

# ADVANCED CRYOGENIC INSULATION SYSTEMS

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# INTRODUCTION

- What is the “best insulation system” for a given design situation?
  - **System design, engineering analysis, and thermal testing** (materials) go hand-in-hand
- Conducting analysis and calculations according to **standard methods**:
  - Essential for fair comparison of different materials and accurate applications of results
- Testing to measure of the total heat transmission ( $Q$ ) into a cryogenic system
  - **Relevant conditions** typically include a large temperature difference ( $\Delta T$ ) and a controlled, steady-state test environment or vacuum level
- Examples of cryogenic storage tanks and transfer piping are analyzed:
  - Determine the **relative importance of both insulation and structural materials** for achieving designs of highest energy efficiency

# CALCULATIONS OF HEAT TRANSMISSION

- Follow the guidance given in *ASTM C1774 Standard Guide for Thermal Performance Testing of Cryogenic Insulation Systems*
  - Large  $\Delta T$ : Warm Boundary Temperature (WBT) - Cold Boundary Temperature (CBT)
- The **heat flow rate\*** ( $Q$ ) in J/s and the **effective thermal conductivity** ( $k_e$ ) in mW/m-K are calculated as follows:
  - Where  $\dot{m}$  is the mass flow rate in g/s,  $h_{fg}$  is the heat of vaporization in J/g,  $A_e$  is the effective area for heat transmission in  $m^2$ ,  $\Delta T$  is the temperature difference in K, and  $\Delta x$  is the thickness in m
  - The system thermal conductivity ( $k_s$ ) is defined as the thermal conductivity value through the **total system** including all ancillary elements such as packaging, supports, seams, joints, piping penetrations, feedthroughs, structures, etc.

$$Q = \dot{m}h_{fg} = k_e A_e \frac{\Delta T}{\Delta x} \qquad k_e = \frac{Q}{A_e} \frac{\Delta x}{\Delta T}$$

\*Heat leak or heat load or heat transmission

# CALCULATIONS OF HEAT TRANSMISSION

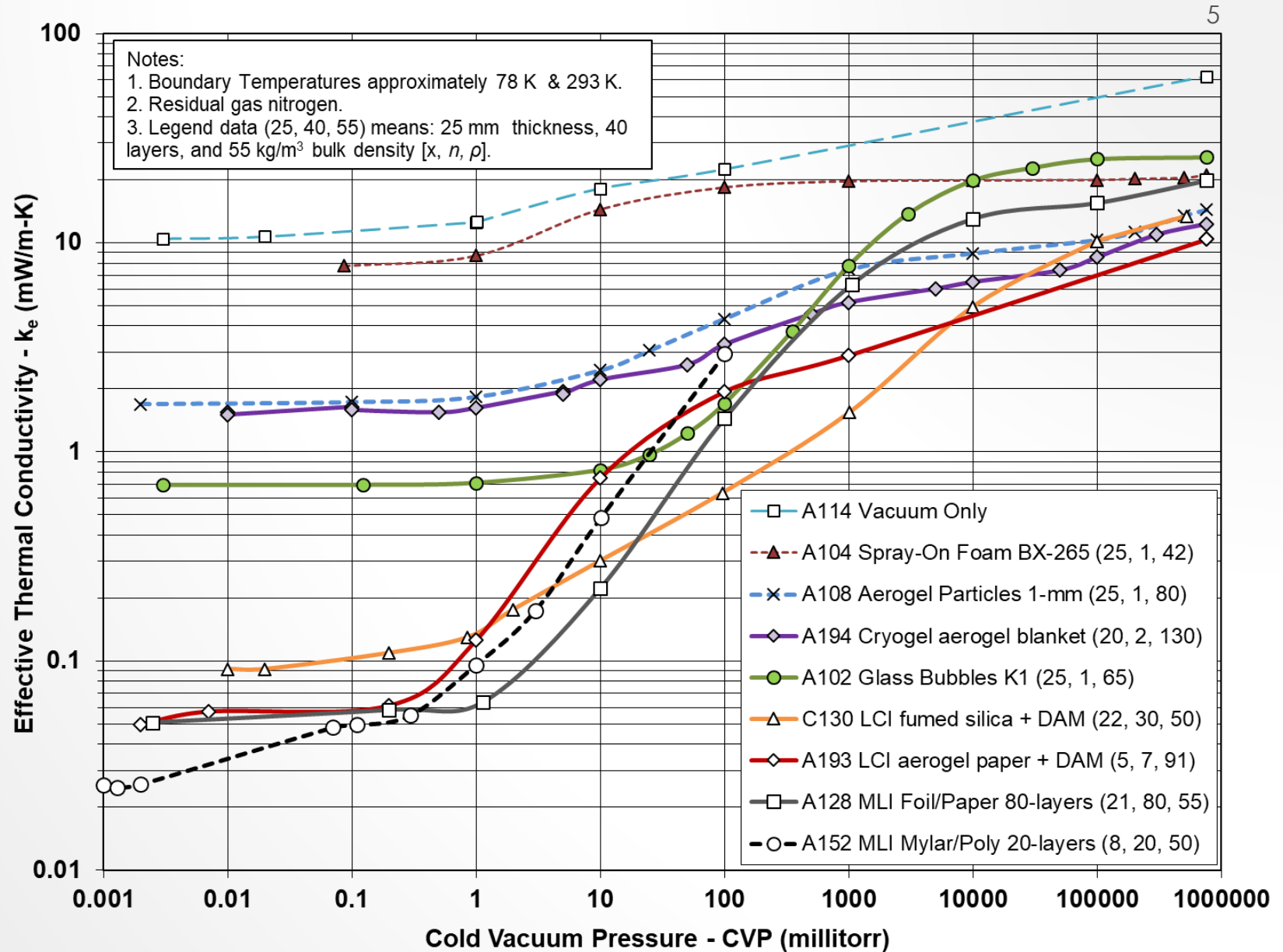
- The  $Q$  is obtained *in a direct way* by boiloff calorimetry
- Effective area of heat transmission ( $A_e$ ):
  - For flat plate geometry, the  $A_e$  is constant through the thickness of the thermal insulation system
  - For cylindrical or spherical geometries, the  $A_e$  is the *log-mean area* between the inner and outer diameters of the thermal insulation system
- From the  $Q$ , the heat flux ( $q$ ) in  $W/m^2$  is calculated:
  - Heat flow rate, under steady-state conditions, through the  $A_e$  in  $m^2$ , in a direction perpendicular to the plane of the thermal insulation system
- For all calculations in this study, the  $k_e$  or  $k_s$  (as applicable) and the  $q$  are based on a standard  $\Delta T$  of 215 K (that is, *for 293 K WBT & 78 K CBT*)

