

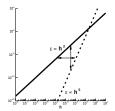


# High-Order Method Development at Langley: Past, Present and Future

NASA Langley Research Center Computational Fluid Dynamics Peer Review

Mark H. Carpenter
Computational AeroSciences Branch

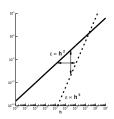
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#### **Outline**



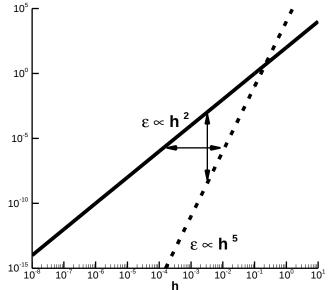
- Why High-Order / History / Impediments
- "Current" Langley Activity
  - Discontinuous Galerkin
  - Energy Stable WENO
- Questions

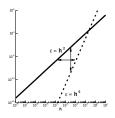


### Why High-Order Methods (HOM)



- Problems are quickly getting larger and unsteady
  - Model whole tunnel with sting
  - Unsteady fluid-structures interaction
- •Kreiss and Oliger:  $u_t + au_x = 0$ ,  $u(x,0) = e^{(i kx)} = ==>>> \epsilon(k) = [\beta k a T_f(\frac{2\pi}{P})^p]$ 
  - $\bullet$  P = points/wavelength;  $T_f$  = final time; beta = constant; p = polynomial order
    - LES of wing, Re =  $10^7$ :
      - "Grand Challenge" in 4 decades



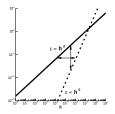


#### **How Mature is CFD?**

(6 orders already; How many more?)



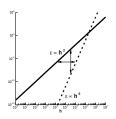
- DISCRETIZATIONS: 2 => 2½ orders
  - High-Order spatial operators yield 1  $\frac{1}{2}$  => 2 orders improvement
    - Limiters / Solvers / Grid adaptation
  - High-Order temporal operators could yield ½ order of improved efficiency
    - Automated integration with 4<sup>th</sup>- order methods
      - Temporal error estimators / Variable timestepping
  - Local complement: 1 1/2 FTE (i.e. Langley CS/Contractor, NIA, . . .)
- SOLVERS: 1 ½ => 2 orders
  - Far from optimal but far from easy! Slow progress.
  - Local complement : 1 FTE
- GRID ADAPTION: 2 orders
  - Adjoint error estimators
  - Local complement : 1 FTE
- David Keyes (Columbia) & Steven Jardin (Princeton): similar est!



### **Barriers and Challenges for HOMs**



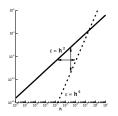
- Circa 2010
  - Discontinuous solutions
    - Unresolved features, bad grids, . . .
    - Mathematical foundation not as mature
    - 95% robustness test
  - Efficient Solvers
    - Inherent stiffness of discrete operator
    - Indefinite nature of matrices
    - Lack inevitable fine-tuning of algorithms
    - Analysis (e.g. Unstructured methods)



## **High-Order Methods at Langley Past: A Long and fruitful history!**



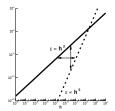
- Applications driving HO method use and development
  - Transition onset prediction and control
  - Turbulence modeling and control
  - Noise prediction and control
  - Airfoil design
  - DNS (temporal / spatial) and LES
- Langley Developers / Users
  - Atkins, Balakumar, Carpenter, Casper, Choudhari, Chang, Kumar, Li,
     Harris, Malik, Rudy, Singer, Streett, Vatsa, Watson, Zang . . .
- FY08 and before: Focused HO Method development group (>= 1 FTE)
  - Project funded
- FY09 and after: Distributed development
  - Project funded



