

# **Design of a Magnetic Gear for NASA's Vertical Lift Quadrotor Reference Vehicle**

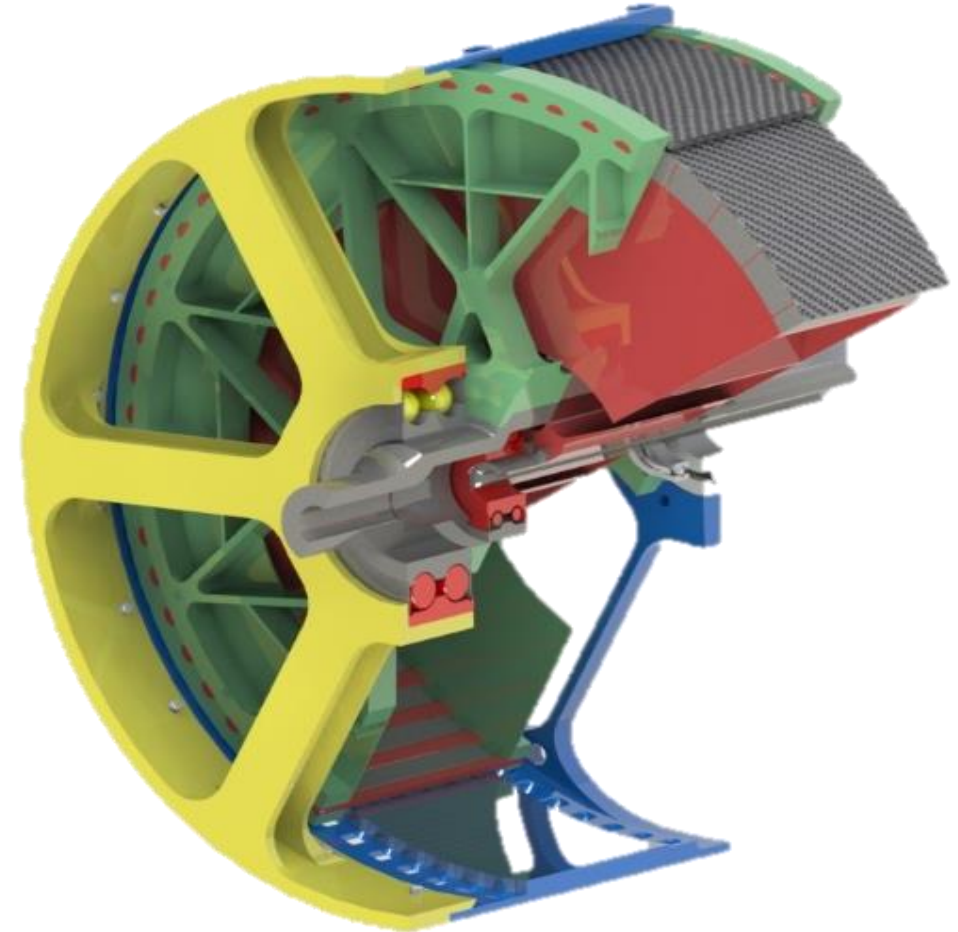
**Thomas Zachary Justin  
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**NASA Glenn Research Center  
Materials and Structures Division  
Rotating and Drive Systems Branch**

# Outline

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- Background and Motivation
- Concentric Magnetic Gears
- Enabling Design Principles
- Preliminary Design
  - Design Code
- Final Design
  - Efficiency
  - Thermal
  - Structural
- Conclusions



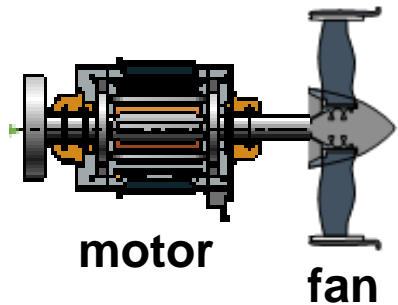
**PT-4**

# Background & Motivation

- NASA set goals for aircraft efficiency, emissions, reliability, and noise
- Parallel large & small aircraft development
  - Economic benefit of alternative propulsion
- Electrified aircraft propulsion is a key enabler
- Most concepts use direct drive
- Geared drives are almost always mass optimal

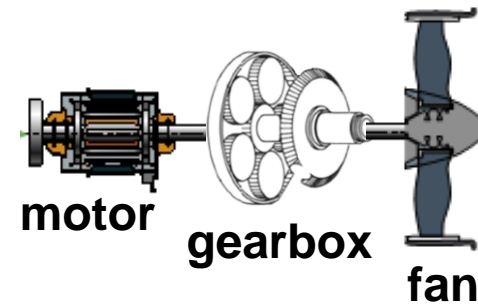


## Direct drive



- + Simpler
- Non-optimal motor and/or fan

## Geared drive



- + Optimized motor & fan
- + Enables cross shafting
- More complex
- Potentially less reliable

# Background & Motivation

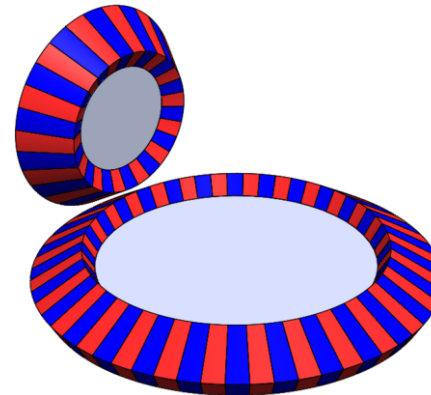
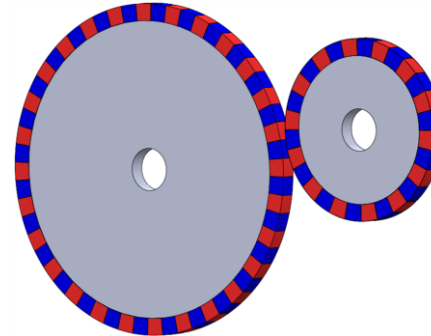
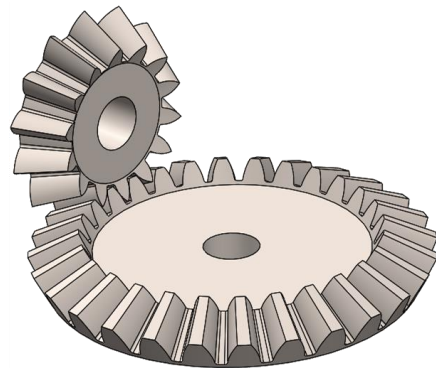
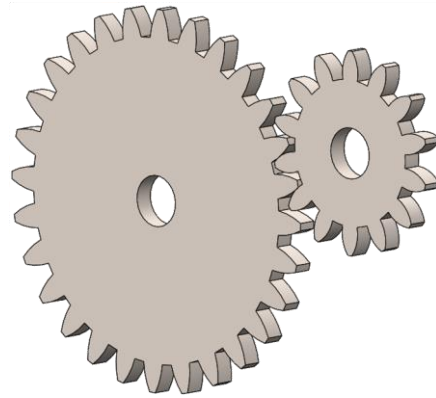
## Mechanical gearing

### Pros

- + High / very high torque/mass (**specific torque**)
- + High / very high efficiency
- + Mature technology

### Cons

- Contact-related wear & failure
  - Requires lubrication system(s)
  - Routine & costly maintenance
- Strong tonal vibration & cabin noise



## Magnetic gearing

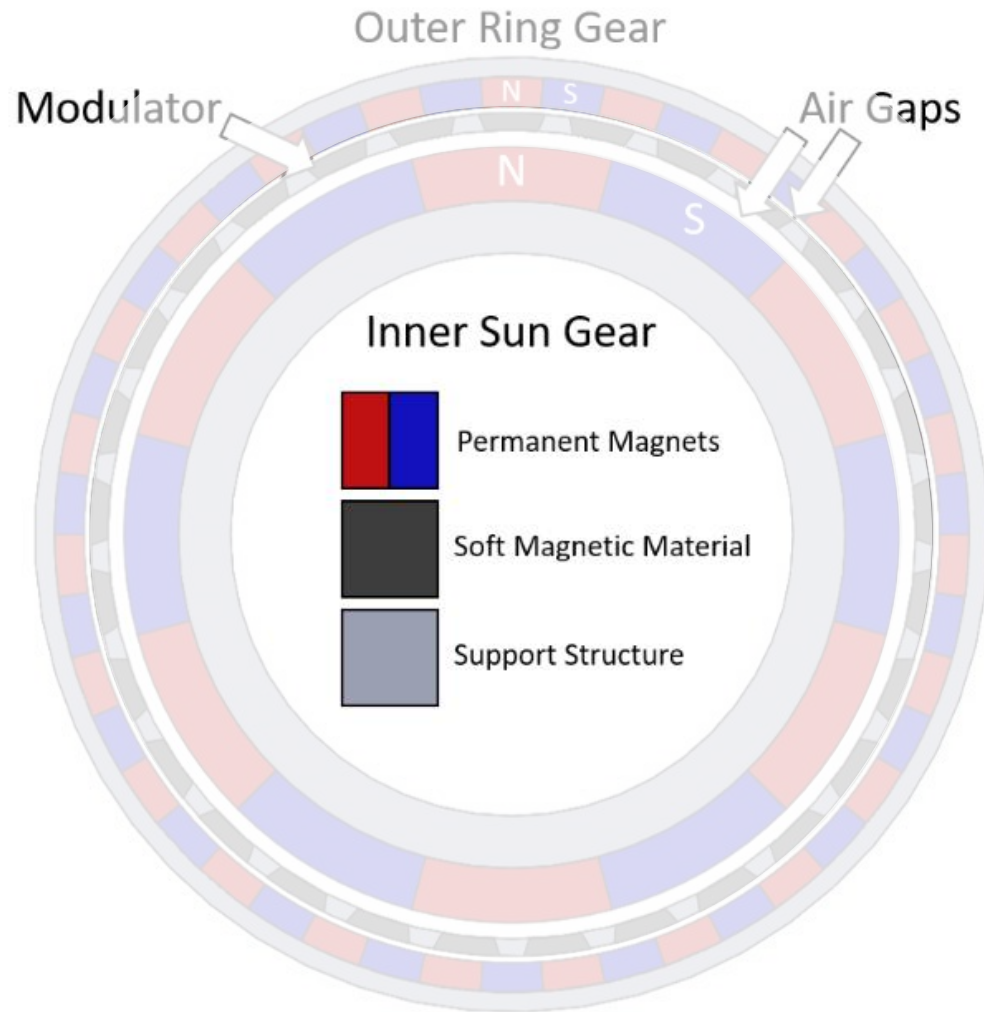
### Pros

- + Non-contact
  - + No lubrication
  - + Low maintenance
- + Easily integrated in electric machines
- + Potentially low vibration

### Cons

- Unknown limits on specific torque & efficiency
- Magnet temperature limit
- Individual magnet interaction weaker than 1 gear tooth pair

