

# Characterizing Damage Accumulation in CMCs

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Bhavana Swaminathan<sup>1</sup>

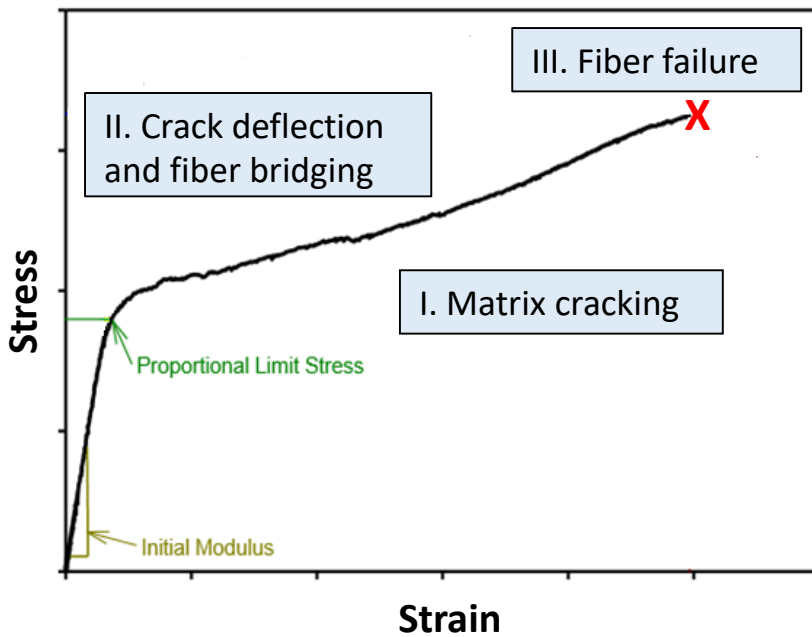
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3. NASA Glenn Research Center (GRC)
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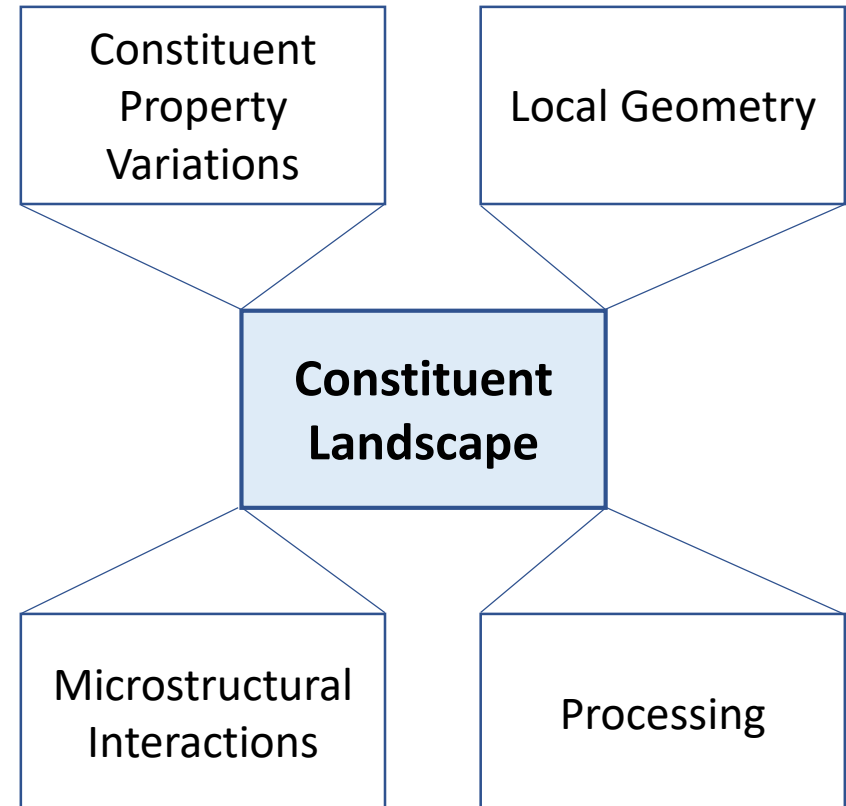


# CMC Damage Behavior is Influenced by Constituent Landscape Effects

**Hypothesis:** Locally averaged structural features are significant drivers in the manner in which damage accumulates, moves, and interacts

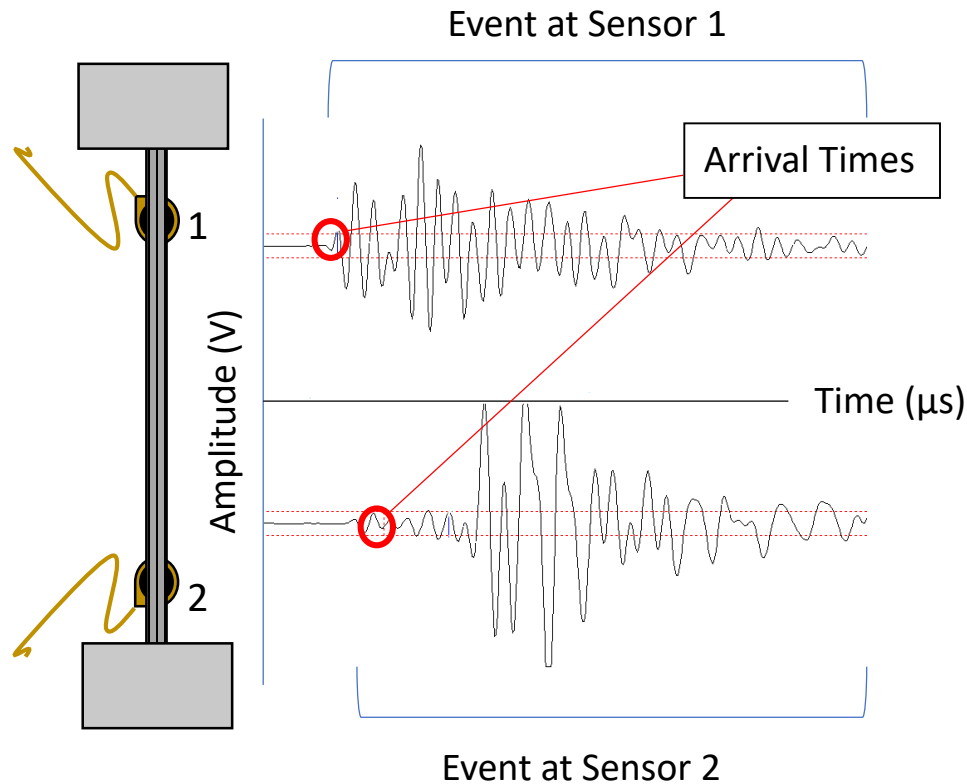


Corman and Luthra, Handbook of Ceramic Composites (2005).