## High-Order methods at Langley Present: A new funding Paradigm



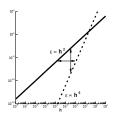
- Applications driving HO method use and development: Same
  - -Transition/Turbulence/Noise/Design/DNS/LES
- Current Langley Efforts
  - -CESE; (4th-order CD): Chang
  - -Vulcan; (ESWENO): Carpenter/White; NN; (DG): Atkins
  - -Overflow; (WENO): Buning
  - -FUN3D; (DG): (K-exact FV):
  - -NoName; (WENO): Balakumar
- Langley External Efforts
  - -Darmofal; MIT Project X; DG-FEM; Monitor: Atkins
  - -Mavripilis; Wyoming; DG-FEM/ HO-Temporal; Monitor Thomas
  - -Wu; NIA; WENO/DNS; Monitor Choudhari
  - -Fasel; Arizona; HO-CD / DNS; Monitor Choudhari
  - -Martin; Princeton; WENO / DNS; Monitor Gnoffo
  - -Zhong/Kim; UCLA; WENO / DNS; Monitor Choudhari



#### **Outline**



- Why High-Order / History / Impediments
- "Current" Langley Activity
  - Discontinuous Galerkin (Harold Atkins)
  - Energy Stable WENO
- Questions



## People and Places 15 years of DG at Langley



#### Lead Discontinuous Galerkin

Harold Atkins (LaRC)

### **Contributors**

Chi-Wang Shu and student (Brown U.)

David Lockard (LaRC)

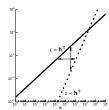
David Keyes and student (ICASE, Columbia U.)

Fang Hu (Old Dominion U.)

Dimitri Mavriplis (U. Wyoming)

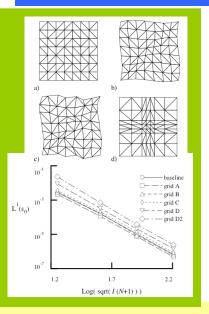
Brian Helenbrook and student (Clarkson U.)

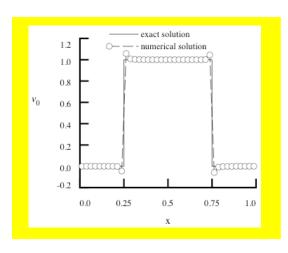
Chau-Lyan Chang (LaRC)

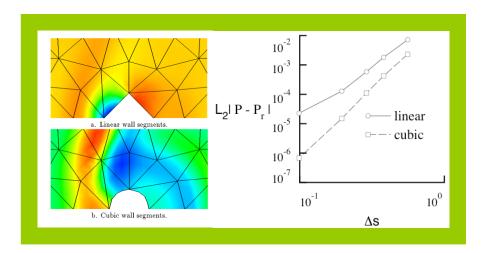


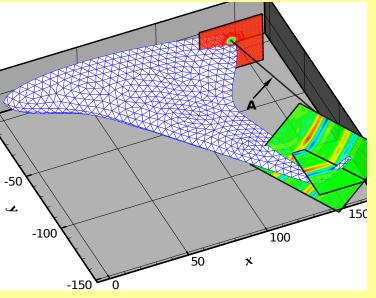
## **One-Page Summary of Early Work**

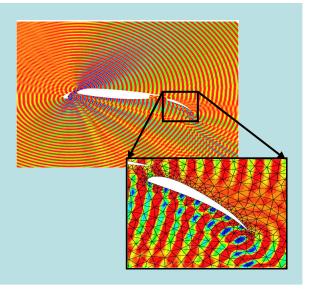


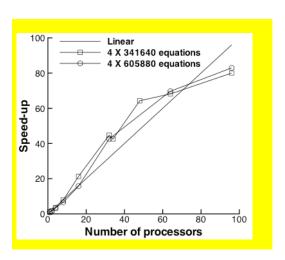


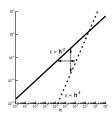








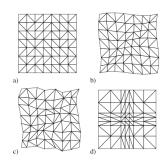


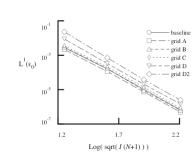


## Rate of Grid Convergence and Sensitivity to Mesh Smoothness



#### **Numerical Test**





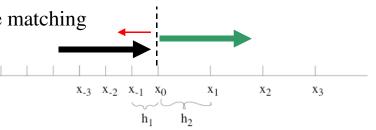
#### **Analysis**

assume:

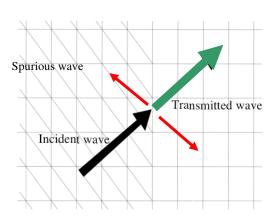
$$\mathbf{u}_h^n(\xi, t) = e^{-i\omega t} \lambda^n P_h(\xi)$$
 where  $\lambda = e^{iK_h}$ 

choose  $\omega$ , solve for  $\lambda$ 

apply interface matching



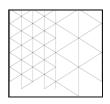
#### Two Dimensional



#### Applicable to:

- change in grid topology
- change in flux function
- change in polynomial order
- boundary conditions





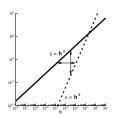


#### **Conclusions:**

Spurious modes ...

- are highly damped.
- amplitude's converge to zero with order of discretization.

Mesh Smoothness is not an issue



## Rate of Grid Convergence continued ... An Unexpected Finding



assume:

$$\mathbf{u}_h^n(\xi, t) = e^{-i\omega t} \lambda^n P_h(\xi)$$
 where  $\lambda = e^{iK_h}$ 

choose  $\omega$ , solve for  $\lambda$ 

$$(1 - \gamma)[G(iK)\lambda - H(iK)] + (-1)^{p+1}(1 + \gamma)[G(-iK)\frac{1}{\lambda} - H(-iK)] = 0$$

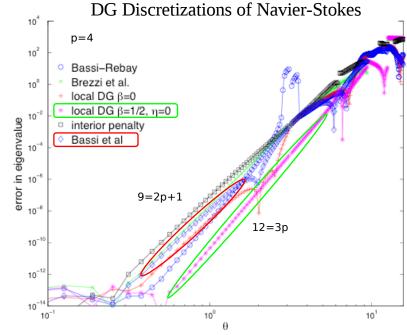
$$\begin{array}{lll} p=1 \Longrightarrow & G(x)=1-\frac{1}{3}x & H(x)=1+\frac{2}{3}x+\frac{1}{6}x^2 \\ p=2 \Longrightarrow & G(x)=1-\frac{2}{5}x+\frac{1}{20}x^2 & H(x)=1+\frac{3}{5}x+\frac{3}{20}x^2+\frac{1}{60}x^3 \\ p=3 \Longrightarrow & G(x)=1-\frac{3}{7}x+\frac{1}{14}x^2-\frac{1}{200}x^3 & H(x)=1+\frac{4}{7}x+\frac{1}{7}x^2+\frac{2}{105}x^3 \\ & & & & & & & & \\ \end{array}$$

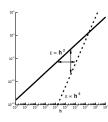
$$\lambda = \frac{H(iK)}{G(iK)} = \text{ exact Pade approximation of } e^{iK} \text{ to order } 2p+2$$

$$Re(K_h) - K = O^{2p+2}$$
 and  $Im(K_h) = O^{2p+1}$ 

### super-convergence

- -- proven analytically
- -- 1D and 2D propagation





## The Paradox of Super-Convergence



eigenvalue is super-convergent

- wavelength
- propagation speed

 $\mathbf{u}_{h}^{n}(\xi,\,t)=e^{-i\omega t}\lambda^{n}P_{h}(\xi)$ 

- growth/decay

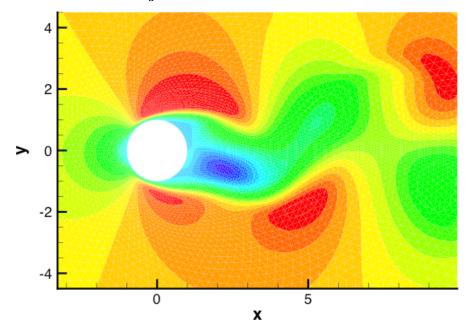
eigenfunction is <u>not</u> super-convergent

