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Development of Lyocell based Phenolic Impregnated Carbon Ablator (PICA-D) for Future NASA Missions

PICA-D and Three Exciting NASA missions

Matt Gasch, Mairead Stackpoole, Ethiraj Venkatapathy, Don Ellerby, Kristina Skokova, Frank Milos, Keith Peterson, Dinesh Prabhu and Greg Gonzales
NASA Ames Research Center, Moffett Field, CA 94035

Steve Violette and Taylor Franklin
Fiber Materials Inc, Biddeford, ME 04005

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Outline



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- **Background**
 - Heritage PICA
 - PICA Sustainability Challenge
- **Sustainability**
 - Lyocell: An alternative precursor to rayon
 - PICA Manufacturing
- **Development of PICA-D**
 - Establish PICA-D as a drop-in replacement for Heritage PICA
 - PICA-D Arc Jet Campaign
 - Lyocell Fiberform/PICA-D Billet and Near-Net-Shape Single-Piece Aeroshell Processing
- **PICA-D and Three Exciting NASA missions**
- **Summary**
- **Acknowledgements**

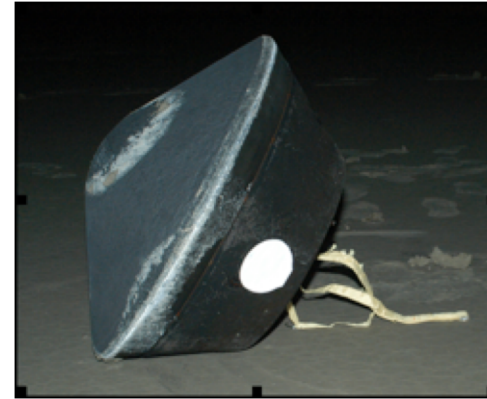
Background – PICA



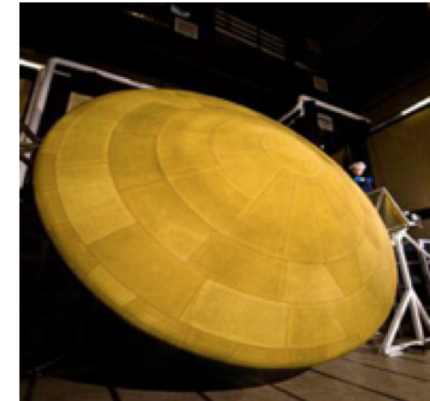
State of the Art Low Density Carbon Phenolic Ablators

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- **Phenolic Impregnated Carbon Ablator (PICA)**
 - First used as forebody single piece heatshield for Stardust
- **Low density coupled with efficient ablative capability at medium-high heat fluxes**
- **Since Stardust-**
 - Under the Orion program PICA was shown to be capable for both ISS and lunar return missions but was not selected as the baseline TPS
 - PICA was transitioned to Mars Science Lab (MSL) post CDR in a tiled configuration when the mission environments went beyond the capabilities of SLA561V
 - OSIRIS-REx sample return capsule as a single piece
 - On Dec. 3, 2018, the OSIRIS-REx spacecraft arrived at its target, near-Earth asteroid Benu



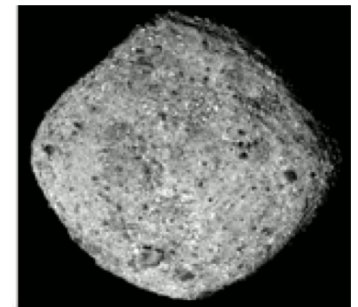
Stardust forebody TPS.
(~0.8m diameter)



MSL Heat Shield
(4.5m diameter)



OSIRIS-REx forebody
TPS. (~0.8m diameter)



Benu taken by the OSIRIS-REx spacecraft from a distance of ~ 50 miles
<https://www.nasa.gov/osiris-rex>

Challenges with PICA Sustainability



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- In 2016 NASA learned that the “heritage” rayon used in PICA was ceasing production, leading to a flight-qualified PICA sustainability concern
 - The carbon fiber precursor for PICA has become obsolete twice since the material was developed and used on Stardust, so a secure source is essential to maintain PICA capabilities for future missions
- In FY16/17, NASA ARC was funded by the Planetary Science Division of the Science Mission Directorate to address PICA rayon sustainability
- Lyocell Based PICA (PICA-D) was manufactured and limited testing performed – initial results indicate Lyocell is a good candidate as a potential replacement for heritage rayon

Mission/ Project	Precursor Type	Rayon Sustainability	Changes /Updates to PICA
Stardust Near Net Shape 0.8m	Liberty rayon	US source – production ceased in the 90s	Developing process to fabricate NNS within the project density specification required
Orion CEV ADP Billets	Multiple sources – settled on SINACE	Multiple international sources evaluated	Optimized densification process for billets, tested the bounds of the density specification and the influence on performance / properties
MSL Billets	SINACE rayon	International source – production ceased in ~ 2017	Leveraged ADP data to allow use on MSL
OSIRIS Rex Near Net Shape 0.8m	SINACE rayon	International source – production ceased in ~ 2017	Spec tightened over Stardust for NNS casting range . Phenolic adjustments based on lessons learned from ADP/MSL
M2020 Billets	SINACE rayon – source depleted	International source – production ceased in ~ 2017	Leveraged MSL
PICA-D Development Billets	Lyocell w/ Ti Dulling Agent	Domestic/international sister plants. Greener processing	MSL specification range – Eliminated use of “re-grind”
PICA-D Development Near Net Shape 0.8m	Lyocell w/ Ti Dulling Agent	Domestic/international sister plants. Greener processing	Leveraged OSIRIS REx/MSL – No re-grind
PICA-D Development Billets	Lyocell	Domestic/international sister plants. Greener processing	MSL/M2020 specification range – No re-grind
PICA-D Development Near Net Shape 1.5m	Lyocell	Domestic/international sister plants. Greener processing	Leveraged OSIRIS REx/MSL – No re-grind