$$q = \frac{Q}{A_e}$$

# CRYOSTAT-100 TEST DATA FOR THERMAL INSULATION SYSTEMS

- Spray-On Foam (NCFI)
- Aerogel Particles (Cabot)
- Aerogel Blanket (Aspen Aerogels)
- Glass Bubbles (3M)
- Layered Composite Insulation (LCI)
- Multilayer Insulation (MLI)

- ✓ Boundary Temperatures = 78 K & 293 K
- ✓ Residual gas = nitrogen
- ✓ Thickness = as noted
- ✓ No. of Layers = as noted
- ✓ Bulk Density = as noted



# SELECT THERMAL CONDUCTIVITY DATA (CRYOSTAT-100) FOR CRYOGENIC THERMAL INSULATION SYSTEMS

- High vacuum (<math><0.1\ \mu</math>)
- Soft vacuum (~100  $\mu$ )
- No vacuum (760,000  $\mu$ )

Note: 1  $\mu$  = 1 millitorr = 0.133 Pa

Thermal Insulation System	Ref. No.	†Density	CVP	*k <sub>e</sub>
		kg·m <sup>-3</sup>	$\mu$	mW/m-K
Glass Bubbles Type K1	A102	65	<math><0.1</math>	0.70
			100	1.7
			760,000	26
Aerogel Particles (1-mm diameter)	A108	80	<math><0.1</math>	1.7
			100	4.3
			760,000	14
LCI with fumed silica & Mylar	C130	50	<math><0.1</math>	0.09
			100	0.64
			530,000	13.4
MLI 80-layers Foil and Paper	A128	55	<math><0.1</math>	0.051
			100	1.5
			760,000	20
MLI 20-layers Mylar and Poly Net	A152	50	<math><0.1</math>	0.026
			100	3.0
			760,000	18 (est.)

†As-tested \*For boundary temperatures of 293 K and 78 K

# STRUCTURAL-THERMAL MATERIALS

- Included are polyimide aerogel AeroZero<sup>®</sup>, Ultem<sup>®</sup>, Foamglas<sup>®</sup>, Divinycell<sup>®</sup>, and Rohacell<sup>®</sup>
- Included for reference: G10 composite, Teflon<sup>™</sup>, balsa wood, and polyisocyanurate spray foam
- $k_e$  data from Macroflash (Cup Cryostat), per ASTM C1774 Annex A4, boundary temperatures of 78 K / 293 K, compressive load of 34 kPa
- Combined properties: Thermal Conductivity + Density + Strength

## Thermal-Structural Figure-of-Merit ( $F_{ST}$ )

where:

$\sigma$  = compressive strength [MPa]

$k_e$  = effective thermal conductivity [mW/mK]