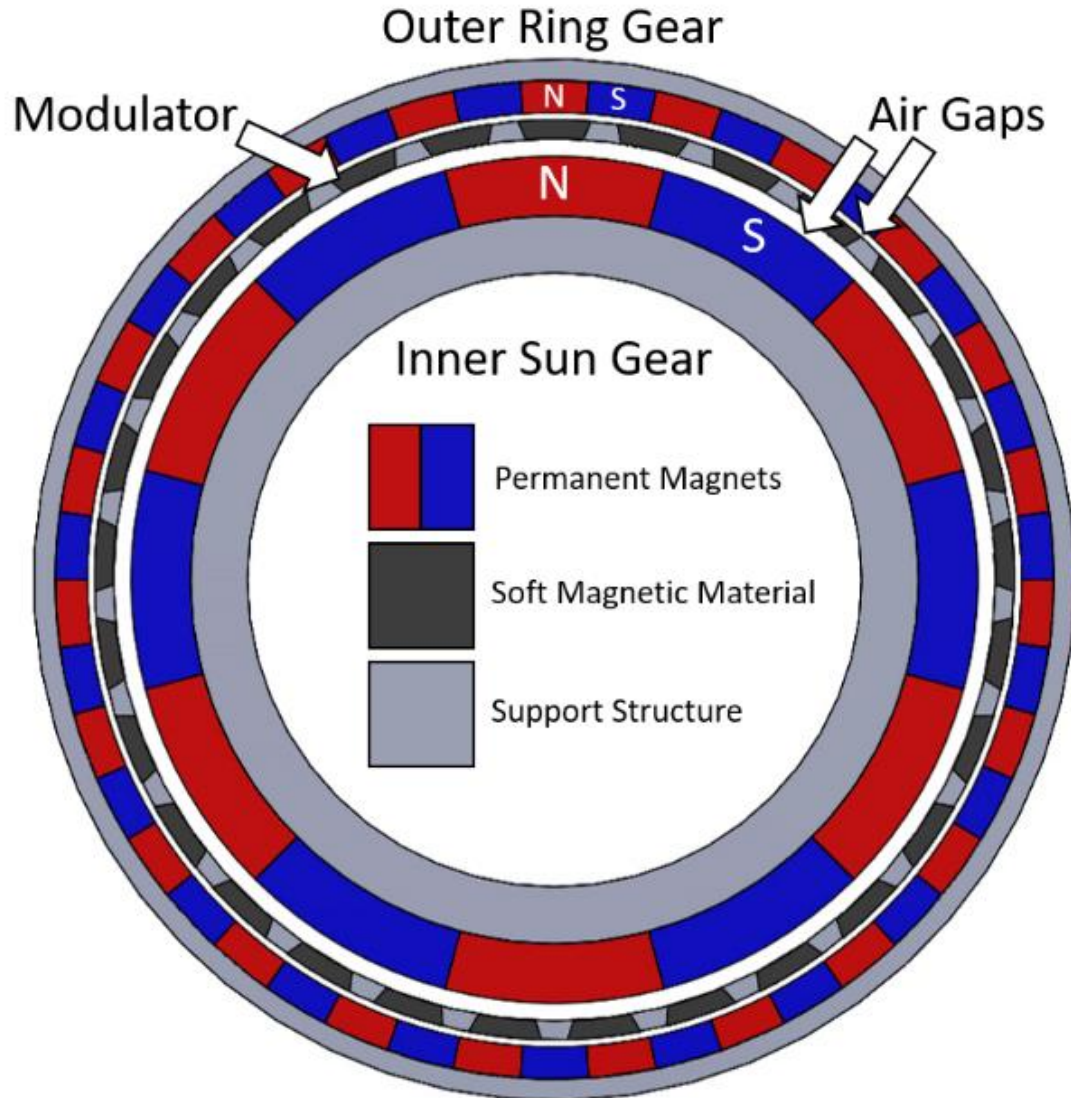
# Concentric Magnetic Gears



- Rule of thumb:  
Magnetic fields with matching spatial harmonic order can couple to transmit torque
- Ring and Sun gear have different pole counts
  - Produce different spatial harmonic
- Modulator “modulates” the flux of each rotor so that that have matching spatial harmonic order in the airgaps



# Concentric Magnetic Gears



$$\cos(\theta) * \cos(\alpha) = \frac{1}{2} (\cos(\theta + \alpha) + \cos(\theta - \alpha))$$

$$B_{rs} = F * \cos(PS * (\theta + \alpha))$$

↳ Number of Sun Gear Pole Pairs

$$u = u_{avg} + u_m * \cos(Q * (\theta + \beta))$$

↳ Number of Pole Pieces

$$B_{rs} * u_m = u_{avg} * F * \cos(PS * (\theta + \alpha)) + \frac{F * u}{2} \cos((Q + PS)\theta + PS * \alpha + Q * \beta) + \frac{F * u}{2} \cos((Q - PS)\theta - PS * \alpha + Q * \beta)$$

$$PR = Q \pm PS \quad \text{or} \quad Q = PR \pm PS$$

↳ Number of Ring Gear Pole Pairs

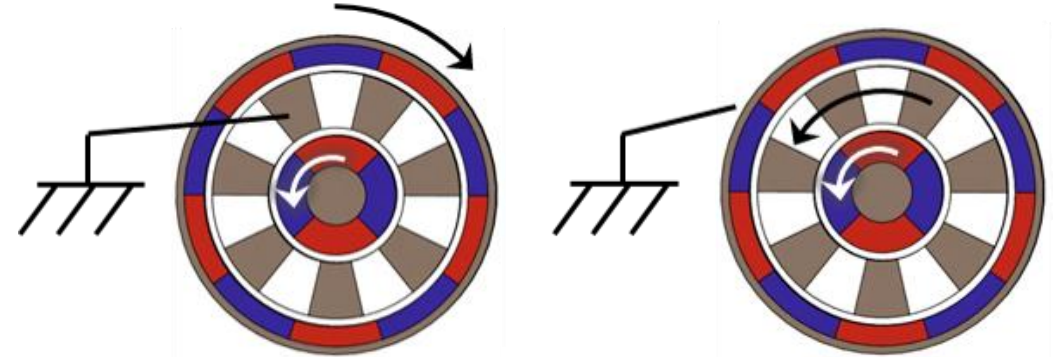
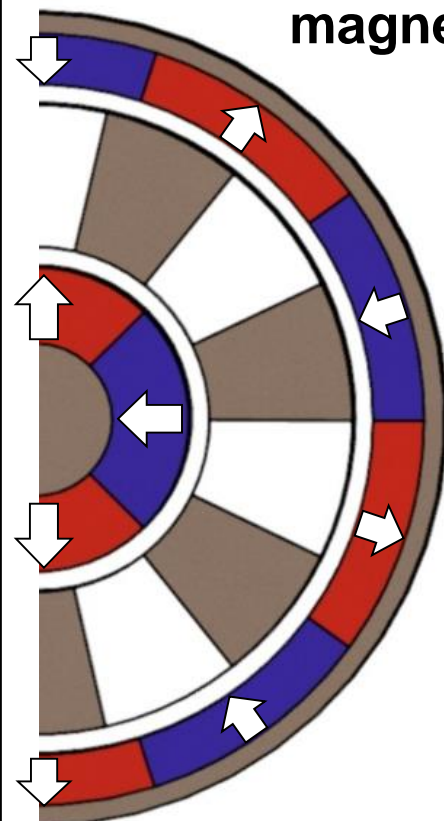
# Concentric Magnetic Gears

## Gear Ratio

Mechanical planetary gear



Analogous concentric magnetic gear



Output	Q Selection	Gear Ratio
Ring Gear	PR-PS	$\frac{PR}{PS}$
	PR+PS	
Modulator	PR-PS	$\frac{Q}{PS} = \frac{PR}{PS} - 1$
	<b>PR+PS</b>	<b><math>\frac{Q}{PS} = \frac{PR}{PS} + 1</math></b>

# Background & Motivation

## Phase I 2017

- How do they work? (PT-1)
- Can they be lightweight? (PT-2)

## Phase II 2018-2019

- Can they be efficient? (PT-3)
- Can they be efficient and light weight? (PT-4)

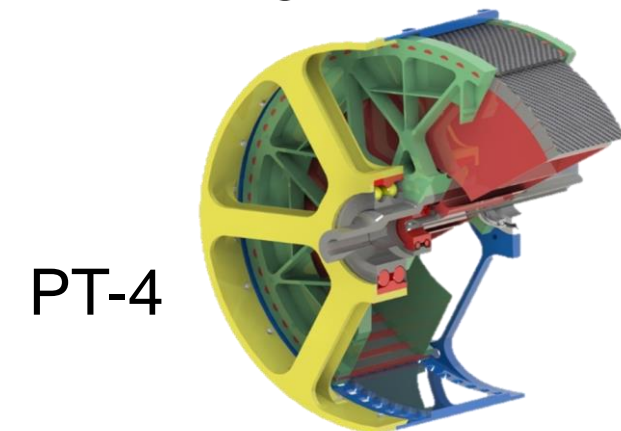
## Phase III 2019-2020

- How to pair them with motors?
- Can they be reliable?