# Acoustic Emission (AE) Enables Real-Time Damage Assessment

AE can be used to locate event sources and characterize damage in the bulk



$$Location = \frac{x}{2} \cdot \left[ \frac{\Delta t}{\Delta t_x} \right]$$

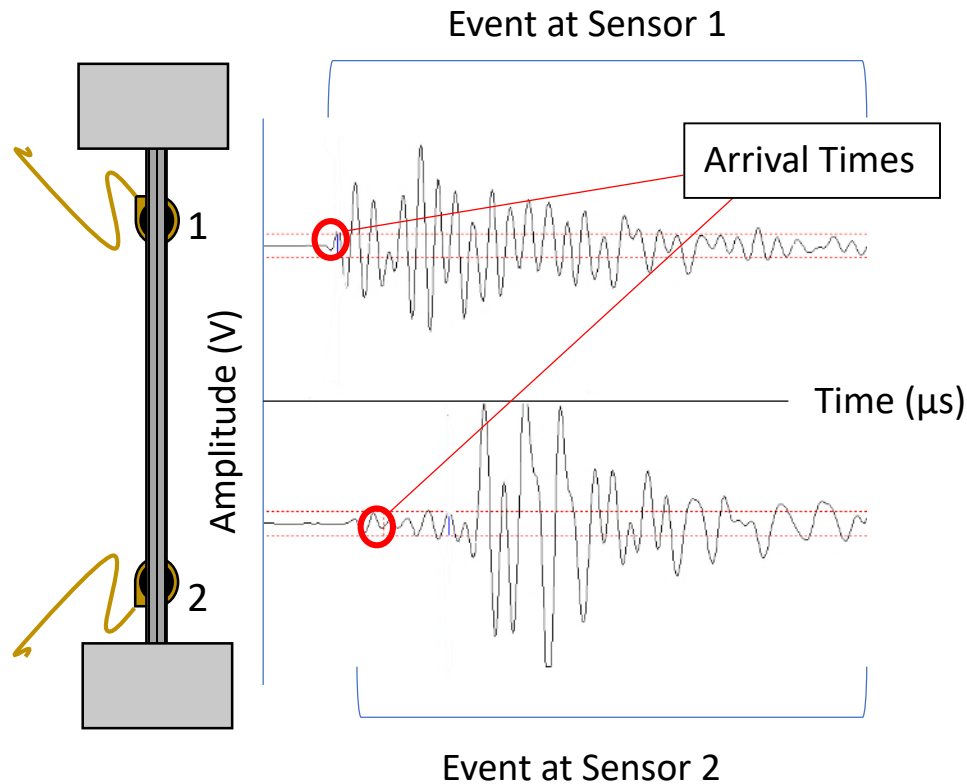
$x$  = sensor separation

$\Delta t$  = arrival time difference

$\Delta t_x$  = difference in arrival times  
across AE window

# Acoustic Emission (AE) Enables Real-Time Damage Assessment

AE can be used to locate event sources and characterize damage in the bulk

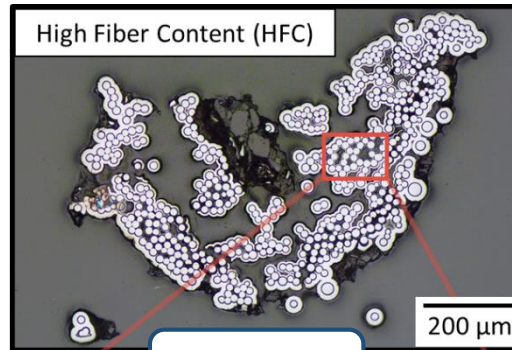
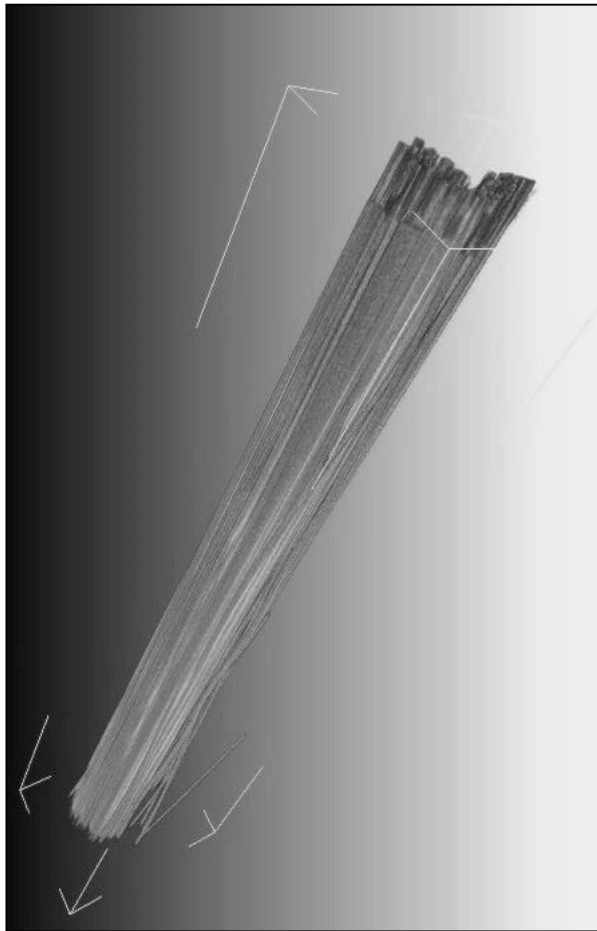


## Advantages

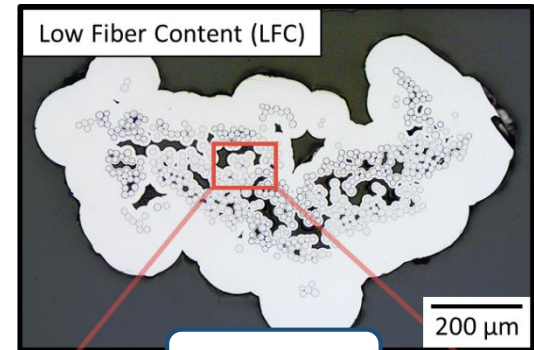
- In-operation monitoring
- Damage localization
- Volumetric inspection
- Early detection of damage

# Minicomposites Provide Simplified Architecture Suited to Study Microscale Damage

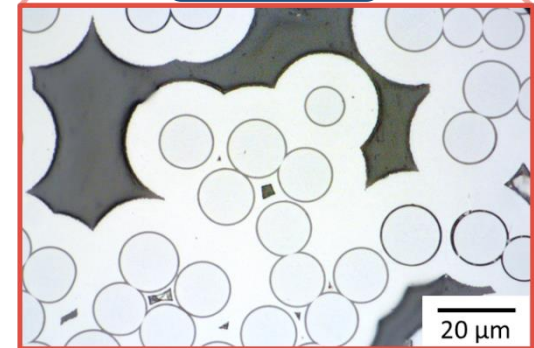
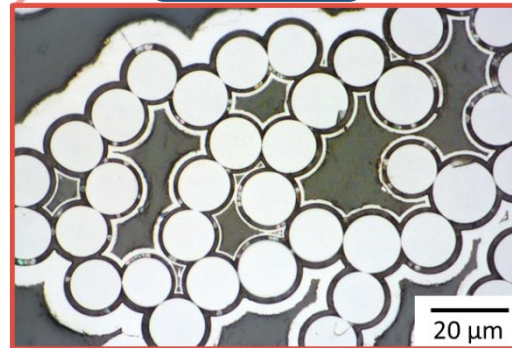
Specimen ID	BN Thickness ( $\mu\text{m}$ )	$V_f$ (%)	$V_{BN}$ (%)	$V_m$ (%)	Area ( $\text{mm}^2$ )
HFC	$1.9 \pm 0.6$	33.5	24.6	41.8	0.167
LFC	$0.3 \pm 0.08$	21.3	2.2	76.5	0.266



High  $\tau$ ,  $\Gamma$



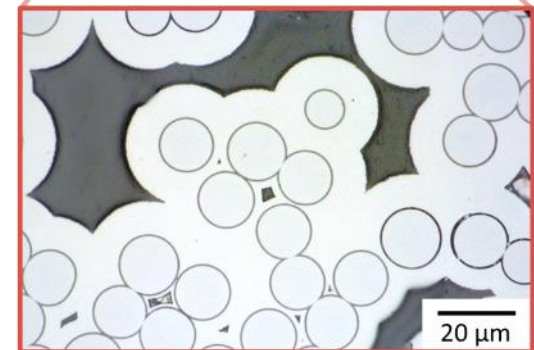
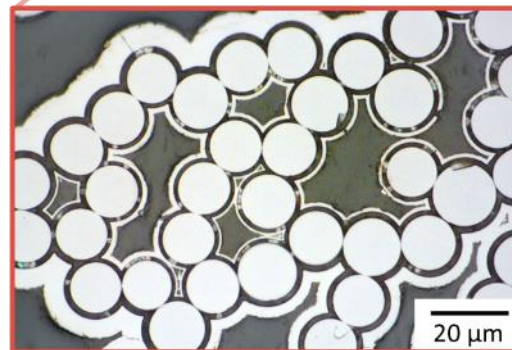
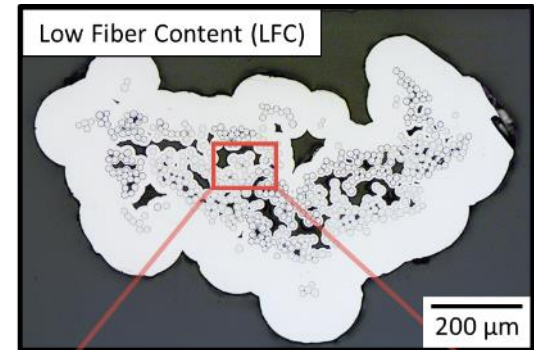
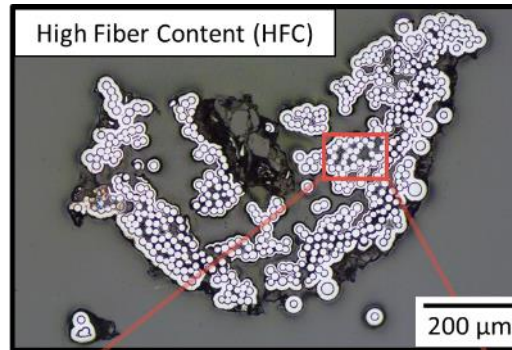
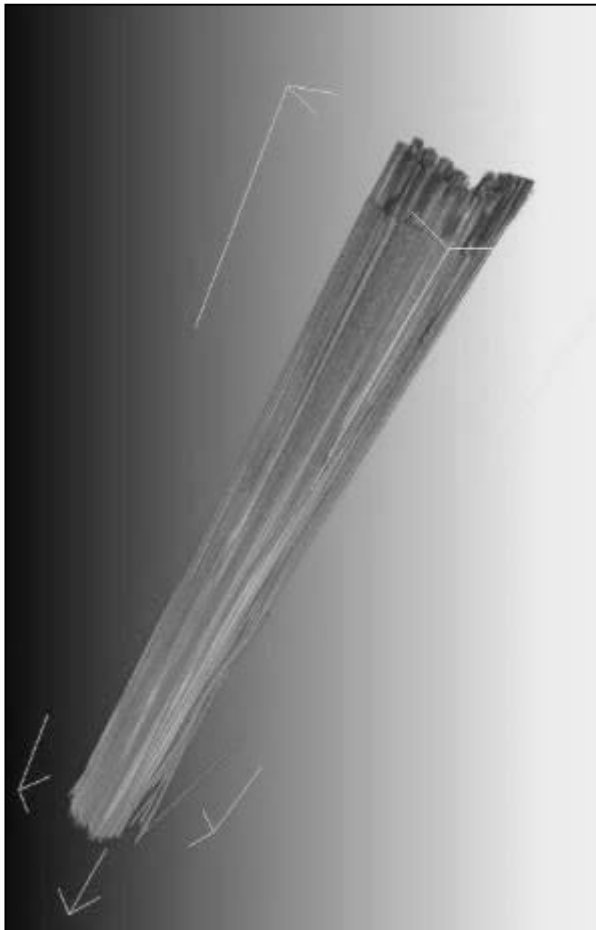
Low  $\tau$ ,  $\Gamma$