$\rho$  = bulk density [kg/m<sup>3</sup>]

$$F_{ST} = \frac{\sigma}{\rho k_e} \times 10^6 \quad \left[ \frac{\text{K} \cdot \text{m} \cdot \text{s}}{\text{g}} \right]$$

# THERMOPHYSICAL DATA FOR STRUCTURAL-THERMAL MATERIALS USED IN CRYOGENIC SYSTEMS

- Effective thermal conductivity data by **Macroflash**
- Bulk density
- Compressive strength

Material	$\dagger\sigma$	$\rho$	$*k_e$	$F_{ST}$
	MPa	kg·m <sup>-3</sup>	mW·m <sup>-1</sup> ·K <sup>-1</sup>	K·m·s·g <sup>-1</sup>
G-10 (transverse direction)	448	1,939	467	495
Ultem® 2300 Glass Filled PEI	221	1,500	212	695
Ultem® 9185 PEI (3-D printed)	100	1,199	145	575
Teflon™ PTFE	24.1	2,120	253	45
Rohacell® WF-300 PMI foam (14 kPa)	17.8	324	42.1	1,305
Balsa Wood (transverse direction)	7.0	166	45.9	919
AeroZero® polyimide aerogel	1.6	150	28.1	380
Foamglas® Cellular Glass Foam	0.8	118	32.3	210
Divinycell® H45 PVC Foam (14 kPa)	0.6	50	23.8	504
Spray Foam Polyiso BX-265 (14 kPa)	0.4	37	22.6	483

†At ambient temperature \*Boundary temperatures 293 K / 78 K; compressive load 34 kPa or as noted.





