

**Energy Tech 2019 Conference**

**IX Center – Cleveland, OHIO**

**10/22 to 10/25, 2019**

***Gas Turbine Energy Conversion Systems for Lunar  
& Planetary Nuclear Power Plants including Ground Based  
Applications using LiFTR “Liquid Fluoride  
Thorium Reactor” Technology***

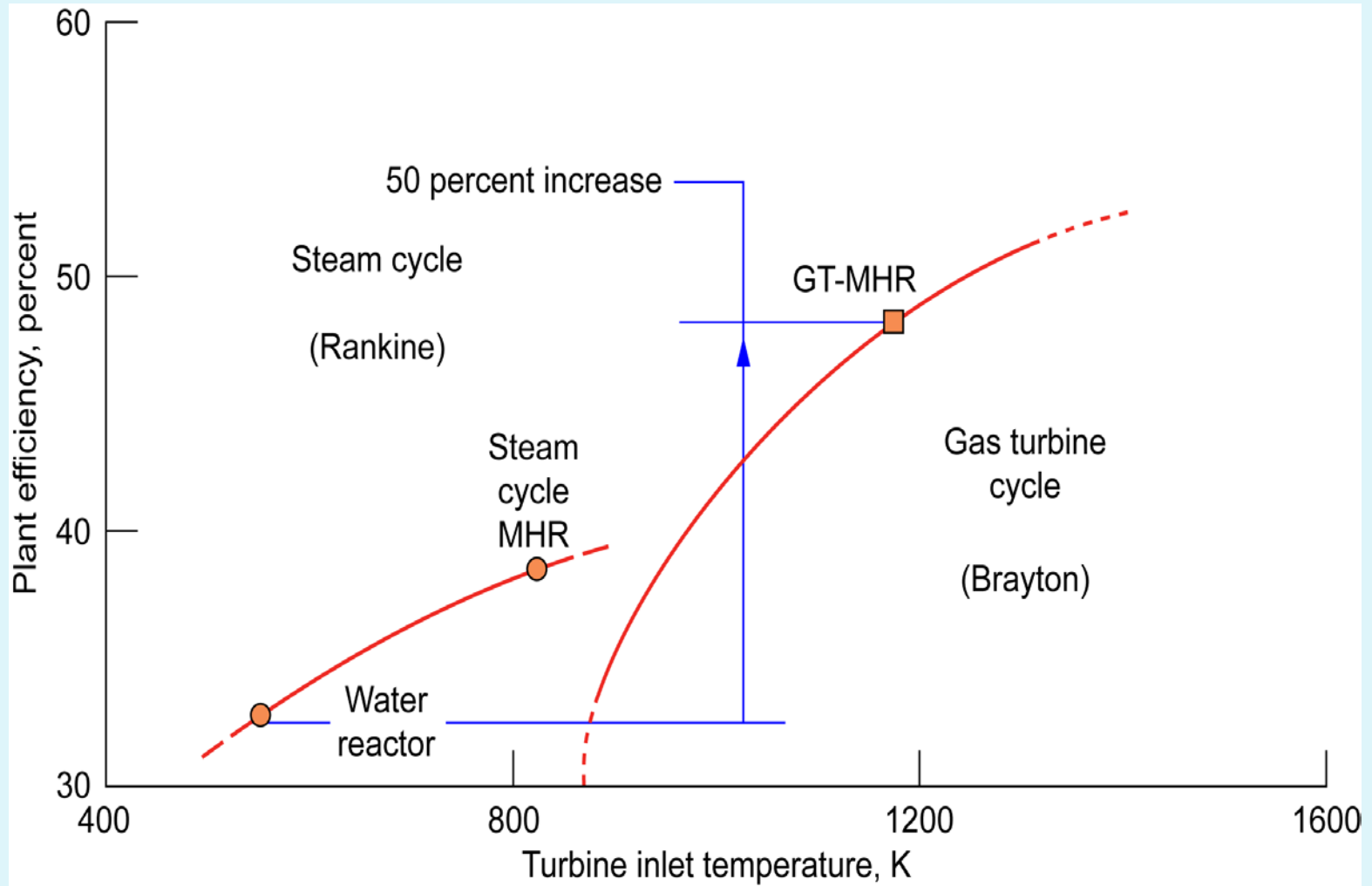
**Session 7 (10/22/19)**

**Albert J. Juhasz – PhD  
NASA - GRC, Cleveland, Ohio**

# Background & Motivation

- Advanced Nuclear Power named in Energy Policy Act of 2005 (Gen IV Nuclear Systems Initiative)
  - Gen. Baseload GW's Power -> Solve **Depletion of Earth's HC** (hydro-carbon) Resources
  - Ameliorate world climate problems by eliminating **Greenhouse Gas Emissions**
- Thorium Energy Independence & Security Act of 2008 (S.3680)
  - Thorium Power Plant Construction (100 GWe over next 25 yrs) would **re-invigorate US and World economies**
- Gen IV Candidate Advanced Nuclear Power Plants
  - Gas Turbine Modular He Reactor (GT-MHR) Systems - **Space**
  - Liquid Fluoride Thorium Reactor (LFTR) GT Systems - **Terrestrial**
- High Temperature Gas Turbine Power plants offer **large Thermal Efficiency improvement** over Steam plants
  - Amenable to formation of nuclear **micro-grid elements** housed in off-shore submarine hulls

# Energy Conversion Cycle Comparisons



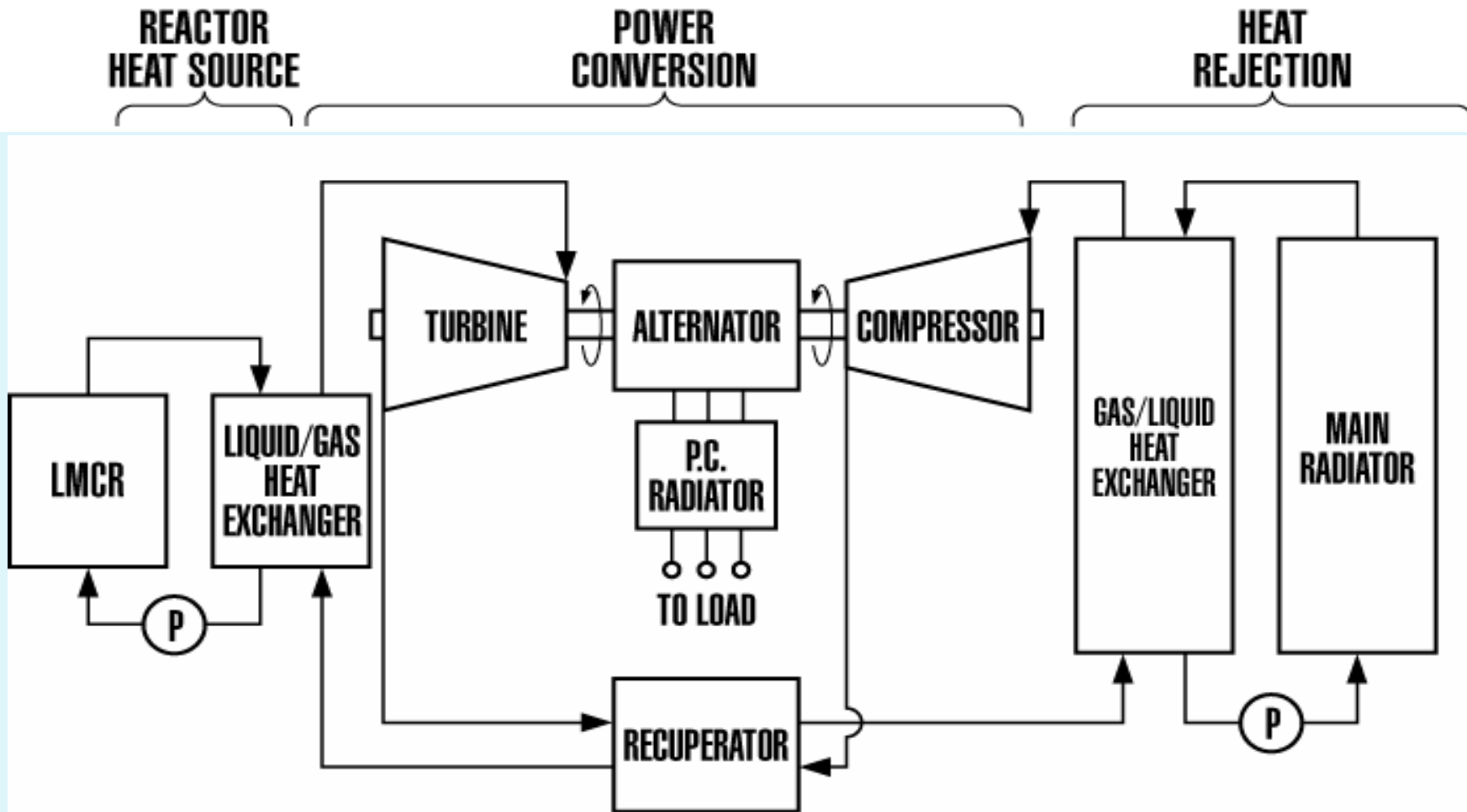
(La Bar , 2002)

# Requirements – i.e. Power System Design Drivers

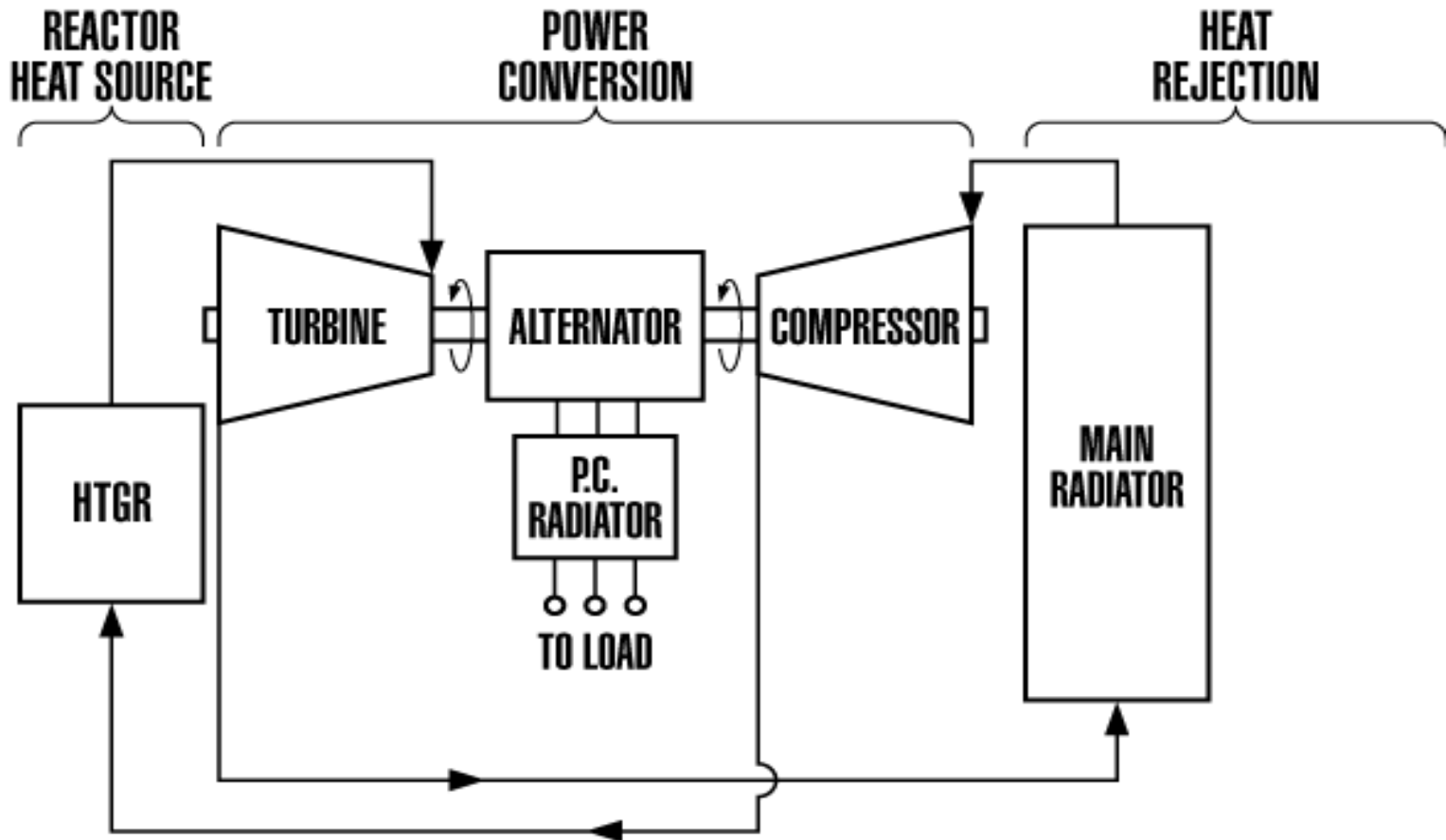
- Space (Lunar-Mars) Power Systems
  - MWe Power Levels require CBC (Closed Brayton Cycle) Conversion
  - Emphasis is on **Minimum System Mass**
  - **High System Reliability**, Autonomy and long Operational Life required to compensate for little or no maintenance
  - Need **least complex systems** w. minimum components
  - **Thermal Efficiency can be traded to achieve Low Mass**, i.e. non-regenerated and direct heated/cooled cycles eliminate heat exchanger (regenerator HX, HSHX, CSHX) mass at reduced Eff.
  - Location in permanently shadowed Lunar craters is ideal for CBC Nuclear Power systems, by enabling “**Low Temperature Heat Rejection**”, i.e. (**High Efficiency at low radiator area**).
- Terrestrial Nuclear Power Systems – e.g. LFTR Power Plant
  - Emphasis is on maximizing Thermal Efficiency,  $\eta_t$  and thus Power Output, Revenue, **Profit & Return on Investment**
  - **High Temp. Materials R&D** enables high TIT, and thus high  $\eta_t$
  - System Maintenance possible during regularly scheduled Periods
  - High System Mass and Complexity are acceptable as long as high Power Plant Availability/Reliability is assured
  - Ideal for Micro – Grid nuclei

# Space CBC Systems and Analysis

**In-direct Heat Input & Heat Rejection via Radiator for Regenerated Closed Brayton Cycle (CBC) Power System via Heat Exchangers**