PT-2  
45 Nm/kg

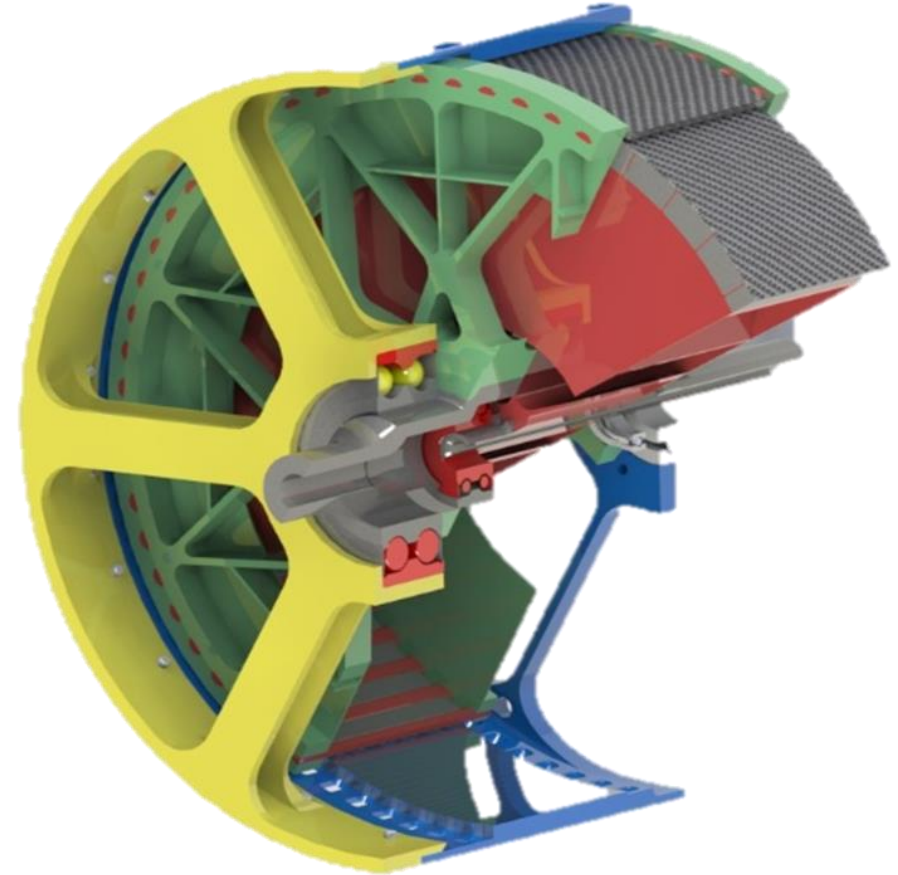
PT-3 >98%  
Efficient





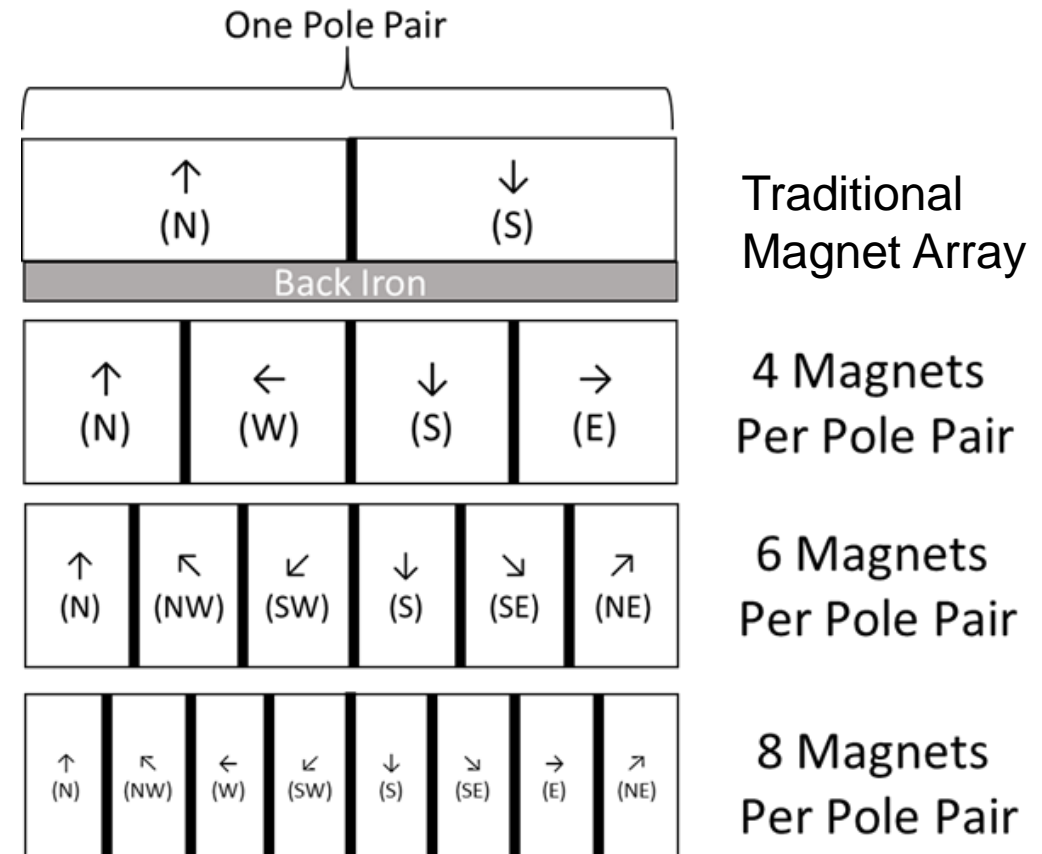
# Lightweight and Efficient Magnetic Gears

1. Halbach Arrays
2. Magnet Laminations
3. Minimize Modulator Thickness
4. Minimize Airgaps



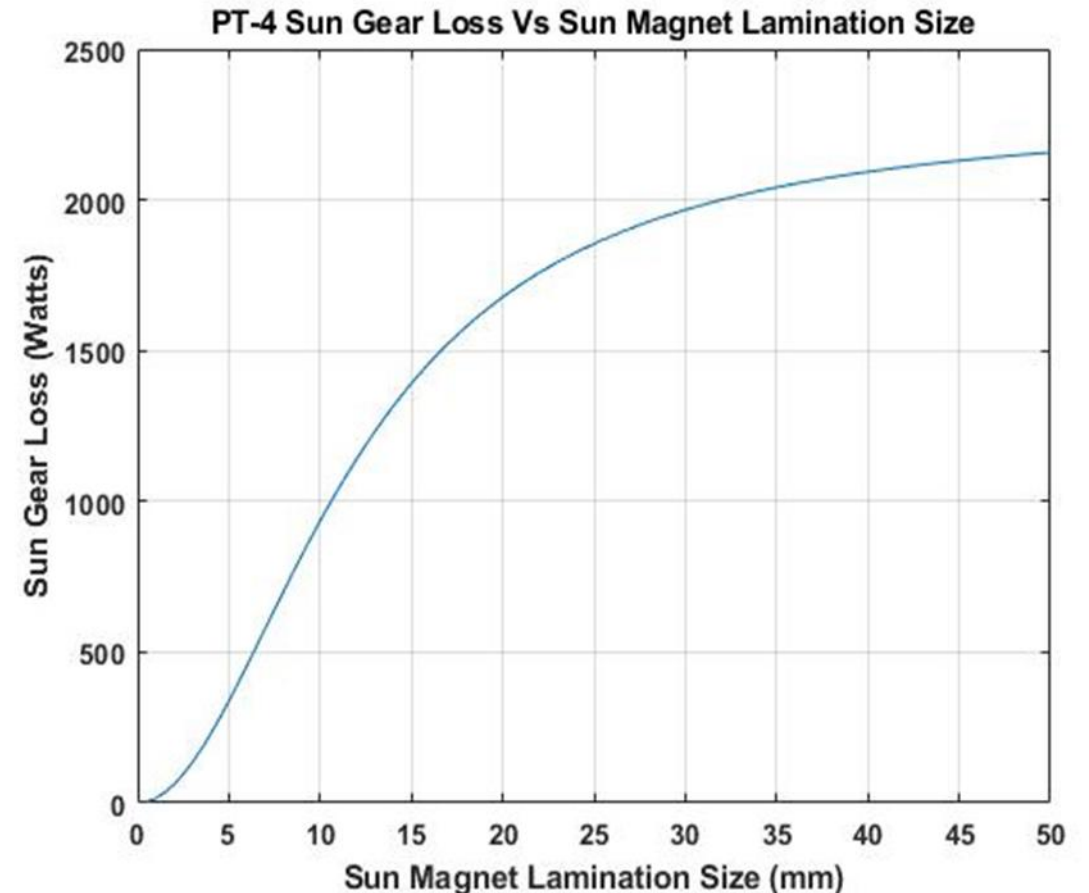
## Halbach Arrays

- Eliminate need for back iron
- Increase magnet per pole count:
  - Improves Array specific flux
  - Suppresses Eddy Current Loss
  - Magnet fill percentage loss



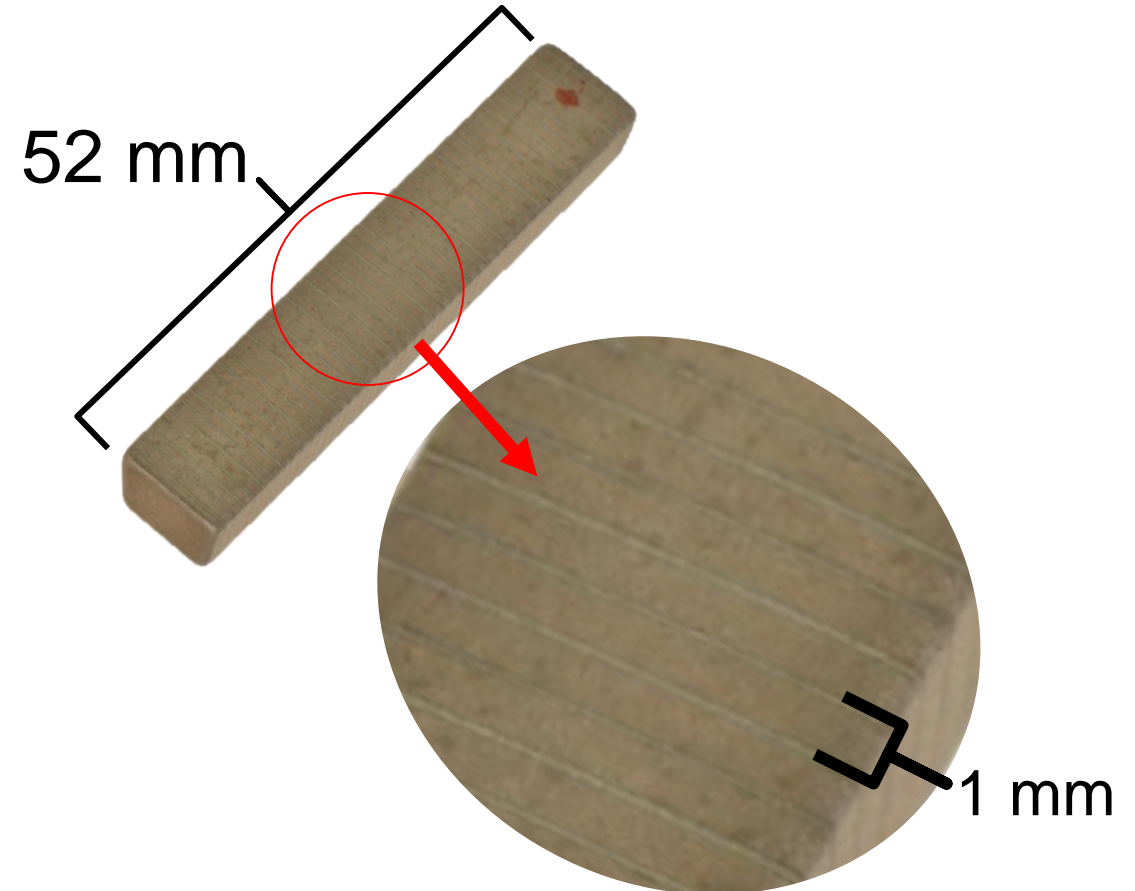
## Magnet Laminations

- Suppress magnet eddy current loss
- Can enable >99% efficiency
- Magnet Fill percentage
- To enable high efficiency
  - High magnets per pole
  - Small magnet laminations



## Magnet Laminations

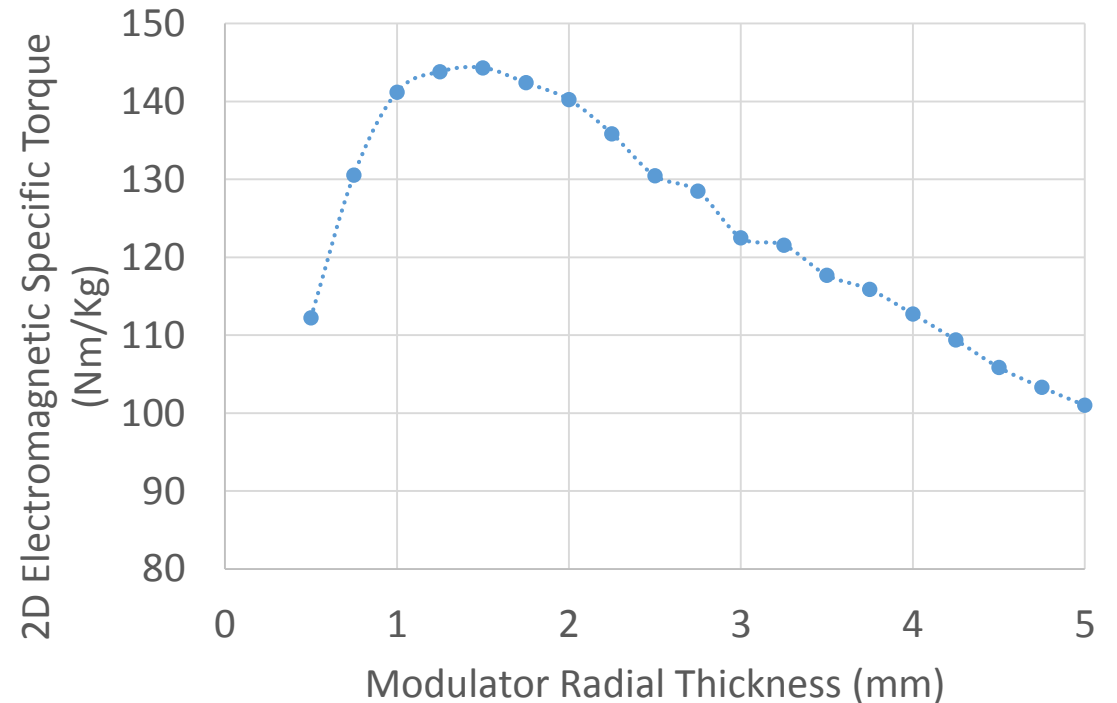
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## Minimize Modulator Thickness

- Electromagnetically:
  - There is an optimum Modulator Thickness
  - Typically ~1.5 mm
- Mechanical structure limits thickness
  - Sandwiched between airgaps
- PT-2: 2.6 mm thickness
- PT-4: 2 mm thickness

