

Variation in Ion Acceleration Characteristics of the HERMeS Hall Thruster during Magnetic Optimization

IEPC-2019-713

Companion papers:

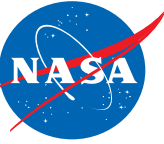
902, Wed 15:45

841, Thu 15:00

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Glenn Research Center

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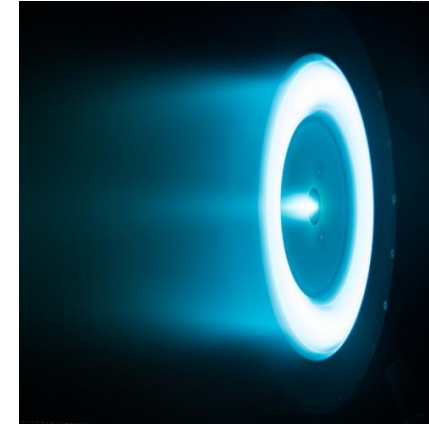


Outline

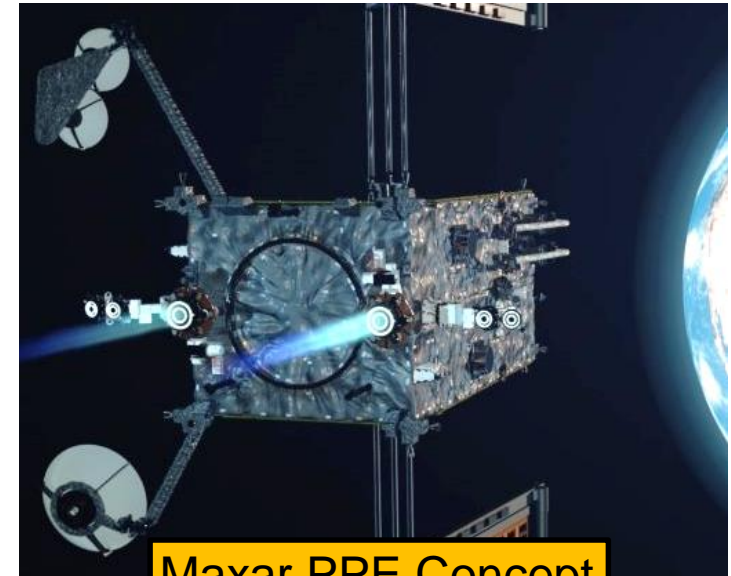
- Introduction
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- Data analysis
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 - Variation in Acceleration Profile
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- Conclusion

Introduction

- A NASA GRC and JPL team developed a 12.5-kW, magnetically-shielded Hall thruster, called Hall Effect Rocket with Magnetic Shielding (HERMeS)
- Flight development continuing in the form of Aerojet Rocketdyne's Advanced Electric Propulsion System (AEPS)
- Propulsion system for the Power and Propulsion Element (PPE), the first element of NASA's Gateway
- Completing risk reduction activities (using HERMeS) and transitioning to Engineering Test Unit (ETU) testing
- Developing a related Plasma Diagnostics Package (PDP)