Courtesy of Richard Rauser (GRC)

# Minicomposites Provide Simplified Architecture Suited to Study Microscale Damage

Specimen ID	Sliding Stress ( $\tau$ )	Debond Toughness ( $\tau$ )
HFC	$34.5 \pm 13.0$ MPa	$5.5 \pm 3.9$ J/m <sup>2</sup>
LFC	$18.1 \pm 4.8$ MPa	$1.2 \pm 0.5$ J/m <sup>2</sup>

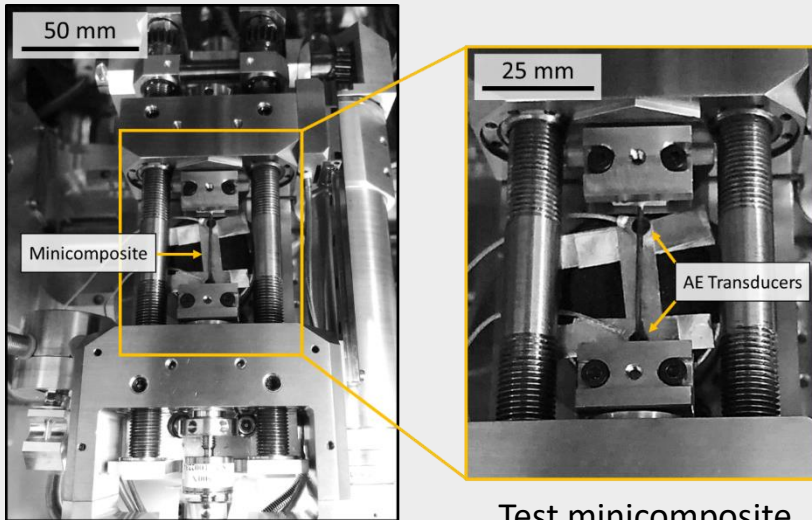


Courtesy of Richard Rauser (GRC)



