

Development of an Electrochemical Oxygen Compressor and Generator for Spacesuit Oxygen Resupply

236th Meeting of the Electrochemical Society

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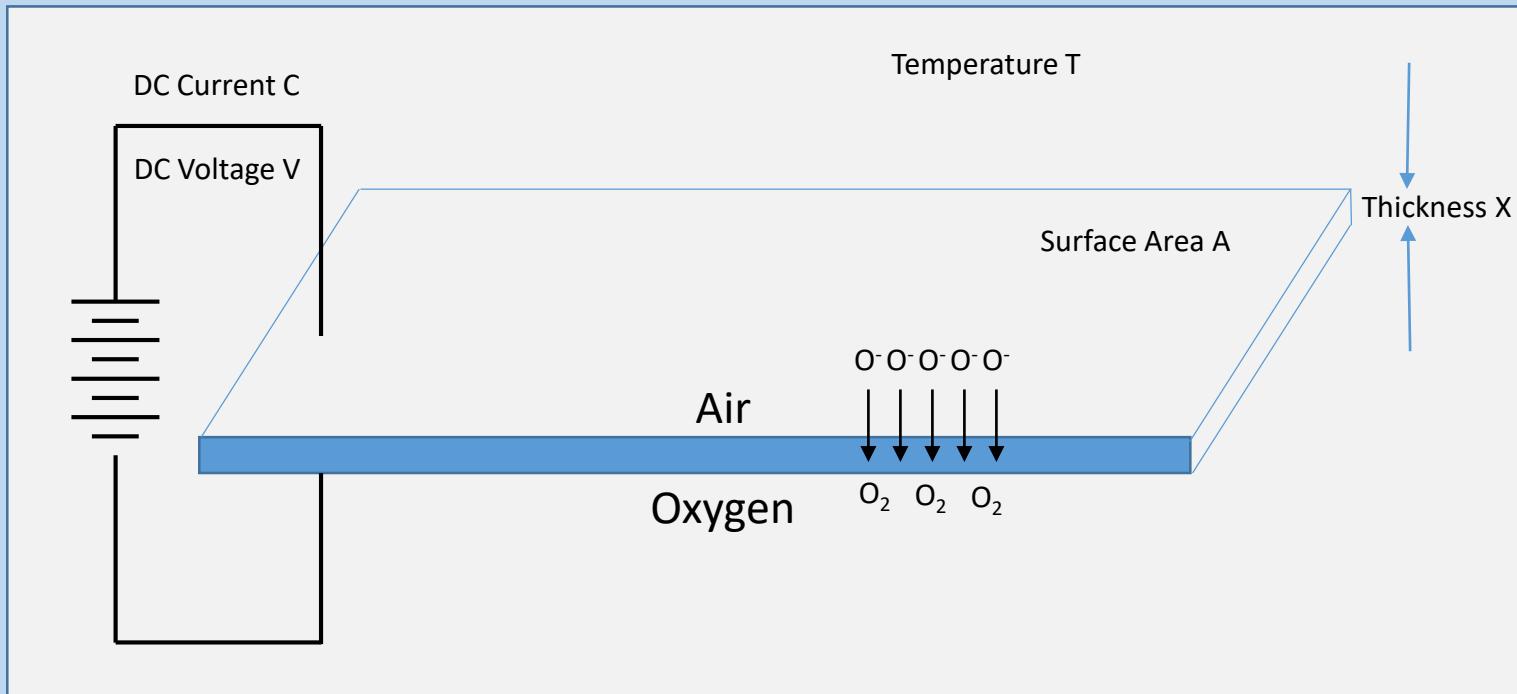
Agenda

- **Oxygen Transport Membrane Fundamentals**
- **Spacesuit Oxygen Tank Recharge Application**
- **eCOG-C Feasibility Assessment**
- **Summary**

Oxygen Transport Membrane Fundamentals

- **Function:** OTM can separate oxygen from other gases, because oxygen ions get electrochemically transported through some oxide materials “infinitely selective”
- **Hot:** Electrical resistance is essentially infinite at room temperature, common process temps are 650-750 C.
- **Thin:** Resistance across OTM is a function of membrane thickness. Thin membranes use less power.
- **Brittle:** The oxide materials that perform the best for OTMs are brittle, especially when they are thin.