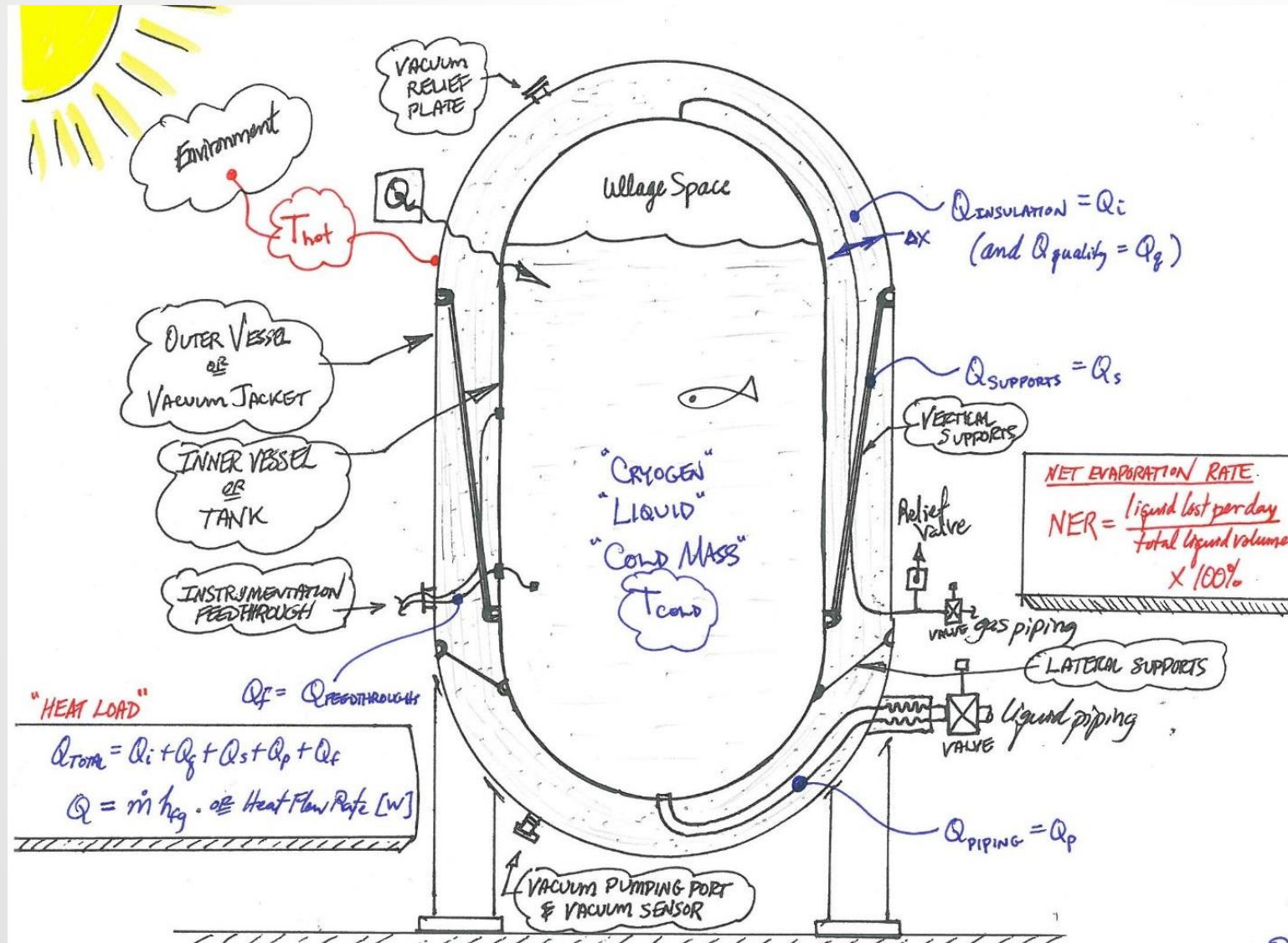
# CRYOGENIC SYSTEM HEAT LEAK ANALYSIS

- Optimum design involves a different combination of materials for each specific case
- For simplicity and comparison, uniform hot and cold surfaces are assumed (testing includes data for LN<sub>2</sub> at 77 K and LH<sub>2</sub> at 20 K)
- Wide range of different cold vacuum pressures (CVP) represent actual working systems
- Cryogenic system example cases:
  1. Cylindrical tank                      high vacuum (<0.1 $\mu$ )
  2. Large spherical tank                moderate vacuum (~10 $\mu$ )
  3. Small composite tank               soft vacuum (~100 $\mu$ )
  4. Transfer piping                        normal vacuum (~1 $\mu$ )
  5. Piping field joint                      no vacuum (760,000 $\mu$ )

# CRYOGENIC SYSTEM EXAMPLE CASES

- Three storage tank systems (LH<sub>2</sub>) analyzed:
  - 1) Medium-size 125,000-liter cylindrical SST tank with a carbon steel outer jacket and a 200-mm annular space
  - 2) Large-size 3,200,000-liter spherical SST tank with a carbon steel outer jacket and a 1,200-mm annular space
  - 3) Small-size 100-liter carbon composite tank with a stainless steel (SST) outer jacket and an annular space of 23-mm thickness
- Two transfer piping systems (LH<sub>2</sub>) analyzed:
  - 4) Transfer line consisting of a DN25x80-mm all SST vacuum-jacketed (VJ) pipe segment of 18-m length
    - Ends are disregarded for the reference case of a very long pipeline
  - 5) One-meter overall length field joint connection between two DN250x300-mm VJ pipe segments

# VACUUM-JACKETED (VJ) TANK BASICS



# CRYOGENIC SYSTEM EXAMPLE CASES: VJ STORAGE TANKS

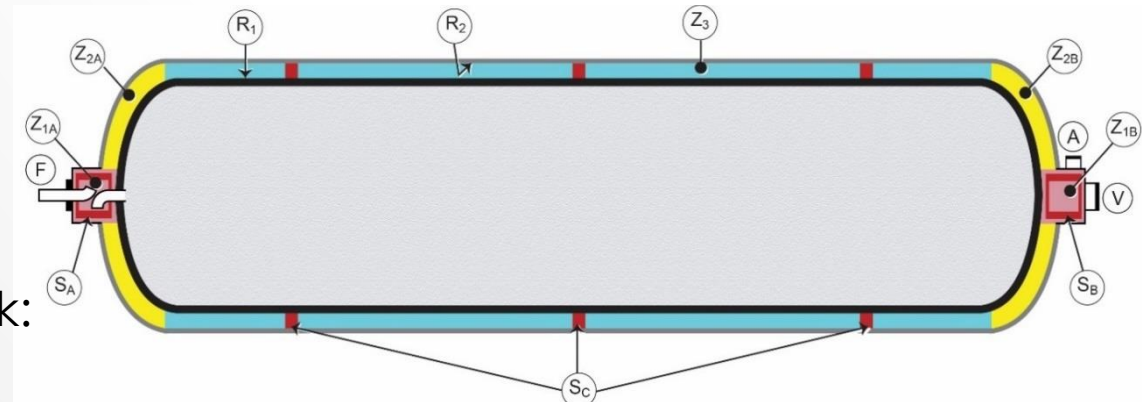


1) Medium cylindrical tank: high vacuum ( $<0.1\mu$ )



2) Large spherical tank: moderate vacuum ( $\sim 10\mu$ )

3) Small composite tank:  
soft vacuum ( $\sim 100\mu$ )

