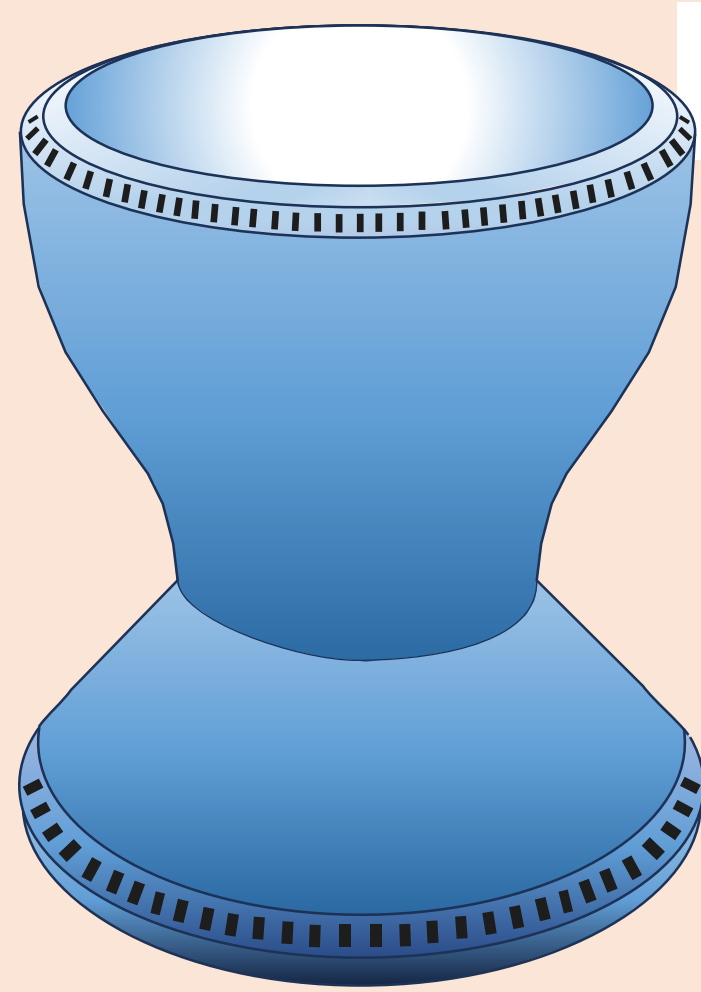
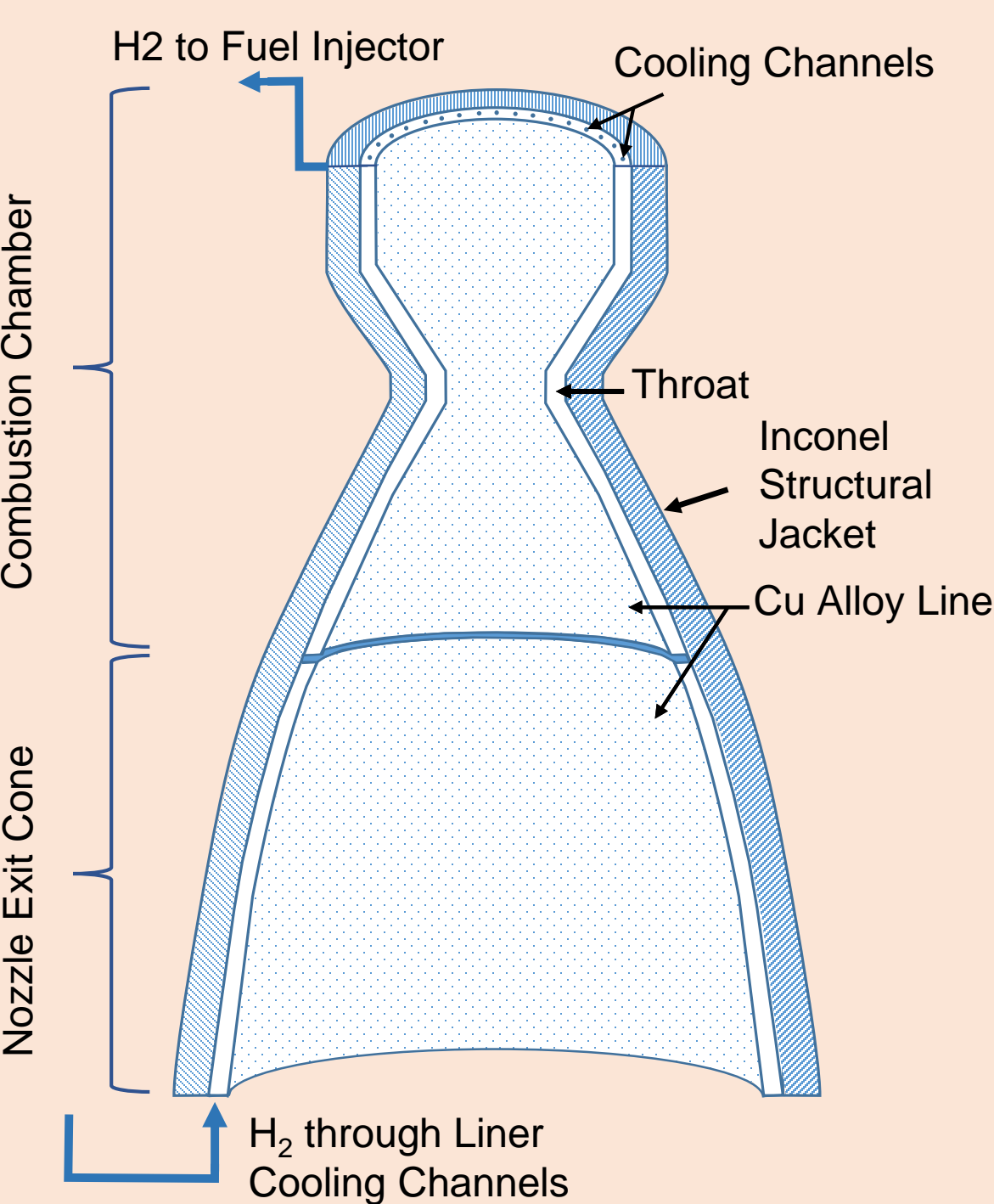