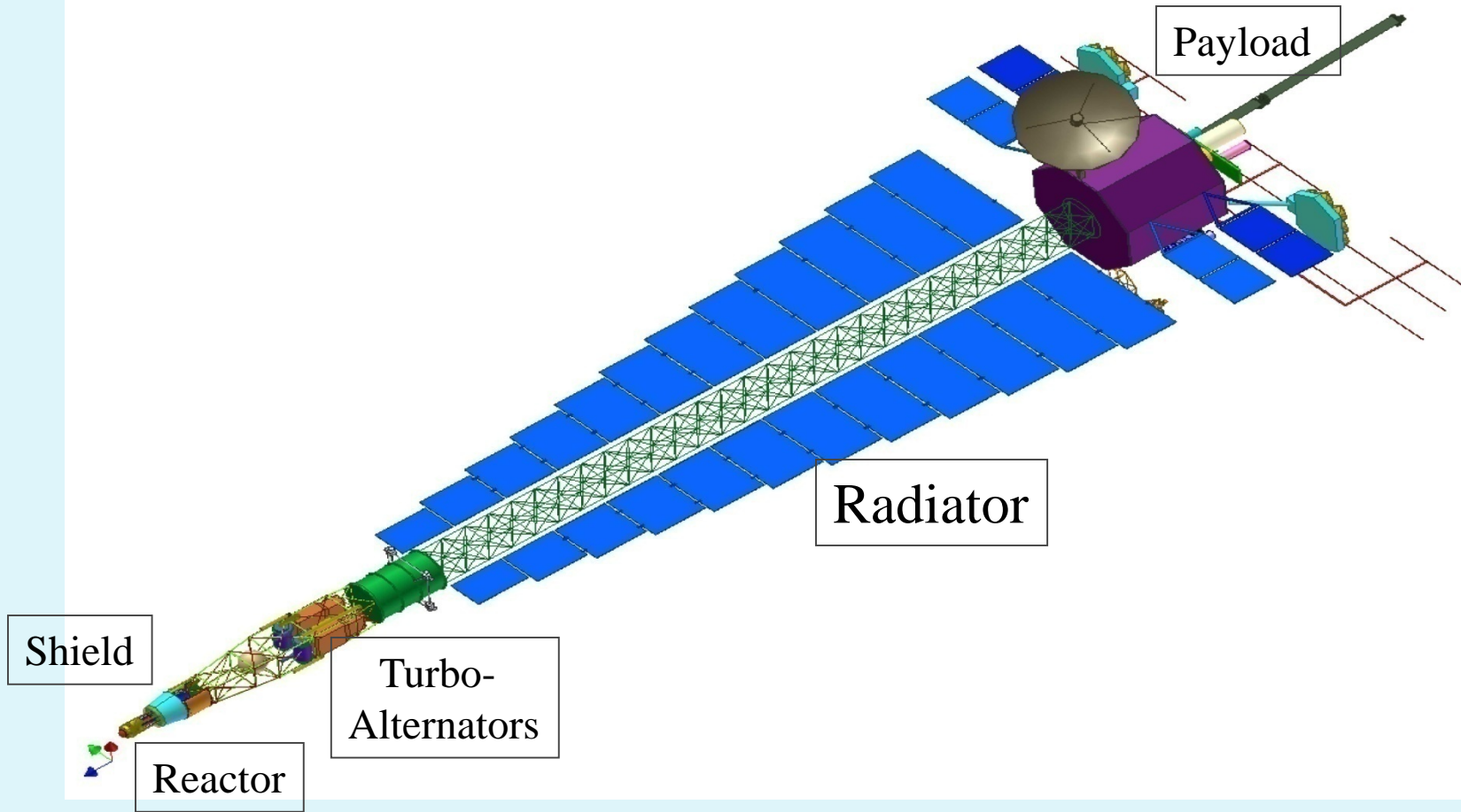


**Direct** Heat Input and Rejection via Radiator for **Non-Regenerated** Closed Brayton Cycle (CBC) Power System



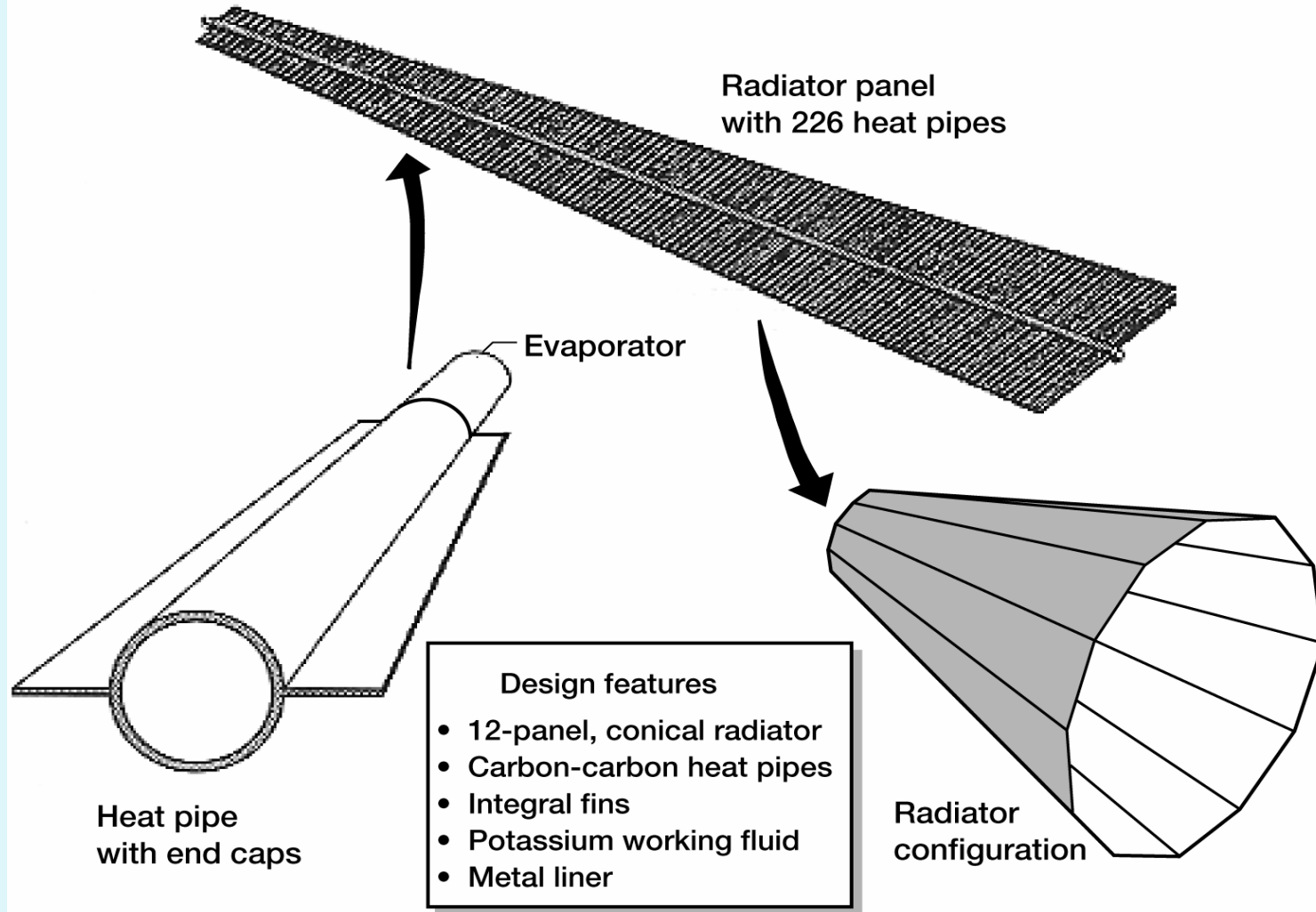
# Spacecraft with Trapezoidal Heat Pipe Radiator

(Ref. SP-100 Program)

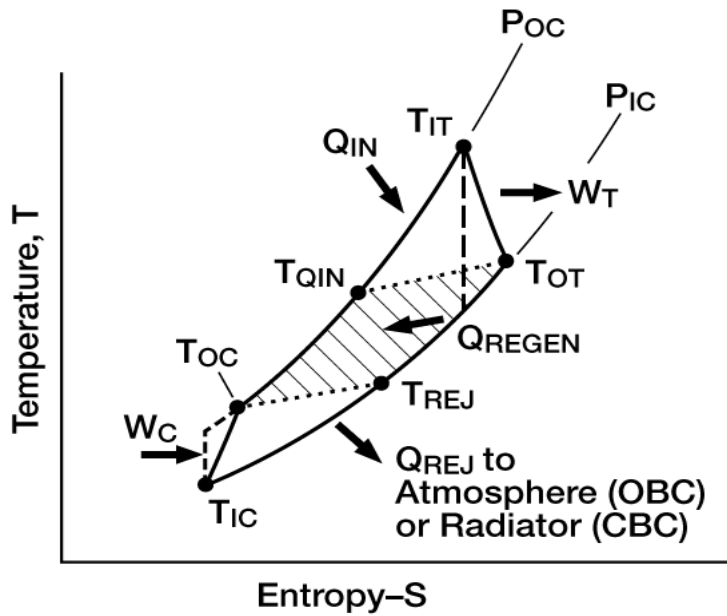
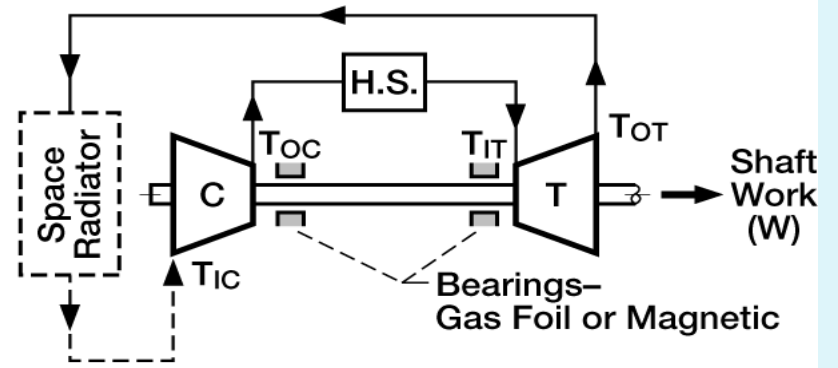
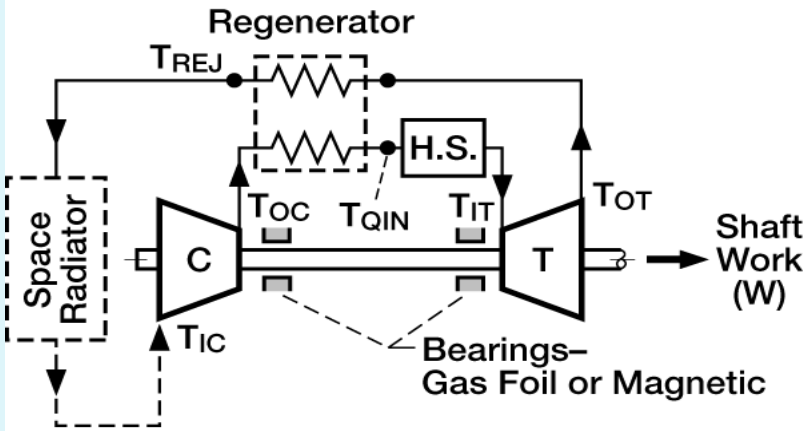




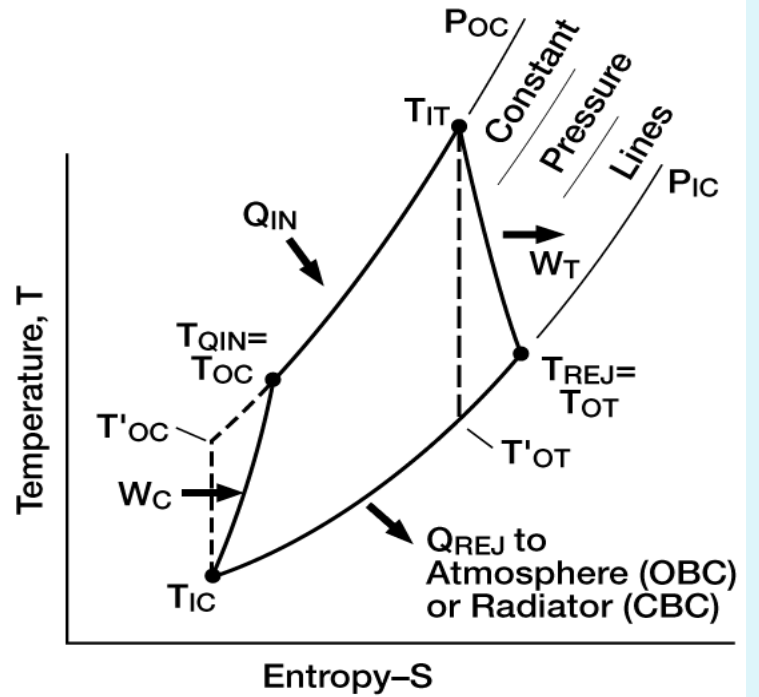
## SP – 100 Radiator Panel/Cone Configuration