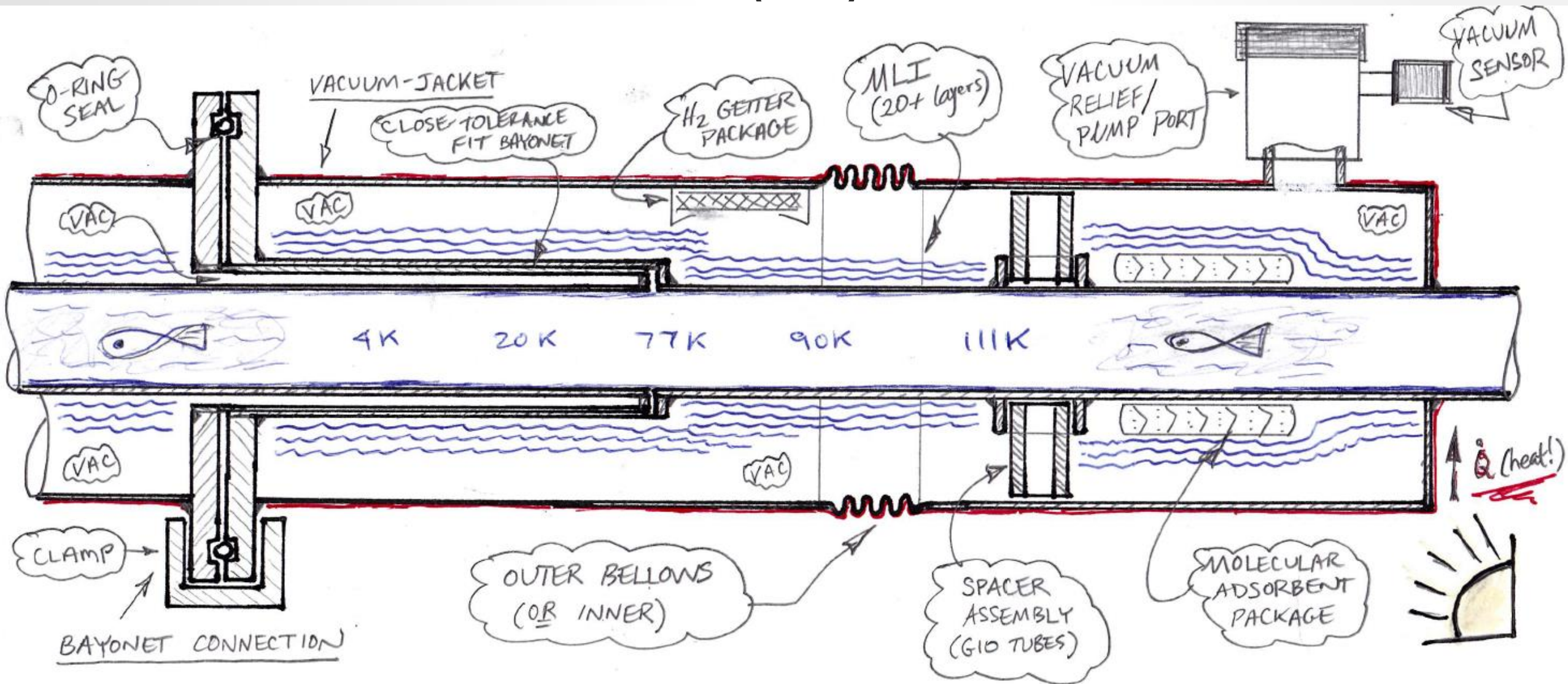


# VJ STORAGE TANKS (LH<sub>2</sub>) AT NASA/KSC LAUNCH COMPLEX 39B

Existing 3,200 m<sup>3</sup> sphere  
(right); new 4,700 m<sup>3</sup>  
sphere under  
construction (left)