# A Multi-Modal Approach for Damage Characterization

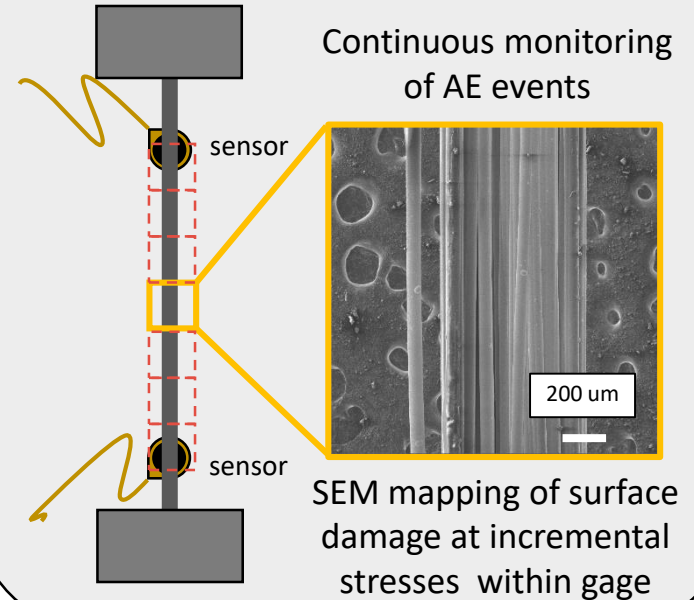
## In-SEM Mechanical Testing



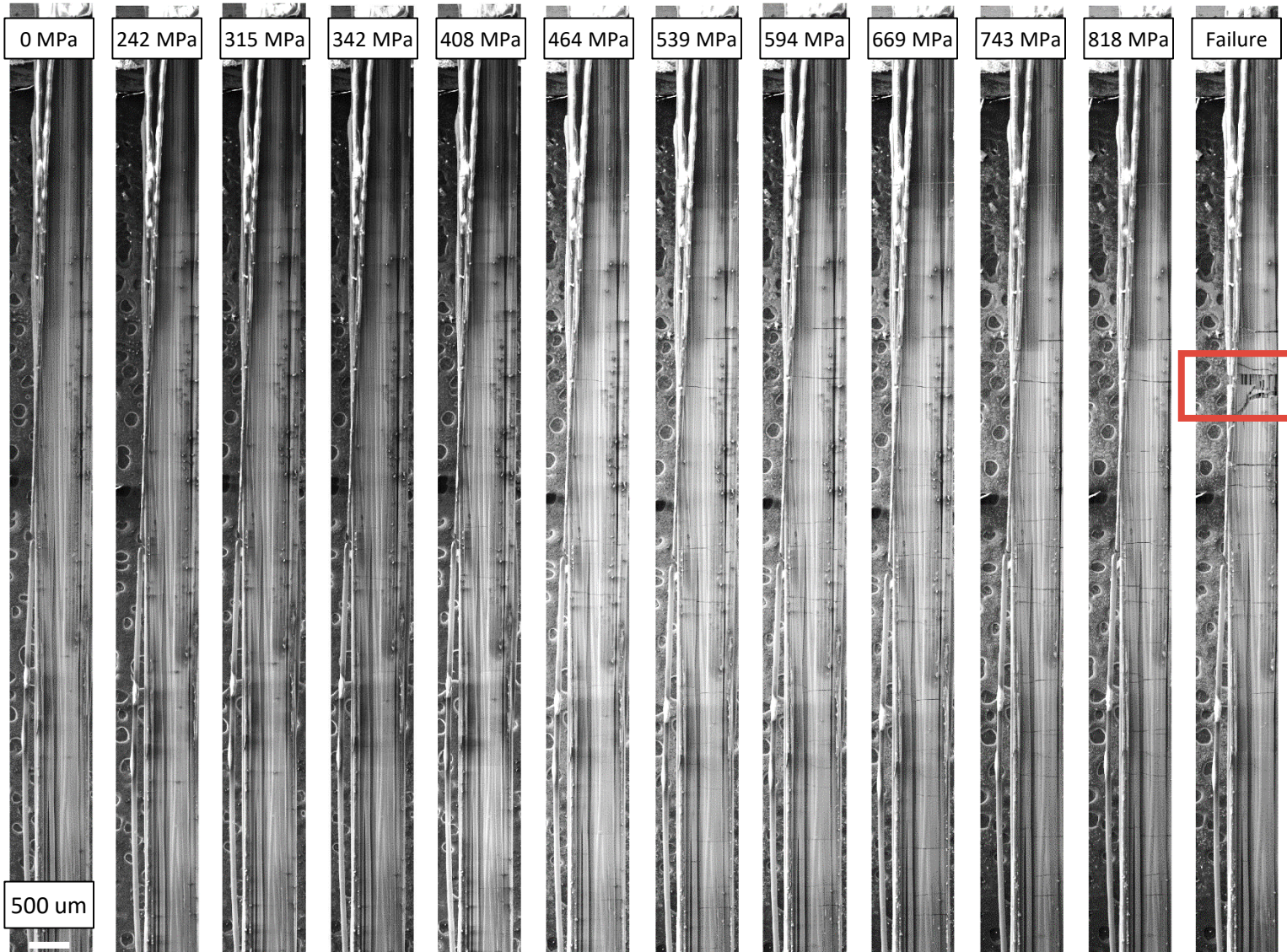
Test minicomposite to failure



## Acoustic Emission (AE)

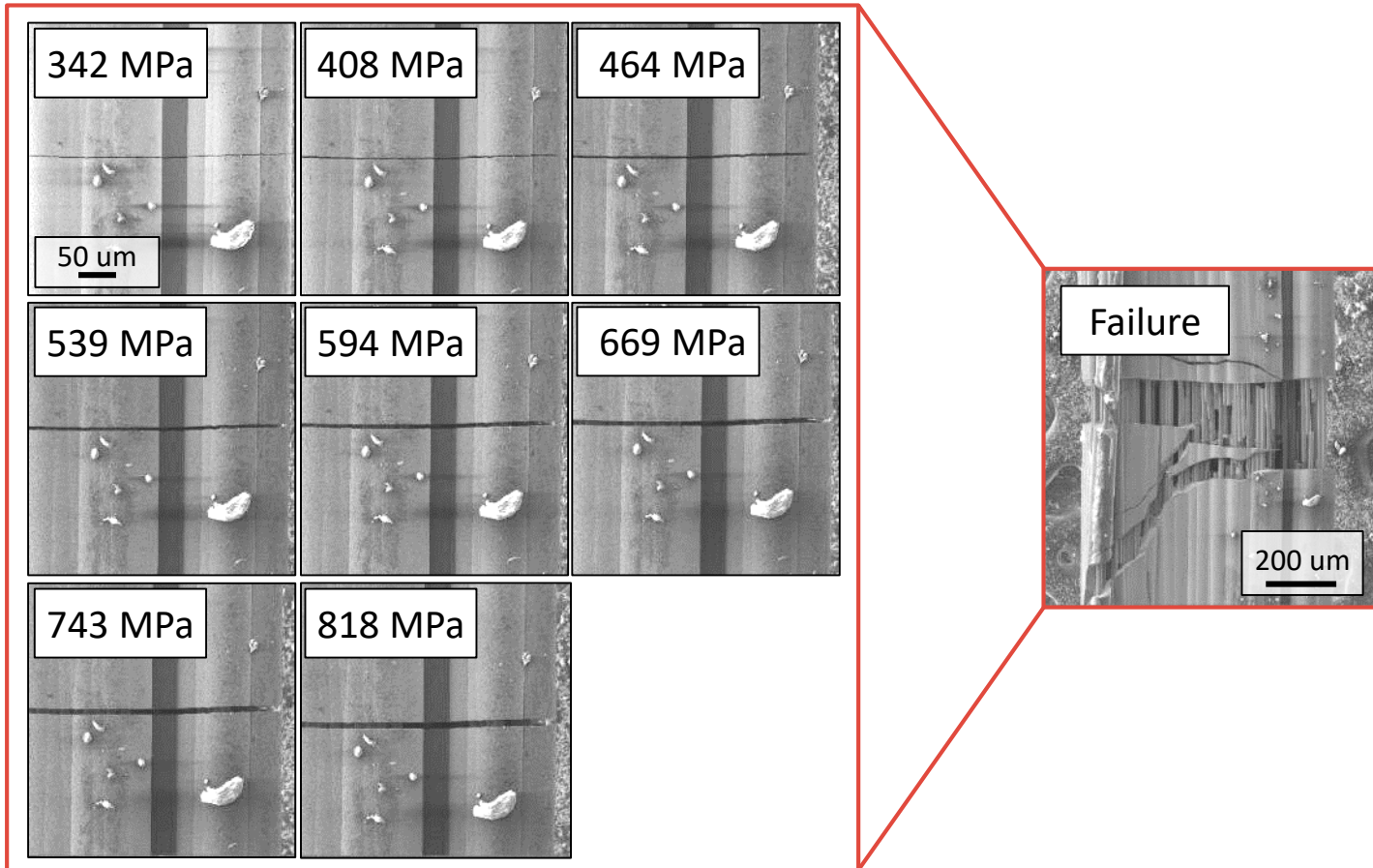


# Incremental Loading to Capture Damage Progression





# Incremental Loading to Capture Damage Progression



## Two Domains: Matrix-Dominated and Fiber-Dominated Damage

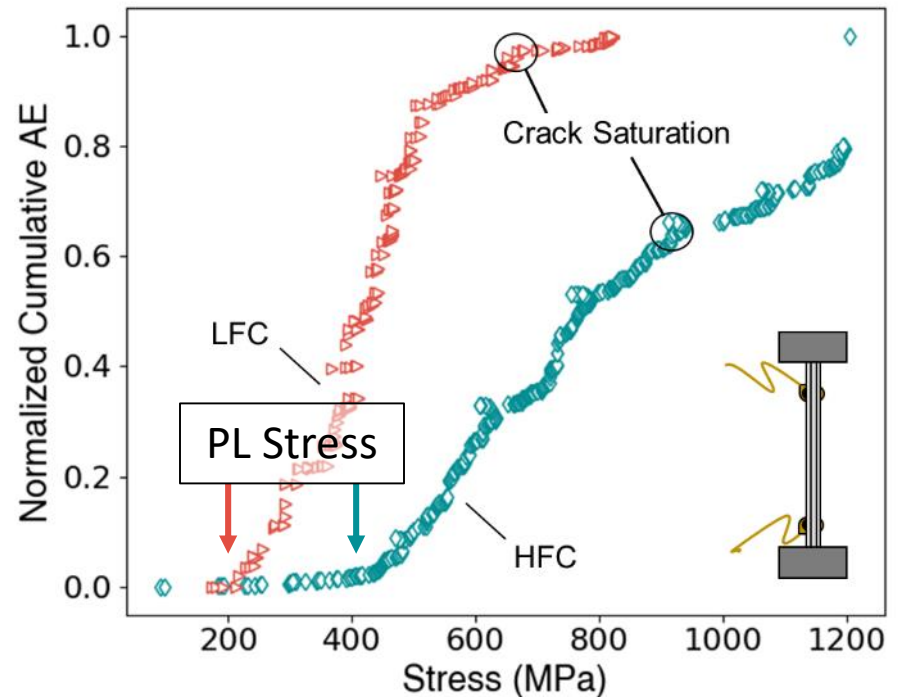
### Matrix-Dominated Damage

- Matrix crack initiation
- Matrix crack propagation
- Interfacial debonding
- Fiber sliding
- Early fiber failures

Crack Saturation

### Fiber-Dominated Damage

- Fiber loading
- Fiber sliding
- Fiber breakage
- Fiber pullout



## Two Domains: Matrix-Dominated and Fiber-Dominated Damage

### Dominant Mechanisms

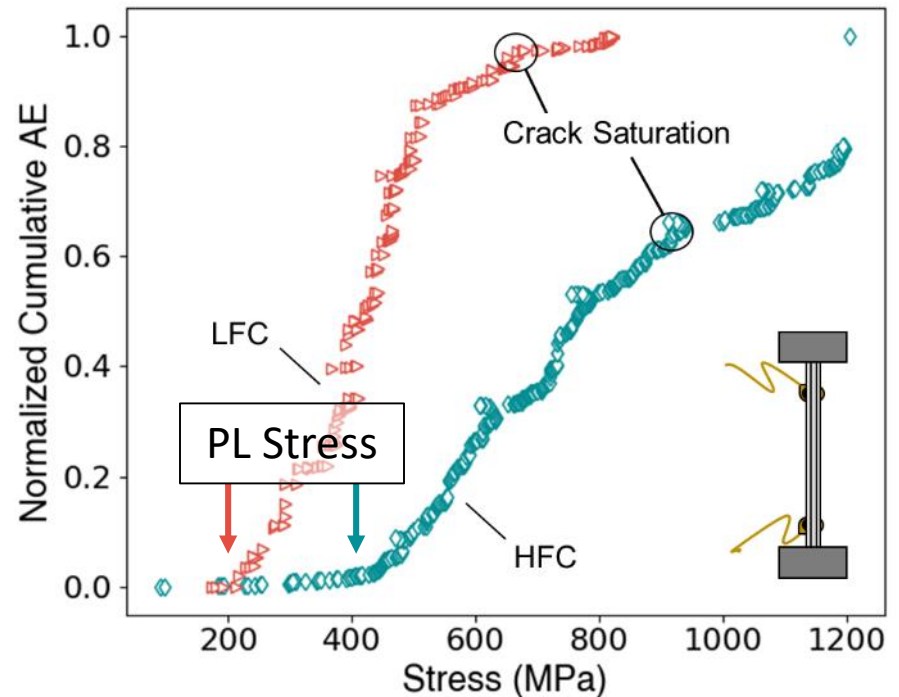
- Matrix crack initiation
- Fiber breakage