# GAS TURBINE (BRAYTON CYCLES)



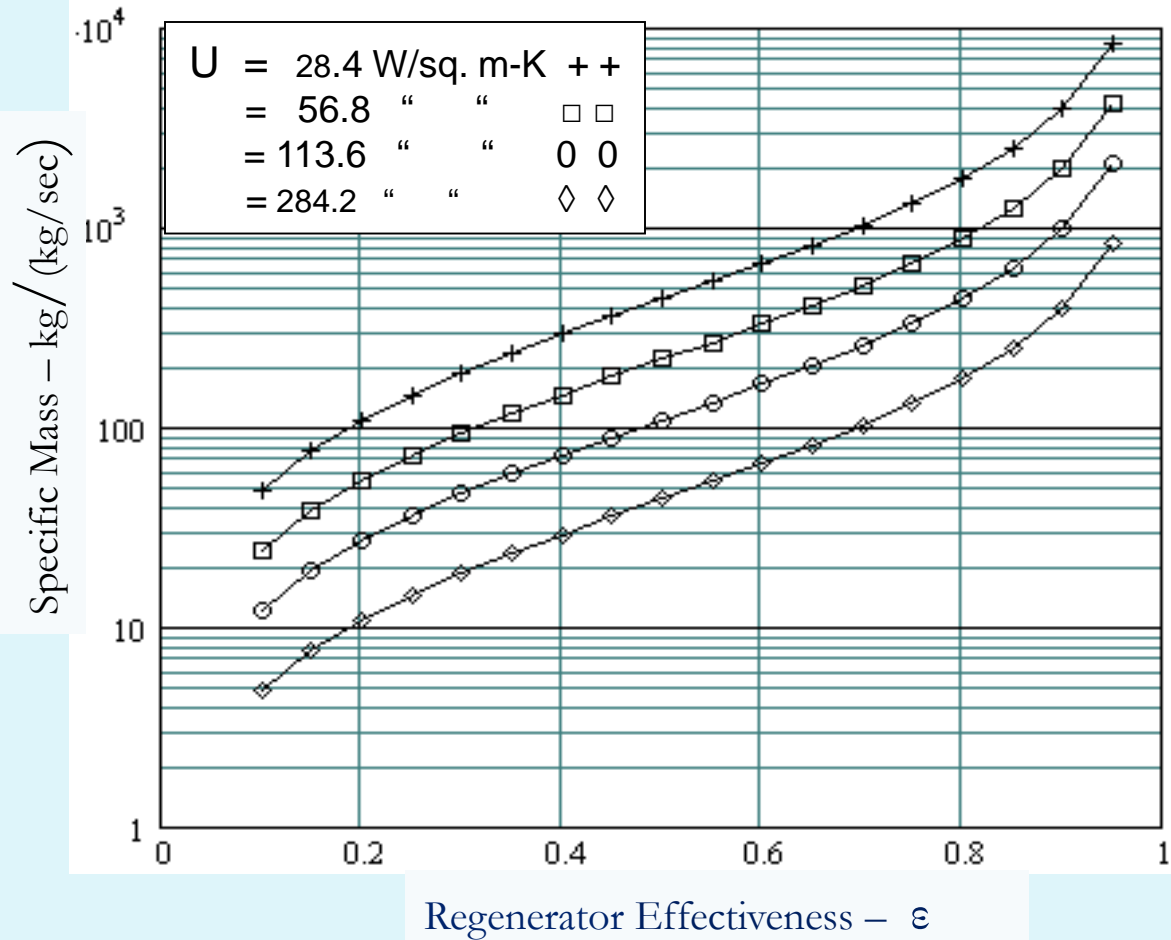
**Regenerated Cycle**



**Non-Regenerated Cycle**

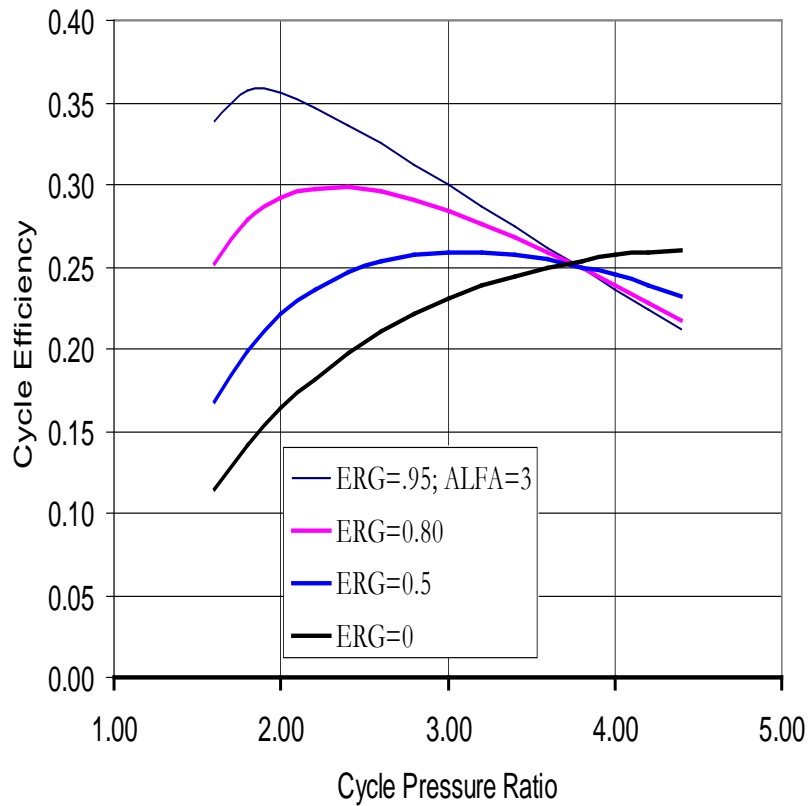
CD-04-82620

## Regenerator Specific Mass vs. Effectiveness with Heat Transfer Coefficient $U$ as a Parameter for He Working Fluid

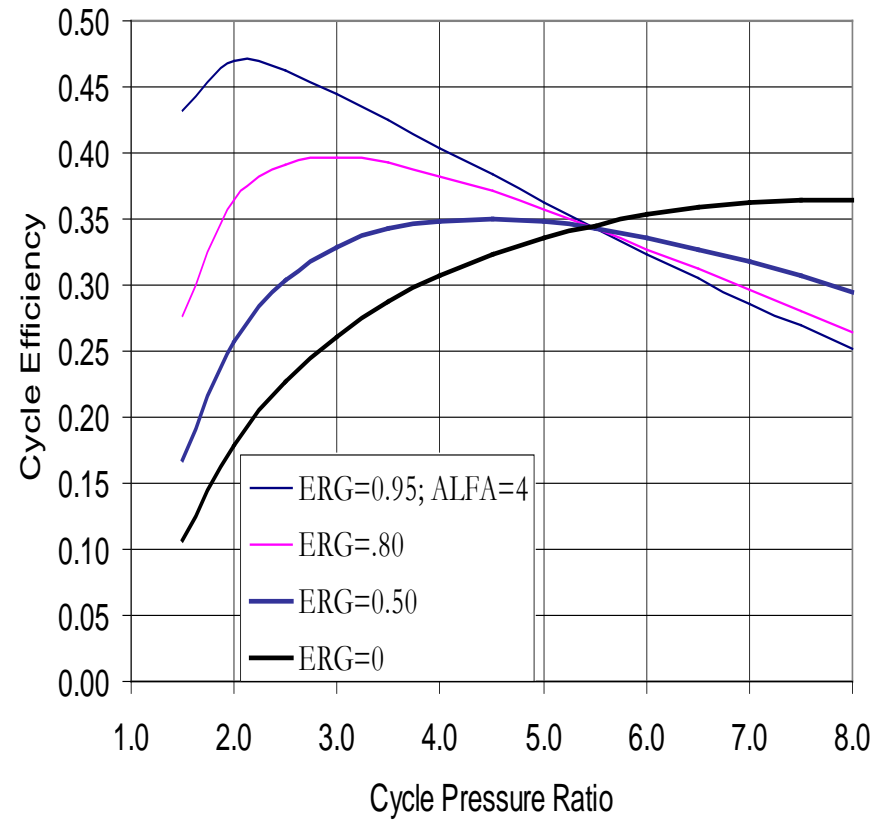


## Influence of Regenerator Effectiveness (ERG) on Cycle Efficiency at Cycle Temp. Ratio of 3.0 and 4.0

$$\eta_{PC} = \eta_{PT} = 0.9; \quad \gamma = 1.666$$

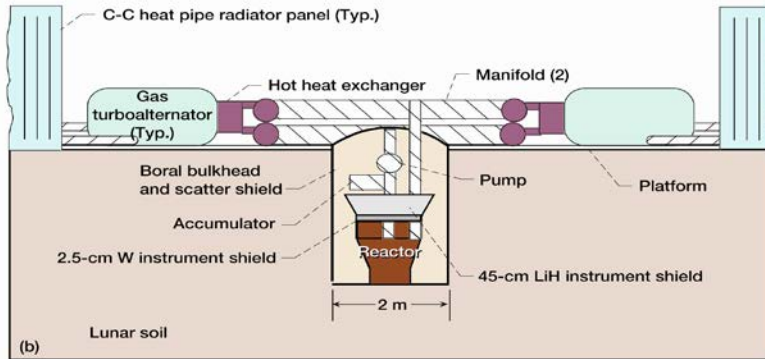
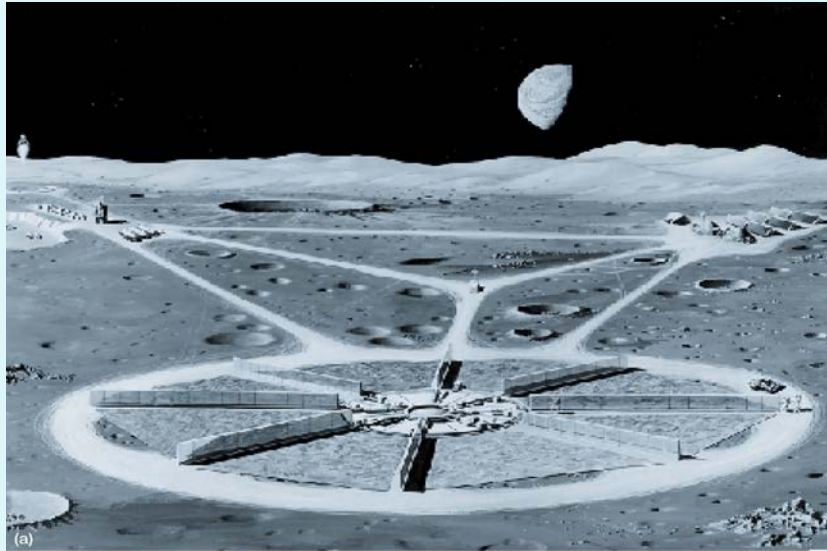


(a) Temp. Ratio = 3.0



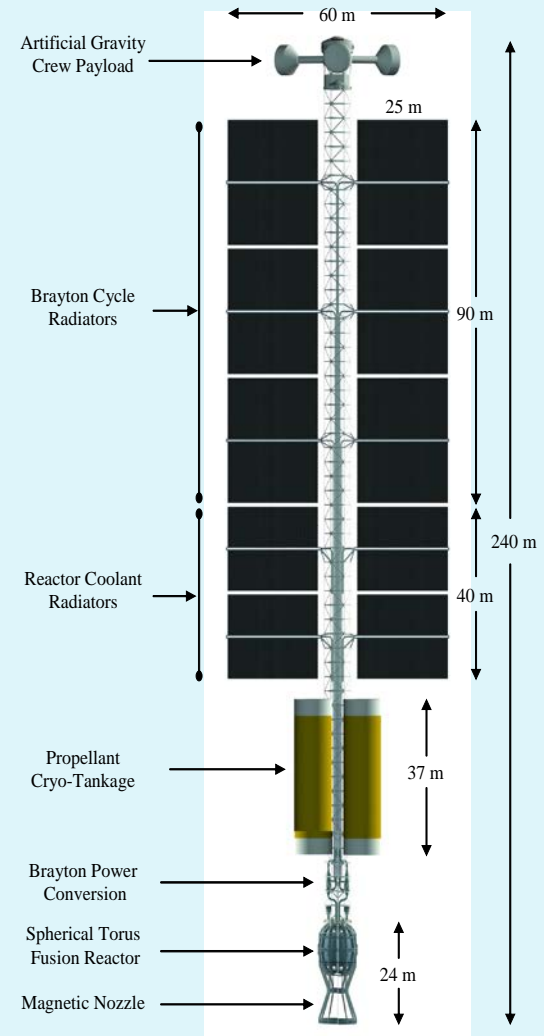
(b) Temp. Ratio = 4.0

# Advanced Power System Applications



Heat transport system  
 [Symbol] Primary  
 [Symbol] Secondary

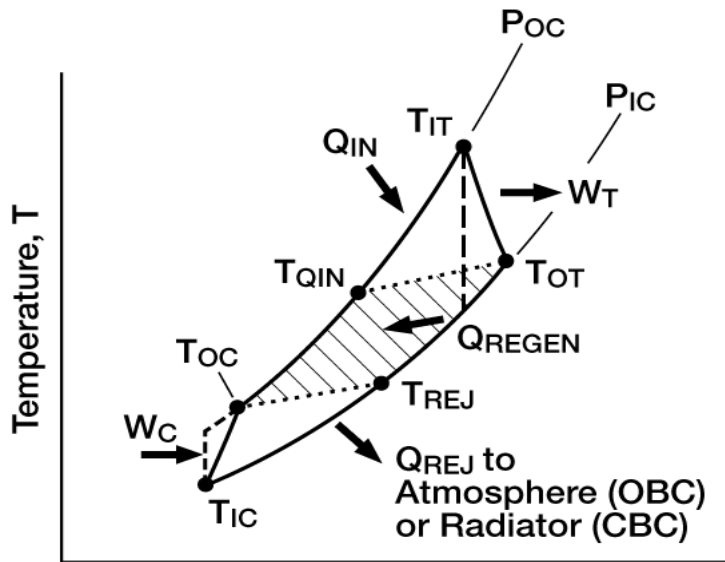
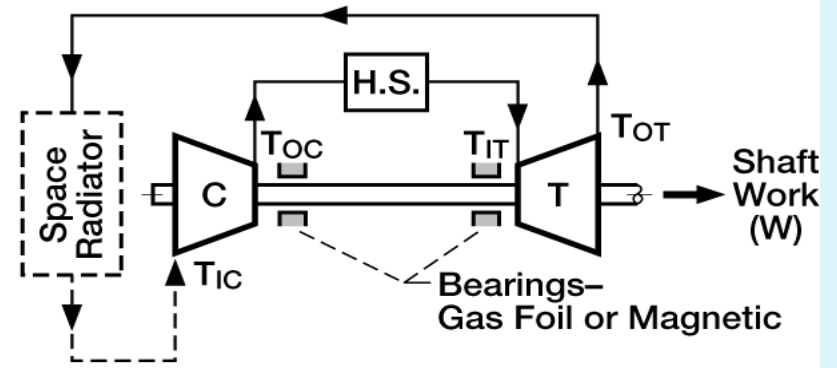
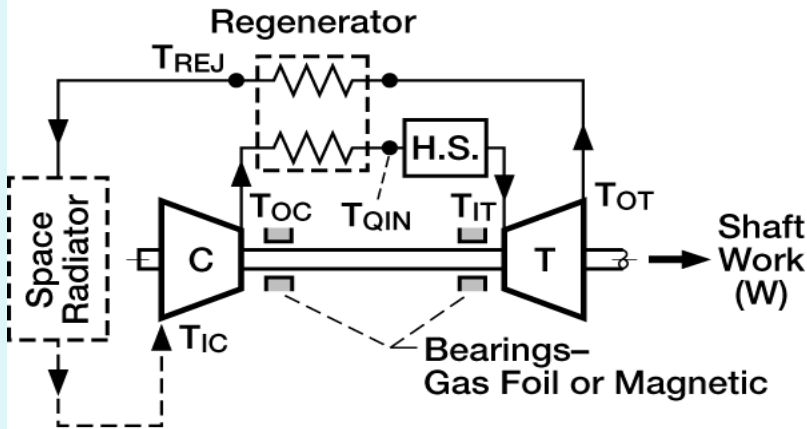
Lunar Base Power System