Oxygen Transport Membrane Fundamentals



Oxygen Transport Membrane Fundamentals

- One solution to the hot, thin, brittle problem: OTM is embedded in a multi-layer wafer, that supports the OTM structurally, and creates a reservoir for oxygen



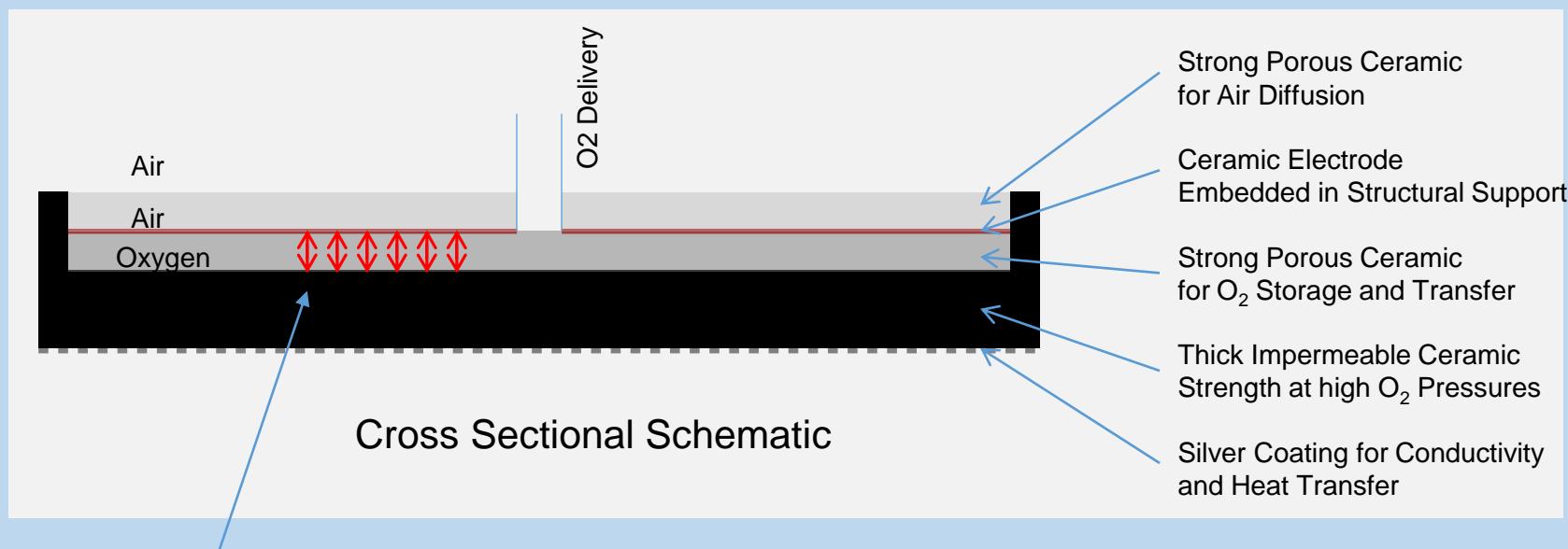
Photo of cell stack



Photo of cutaway

Oxygen Transport Membrane Fundamentals

- The current OTM has an internal reservoir in each wafer. Oxygen puts the wafer in tension. The wafer can hold 300 psi, but not 600 psi.



