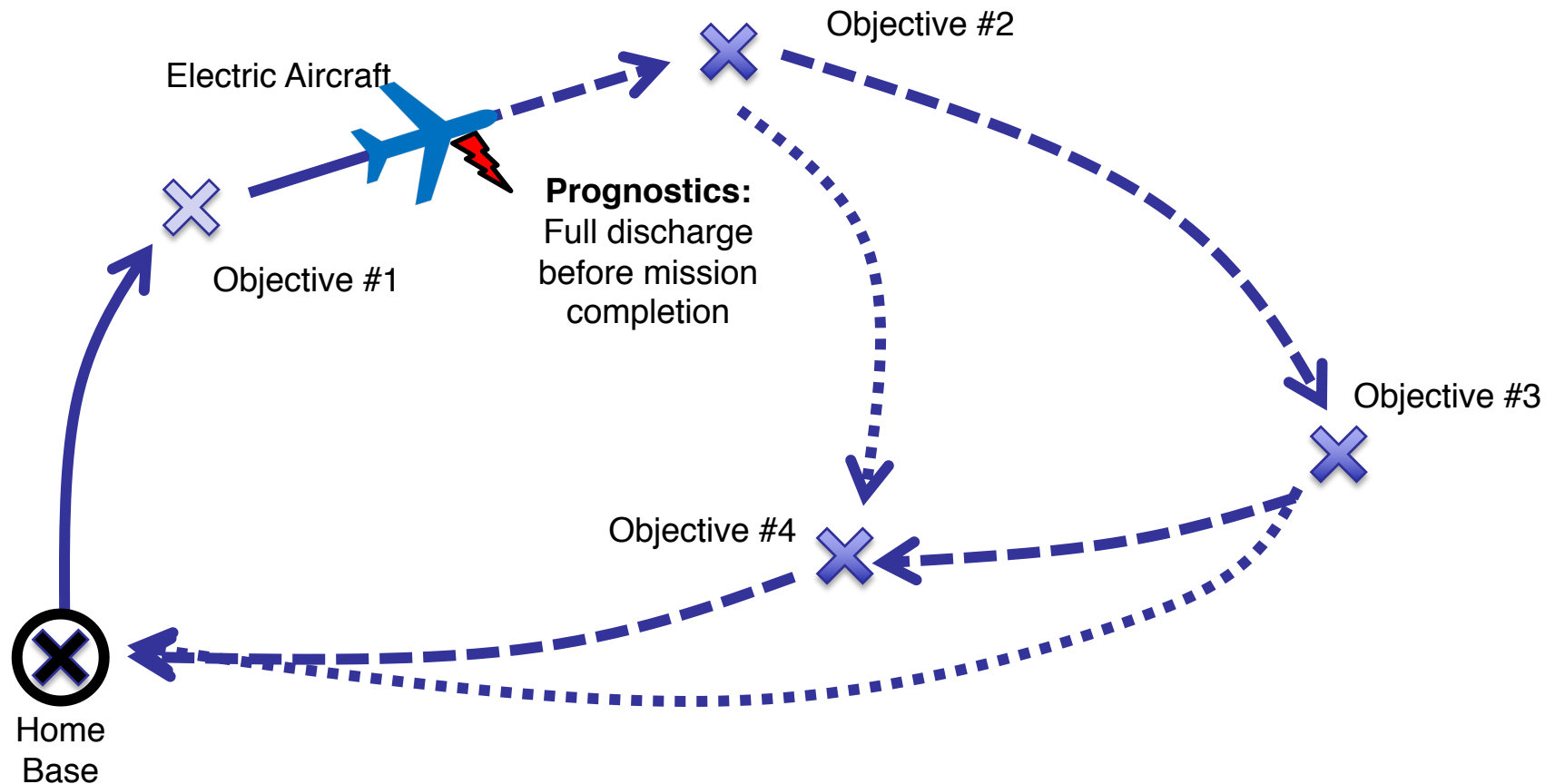
# The Basic Idea



# Why Prognostics?

## Example: UAV Mission

Visit waypoints to accomplish science objectives. Predict aircraft battery end of discharge to determine which objectives can be met. Based on prediction, plan optimal route. Replan if prediction changes.

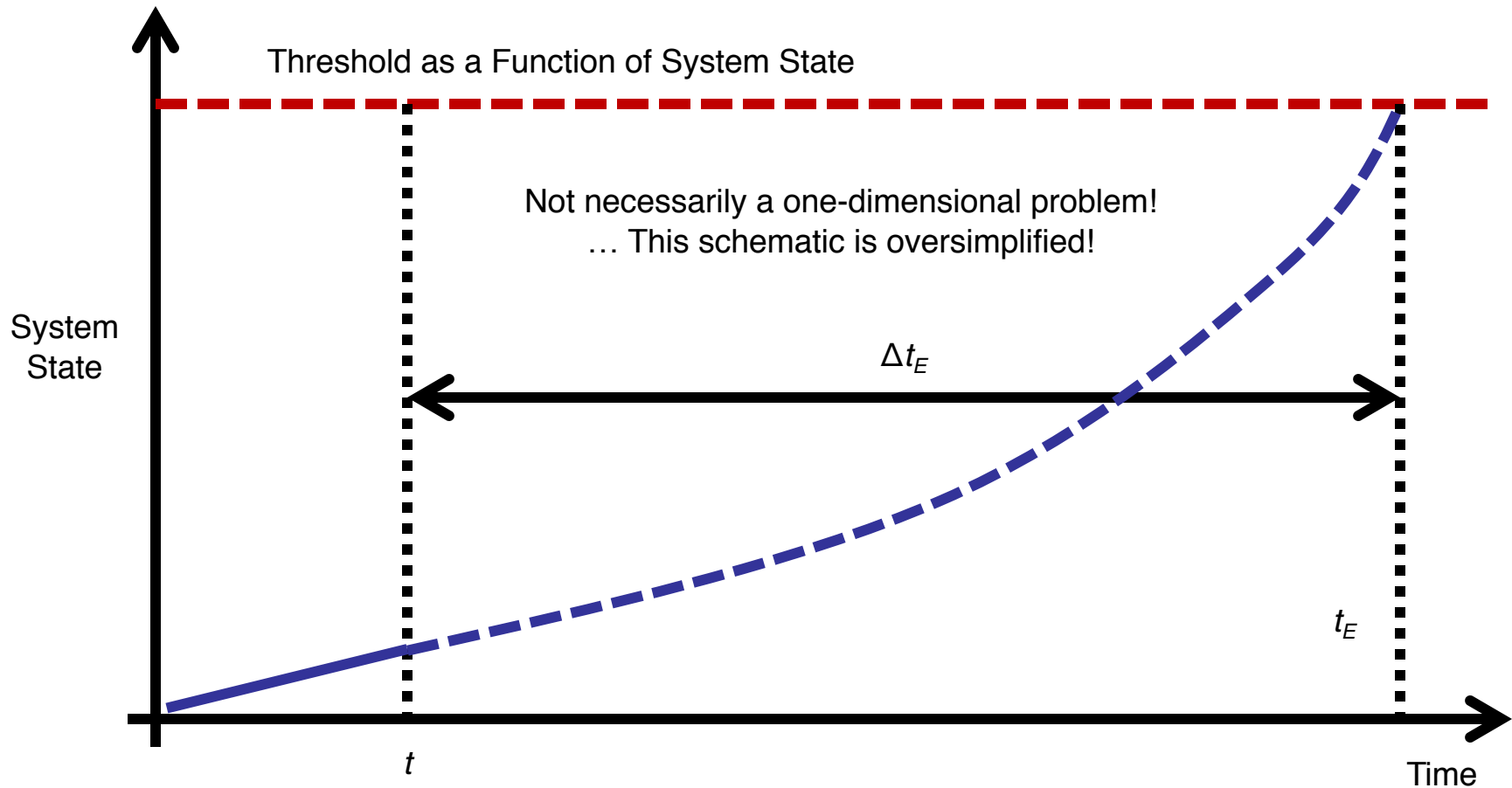


# Why Prognostics?

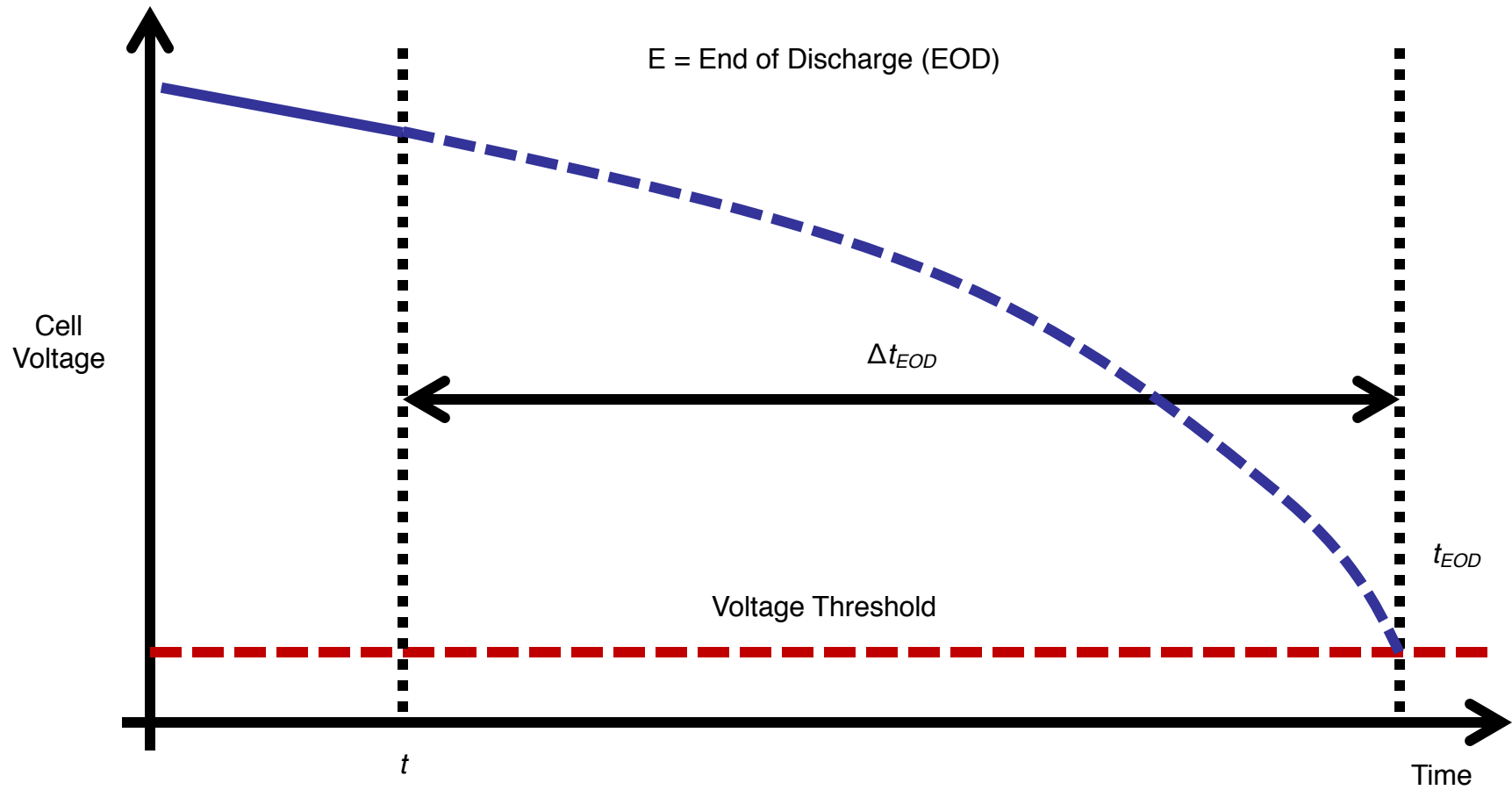
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- Prognostics can enable:
  - Adopting condition-based maintenance strategies, instead of time-based maintenance
  - Optimally scheduling maintenance
  - Optimally planning for spare components
  - Reconfiguring the system to avoid using the component before it fails
  - Prolonging component life by modifying how the component is used (e.g., load shedding)
  - Optimally plan or replan a mission
- System operations can be optimized in a variety of ways

# The Basic Idea Revisited



# The Basic Idea : Batteries Example



# Prognostic Algorithm Categories

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- Type I: Reliability Data-based
  - Use population based statistical model
  - These methods consider historical time to failure data which are used to model the failure distribution. They estimate the life of a typical component under nominal usage conditions.
  - Ex: Weibull Analysis
- Type II: Stress-based
  - Use population based fault growth model – learned from accumulated knowledge
  - These methods also consider the environmental stresses (temperature, load, vibration, etc.) on the component. They estimate the life of an average component under specific usage conditions.
  - Ex: Proportional Hazards Model
- Type III: Condition-based
  - Individual component based data-driven model
  - These methods also consider the measured or inferred component degradation. They estimate the life of a specific component under specific usage and degradation conditions.
  - Ex: Cumulative Damage Model, Filtering and State Estimation

# Data-Driven Methods

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- Model is based solely on data collected from the system
- Some system knowledge may still be handy:
  - What the system ‘is’
  - What the failure modes are
  - What sensor information is available
  - Which sensors may contain indicators of fault progression (and how those signals may ‘grow’)
- *General* steps:
  - Gather what information you can (if any)
  - Determine which sensors give good trends
  - Process the data to “clean it up” – try to get nice, monotonic trends
  - Determine threshold(s) either from experience (data) or requirements
  - Use the model to predict RUL
    - Regression / trending
    - Mapping (e.g., using a neural network)
    - Statistics

# Data-Driven Methods

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- Pros
  - Easy and Fast to implement
    - Several off-the-shelf packages are available for data mining
  - May identify relationships that were not previously considered
    - Can consider all relationships without prejudice
- Cons
  - Requires lots of data and a “balanced” approach
    - Most of the time, lots of run-to-failure data are not available
    - High risk of “over-learning” the data
    - Conversely, there’s also a risk of “over-generalizing”
  - Results may be counter- (or even un-)intuitive
    - Correlation does not always imply causality!
  - Can be computationally intensive, both for analysis and implementation
- Example techniques
  - Regression analysis
  - Neural Networks (NN)
  - Bayesian updates
  - Relevance vector machines (RVM)

# Physics-Based Methods

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- Description of a system's underlying physics using suitable representation
- Some examples:
  - Model derived from “First Principles”
    - Encapsulate fundamental laws of physics
      - PDEs
      - Euler-Lagrange Equations
  - Empirical model chosen based on an understanding of the dynamics of a system
    - Lumped Parameter Model
    - Classical 1<sup>st</sup> (or higher) order response curves
  - Mappings of stressors onto damage accumulation
    - Finite Element Model
    - High-fidelity Simulation Model
- Something in the model correlates to the failure mode(s) of interest

# Physics-Based Models

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- Pros
  - Results tend to be intuitive
    - Based on modeled phenomenon
    - And when they're not, they're still instructive (e.g., identifying needs for more fidelity or unmodeled effects)
  - Models can be reused
    - Tuning of parameters can be used to account for differences in design
  - If incorporated early enough in the design process, can drive sensor requirements (adding or removing)
  - Computationally efficient to implement
- Cons
  - Model development requires a thorough understanding of the system
  - High-fidelity models can be computationally intensive
- Examples
  - Paris-Erdogan Crack Growth Model
  - Taylor tool wear model
  - Corrosion model
  - Abrasion model

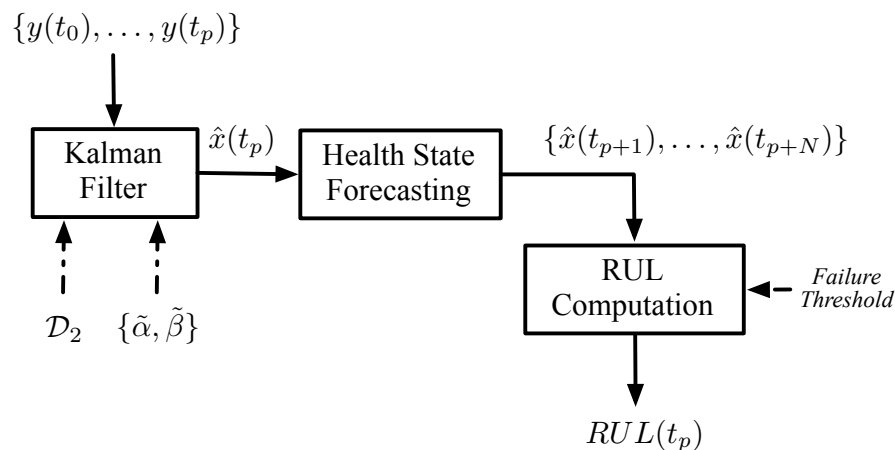
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# **INTRODUCTION TO MODEL- BASED PROGNOSTICS**

# Model-based prognostics (1/2)

$$\dot{\mathbf{x}}(t) = f(\mathbf{x}(t), u(t)) + w(t)$$
$$y(t) = h(\mathbf{x}(t), u(t)) + v(k)$$

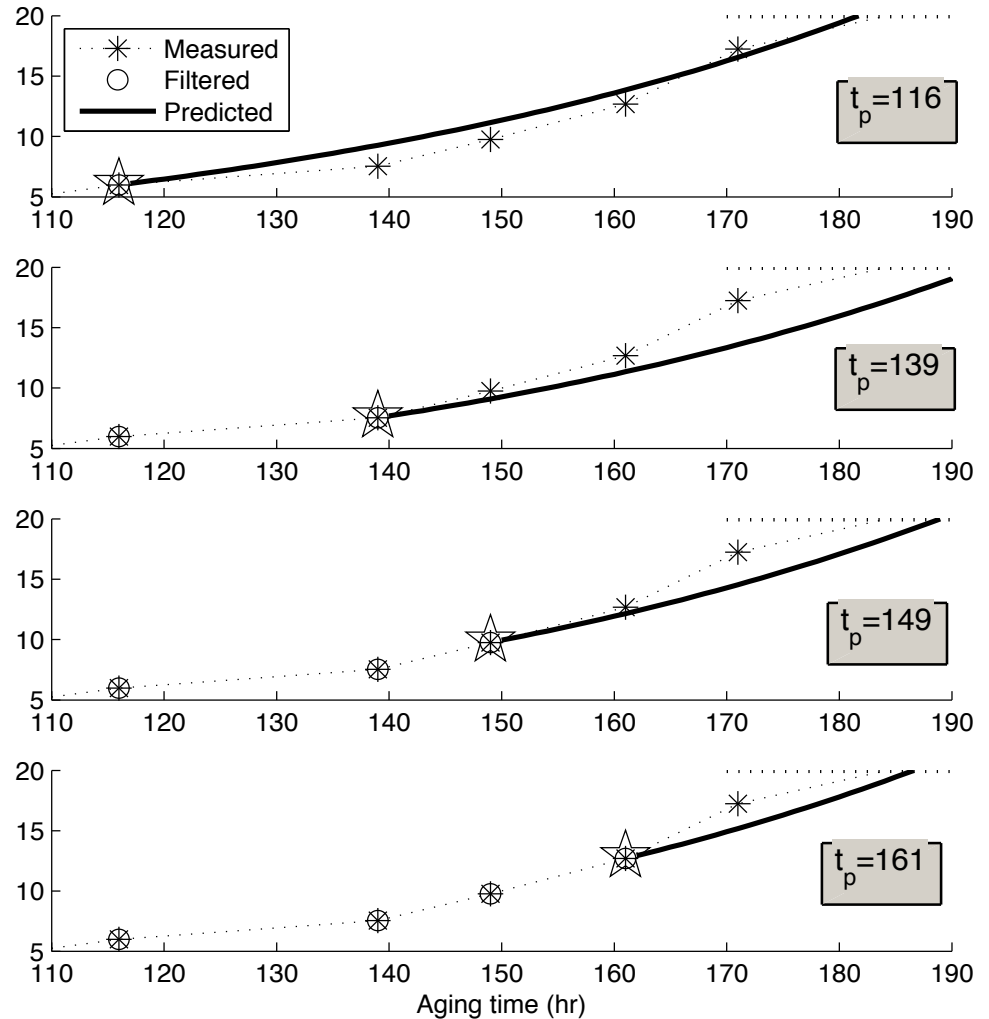
$$R(t_p) = t_{EOL} - t_p$$



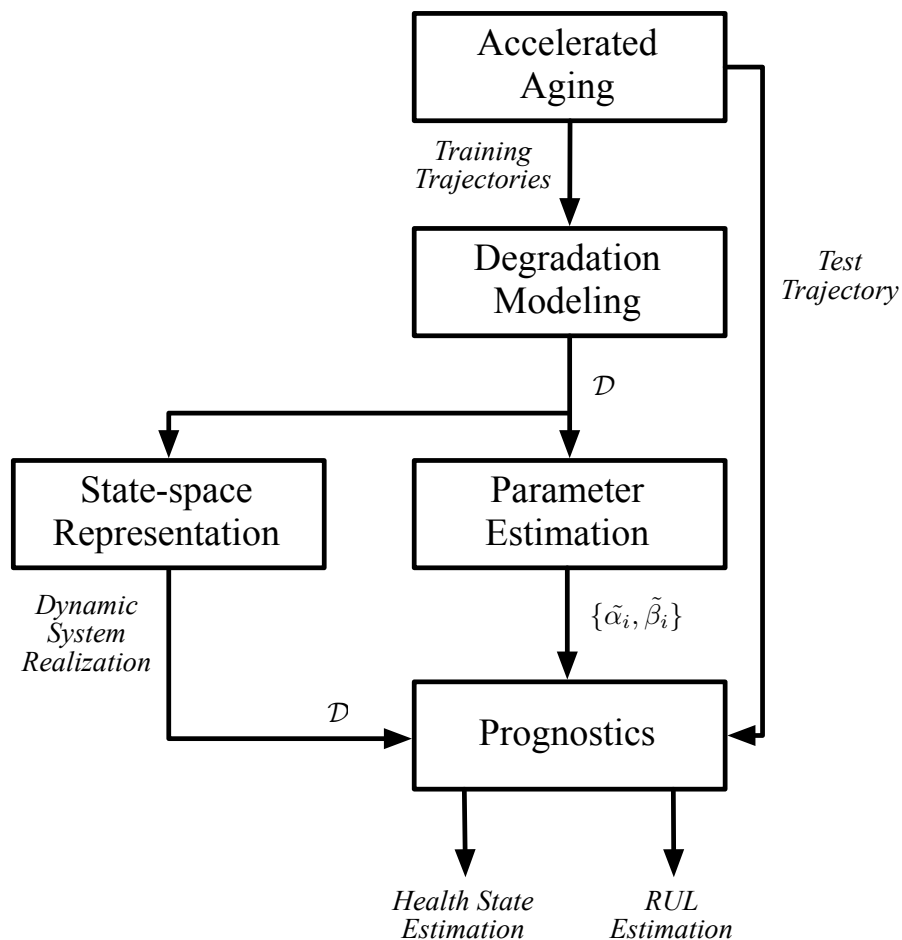
- State vector includes dynamics of the degradation process
- It might include nominal operation dynamics
- EOL defined at time in which performance variable cross failure threshold
- Failure threshold could be crisp or also a random variable

# Model-based prognostics (2/2)

- Tracking of health state based on measurements
- Forecasting of health state until failure threshold is crossed
- Compute RUL as function of EOL defined at time failure threshold is crossed