Background

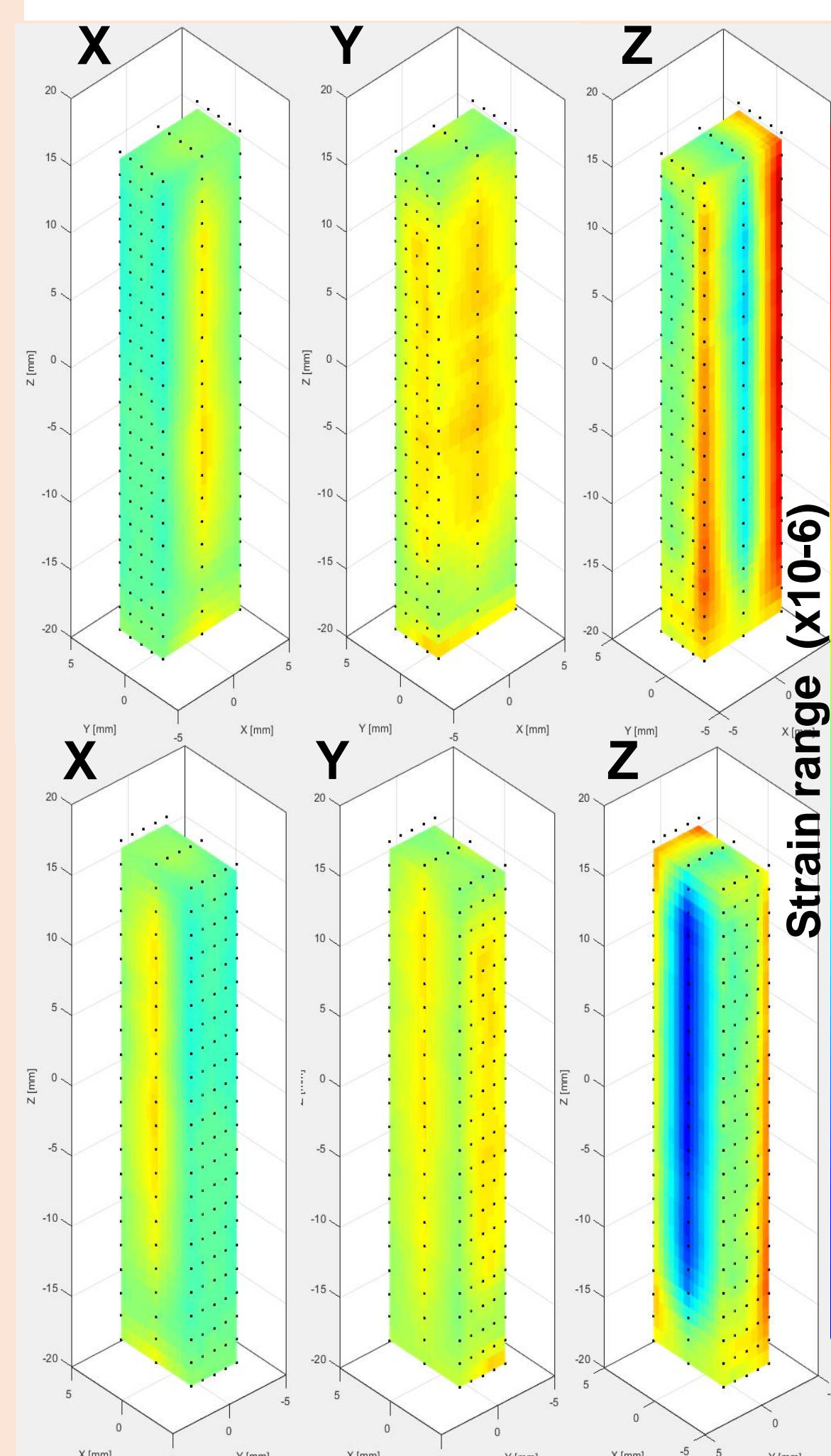
GRCo-84 is a precipitation strengthened alloy composed of Cu-8 Cr-4 Nb at% with Cr₂Nb precipitates that provide dispersion and precipitation strengthening characteristics and limited solubility in the Cu matrix. The particle role of Cr₂Nb (C15 Laves) is unusual only contributing 1/3 of strengthening at high temperatures while the matrix provides the remainder. The particles mechanically and thermally stabilize the matrix retaining purity and preventing coarsening and loss of strength. At high temperatures (50-85% T_{m,Cu}), GRCo-84 provides the best thermal and mechanical properties of available alloys.