◀ HERMeS in operation

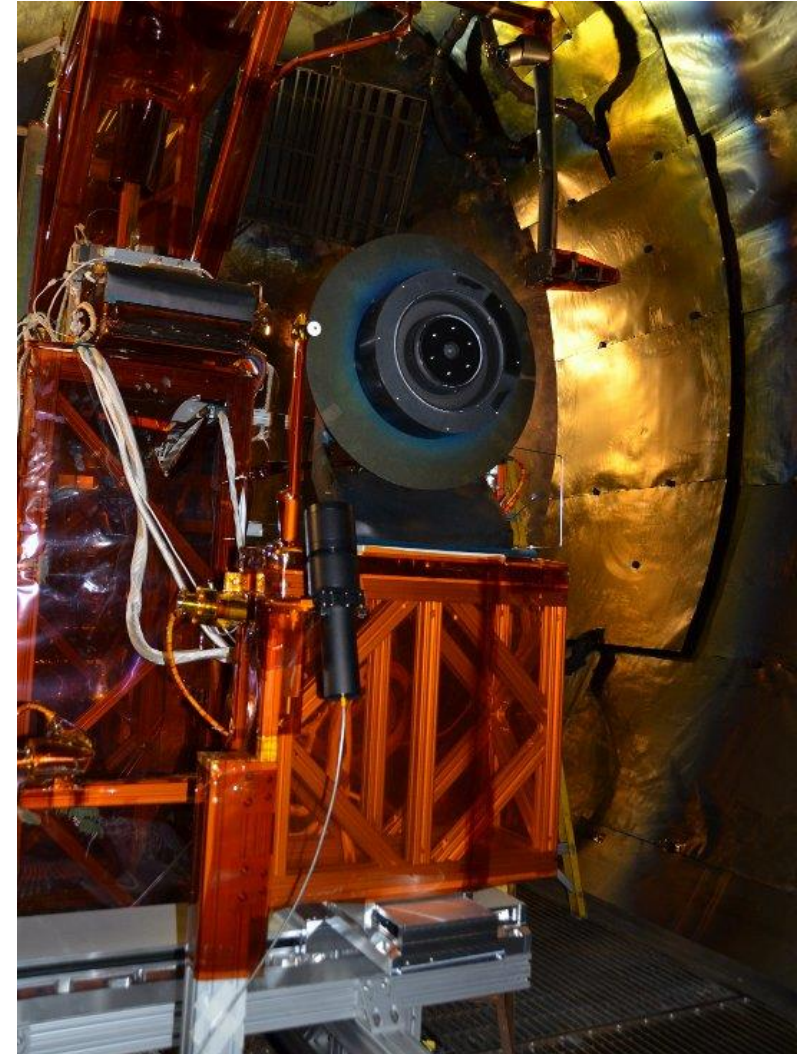


Maxar PPE Concept

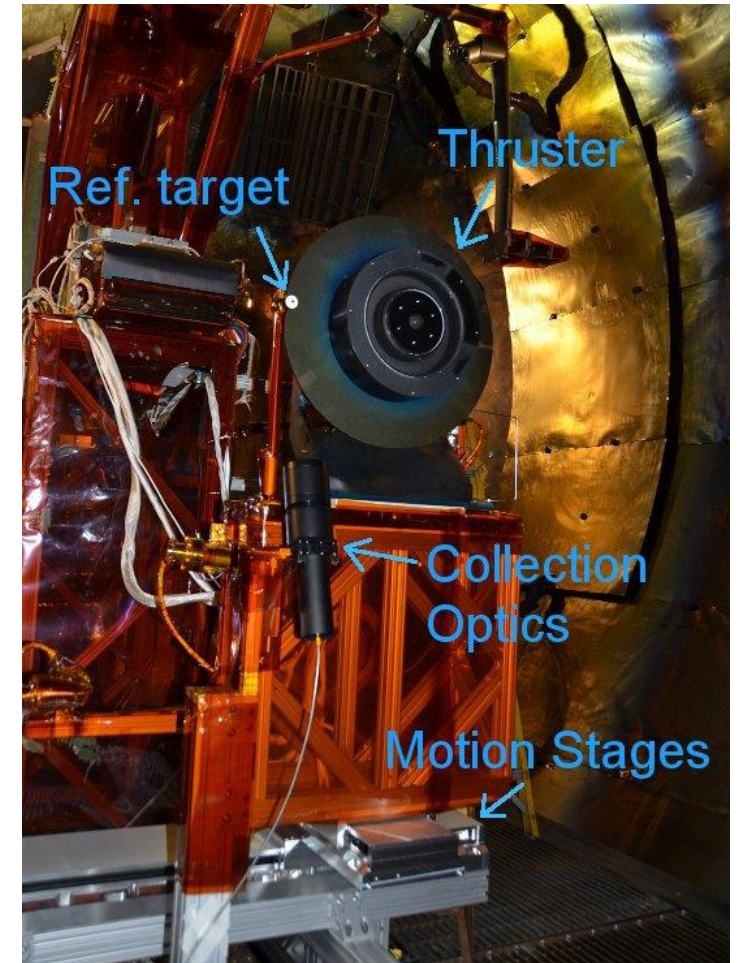
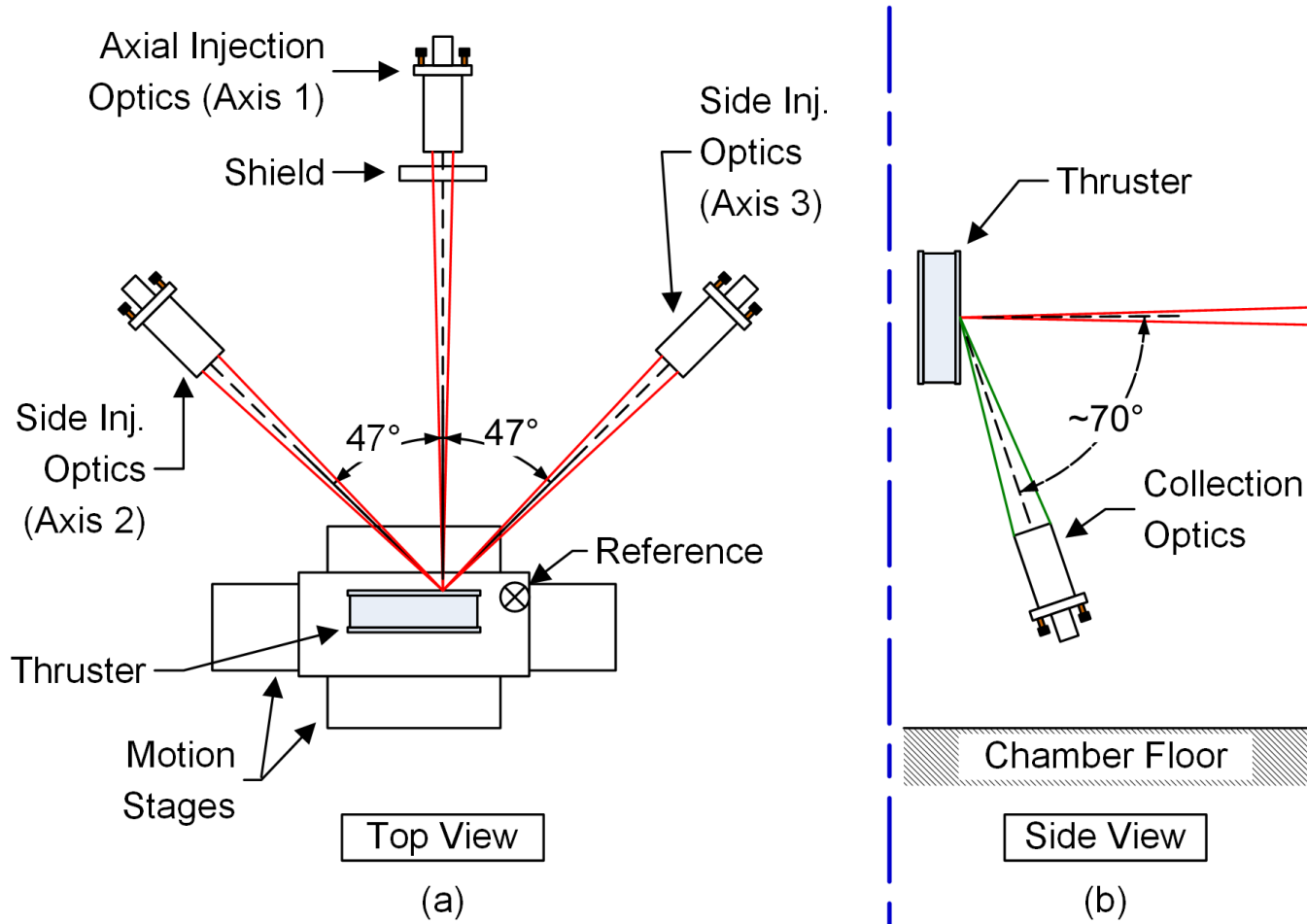
Magnetic Optimization Test

- HERMeS TDU1
 - Throttle range from 0.6 to 12.5 kW, 2000 to 3000 sec
 - Centrally mounted cathode, 7% cathode flow fraction
 - Cathode tied to thruster body
 - Test was in VF6, $\sim 1.2e-5$ Torr near thruster
- Four configurations in order of decreasing magnetic shielding: B0, B1, B2, B4
 - B0 is TDU baseline
 - Max radial magnetic field along channel centerline shifted upstream by about the same amount between each configuration in the sequence

Label	Discharge voltage, V	Discharge power, kW
300-6.3	300	6.25
600-12.5	600	12.50