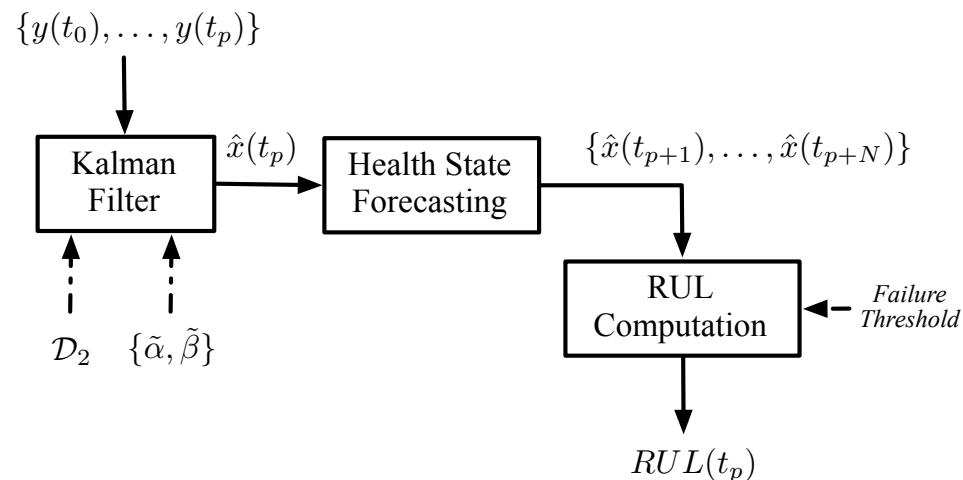


# Methodology



$$x_k = Ax_{k-1} + Bu_{k-1} + w_{k-1}$$

$$y_k = Hx_k + v_k$$



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# **RESEARCH APPROACH**

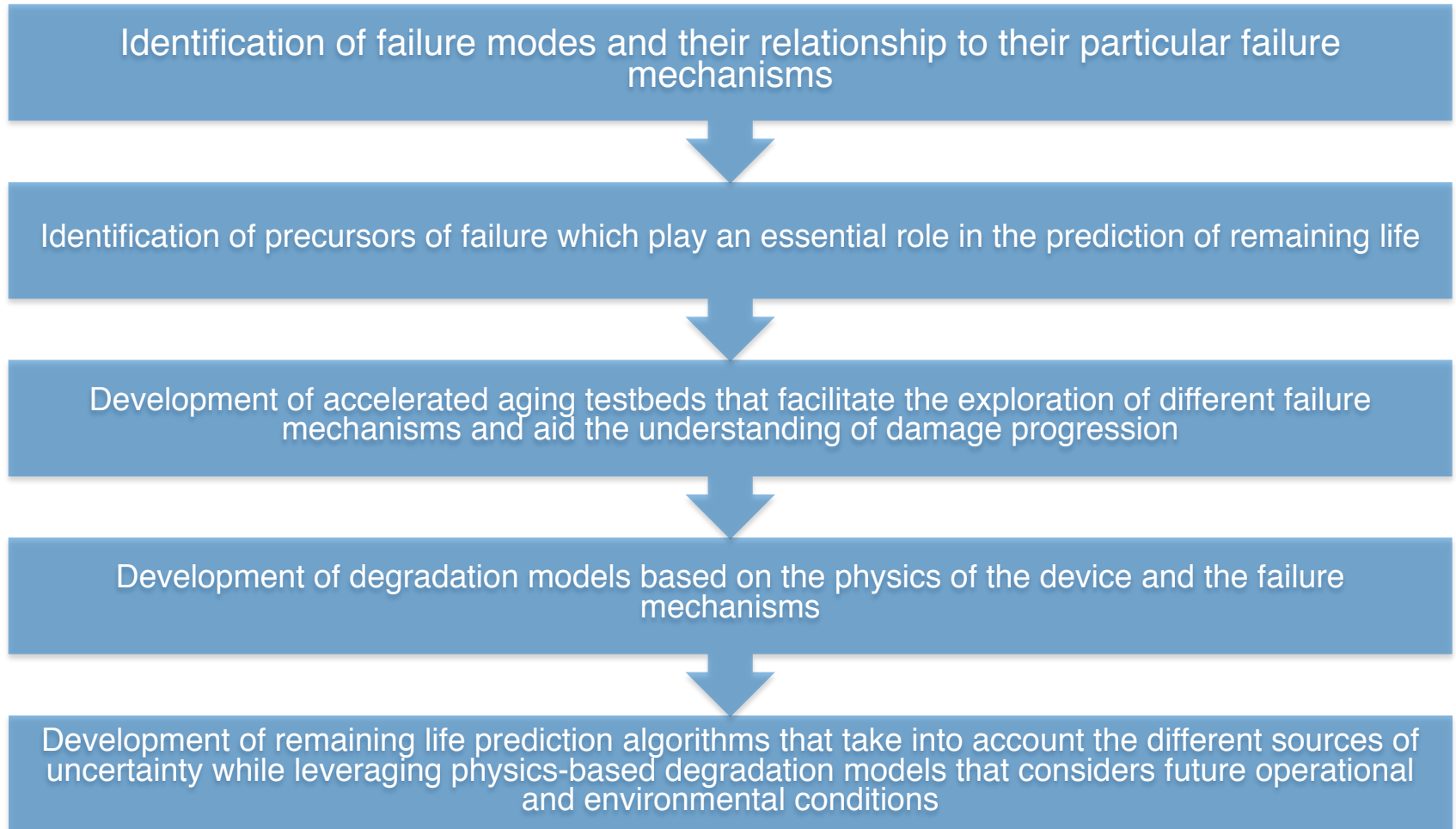
# High level research efforts

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- **Prognostics models and algorithms**
  - Identification of precursors of failure for MOSFETs under different failure mechanism conditions
  - Identification of precursors of failure for different IGBT technologies
  - Modeling of degradation process MOSFETs
  - Development of prognostics algorithms
- **Prognostics for output capacitor in power supplies (ARC)**
  - Electrical overstress and thermal overstress
  - Development of prognostics algorithms
- **Accelerated Life Testing**
  - Thermal overstress aging of MOSFETs and IGBTs
  - Electrical overstress aging testbed MOSFETs
  - Electrical overstress aging testbed for Capacitors
- **Effects of lightning events of MOSFETS (LaRC)**
- **Battery Degradation and ageing ( ARC – LaRC)**
- **Ageing Effecting on ESC's ( ARC – LaRC)**

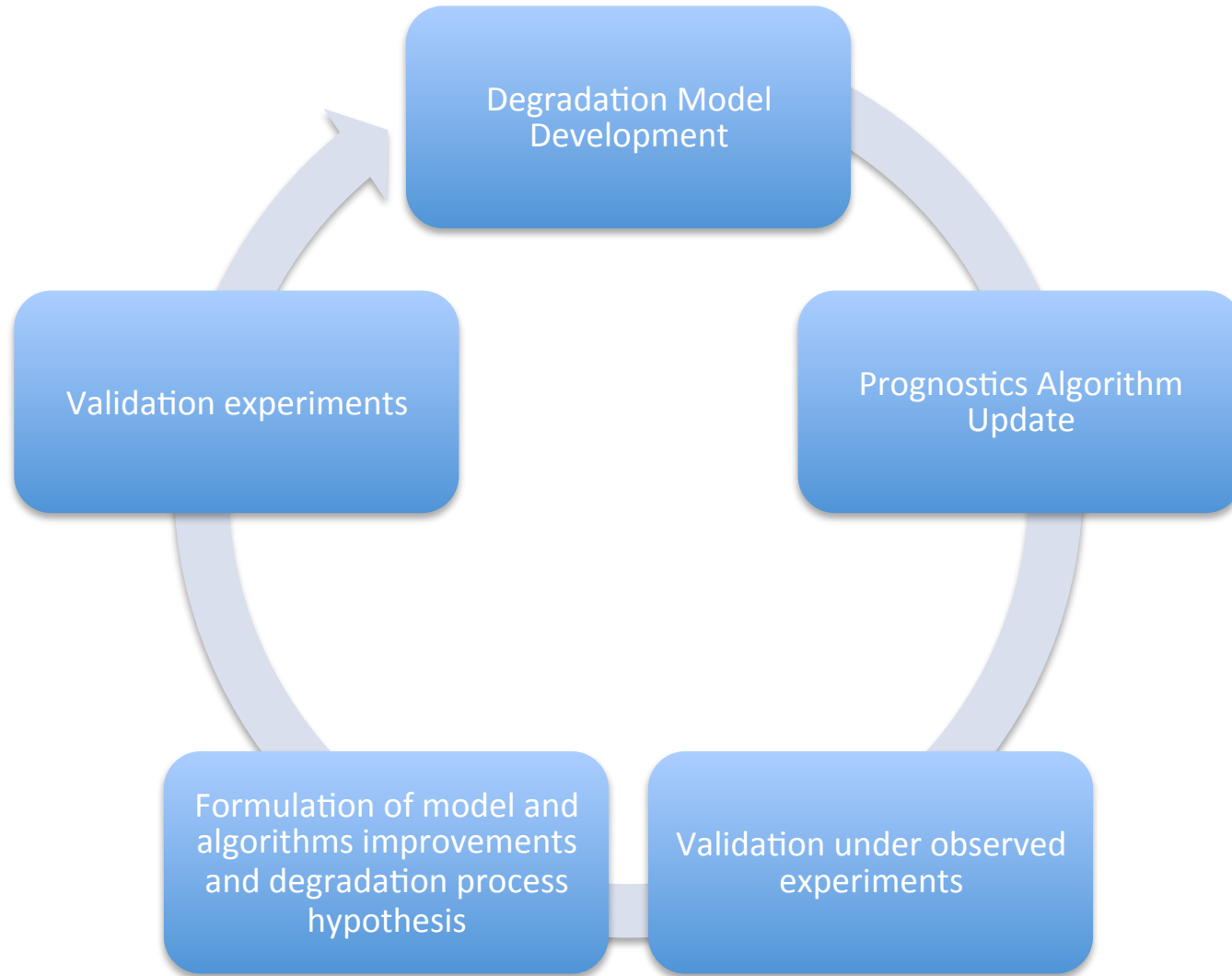
# Research Approach

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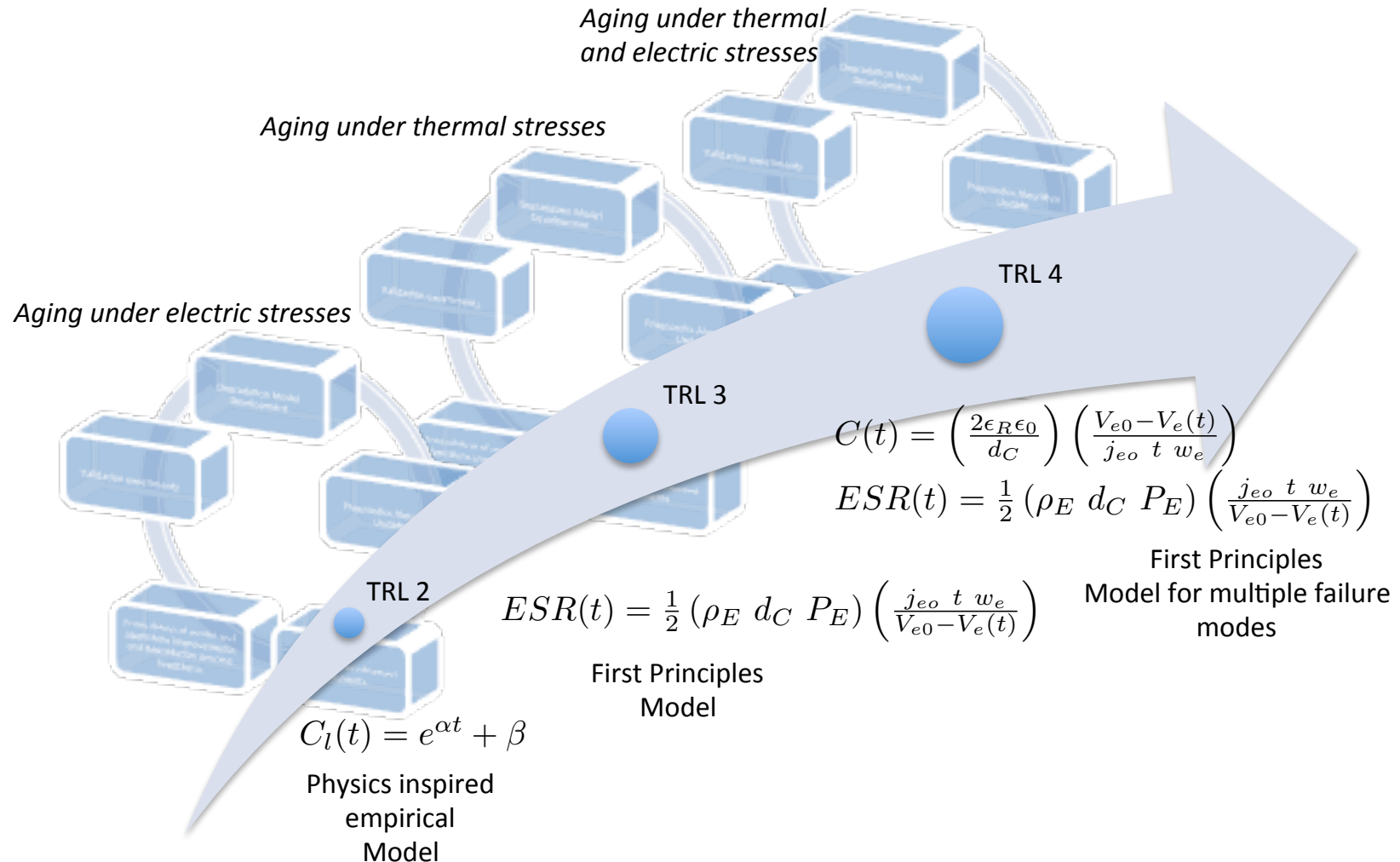


# Prognostics Algorithm Maturation through Validation Experiments

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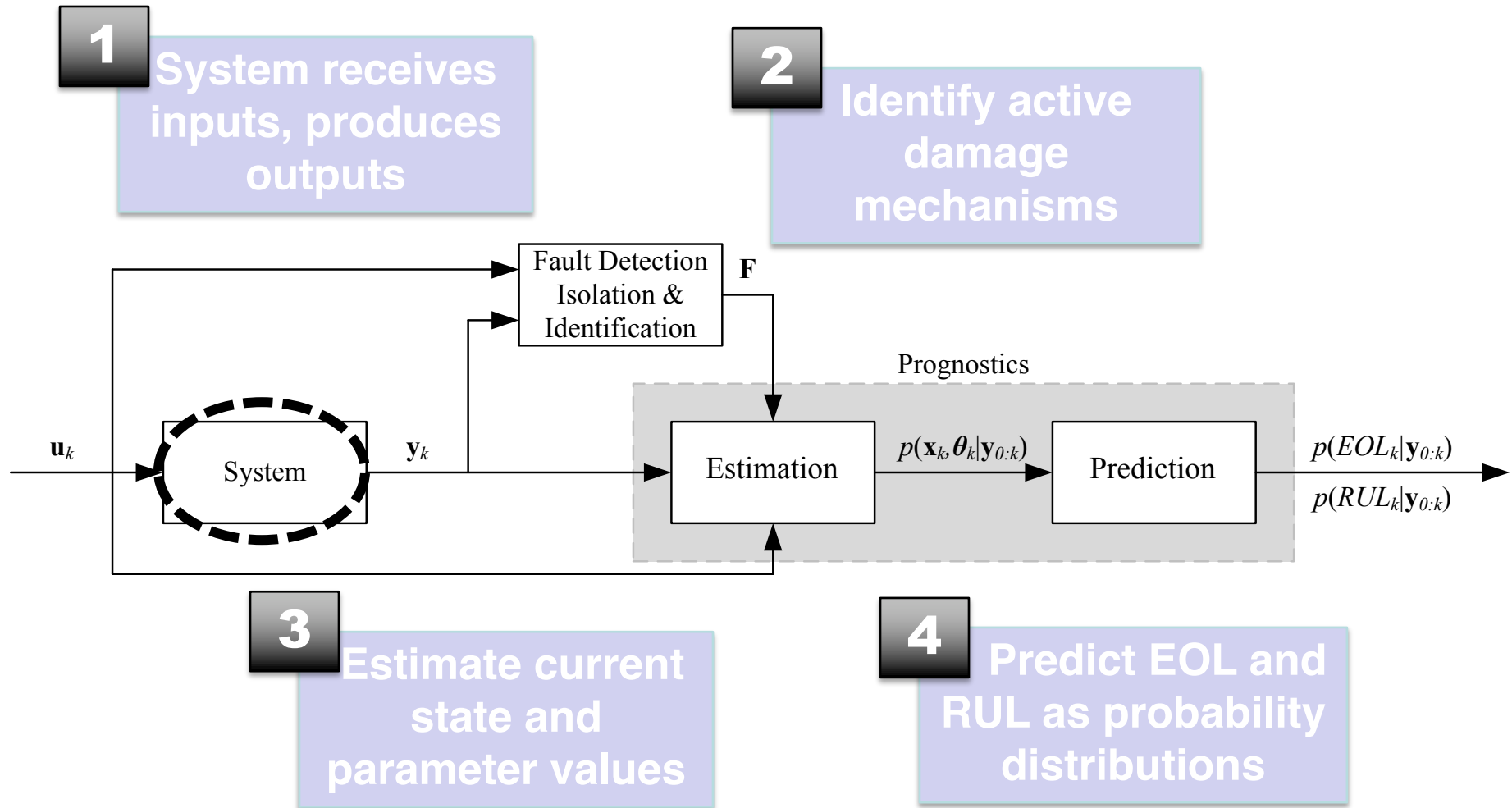
# Prognostics Algorithm Maturation through Validation Experiments



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# **ARCHITECTURE**

# Model-Based Architecture



# Problem Requirements

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- System model
  - System state space
  - Partition into nonfailure and failure states
  - System inputs
  - State update equation
- Prediction inputs
  - Initial time  $k_o$
  - Prediction horizon  $k_h$
  - System inputs from  $k_o$  to  $k_h$

# System Model

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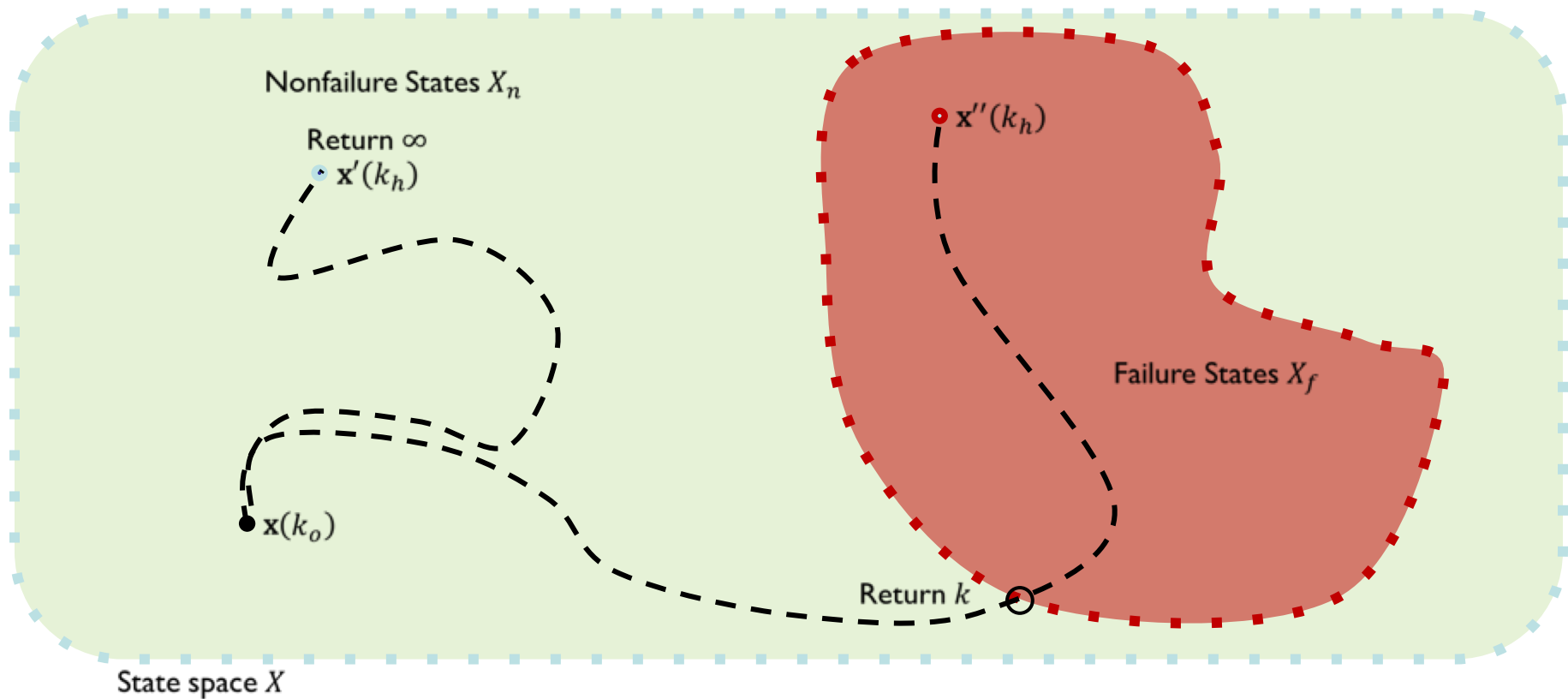
- Assume system can be modeled using
  - $x(k + 1) = f(x(k), u(k), v(k))$
  - $k$  is the discrete time variable
  - $x$  is the state vector
  - $u$  is the input vector
  - $v$  is the process noise vector
  - $f$  is the state update equation
- Define a function that partitions state-space into nonfailure and failure states
  - $T_f: \mathbb{R}^{n_x} \rightarrow \{true, false\}$
  - That is,  $T_f(x(k))$  returns true when it is a failure state, false otherwise

# Initial Problem Formulation

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- Assume we know
  - Initial state,  $\mathbf{x}(k_o)$
  - Future input trajectory,  $\mathbf{U}_{k_o, k_h} = [u(k_o), u(k_o + 1), \dots, u(k_h)]$
  - Process noise trajectory,  $\mathbf{V}_{k_o, k_h} = [v(k_o), v(k_o + 1), \dots, v(k_h)]$
- Problem definition
  - Given  $k_o, k_h, \mathbf{x}(k_o), \mathbf{U}_{k_o, k_h}, \mathbf{V}_{k_o, k_h}$
  - Compute EOL
    - $\text{EOL}(k) = \inf\{k' : k' \geq k \text{ and } T_f(\mathbf{x}(k))\}$

# Concept: ComputeEOL



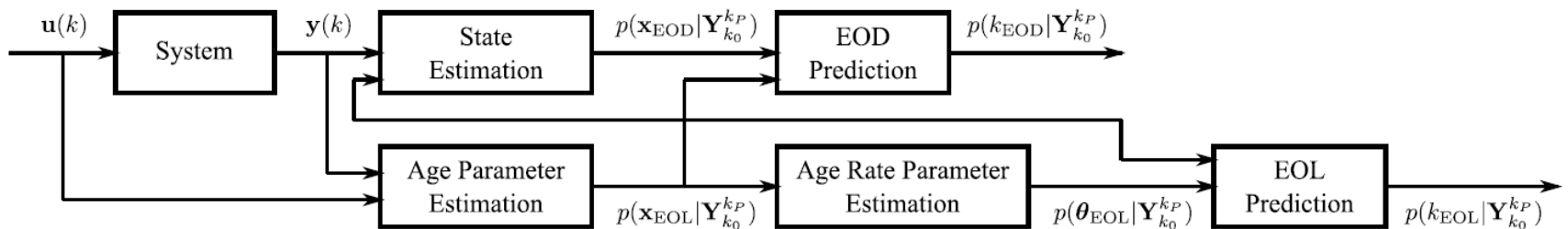
# Computational Algorithm

ComputeEOL( $k_o, k_h, \mathbf{x}(k_o), \mathbf{U}_{k_o, k_h}, \mathbf{V}_{k_o, k_h}$ )

1.  $\mathbf{X}_{k_o, k_h}(k_o) \leftarrow \mathbf{x}(k_o)$  *// Set initial state*
2. **for**  $k = k_o$  **to**  $k_h - 1$  **do**
3.   **if**  $T_f(\mathbf{X}_{k_o, k_h})(k)$  *// Check if failure state*
4.     **return**  $k$  *// Return current time as EOL*
5.   **end if**
6.    $\mathbf{X}_{k_o, k_h}(k + 1) \leftarrow f(\mathbf{X}_{k_o, k_h}(k), \mathbf{U}_{k_o, k_h}(k), \mathbf{V}_{k_o, k_h}(k))$  *// Update state*
7. **end for**
8. **if**  $T_f(\mathbf{X}_{k_o, k_h})(k_h)$  *// Check if failure state*
9.   **return**  $k_h$  *// Return current time ( $k_h$ ) as EOL*
10. **else**
11.   **return**  $\infty$  *// Return infinity*
12. **end if**

# Integrated Prognostics Architecture

- System (battery) gets inputs (current) and produces outputs (voltage)
- State estimation computes estimate of state given estimates of age parameters
- EOD prediction computes prediction of time of EOD, given state and age parameter estimates
- Age parameter estimation computes estimates of age parameters
- Age rate parameter estimation computes parameters defining aging rate progression
- EOL prediction computes prediction of time of EOL, given age parameter and age rate parameter estimates



# State Estimation

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- What is the current system state and its associated uncertainty?
  - Input: system outputs  $y$  from  $k_0$  to  $k$ ,  $y(k_0:k)$
  - Output:  $p(x(k), \theta(k) | y(k_0:k))$
- Battery models are nonlinear, so require nonlinear state estimator (e.g., extended Kalman filter, particle filter, unscented Kalman filter)
- Use unscented Kalman filter (UKF)
  - Straight forward to implement and tune performance
  - Computationally efficient (number of samples linear in size of state space)

# Prediction

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- Most algorithms operate by simulating samples forward in time until  $E$
- Algorithms must account for several sources of uncertainty besides that in the initial state
  - A representation of that uncertainty is required for the selected prediction algorithm
  - A specific description of that uncertainty is required (e.g., mean, variance)

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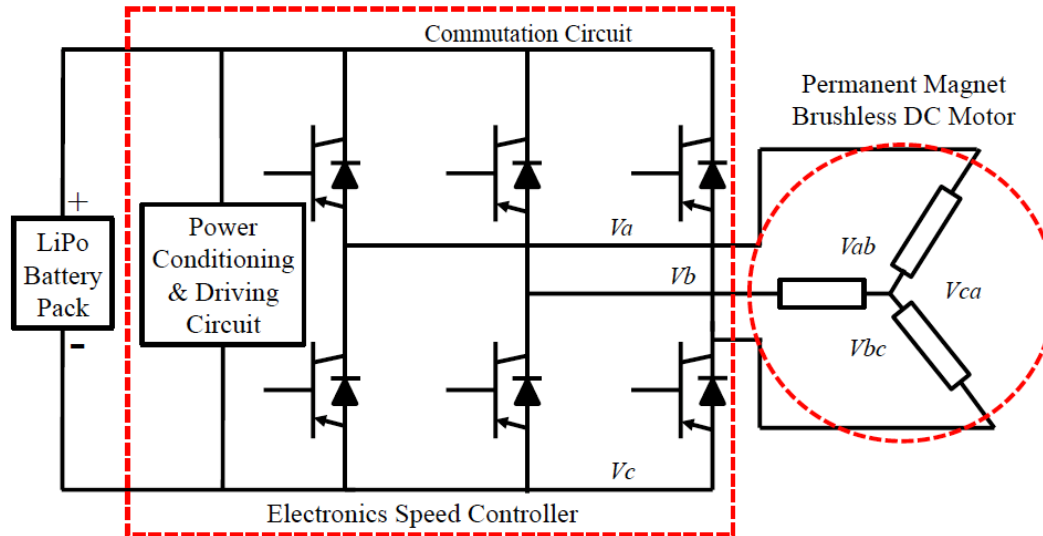
End of Introduction Section