GRCo-84 is currently in development for reusable launch vehicles including the Space Launch System (SLS) with a focus on fabrication with additive manufacturing (AM) techniques. GRCo-84 is an optimal material for consolidating with AM. The base material is costly, the production times are long, and more geometry control can considerably improve cooling efficiency. Development of AM GRCo-84 with selective laser melting (SLM) has rapidly progressed due to ease of printing and limited operator adjustment between builds, but the necessary knowledge-base of thermal history and stress state during builds is still under development.



Motivation



Previous neutron diffraction experiments on NRSF-2 revealed high residual stress in the Cu matrix (311) of as-built SLM GRCo-84. Higher even than reported yield strain values. However, GRCo-84 is a two-phase alloy, and almost no literature exists to explain stress evolution in SLM GRCo-84. It is critical to understand stress sharing behavior between the two phases to accurately assess previous and future neutron results on a single peak, and the future of the alloy. Literature suggests 1/3 Orowan strengthening and 2/3 Hall-Petch for traditionally fabricated GRCo-84, but this might not be the case with SLM as its grain structure should be significantly different [1].

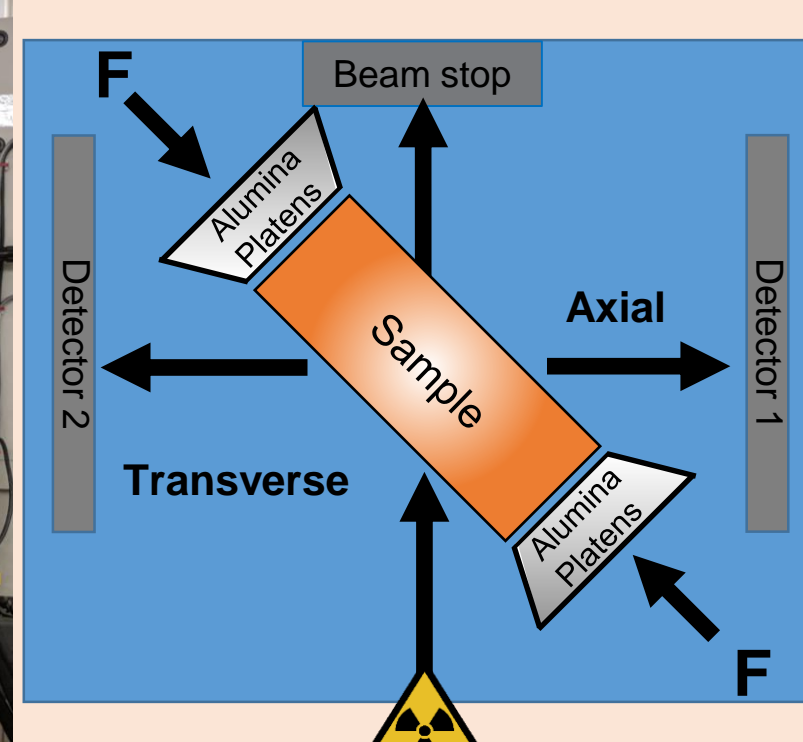
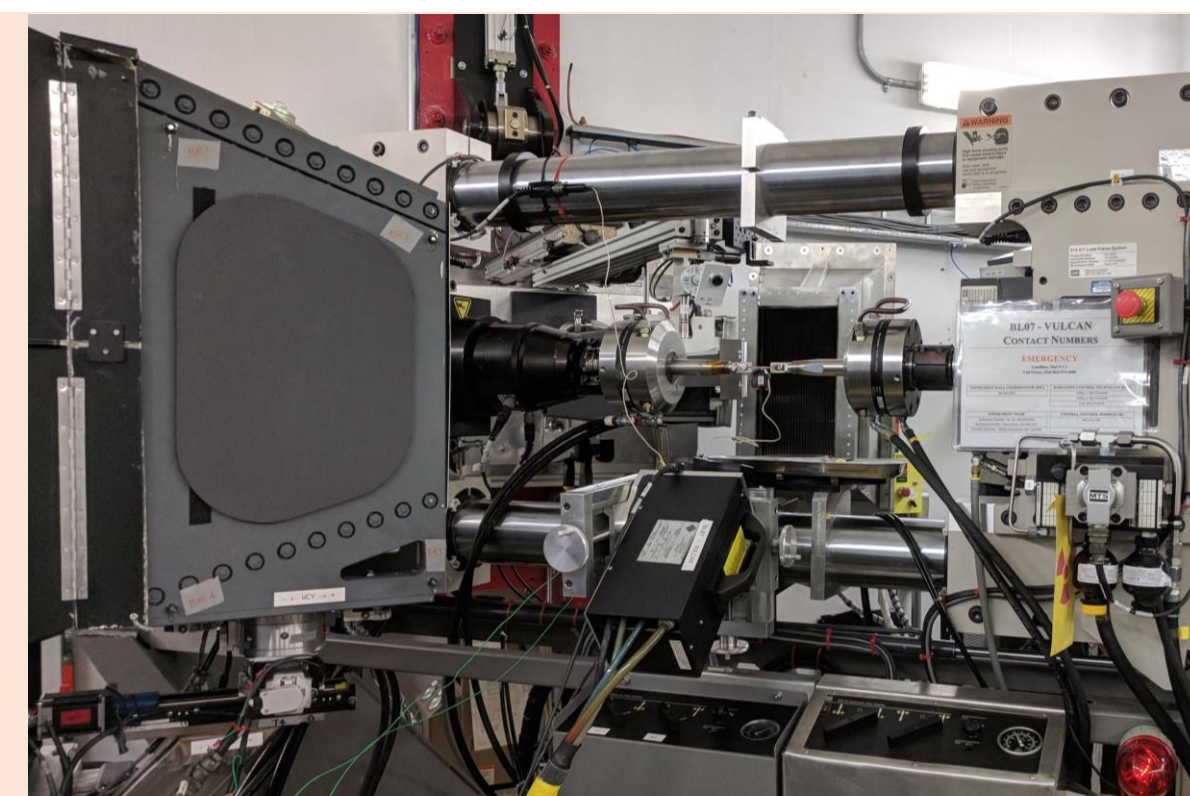
Stress/Strain Estimates/Assumptions

