

# Comparison of Risk from Orbital Debris and Meteoroid Environment Models on the Extravehicular Mobility Unit (EMU)

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## Overview



- A well-known hazard associated with exposure to the space environment is the risk of failure from an impact from a meteoroid and orbital debris (MMOD) particle
- An extravehicular mobility unit (EMU) "spacesuit" impact during a US extravehicular activity (EVA) is of great concern as a large leak could prevent an astronaut from safely reaching the airlock in time resulting in a loss of life
- A risk assessment is provided to the EVA office at the Johnson Space Center (JSC) by the Hypervelocity Impact Technology (HVIT) group prior to certification of readiness for each US EVA
- Need to understand the effect of updated meteoroid and orbital debris environment models to EMU risk

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**EVA Risk Assessment Methodology** 





**EVA Timeline Risk Assessment Flowchart** 



## **EMU Finite Element Model (FEM)**



- A detailed finite element model (FEM) of the EMU was created with regions for the various shielding configurations
- 42 different surface property ID (PID) types representing the different shielding configurations
- Two main groups of shielding configurations:
  - 1. Soft goods:
    - TMG over a pressure garment
    - maintains the acceptable atmospheric environment for the astronaut
  - 2. Hard goods:
    - TMG (except for helmet) over metallic, composite and/or plastic components.



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#### **EMU FEM Continued**





\* Color corresponds to the regional risk breakdown bar charts



## **Ballistic Limit Equations**



- EMU thermal meteoroid garment (TMG) performs as a mini bumper shield
  - Ortho-fabric layers induce a shock pulse that breaks up the projectile and creates an expanding debris cloud
  - Inner layers (MLI and ripstop) and the pressure garment restraint layer help with further particle breakup and create spacing for the debris cloud to expand before reaching the bladder or underlying critical component







## **EMU Failure Modes**



- The risk of a meteoroid or orbital debris particle penetrating the thermal meteoroid garment (TMG) and the pressure garment is determined for two failure modes:
  - 1. Perforation threshold (EVA abort): any size leak risk of the bladder layer of the pressure garment
  - 2. Critical hole size threshold (catastrophic leak): uncontrolled leak risk caused by a >4mm hole in the bladder



## **EVA Timeline Analysis**



- EVA office provides a summary of EMU positions (including body orientation) for the specific EVA worksite locations on the International Space Station (ISS), and the duration at each location
  - using detailed EVA summaries/presentations and/or EVA training run videos from the Neutral Buoyancy Lab (NBL)
- Timeline analysis FEMs are built by orienting an EMU FEM at each worksite location on a simplified ISS FEM. When one or more worksites require multiple body orientations, additional analysis FEMs are built.









- Meteoroid models (provided by MSFC Meteoroid Environment Office):
  - MEM R2 previous meteoroid environment model used from 2014 through August 2019
    - All meteoroids assumed to be 1.0 g/cm<sup>3</sup>
  - MEM 3 latest released meteoroid environment model and used in EVA risk assessments starting in August 2019
    - Meteoroid density in two distributions (low-density and high-density) with density varying from 0.125 g/cm<sup>3</sup> up to 7.975 g/cm<sup>3</sup>
- Orbital debris models (provided by JSC Orbital Debris Program Office):
  - ORDEM 3.0 current approved debris model
  - ORDEM 3.1 new debris model not yet released
    - Risk assessments provided here are preliminary, for indication only (may be changed after ORDEM 3.1 is finalized and released)

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#### Meteoroid Model Comparison MEMR2 to MEM-3





EMU MMOD Study - Risk by Component				
Region	MEM-3 to MEMR2 Factor	% of Total Difference		
Lower Torso Assembly	1.9	46%		
Gloves	1.8	27%		
Arms	1.9	18%		
Primary Life Support System	1.3	7%		
SAFER Assembly	1.3	1%		
Helmet	1.3	0.4%		
Display and Control module	1.2	0.3%		
Hard Upper Torso	1.3	0.2%		
Softgoods	1.9	89%		
Hardgoods	1.3	11%		
Total	1.7	100%		

- MEM-3 meteoroid environment model contributes <u>71% more risk</u> (any size leak penetration) than MEM-R2
- Risk difference is attributable to the addition of high density populations to the MEM-3 meteoroid environment
- Softgoods shielding configuration accounts for 89% of the risk difference

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![](_page_10_Picture_2.jpeg)

#### Orbital Debris Model Comparison ORDEM 3.0 vs. ORDEM 3.1

![](_page_10_Picture_4.jpeg)

![](_page_10_Figure_5.jpeg)

EMU MMOD Study - Risk by Component				
	OD3.0 to			
	OD3.1	% of Total		
Region	Factor	Difference		
Lower Torso Assembly	0.9	52%		
Arms	0.9	22%		
Primary Life Support System	0.8	17%		
SAFER Assembly	0.8	4%		
Gloves	1.0	3%		
Helmet	0.9	1%		
Hard Upper Torso	0.9	1%		
Display and Control module	0.9	1%		
Softgoods	0.9	72%		
Hardgoods	0.8	28%		
Total	0.9	100%		

- ORDEM 3.1 environment model contributes <u>13% less risk</u> of (any size leak penetration) than ORDEM 3.0
- 99.7% of this risk difference comes from the high and medium density populations
- Softgoods shielding configuration accounts for 72% of the risk difference

![](_page_11_Picture_1.jpeg)

## **Combined MMOD Comparison**

![](_page_11_Picture_3.jpeg)

- <u>11% more</u> cumulative MMOD penetration risk for MEM-3 and ORDEM 3.1 versus MEMR2 and ORDEM 3.0
  - Risk decrease from ORDEM 3.1 orbital debris environment offset by risk increase from MEM-3 meteoroid environment
- 78% of the OD3.1/MEM-3 risk is in the softgoods shielding configuration regions

EMU Risk Assessment	Number of Failures (400km, 100 hour exposure, 24 attitude/2018-2024 average)							
	Orbital Debris				Meteoroid	MMOD		
Description	NaK	LD	MD	HD	Intacts	Total	Total	Total
ORDEM 3.1 and MEM-3	0.00E+00	4.22E-07	3.43E-04	1.15E-03	2.18E-09	1.50E-03	1.13E-03	2.63E-03
ORDEM 3.0 and MEMR2	2.71E-09	1.01E-06	4.54E-04	1.26E-03	1.18E-09	1.71E-03	6.63E-04	2.38E-03
Factor	0.0	0.4	0.8	0.9	1.8	0.9	1.7	1.1

![](_page_12_Picture_1.jpeg)

#### MMOD Risk odds for typical EVA (6.5 hour duration, 2 crew)

![](_page_12_Picture_3.jpeg)

- MMOD risks are relatively small for typical ISS EVAs, no matter what environment models are used
  - Example only, MMOD risks vary by location, duration, year of EVA, and other factors

	MMOD Risk Odds		
Failure Mode	MEM R2 and ORDEM 3.0	MEM 3 and ORDEM 3.1	
Any size leak	1 in 5,000	1 in 4,500	
Critical leak	1 in 28,000	1 in 26,000	

![](_page_13_Picture_1.jpeg)

## Conclusions

![](_page_13_Picture_3.jpeg)

- Assessed the change in risk using updated MMOD environment models (MEM-3 and ORDEM 3.1)
  - 11% more cumulative MMOD penetration risk due to risk increase from high-density component of MEM-3
- Soft goods regions of the EMU continue to drive risk, contribute 78% of cumulative MMOD penetration risk
- MMOD risks remain small for typical 6.5 hour EVA, no matter what MMOD environments are used