High Energy AE

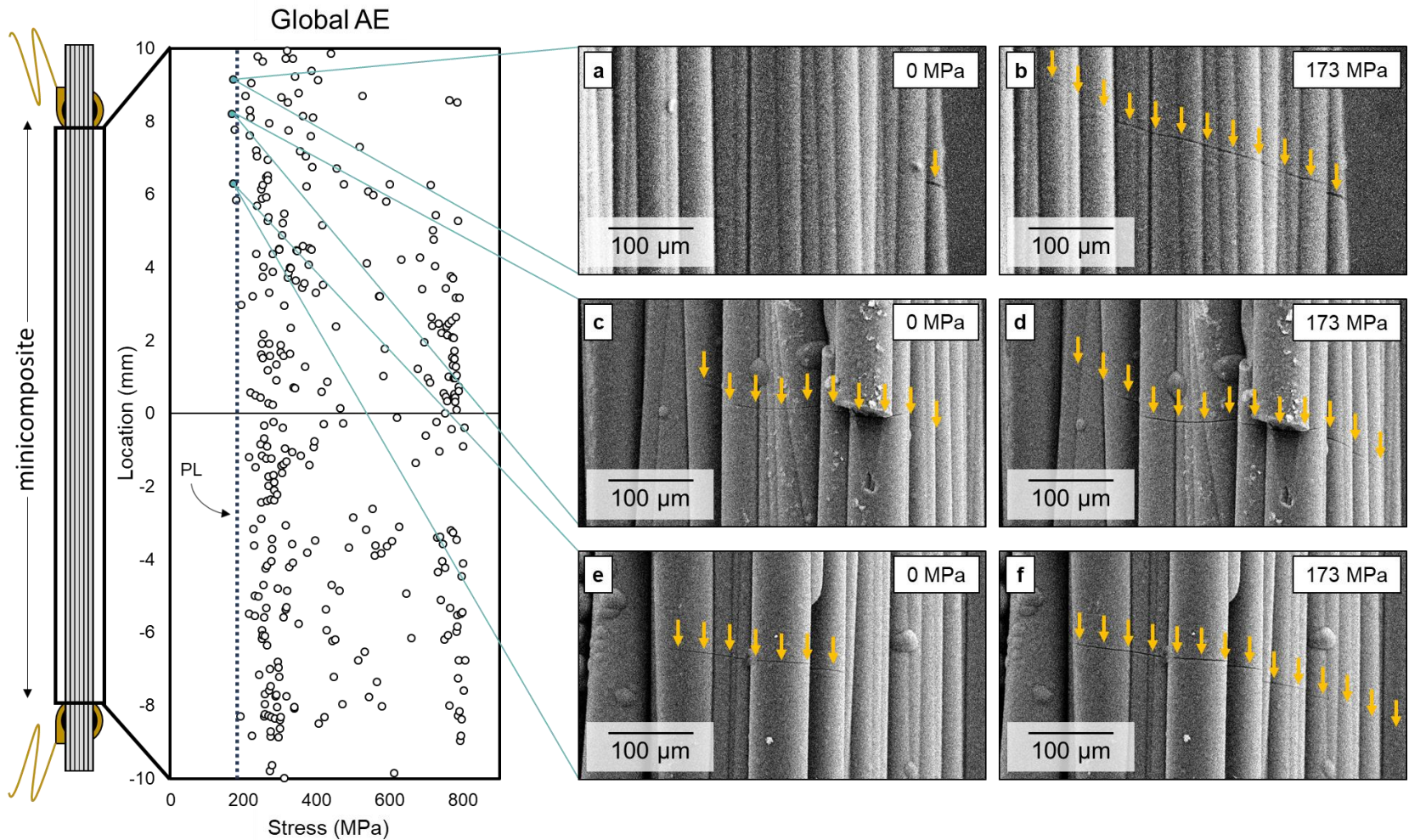
Low Energy AE

### Nondominant Mechanisms

- Matrix crack propagation
- Interfacial debonding
- Fiber sliding
- Fiber pullout

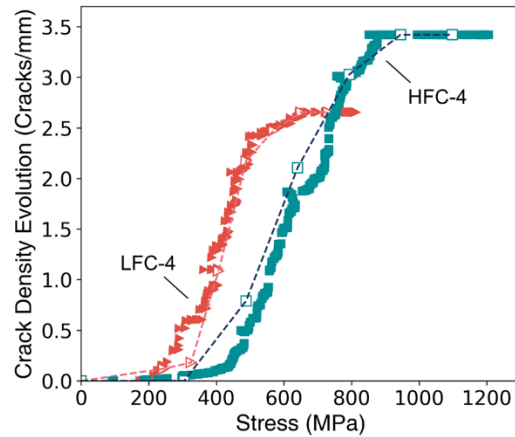
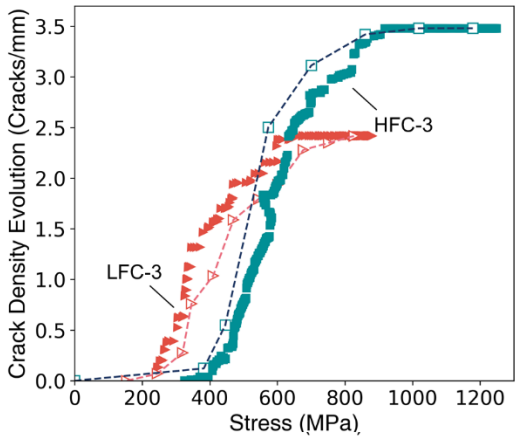
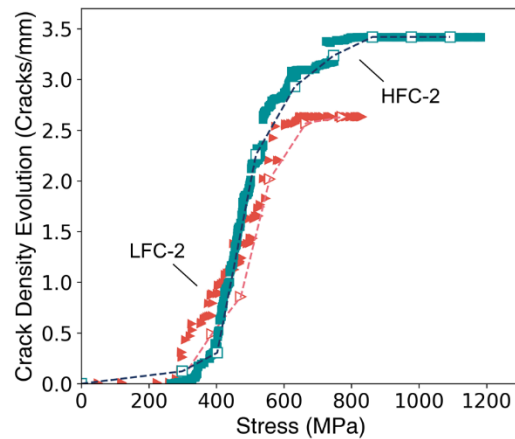
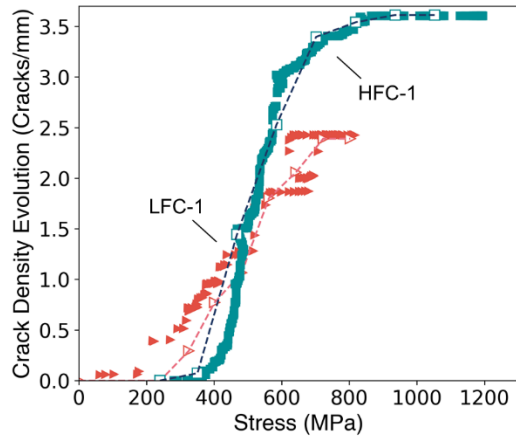