Lyocell – A Sustainable Precursor



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- Traditional rayon manufactured from wood pulp involves many steps and the conversion of wood pulp into rayon or regenerated cellulose results in toxic byproducts
 - rayon manufacturing was discontinued and is no longer a viable process in the US and Europe
- Lyocell - solvent spinning technique is simpler and more environmentally sound
 - uses a non-toxic solvent chemical that is 99% recycled in the manufacturing process
- Lenzing – sister factories in US, Austria and UK able to provide the same Lyocell precursor – multiple supply routes alleviate future sustainability concern

Refer to below links if interested in information on how fibers are made from wood pulp:

<https://www.youtube.com/watch?v=tHdJGFv99fE>

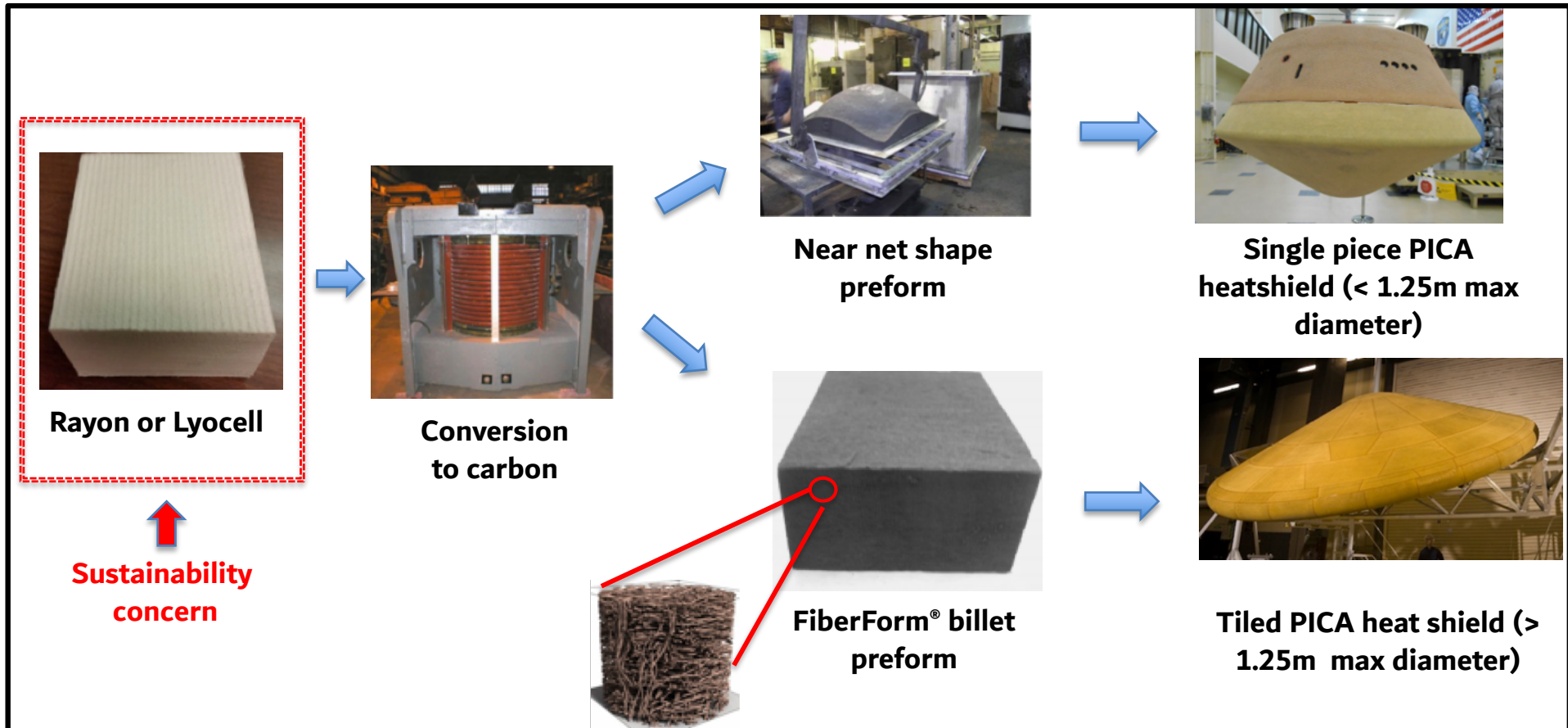
<https://www.youtube.com/watch?v=14PZNgRoEUM>

PICA Manufacturing Overview

Role of Rayon/Lyocell in PICA Manufacturing



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- Chopped, graphitized rayon or Lyocell - based carbon fiber slurry-cast into either block (billet) or single piece heatshield preforms
- Single piece cast heatshields have fiber oriented to optimize through-thickness thermal conductivity
- Lightweight phenolic sol-gel matrix is infiltrated into preform

Establishment of PICA-D as a Replacement for Heritage PICA



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- **FY17 - SMD-PSD funded NASA Ames to manufacture & perform limited property/aerothermal characterization of Lyocell-based PICA (PICA-D)**
 - Fiber Processing, billet fabrication, single piece heatshield preform fabrication, conversion to PICA (billets and single piece preform)
 - PICA property testing and arc jet testing
- **FY17 task with limited testing indicated PICA-D had the potential to be a replacement for heritage PICA**
- **FY18/FY19+ – further efforts to characterize and extend the capability of PICA-D and establish Lyocell PICA as a replacement for heritage PICA**
 - Establishing PICA-D as a “drop in replacement” will allow missions to depend on and design with PICA-D without having to address further sustainability risks.
 - Establishing extended capability of PICA-D will allow Sample Return Missions with higher entry speed that were not considered before.
 - Extended operational capability
 - Extended single piece heatshield manufacturing

PICA-D Property Testing



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- Coupon level testing of PICA-D provides an initial material screening that will:
 - Allow for identifying differences with heritage PICA
- Structural testing objectives:
 - Determine moduli, strength, and thermal expansion over a range of temperatures.
 - Develop and deliver a preliminary material property database that can be used in future mission analyses
 - Identify any differences in material properties and allowables between PICA-D and heritage PICA
- Thermal properties test objectives:
 - Determine in-plane and through-the-thickness thermal conductivity over a range of temperatures
 - Identify any differences in thermal conductivity between PICA-D and heritage PICA