- Young's modulus, E, 111-130 GPa
 - Poisson's ratio, ν , of 0.29
 - d_0 of 1.09152 Å
- $$\epsilon_{hkl} = \frac{d_{hkl} - d_{hkl}^0}{d_{hkl}^0}$$
- $$\sigma_{ij} = \frac{E}{1+\nu} \left(\epsilon_{ij} + \frac{\nu}{1-2\nu} (\epsilon_{11}^{hkl} + \epsilon_{22}^{hkl} + \epsilon_{33}^{hkl}) \right)$$

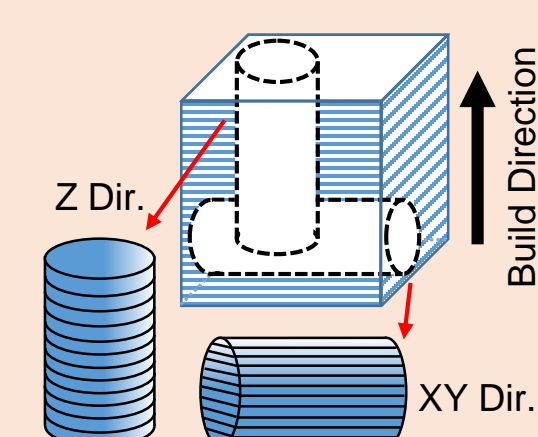
Strain mapping of as-built GRCo-84 Cu (311) obtained on HFIR's NRSF-2 showing arbitrary x, y, and z strain in two orientations