# **DISCUSSION AND QUESTIONS?**

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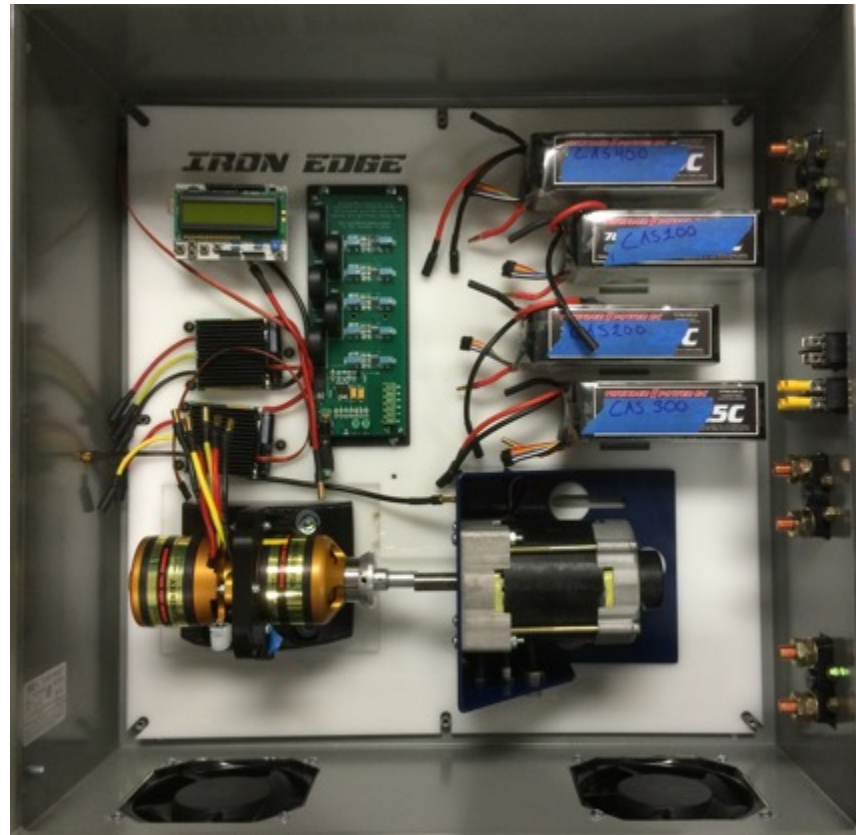
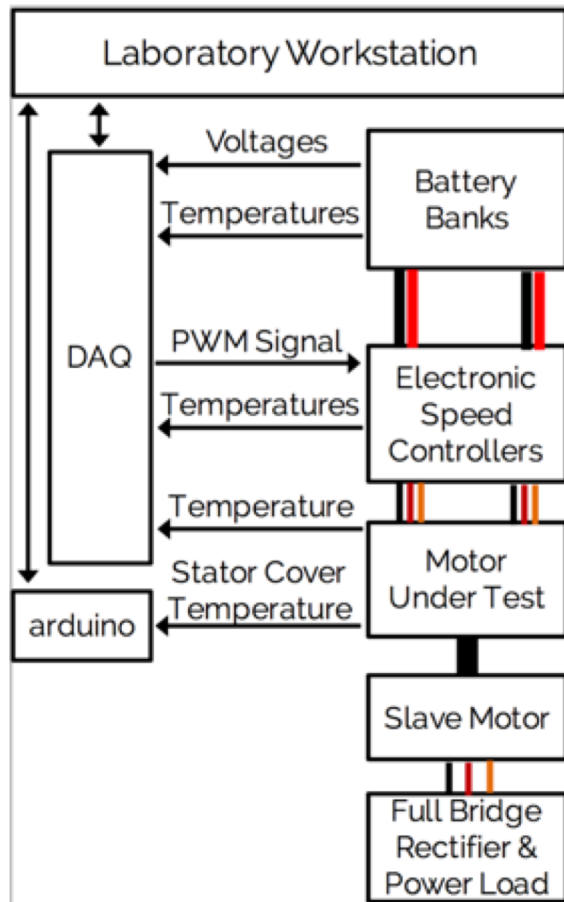
# **ELECTRIC VEHICLE POWERTRAIN**

# Electric Propulsion System



- **LiPo Batteries**
  - Lithium corrosion, plating, electrolyte layer formation, and contact losses
- **Permanent Magnet Brushless DC Motors**
  - Bearing wear, and electrical faults in the form of poor contacts and insulation deterioration
- **Electronics Speed Controllers**
  - MOSFETs are not synchronized while operating, or when the switching circuit is malfunctioning
- **Study Cascading faults**
- **Effects of component level aging/degradation on system performance**

# Hardware in Test Loop



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# **ACCELERATED AGING TOOL FOR PROGNOSTICS RESEARCH**

# Accelerated Aging

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- Traditionally used to assess the reliability of products with expected lifetimes in the order of thousands of hours
  - in a considerably shorter amount of time
- Provides opportunities for the development and validation of prognostic algorithms
- Such experiments are invaluable since run-to-failure data for prognostics is rarely or never available
- Unlike reliability studies, prognostics is concerned not only with time to failure of devices but with the degradation process leading to an irreversible failure
  - This requires in-situ measurements of key output variables and observable parameters in the accelerated aging process with the associated time information
- Thermal, electrical and mechanical overstresses are commonly used for accelerated aging tests of electronics

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# **Example: Electrical overstress aging of Power Transistors**

# Accelerate aging strategy (1/2)

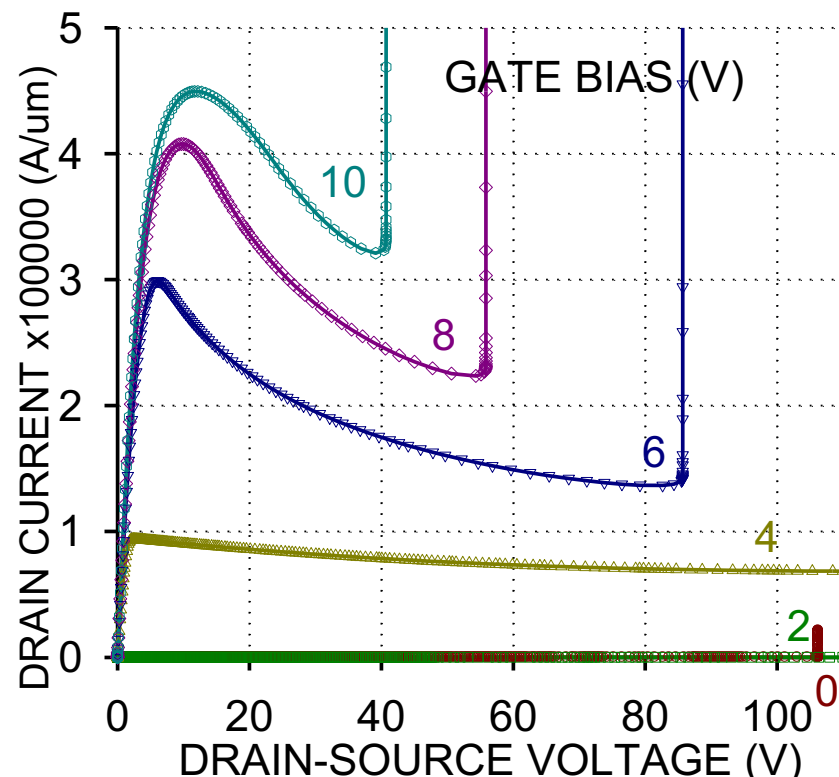
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- Main strategy
  - application of electrical overstress
  - fixed junction temperature in order to avoid thermal cycles
  - avoid package related failures
- Accelerated test conditions are achieved by electrical operation regime of the devices at temperatures within the range below maximum ratings and above the room temperatures.

# Accelerate aging strategy (2/2)

- The highest acceleration factor for aging can be achieved in the proximity of the Safe Operation Area (SOA) boundary
- Instability points represent the critical voltages and currents limiting the SOA
- An electrical regime close to the SOA boundary serves as the accelerator factor (stressor) and it is expected to reduce the life of the device
- The SOA boundary shifts closer to the origin as the temperature increases

Simulated I-V characteristics and instability boundary at 300° K for power MOSFET.



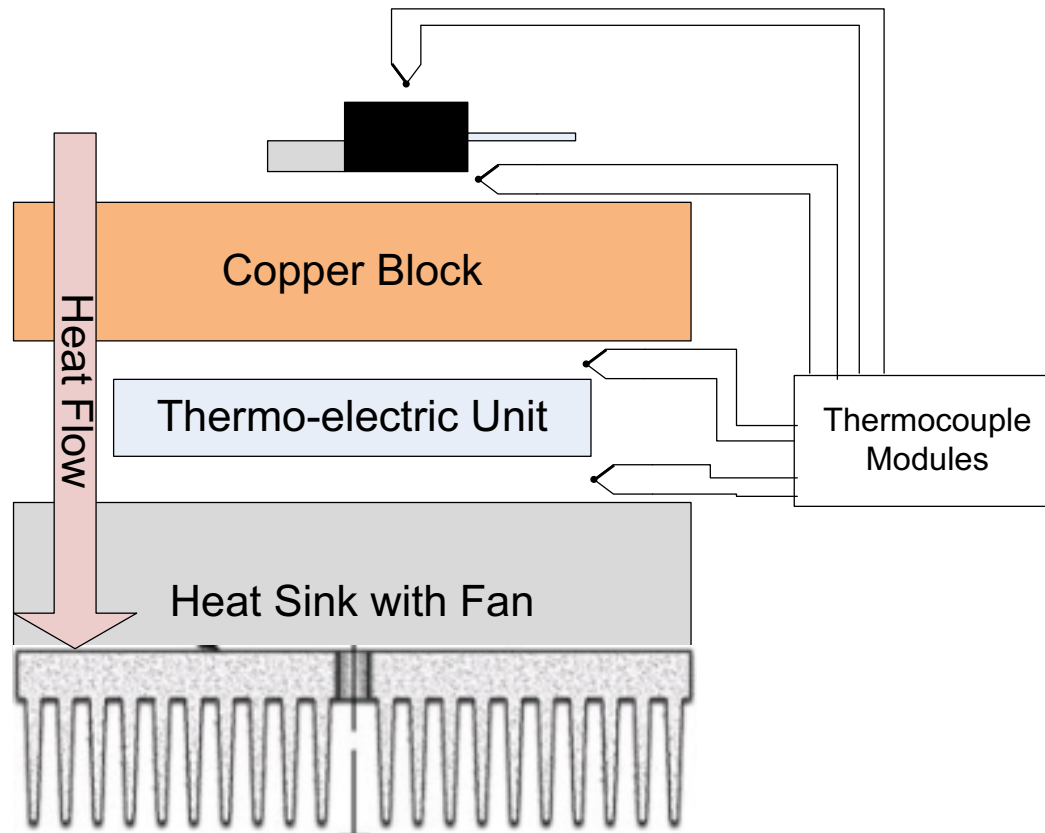
# Aging system description (1/3)

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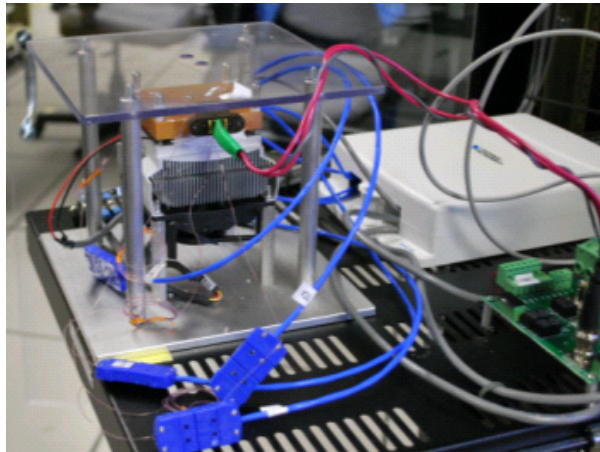
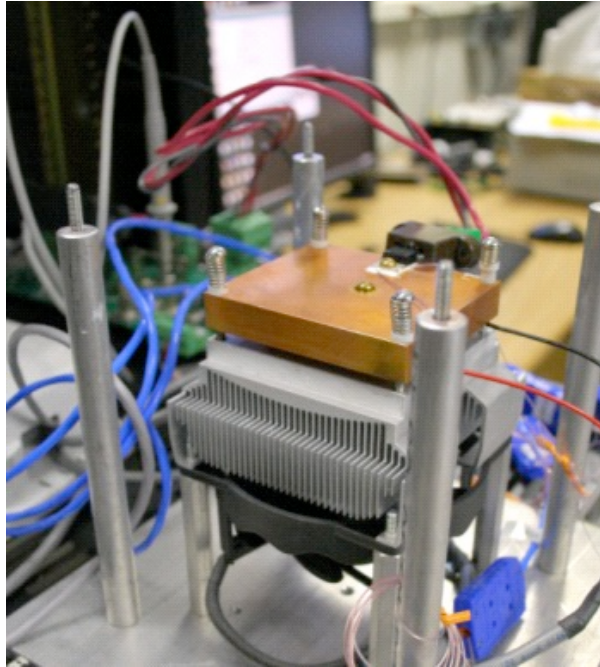
- Three main components in terms of hardware
  - Electrical operation unit of the device
    - custom made printed circuit boards for the instrumentation circuitry and gate drivers
    - commercially available power supplies and function generator to control the operation of the DUT
  - An in-situ measurement unit of key electrical and thermal parameters
    - commercially available measurement and data acquisition for slow and high speed measurements
  - Thermal block section for monitoring and control of the temperature

# Aging system description (2/3)

Thermal block for measurement and control of device temperature



# Aging system description (3/3)



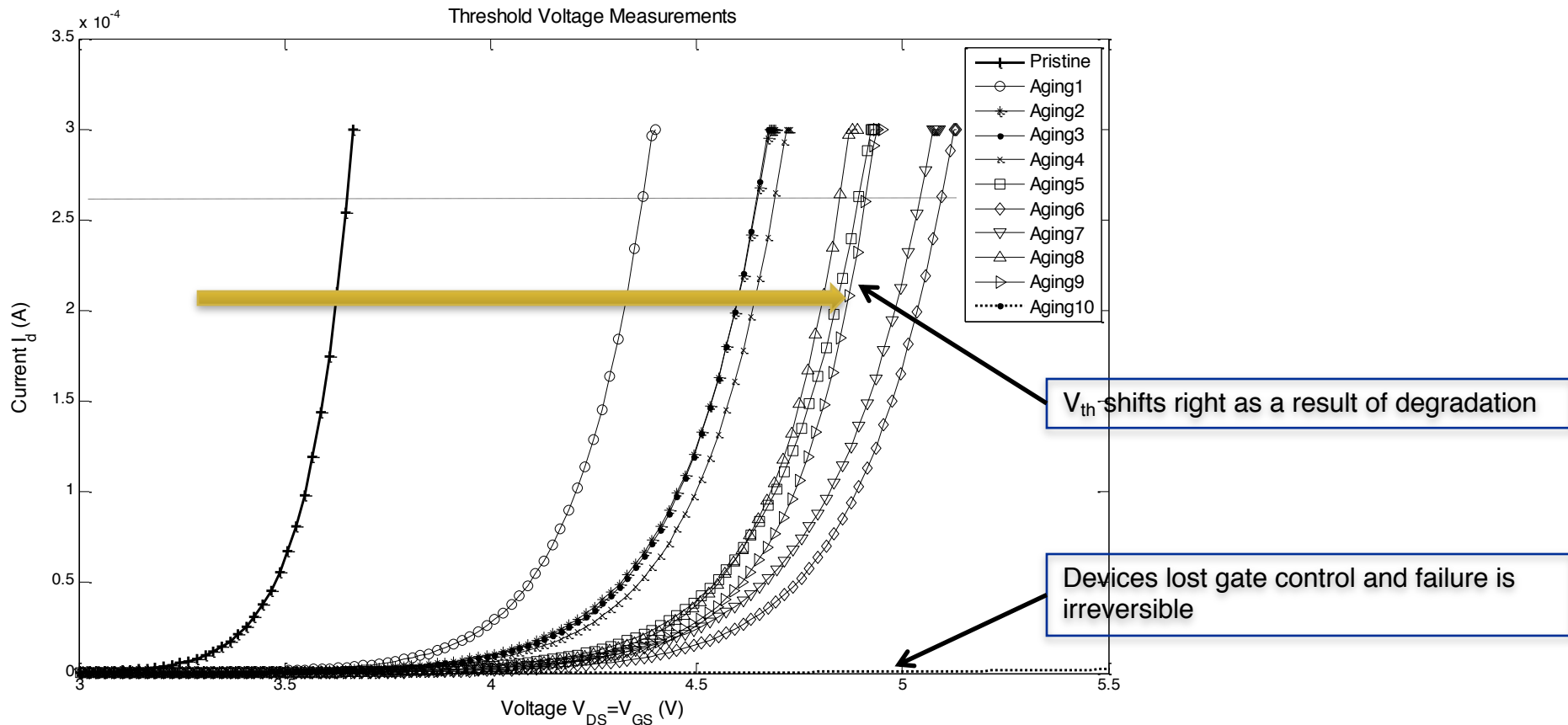
# Experiment on power MOSFET (1/2)

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- IRF520Npbf power MOSFET
  - TO220 package, 100V/9A.
- Electrical overstress used as acceleration factor.  
High potential at the gate
  - $V_{gs} = 50V$ ,  $V_{gs}$  rating is 20V max.
  - $V_{ds} = 2.4V$  with a 0.2 ohm load.
- Temperatures kept below maximum rating  
 $T_{j,max} = 175^{\circ} C$
- Objective is to induce failure mechanism on the gate structure

# Experiment on power MOSFET (2/2)

- Degradation process as observed on threshold voltage ( $V_{th}$ )



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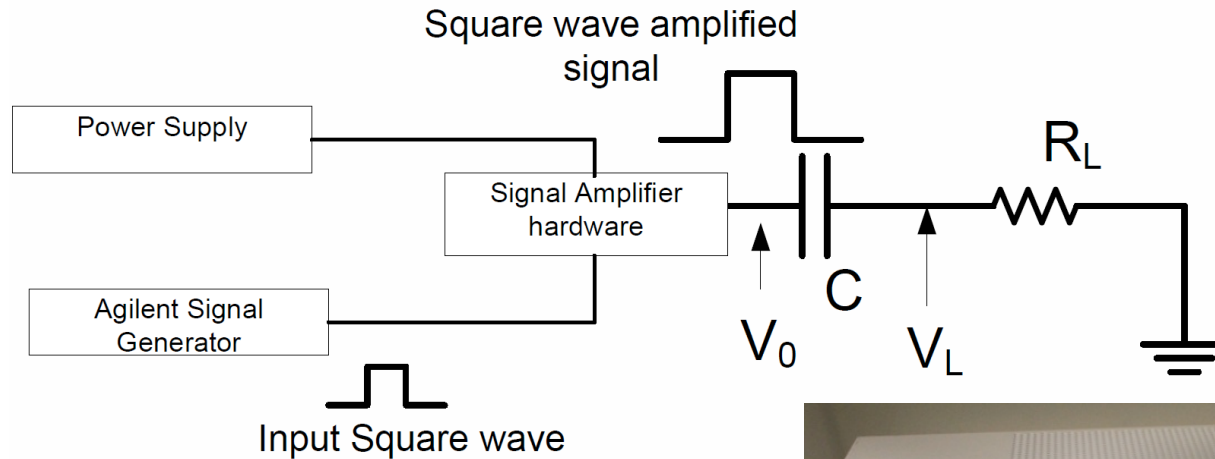
# **Example: Electrical overstress aging of Electrolytic Capacitors**

# Accelerated aging system

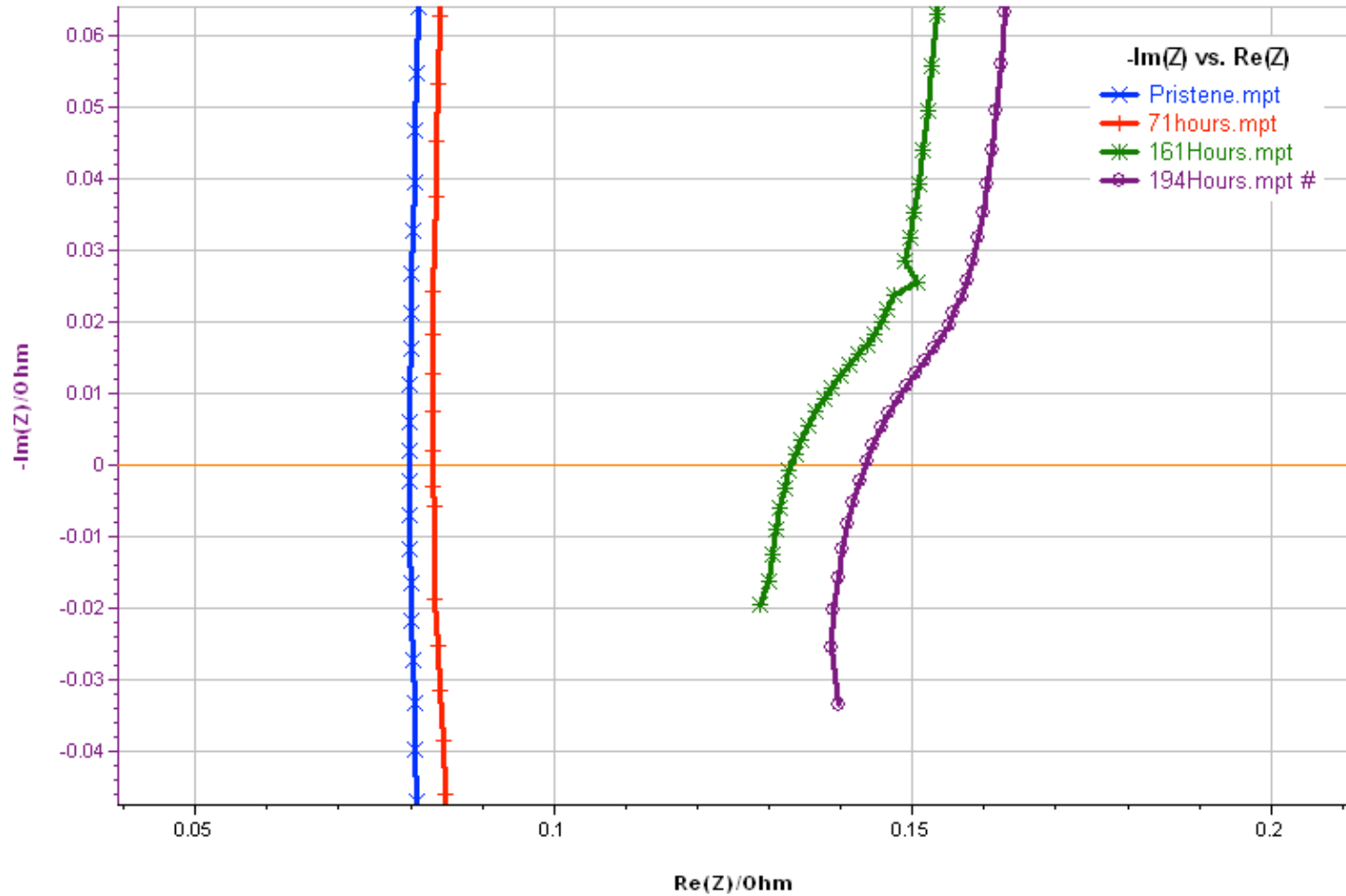
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- Allows for the understanding of the effects of failure mechanisms, and the identification of leading indicators of failure essential for the development of physics-based degradation models and RUL prediction
- Electrolytic capacitor 2200 $\mu$ F, 10V and 1A
- Electrical overstress >200 hrs
  - Square signal at 200 mHz with 12V amplitude and 100 ohm load