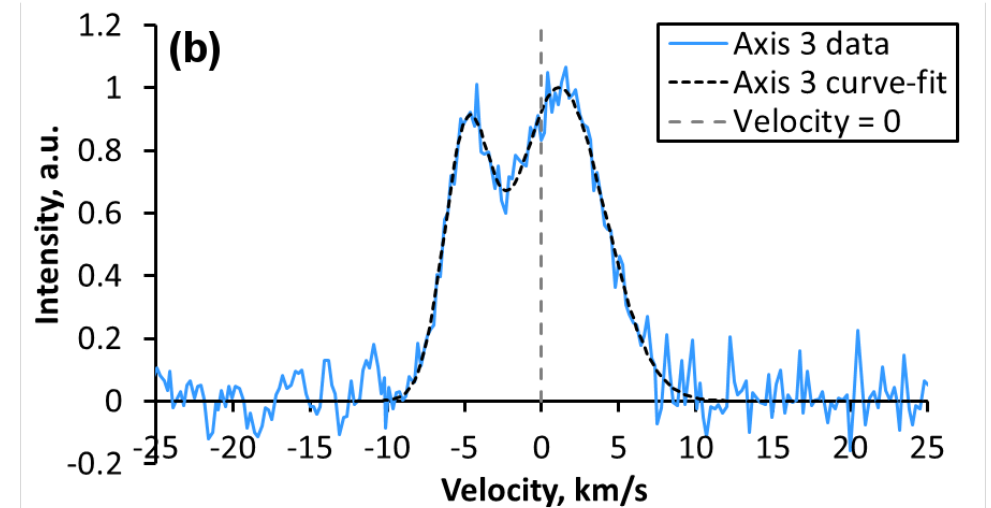
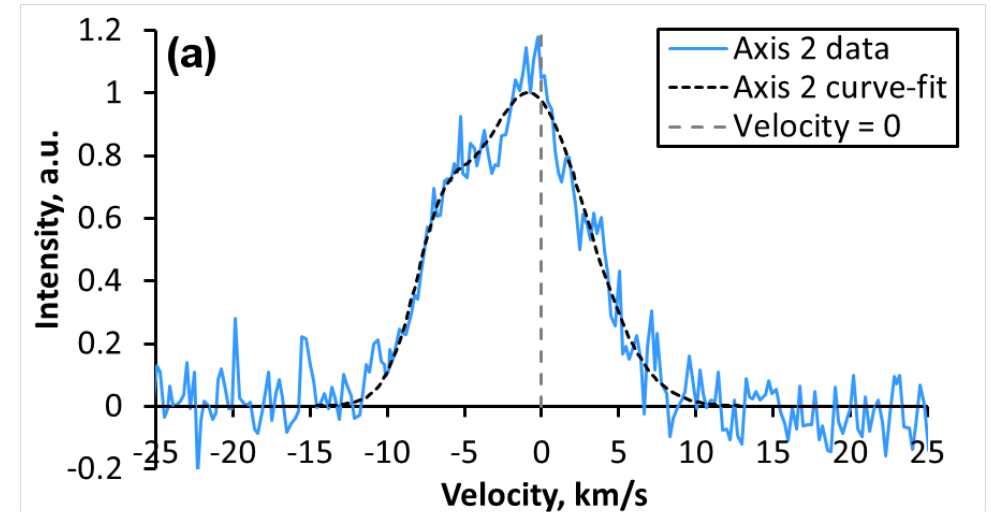
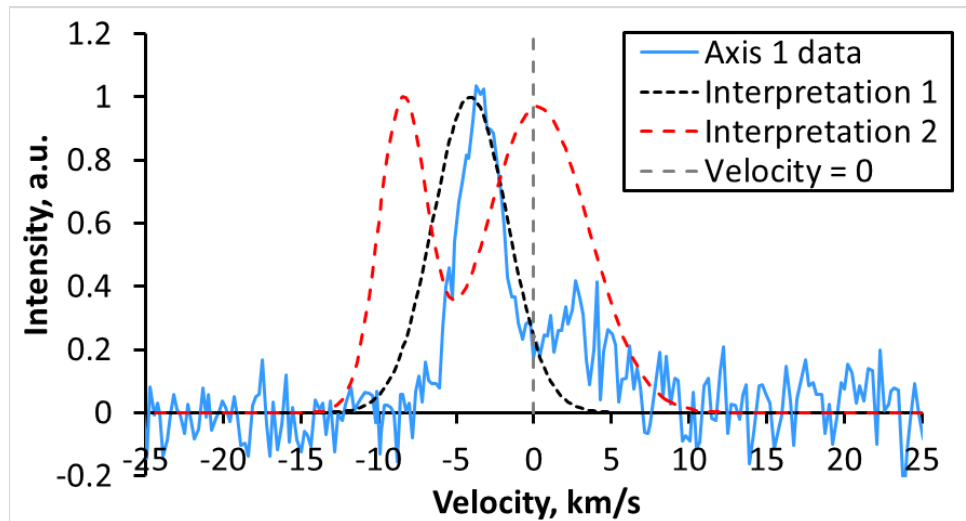
Experimental Setup – Vacuum Side Optics