# Damage Accumulation below PL due to Propagation of Flaws



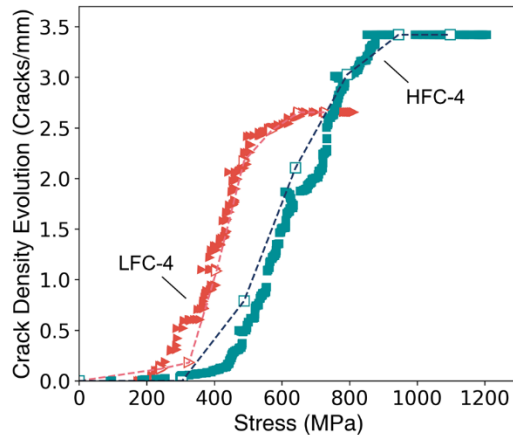
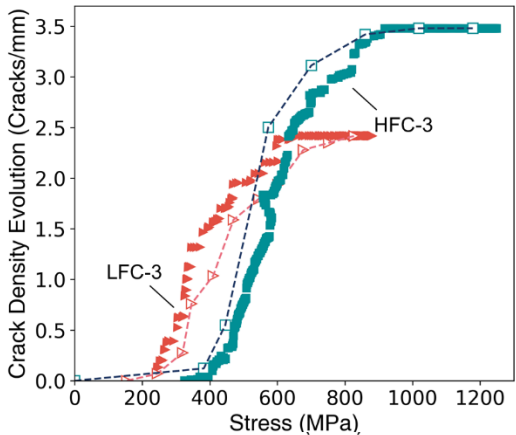
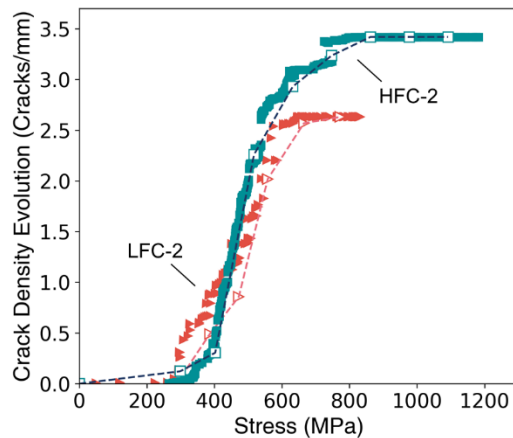
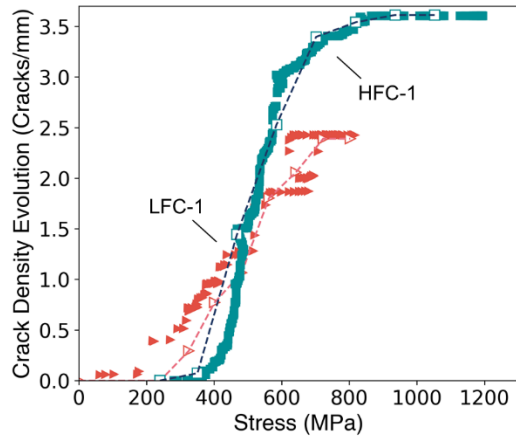


# Agreement between In-SEM Measured and AE-Predicted Crack Density Evolution



$$CDE(N) = CD_{rup.} \cdot \frac{\text{Cumulative AE (N)}}{\text{Cumulative AE at saturation}}$$

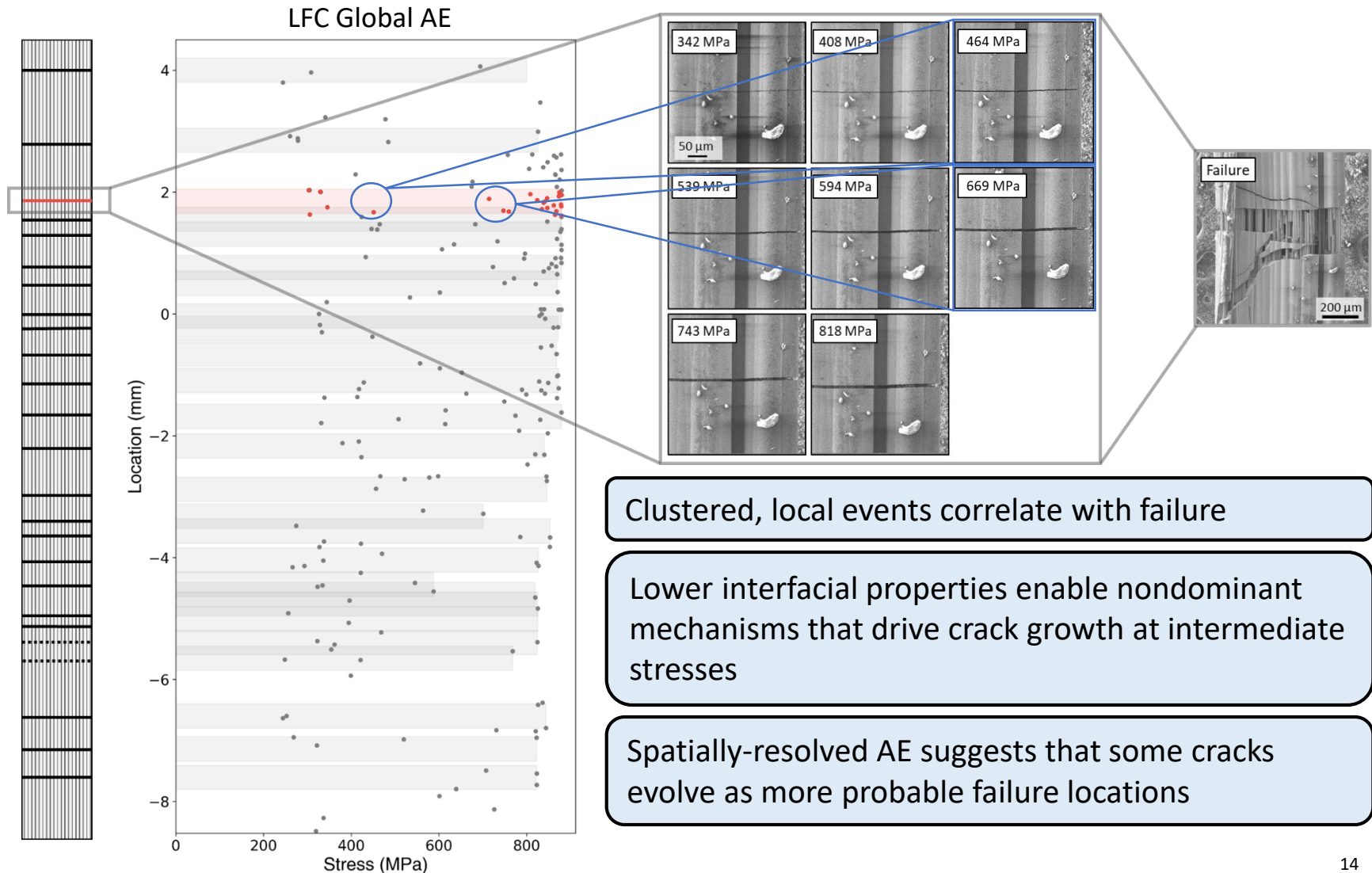
# Agreement between In-SEM Measured and AE-Predicted Crack Density Evolution



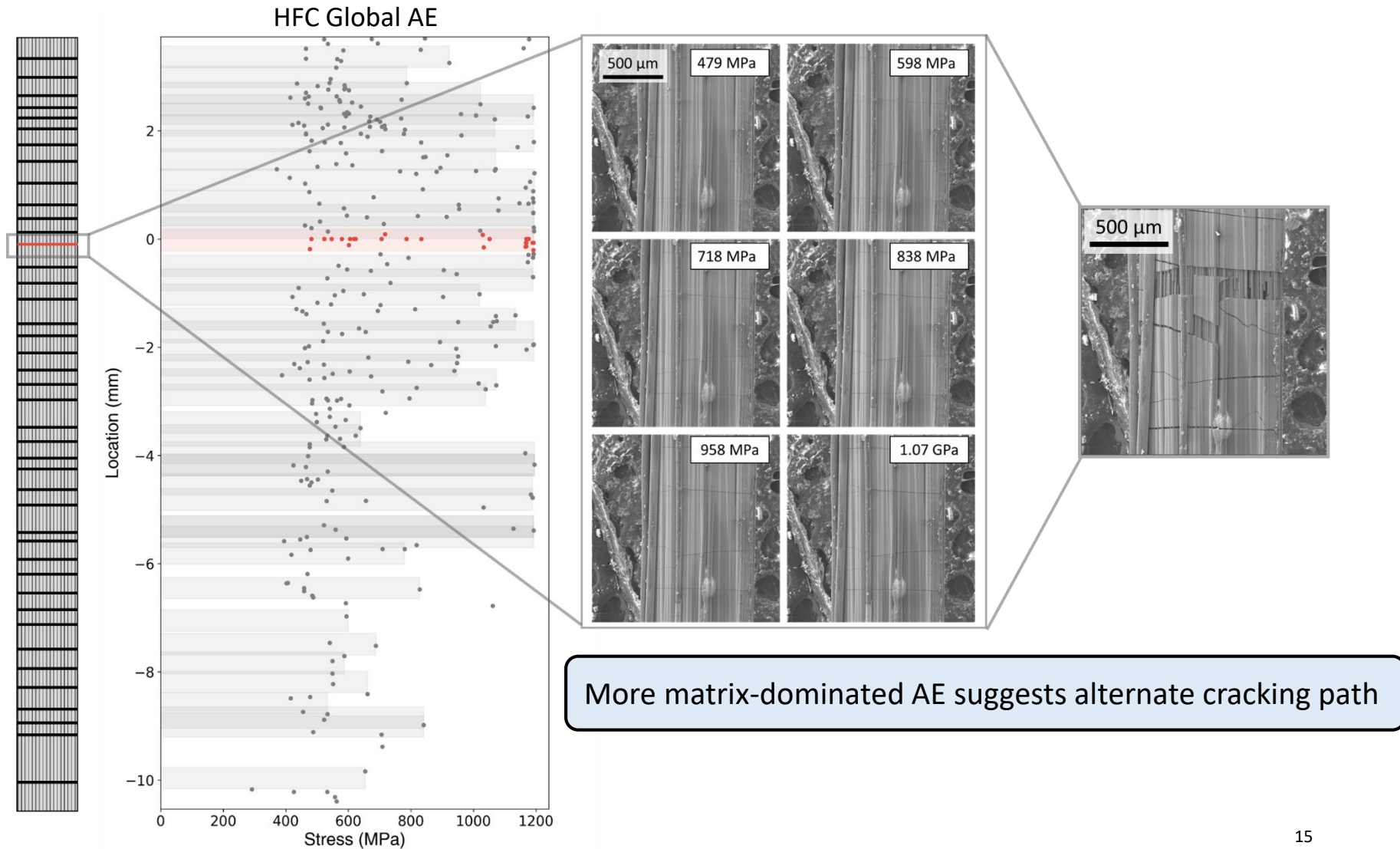
- Agreement at early and intermediate stresses shows transverse matrix cracking dominance