Test Type	Virgin Property Testing			Fiberform Property Testing
	-250°F	RT	350°F	RT
IP tension	3	10	3	4
TTT Tension	3	10	3	4
IP Compression	3	3	3	
TTT Compression	3	3	3	
IP Shear	3	3	3	4
Transverse Shear	3	10	3	4
IP CTE	2			2
TTT CTE	2			2

Structural Test: Matrix 1

Test Type	PICA-D Char		FiberForm	
	3300°F	4500°F	3300°F	4500°F
TTT Compression	3	3	3	3

High-Temp Structural Test: Matrix 2

Test Type	Virgin Property Testing	
	100°F	350°F
TTT Conductivity	2	2
IP Conductivity	2	2

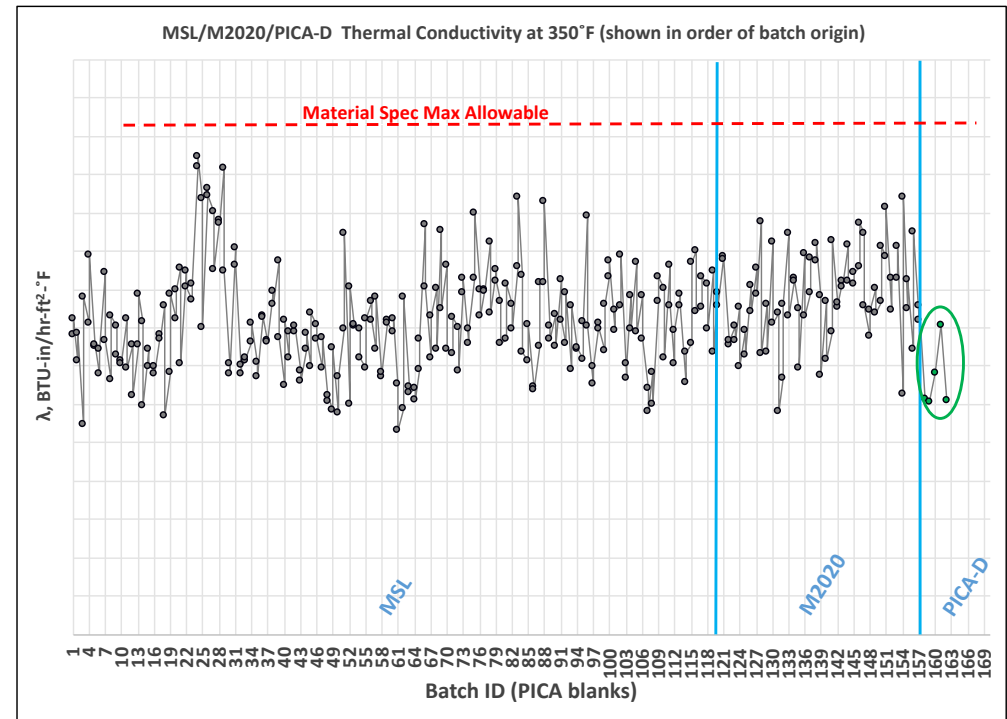
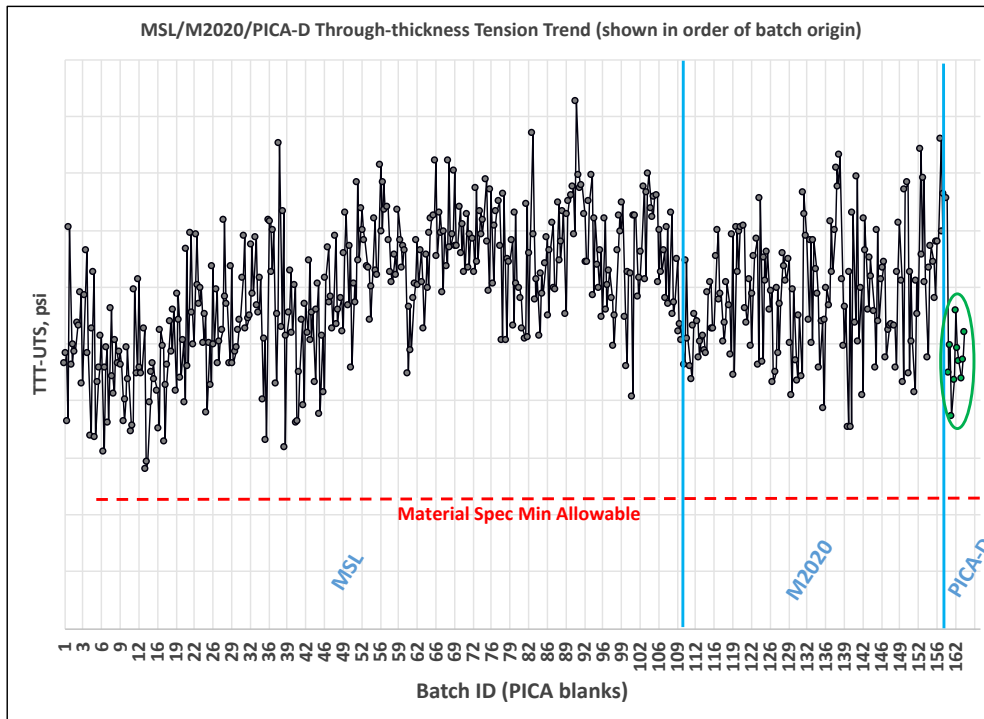
Thermal Test Matrix



PICA-D Property Testing

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- PICA lot acceptance test (LAT) results: in-plane (IP) tension, through-thickness (TT) tension, and through thickness thermal conductivity at 100F and 350F
- The data was compared to Mars Science Laboratory (MSL) and Mars 2020 production LAT results


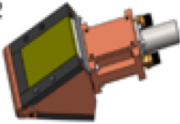
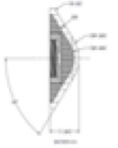