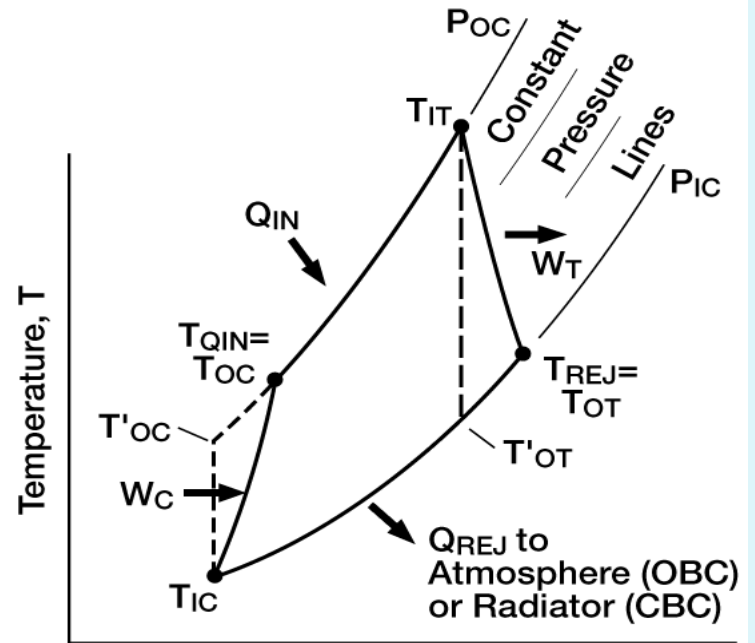
Interplanetary Fusion Propulsion Space Vehicle

# Ground Based CBC Systems

# Recall GAS TURBINE (BRAYTON CYCLES) - w/o RH & IC



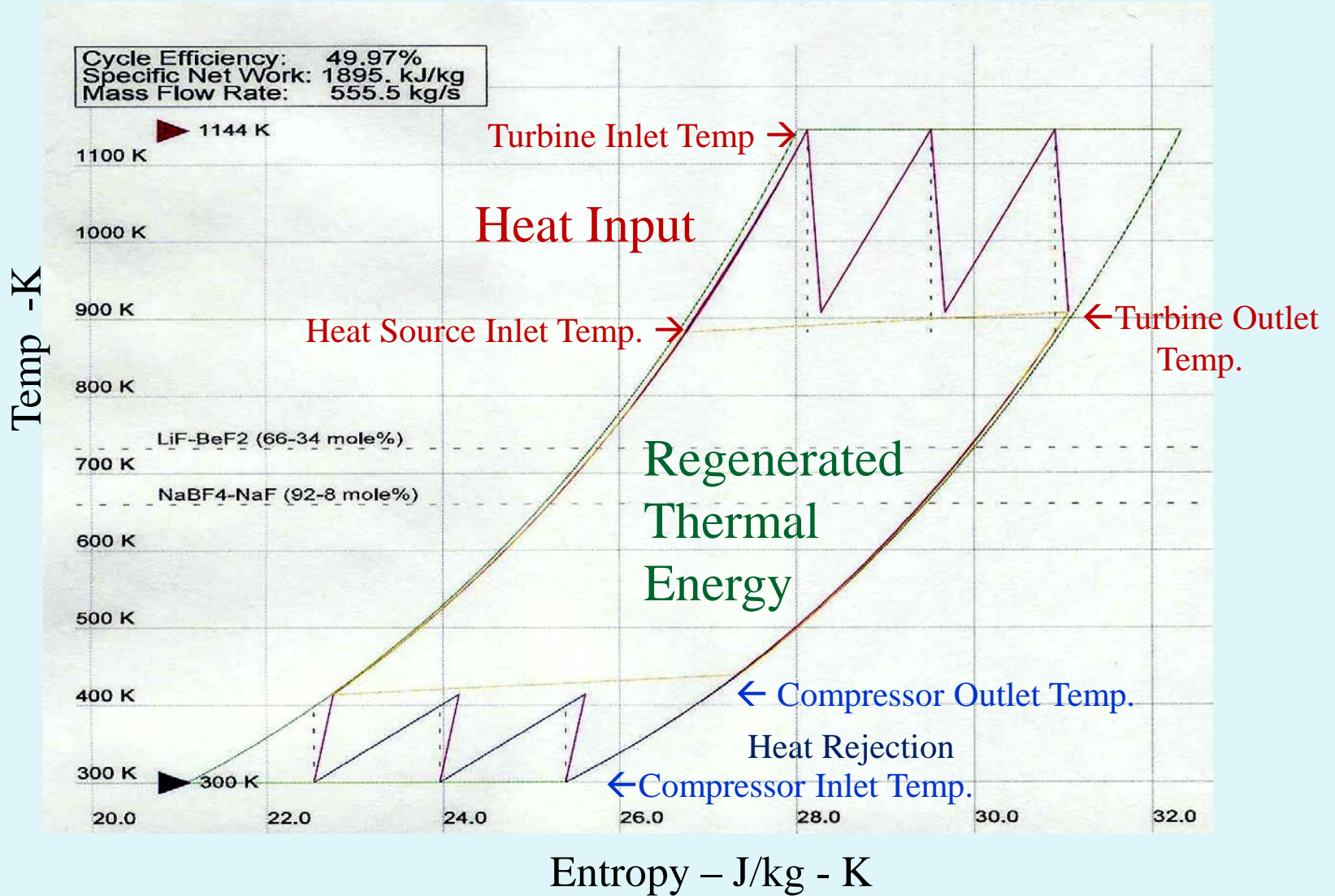
**Regenerated Cycle**



**Non-Regenerated Cycle**

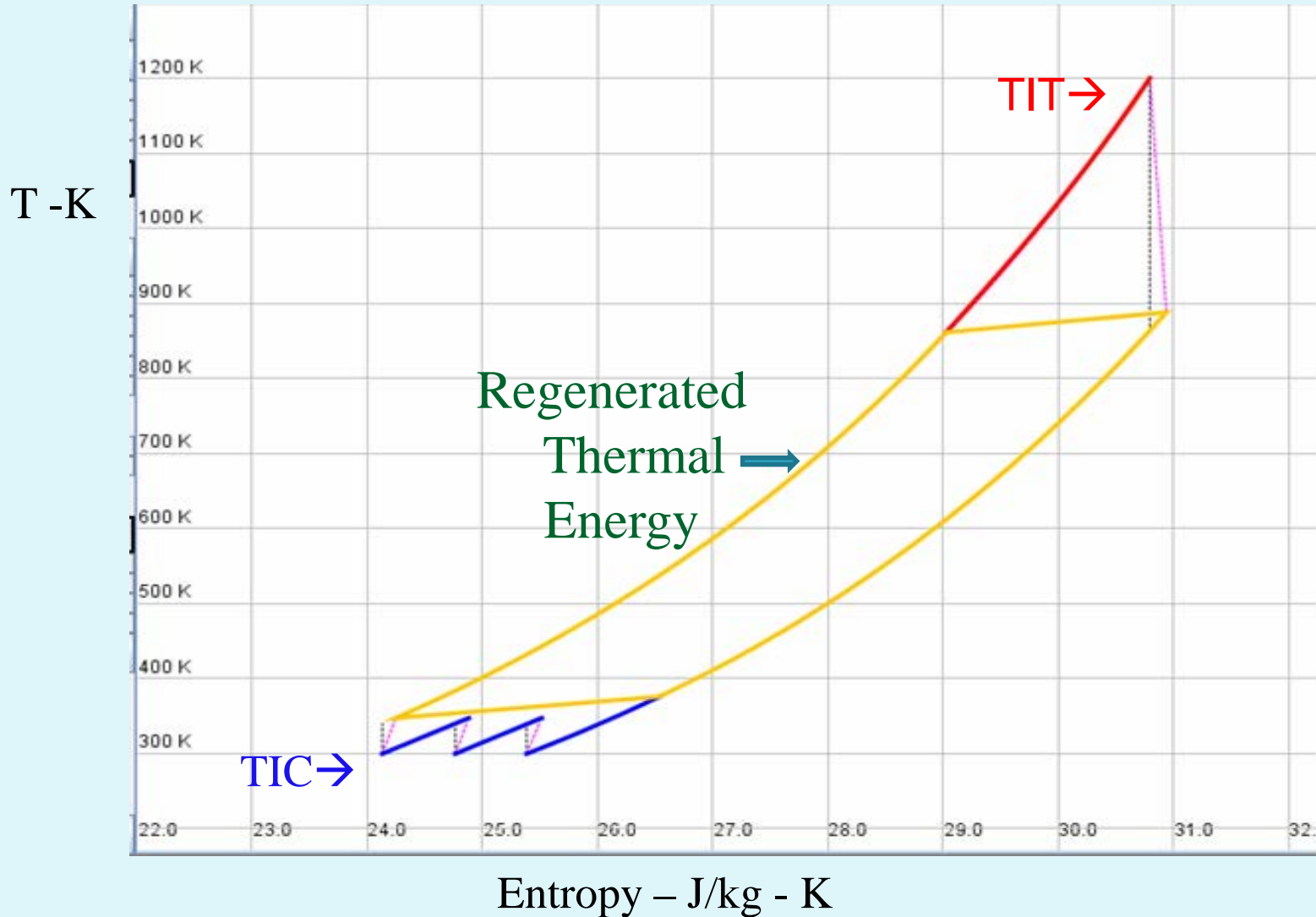
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# Three Stage Reheat & Intercool Brayton Cycle Temperature – Entropy Diagram