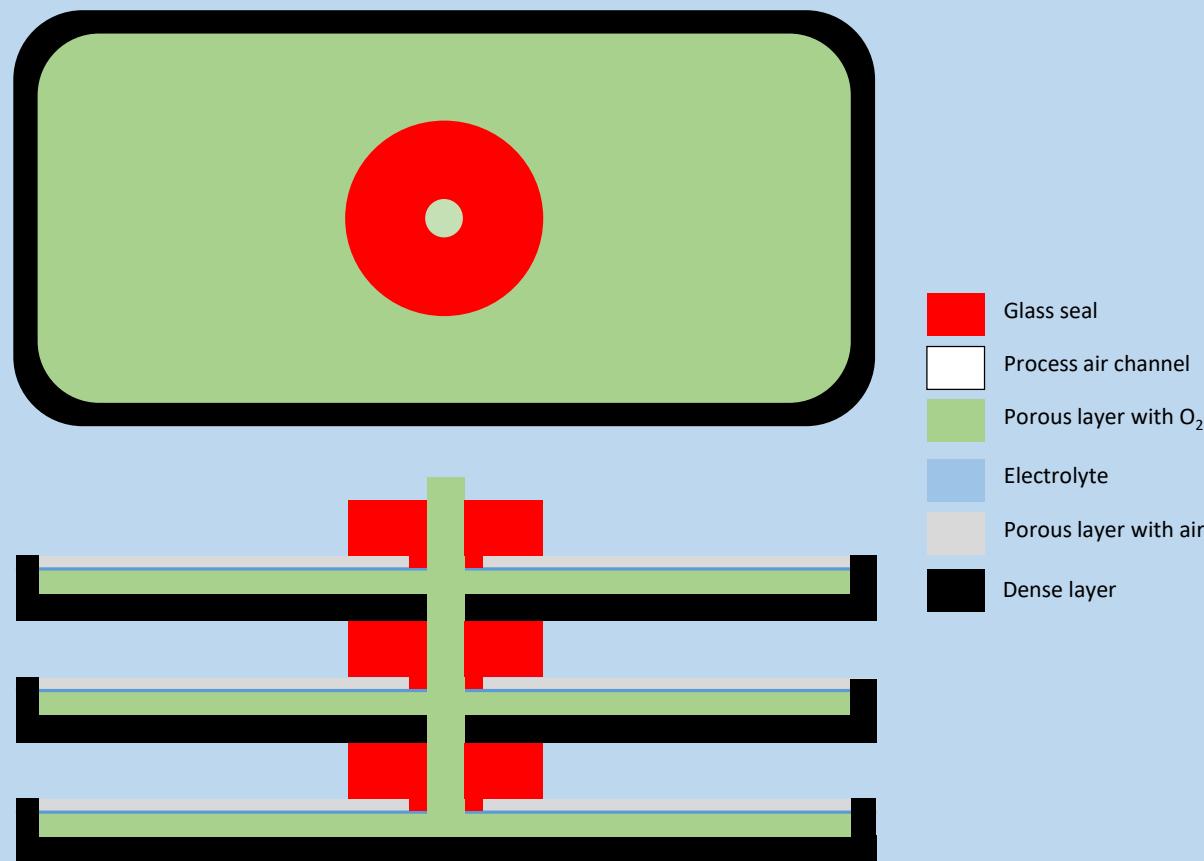
Oxygen exerts pressure force on the wafer