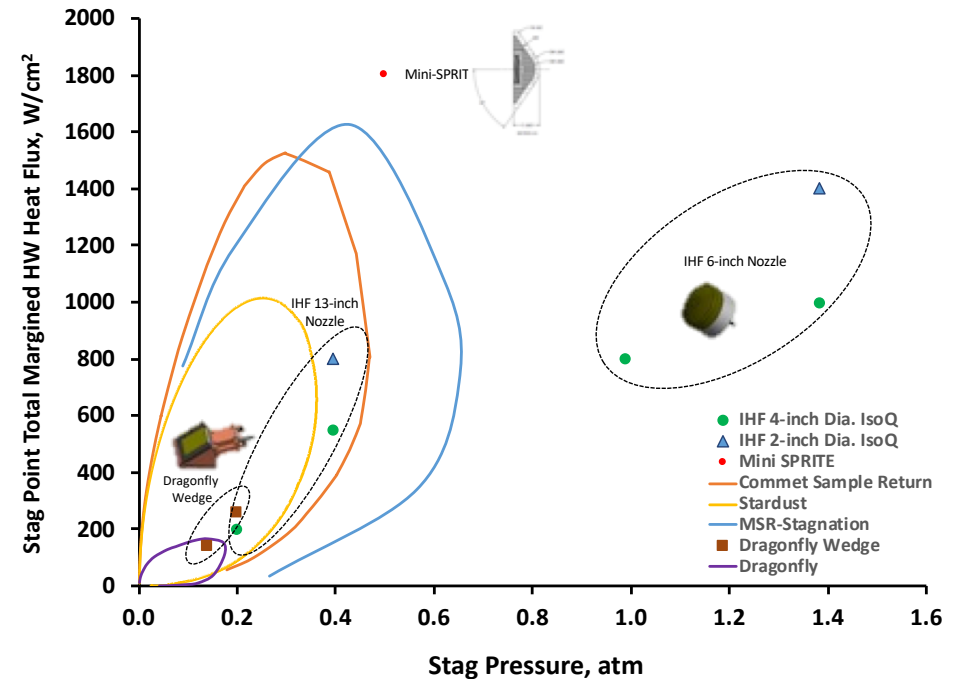
PICA-D LAT results are comparable to Heritage PICA and within the expected scatter indicating potential updates to Thermal and/or Structural performance models are not required

Test Campaign to Establish/Extend Capabilities



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<p>1</p> 	<p>Drop-in replacement testing on 4in and 2in diameter stagnation models</p> <ul style="list-style-type: none"> Objective: provide experimental evidence that PICA-D could be used with high confidence as a TPS material on variety of missions Testing from ~ 200 W/cm² up to ~ 1500 W/cm² (cold wall) heat flux and 1.4 atm pressure AHF and IHF 6in nozzle tests Compare recession and in-depth temperature data with FIAT predictions Some, but limited, comparison to heritage (MSL) PICA is also planned
<p>2</p> 	<p>Exploring PICA-D performance limits through testing at extreme environments</p> <ul style="list-style-type: none"> Objective: expanding PICA limits and thermal performance modeling capability TBD condition, possibly ~ 1000 W/cm², 0.8 atm pressure, 1400 Pa shear, wedge test model Testing at AEDC H2S facility
<p>3</p> 	<p>Exploring PICA-D performance limits through new test article design</p> <ul style="list-style-type: none"> Objective: expanding PICA limits and thermal performance modeling capability Mini-SPRITE test model: ~2200 W/cm² (cold wall) stagnation heating, 500 Pa shear on the flank IHF 6in nozzle test (limited number of coupons)



Arc jet campaign objectives

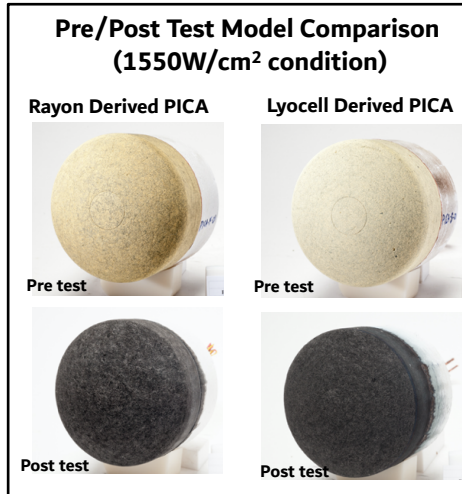
- Compare the thermal response and recession behavior of Lyocell derived PICA to rayon derived PICA
- Initial look at any performance differences or off-nominal behavior in PICA-D
- Establishing the extended capability of PICA-D will allow Sample Return Missions with higher entry speed that were not considered before



PICA-D Arc Jet Testing

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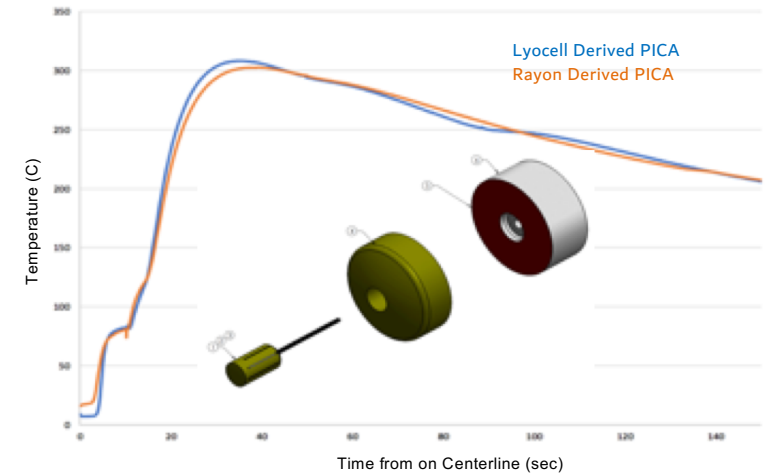
Runs in Air 2017