# Electrical Overstress Aging System



# Degradation observed on EIS measurements



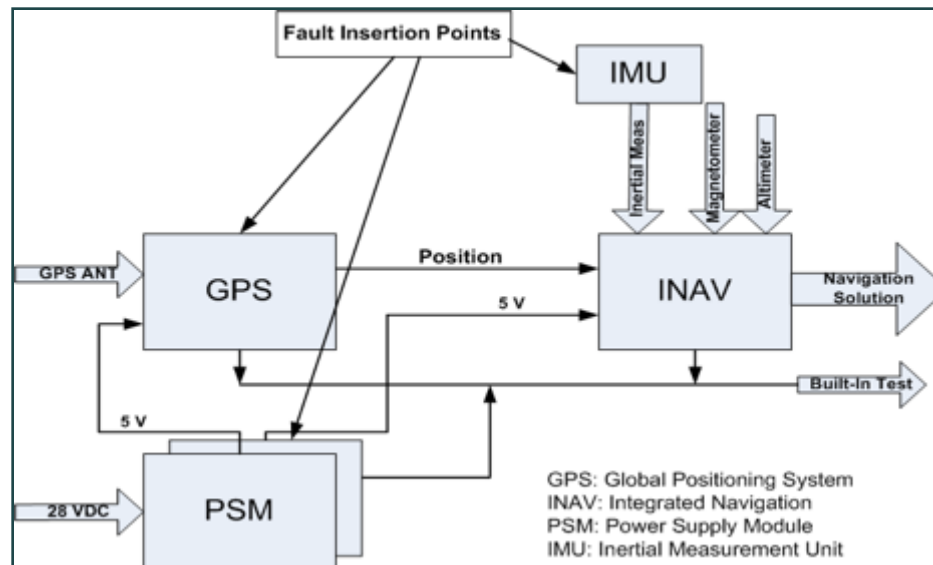
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# **CASE STUDY I: PROGNOSTICS OF ELECTROLYTIC CAPACITORS**

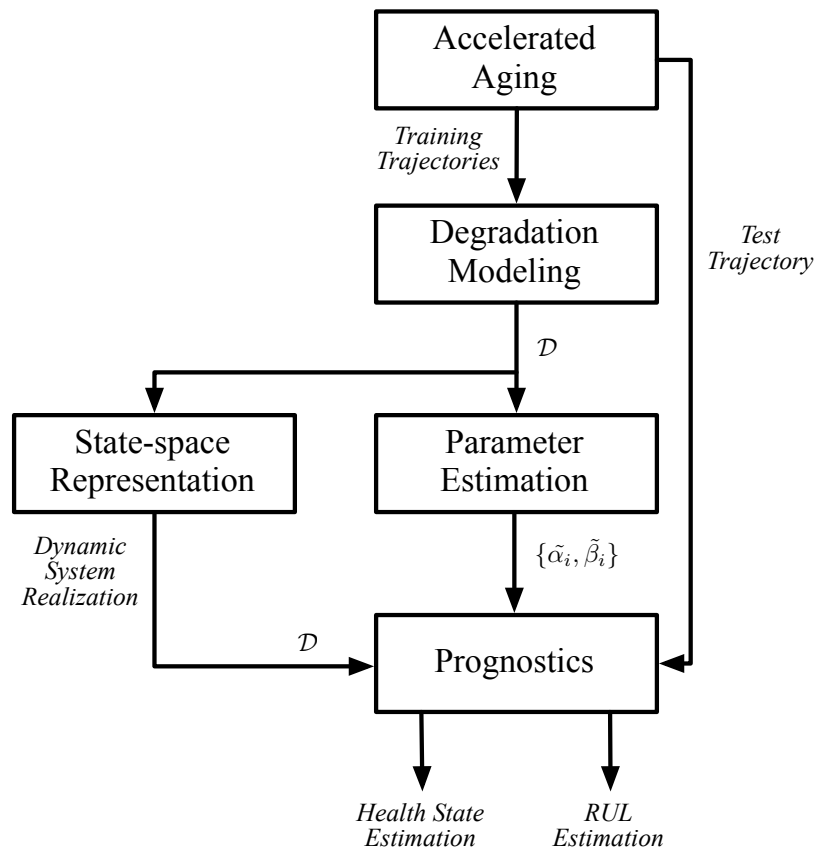
**MODEL-BASED APPROACH EXAMPLE**

# Case Study: Avionics System

- Integrated Avionics systems consists of:
  - Global Positioning System (GPS) module
  - Integrated navigation (INAV) module combines output of the GPS model and Inertial measurement unit
  - Power Supply module

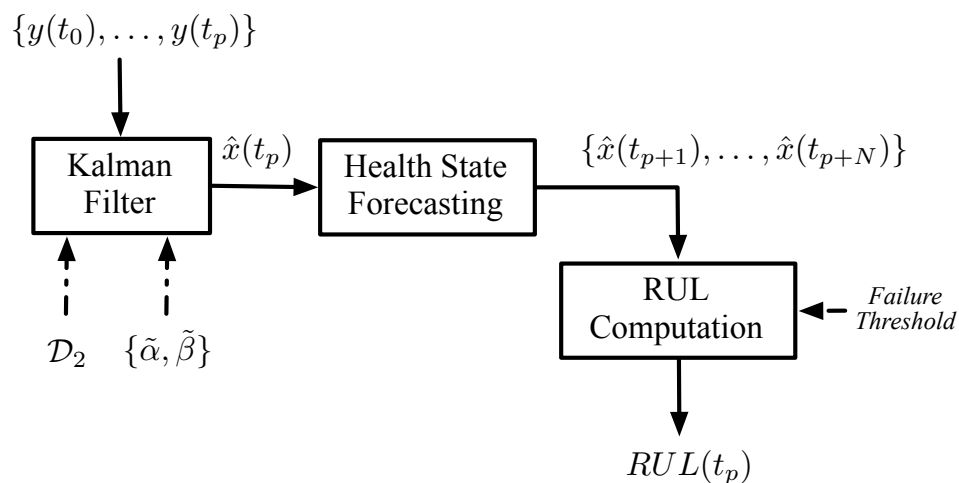


# Methodology



$$x_k = Ax_{k-1} + Bu_{k-1} + w_{k-1}$$

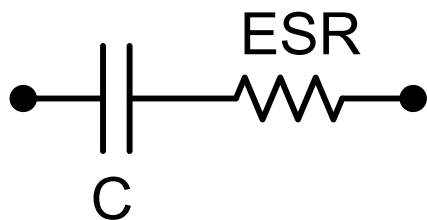
$$y_k = Hx_k + v_k$$



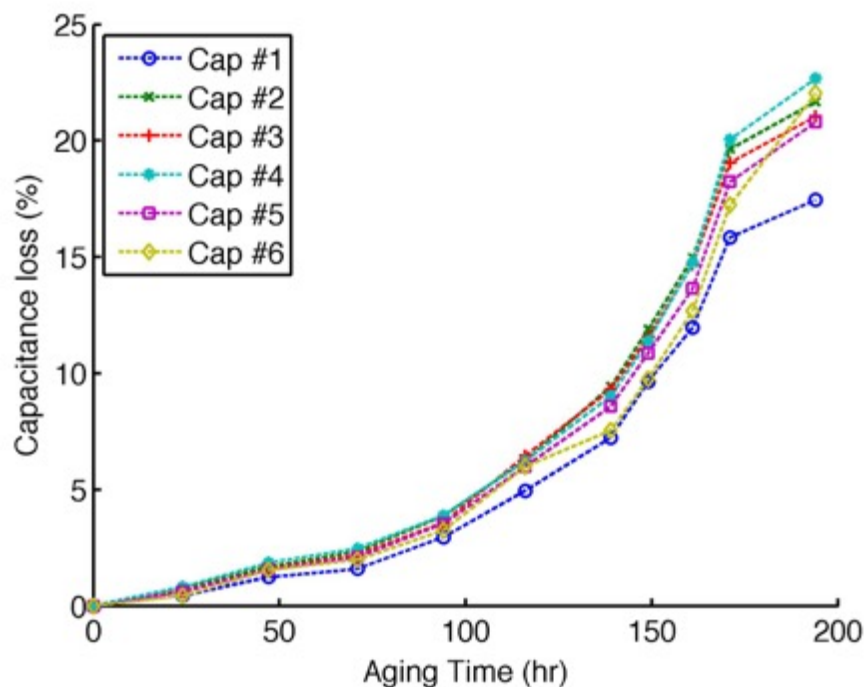
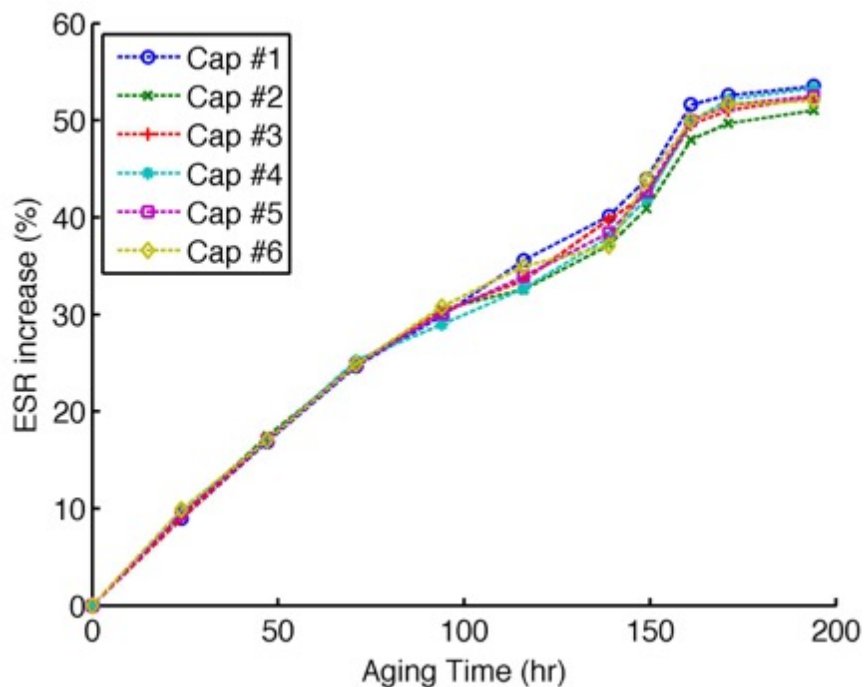
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# **Accelerated Aging and Precursors of Failure Features**

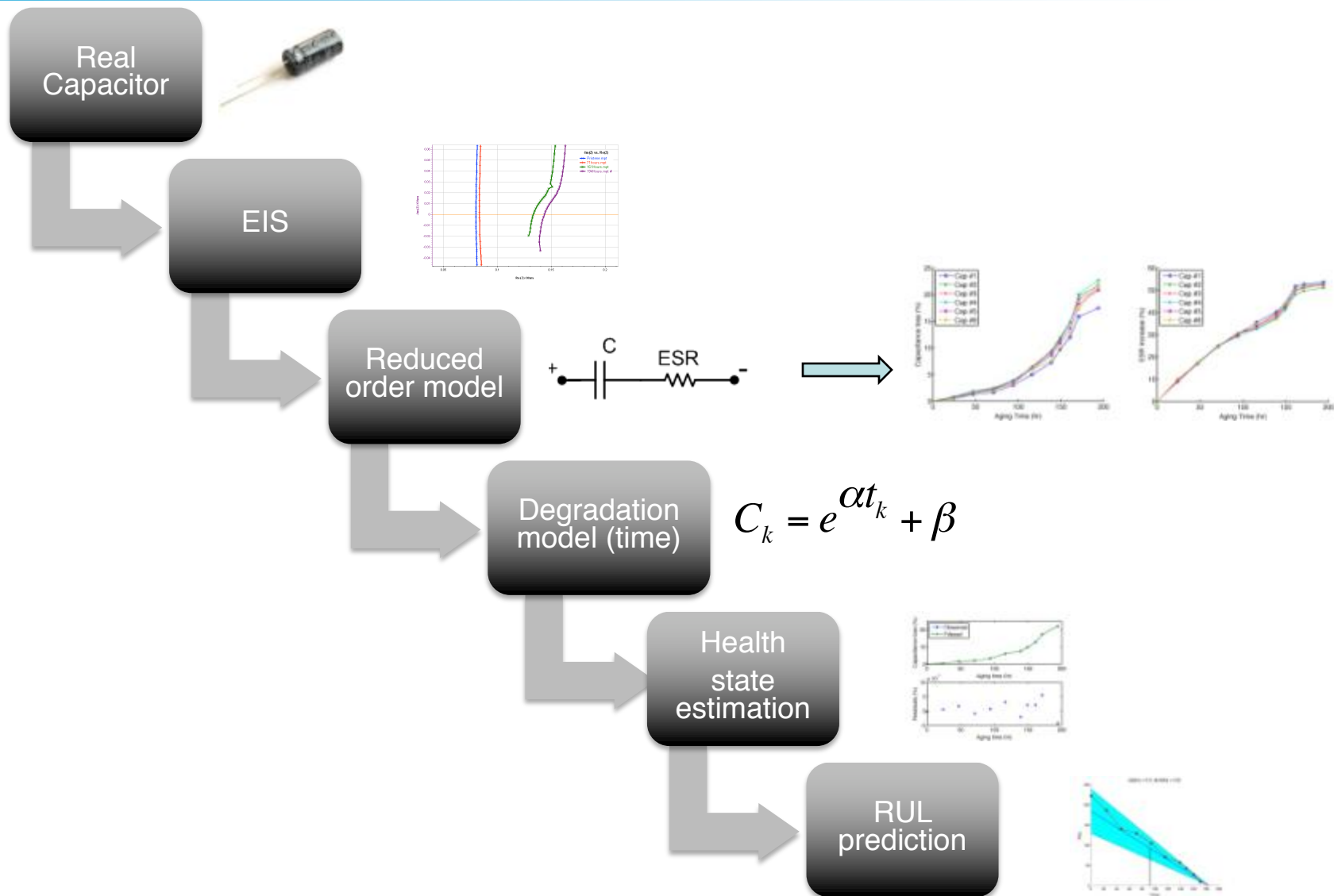
# Degradation on lumped parameter model



C and ESR are estimated from EIS measurements



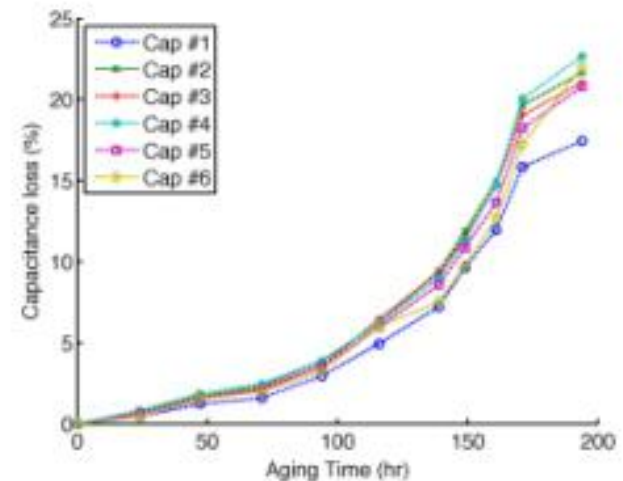
# Methodology



# Empirical degradation model

- Based on observed degradation from capacitance parameter
- Using training capacitor data to estimate degradation model parameters
- Assumed exponential model based on capacitance loss
- Parameter estimation with least-squared regression

$$C_k = e^{\alpha t_k} + \beta$$



# Degradation model results

| Validation test | Test capacitor | Training capacitor | $\alpha$<br>(95% CI)              | $\beta$<br>(95% CI)                  | $\sigma_v^2$ |
|-----------------|----------------|--------------------|-----------------------------------|--------------------------------------|--------------|
| $T_2$           | #2             | #1, #3–#6          | <b>0.0162</b><br>(0.0160, 0.0164) | <b>-0.8398</b><br>(-1.1373, -0.5423) | 1.8778       |
| $T_3$           | #3             | #1, #2, #4–#6      | <b>0.0162</b><br>(0.0160, 0.0164) | <b>-0.8287</b><br>(-1.1211, -0.5363) | 1.9654       |
| $T_4$           | #4             | #1–#3, #5, #6      | <b>0.0161</b><br>(0.0159, 0.0162) | <b>-0.8217</b><br>(-1.1125, -0.5308) | 1.8860       |
| $T_5$           | #5             | #1–#4, #6          | <b>0.0162</b><br>(0.0161, 0.0164) | <b>-0.7847</b><br>(-1.1134, -0.4560) | 2.1041       |
| $T_6$           | #6             | #1–#5              | <b>0.0169</b><br>(0.0167, 0.0170) | <b>-1.0049</b><br>(-1.2646, -0.7453) | 2.9812       |

- The optimal parameter presented along the 95% confidence interval.
- The residuals are modelled as a normally distributed random variable with zero mean and variance

# Prognostics algorithm

---

- Implementation of prognostics algorithm with Kalman filter
- Capacitance loss considered as state variable
- EIS measurements and lumped parameter model used to obtain measured capacitance loss values
- Empirical degradation model used to generate the state transition equation
- Use one Capacitor for testing and the rest for model parameter estimation (leave one out test)
- Failure threshold of 20% drop on capacitance based on MIL-C-62F

# Kalman filter implementation

- State transition equation derived from degradation model

$$C_k = e^{\alpha t_k} + \beta$$



$$\frac{\partial C}{\partial t} = \alpha C - \alpha \beta$$

$$\frac{C_t - C_{t - \Delta t}}{\Delta t} = \alpha C_{t - \Delta t} - \alpha \beta$$

$$C_t = (1 + \alpha \Delta t) C_{t - \Delta t} - \alpha \beta \Delta t$$

$$C_k = (1 + \alpha \Delta_k) C_{k-1} - \alpha \beta \Delta_k$$



- State-space model for filter implementation

$$C_k = A_k C_{k-1} + B_k u + v$$

$$y_k = h C_k + w, \text{ where}$$

$$A_k = (1 + \Delta t),$$

$$B_k = -\alpha \beta \Delta_k,$$

$$h = 1, u = 1.$$

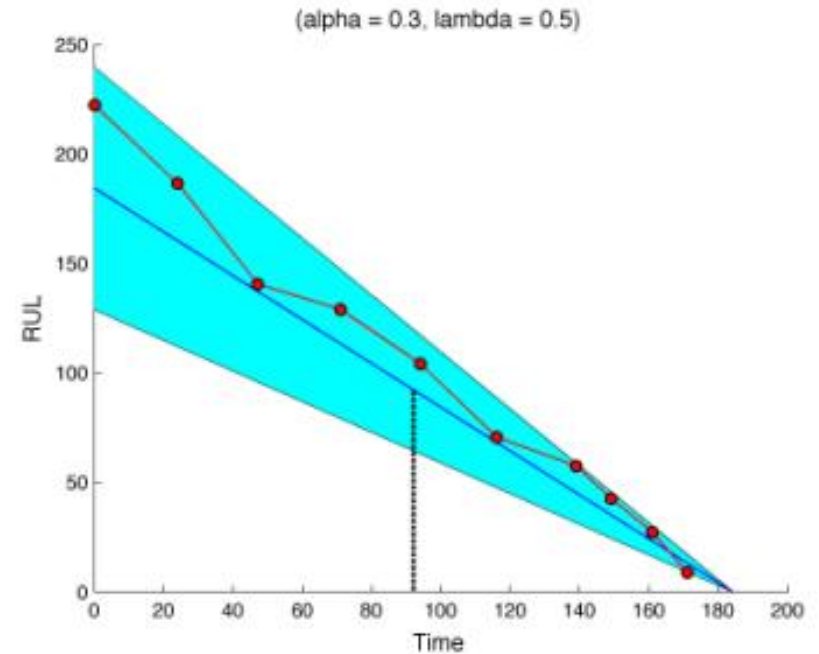
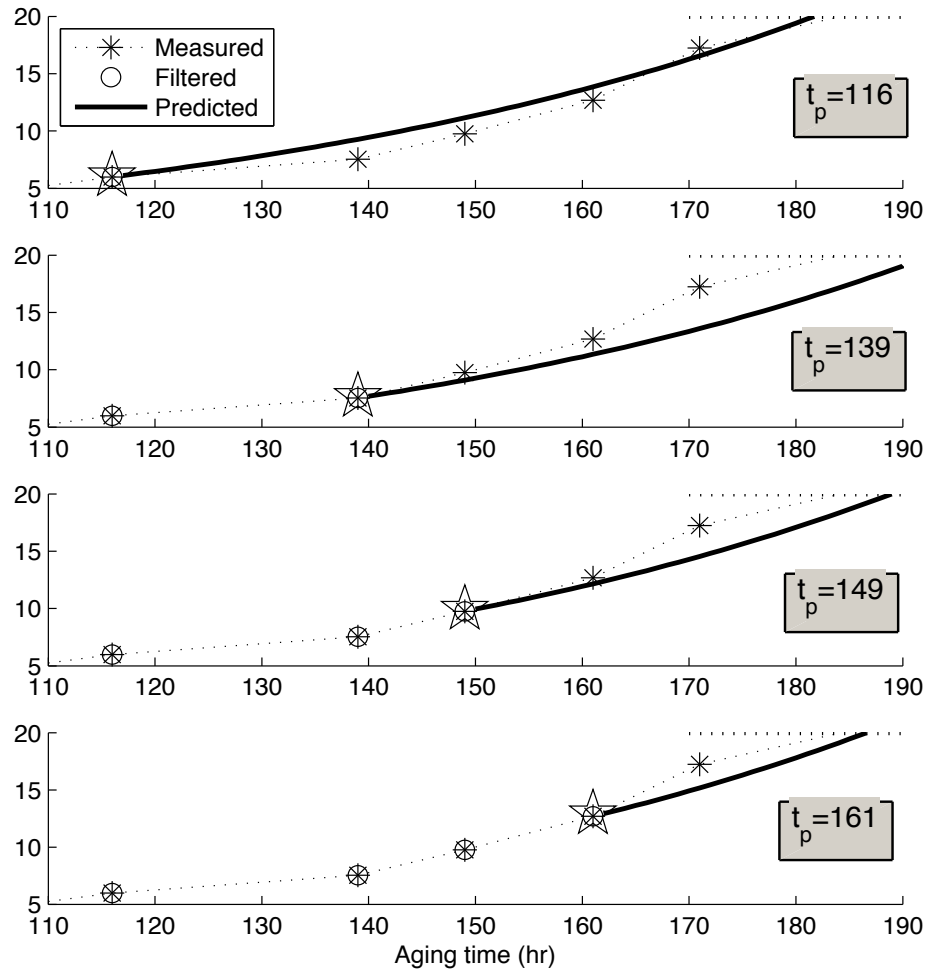
# Prediction mode

---

- Assumed measurements are not available at some point in time
- Filter used in forecasting mode to predict future states
- Predictions done at 1 hr. intervals
- State transition equation used to propagate state (n: number of prediction steps,  $l$ : last measurement at  $t_l$ )

$$\hat{C}_{l+n} = A^n C_l + \sum_{i=0}^{n-1} A^i B$$

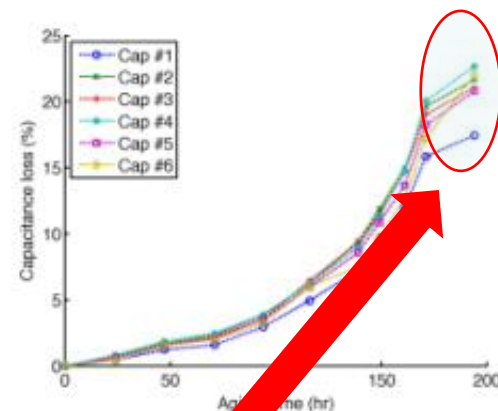
# Tracking and forecasting (Cap. #6)



# Relative Accuracy

$$RA = 100 \left( 1 - \frac{RUL^* - RUL'}{RUL^*} \right)$$

| $t_p$ | $RA_{T2}$ | $RA_{T3}$ | $RA_{T4}$ | $RA_{T5}$ | $RA_{T6}$ | $\widetilde{RA}$ |
|-------|-----------|-----------|-----------|-----------|-----------|------------------|
| 24    | 94.8      | 95.5      | 91.9      | 96.9      | 99.7      | <b>95.5</b>      |
| 47    | 97.4      | 99.3      | 96.4      | 96.7      | 91.7      | <b>96.7</b>      |
| 71    | 87.5      | 91.9      | 84.5      | 94.1      | 97.1      | <b>91.9</b>      |
| 94    | 85.6      | 90        | 78.9      | 94.8      | 94.2      | <b>90</b>        |
| 116   | 86        | 99.1      | 76.5      | 98        | 96.2      | <b>96.2</b>      |
| 139   | 77.8      | 95.8      | 53.1      | 96.7      | 81.1      | <b>81.1</b>      |
| 149   | 82.1      | 98.4      | 46.9      | 94.8      | 86.6      | <b>86.6</b>      |
| 161   | 77.2      | 87.3      | 16.6      | 87.5      | 89.8      | <b>87.3</b>      |
| 171   | 26.6      | 26.4      | N/A       | 34.8      | 63.7      | <b>30.7</b>      |



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End of Case Study I:

# **DISCUSSION AND QUESTIONS?**

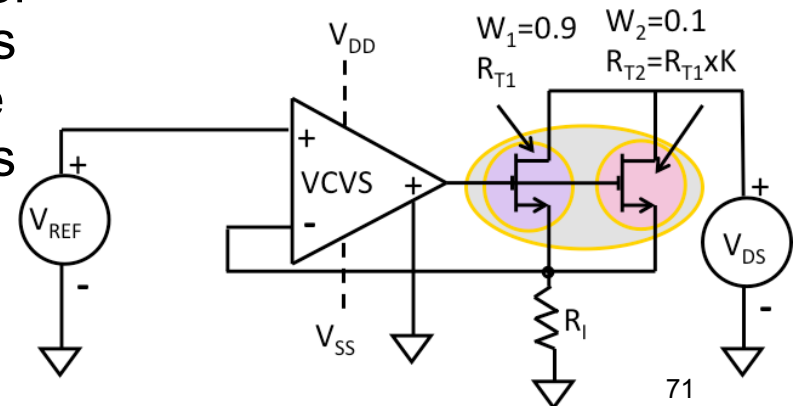
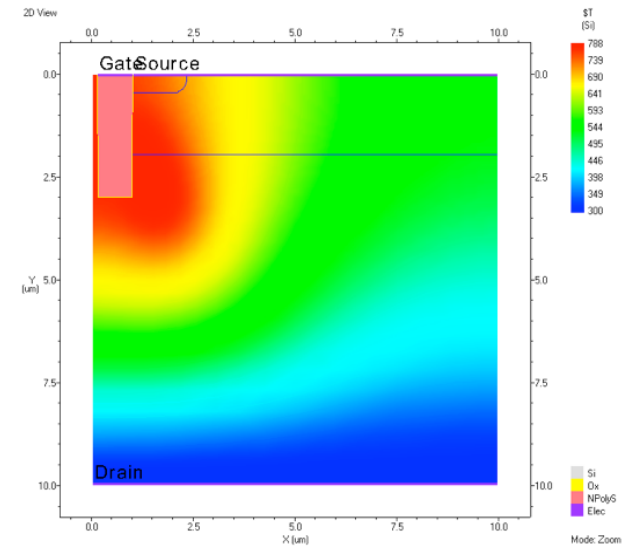
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# **CASE STUDY II: PROGNOSTICS OF POWER TRANSISTORS**

**PRECURSORS OF FAILURE EXAMPLE**

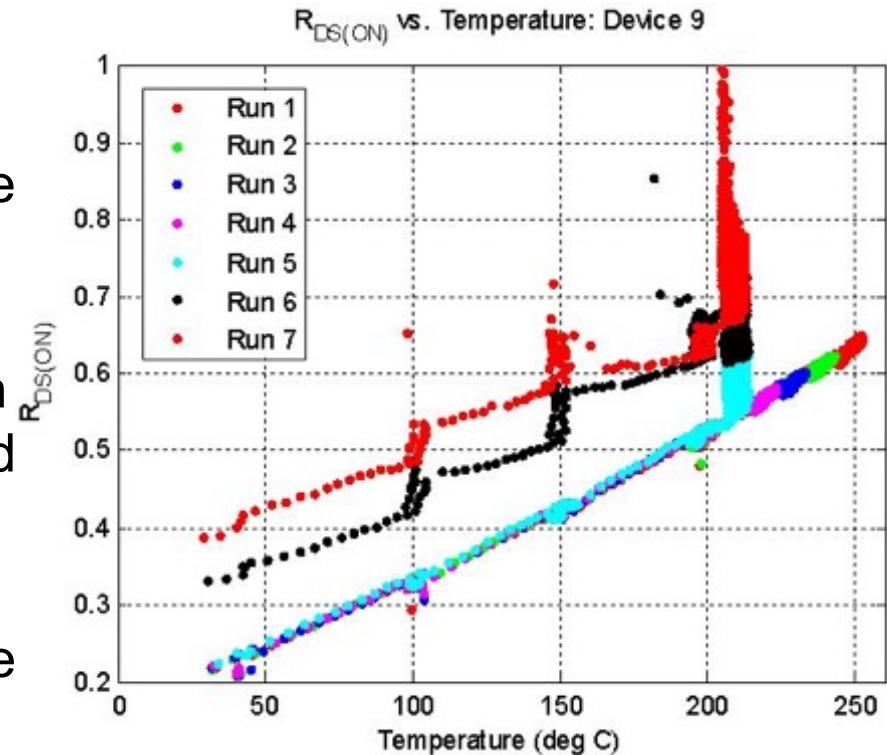
# Modeling for Power MOSFET under electrical overstress

- Two-transistor model is shown to be a good candidate for a degradation model for model-based prognostics
- The model parameters  $K$  and  $W1$  could be varied as the device degrades as a function of usage time, loading and environmental conditions
- Parameter  $W1$  defines the area of the healthy transistors, the lower this area, the larger the degradation in the two-transistor model. In addition, parameter  $K$  serves as a scaling factor for the thermal resistance of the degraded transistors, the larger this factor, the larger the degradation in the model.



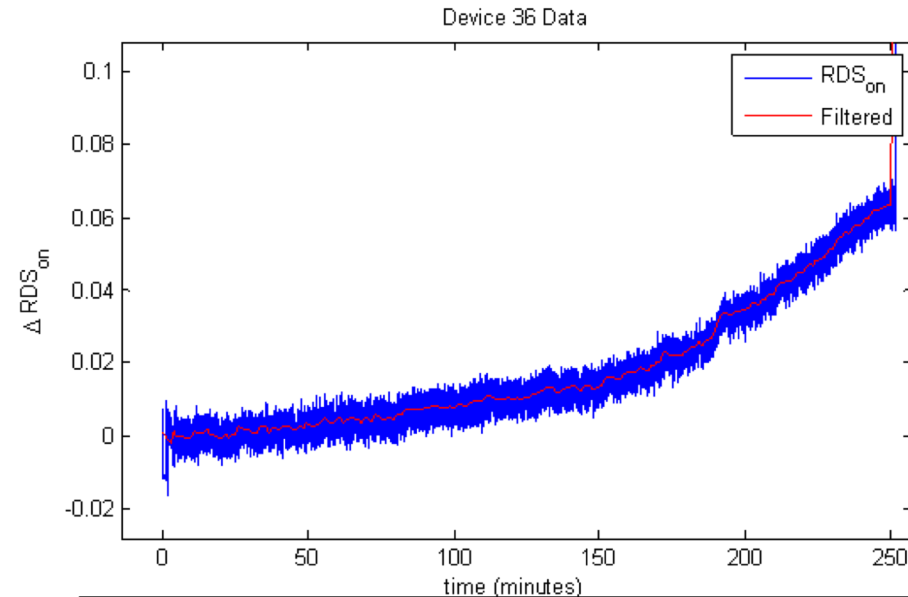
# Precursor of Failure

- As case temperature increases, ON-resistance increases
- This relationship shifts as the degradation of the device increases
- For a degraded state, ON-resistance will be higher at any given case temperature
- This is consistent with the die-attach damage since it results on increased junction temperature operation
- This plot can be used directly for fault detection and diagnostics of the die-attach failure mechanism



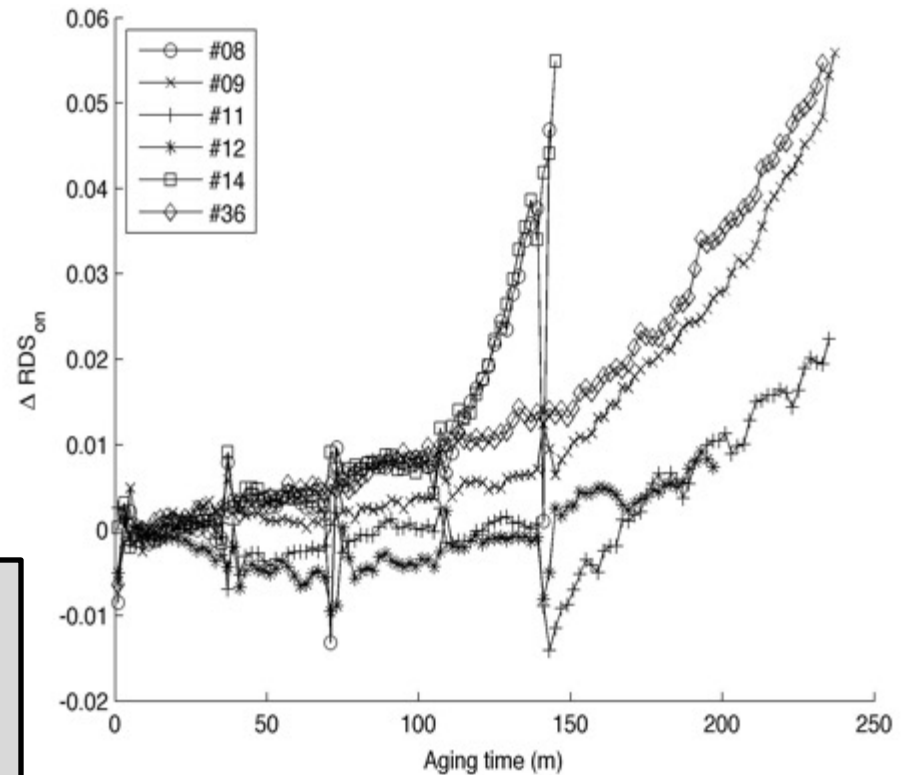
# Degradation process data

Normalized ON-state resistance ( $\Delta R_{DS(ON)}$ ) and filtered trajectory for device #36



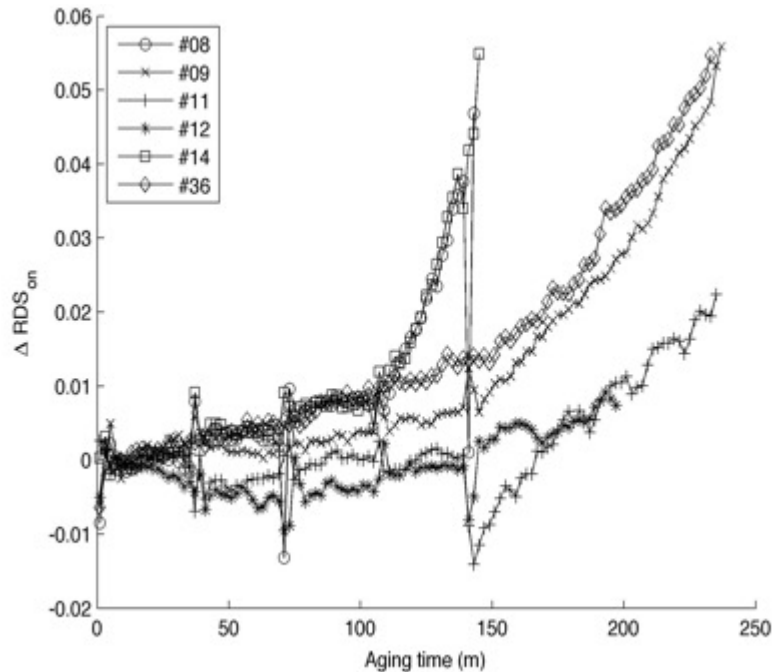
- Cases #08, #09, #11, #12 and #14 are used for algorithm development purposes.
- Case #36 is used to test the algorithms.

Normalized ON-state resistance ( $\Delta R_{DS(ON)}$ ) and filtered trajectory for device #36



# Empirical Degradation Model

- An empirical degradation model was selected for the model-based algorithms
- Exponential based function to capture degradation process
- Two parameters in the model which will be estimated on-line



$$\Rightarrow \Delta R_{DS(ON)} = \alpha(e^{\beta t} - 1)$$

---

# **Prediction of Remaining Life**

# RUL Prediction Methodology Considerations

---

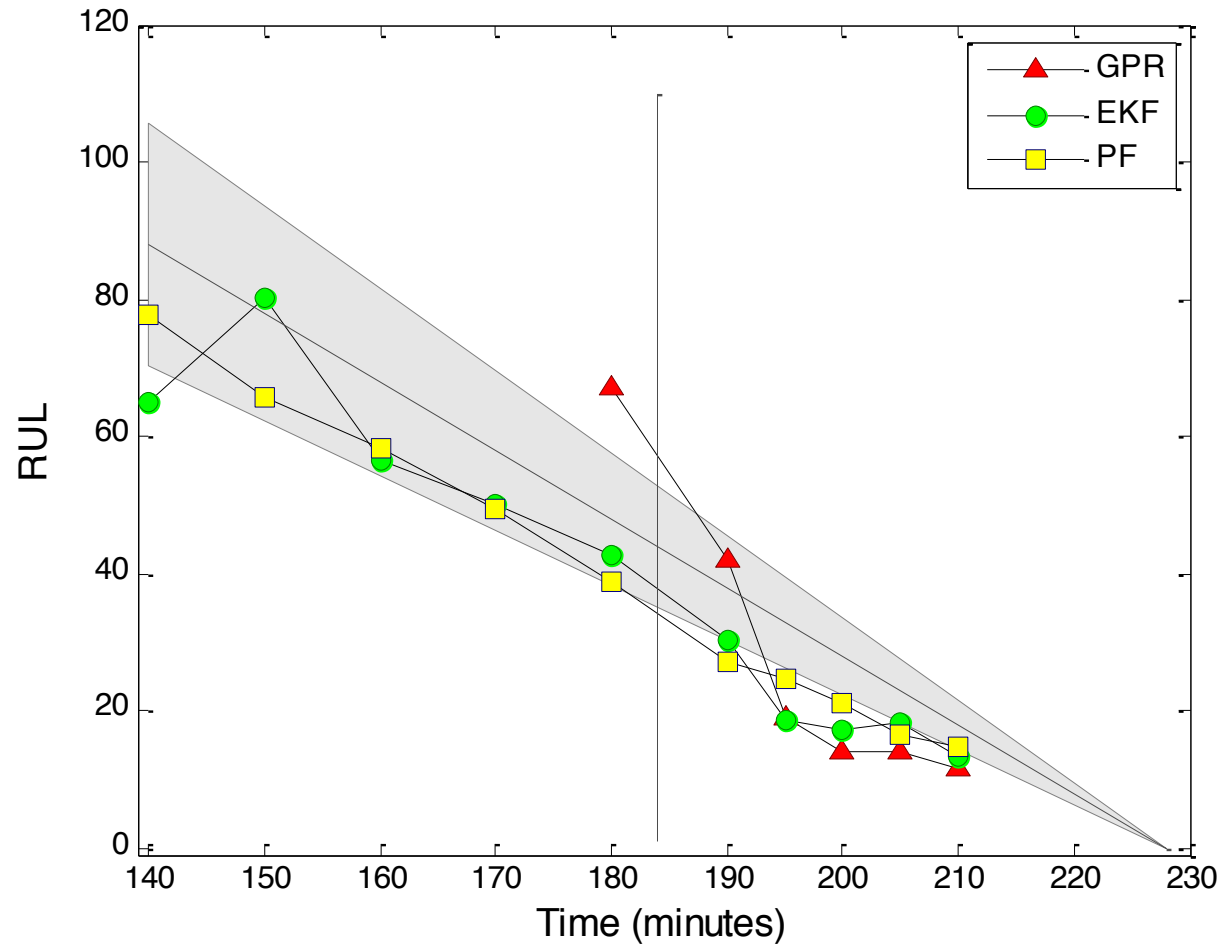
- A single feature is used to assess the health state of the device ( $\Delta R_{DS(ON)}$ )
- It is assumed that the die-attached failure mechanism is the only active degradation during the accelerated aging experiment
- Furthermore,  $\Delta R_{DS(ON)}$  accounts for the degradation progression from nominal condition through failure
- Periodic measurements with fixed sampling rate are available for  $\Delta R_{DS(ON)}$
- A crisp failure threshold of 0.05 increase in  $\Delta R_{DS(ON)}$  is used
- The prognostics algorithm will make a prediction of the remaining useful life at time  $t_p$ , using all the measurements up to this point either to estimate the health state at time  $t_p$  in a regression framework or in a Bayesian state tracking framework
- It is also assumed that the future load conditions do not vary significantly from past load conditions

# RUL Prediction Algorithms

---

- Gaussian Process Regression
  - Algorithm development cases used to select covariance matrix structure and values
- Extended Kalman filter
  - Empirical degradation model
  - State variable: Normalized ON-resistance and degradation model parameters
  - Arbitrary values for measurement and process noise variance
- Particle filter
  - Empirical degradation model
  - State variable: Normalized ON-resistance, degradation model parameters
  - Exponential growth model used for degradation model parameters
  - Arbitrary values for measurement and process noise variance

# RUL estimation results



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End of Case Study II:

# **DISCUSSION AND QUESTIONS?**

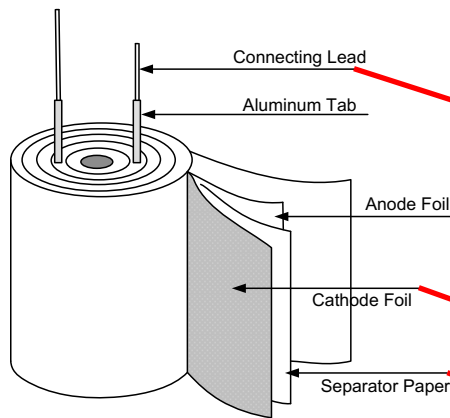
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# **CASE STUDY III: PHYSICS- BASED PROGNOSTICS OF CAPACITORS**

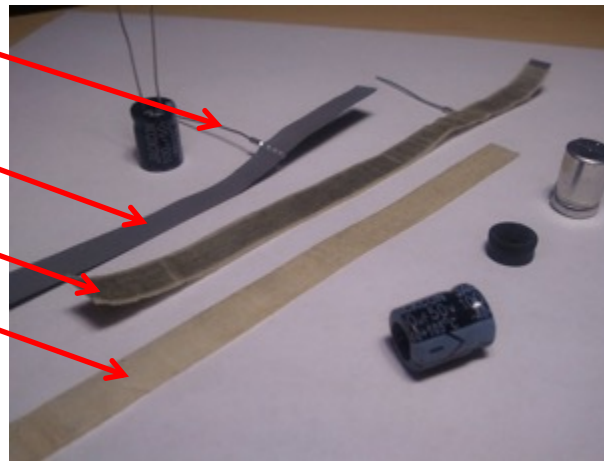
**DEGRADATION MODELING EXAMPLE**

# Capacitor Structure

- An aluminum electrolytic capacitor, consists of
  - Cathode aluminum foil,
  - Electrolytic paper, electrolyte
  - Aluminum oxide layer on the anode foil surface, which acts as the dielectric.
  - Equivalent series resistance (ESR) and capacitance(C) are electrical parameters that define capacitor health

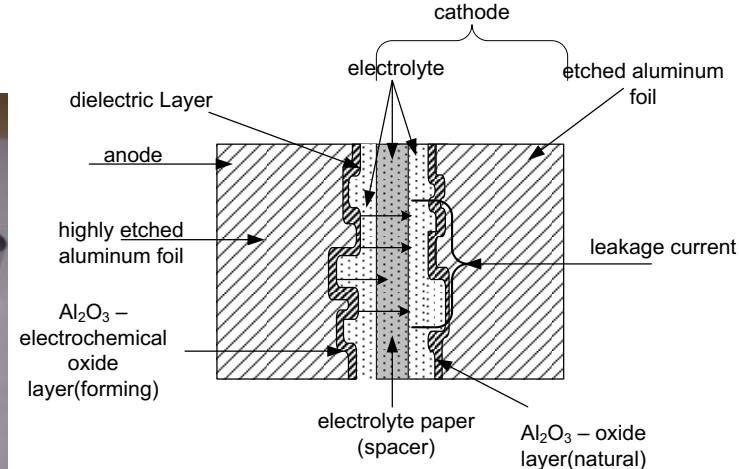


Physical Structure



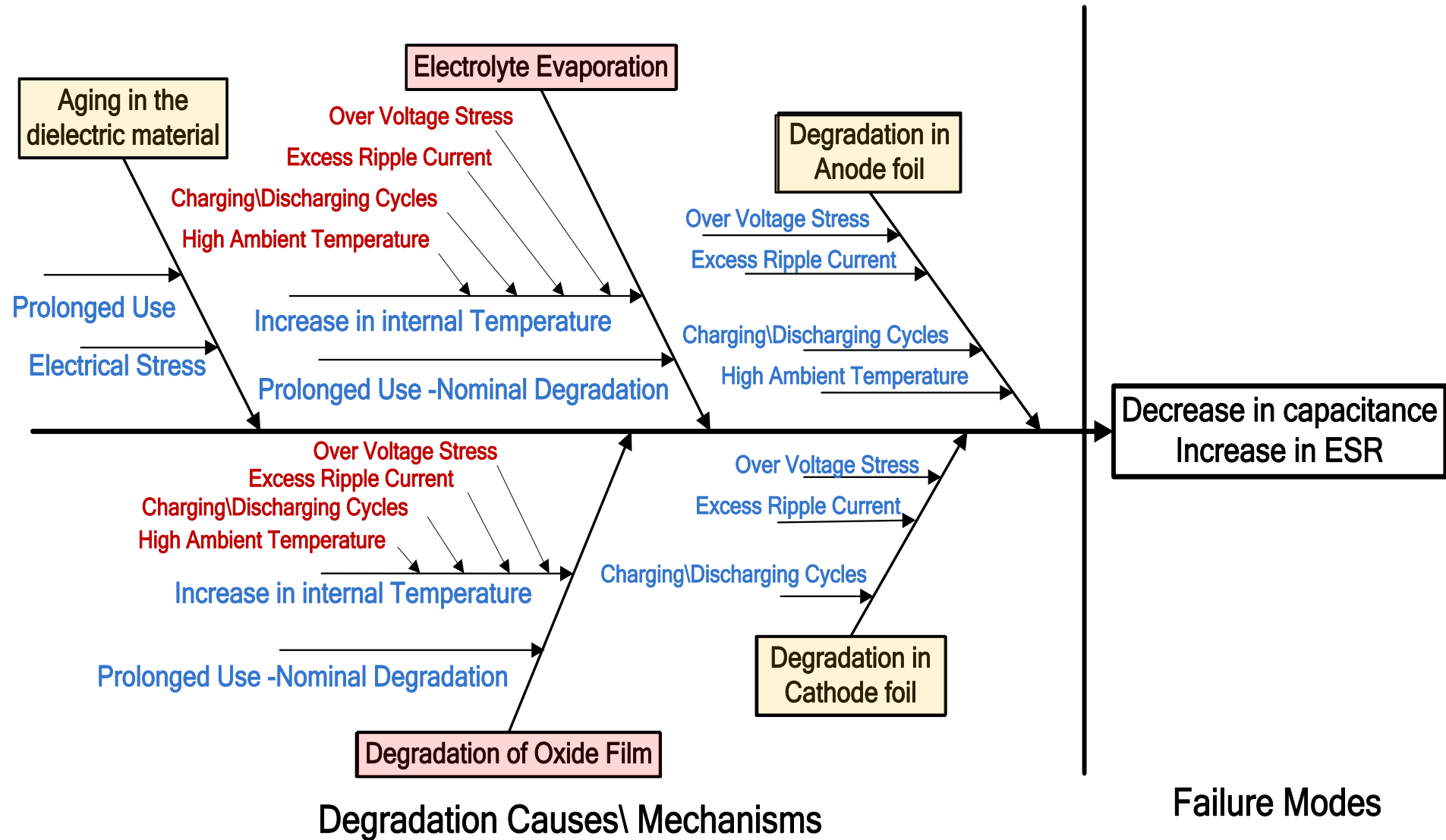
Ref :<http://en.wikipedia.org/wiki/File:ElectrolyticCapacitorDisassembled.jpg>

Open Structure



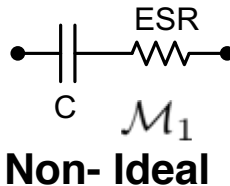
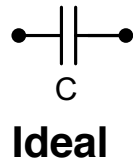
Internal Structure

# Degradation Mechanisms

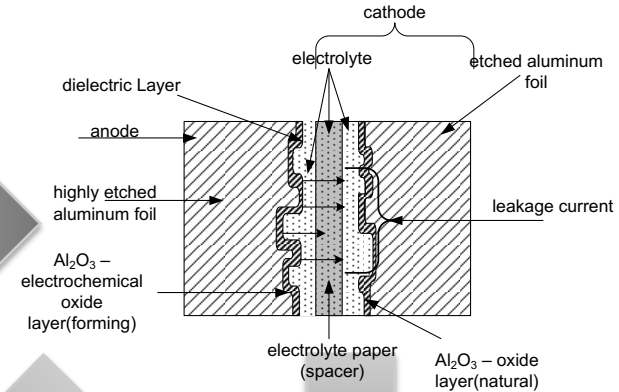


# Capacitor Degradation Model

Pristine Capacitor



**Degradation**



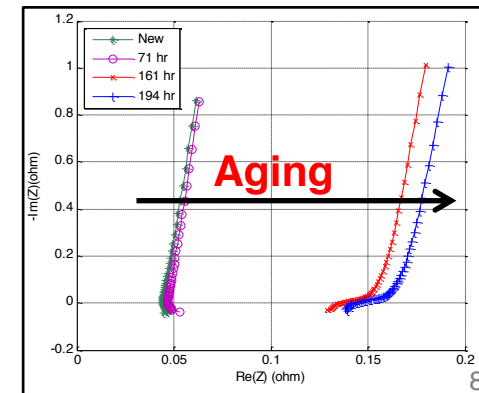
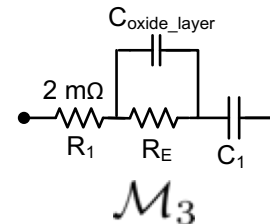
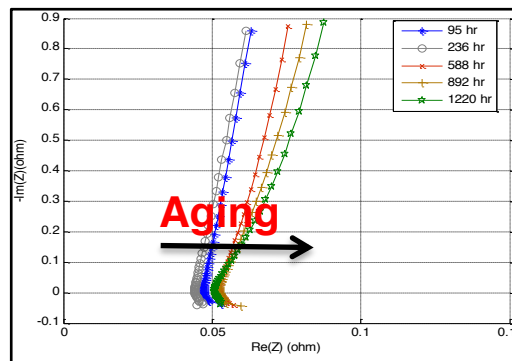
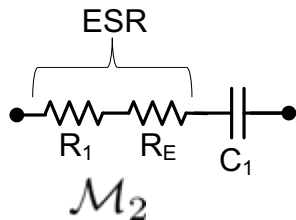
**Electrolyte volume  $V_e$  maximum  
Capacitance Value maximum**