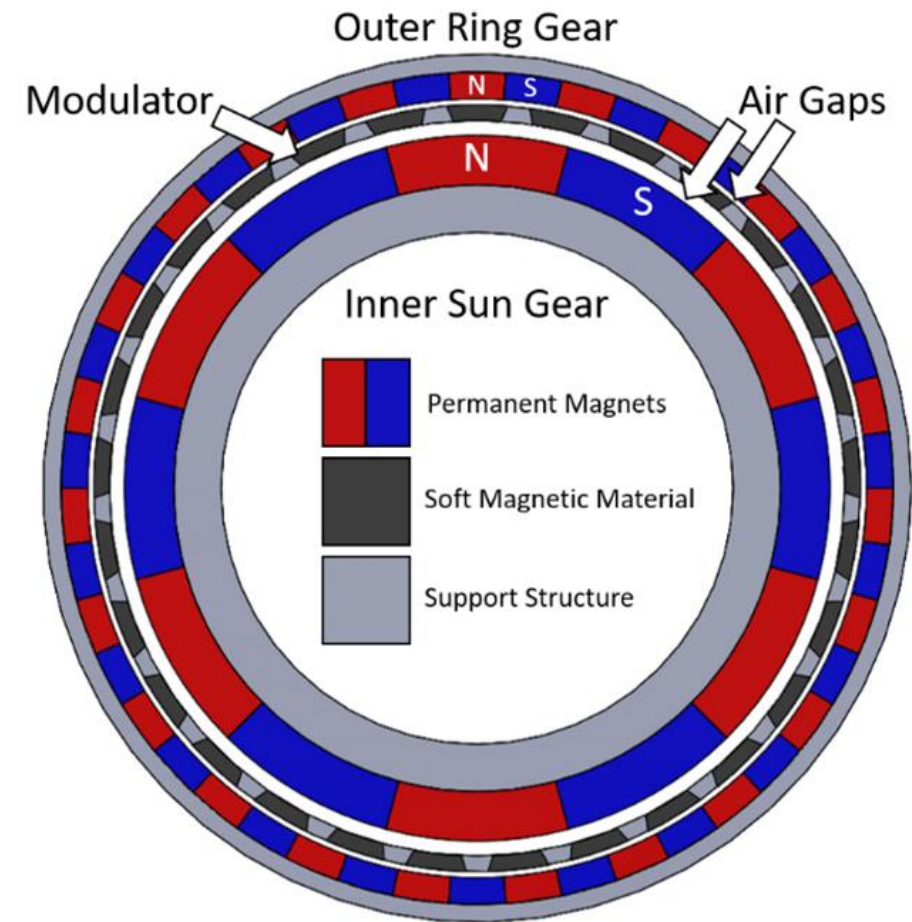
Specific Torque Vs Modulator Thickness





## Minimize Modulator Thickness

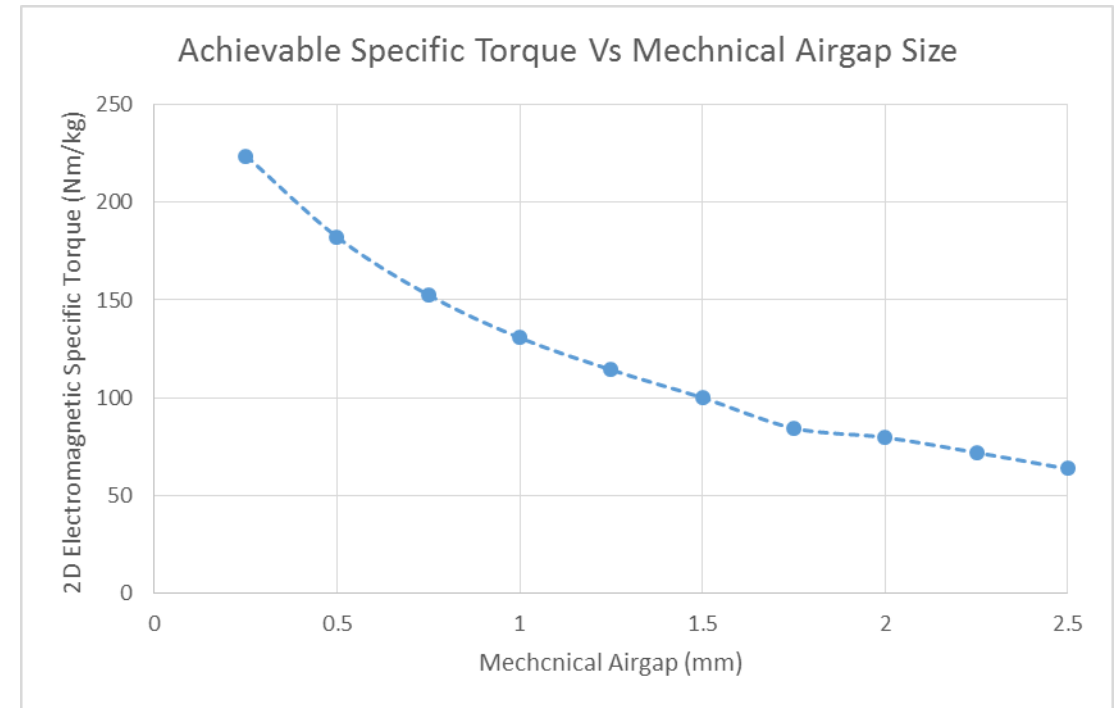
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## Air Gap Thickness

- Potential to double magnetic gear specific torque
- Smaller airgaps reduces optimal modulator thickness
- Development area to improve magnetic gears:

Modulator structure that enables smaller airgaps and smaller modulators



## PT – 4 Design

- Quadrotor Reference Vehicle
  - NASA's RVLТ Project
  - Single Passenger Air Taxi
  - 4 Rotors
    - 680 RPM (low noise)
    - 16.1 kilowatts
    - ~8000 RPM motor



NASA's Vertical Lift Quadrotor Reference Vehicle

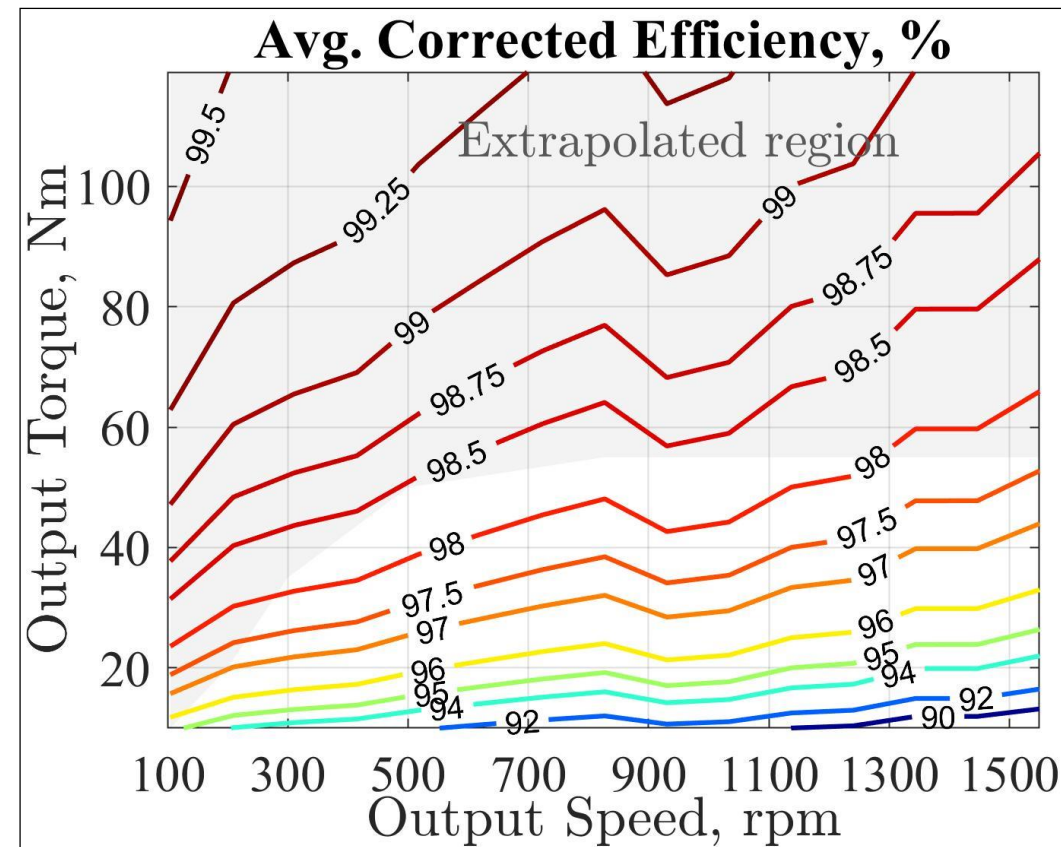
# Preliminary Design

	Requirement	Value
Performance Targets	Target Total Gear Mass	4.5 kg
	Target Gear Efficiency	>97%
Nominal Operating Point	Gear Ratio	~12:1
	Output Torque	226 Nm
	Output Speed	680 RPM
Quadrotor Load Estimates	Required Bearing Life (99% Reliability)	10,000 Hours
	Thrust Load	1400 N
	Propeller Mass	10.5 Kg
	Propeller Hub Moment (Worst Case)	1203 Nm
	Propeller Hub Moment (Nominal Case)	604 Nm
	Nominal Propeller Hub Drag	100 N
	Max Turning Acceleration	2 G's
	Nominal Yaw Rate	0.25 rad/s
	Nominal Pitch Rate	0.25 rad/s

## Preliminary Design

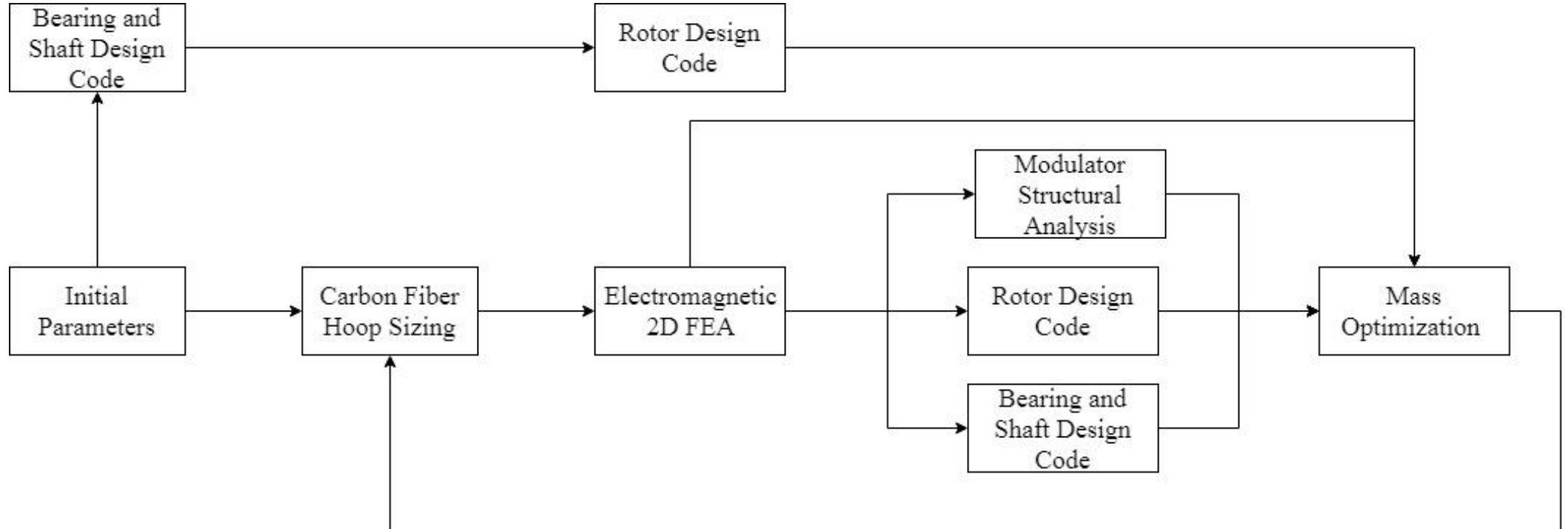
- Preliminary Studies
  - > 97% possible with magnet laminations
    - Validated by PT-3 testing
  - Thermal closed
  - Mass was the question mark
- Electromagnetic and Structural Design Code Developed
  1. Total Gear Mass Effects
  2. Modulator Structure
  3. Sun magnet retaining hoop