Neutron Diffraction HIP vs As-Built

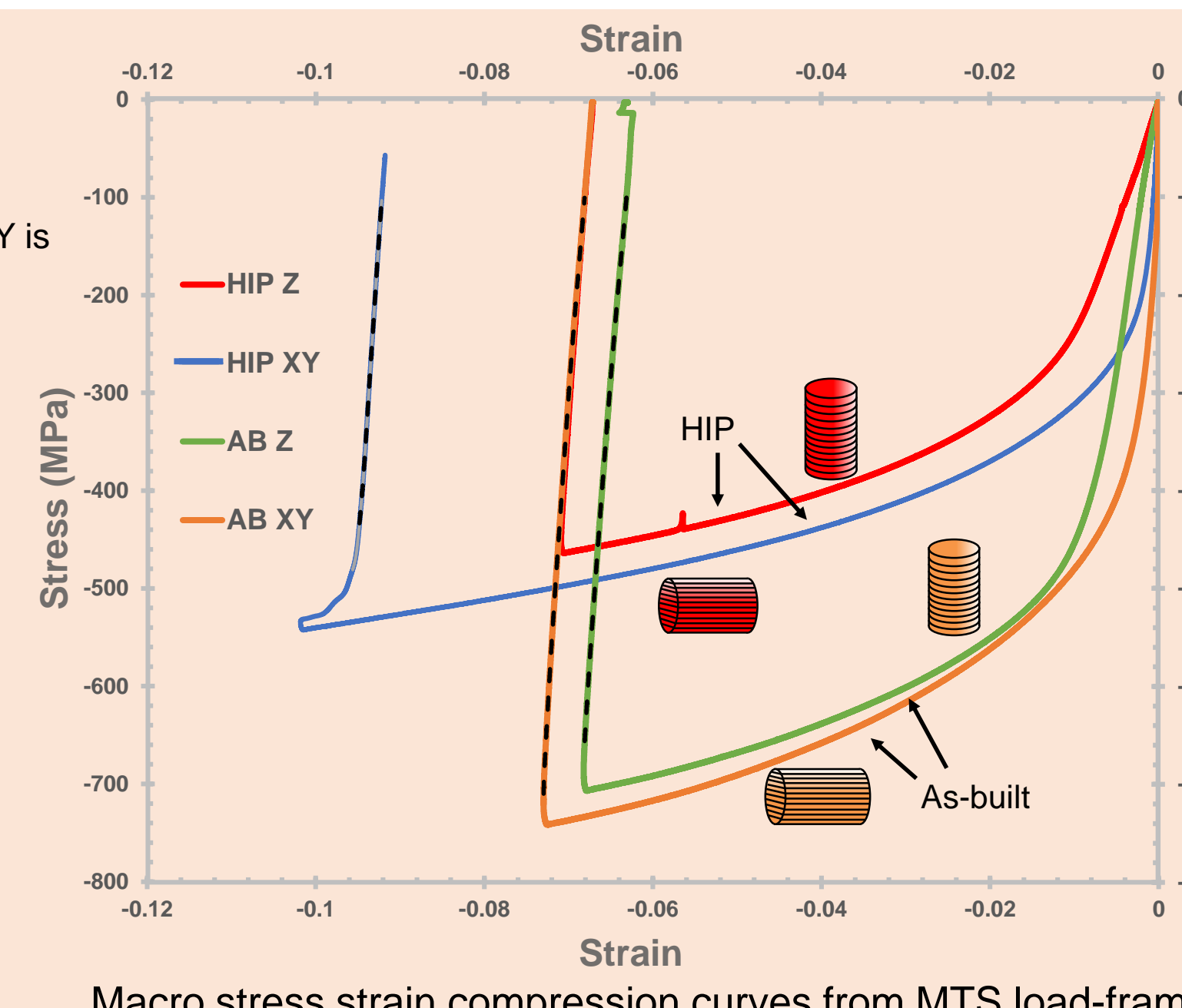
Sample & VULCAN Environment



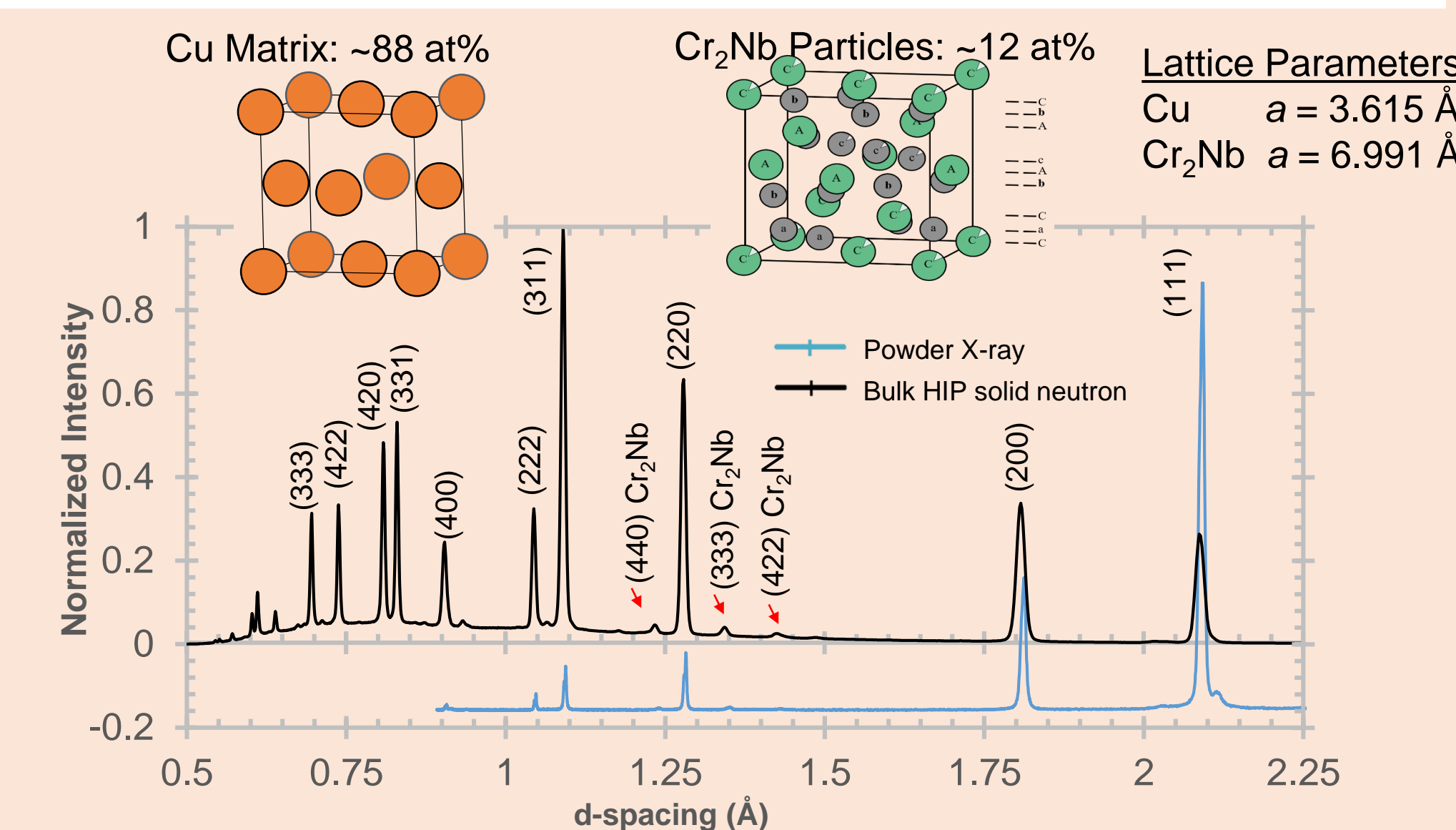
- Compression samples EDMed from bulk cubes
- XY and Z, but only XY is presented here
- 8 mm diameter
- 16 mm length



Elastic Modulus (unloading)
AB XY: 123 GPa HIP XY: 119 GPa
AB Z: 111 GPa HIP Z: 130 GPa
(all R² > .99)