Recession Comparison

Material	Average centerline recession (1550W/cm ² and 1.3 atm)	Average centerline recession (400W/cm ² and 0.3atm)	Average centerline recession (220W/cm ² and 0.08atm)
Lyocell PICA	4.0mm	6.02mm	3.79mm
Rayon PICA	4.2mm	5.97mm	3.89mm

Example in-depth TC Trace @ 1550W/cm²



Run condition very relevant for proposers considering PICA as a forebody or backshell material

Runs in N₂ 2018/19

Heat Flux = 140 W/cm²,
Pressure= 14 kPa

Heat Flux = 260 W/cm²,
Pressure= 19 kPa



- Previous testing of PICA with RTV seams was only done in air under MSL and Orion programs
- In support of Dragonfly Phase A study, PICA-D built 2 wedge shear models with RTV seams for testing in a nitrogen environment

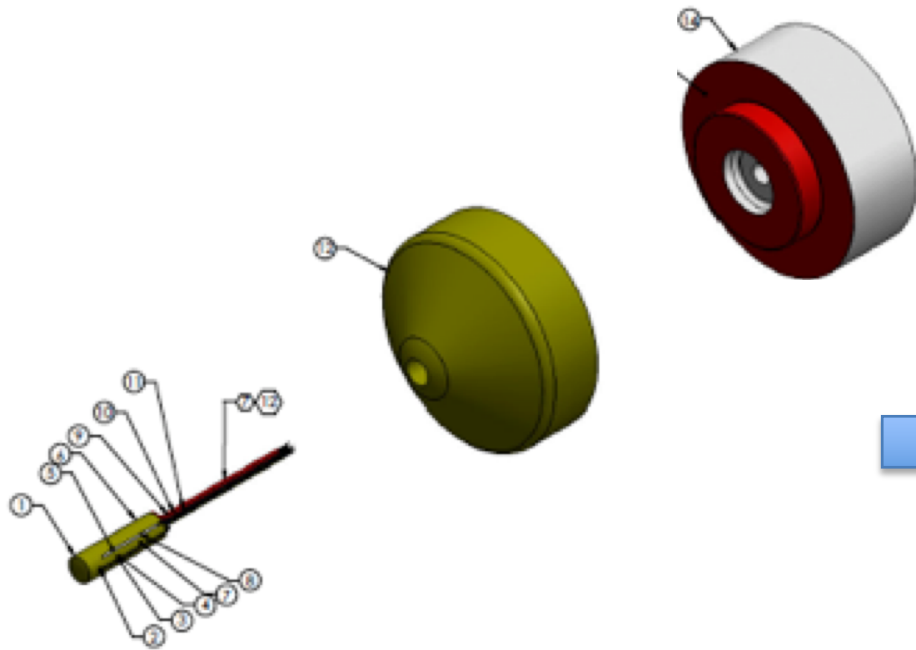
For a Given Test Condition (Same Run Time) Initial Results Indicate that Recession and In-depth Temperature Between a Lyocell-Derived PICA and a Heritage Rayon-Derived PICA are Comparable, in Both Oxygen and Nitrogen.

PICA-D Mini-SPRITE Arc Jet Testing

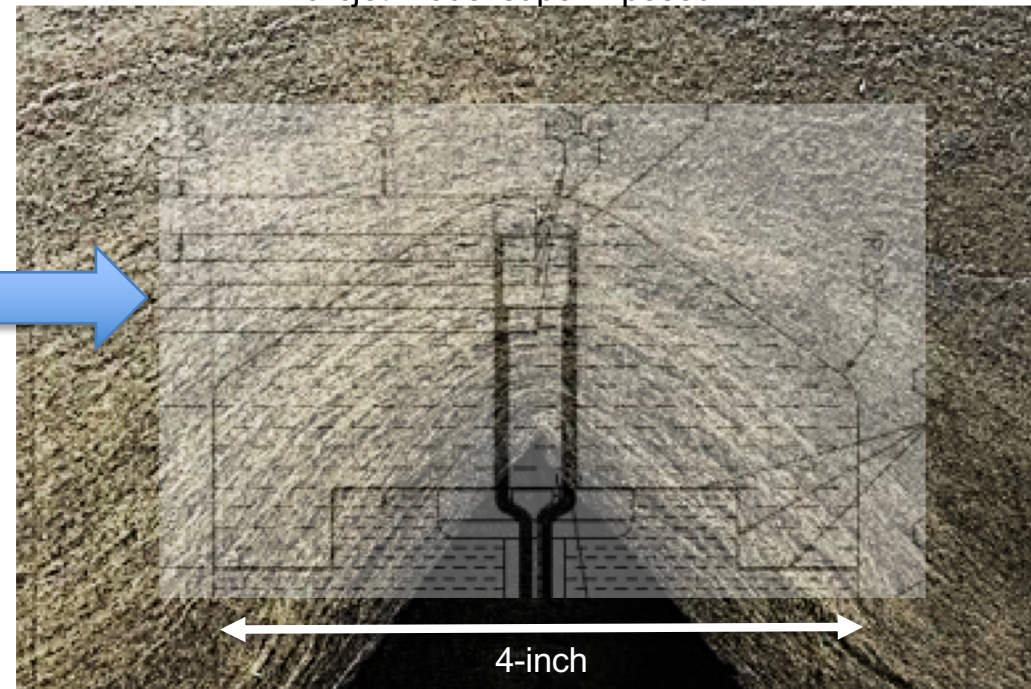


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- Exploring PICA-D performance limits through new test article design
 - Objective: expanding PICA limits and thermal performance modeling capability
 - Mini-SPRITE test model: $\sim 2200 \text{ W/cm}^2$ (cold wall) stagnation heating, 500 Pa shear on the flank
 - 4-inch diameter coupons