- Agreement at early stresses shows transverse cracking
- Variations in real and predicted response[s] due to other phenomena detected by AE

# Correlation of Spatially-Resolved AE and In-SEM Observed Damage

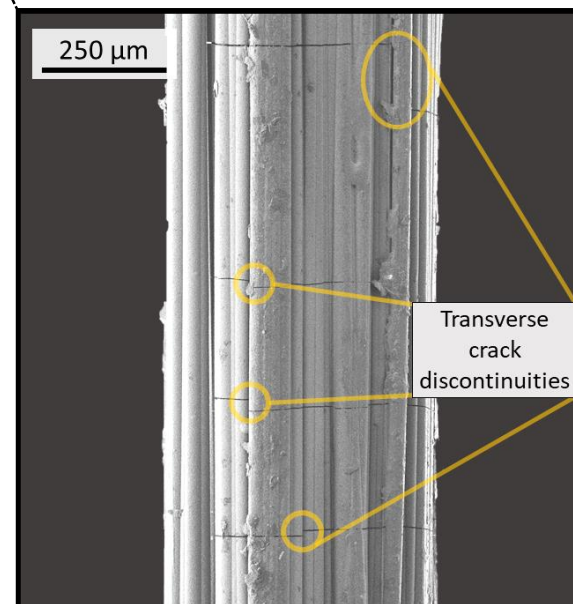
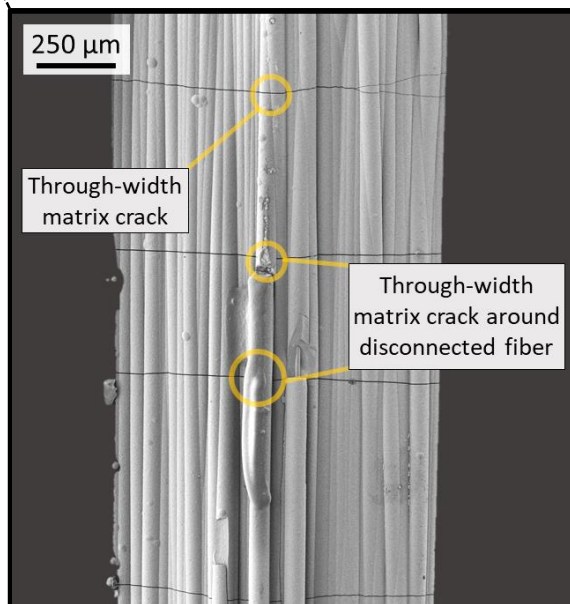
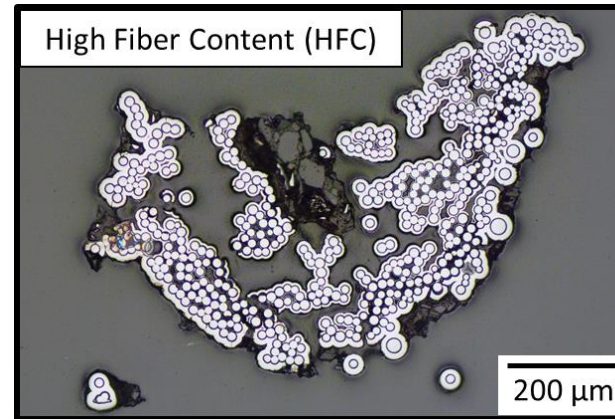
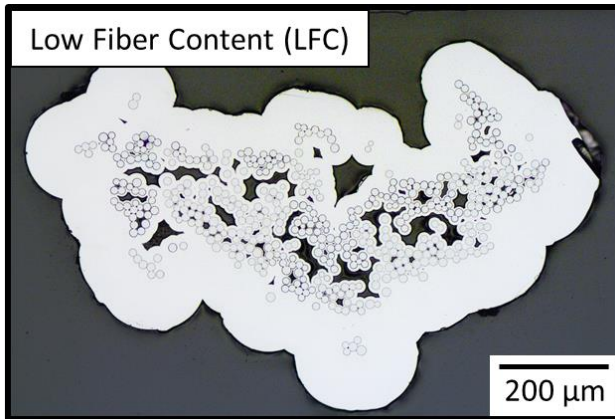


# Correlation of Spatially-Resolved AE and In-SEM Observed Damage



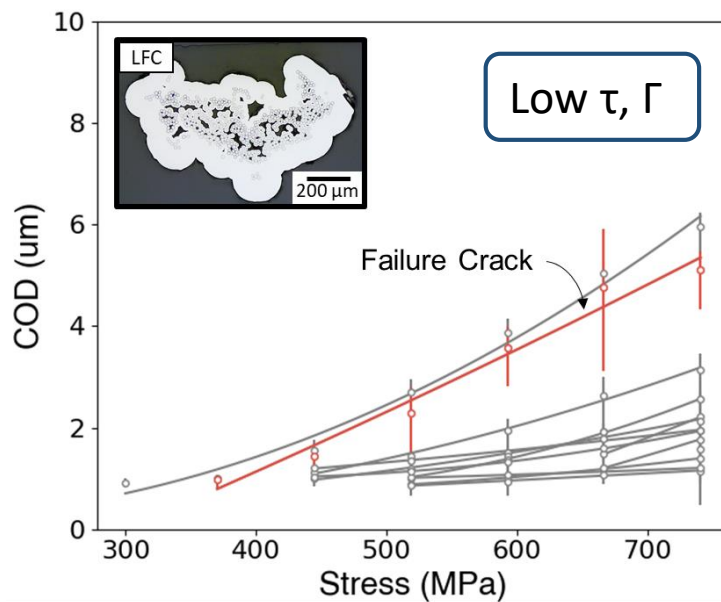


# Incremental vs. Through-Thickness Crack Propagation

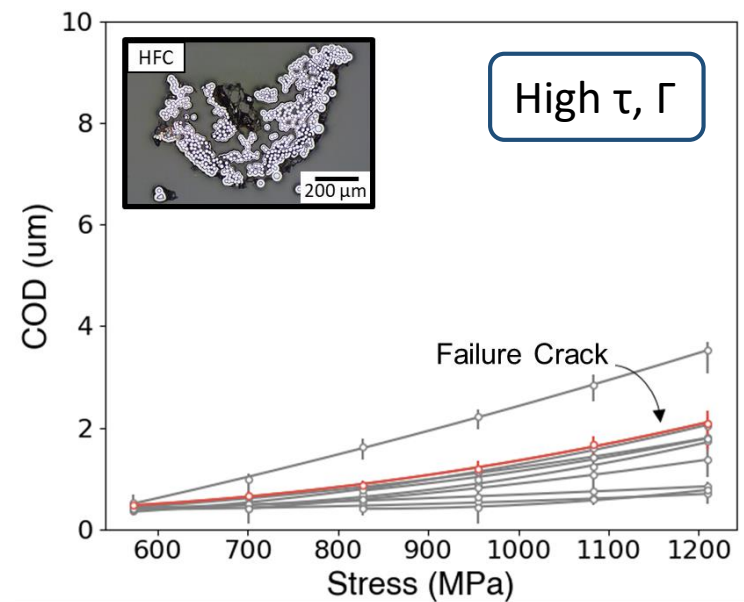


## Local Microstructural Effects Influence Crack Opening Displacement (COD)

- Earliest crack openings: 600 nm
- Fiber sliding w/ larger debond lengths to accommodate opening
- Saturation of non-dominant phenomena prior to fiber breaks
- Fiber breaks redistribute load to tow