Additional setup info in AIAA-2018-4723

Data Analysis

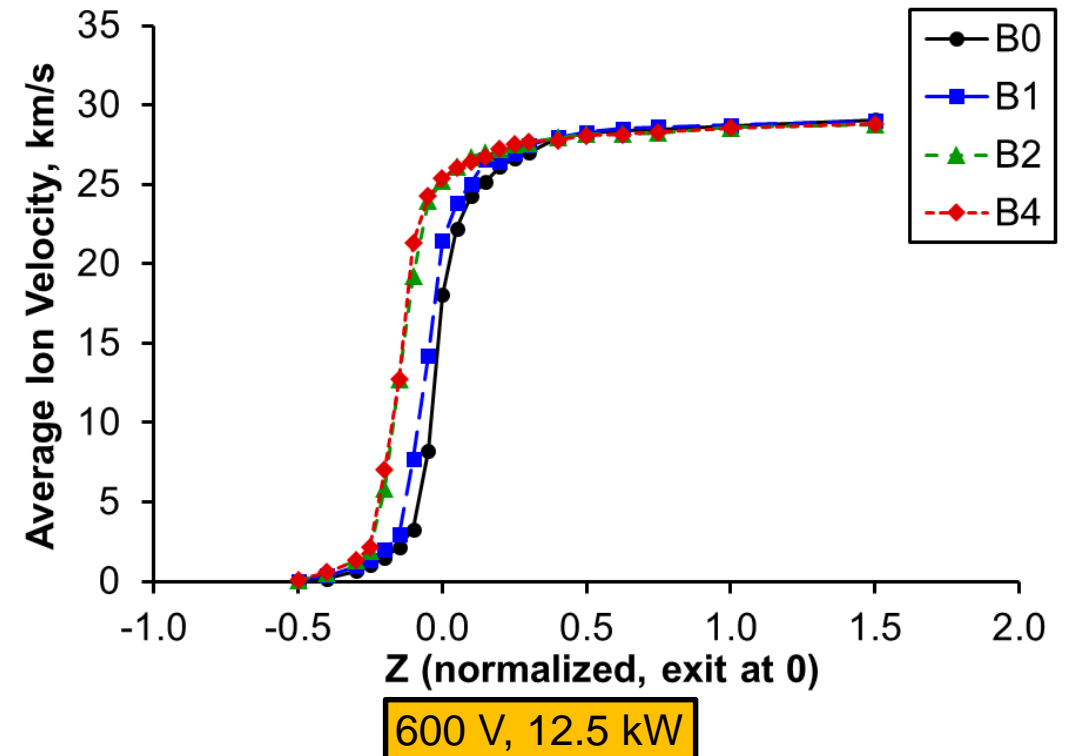
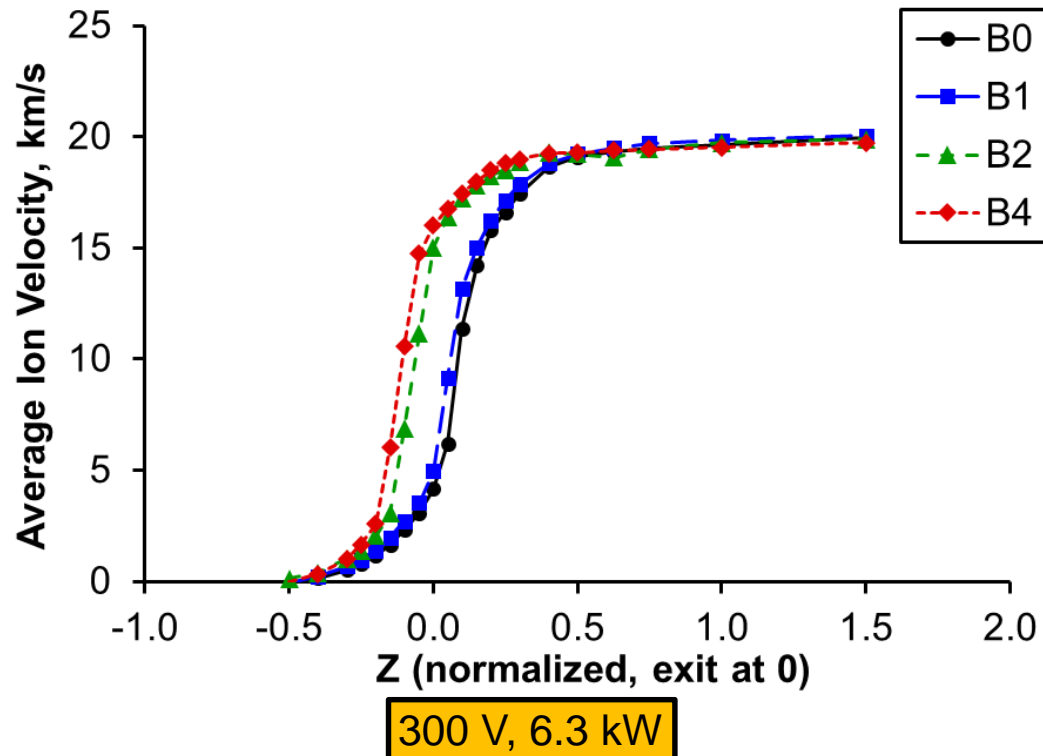
- Curve fit-based approach accounting for Zeeman Effect (See paper)
- Spatial uncertainty: 0.5 mm
- Velocity uncertainty: ± 112 m/s typical (± 600 m/s for noisiest scans)
- On Inner Front Pole Cover (IFPC), two peaks found on axes 2 and 3; comparison to axis 1 shows two ion populations pointed into the IFPC
 - One from the discharge channel, one from the cathode
 - See AIAA-2019-3897 for additional details