Results of PT-3 Dynamic Testing



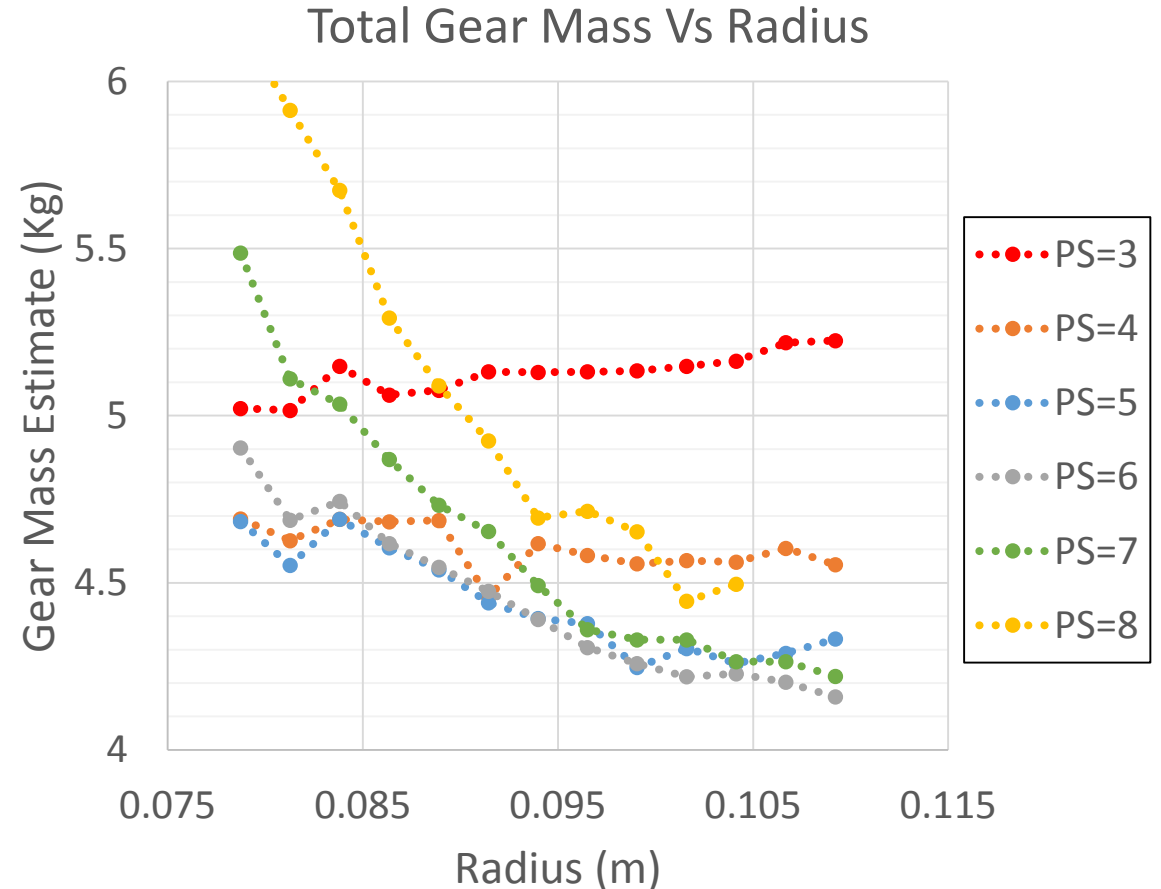


## Design Code Architecture



## Design Code Results

- Assumed 80C Operating Temp
- N52M Neodymium Magnets
  - Highest Grade with 80C operating T
- $\text{Fe}_{49}\text{Co}_{49}\text{V}_2$  Modulator
- Parametric sweeps of radius and PS
- Designs under 4.5 Kg
  - 5-7 Sun Gear Pole Pairs (PS)
  - $>.09$  m radius



## Selected Electromagnetic Design

### PT-4 Electromagnetic Design Parameters

Limits electrical frequency

**Sun Pole Pairs**

5

**Magnetic OR (mm)**

104.1

**Axial Length (mm)**

52

Reasonable Modulator Thickness

**Sun Magnet Thickness (mm)**

7.878

Good Ring Magnet Thickness

**Modulator Thickness (mm)**

2

**Ring Magnet Thickness (mm)**

3.302

Eliminates Symmetry

**Modulator Pole Pieces**

61

To reduce torque ripple

**Ring Pole Pairs**

56

**Inner Pole Piece Span Angle**

4

**Mid Pole Piece Span Angle**

2.3

**Outer Modulator Span Angle**

5.44

**Pole Piece Fillet Radius (mm)**

0.127

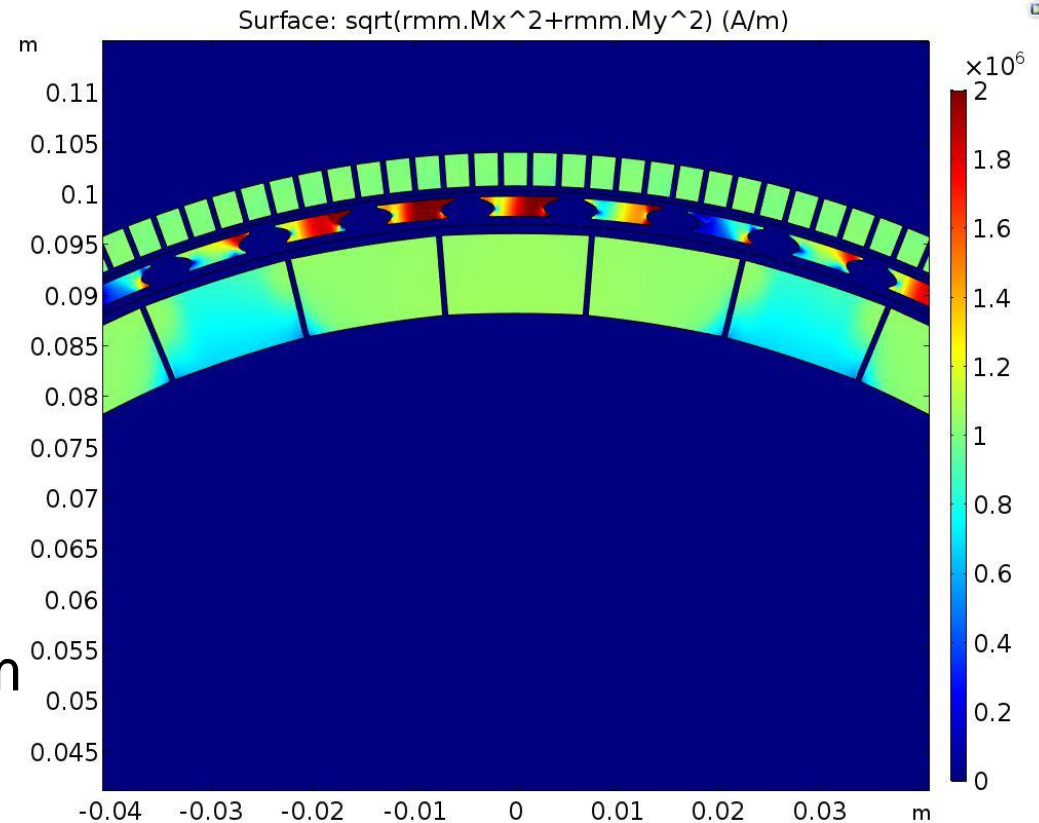
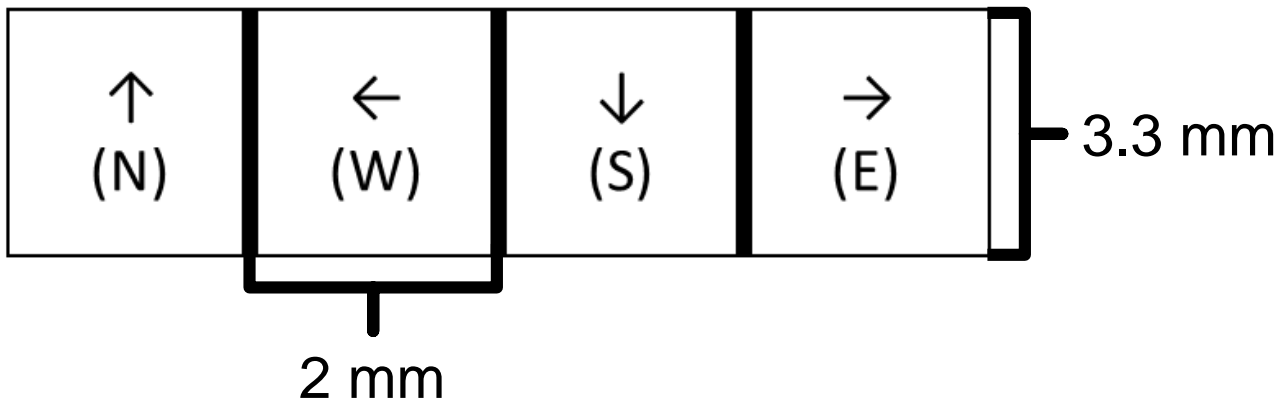
**Magnetic Mass (kg)**

2.705

Re-optimized Pole Piece Geometry

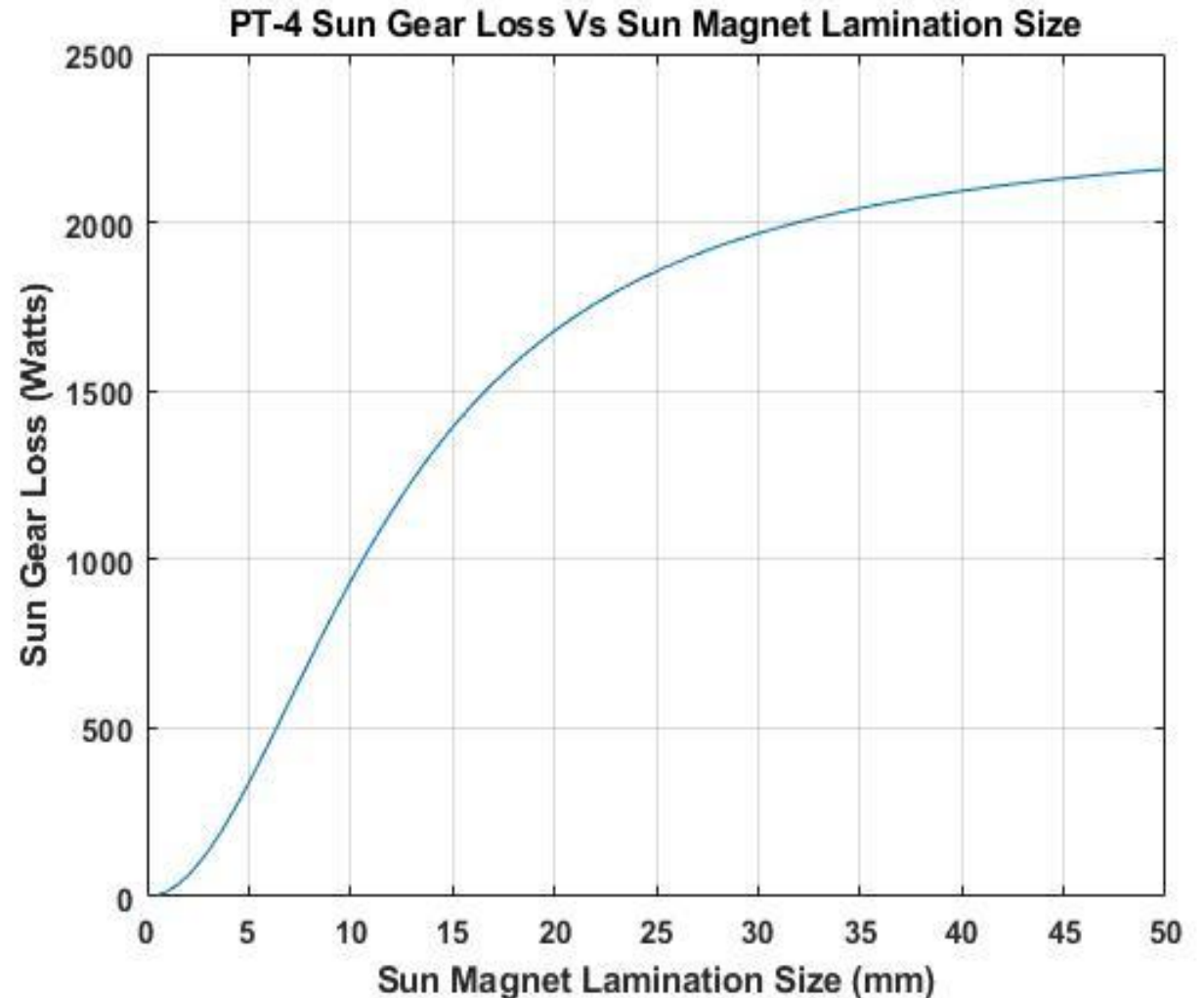
## Demagnetization of Ring Gear Magnet

- Ring gear uses 4-magnet Halbach Array
- 3.3 mm Ring Magnet Thickness
  - Prevents N-S Demagnetization
- 2 mm Ring Magnet Width
  - Allows E-W Demagnetization
- Fixed with Material Change
  - N48SH Ring Magnets