**Thermal Stress**

**Electrical Stress**

**Avg. surface area decreases ( $A_s$ ) +  
oxide layer breakdown**

**Electrolyte degradation + Decrease  
in ( $A_s$ ) + crystallization +oxide layer  
breakdown**



# Empirical Model with static parameters

- This empirical model represents an approximation of lumped parameter model  $\mathcal{M}_1$

$$\mathcal{E}_1 : C_l(t) = e^{\alpha t} + \beta,$$

- $\alpha$  and  $\beta$  are degradation model parameters estimated from the experimental data.
- The following system structure is used in the implementation of the filtering and the prediction using the Kalman filter.

$$x_k = A_k x_{k-1} + B_k u + v,$$

$$y_k = h x_k + w,$$

$$A_k = (1 + \Delta_k),$$

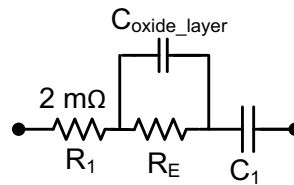
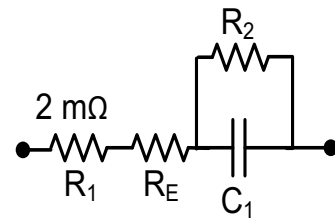
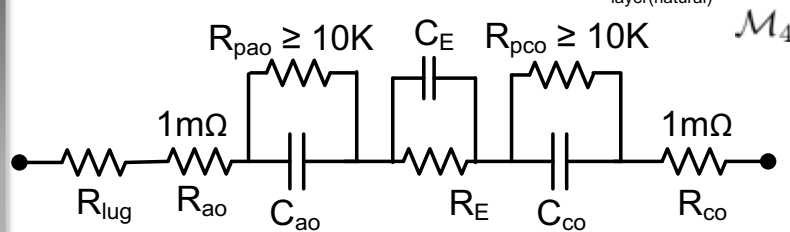
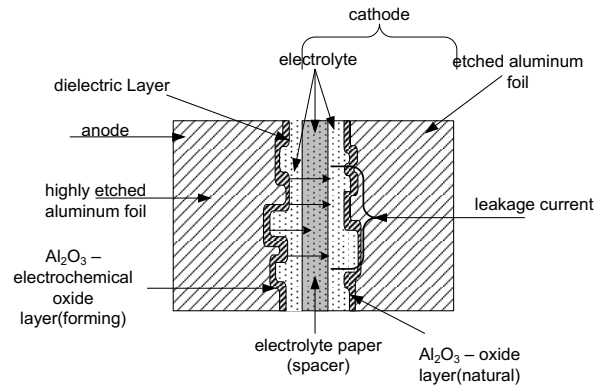
$$B_k = -\alpha \beta \Delta_k,$$

$$h = 1,$$

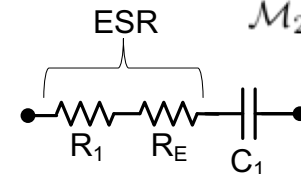
- The state variable ( $x_k$ ) at aging time ( $t_p$ ) is the percentage loss in Capacitance.
- Process noise was estimated from the model regression for the empirical model
- Measurement noise was estimated from the EIS measurements

# Degradation Model: Electrical Circuit Equivalent

## Granularity of Degradation Models

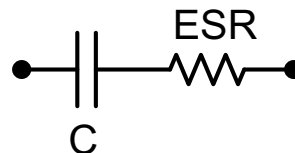


$\mathcal{M}_3$



$\mathcal{M}_2$

$\mathcal{M}_1$



# Capacitance Degradation Model

- Decrease in electrolyte volume :

$$V_e(t) = V_{e0} - (w_e A_s j_{eo} t)$$

(1)

where:

$V$ : dispersed volume at time  $t$ ,  $V_e$ : initial electrolyte volume

$A_s$ : surface area of evaporation,  $j_{eo}$  : evaporation rate

$t$  : time in minutes,  $w_e$  = volume of ethyl glycol molecule

- Capacitance (C) ): Physics-Based Model:

$$C = (2\epsilon_R \epsilon_O A_s) / d_C$$

(2)

- Electrolyte evaporation dominant degradation phenomenon
  - First principles: Capacitance degradation as a function of electrolyte loss

where:

$$\mathcal{D}_1 : C(t) = \left( \frac{2\epsilon_R \epsilon_0}{d_C} \right) \left( \frac{V_{e0} - V_e(t)}{j_{eo} t w_e} \right),$$

$C$  : capacitance of the capacitor,  
 $\epsilon_R$  : relative dielectric constant,  
 $\epsilon_O$  : permittivity of free space,  
 $d_C$  : oxide thickness.

(3)

# Capacitance Degradation Model

- Oxide breakdown observed - experimental data
- The breakdown factor is exp. function of electrolyte evaporation

$$C_{bk(t)} = \exp f(V_{eo} - V_{e(t)})$$

- Updated in capacitance degradation model :

$$C = (2\epsilon_R\epsilon_0 A_s c_{bk})/d_C,$$
$$\mathcal{D}_{11} : C(t) = c_{bk(t)} \left( \frac{2\epsilon_R\epsilon_0}{d_C} \right) \left( \frac{V_{eo} - V_e(t)}{j_{eo} t w_e} \right)$$

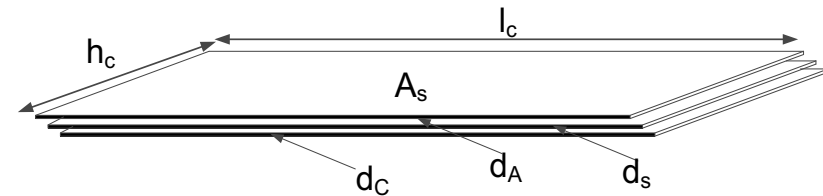
# Dynamic Model of Capacitance

From the structure of capacitor we have the electrolyte volume( $V_e$ ) expressed in the form of oxide surface area ( $A_s$ ) as :

(4)

$$V_e = A_s \cdot d_C,$$

$$A_s = \frac{V_e}{d_C}.$$



The first order discrete approximation for change in electrolyte volume can be expressed as:

$$\frac{dV_e}{dt} = -(w_e A_s j_{eo}),$$

(5)

$$V_{e(k+1)} = V_{e(k)} + \frac{dV_e}{dt} \Delta t,$$

$$V_{e(k+1)} = V_{e(k)} - (w_e A_s j_{eo}) \Delta t.$$

# Dynamic Model of Capacitance

$$V_{e(k)} = \frac{C_k}{2\epsilon_R\epsilon_0 c_{bk}} d_C^2,$$

$$V_{e(k)} = (C_k)\alpha$$

(6)

Similarly Capacitance can be expressed as :

$$C_{k+1}\alpha = C_k\alpha + \frac{dV_e}{dt}\Delta t,$$

$$C_{k+1}\alpha = C_k\alpha - (w_e A_s j_{eo})\Delta t, \text{ hence}$$

$$C_{k+1} = C_k - \frac{(w_e A_s j_{eo})}{\alpha} \Delta t.$$

(7)

The complete discrete time dynamic model for capacitance degradation can be summarized as :

$$\mathcal{D}_4 : C_{k+1} = C_k - \left( \frac{2\epsilon_R\epsilon_0 w_e A_s j_{eo} c_{bk}}{d_C^2} \right) \Delta t$$

# Dynamic Model of ESR

- Decrease in electrolyte volume :

$$V_e(t) = V_{e0} - (w_e A_s j_{eo} t)$$

- ESR
  - Based on mechanical structure and electrochemistry.
  - With changes in  $R_E$  (electrolyte resistance )

$$ESR = \frac{1}{2} \left( \frac{\rho_E d_C P_E e_{bk}(t)}{A_s} \right)$$

$$\mathcal{D}_2 : ESR(t) = \frac{1}{2} (\rho_E d_C P_E) \left( \frac{j_{eo} t w_e e_{bk}(t)}{V_e(t)} \right) \quad (8)$$

Dynamic ESR degradation Model :

$$\mathcal{D}_5 : \frac{1}{ESR_{k+1}} = \frac{1}{ESR_k} - \left( \frac{2w_e A_s j_{eo}}{\rho_E P_E d_C^2 e_{bk}(t)} \right) \Delta t$$

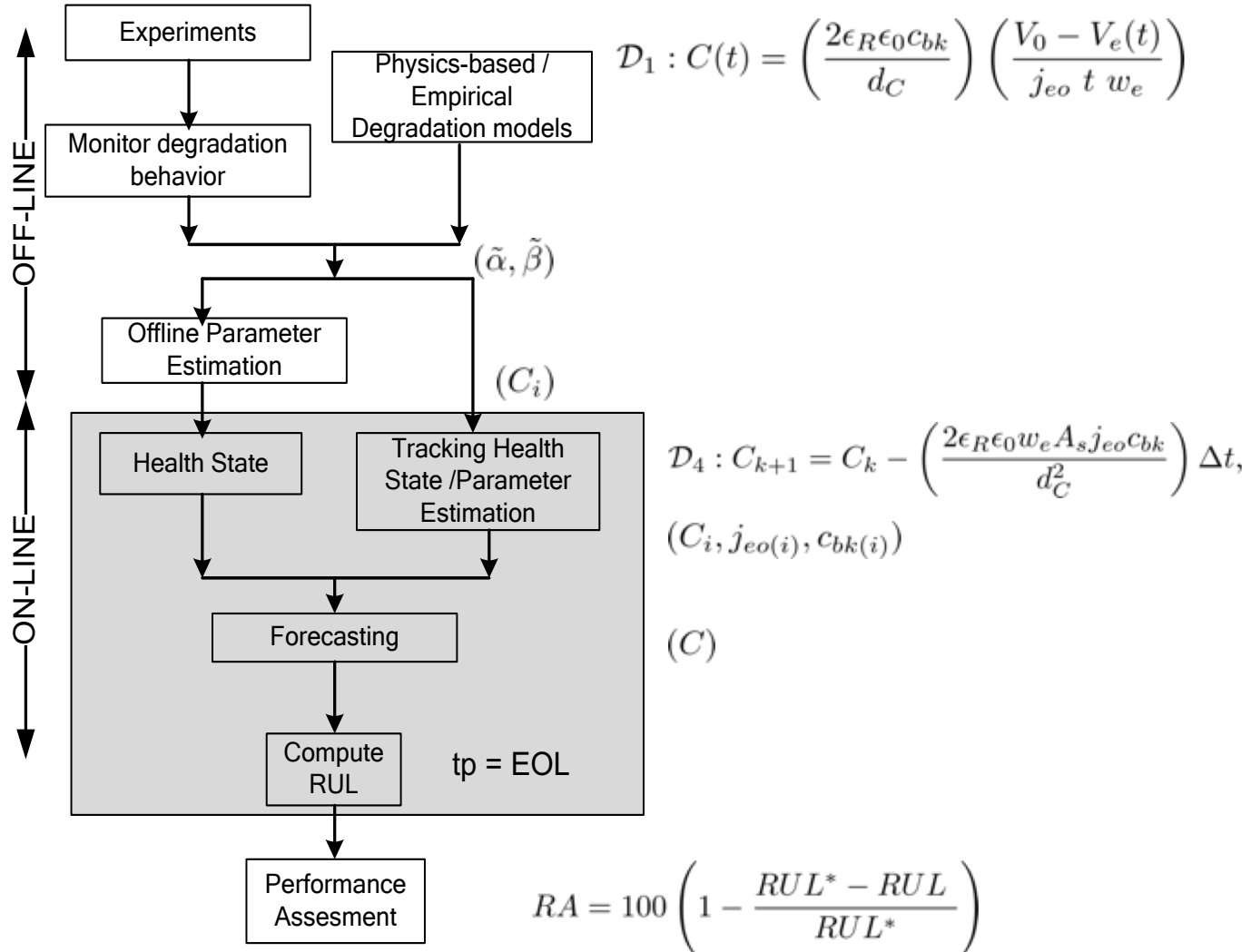
where:

$\rho_E$  : electrolyte resistivity,

$P_E$  : correlation factor related to electrolyte spacer porosity and average liquid pathway,

$e_{bk}(t)$  : resistance dependence oxide breakdown factor

# Process Flow



# Unscented Kalman Filter for State Estimation

$$\mathcal{D}_4 : C_{k+1} = C_k - \left( \frac{2\epsilon_R \epsilon_0 w_e A_s j_{eo} c_{bk}}{d_C^2} \right) \Delta t$$

- Derived physics-based degradation model
- The following system structure is implemented for state estimation

$$\begin{aligned}\mathbf{x}_k &= A_k \mathbf{x}_{k-1} + B_k u + \mathbf{v}, \\ \mathbf{y}_k &= H_k \mathbf{x}_k + \mathbf{w}.\end{aligned}$$

$$\begin{aligned}A &= 1, \\ B &= -\frac{(2\epsilon_R \epsilon_0 w_e A_s j_{eo} c_{bk})}{d_C^2} \Delta t, \\ H &= 1, \\ u &= j_{eo}, c_{bk}.\end{aligned}$$

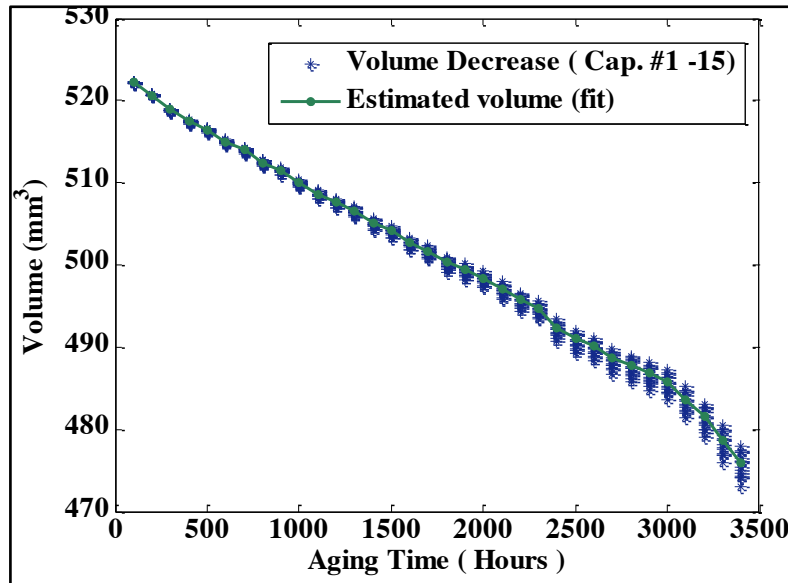
- The state variable ( $\mathbf{x}_k$ ) is the current health state at aging time ( $t_p$ )

Process noise was estimated from the model regression for the empirical model  
Measurement noise was estimated from the EIS measurements

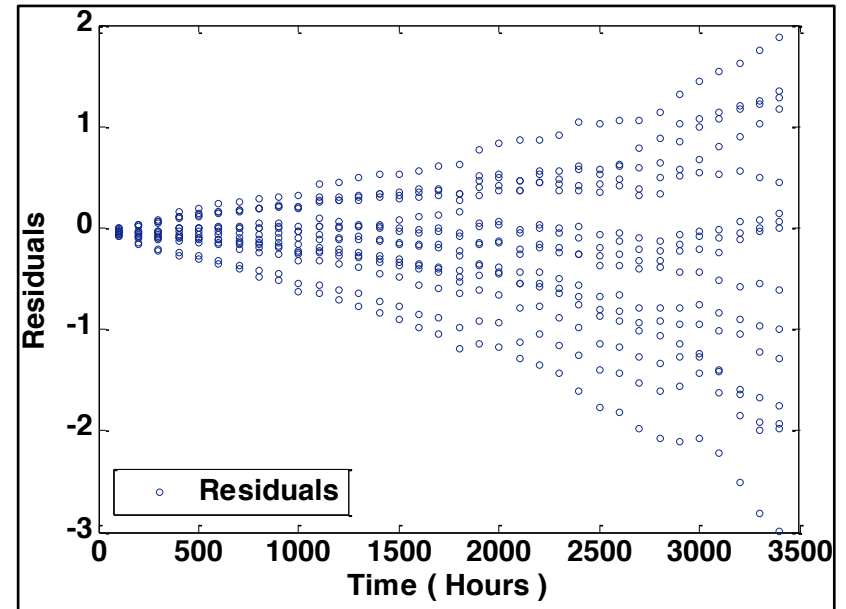
# Electrolyte Volume Estimation for TOS Experiment

| Parameter                | $\hat{X}$               | $\tilde{X}$             | S.D                     | C.I                                              |
|--------------------------|-------------------------|-------------------------|-------------------------|--------------------------------------------------|
| $\hat{\theta}_1(mm^3)$   | 523.6112                | 523.6113                | 0.0026                  | [523.6098, 523.6127]                             |
| $\hat{\theta}_2(mm^2/t)$ | 0.0161                  | 0.0161                  | $1.8748 \times 10^{-5}$ | [0.01614, 0.01611]                               |
| $\hat{\theta}_3(mm/t^2)$ | $3.8077 \times 10^{-7}$ | $3.8072 \times 10^{-7}$ | $6.9373 \times 10^{-9}$ | $[0.3769 \times 10^{-6}, 0.3846 \times 10^{-6}]$ |
| RMSE                     | 26.2232                 | 26.2277                 | 0.0483                  | [26.1965, 26.2500]                               |
| RMSPE                    | 0.8589                  | 0.8591                  | 0.0016                  | [0.8580, 0.8598]                                 |

Summary for Linear Regression Electrolyte Degradation Model



Volume estimation for all capacitors



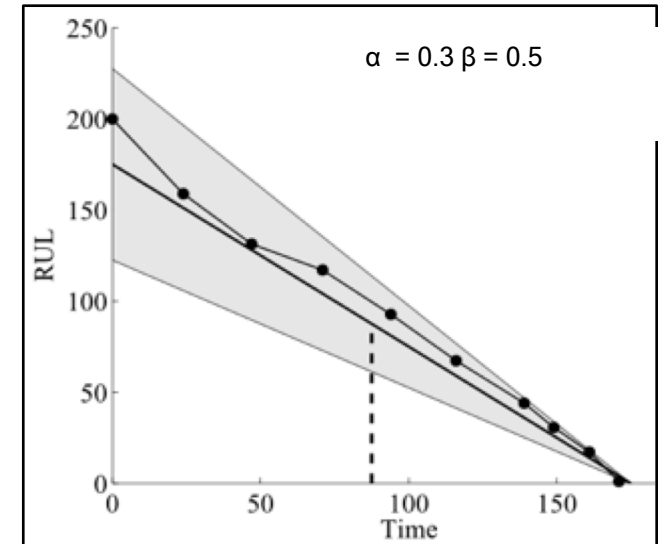
Volume estimation Error

# EOS Experiment RA Results – Discussion

Capacitance - Over RA summary for model  $\mathcal{E}_1$

| Aging Time | $\overline{RA}_a$ |
|------------|-------------------|
| 24         | 95.5              |
| 47         | 96.7              |
| 71         | 91.9              |
| 94         | 90                |
| 116        | 96.2              |
| 139        | 81.1              |
| 149        | 86.6              |
| 161        | 87.3              |
| 171        | 30.7              |

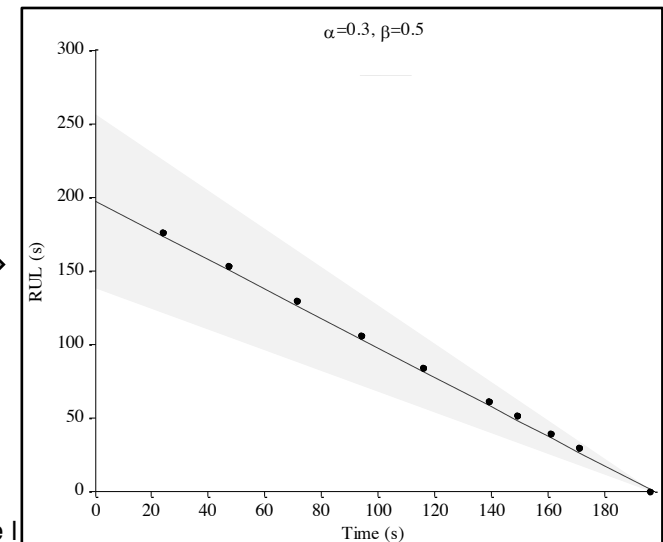
$$\mathcal{E}_1 : C_l(t) = e^{\alpha t} + \beta,$$



Capacitance - Over RA summary for model  $\mathcal{D}_4$

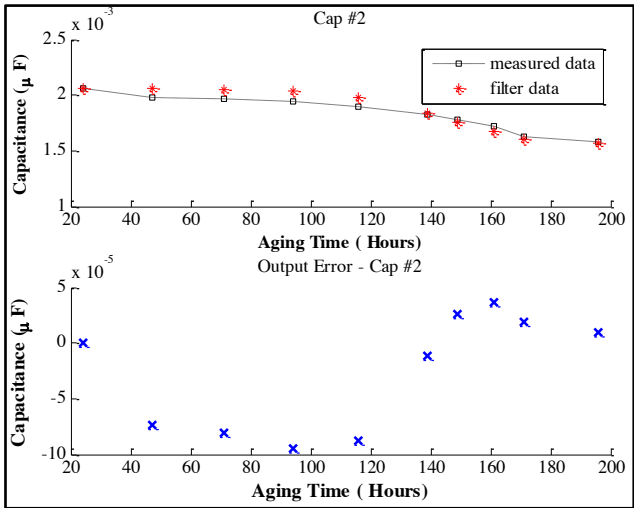
| Aging Time | $\overline{RA}_a$ |
|------------|-------------------|
| 24         | 98.06             |
| 47         | 97.76             |
| 71         | 97.34             |
| 94         | 96.73             |
| 116        | 95.84             |
| 139        | 94.16             |
| 149        | 92.92             |
| 161        | 90.49             |
| 171        | 86.67             |

$$\mathcal{D}_4 : C_{k+1} = C_k - \left( \frac{2\epsilon_R \epsilon_0 w_e A_s j_{eo} c_{bk}}{d_C^2} \right) \Delta t$$

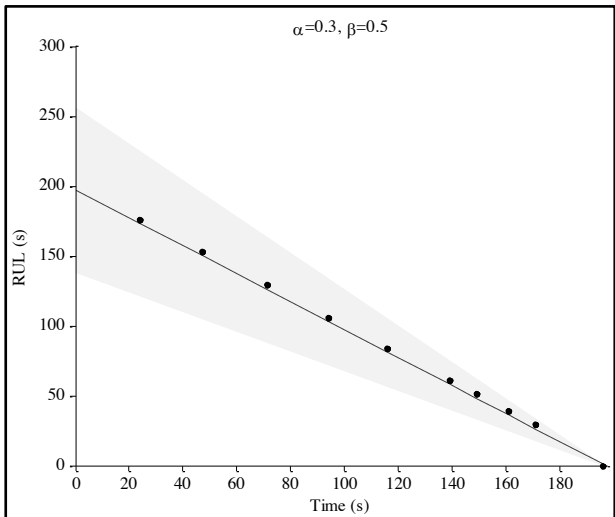


# RUL and Validation – EOS -Experiment – Capacitance Degradation Model

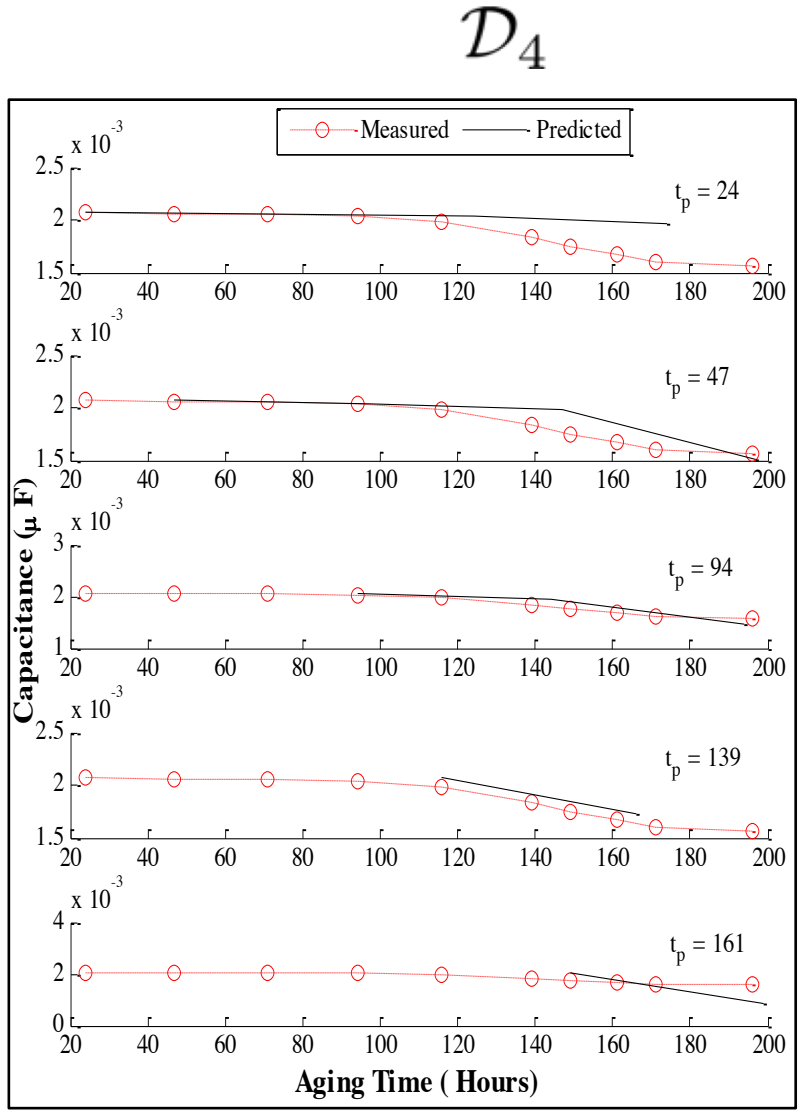
Tracking



Alpha Lambda



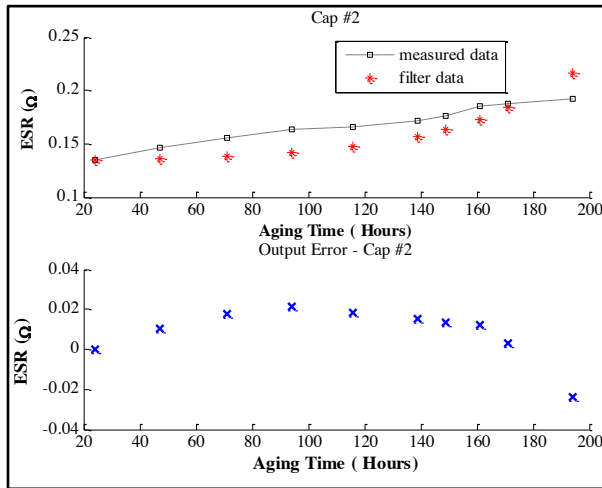
Predictions at different aging time



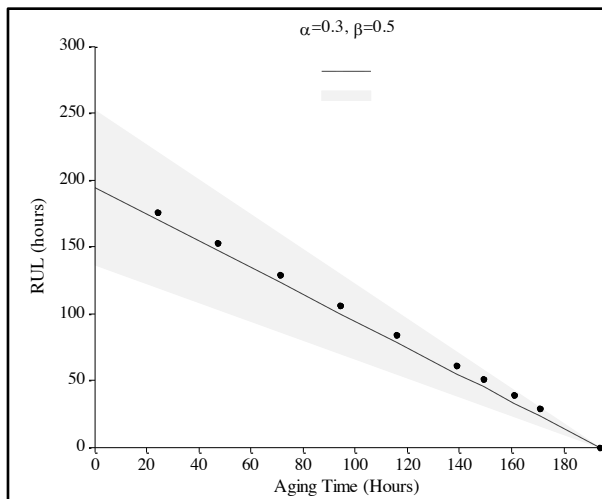
# RUL and Validation – EOS -Experiment – ESR Degradation Model