Mini-SPRITE Test Casting with
arcjet model superimposed

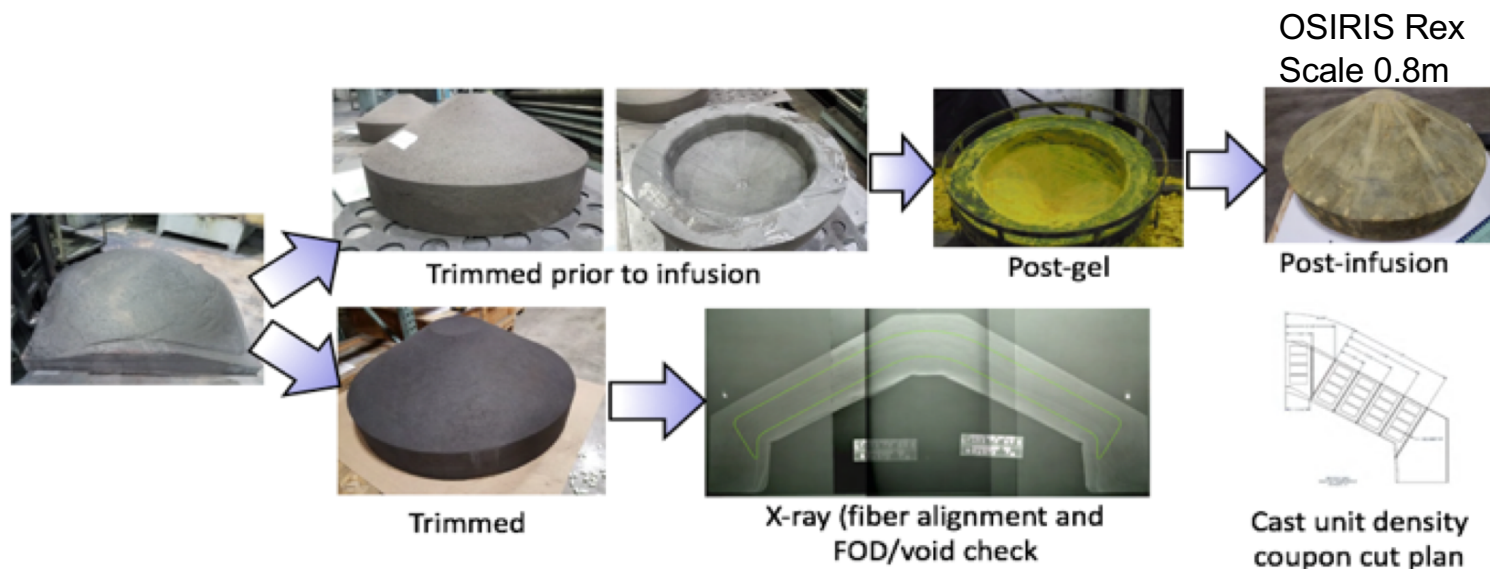


Lyocell Fiberform/PICA Billet and Near Net Shape Cast Processing



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- Fabricated 3 near-net-shaped Fiberform heatshield blanks (OSIRIS REx scale) in FY17
- Fabricated 4 near-net-shaped ~ 1.5m single piece FiberForm castings (FY18/19)
 - Non Destructive Evaluation (NDE) on the near net shape Fiberform unit to evaluate fiber alignment indicated good alignment from nose to shoulder
 - Density
 - Converted one into 1.4 m PICA heatshield
- Significant number of lessons learned captured/implemented and substantial risk reduction achieved



PICA-D and Three Exciting Future NASA missions



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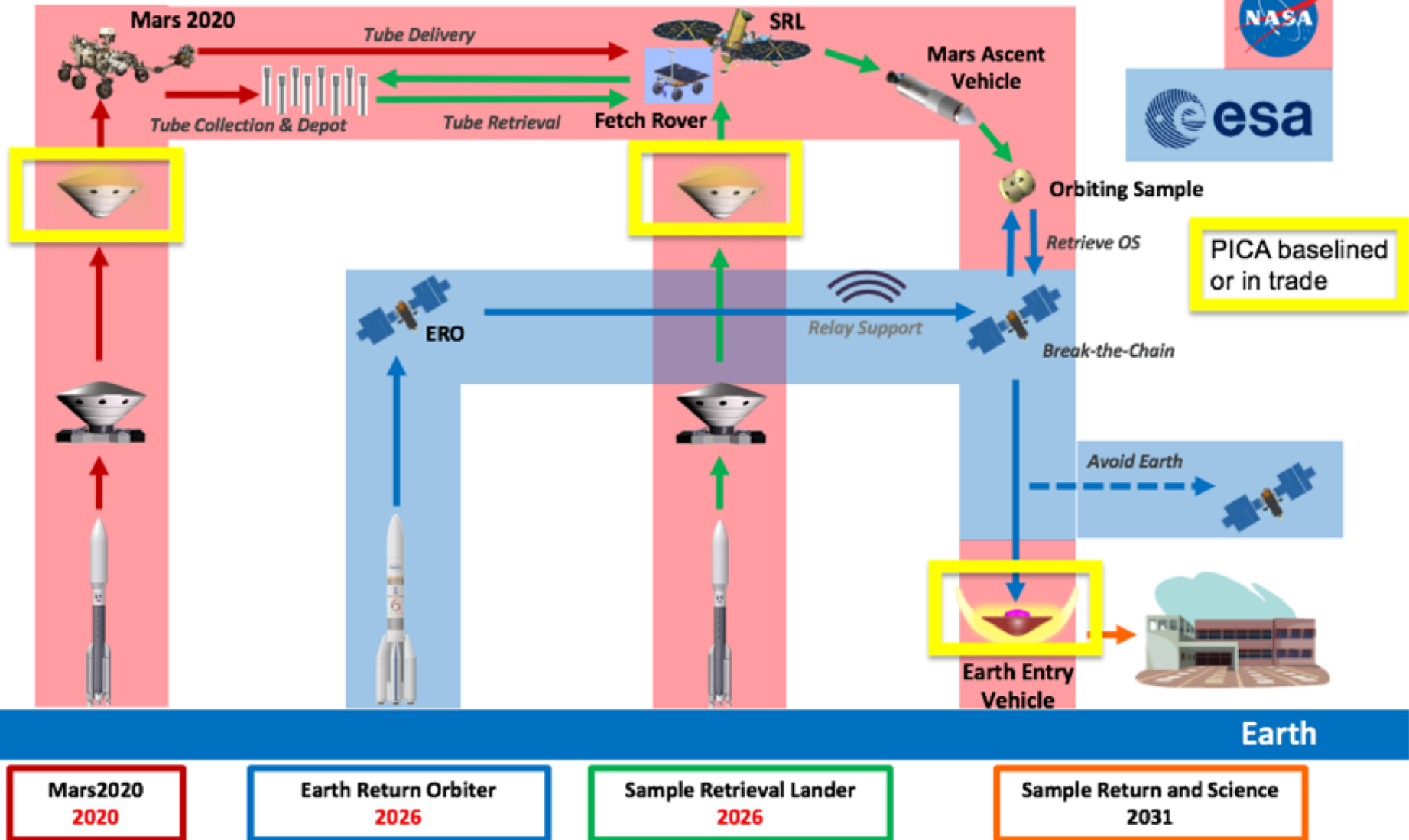
- Mars Sample Return **Sample Return Lander** (MSR SRL)
- Mars Sample Return **Earth Entry Vehicle** (MSR EEV)
- Dragonfly