## Effect of Lamination Size on PT-4 Efficiency

- $P_c = \frac{1}{16} \frac{V}{\rho} \frac{w^2 l^2}{w^2 + l^2} \frac{1}{T} \int_0^T \left( \frac{dB}{dt} \right)^2 dt$
- Ring Magnet Width ~ 2 mm
  - $P_c \sim w^2$
- Sun Magnet Width ~ 12 mm
  - $P_c \sim l^2$
- For PT-4 selected
  - 2 mm Sun Laminations
  - No Ring Laminations

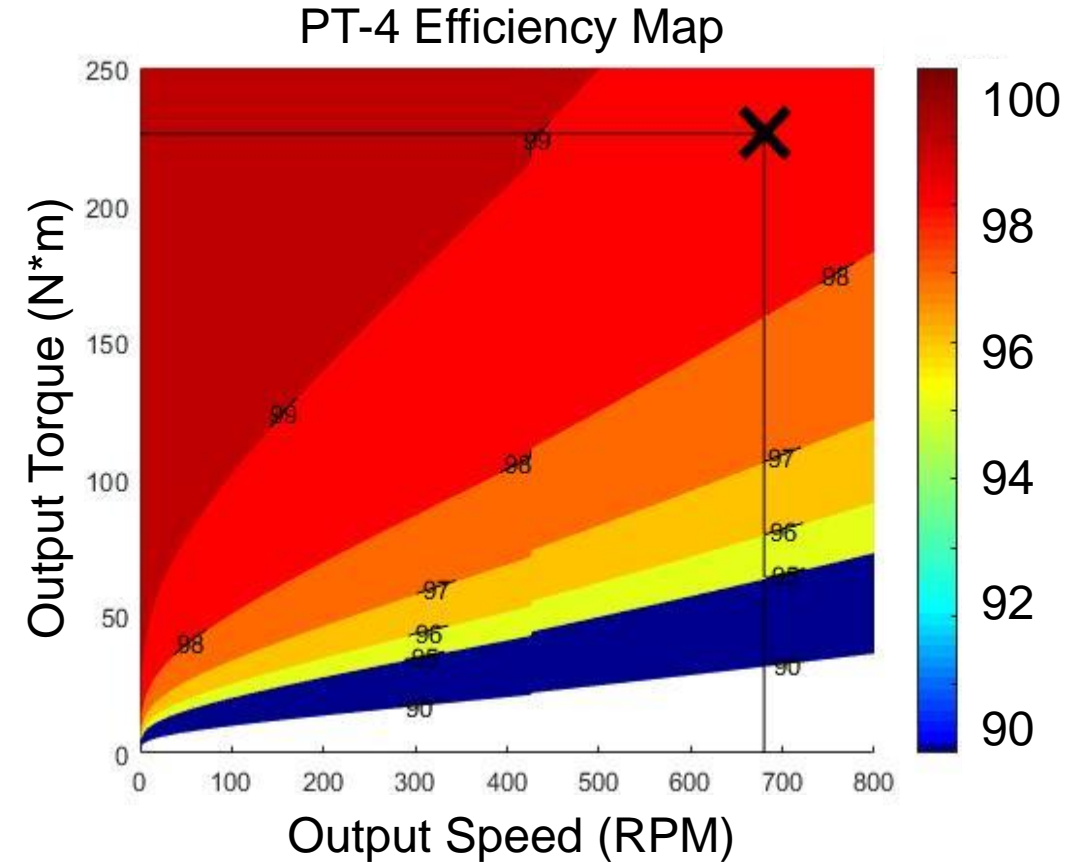




## Efficiency Analysis

- Included Analytical Predictions of Bearing and Windage Losses
- At 20 C ~ 98.5 %
- At 80 C ~ 99%

Temperature (°C)	Maximum 2D Output Torque (Nm)
20	370
80	303
100	270

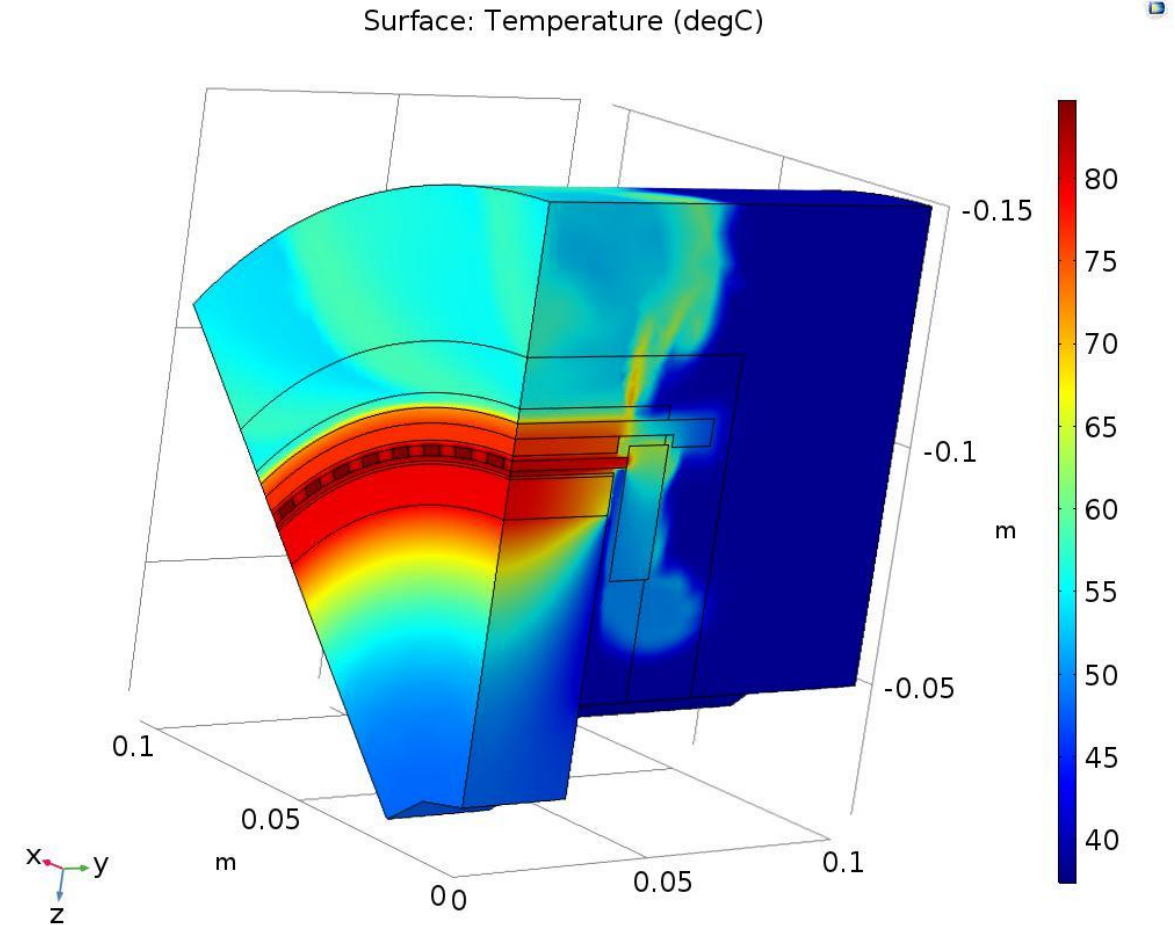


**Magnetic Gear Efficiency Improves as Temperature Increases, but Torque Capacity is Lost**

## Thermal Analysis

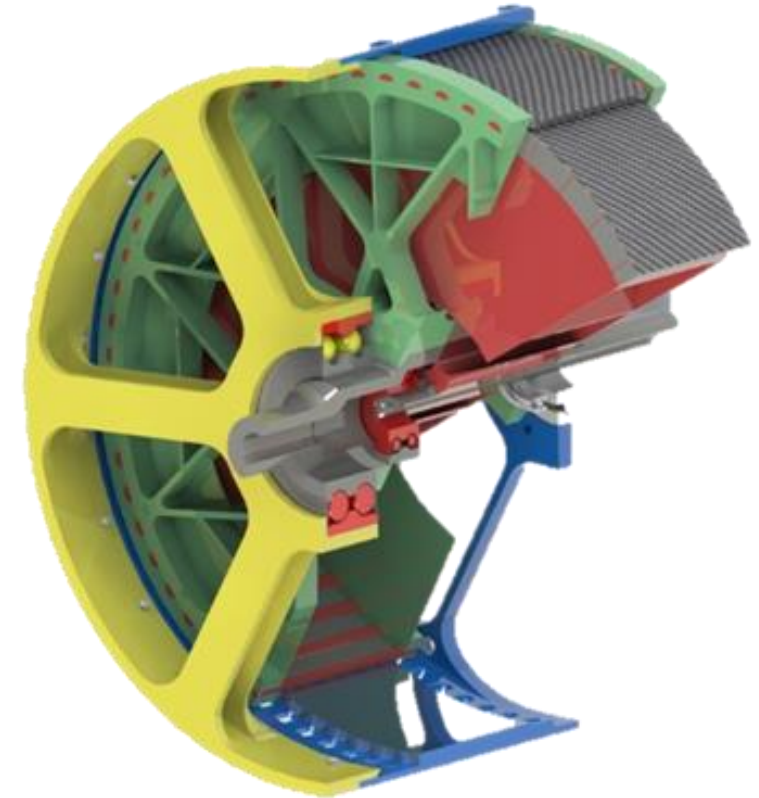
- Centripetally Pumped Cooling
  - Sun Gear tip speed = 50m/s
  - Self cooled
- Assumed 40C Ambient

Magnetic Component	Max Temperature (°C)
Sun Gear Magnets	80
Pole Pieces	85
Ring Gear Magnets	77



## Final Mechanical Design

- Final Mass= 4.6 Kg
- 8% higher than Design Code Prediction
  - Shaft Mass
  - Ring Gear Structure
- Carbon Hoop added to Modulator
  - Deflection neglected in code



# Conclusions

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- PT-4 Final Design
  - 49 Nm/kg at Nominal Operating Condition
    - Expected to be higher at 20C
  - >98.5% Efficiency
- Design Code Developed
  - Creates preliminary design in < 1 Day
  - Under predicted mass by ~8%
    - Some improvements needed

- Build and Test PT-4
- PT-5
  - Designed for X-57
  - Risk Reduction For PT-4
  - >97% efficiency without magnet laminations
- Update design code
- Magnetically geared motors
  - How best to share magnetic and structural components between a motor and a magnetic gear?