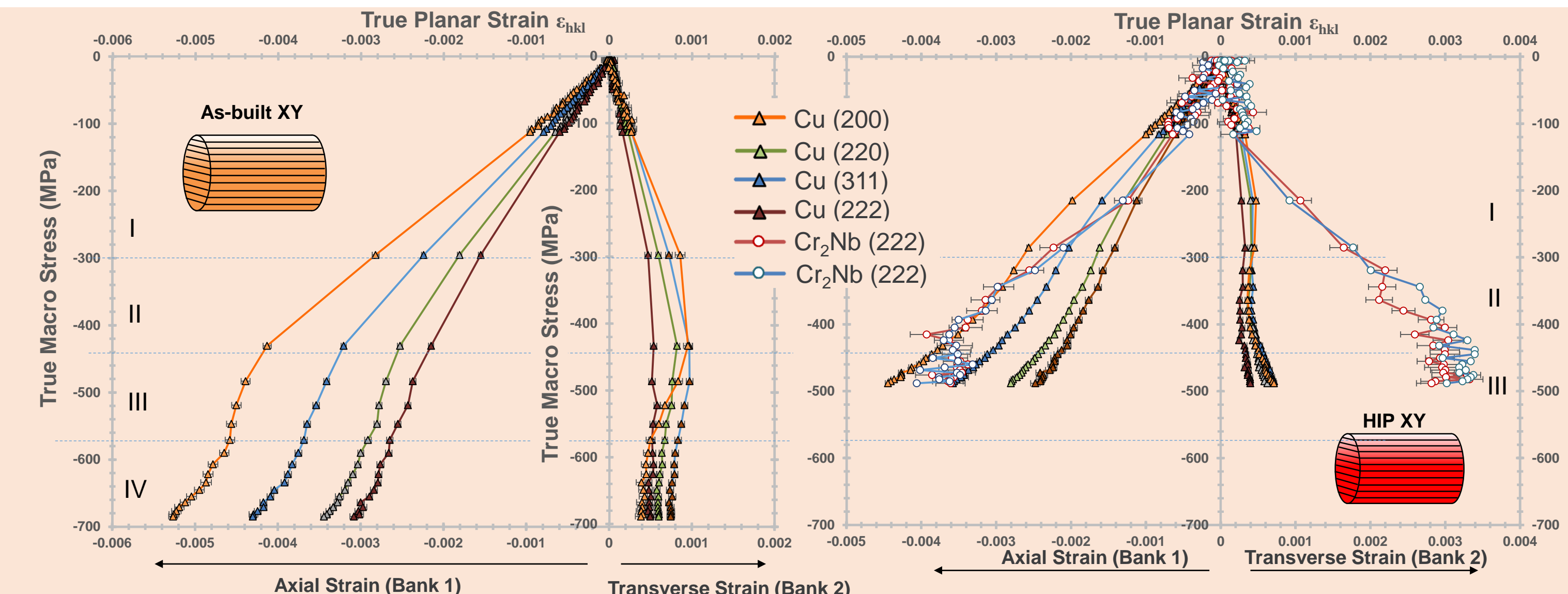


Diffraction Profile

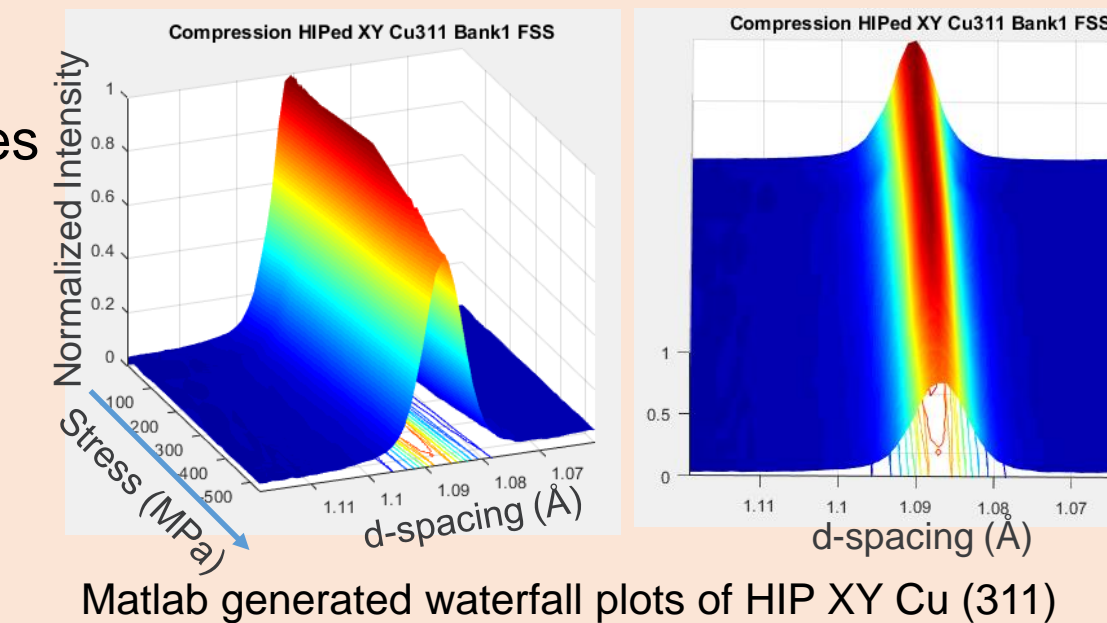


- Limited as-built Cr₂Nb diffraction
- Effect of strain, crystallite size, & texture
- 0.1% to 12% phase %
- As-built to HIP