# VACUUM-JACKETED (VJ) PIPING BASICS



# CRYOGENIC SYSTEM EXAMPLE CASES: VJ TRANSFER PIPING



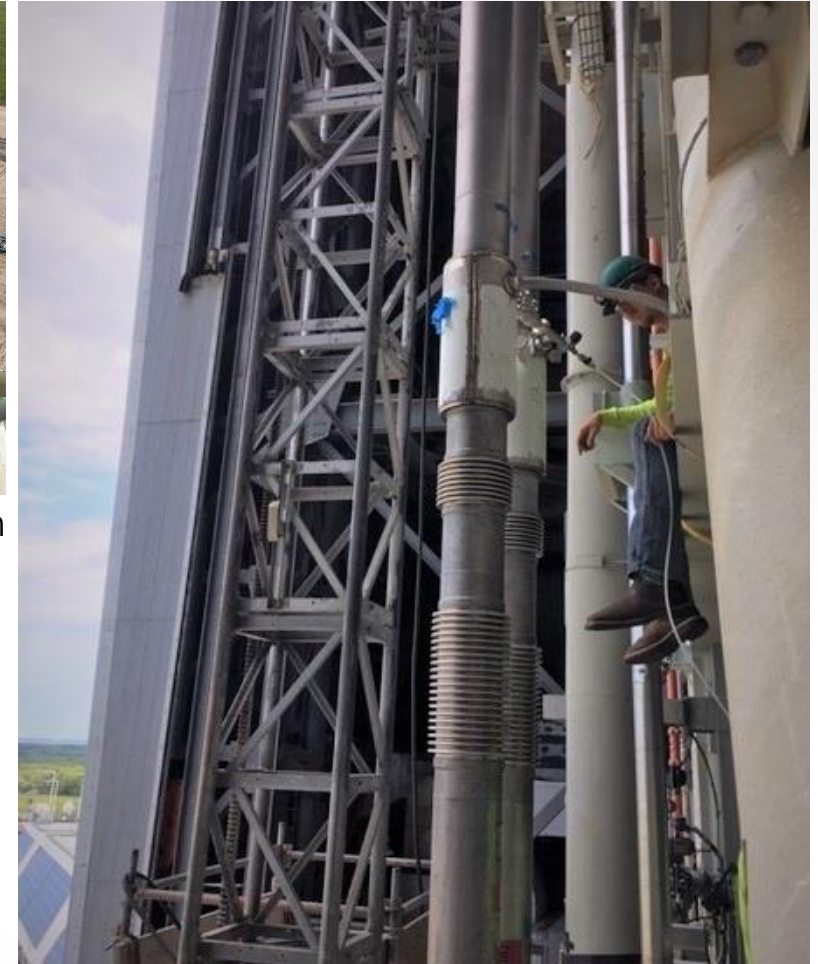
Transfer piping: normal vacuum ( $\sim 1\mu$ )



Before close-out with cans and aerogel in the field

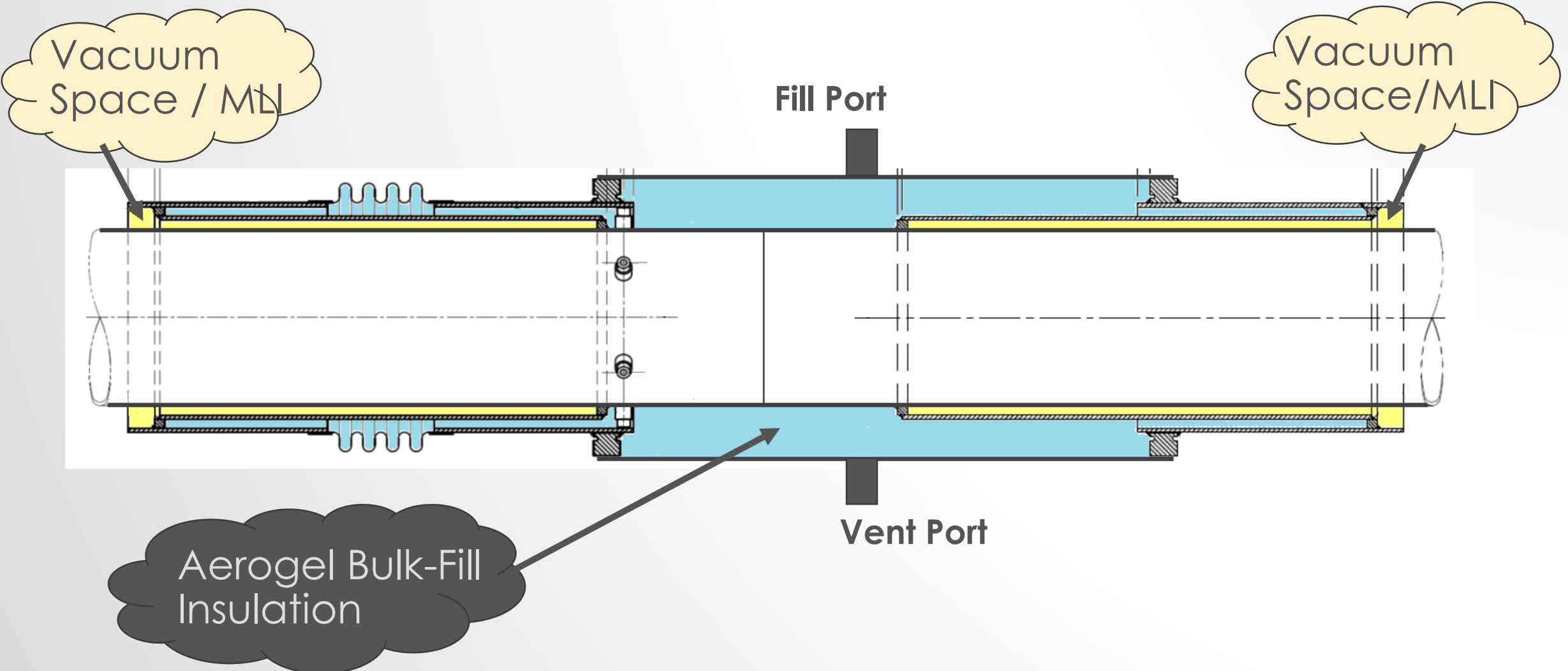


MLI wrapping in the shop



Piping field joint: no vacuum ( $760,000\mu$ )