Mission timelines highlight why a long-term sustainable PICA TPS option is needed

Mars Sample Return Campaign



Mars



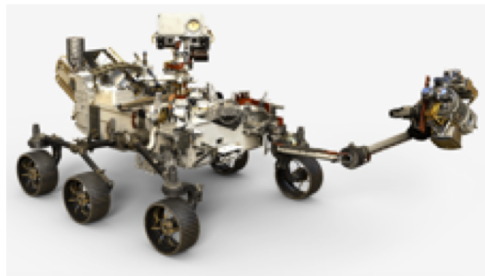
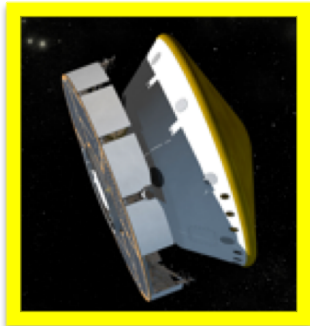
Mars2020 and Sample Retrieval Lander (SRL)



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M2020 Mission

- 4.5m tiled configuration
- Last Sniace rayon mission



Rover with science instruments will collect samples from the Martian surface and deposit tubes filled with samples at location for sample fetch rover to collect

- MSL “built to print” heatshield

Courtesy: Jet Propulsion Laboratory

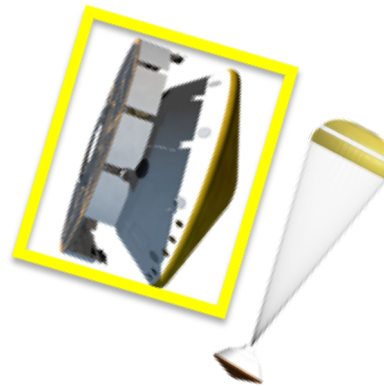
SRL Mission

Platform

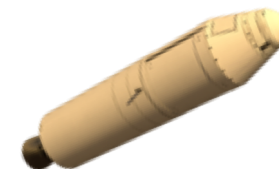


- 5+m tiled configuration
- Lyocell derived PICA

Cruise & EDL



Mars Ascent Vehicle

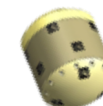


Sample Fetch Rover

(JPL reference concept)



Orbiting Sample

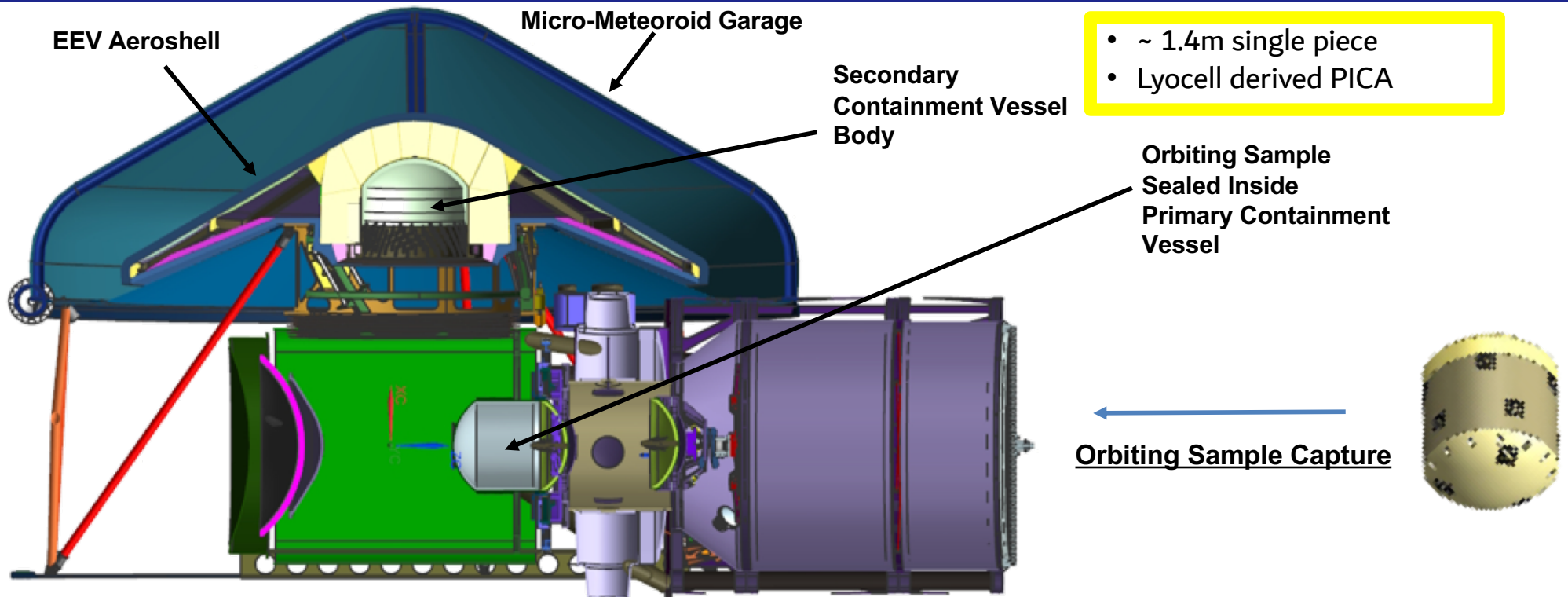


- Vehicle has a different shape than Mars 2020
- Entry environment in family with Mars 2020/ MSL