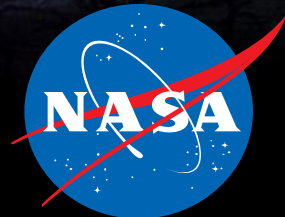


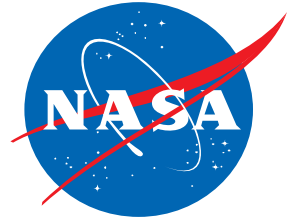
# Acknowledgements

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- NASA Revolutionary Vertical Lift Technology (RVLT) Project
- NASA Internal Research & Development (IRAD) Project
- Glenn Research Center Composites Group
  - Sandi Miller
  - Paula Heimann

# QUESTIONS ?

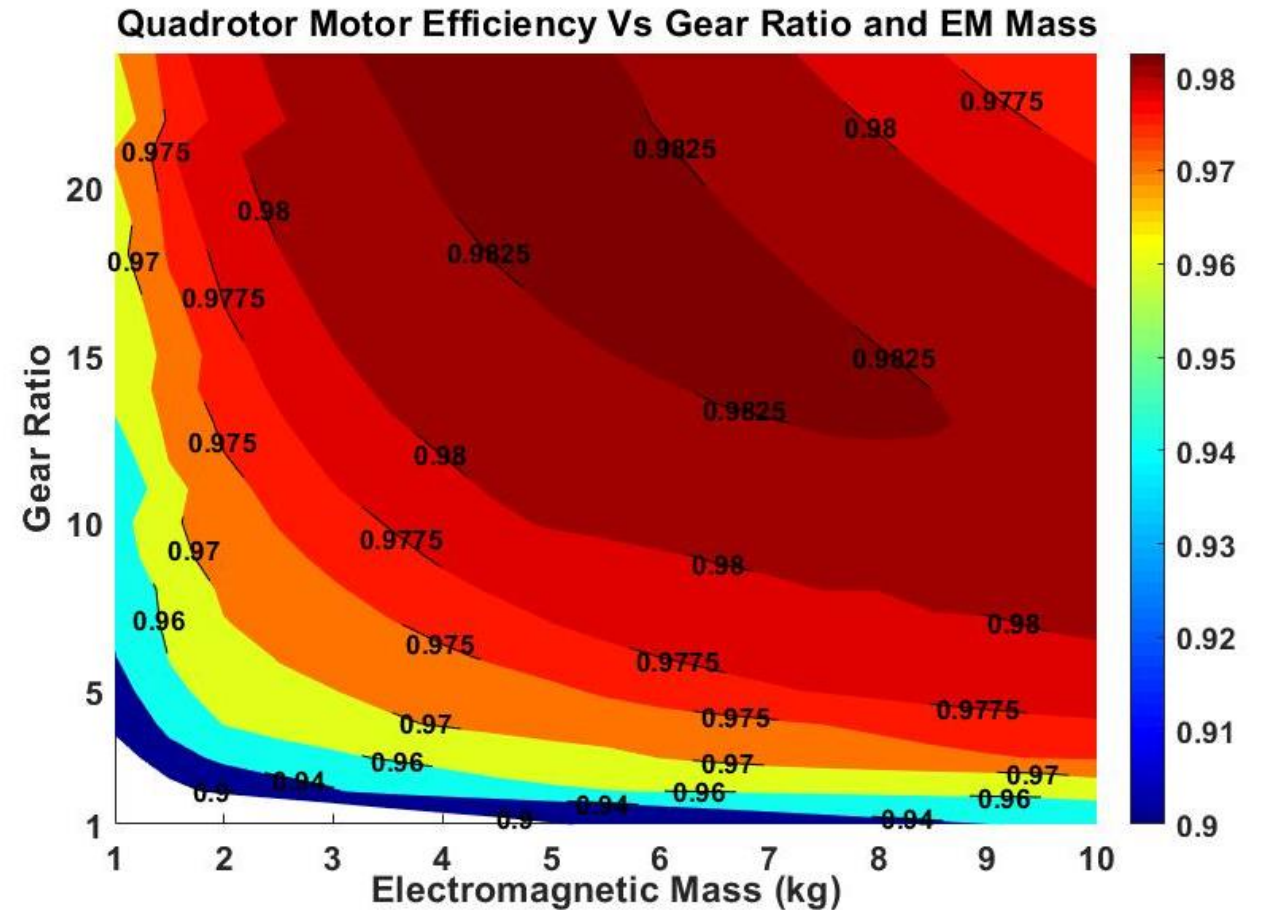




# Conclusions

## Effect of Gearbox on Motor for Quadrotor

- Fan Speed = 680 RPM
- Shaft Power = 16,100 Watts
- Analytical Equations for:
  - Torque
  - Copper Loss
  - Iron Loss
  - Windage Loss
- Losses under predicted
- No Thermal Considerations

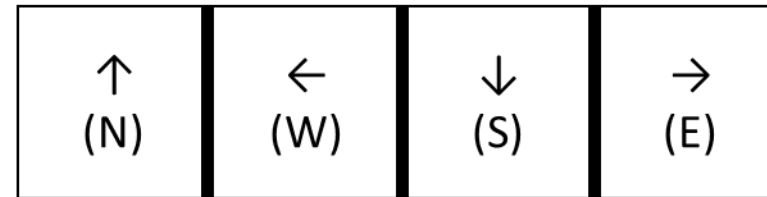


## Halbach Arrays

- Improve Specific Torque
  - Don't need back Iron
  - Higher Specific Flux than traditional array
- Improve efficiency
  - Lower Harmonic Distortion
  - More Magnets per pole
    - $P_c = \frac{1}{16} \frac{V}{\rho} \frac{w^2 l^2}{w^2 + l^2} \frac{1}{T} \int_0^T \left( \frac{dB}{dt} \right)^2 dt$
    - $w \ll l$
    - $P_c \sim w^2$

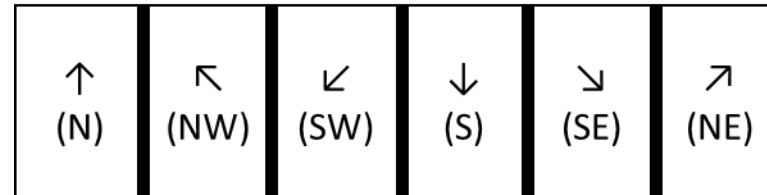
One Pole Pair

Working Face

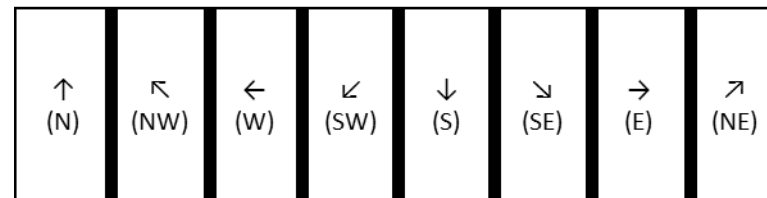


4 Magnets  
Per Pole Pair

Back Face



6 Magnets  
Per Pole Pair



8 Magnets  
Per Pole Pair

Arrows denote magnetization direction\*



## Minimize Modulator Thickness

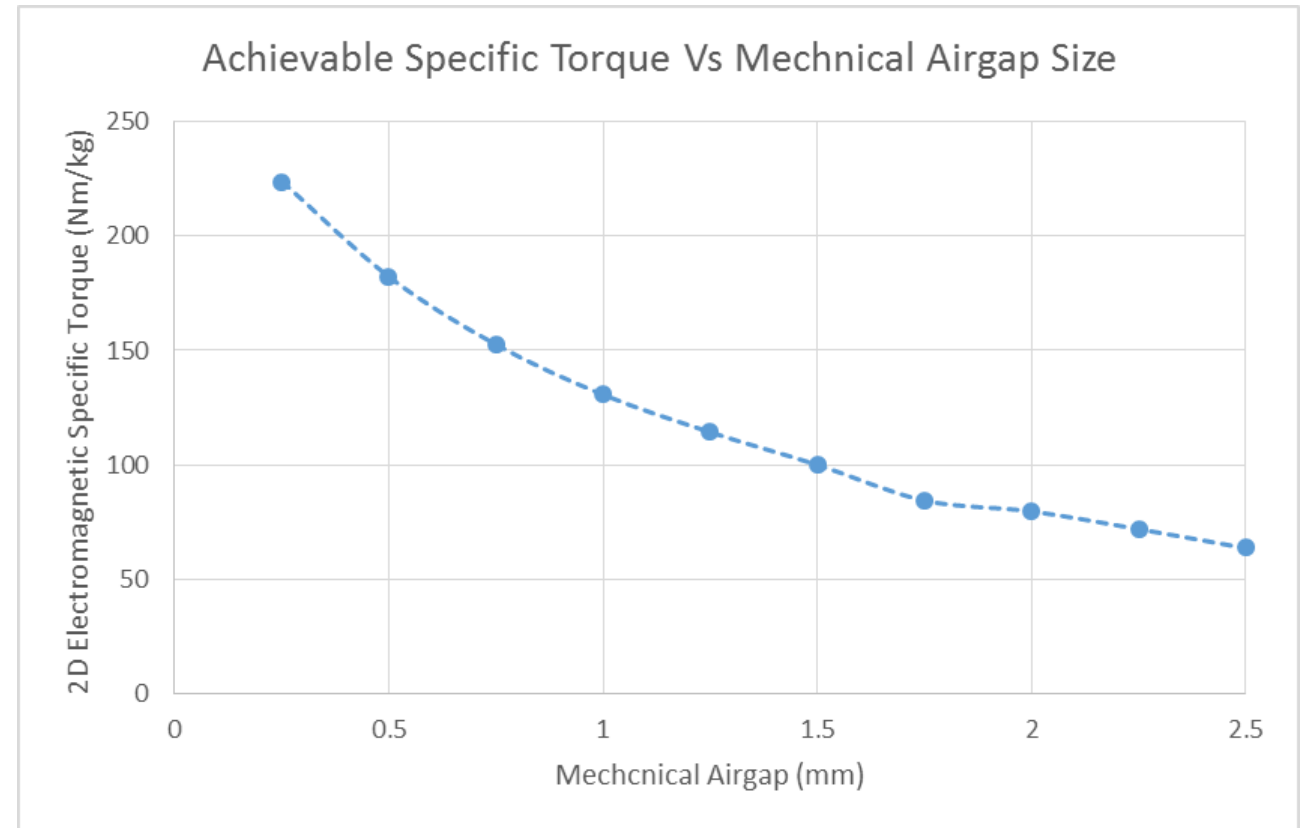
- Electromagnetically there is a specific torque optimum modulator thickness
- That thickness is typically less than what can be achieved mechanically.
  - Subjected to high magnetic forces
    - Gear's output torque
    - Radial force
  - Modulator is sandwiched by airgaps
  - Pole pieces do not provide structure
    - Point of failure in PT-2 and PT-3