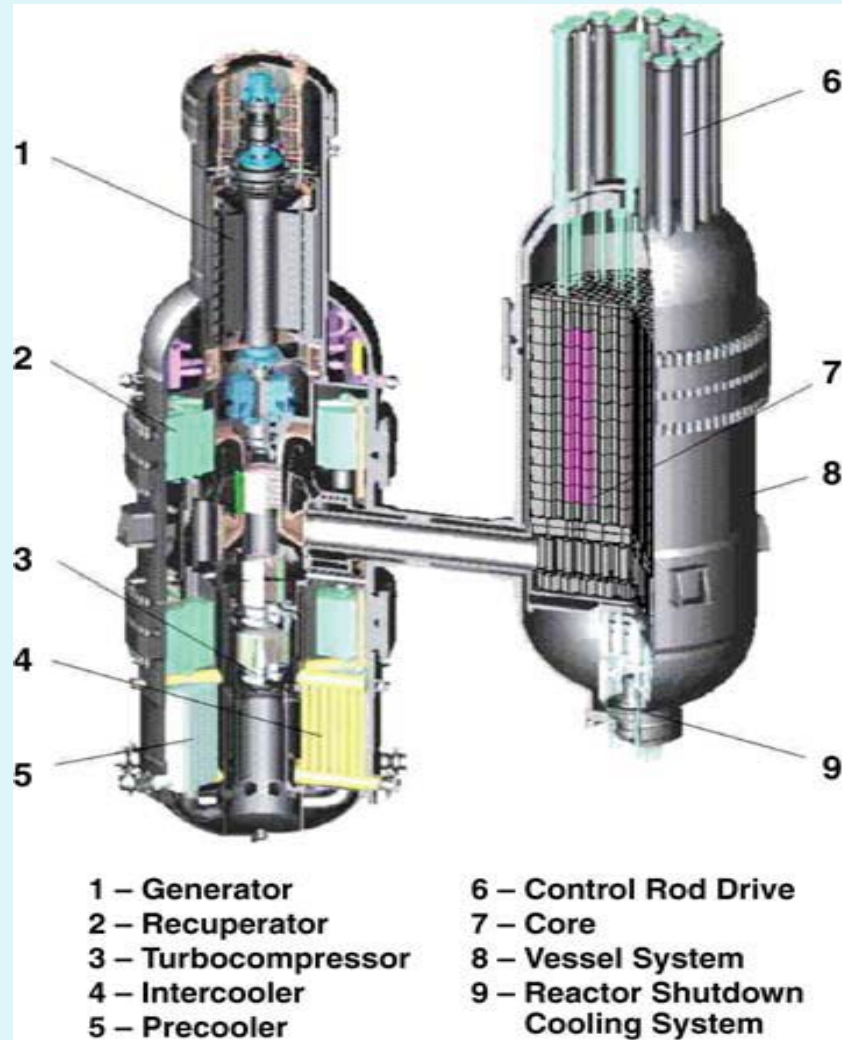




# Three Stage Intercool Only Brayton Cycle Temperature – Entropy Diagram

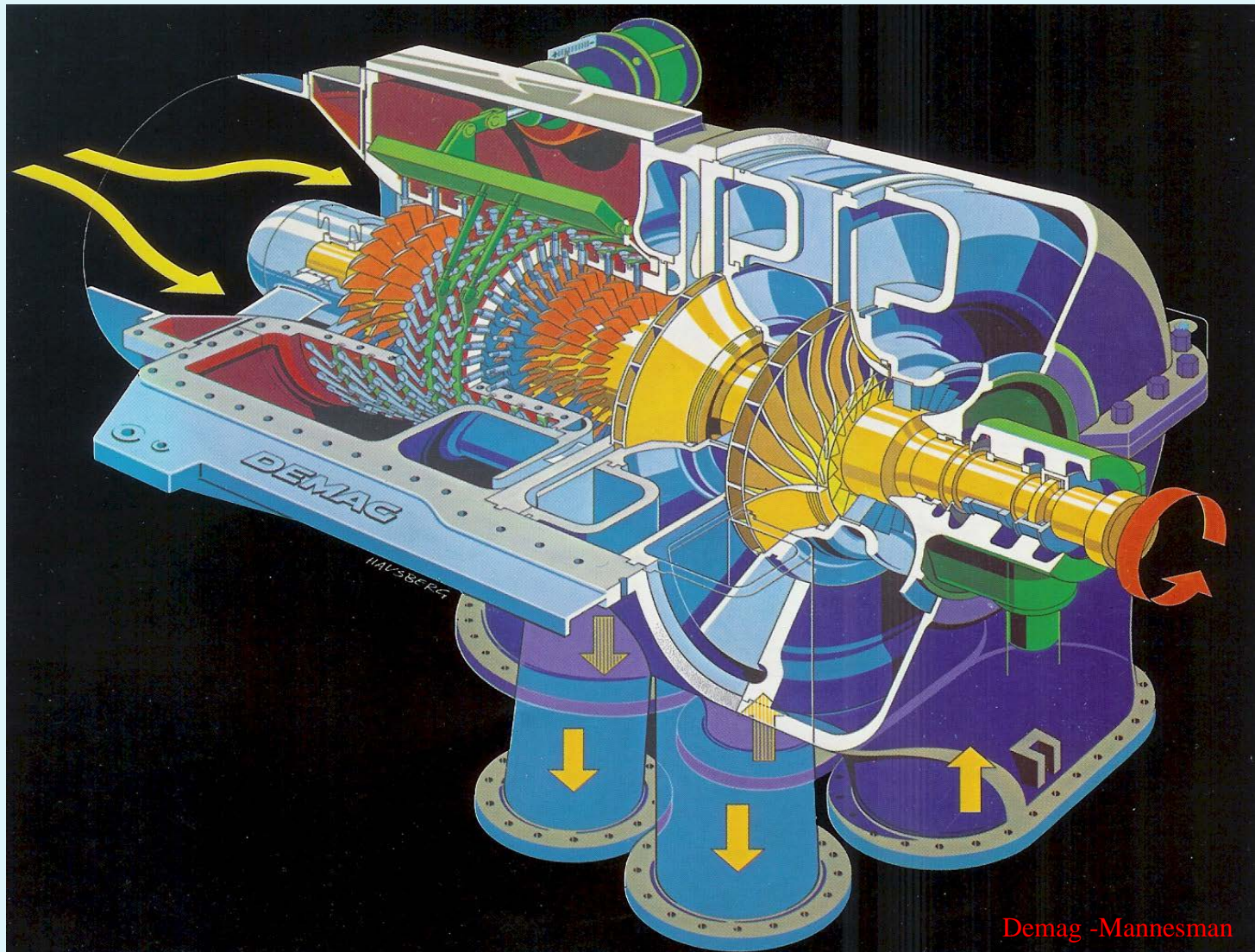


# Proposed Vertical Orientation of GT-MHR\* Turbomachinery



\* Gas Turbine Modular Helium Reactor

# Typical Axial Radial Turbo-compressor

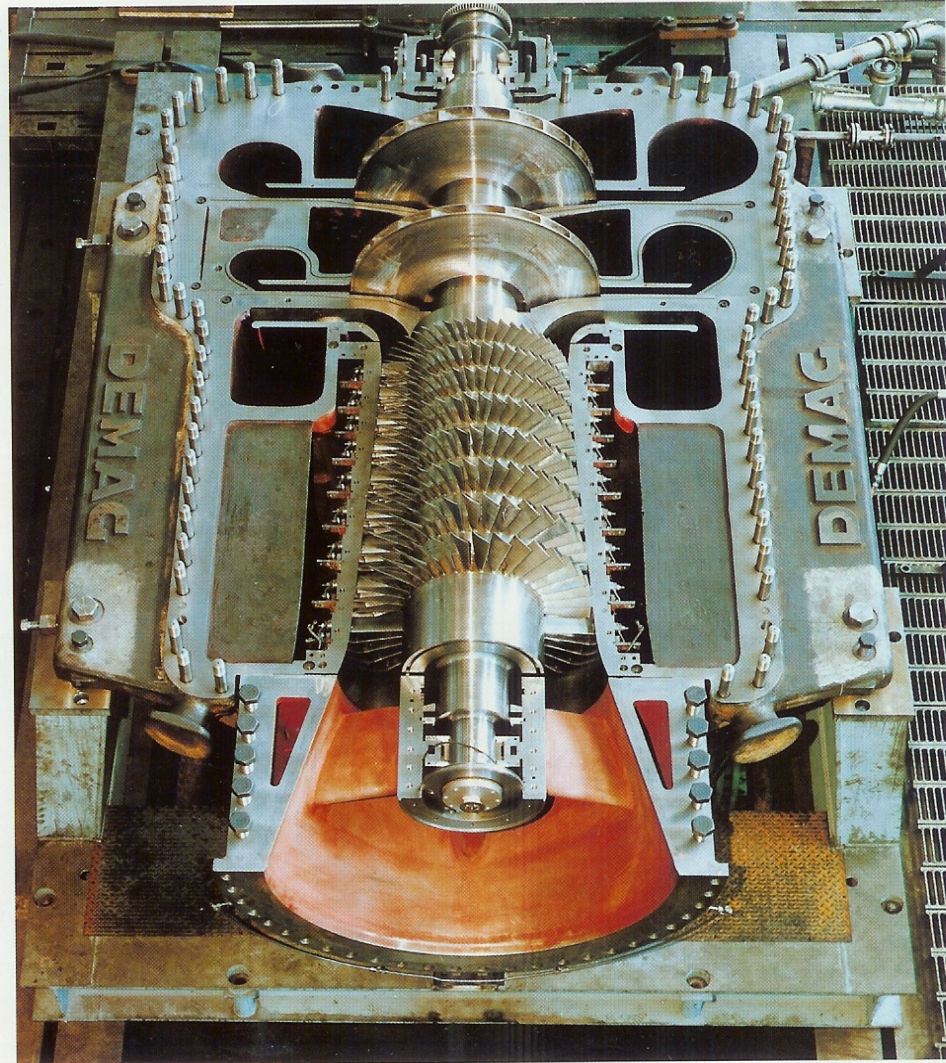


Demag -Mannesman

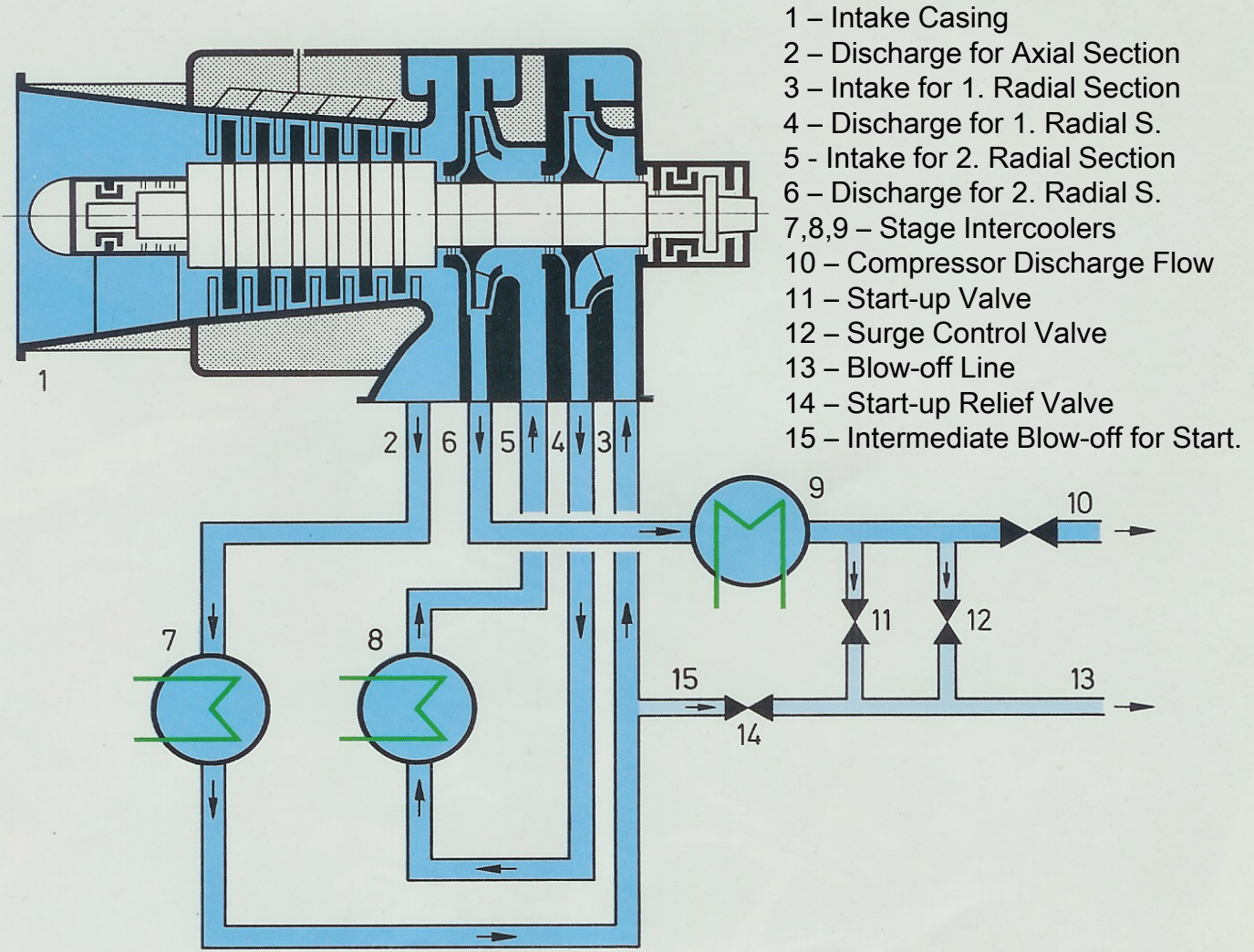
# Axial/Radial Compressor with Axial Intake

Type AR 250-8-2 axial-centrifugal  
compressor with axial intake

Medium	Air
Flow volume	272 000 m <sup>3</sup> /h
Intake pressure	1.01 bar
Discharge pressure	6.4 bar
Rotational speed	4 550 rpm
Drive rating	22 700 kW (turbine)



# Flow Diagram for AR Compressor with Axial Intake



- 1 – Intake Casing
- 2 – Discharge for Axial Section
- 3 – Intake for 1. Radial Section
- 4 – Discharge for 1. Radial S.
- 5 – Intake for 2. Radial Section
- 6 – Discharge for 2. Radial S.
- 7,8,9 – Stage Intercoolers
- 10 – Compressor Discharge Flow
- 11 – Start-up Valve
- 12 – Surge Control Valve
- 13 – Blow-off Line
- 14 – Start-up Relief Valve
- 15 – Intermediate Blow-off for Start.

# Energy from Thorium

via

## Liquid Fluoride **Thorium Reactor**- LFTR for Terrestrial Power

References: Kirk Sorensen, W. Thesling  
R. Hargraves – “Thorium energy cheaper than coal”