600 V, 12.5 kW, around radial middle of the IFPC

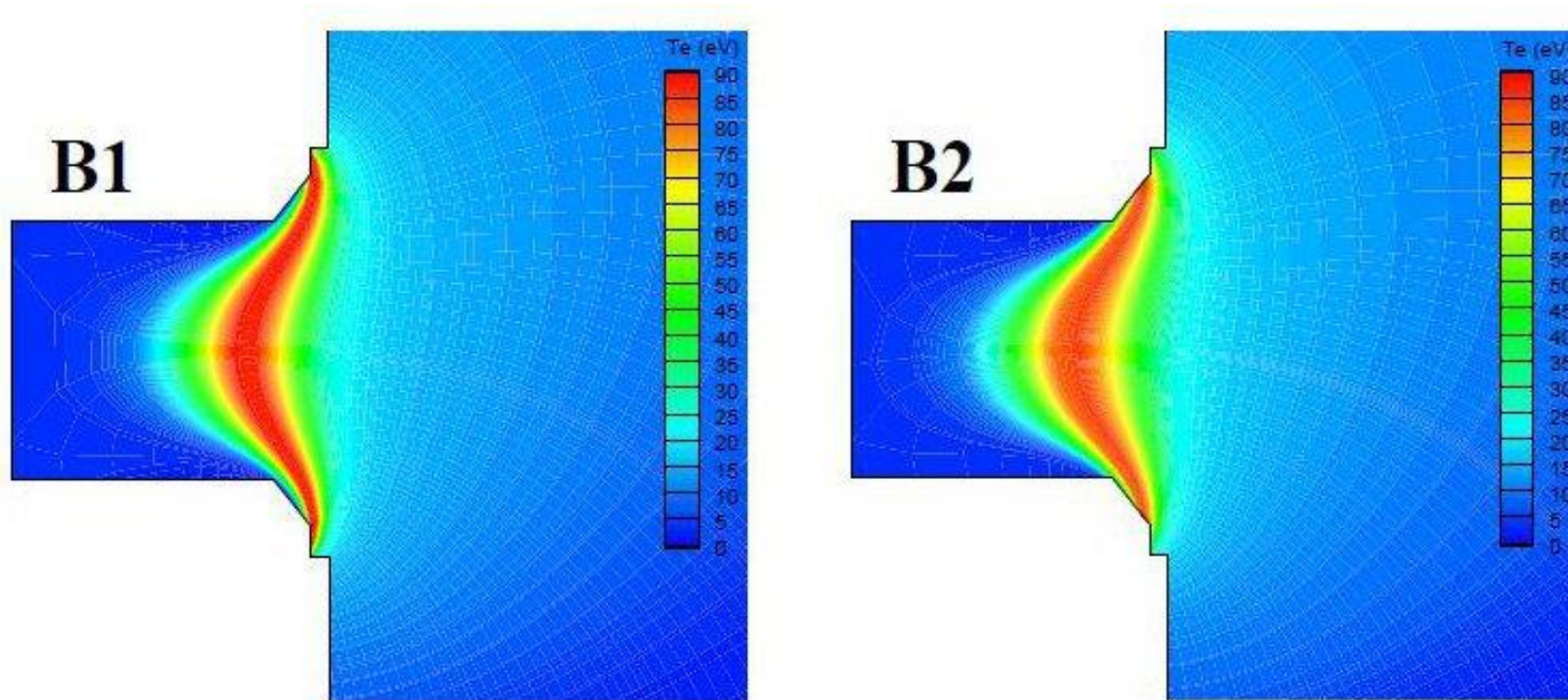
Variation in Acceleration Profile

- Noticeable jump in acceleration zone between B1 and B2



Prior Modeling Work Points to Possible Cause

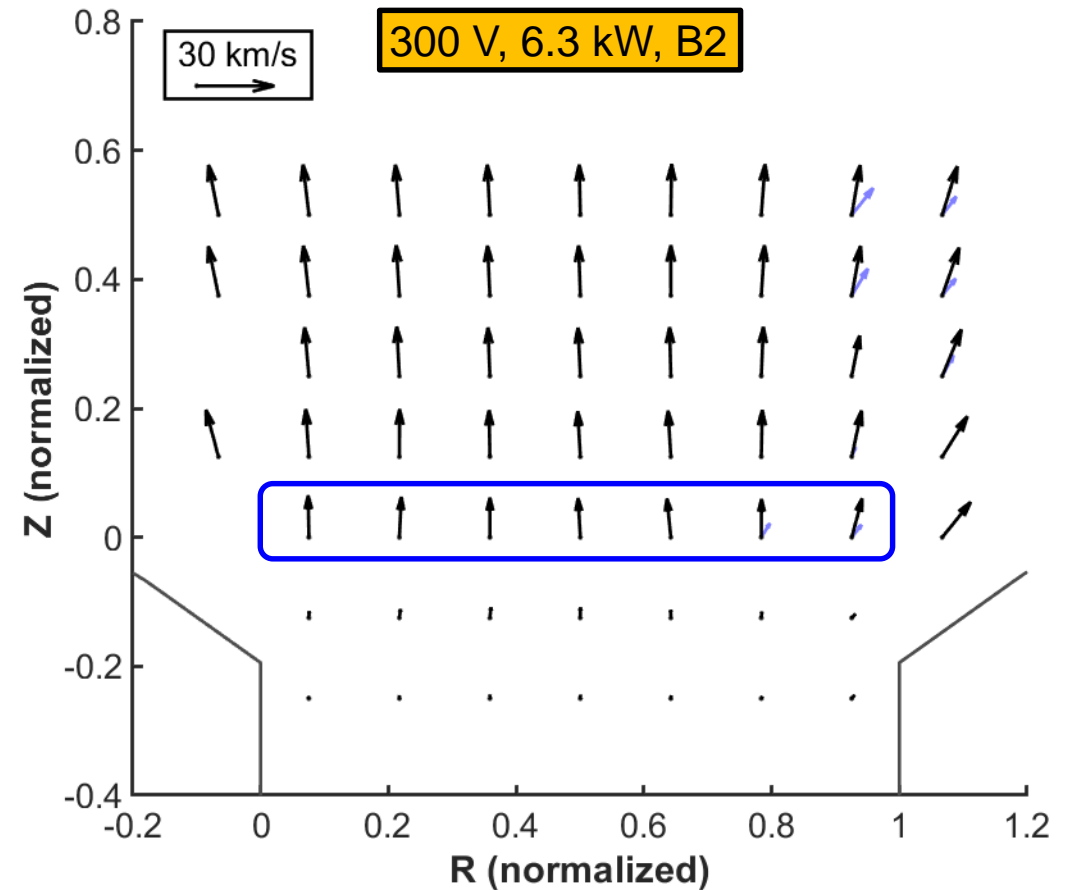
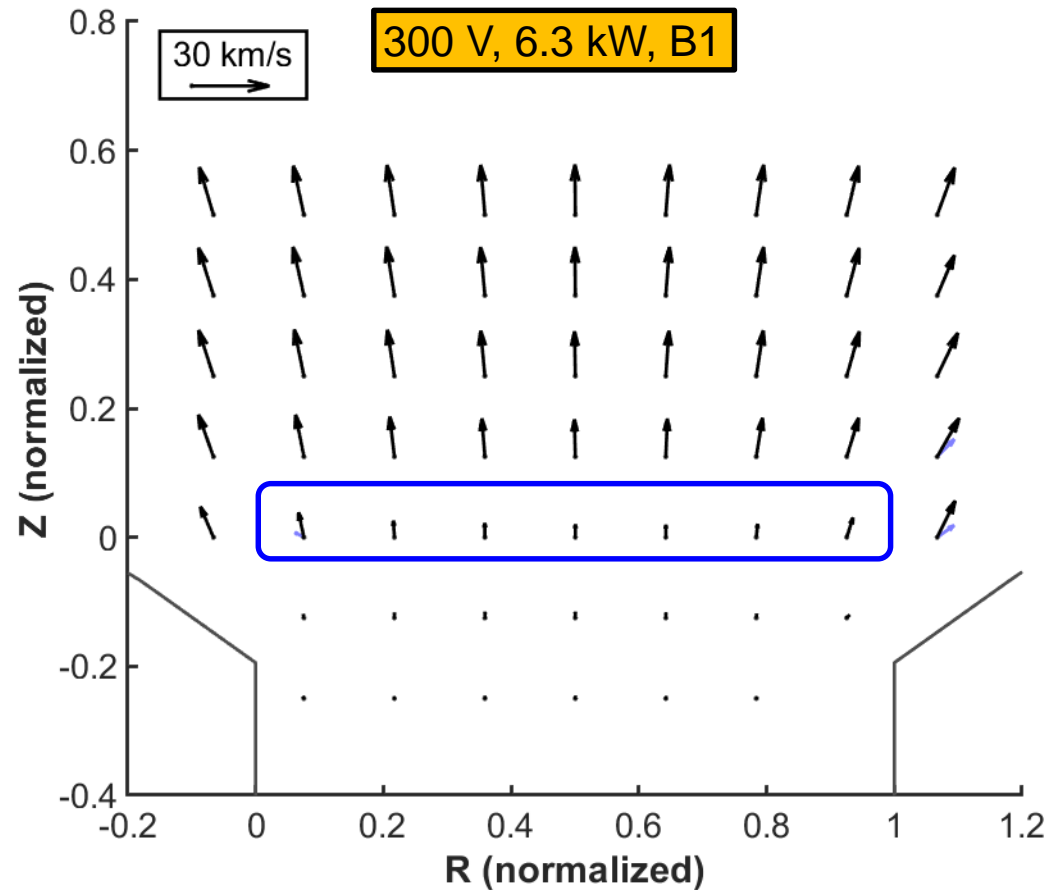
- Hall2De simulation from AIAA-2018-4720 shows a different attachment point for the high-energy plasma (red in plots below) for B1 and B2
- Whereas B0 and B1 were more like magnetically-shielded topology, B2 and B4 were more like reduced / un-shielded topology