$\mathcal{D}_5$

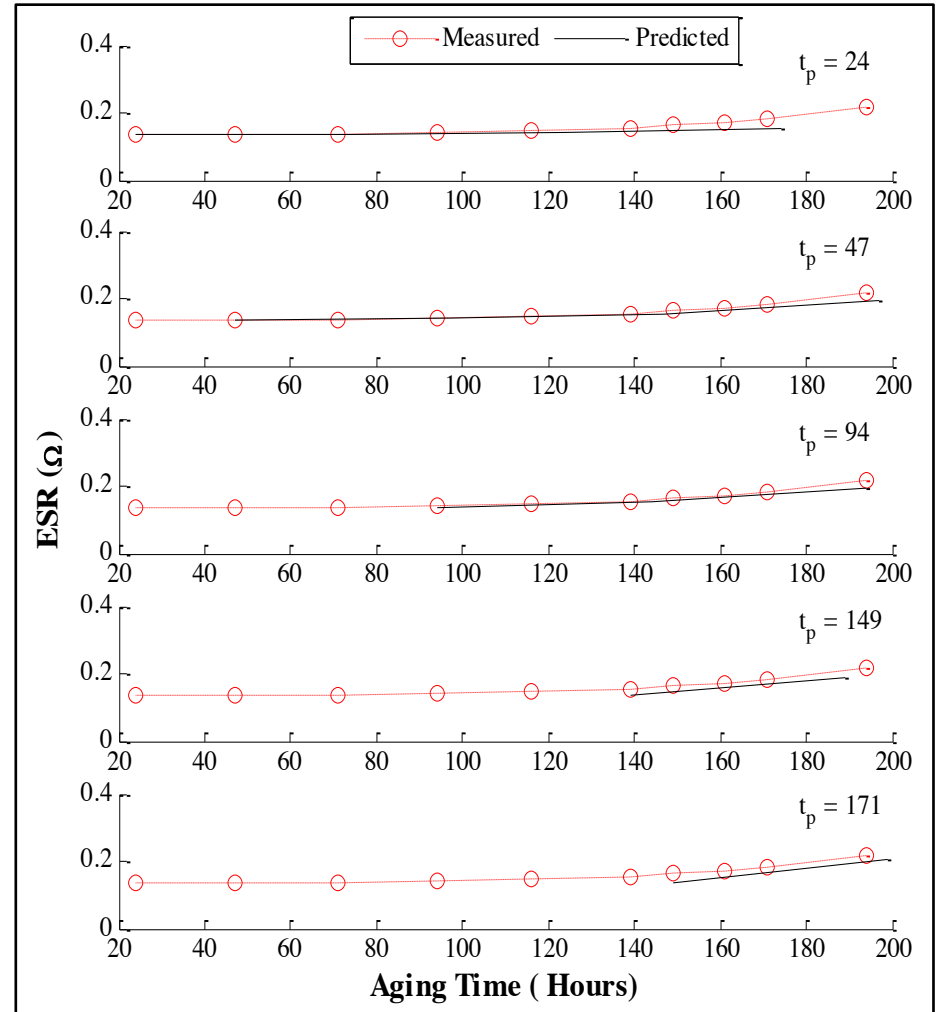
Tracking



Alpha Lambda

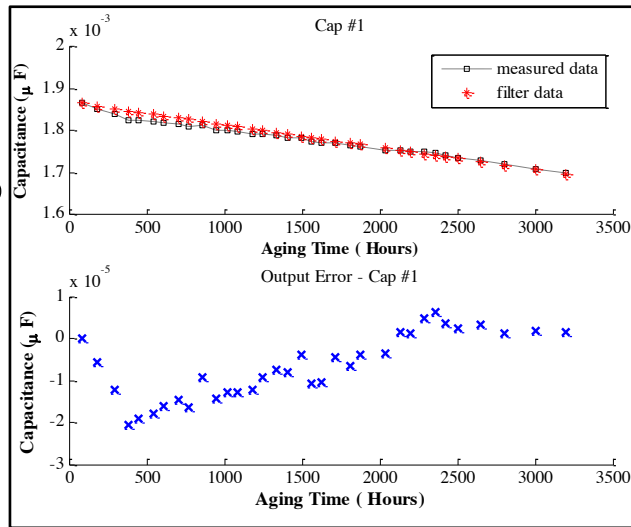


Predictions at different aging time

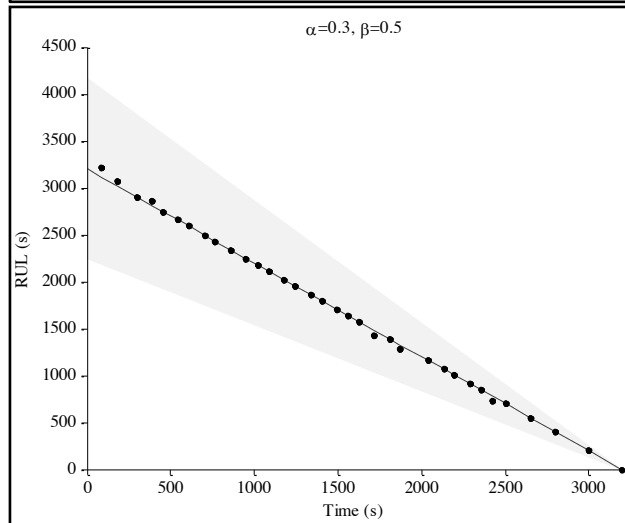


# RUL and Validation – TOS -Experiment - Capacitance

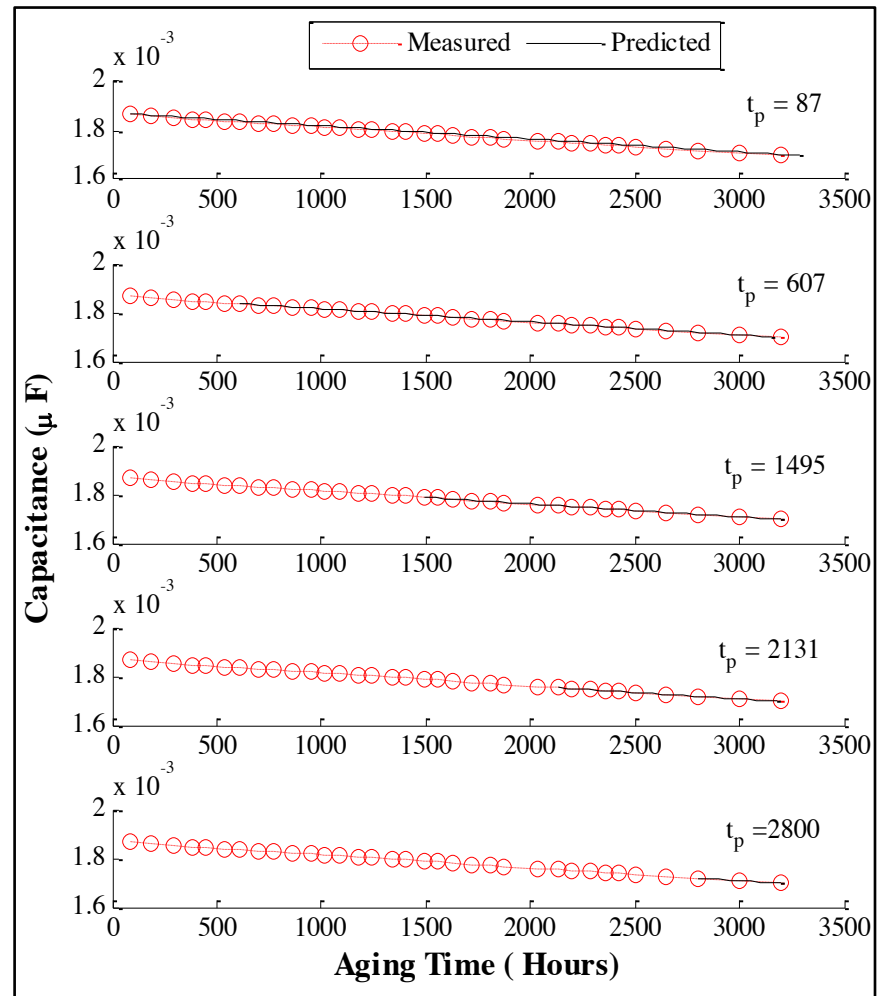
Tracking



Alpha Lambda



Predictions at different aging time



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End of Case Study III:

# **DISCUSSION AND QUESTIONS?**

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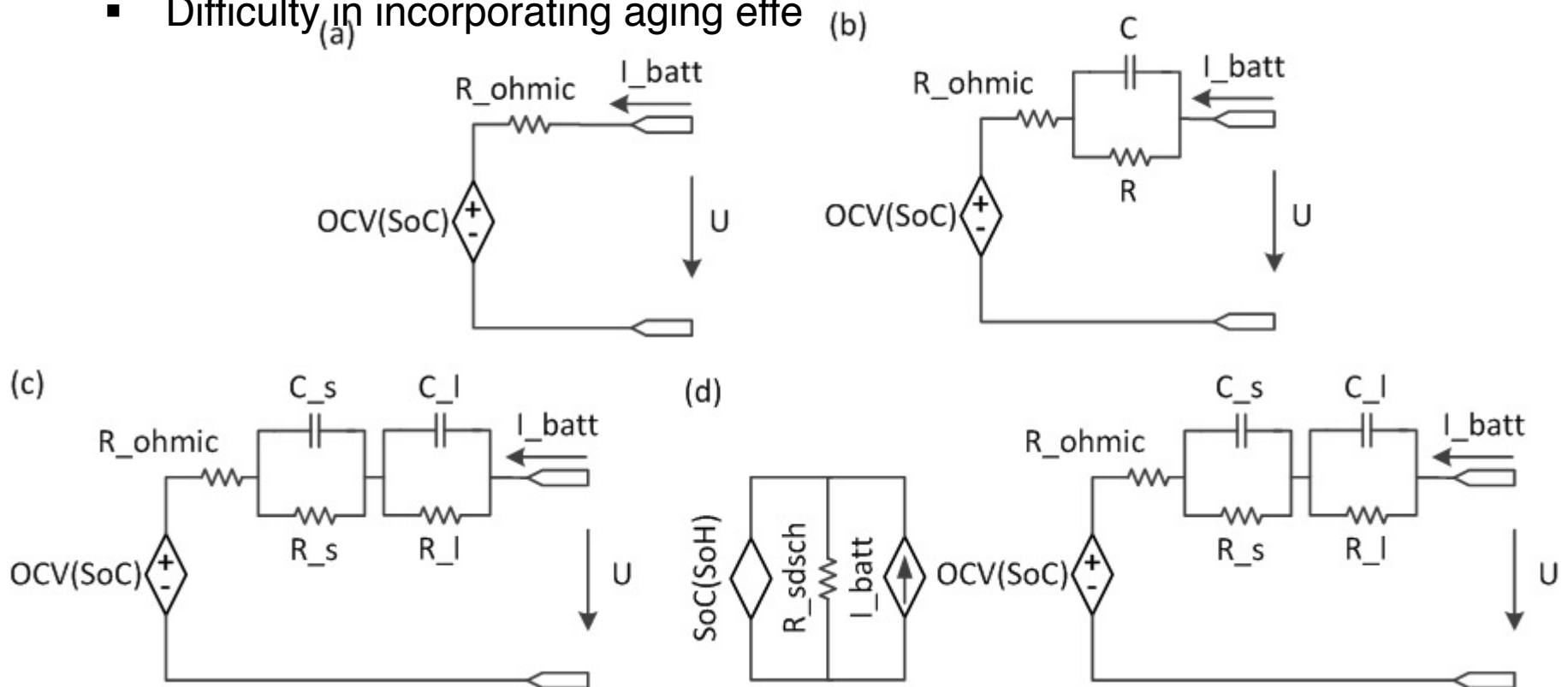
# **CASE STUDY IV: PROGNOSTICS OF LI-ION BATTERIES**

**DEGRADATION MODELING EXAMPLE**

# Battery Modeling

## – Equivalent Circuit Empirical Models

- Most common approach
- Various model complexities used
- Difficulty in incorporating aging effects



# Battery Model – Tuned using Lab Data

- An equivalent circuit battery model is used to represent the battery terminal voltage as a function of current and the charge stored in 3 capacitive elements

$$x = [q_b \ q_{cp} \ q_{Cs}]^T$$

$$\dot{x} = \begin{bmatrix} 0 & 0 & 0 \\ 0 & -\frac{1}{R_{cp}C_{cp}} & 0 \\ 0 & 0 & -\frac{1}{R_sC_s} \end{bmatrix} x + \begin{bmatrix} -1 \\ 1 \\ 1 \end{bmatrix} i + \xi$$

$$y = V = \left[ \frac{1}{C_b} - \frac{1}{C_{cp}} - \frac{1}{C_s} \right] \cdot x$$

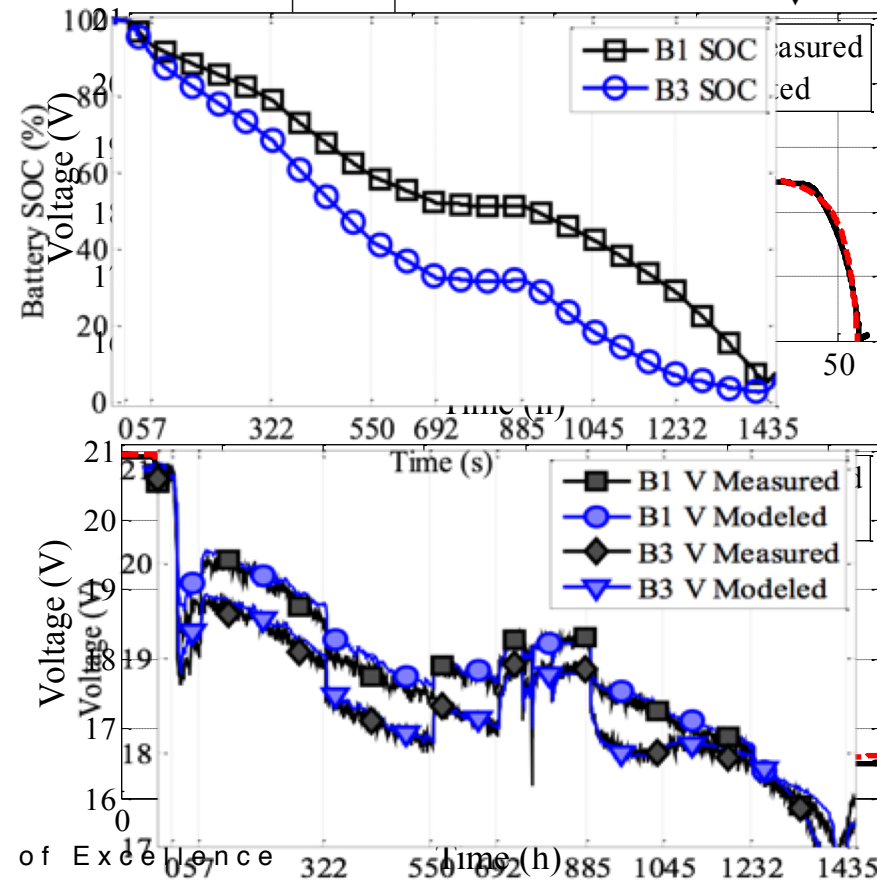
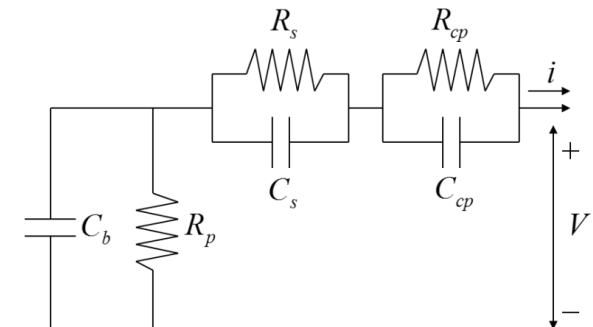
- Two laboratory loading experiments are used to fit the following parameterization coefficients

$$SOC = 1 - \frac{q_{max} - q_b}{C_{max}}$$

$$C_b = C_{Cb0} + C_{Cb1} \cdot SOC + C_{Cb2} \cdot SOC^2 + C_{Cb3} \cdot SOC^3$$

$$C_{cp} = C_{cp0} + C_{cp1} \cdot \exp(C_{cp2}(1 - SOC))$$

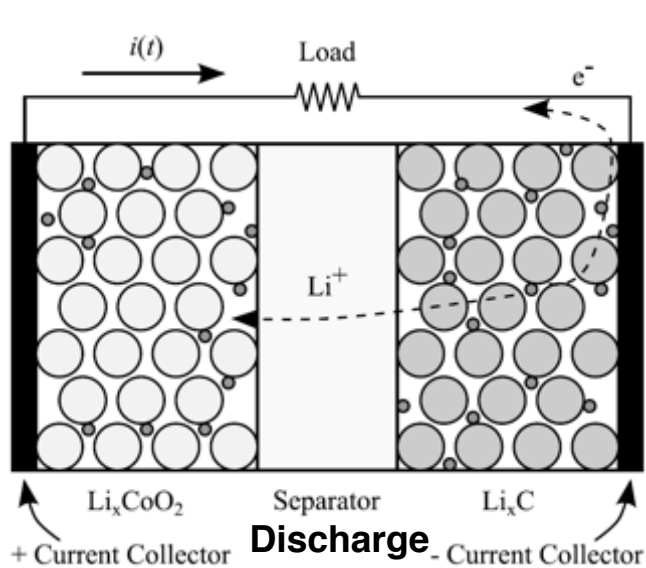
$$R_{cp} = R_{cp0} + R_{cp1} \cdot \exp(R_{cp2}(1 - SOC))$$



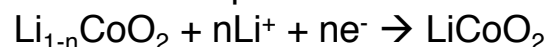
# Battery Modeling

## – Electrochemical Models vs. Empirical Models

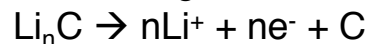
- Battery physics models enable more direct representation of age-related changes in battery dynamics than empirical models
- Typically have a higher computational cost and more unknown parameters



Reduction at pos. electrode:



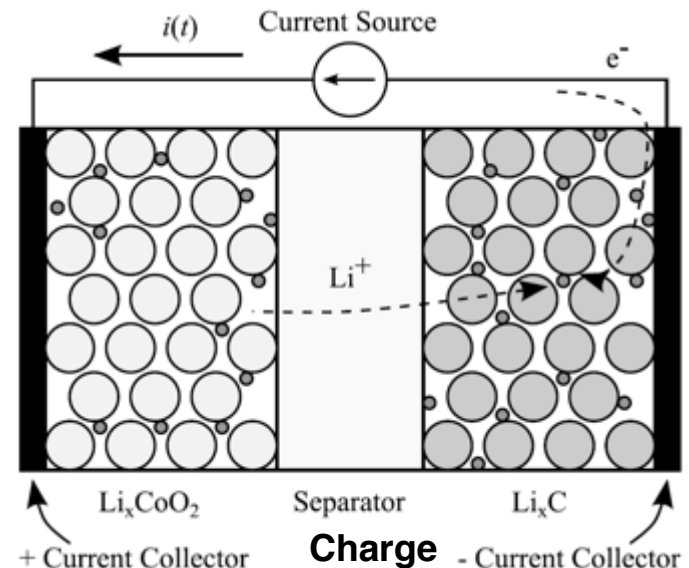
Oxidation at neg. electrode:



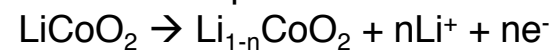
Current flows + to -

Electrons flow - to +

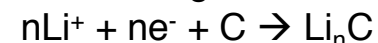
Lithium ions flow - to +



Oxidation at pos. electrode:



Reduction at neg. electrode:



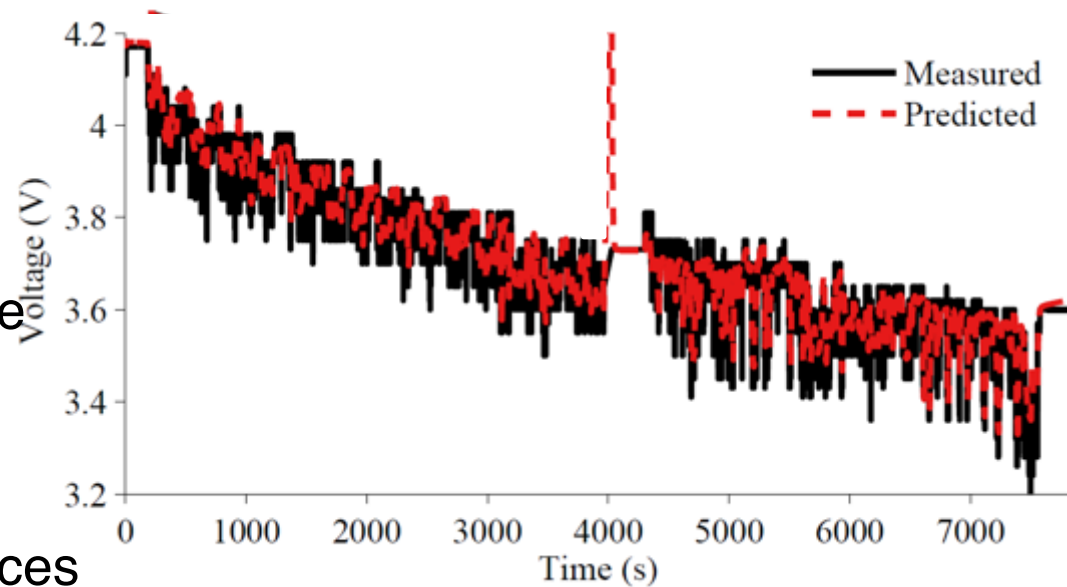
Current flows - to +

Electrons flow + to -

Lithium ions flow + to -

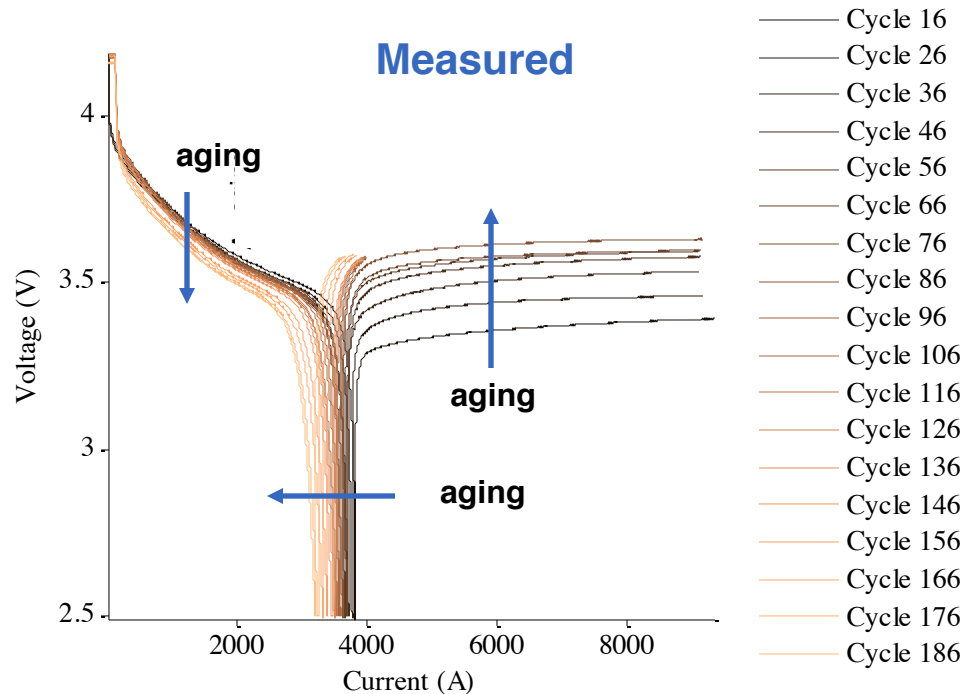
# Electrochemical Li-ion Model

- Lumped-parameter, ordinary differential equations
- Capture voltage contributions from different sources
  - Equilibrium potential → Nernst equation with Redlich-Kister expansion
  - Concentration overpotential → split electrodes into surface and bulk control volumes
  - Surface overpotential → Butler-Volmer equation applied at surface layers
  - Ohmic overpotential → Constant lumped resistance accounting for current collector resistances, electrolyte resistance, solid-phase ohmic resistances