# FIELD JOINT FOR VJ PIPING CONNECTION





# COLD TRIANGLE DESIGN APPROACH

The total heat leak into any cryogenic tank or piping assembly is comprised of three parts in addition to the baseline heat leak through the insulation ( $Q_i$ ):

- 1) Heat leak through the structural supports ( $Q_s$ )
- 2) Heat leak through piping connections, penetrations, and feedthroughs ( $Q_p$ )
- 3) Insulation Quality Factor or IQF ( $Q_{IQF}$ )
  - a. Heat leak due to practical constraints of fabrication, installation, and assembly of the insulation system on the cold mass
  - b. Heat leak due to the negative effects of the supports and connections on the insulation performance
  - c. Other parasitic heat leaks or real-world factors

$$Q = Q_{\text{total}} = Q_i + Q_s + Q_p + Q_{IQF}$$



**Cold Triangle**  
 Insulation +  
 Supports +  
 Piping +  
 IQF

# Cold Triangle



$$Q_{\text{total}} = Q_i + Q_s + Q_p + Q_{\text{IQF}}$$



# USE OF THE COLD TRIANGLE

- Provides a basis for:
  - Evaluating performance benefits of new materials
  - Analyzing the cost effectiveness in overall system design
- Insulation Quality Factor (IQF) for IQF HEAT LOAD or  $Q_{IQF}$ 
  - $Q_{IQF} = 0$  is perfection or perhaps idealized laboratory conditions
  - $Q_{IQF} = 0$  means no additional heat leak
- Use of the IQF:
  - Determined by testing or estimated by analysis
  - Represents the true additional heat load of real systems
  - Set up as an individual parameter or individualized for the insulation, the piping, and the supports
    - Account for as a percentage of each part, if data are available
    - Or use as a placeholder, if data are not available

# ANALYSIS: FIVE EXAMPLE CASES

- Breakdowns of the total heat leaks for the five different systems (LH<sub>2</sub> and/or LN<sub>2</sub>) are summarized in the following table
- The insulation materials chosen and the data given represent **preferred solutions in each case** (or context) among this vast array of different applications
- The heat leakage rate (Q) for each system is comprised of three main parts: insulation, supports, and piping penetrations:
  - Plus another crucial part: the insulation quality factor (IQF)

# Total heat flow rate analysis for different examples of cryogenic systems

Cryogenic System Description	Thermal Insulation System	CVP	Dimensions		Q <sub>i</sub>	Q <sub>s</sub>	Q <sub>p</sub>	^Q <sub>IQF</sub>	Q <sub>total</sub>	q	k <sub>s</sub>
			†	*A <sub>e</sub>							
			μ	mm							
<b>1) Medium tank, VJ, factory built (125-m<sup>3</sup>)</b>	MLI (80 layers) for high vacuum	<b>&lt;0.1</b>	250	229	33.7	17	43.3	6.3	<b>300</b>	1.3	1.5
<b>2) Large tank, VJ, site-built (3,200-m<sup>3</sup>)</b>	Bulk-fill glass bubbles, moderate vac	<b>10</b>	1200	1194	74.1	12.9	12.9	0.0	<b>240</b>	0.2	1.1
<b>3) Small composite tank, VJ, conceptual (0.1-m<sup>3</sup>)</b>	LCI for soft vacuum	<b>100</b>	23	1.6	68.9	21	10.4	?	<b>14.4</b>	9.0	1.0
<b>4) Piping System, DN25x80-mm, VJ, with 2 bayonet joints, factory built (18.3-m)</b>	MLI (20 layers) for normal vacuum	<b>1</b>	26	3.2	12.0	15	38.3	34.4	<b>20.9</b>	6.5	0.8
<b>5) Piping Field Joint, DN250x300-mm, double-wall, field fabricated (1-m)</b>	Bulk-fill aerogel for non-vacuum	<b>760000</b>	50	0.88	54.1	27	19	0.0	<b>49.0</b>	55.7	12.9

\* Effective surface area for heat transmission between inner and outer shells

^ IQF = Insulation Quality Factor

† ΔT = 293 K - 78 K = 215 K

# ANALYSIS: SUMMARY

- These examples give **first order approximations** and the interplay among materials, design, installation, and manufacturing
- There are many more details and materials choices to be made in the final design for the optimum solution
- The IQF is a first order way to quantify the combined effects of the insulation's thermal performance due to:
  - a) Practical limitations of its installation and
  - b) Increased heat transmission due to supports and piping
- Note: the IQF is not a degradation factor but a very real heat load