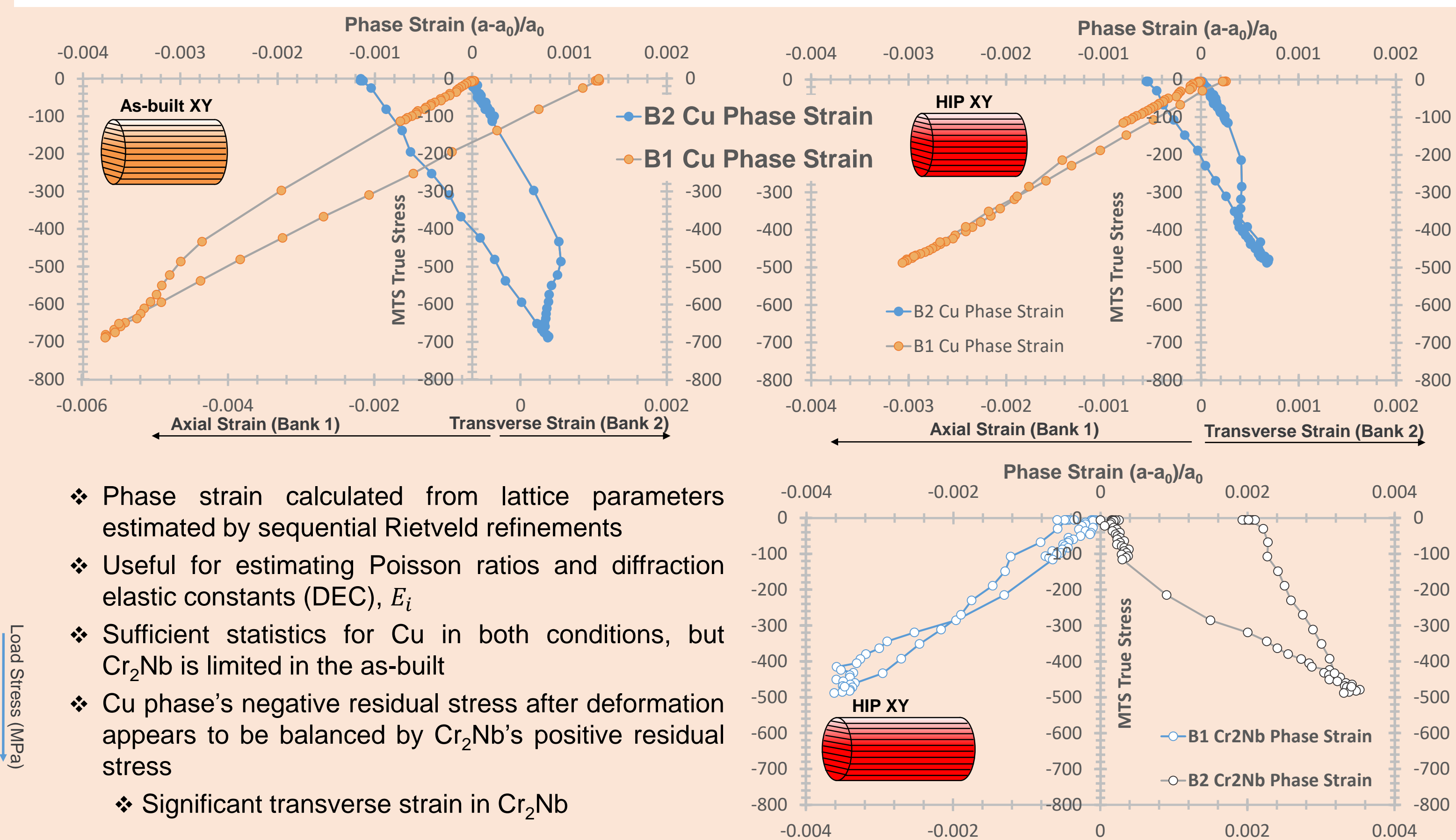
Planar strain



- Three distinct regions (I-III) of slope changes in Cu matrix
- Should correlate to different yielding behavior between planes and phases
- Cu (200) carries the most stress while (222) carries the least
- Cr₂Nb peaks are too broad to plot in as-built condition
- Cr₂Nb increases rapidly in strain in region II
- As-built and HIP have significantly different behavior in transverse direction
- Could be effect of matrix or precipitate change after HIP