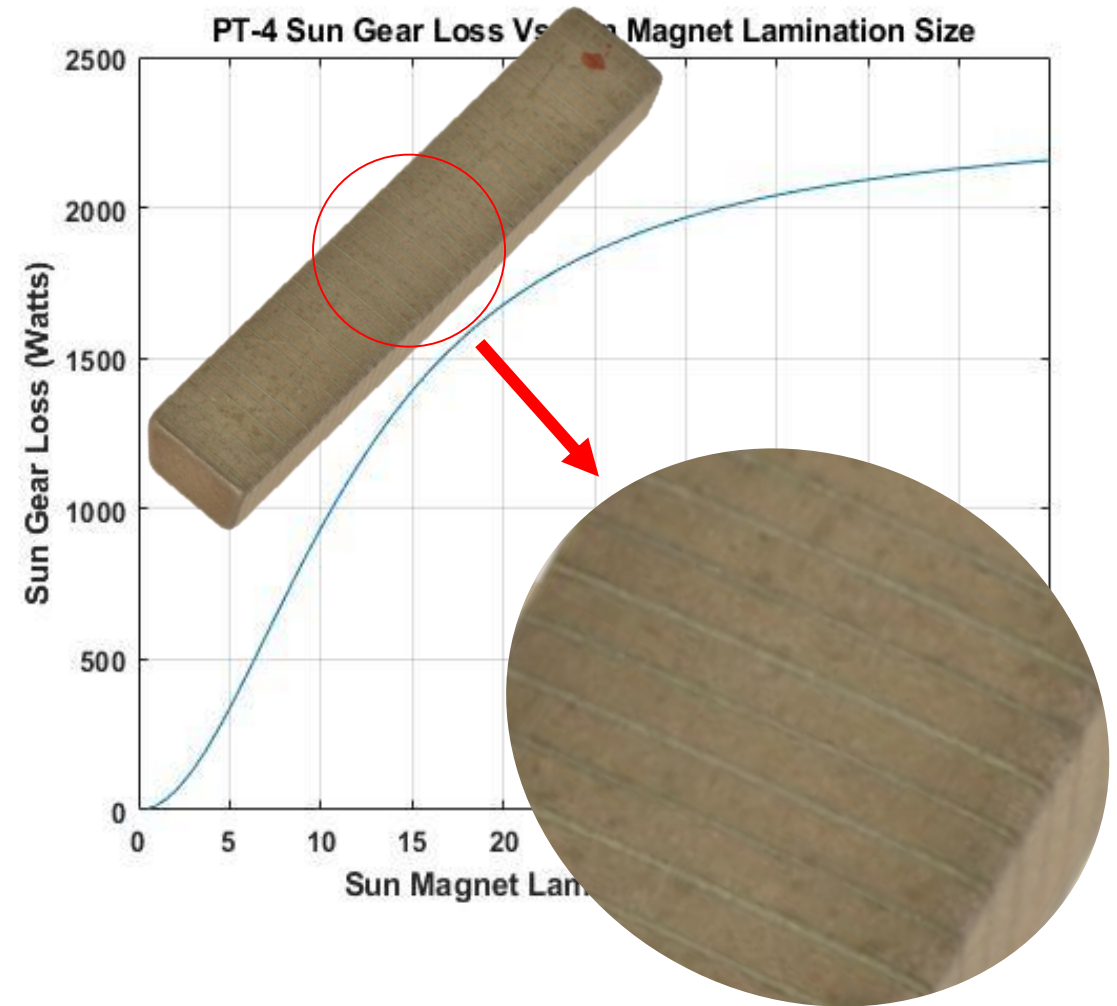
## Minimizing Airgap

- Less Reluctance Between Rotors
  - Lowers optimum modulator thickness
- Less Pole to Pole Leakage
  - Increases optimum rotor pole counts
  - Lower reluctance between poles
  - Lowers optimum modulator thickness
- Reduces Efficiency
  - More unmodulated flux crosses airgaps



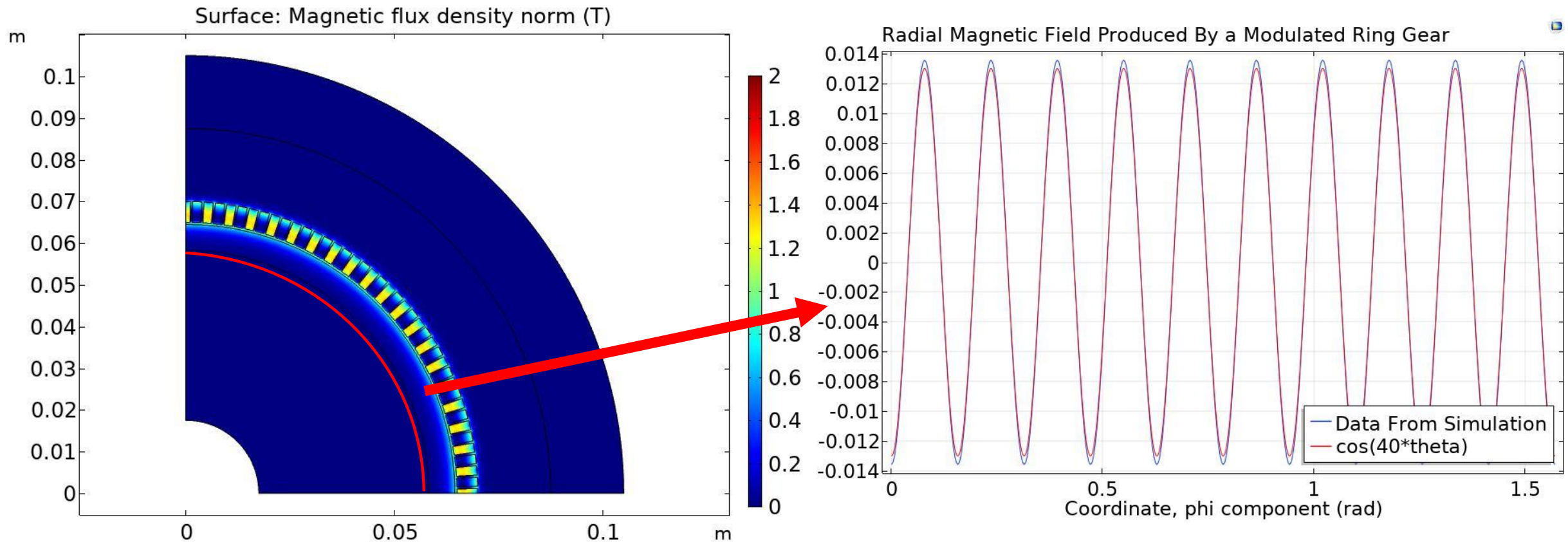
## Magnet Laminations

- Enables >99% Efficiency
  - $P_c = \frac{1}{16} \frac{V}{\rho} \frac{w^2 l^2}{w^2 + l^2} \frac{1}{T} \int_0^T \left( \frac{dB}{dt} \right)^2 dt$
  - $l \ll w$
  - $P_c \sim l^2$
- Magnet Fill Percentage Decreases
  - Lowers Torque
- PT-3
  - 1mm Laminations
  - >98% Efficiency
  - ~80% magnet fill



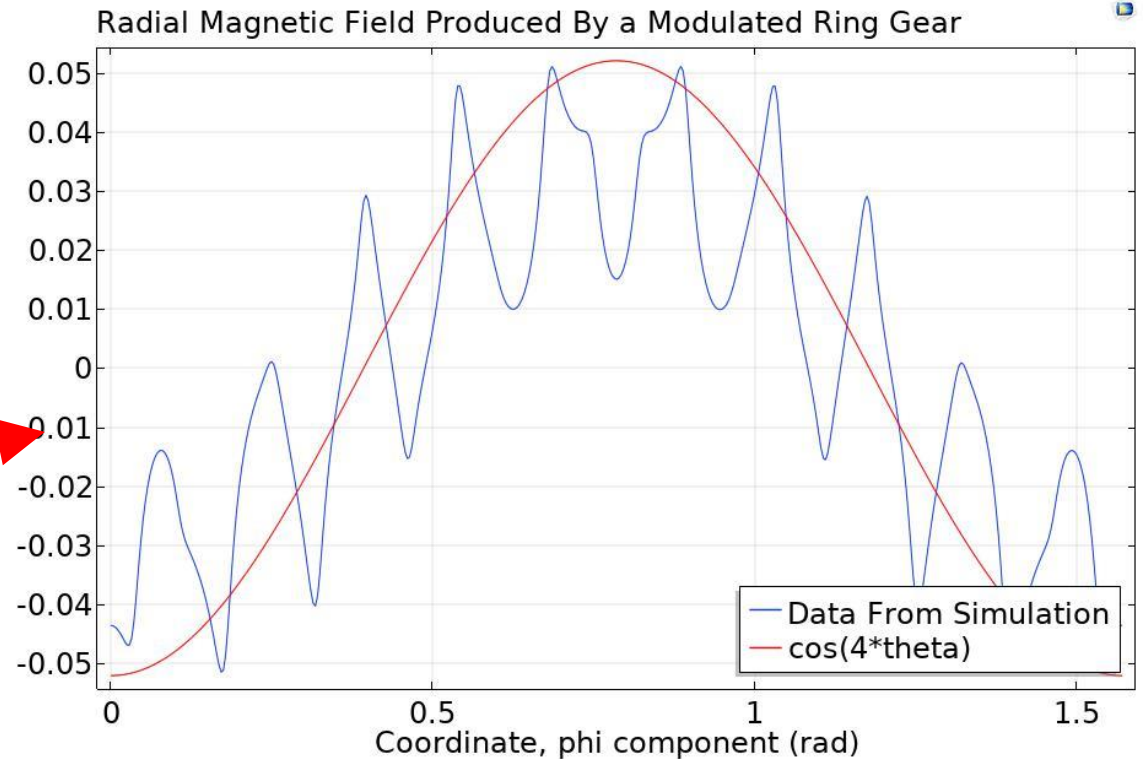
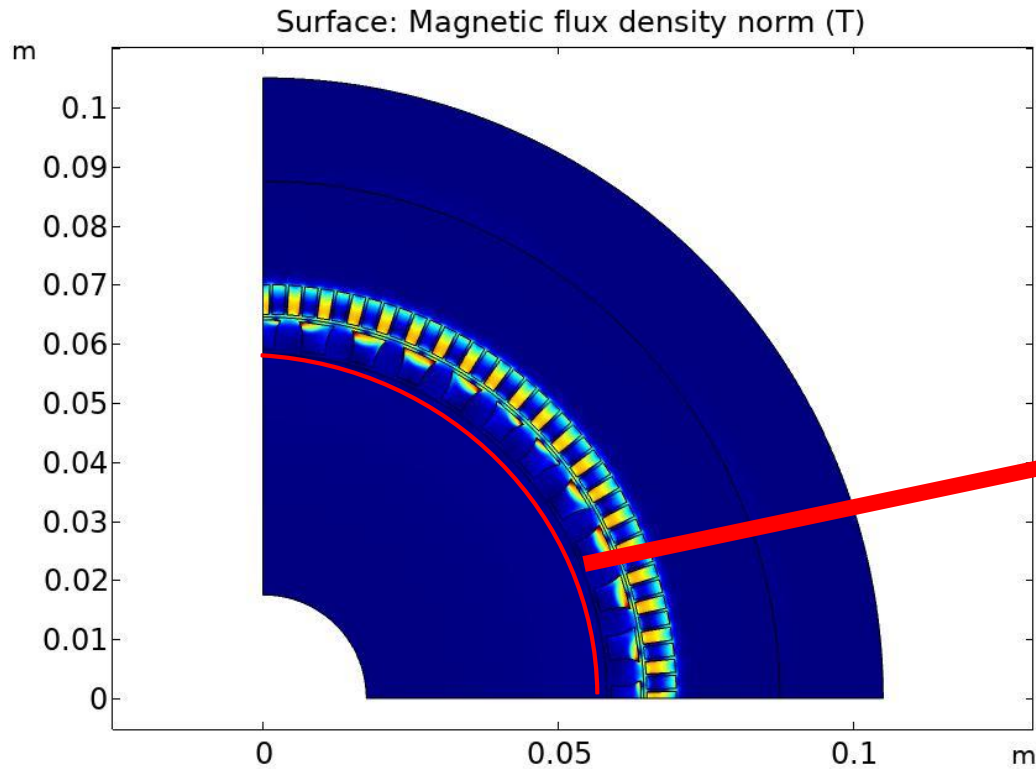
# Concentric Magnetic Gears

## Flux Modulation Example: 10 Pole Pair Ring Gear Only



# Concentric Magnetic Gears

## Flux Modulation Example: Add 11 Pole Piece Modulator



$$PS = Q - PR = 11 - 10 = 1$$