Spacesuit O₂ Tank Recharge

High Pressure O ₂ Gas Tanks	Pressure: No Issues: Pressurization is done prior to launch Purity: No Issues: Purity can be validated prior to launch, risk of contamination after launch is low Safety: Favorable: No energy is added to the system during the mission Summary: Great for missions with few planned EVAs. Concerns about size and weight for exploration
Cryogenic Liquid O ₂ (LOX)	Pressure: Pressurization occurs as heat is added to the system Purity: No Issues: Purity can be validated prior to launch, risk of contamination after launch is low Safety: Lots of stored energy. Apollo 13 mishap involved LOX Summary: Best suited for short duration missions with many planned EVAs
Low Pressure Water Electrolysis + Drier + Mechanical Compressor	Pressure: Demonstrated with technology demonstrator systems Purity: In-flight purity verification will likely be required; contamination risk is credible (compressor) Safety: Adding energy to O ₂ during mission – compressors that keep O ₂ cool can be large Summary: May trade well for exploration missions, especially if reliable purity verification is developed
Low Pressure Water Electrolysis + PSA O ₂ Concentrator + Mechanical Compressor	Pressure: Demonstrated with technology demonstrator systems Purity: In-flight purity verification will likely be required; severe concerns about PSA product purity Safety: Adding energy to O ₂ during mission – compressors that keep O ₂ cool can be large Summary: PSA product purity concerns need to be addressed before this can be evaluated
High Pressure Water Electrolysis + O ₂ Drier	Pressure: Demonstrated with technology demonstrator systems (thick end caps) Purity: Must remove water, verification may be easier if only water needs to be measured Safety: Fundamental safety issue: stack has hydrogen, high pressure O ₂ , and ignition source Summary: Safety issues cannot be designed out of the system
PEM Electrolyte Electrochemical O ₂ Compressor	Pressure: 14,000 kPa demonstrated with lab prototype Purity: Must remove water, verification may be easier if only water needs to be measured Safety: Fundamental, but less severe: solid fuel (not H ₂), high pressure O ₂ , and ignition source Summary: Kinetics needs to be demonstrated before this can be evaluated
Electrochemical Oxygen Generator & Compressor (eCOG-C)	Pressure: 14,000 kPa demonstrated with lab prototype Purity: No in-flight purity verification needed Safety: Adding energy to O ₂ during mission - high temp hazards, but fire triangle ok (no fuel) Summary: Good potential for human exploration: solid state, high purity, no fuel near O ₂