Reproduced from AIAA-2018-4720 for convenience

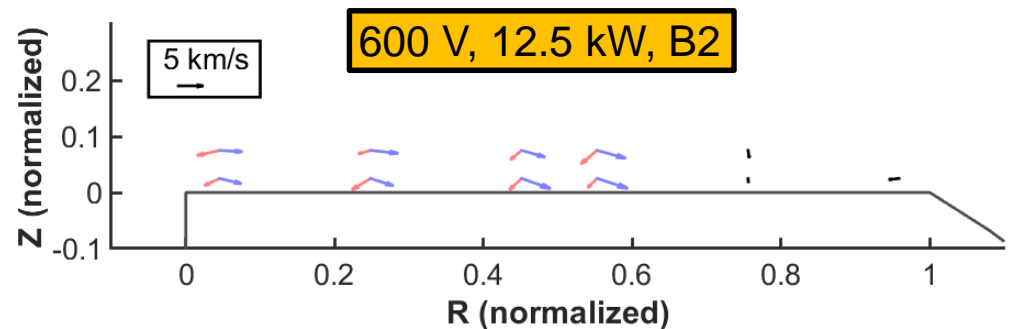
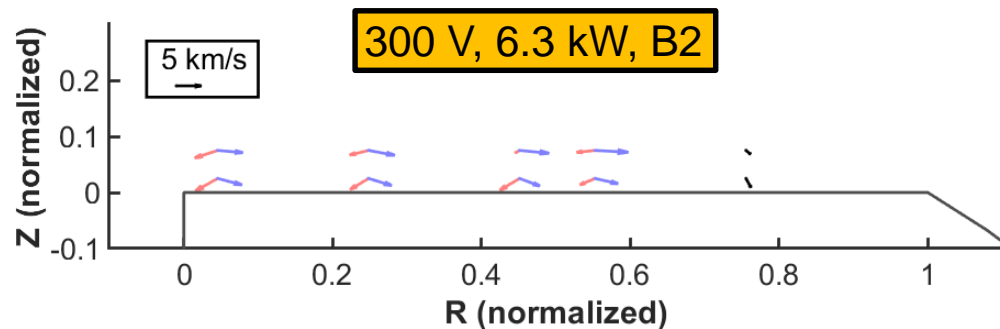
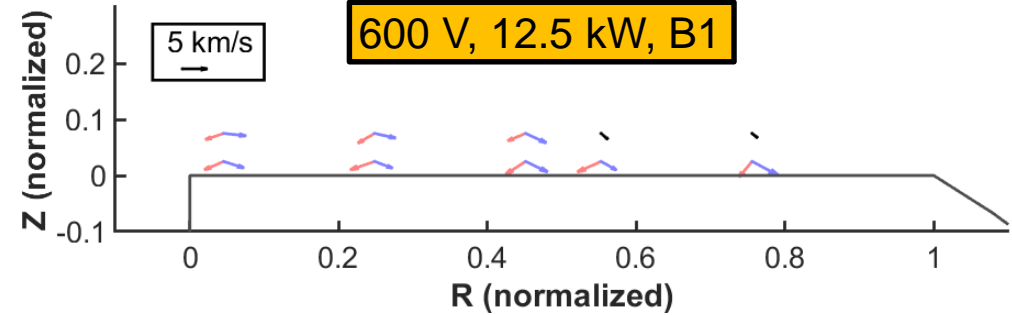
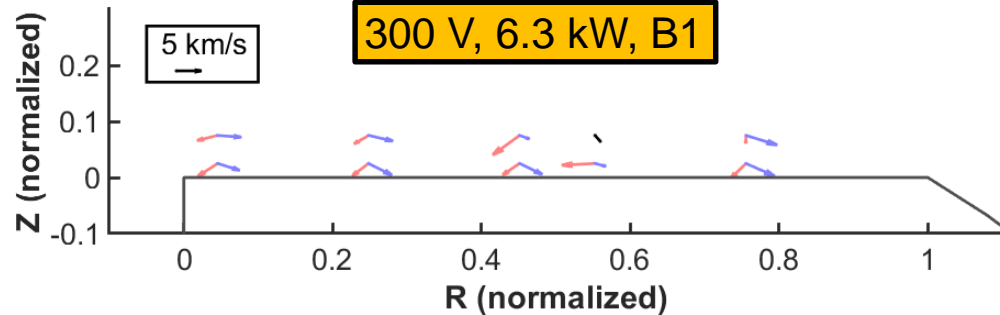
Ion Characteristics in Discharge Channel