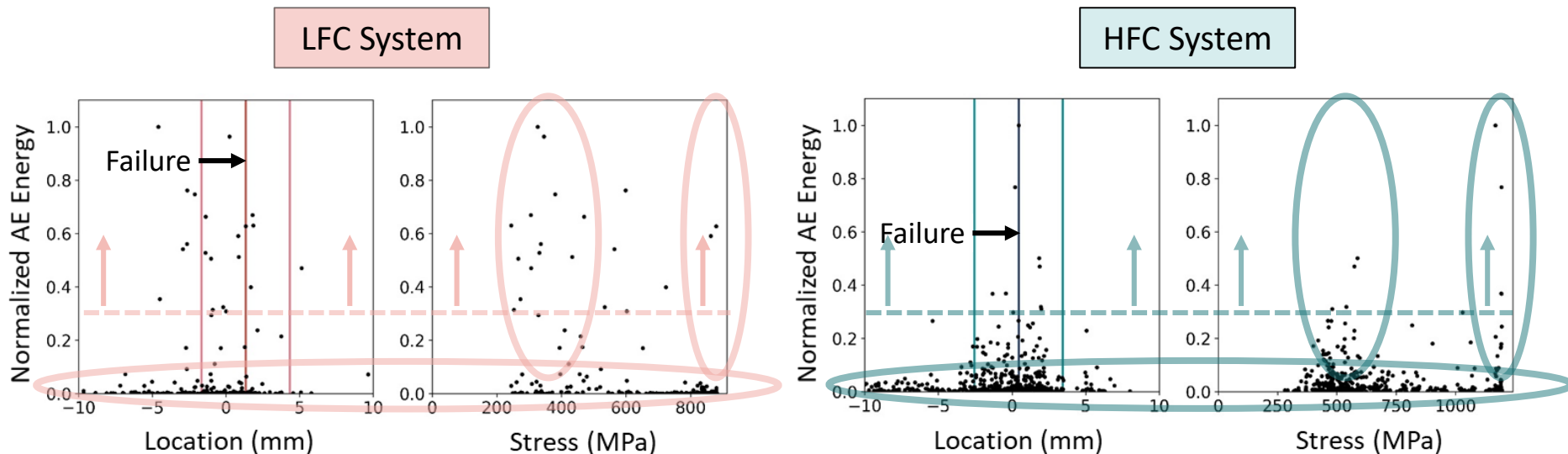


- Earliest crack openings: 300 nm
- High fiber content for initial crack bridging
- Fiber breaks redistribute load to pillar
- Fiber reloading through sliding



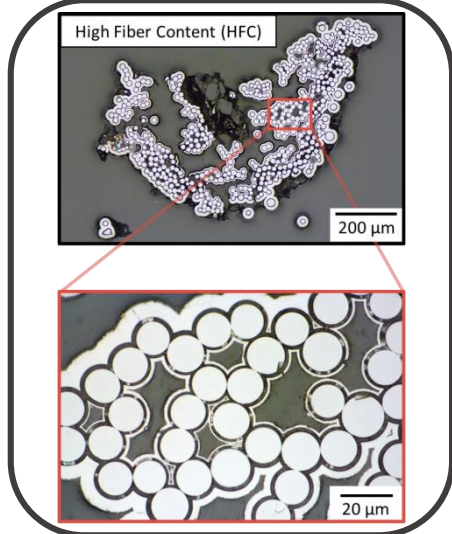
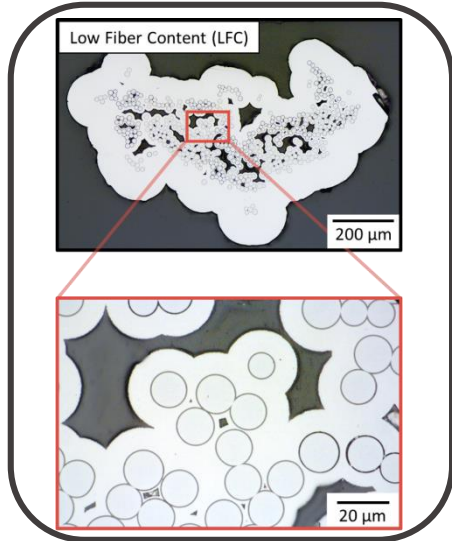
## AE Event Energy May Correlate with Source Damage Mechanism

Correlation between high-energy AE and proximity to failure location

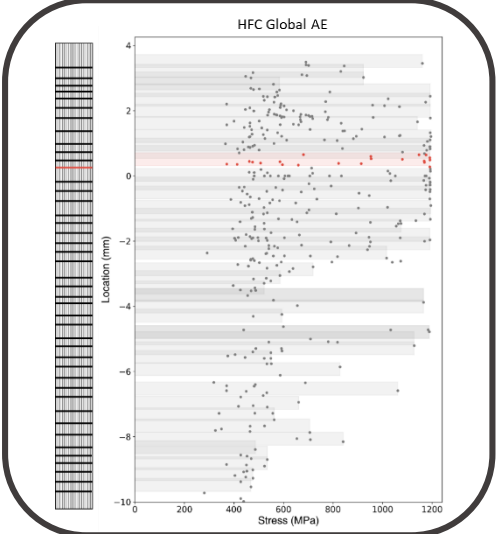
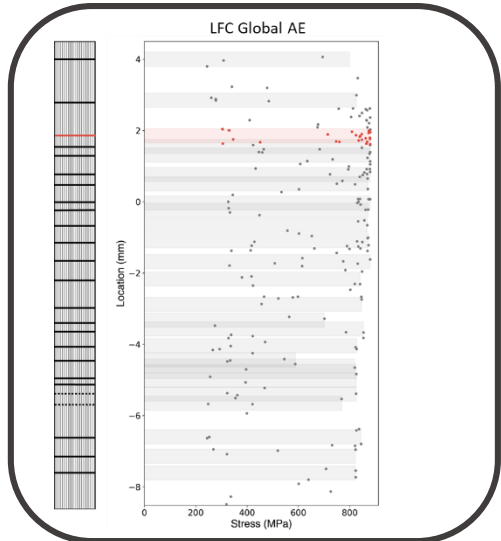


- High-energy AE activity from dominant damage mechanisms (matrix cracking / fiber breakage)
- Ultra-high-energy AE activity from large crack formation
- Low-energy AE activity from nondominant damage mechanisms

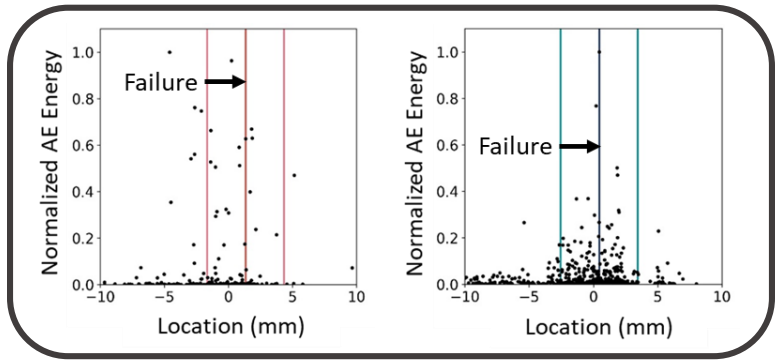
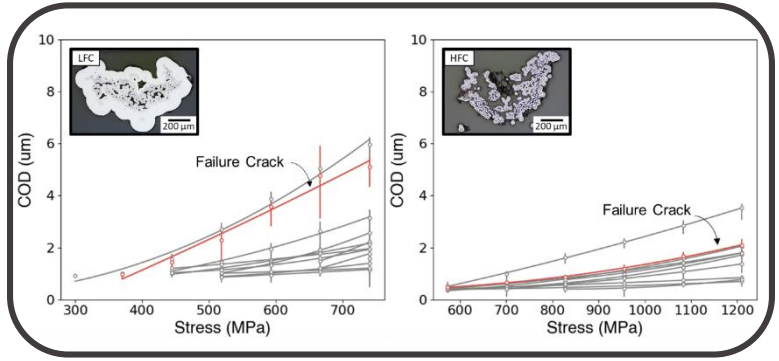
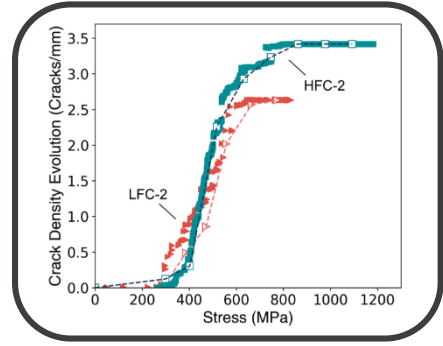
Damage characterization in two minicomposite systems



Correlation of spatially-resolved AE to microscale surface damage



Structural features and local phenomena drive damage response





# Acknowledgements

## Collaborators at UCSB

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