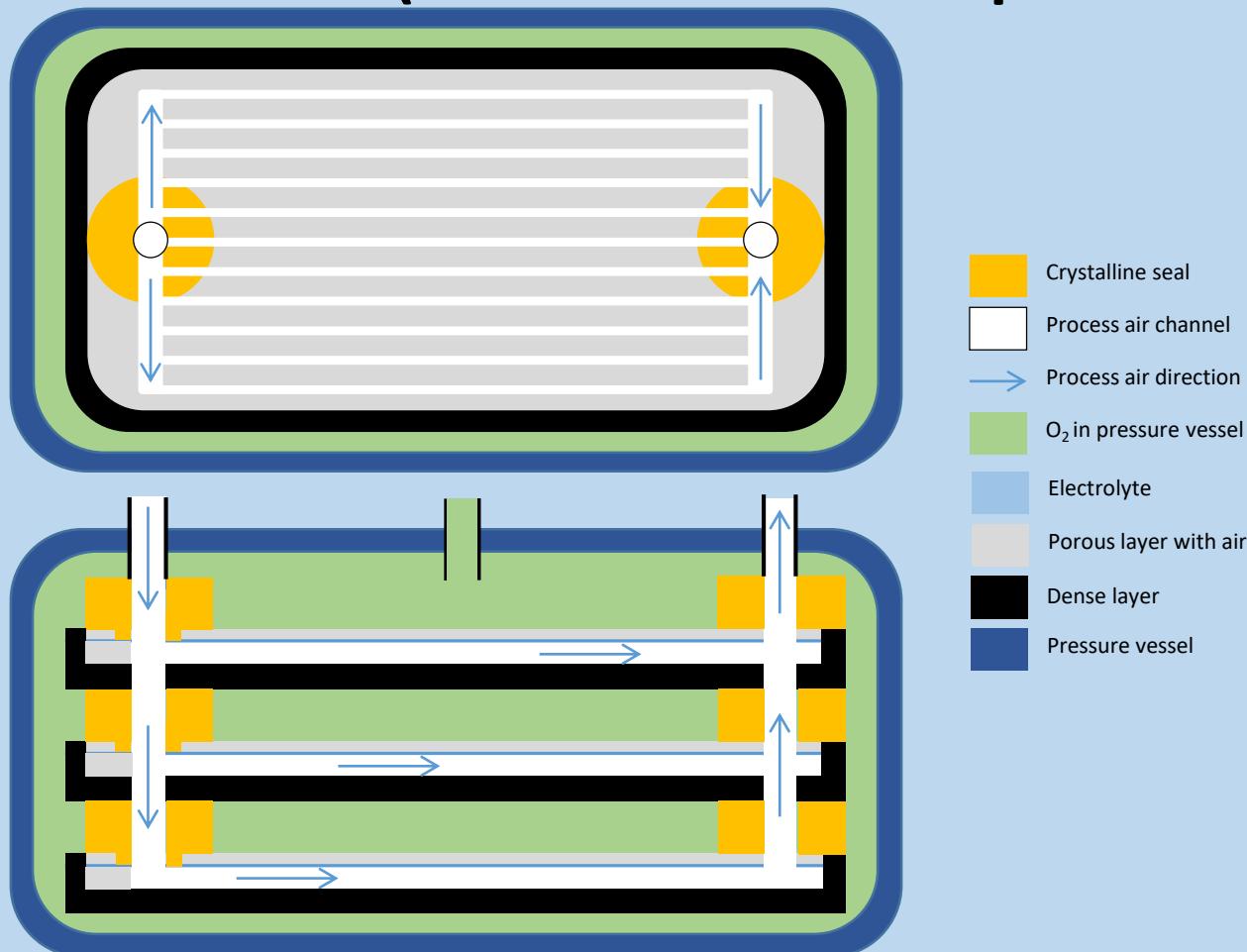
Feasibility Assessment

- The current OTM wafer has an oxygen reservoir internal to the wafer. Glass seals connect the wafers

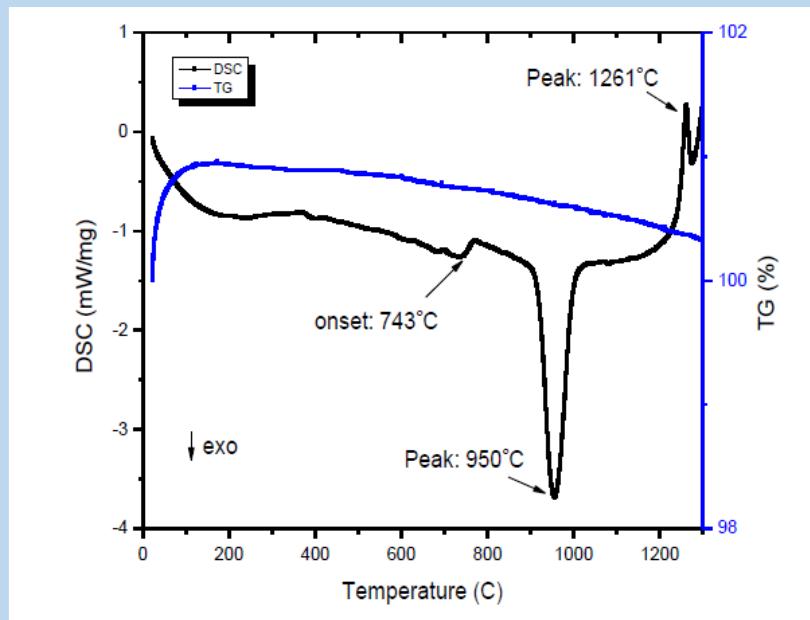
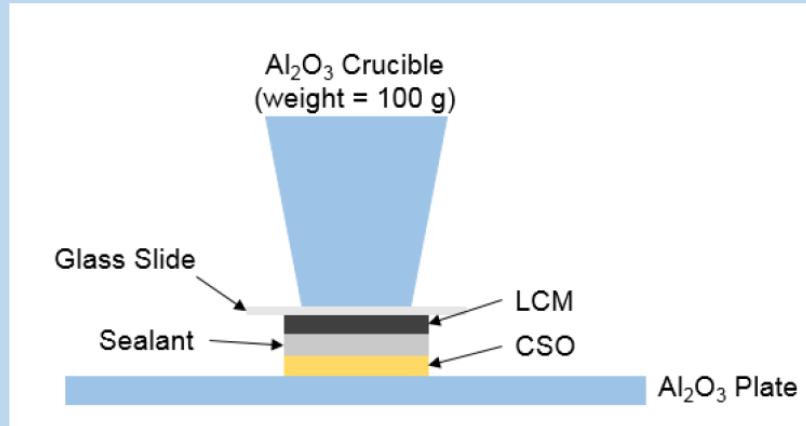