Phase Strain

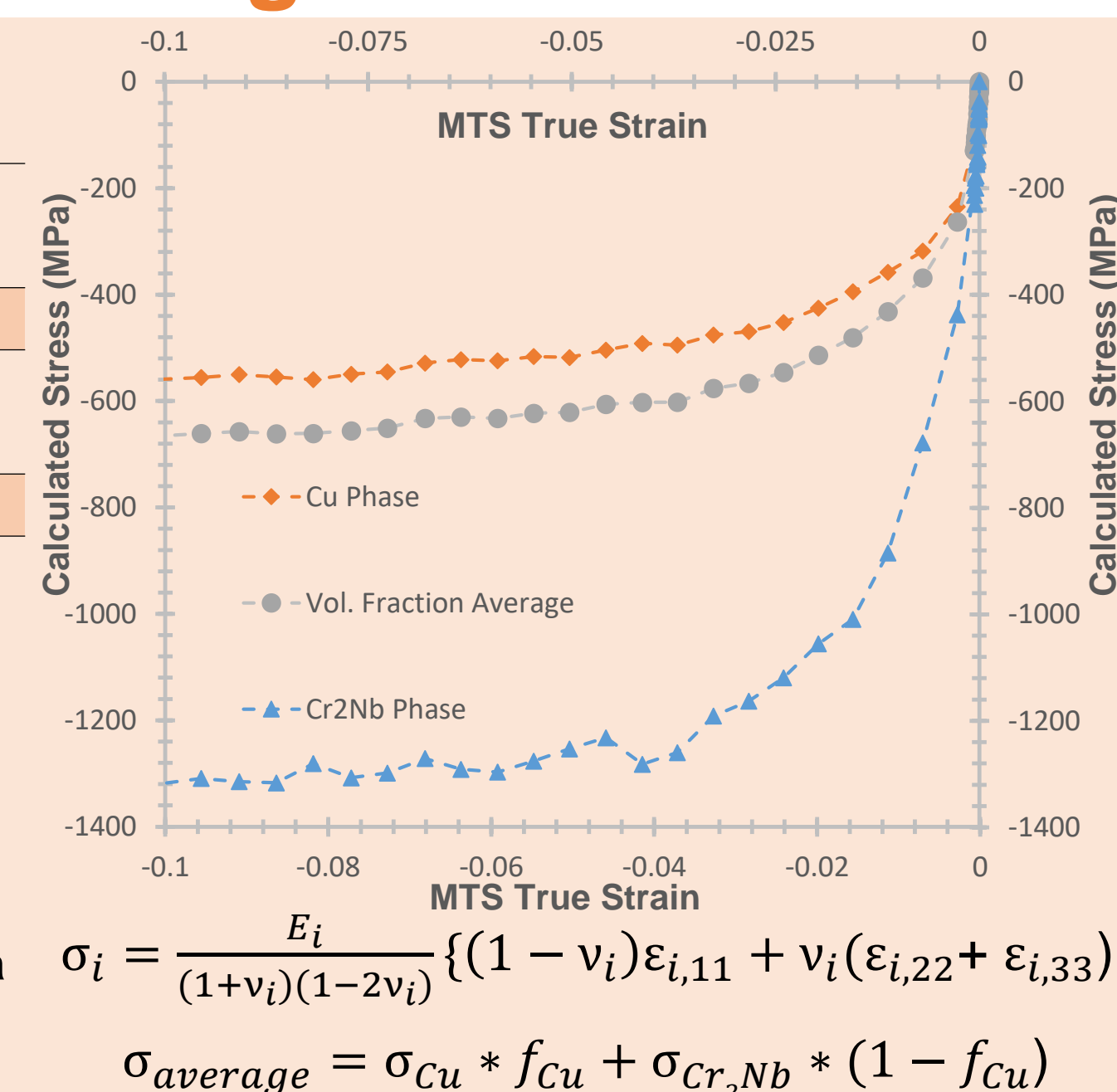


- Phase strain calculated from lattice parameters estimated by sequential Rietveld refinements
- Useful for estimating Poisson ratios and diffraction elastic constants (DEC), E_i
- Sufficient statistics for Cu in both conditions, but Cr₂Nb is limited in the as-built
- Cu phase's negative residual stress after deformation appears to be balanced by Cr₂Nb's positive residual stress
- Significant transverse strain in Cr₂Nb

Phase Partitioning

Poisson ratio, ν	Load	As-built XY		HIP XY	
		Cu	Cr ₂ Nb	Cu	Cr ₂ Nb
DEC (GPa)	Load	147	---	141	162
	Unload	129	---	146	162
	Average	138	---	143.5	162

- Poisson ratio estimated from comparison of axial and transverse lattice parameters
- DEC estimated from elastic regions
- HIP XY Partitioning $f_{Cu} \approx 0.86$
- Both phases share similar elastic regions
- Cr₂Nb rapidly strain hardens until ~1200 MPa
- Cu elongates steadily with limited hardening after ~600 MPa
- As-built Cr₂Nb statistics insufficient to plot, Cu shows similar trends to HIP



Conclusions

Previous strain mapping showed evidence of strong compressive and tensile Cu matrix strain in the (311) which as the primary matrix component led to the assumption that the macro-stress behavior was similar; however, the stress partitioning behavior shown in this in-situ experiment requires re-evaluation of the strain mapping data. Wherever there is residual thermal strain in the matrix, there is likely balancing strain in the Cr₂Nb phase that reduces the overall stress in these areas.

The as-built's Cr₂Nb particles are significantly smaller and less uniformly distributed than the HIP condition and exhibits more brittle behavior. These could be accounted for by the Cr₂Nb being in a more unstable or less crystalline form than after it has been HIPed, or there could be a significant fraction of dissolved Cr₂Nb in the Cu matrix.

Future work includes analysis of EBSD on primary samples, further metallography, in-situ HTXRD, and in-situ mechanical EBSD to observe grain interactions in collaboration with current neutron data.

References

1. Anderson, K.R. and J.R. Groza, *Microstructural Size effects in high-strength high-conductivity Cu-Cr-Nb alloys*. Metallurgical and Materials Transactions A, 2001, 32(8), p. 1211-1224.
2. "Neutron Scattering Lengths and Cross Sections." Neutron Scattering Lengths and Cross Sections. NIST, 07 Jan. 2013. Web. 27 Sept. 2016.
3. Ellis, David L. "GRCo-84: A high-temperature copper alloy for high-heat-flux applications." (2005).

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Acknowledgements

RPM would like to acknowledge partial support through the Manufacturing and Materials Joining Innovation Center (Ma²JIC) University of Tennessee, Knoxville site. Ma²JIC is funded by the National Science Foundation (NSF) through the Industry/University Cooperative Research Center (I/UCRC) program award number IIP 1540000 (Phase I) and 1822186 (Phase II).

This research used resources at the High Flux Isotope Reactor and Spallation Neutron Source, a DOE Office of Science User Facility operated by the Oak Ridge National Laboratory. RPM would also like to acknowledge Oak Ridge National Laboratory for its partial support as a Go! Student on the NRSF-2 HB-2B beamline.