- Going from B1 to B2: Accel zone move upstream, plume becomes less divergent



Ion Characteristics near IFPC

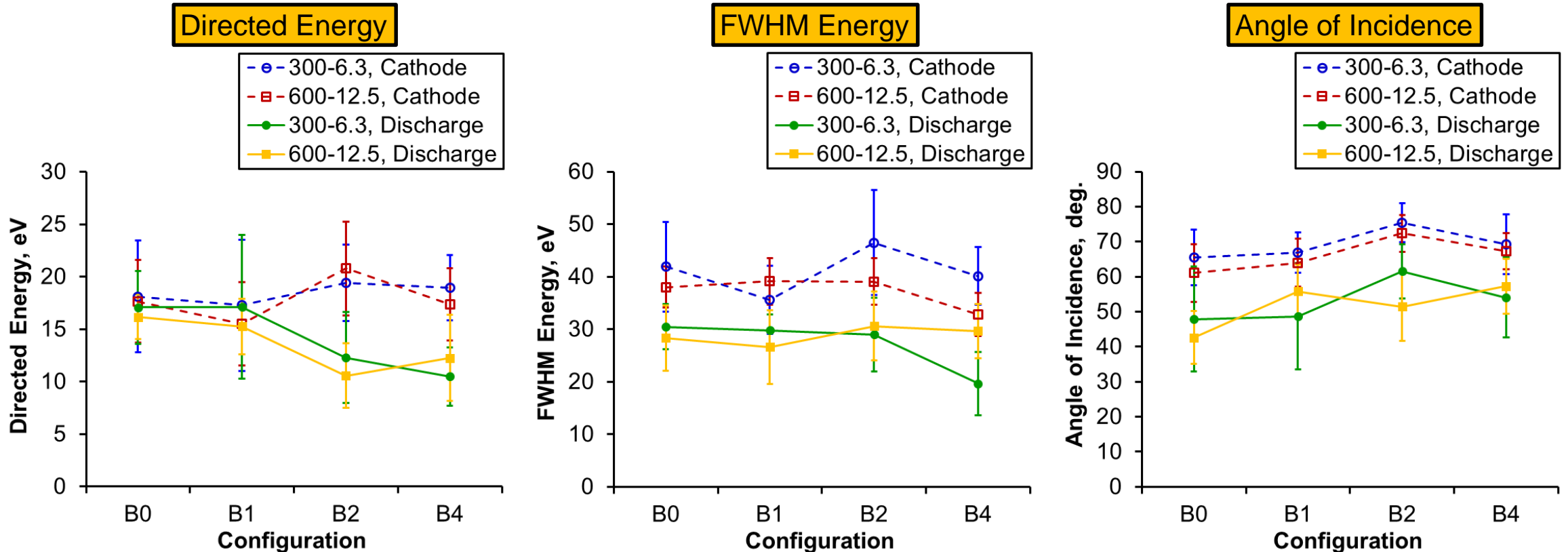
- Red vectors represent discharge channel stream, blue vectors represent cathode stream
- Both streams point into the inner pole cover at high oblique angles

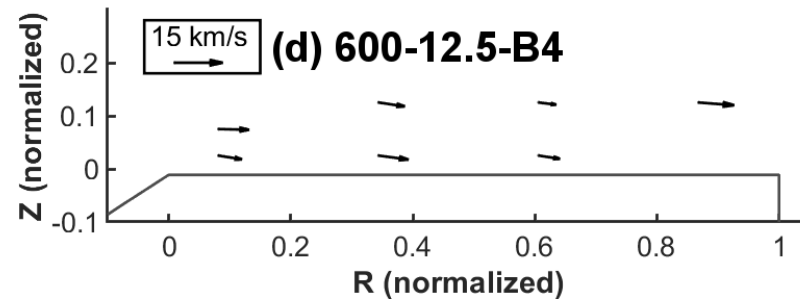
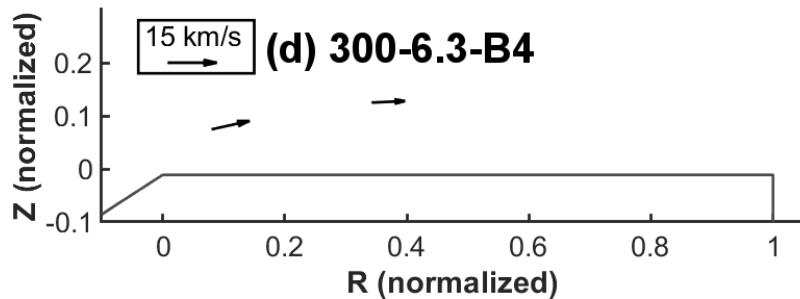
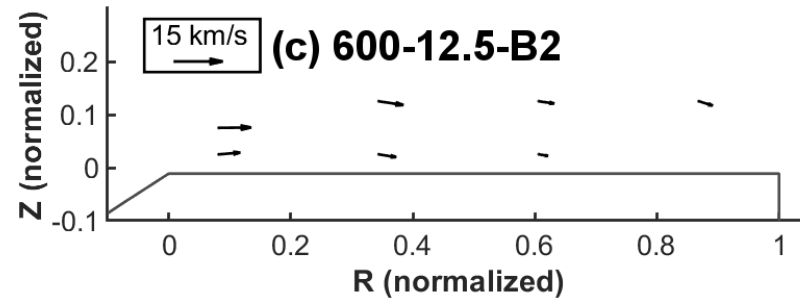
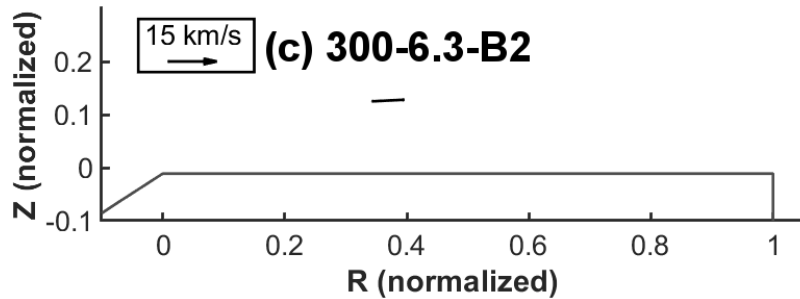
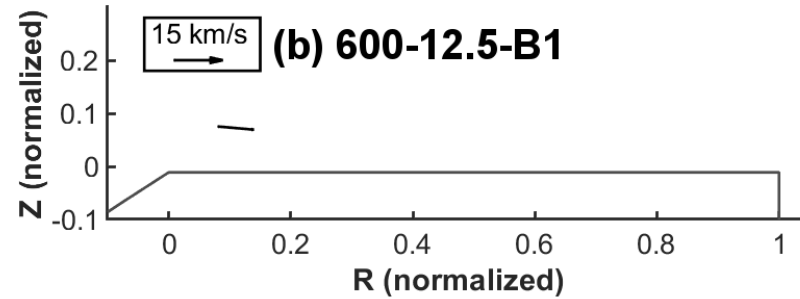
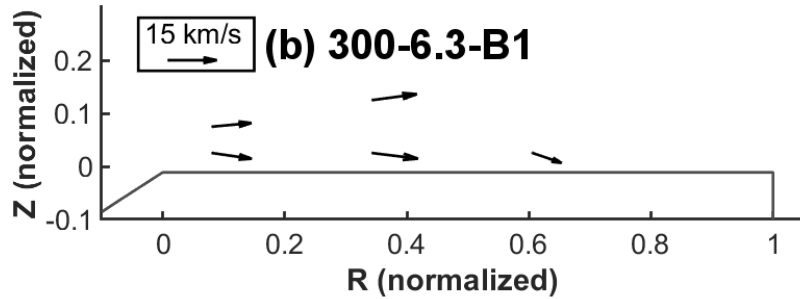
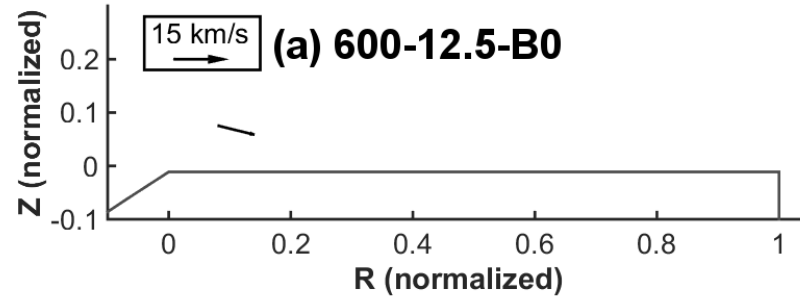
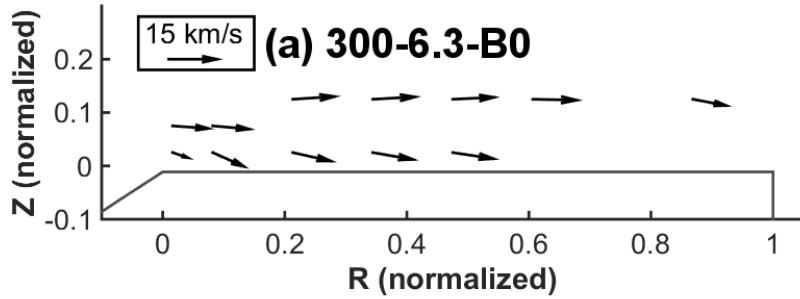