MSR Earth Entry Vehicle (EEV)



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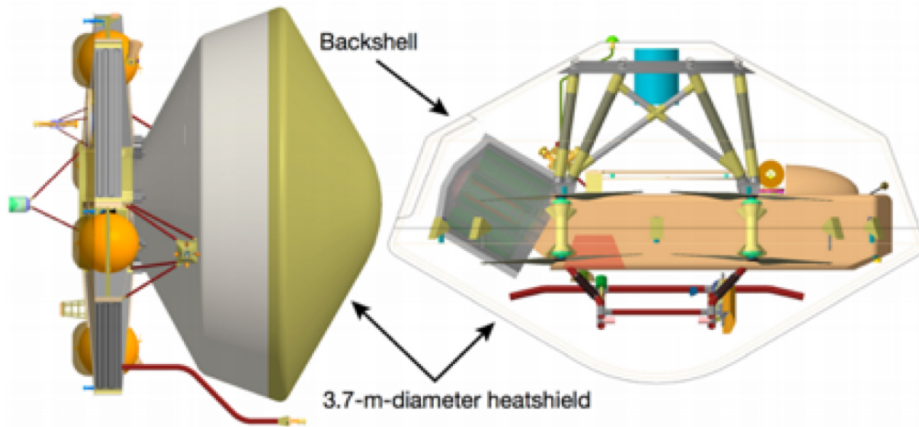
- Single piece PICA-D considered for MSR EEV forebody
 - Requirements for MSR EEV – up to 1.4m which is greater than Stardust (0.8m)
 - Leveraging PICA-D for manufacturing scale-up demonstration
- Tiled backshell PICA baselined
- Expected conditions ($\sim 1600 \text{ W/cm}^2$, $\sim 80 \text{ kPa}$, $\sim 1000 \text{ Pa}$ shear)
- PICA performance beyond thermal needs to be assessed - potential MMOD impact and ground impact

Dragonfly

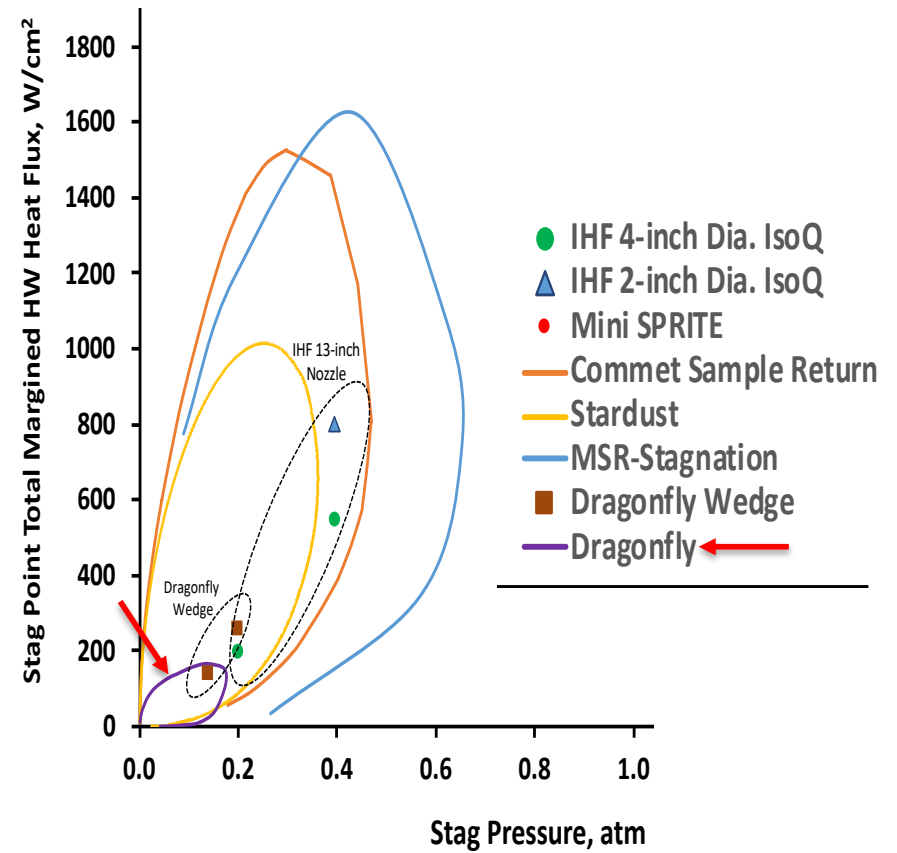


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- Mobile robotic rotorcraft lander to Titan,
- Study prebiotic chemistry and extraterrestrial habitability at various locations
- Perform vertical-takeoff and landings



- Dragonfly forebody TPS (~4 m diameter)
- Currently NF – Phase A
- Benign environment for PICA
- Leverage MSL integration approach



- 4+m tiled configuration
- Lyocell derived PICA

Summary



- **PICA has become a workhorse TPS for NASA and sustainment is essential**
- **NASA ARC / FMI have been and will continue to work together and address any PICA sustainability concerns**
 - **Lyocell, a viable precursor, will be available for decades**
- **Lyocell Based PICA (PICA-D) was manufactured and testing performed to-date show it to be a viable replacement for heritage rayon**
- **Establishing the extended capability of PICA-D, scale-up to 1.4m, will allow Sample Return Missions with higher entry speeds and larger payload in the future including Mars Sample Return Mission**
- **Exciting future NASA missions need PICA (SRL, MSR EEV and Dragonfly) and NASA TPS sustainability effort will have a payoff for these missions**

Acknowledgement



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- PICA sustainability activities are funded by NASA's Planetary Science Division of the Science Mission Directorate