# Thorium and Uranium Abundance in the Earth's Crust

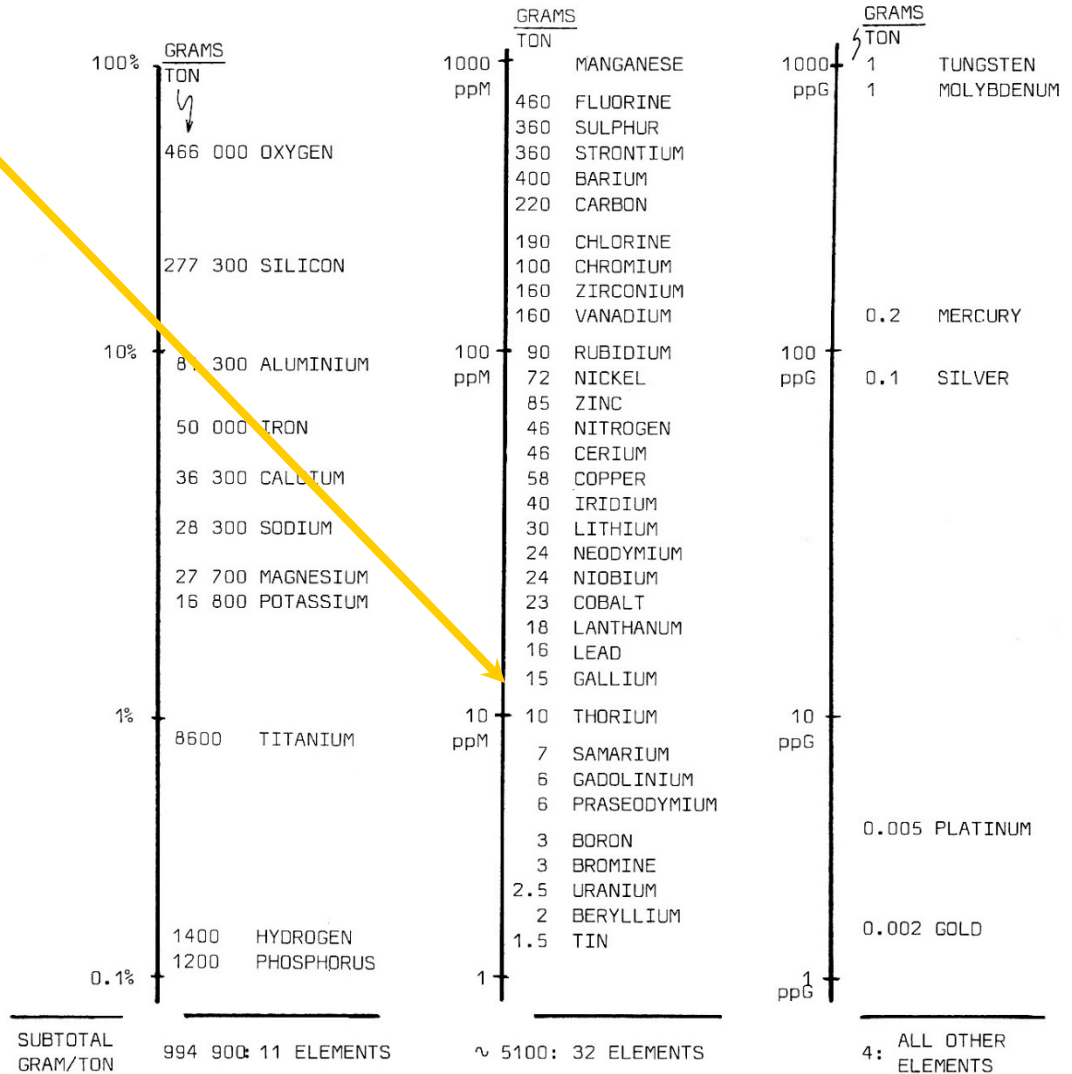
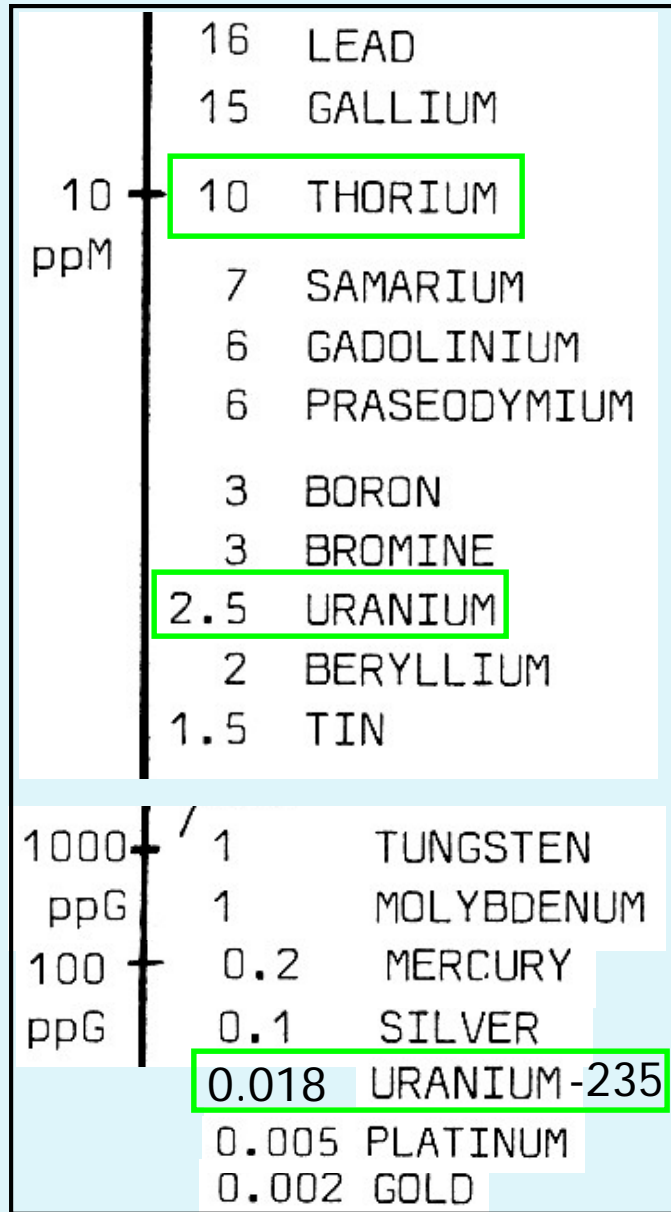
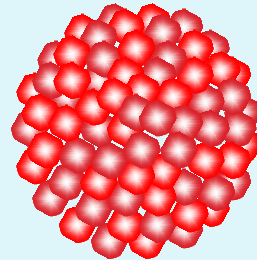


Fig. 5.13. The chemical composition of the Earth's crust.

# Thorium<sub>232</sub> - Uranium<sub>233</sub> Breeding Cycle

Thorium-233 decays quickly (half-life of 22.3 min) to protactinium-233 by emitting a beta particle (i.e. an electron).

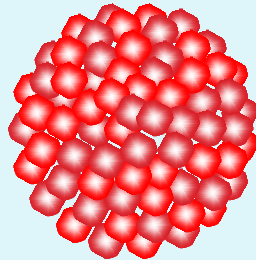


Pa-233  
New Element

Protactinium-233 decays more slowly (half-life of 27 days) to uranium-233 by emitting a beta particle (an electron).

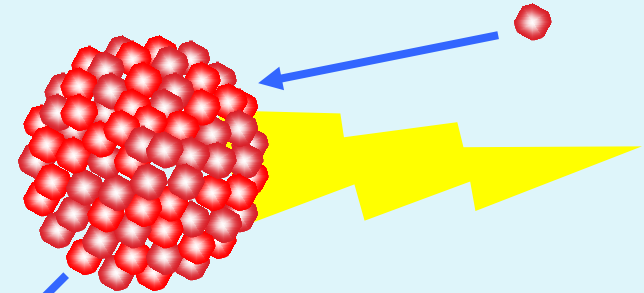
It is important that Pa-233 NOT absorb a neutron before it decays to U-233—it should be shielded from any neutrons until it decays.

Th-233  
(New Iso-  
tope)

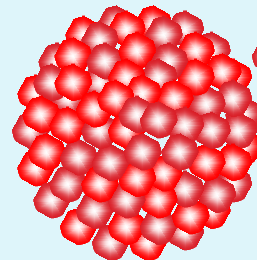


Thorium-232 absorbs a neutron from fission and becomes thorium-233.

U-233  
(Fissile)



Uranium-233 is fissile and will fission when struck by a neutron, releasing energy and 2 to 3 neutrons. One neutron is needed to sustain the chain-reaction, one neutron is needed for breeding, and any remainder can be used to breed additional fuel.



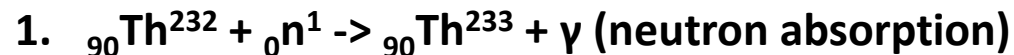
Th-232  
(Fertile)  
24



# Thorium –Uranium Fuel Cycle

## Three Step process

Step 1 - Change of **Atomic Mass (Isotope)** via **neutron** absorption



Steps 2 & 3 - Change of **Atomic Number (Element)** via  **$\beta$  decay**



\*(where  $\lambda$  is Decay Constant  
HL is Half Life)

)

## Time for 99.9% Beta Decay of Protactinium to U-233

- **Looking at Step 3**, the time required for 99.9% of Pa<sup>233</sup> decaying to U<sup>233</sup> and 0.1% remaining Pa<sup>233</sup>, let  $N(t) = 0.1$  and  $N_0 = 99.9$  in eq. (1)

$$N(t) = N_0 e^{-\lambda t} \quad (1)$$

where  $\lambda$  is the *decay constant* computed from

$$\lambda = (\ln .5)/T_{0.5}, \text{ with } T_{0.5} \text{ being the } \textit{Half Life} = 27 \text{ days}$$

- So  $\lambda = -0.693/27 = -0.02567 \text{ days}^{-1}$

Substituting in (1) :  $0.1 = 99.9 e^{-0.02567 * t} \quad (2)$

Dividing (2) by 99.9  $\sim 100$ , and taking the **ln** of both sides, we have

$$\ln .001 = - .02567 t$$

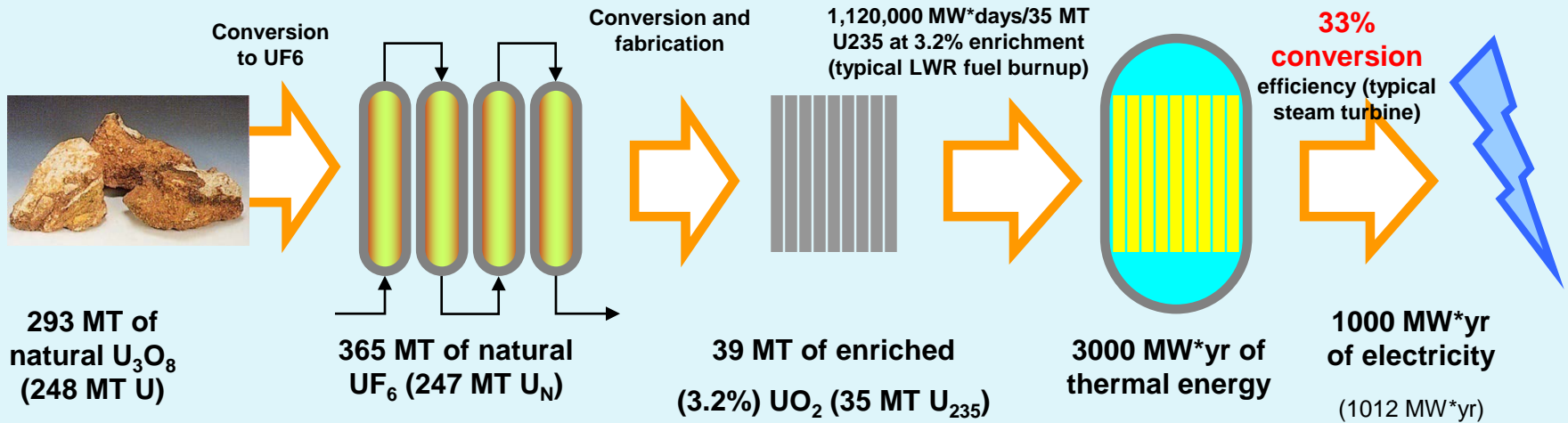
$$t = - 6.9077 / - .02567$$

$$= 269 \text{ days, or } \sim 9 \text{ months for } 99.9\% \text{ of Pa} \rightarrow \text{U}^{233}$$

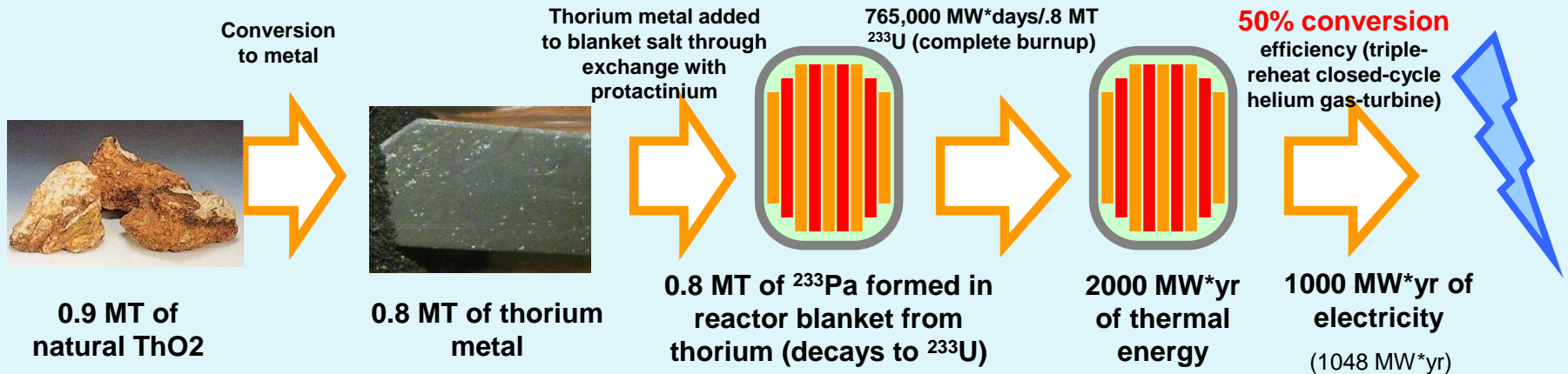
- **Note that for 99 % transmutation to U<sup>233</sup> only 179.5 days would be required**

# Energy Extraction Comparison for $U_{238}$ and $Th_{232}$

## Uranium-fueled light-water reactor: 35 GW\*hr/MT of natural uranium



## Thorium-fueled liquid-fluoride reactor: 11,000 GW\*hr/MT of natural thorium



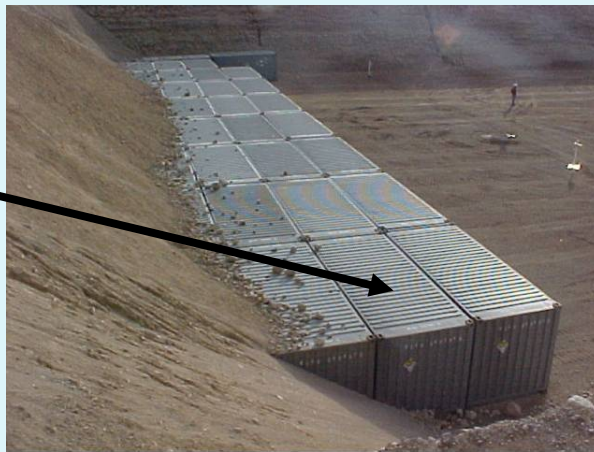
# Thorium: Virtually **Limitless** Energy

## World Thorium Resources

Country	Reserve Base (tons)
Australia	340,000
India	300,000
<b>USA</b>	<b>300,000</b>
Norway (“Thorcon” in Oslo)	180,000
Canada	100,000
South Africa	39,000
Brazil	18,000
Other countries	100,000
World total	1,400,000

Source: U.S. Geological Survey, Mineral Commodity Summaries, January 2008

- Thorium is abundant around the world:
  - Found in trace amounts in most rocks and soils
  - India, Australia, Canada, US have large minable concentrations. (Main Ore “**Monazite**”)
  - US has about 20% of the world reserve base
- No need to hoard or fight over this resource:
  - A single mine site in Idaho could produce 4500 MT of thorium per year
  - Replacing the total US electrical energy consumption would require ~400 MT of thorium