Feasibility Assessment

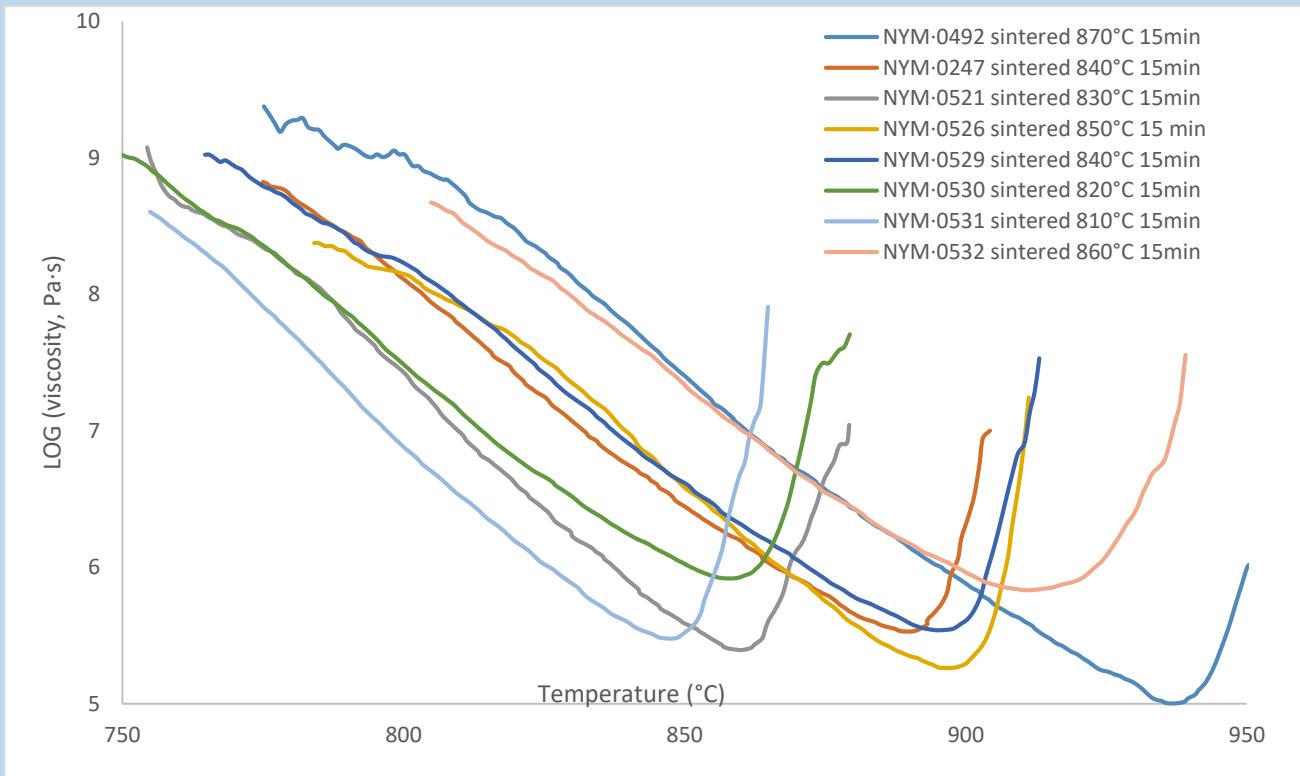
- eCOG-C flows low pressure air internally, oxygen is external to the wafer (contained inside a pressure vessel)