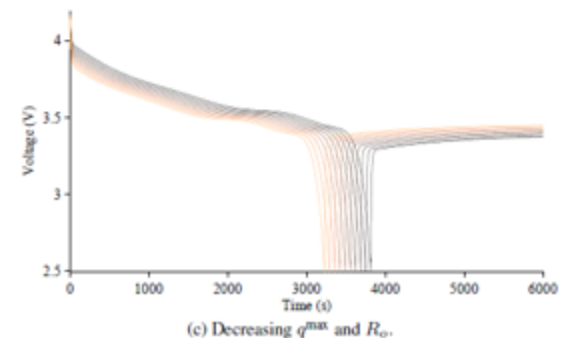
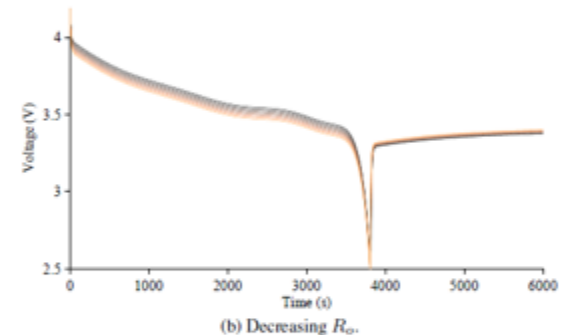
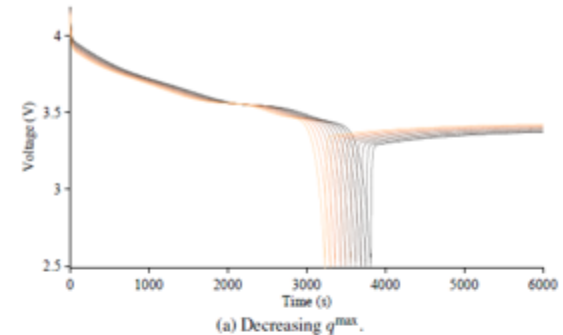


# Battery Aging

- Contributions from both decrease in mobile Li ions (lost due to side reactions related to aging) and increase in internal resistance
  - Modeled with decrease in " $q^{max}$ " parameter, used to compute mole fraction
  - Modeled with increase in " $R_o$ " parameter capturing lumped resistances

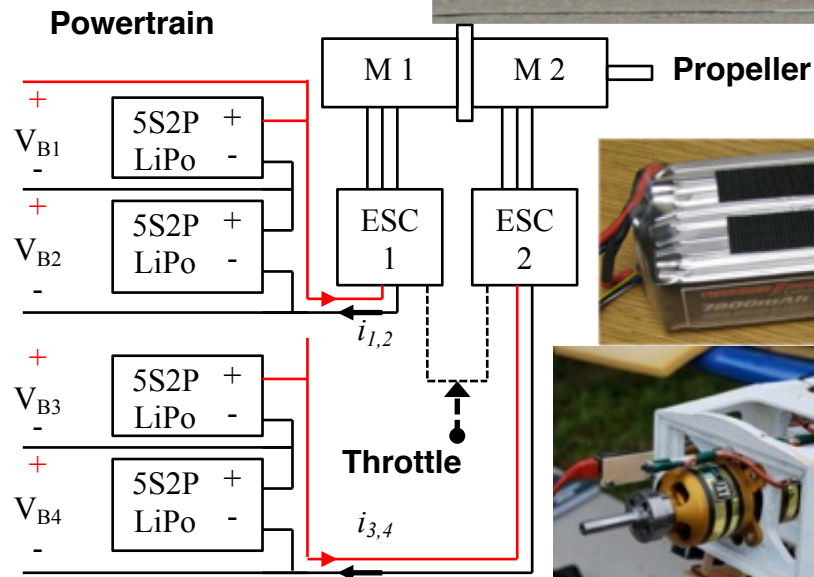


## Simulated



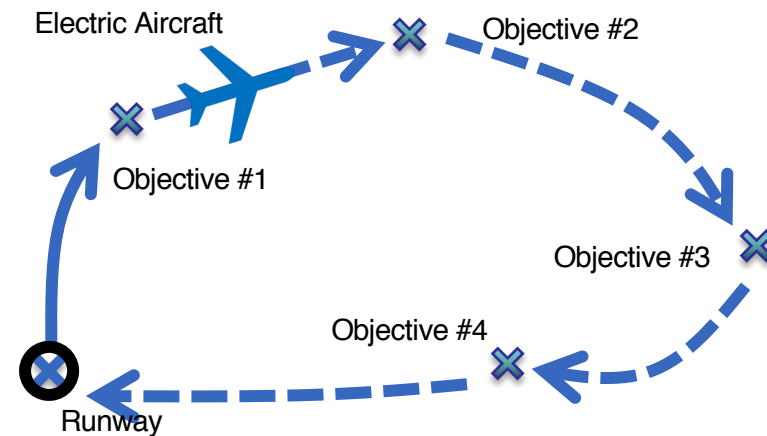
# Edge 540-T

- Subscale electric aircraft operated at NASA Langley Research Center
- Powered by four sets of Li-polymer batteries
- Estimate SOC online and provide EOD and remaining flight time predictions for ground-based pilots



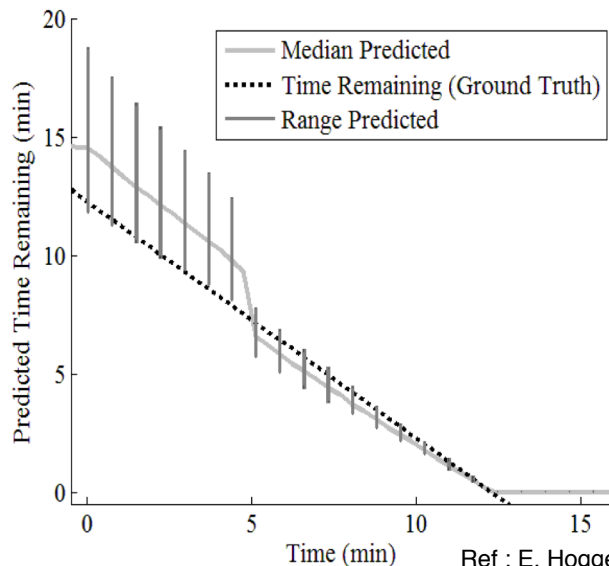
# Edge UAV Use Case

- Piloted and autonomous missions, visiting waypoints
- Require 2-minute warning for EOD so pilot/autopilot has sufficient time to land safely
  - This answer depends on battery age
  - Need to track both current level of charge and current battery age
  - Based on current battery state, current battery age, and expected future usage, can predict EOD and correctly issue 2-minute warning

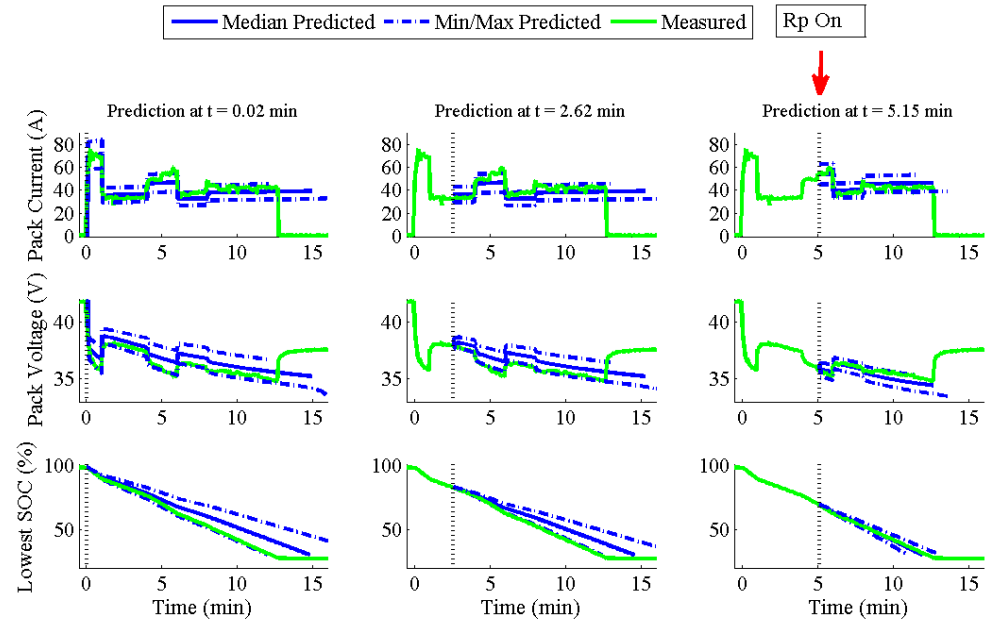


# Predication over Flight Plan

- Measured and predicted battery current, voltage and SOC different time steps
- The min, max and median predictions are plotted from each sample time until the predicated SOC reaches 30%



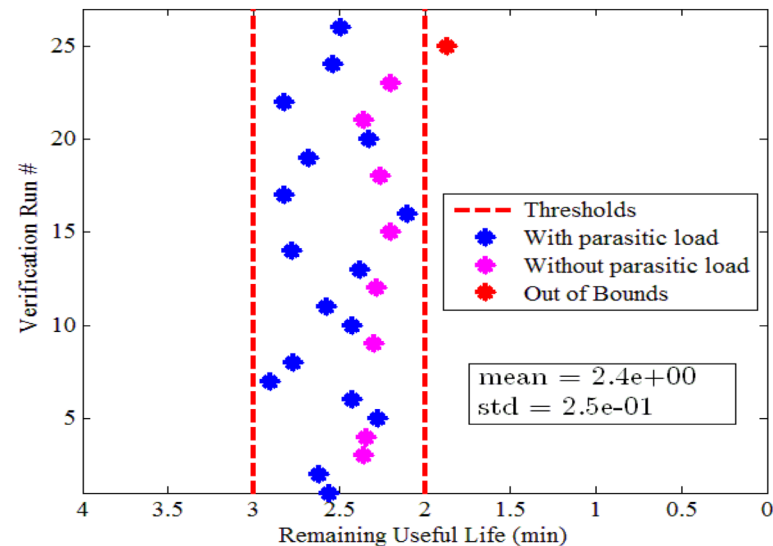
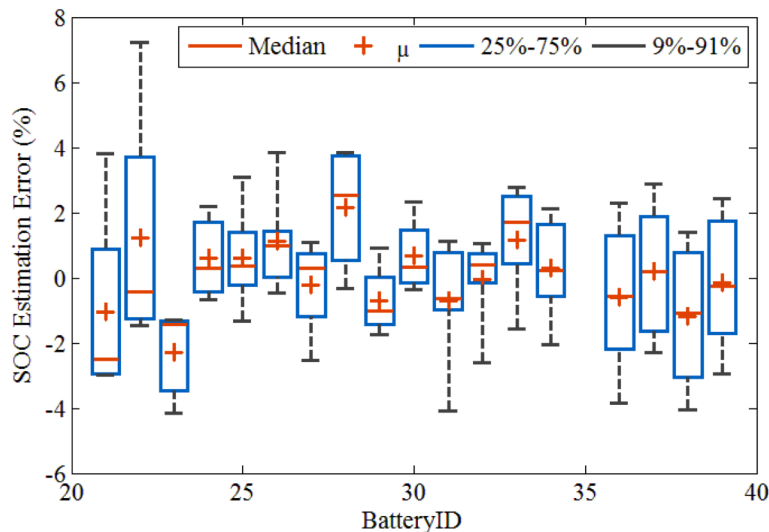
Ref : E. Hogge et al, "Verification of a Remaining Flying Time Prediction System for Small Electric Aircraft", PHM 2015



- Predictions for remaining flight time for entire flight plan
- Overestimate till parasitic load is injected
- Once the parasitic load is detected the remaining flying time prediction shifts down.

# Performance Requirements

- Accuracy requirements for the two minute warning were specified as:
  - *The prognostic algorithm shall raise an alarm no later than two minutes before the lowest battery SOC estimate falls below 30% for at least 90% of verification trial runs.*
  - *The prognostic algorithm shall raise an alarm no earlier than three minutes before the lowest battery SOC estimate falls below 30% for at least 90% of verification trial runs.*
  - *Verification trial statistics must be computed using at least 20 experimental runs*



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End of Case Study IV:

# **DISCUSSION AND QUESTIONS?**

# Data Sets Available for Download

- <https://ti.arc.nasa.gov/tech/dash/pcoe/prognostic-data-repository/>

## Randomized Battery Usage Data Set

### Publications using this data set

|                             |                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              |
|-----------------------------|----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| <b>Description</b>          | Batteries are continuously cycled with randomly generated current profiles. Reference charging and discharging cycles are also performed after a fixed interval of randomized usage in order to provide reference benchmarks for battery state of health.                                                                                                                                                                                                                                                                                                                                    |
| <b>Format</b>               |                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              |
| <b>Datasets</b>             | + <a href="#">Download Randomized Battery Usage Data Set 1</a> (1285 downloads)<br>+ <a href="#">Download Randomized Battery Usage Data Set 2</a> (936 downloads)<br>+ <a href="#">Download Randomized Battery Usage Data Set 3</a> (906 downloads)<br>+ <a href="#">Download Randomized Battery Usage Data Set 4</a> (4217 downloads)<br>+ <a href="#">Download Randomized Battery Usage Data Set 5</a> (825 downloads)<br>+ <a href="#">Download Randomized Battery Usage Data Set 6</a> (890 downloads)<br>+ <a href="#">Download Randomized Battery Usage Data Set 7</a> (857 downloads) |
| <b>Dataset Citation</b>     | B. Bole, C. Kulkarni, and M. Daigle "Randomized Battery Usage Data Set", NASA Ames Prognostics Data Repository ( <a href="http://ti.arc.nasa.gov/project/prognostic-data-repository">http://ti.arc.nasa.gov/project/prognostic-data-repository</a> ), NASA Ames Research Center, Moffett Field, CA                                                                                                                                                                                                                                                                                           |
| <b>Publication Citation</b> | B. Bole, C. Kulkarni, and M. Daigle, 'Adaptation of an Electrochemistry-based Li-Ion Battery Model to Account for Deterioration Observed Under Randomized Use', Annual Conference of the Prognostics and Health Management Society, 2014                                                                                                                                                                                                                                                                                                                                                     |

## HIRF Battery Data Set

### Publications using this data set

|                             |                                                                                                                                                                                                                                                                                                                                                                                                                                  |
|-----------------------------|----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| <b>Description</b>          | Battery Data collected from the Experiments on the Edge 540 Aircraft in HIRF Chamber. Reference document can be downloaded <a href="#">here</a>                                                                                                                                                                                                                                                                                  |
| <b>Format</b>               | The set is in .mat format and has been zipped.                                                                                                                                                                                                                                                                                                                                                                                   |
| <b>Datasets</b>             | + <a href="#">Download HIRF Battery Data Set 1</a> (184 downloads)<br>+ <a href="#">Download HIRF Battery Data Set 2</a> (127 downloads)<br>+ <a href="#">Download HIRF Battery Data Set 3</a> (131 downloads)<br>+ <a href="#">Download HIRF Battery Data Set 4</a> (125 downloads)<br>+ <a href="#">Download HIRF Battery Data Set 5</a> (149 downloads)<br>+ <a href="#">Download HIRF Battery Data Set 6</a> (135 downloads) |
| <b>Dataset Citation</b>     | C. Kulkarni, E. Hogge, C. Quach and K. Goebel "HIRF Battery Data Set", NASA Ames Prognostics Data Repository ( <a href="http://ti.arc.nasa.gov/project/prognostic-data-repository">http://ti.arc.nasa.gov/project/prognostic-data-repository</a> ), NASA Ames Research Center, Moffett Field, CA                                                                                                                                 |
| <b>Publication Citation</b> | Verification of a Remaining Flying Time Prediction System for Small Electric Aircraft. Edward F. Hogge, Brian M. Bole, Sixto L. Vazquez, Jose R., Annual Conference of the Prognostics and Health Management, PHM 2015                                                                                                                                                                                                           |

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# **CLOSING REMARKS**

# Remarks (1/2)

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- Electrical and Electronics PHM Maturity - scientific and engineering challenges
- Research approach challenges
  - How to balance lack of knowledge of the system vs own expertise on particular PHM tools
  - Data-driven or model-based?
    - Data is always needed but more important, information about degradation/aging processes is key
    - Experiments and field data

## Remarks (2/2)

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- Aging systems as a research tool
  - Value in terms of exploration of precursors of failure and their measurements is evident
  - Still an open question on how degradation models and algorithms are translated to the real usage timescale
- In the use of physics
  - It should be embraced
- Validate models and algorithms with data from lab experiments and fielded systems
- A success in developing PHM methodologies in an real usage application will require the right team

# Publications (1/3)

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- [1] C. Kulkarni, G. Biswas, X. Koutsoukos, J. Celaya, and K. Goebel, “Integrated diagnostic/prognostic experimental setup for capacitor degradation and health monitoring,” in *IEEE AUTOTESTCON, 2010*, (Big Sky, MT), pp. 1–7, 2010.
- [2] C. Kulkarni, G. Biswas, X. Koutsoukos, J. Celaya, and K. Goebel, “Aging methodologies and prognostic health management for electrolytic capacitors,” in *Annual Conference of the Prognostics and Health Management Society 2010*, (Portland, OR), 2010.
- [3] C. Kulkarni, G. Biswas, X. Koutsoukos, J. Celaya, and K. Goebel, “Diagnostic/prognostic experiments for capacitor degradation and health monitoring in DC-DC converters,” in *ASME 2010 Conference on Smart Materials, Adaptive Structures and Intelligent Systems*, (Philadelphia, PA), 2010.
- [4] J. Celaya, C. Kulkarni, G. Biswas, S. Saha and K. Goebel, “A Model-based Prognostics Methodology for Electrolytic Capacitors Based on Electrical Overstress Accelerated Aging”, Proceedings of Annual Conference of the Prognostics and Health Management Society, September 25-29, 2011, Montreal, Canada.
- [5] C. Kulkarni, J. Celaya, G. Biswas, and K. Goebel, “Physics Based Degradation Models for Capacitor Prognostics under Thermal Overstress Conditions”, *International Journal of Prognostics and Health Management*, 2013 Vol 4 (1) 005.
- [6] G. Sonnenfeld, K. Goebel, and J. R. Celaya, “An agile accelerated aging, characterization and scenario simulation system for gate controlled power transistors,” in *IEEE AUTOTESTCON 2008*, pp. 208–215, 2008.
- [7] J. R. Celaya, N. Patil, S. Saha, P. Wyscoki, and K. Goebel, “Towards accelerated aging methodologies and health management of power MOSFETs,” in *Annual Conference of the Prognostics and Health Management Society, 2009*, (San Diego, CA), 2009.
- [8] J. R. Celaya, S. Saha, P. Wyscoki, and K. F. Goebel, “Effects of lightning injection on power-MOSFETs,” in *Annual Conference of the Prognostics and Health Management Society, 2009*, (San Diego, CA), 2009.

# Publications (2/3)

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- [9] N. Patil, J. Celaya, D. Das, K. Goebel, and M. Pecht, "Precursor parameter identification for insulated gate bipolar transistor (IGBT) prognostics," *IEEE Transactions on Reliability*, vol. 58, no. 2, pp. 271–276, 2009.
- [10] B. Saha, J. R. Celaya, P. F. Wysocki, and K. F. Goebel, "Towards prognostics for electronics components," in *IEEE Aerospace conference 2009*, (Big Sky, MT), pp. 1–7, 2009.
- [11] P. Wysocki, V. Vashchenko, J. Celaya, S. Saha, and K. Goebel, "Effect of electrostatic discharge on electrical characteristics of discrete electronic components," in *Annual Conference of the Prognostics and Health Management Society, 2009*, (San Diego, CA), 2009.
- [12] J. Celaya, A. Saxena, P. Wysocki, S. Saha, and K. Goebel, "Towards prognostics of power MOSFETs: Accelerated aging and precursors of failure," in *Annual Conference of the Prognostics and Health Management Society 2010*, (Portland, OR), 2010.
- [13] J. J. Ely, T. X. Nguyen, G. N. Szatkowski, S. V. Koppen, J. J. Mielnik, R. K. Vaughan, P. F. Wysocki, J. R. Celaya, and S. Saha, "Lightning pin injection testing on MOSFETS," *NASA Technical Memorandum*, vol. TM-2009-215794, 2009.
- [14] J. R. Celaya, P. Wysocki, V. Vashchenko, S. Saha, and K. Goebel, "Accelerated aging system for prognostics of power semiconductor devices," in *IEEE AUTOTESTCON, 2010*, (Orlando, FL), pp. 1–6, 2010.
- [15] A. E. Ginart, I. N. Ali, J. R. Celaya, P. W. Kalgren, D. P. S, and M. J. Roemer, "Modeling SiO<sub>2</sub> ion impurities aging in insulated gate power devices under temperature and voltage stress," in *Annual Conference of the Prognostics and Health Management Society 2010*, (Portland, OR), 2010.
- [16] S. Saha, J. Celaya, B. Saha, P. Wysocki, and K. Goebel, "Towards modeling the effects of lightning injection on power MOSFETs," in *Annual Conference of the Prognostics and Health Management Society 2010*, (Portland, OR), 2010.
- [17] S. P. Bharadwaj, A. E. Ginart, I. N. Ali, P. W. Kalgren, J. Celaya, and S. Poll, "Solar cells aging estimation based on impedance characterization," in *IEEE Aerospace Conference 2011*, (Big Sky, MT), 2011.

# Publications (3/3)

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- [18] J. R. Celaya, C. Kulkarni, G. Biswas, and K. Goebel, "Towards prognostics of electrolytic capacitors," in *AIAA Infotech@Aerospace 2011*, (St. Louis, MO), 2011.
- [19] J. R. Celaya, A. Saxena, V. Vashchenko, S. Saha, and K. Goebel, "Prognostics of power MOSFET," in *23rd International Symposium on Power Semiconductor Devices and ICs*, (San Diego, CA), 2011.
- [20] J. J. Ely, T. X. Nguyen, G. N. Szatkowski, S. V. Koppen, J. J. Mielnik, R. K. Vaughan, P. F. Wysocki, J. R. Celaya, and S. Saha, "Lightning pin injection test: MOSFETS in "ON" state," *NASA Technical Memorandum*, 2011.
- [21] Raj Bharadwaj, K. Kim, C. Kulkarni, G. Biswas, "Model-Based Avionics Systems Fault Simulation and Detection ", American Institute of Aeronautics and Astronautics, Infotech@Aerospace 2010, April 2010, Atlanta, GA.
- [22] C. Kulkarni, G. Biswas, J. Celaya, and K. Goebel, "Prognostics techniques for capacitor degradation and health monitoring," in *The Maintenance and Reliability Conference (MARCON)*, (Knoxville TN), 2011.
- [23] E. Hogge et al, "Verification of a Remaining Flying Time Prediction System for Small Electric Aircraft", PHM 2015
- [24] E. Hogge, , C. Kulkarni, S. Vazquez, K. Smalling, T. Strom, B. Hill, C. Quach "Flight Tests of a Remaining Flying Time Prediction System for Small Electric Aircraft', Proceedings of the Annual Conference of the Prognostics and Health Management Society 2017 (to be published)
- [25] G. Gorospe, C. Kulkarni, E. Hogge, A. Hsu, N. Ownby, "A Study of the Degradation of Electronic Speed Controllers for Brushless DC Motors", Proceedings Asia Pacific Conference of the Prognostics and Health Management Society 2017, Jeju, South Korea (to be published)

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# **THANK YOU!**

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