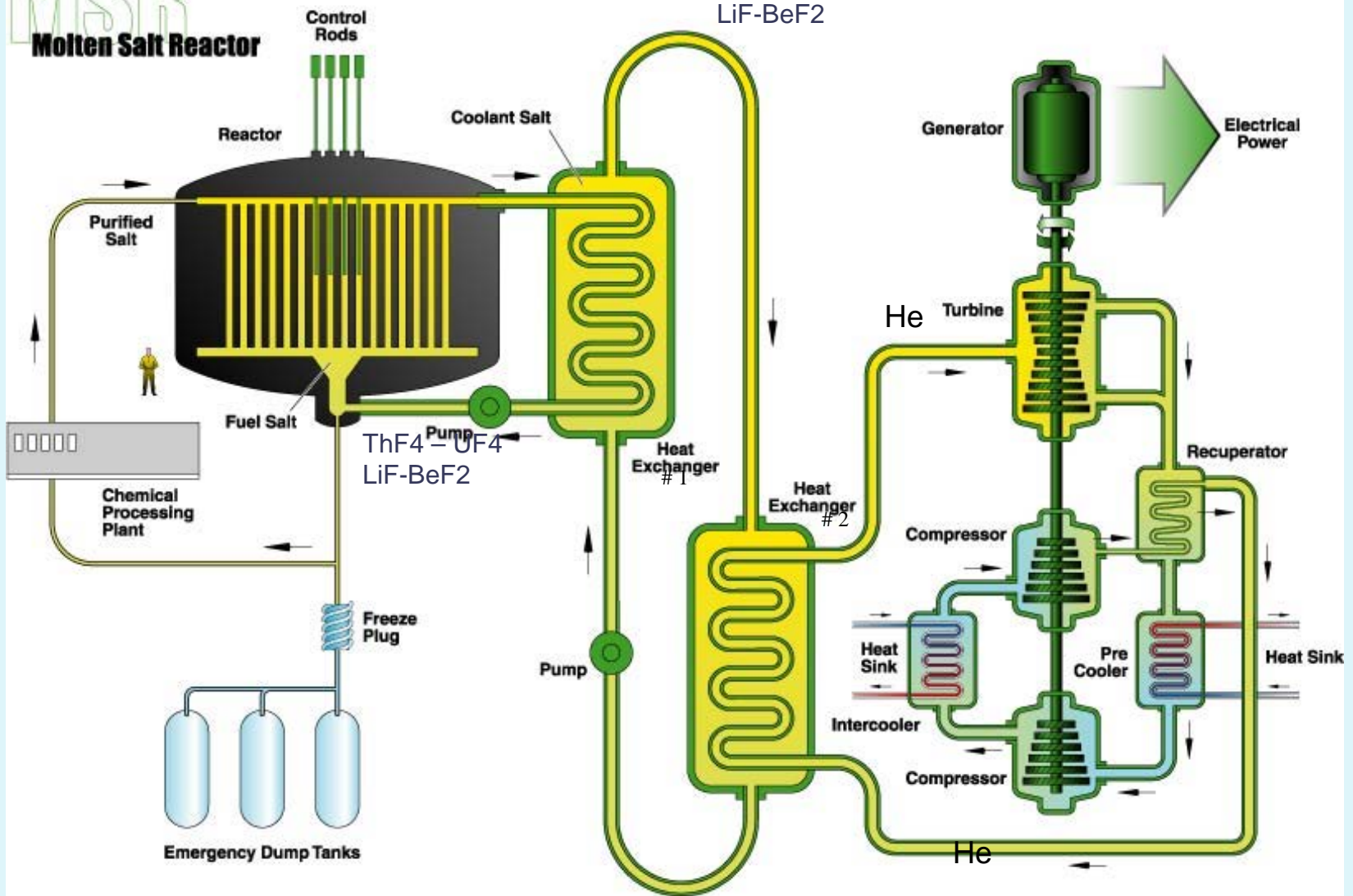
**The United States has buried 3200 metric tonnes of thorium nitrate in the Nevada desert.**

**There are 160,000 tonnes of economically extractable thorium in the US, even at today’s “worthless” prices!**

# Gen-4 Liquid 2 Salt Configuration Reactor Concept – ORNL

Developers : Jerome Wigner and Saul Weinberg

**MSR**  
**Molten Salt Reactor**



# CBC Energy Conversion System Analysis

(with Comp. IC & Turbine Re-Ht)

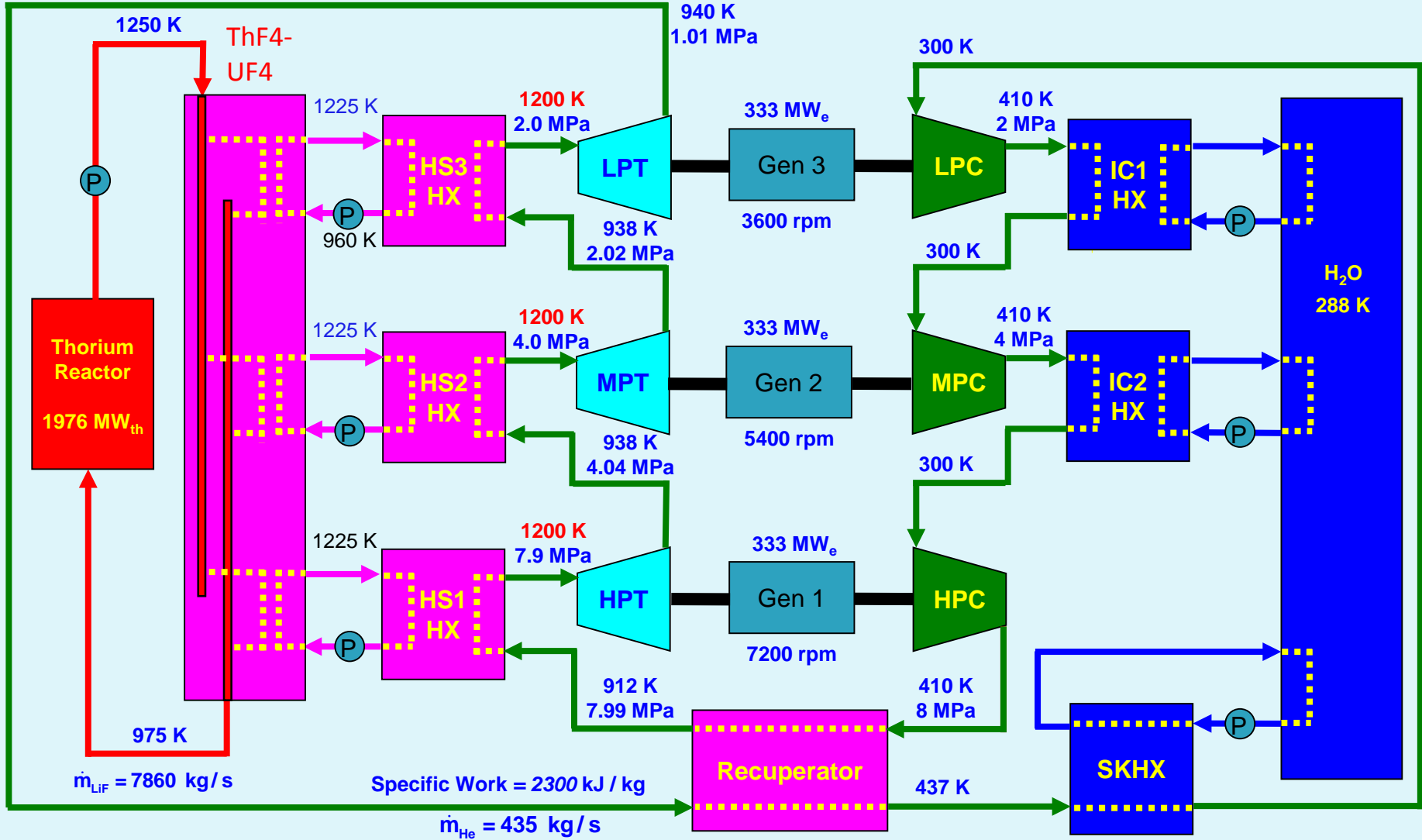
## Key Cycle Input Parameters

- Compressor Inlet Temperature (TIC), K 300
- Cooling Water Temperature, K 288
- Reactor Heat Loss, percent 1.0
- Polytropic Efficiency—Compressor, percent 86
- Polytropic Efficiency—Turbine, percent 92
- Recuperator Effectiveness, percent 95
- Intercooler HX Pressure Loss, percent 0.5
- Reheat HX Pressure Loss, percent 0.8
- Turbine Pressure Ratio Fraction, percent 96
- Generator Efficiency, percent 98

Temp Ratio = 4.0  
 Efficiency = 50.6 %  
 Turbine Power = 1758 MW  
 Compressor Power = -738 MW

**1000 MWe Power Plant 2 Salt Configuration**  
**Thorium Molten Salt Reactor**  
**He CBC w. Rht. & Intcl.- 1200 K Turbine Inlet Temp**

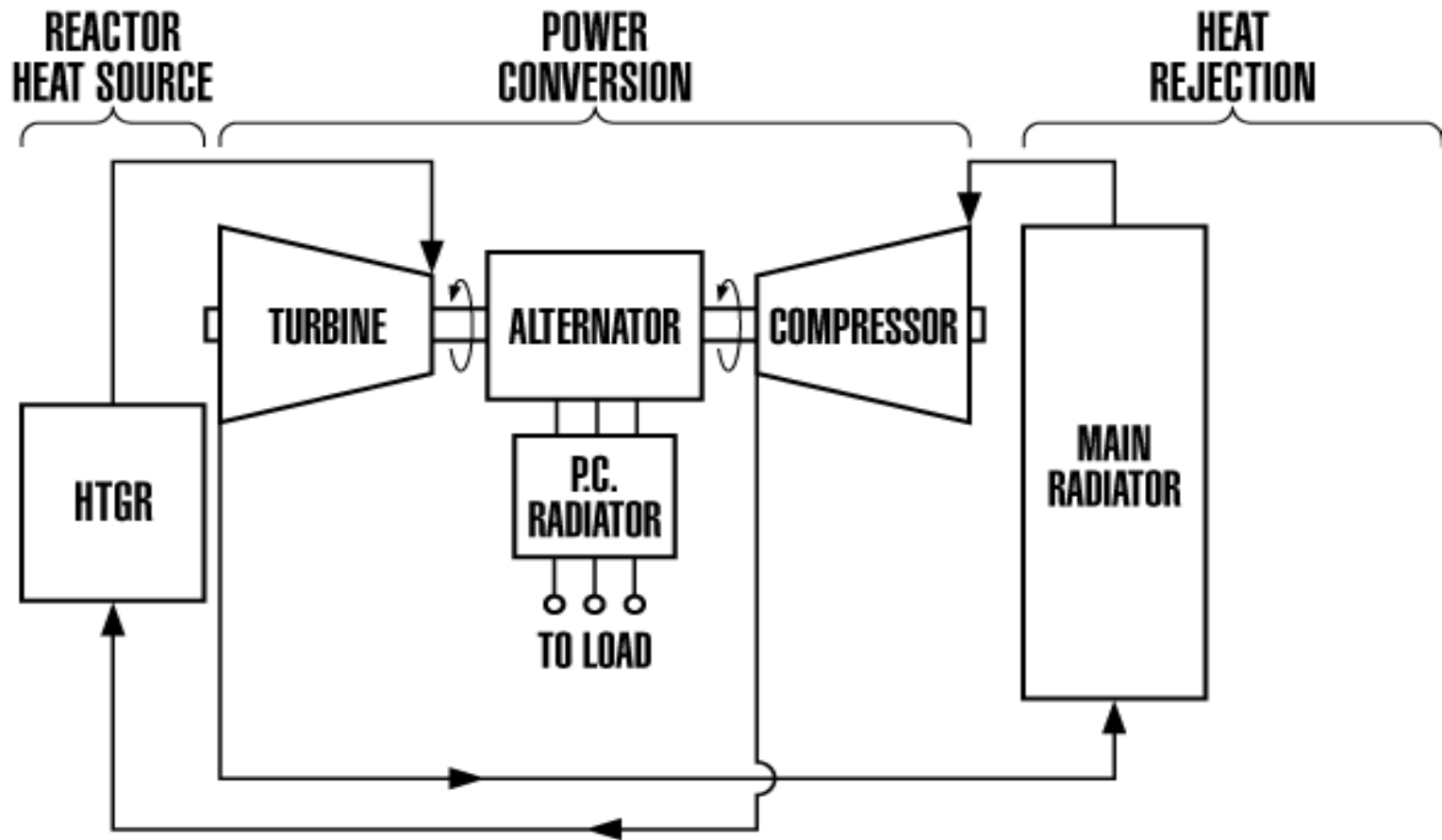
He ---  
 LiF - BeF2  
 ThF4-UF4





- Comparison to Space System-

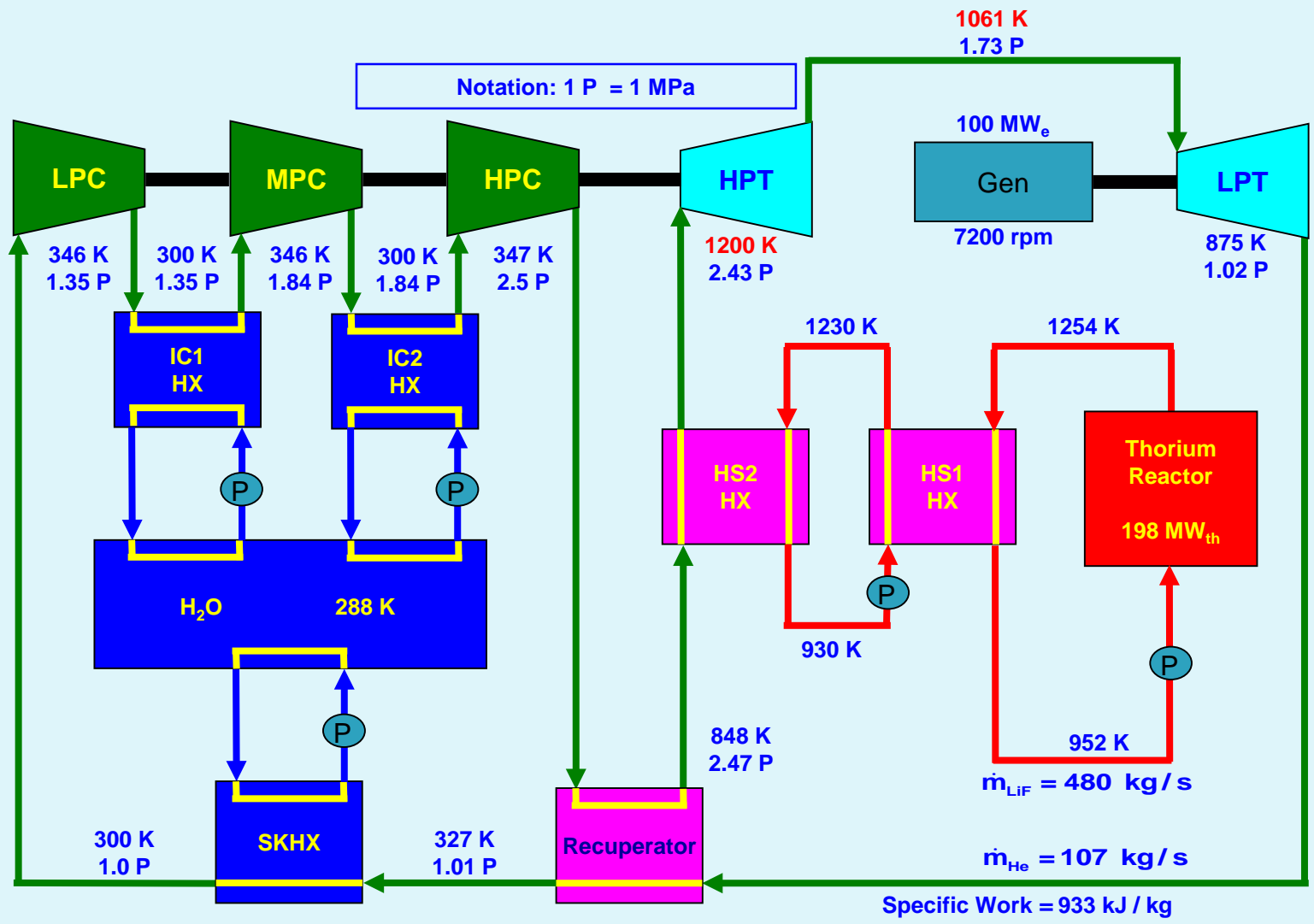
Direct Heat Input and Rejection via Radiator for Non-Regenerated Closed Brayton Cycle (CBC) Power System