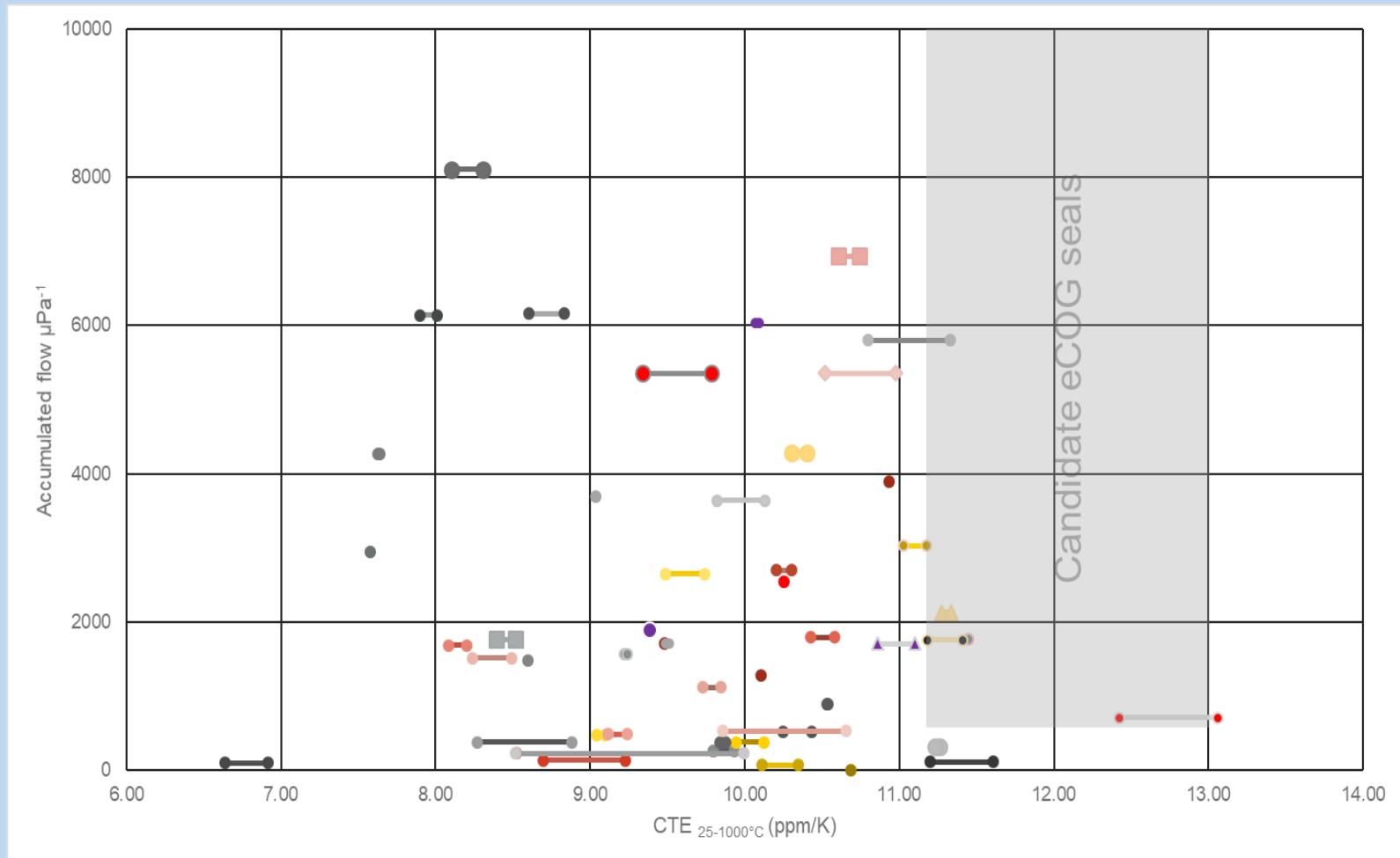
Feasibility Assessment



Feasibility Assessment

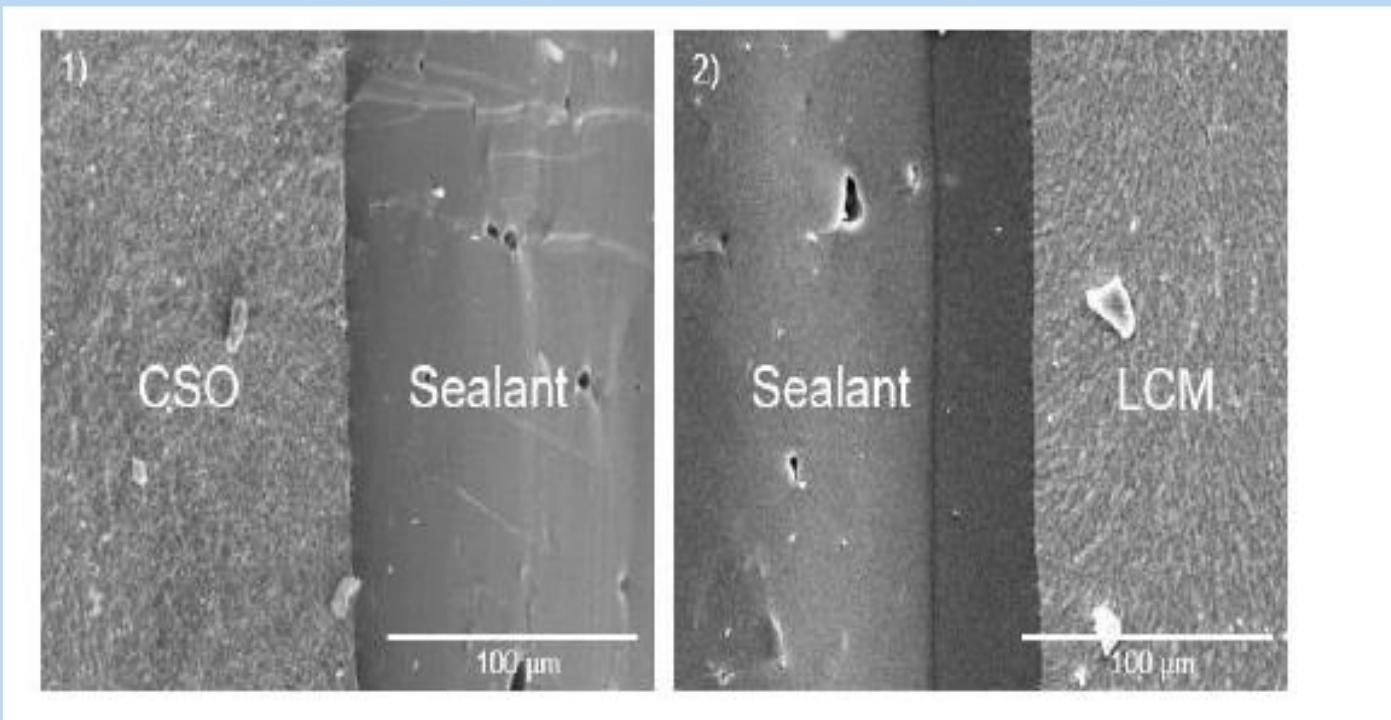


Feasibility Assessment



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Feasibility Assessment



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Summary

- Wafers can be reconfigured to place the oxygen external to the wafer. This puts the wafer in compression and makes >3000psi oxygen delivery possible.
- The #1 technical risk is developing a crystalline interconnecting seal that is strong, adheres to the wafers, and has a coefficient of thermal expansion that matches the wafer.
- The #2 technical risk is developing a modified wafer that has internal flow channels.
- Preliminary assessment of seal and wafer is favorable

Thank You