Summary of Ion Energy and Angle near IFPC

- In the labels of these plots, “Cathode” refers to cathode stream, “Discharge” refers to discharge channel stream
- About the same energy and angles across configurations to within uncertainty



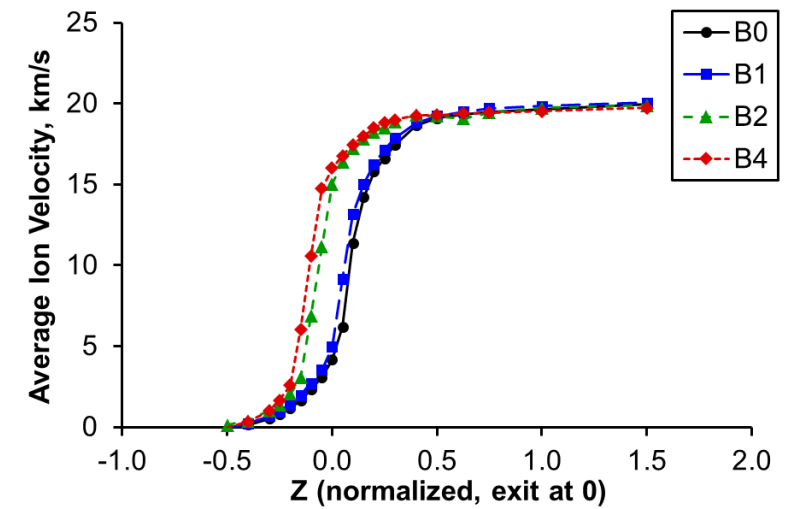
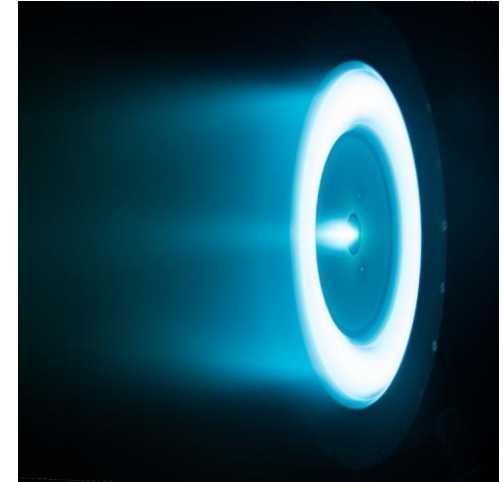


Ion Characteristics near OFPC

- OFPC = Outer Front Pole Cover
- Energy appear to decrease with configuration but limited number of samples
- Signal to noise ratio was low for conditions with limited samples

Conclusion

- Obtained ion velocity data while shifting the magnetic field upstream
- Observed a large jump in ion characteristics between B1 and B2 that suggest a change in how the plasma interacted with the channel walls and poles
- Ion energies near IFPC were constant to within the measurement uncertainty
- Ion energies near OFPC appeared to decrease as magnetic field shift upstream but data set was limited
- Combined with companion papers (902, Wed 15:45 and 841, Thu 15:00), demonstrate that one can trade pole erosion versus channel wall erosion





Acknowledgment

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 - NASA Space Technology Mission Directorate Solar Electric Propulsion Technology Demonstration Mission project for funding this work,
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