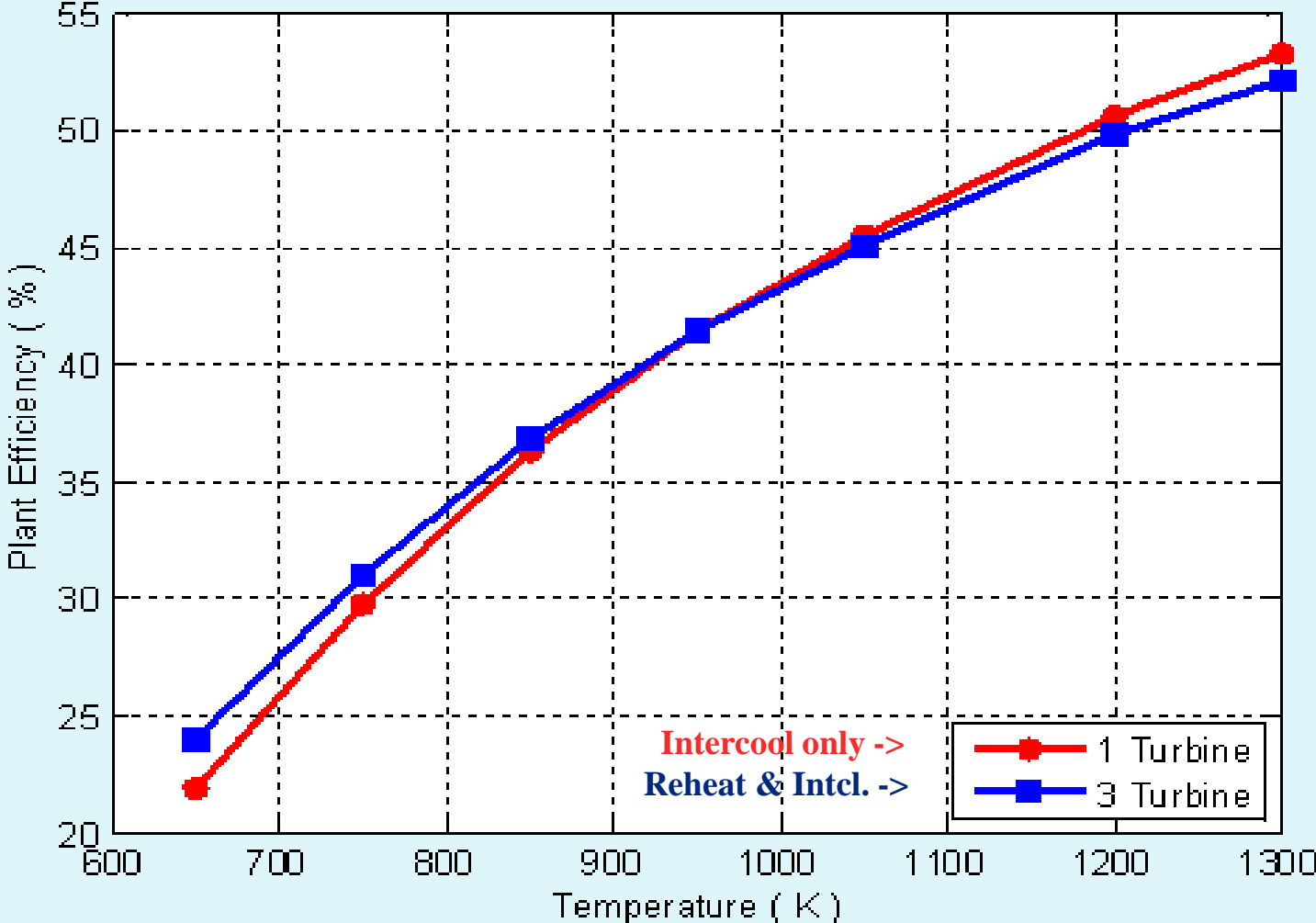
Temp Ratio = 4.0  
 Efficiency = 50.5 %  
 Turbine Power = 178.9 MW  
 Compressor Power = - 76.9 MW

# 100 MWe Power Plant – 2 Salt Configuration Thorium Molten Salt Reactor - Helium Brayton Cycle, 1200 K Turbine Inlet Temp

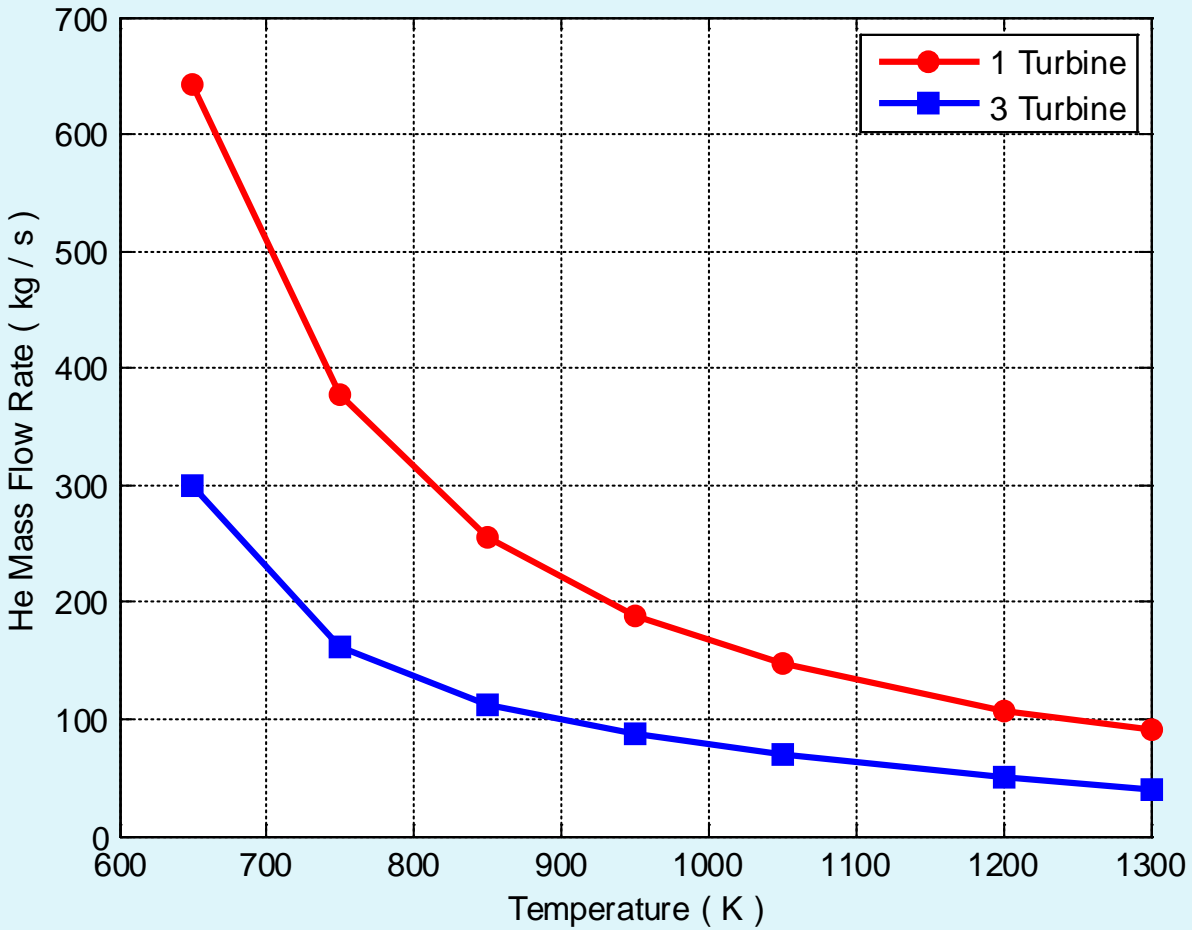
He ---  
 ThF4-UF4 ----  
 & BeF2 -----



# 100 MWe Power Plant Efficiency w. Intercool & Reheat Cycles ( 3 Inter-cooled Compressors in Series)



# 100 MWe Power Plant Flowrate w. Intercool & Reheat Cycles ( 3 Inter-cooled Compressors in Series )



# Submarine Based Power Plants

## Compact, Portable Thorium Reactors



Proposal to use US mothballed shipyards to produce hundreds of portable thorium nuclear gas turbine power plants



## Concluding Remarks

- Numerically confirmed that Nuclear Power Plants with CCGT Conversion Technology can achieve **> 50% Thermal Efficiency at TIT ~ 1200 K.**
- Above result obtained for both 'Intercool + Reheat' and 'Intercool Only' Cycle Configurations
- '**Intercool + Reheat**' Configurations have higher **Complexity** (number of ducts and heat exchangers) **but lower Working Fluid Mass Flow (He)** requirements thus reducing Ducting and Heat Exchanger Size
- **Liquid Fluoride Thorium Reactor Technology (LFTR)** can meet the goals of the **Gen IV Nuclear Energy Systems Initiative**—Energy Policy Act '005
  - Uses fertile Th232 breeding to fissile U233
  - **Can meet world energy demands for tens of millennia**
  - ~300 times Energy Density of current LWR Nuclear Power Plants with corresponding reduction in fission products. **Decay <300 yrs.**
  - Inherently Safe due to negative temp. coefficient of reactivity
  - Load Leveling operation to produce H2, & Sea Water Desalination
- Submarine based Power Plants located “off-shore” could serve as decentralized micro-grid elements, **hardened against cyber threats (EMP).**

# Backup Slides

# Typical Machine Sizes for 1000 MWe He Plant

- Single Turbo-Alt at 10 MP a and  $Pr=2$ ; (TIT=1200K; TR=4)
  - Mass Flowrate ~ 1420 kg/sec
  - Dia. = 6.5 m; L = ~20 m; Speed = 1800 rpm
  - Recuperator Volume ~ 360 m<sup>3</sup>
  - Thermal Eff. = 48%
- Three Reheat/Intercooled Turbo-Alt's
  - Mass Flowrate ~ 474 kg/sec
  - P=20 Mpa ( $Pr=2$ ); Dia = 1.9 m, L = 4.5m, Speed = 72000 rpm
  - P=10 Mpa ( $Pr=2$ ); Dia = 2.7 m, L = 6.3m, Speed = 5400 rpm
  - P= 5 Mpa ( $Pr=2$ ); Dia = 3.8 m, L = 8.5m, Speed = 3600 rpm
  - Recuperator Volume ~ 120 m<sup>3</sup>
  - Thermal Eff. = 51.5%



# Partial BRMAPS Code Output Results for 5 MWe Lunar Power Plant

```

BRAYTON CYCLE CALCULATIONS   - NON REGENERATED - 1500 K- POWER LEVEL =    5.00 MWE    TSINK-K =   190
TEMP RATIO      ETAB      ETAPC      ETAPT      ERG      GAMMA      LPC      ETM      EPSIL      TIT-K
      3.000      .990      .900      .900      .000      1.667      .980      .950      .900      1500
OPTIMUM PRESSURE RATIOS (MAX THERM EFF; MIN ARP,MASS ) =    4.550    3.200    3.400    TIC-K =   500
PR RATIO THERM EFF.  ARP(M2/KW) MSYS (MG) W(KM/S-MW)  TREJ-K  TREFF-K  TOC-K  TOT-K  ETAC  ETAT
4.5500    .2604    .4076  20.8974    .3895    862.88    630.73  1005.81  862.88  .867  .925
3.2000    .2382    .3554  19.6747    .3122    985.29    663.87  850.84  985.29  .875  .919
3.4000    .2446    .3566  19.6355    .3170    963.13    658.18  875.51  963.13  .874  .920
3.2000    .2382    .3554  19.6747    .3122    985.29    663.87  850.84  985.29  .875  .919
THEORETICAL OPTIMUM PRESSURE RATIO (PROPTIM) =    4.579
NUMBER OF ITERATIONS (ICTE,ICTA,ICTM) =    7    9    9
  
```

```

          TURBO - ALTERNATOR POWER DISTRIBUTION
GAS MOLECULAR WEIGHT (KG/MOL) =    4.000
MASS FLOWRATE (KG/SEC) =    6.178
COMPR. PRESSURE RATIO =    3.200
NO. INTC. COMP. STAGES =    1
NO. REHEAT TURB. STAGES =    1
          COMPR. BLEED PCT. =    .500
          COMPRESSOR POWER (MW) =   -10.921
          TURBINE POWER (MW) =    16.190
          POWER LOSSES (MW) =    - .269
          POWER SUM BALANCE (MW) =    5.000
(= GENERATOR TERMINAL
ELECTRICAL POWER-(MW))
  
```