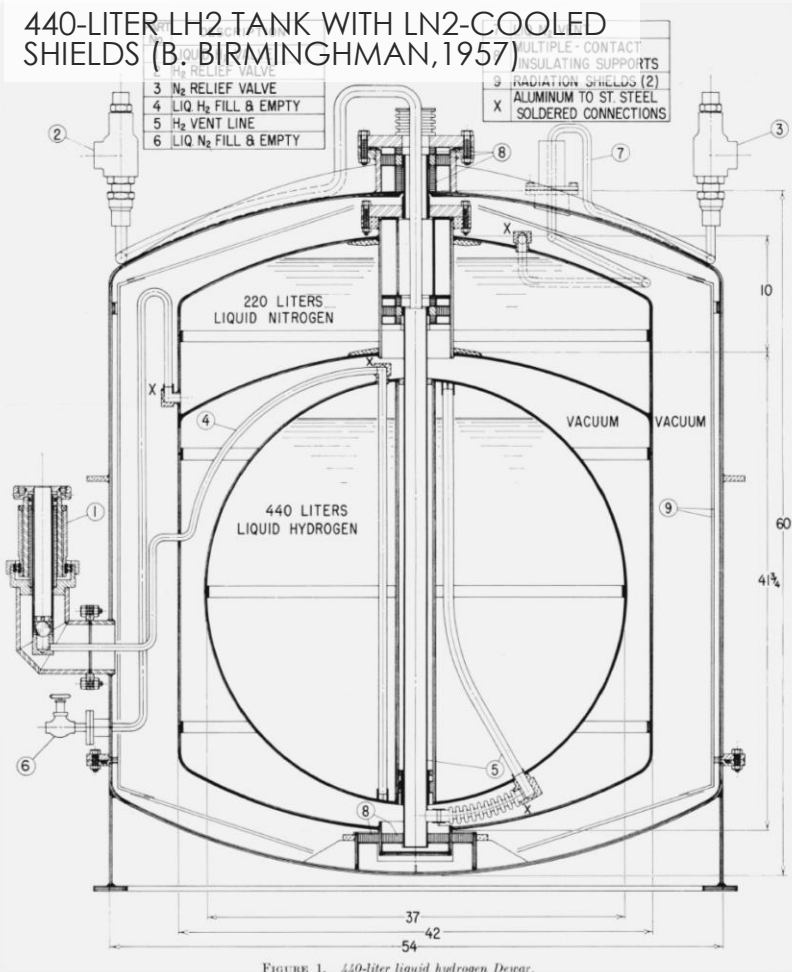
# DESIGN + ANALYSIS + TESTING

- Standardized laboratory testing of materials/systems is the start
- Thermophysical data for **aerogels, aerogel composites, multilayer insulation, novel layered composites, and glass bubbles** for standard test conditions of 293 K / 78 K under conditions from high vacuum to ambient pressure
  - **Reference materials data** are included for direct analytical comparison
  - Basic data for structural-thermal materials, including ranking of their **structural-thermal figure of merit** (FST) are given
- Testing of VJ tanks and VJ piping systems with LN<sub>2</sub> and LH<sub>2</sub> under real-world conditions (prototypes and field demonstrations)
- Cold Triangle analysis of a cryogenic storage tanks system shows the total heat load and its constituent parts
  - Heat load (Q) in W and heat flux (q) in W/m<sup>2</sup>
  - Total system k-factor (k<sub>s</sub>) in mW/m-K

# CONCLUSION

- System design, engineering analysis, AND thermal testing (materials):
  - These go hand-in-hand for determining the “best insulation system” for a given design situation
- Practical methodology for cryogenic system design: “cold triangle” approach of insulation, supports, and piping plus the insulation quality factor (IQF)
  - Total heat leak is what matters: to minimize it in a cost-effective way, the thermal performance must be understood as a summation of its parts
  - Basis for evaluating performance benefits of new materials and analyzing the cost effectiveness in the total system design
- Examples of different cryogenic storage tanks and transfer piping systems:
  - These show the relative importance of both insulation and structural materials for achieving designs of highest energy efficiency





- The elements of heat transmission into a cryogenic system are highly interdependent
- Analysis of the total heat leak is an iterative process
- The right design approach depends on....everything\* and, of course,

....the economic objectives

\*The shape, the size, the components, the environment, the materials, the temperatures, the process, the cryogen, the fabrication constraints, the operational objective, the duty cycle, etc., etc.

# END

Advanced cryogenic insulation systems

# THANK YOU

for your attention

Questions?

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