- instantaneous point values
- precise mode shape

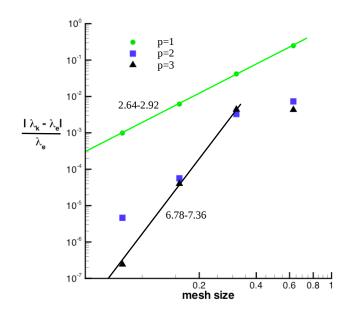
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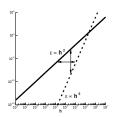
### Naturally occurring features

Vortex shedding from a viscous cylinder Mach = 0.2, Re<sub>d</sub>=200



#### Convergence of period

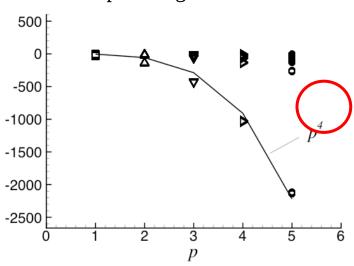




## **Solvers:** Challenges of DG for Navier Stokes

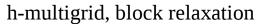


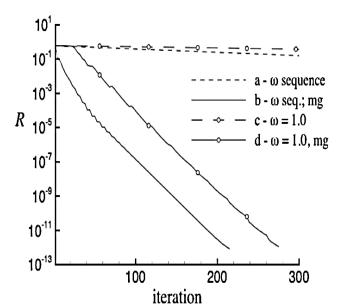
#### **Explicit Eigenvalues**



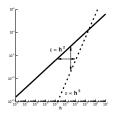
 $dt \propto h^2/|max \text{ eigenvalue}|$ 

Stiffness  $\approx$  |(max eigenvalue)/(min eigenvalue)|





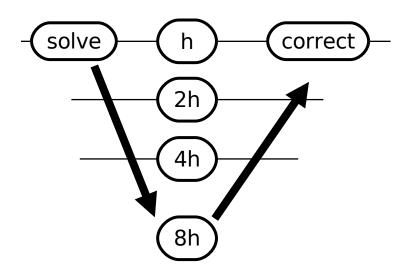
Theoretically possible... but implementation on unstructured grids is difficult.



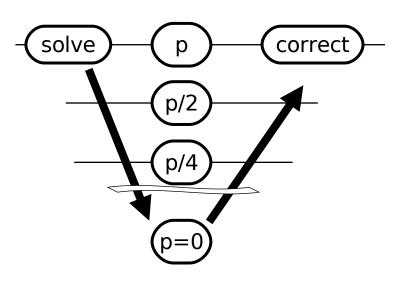
## **P-multigrid Solution Method**



#### h-Multigrid



#### P- Multigrid



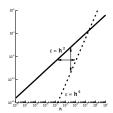
### Each Coarsening:

- -reduces number of unknowns
- -reduces stiffness

Both contributions are critical

At the p=0 level, apply:

- -h-multigrid
- -GMRES
- -direct solver
- -relaxation

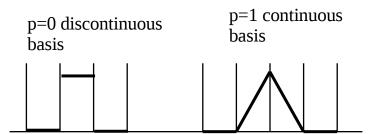


### **Analysis of restriction to p=0**

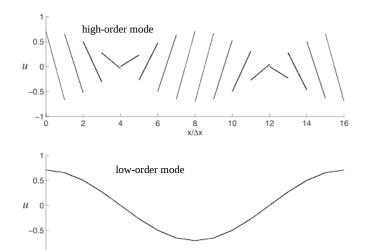


## Convergence Analysis: 2-level Versus Multi-level to p=0

p	Jacobi		Gauss-Seidel		
	2-level	multi- level	2-level	multi- level	0.14
1	0.70	0.70	0.80	0.80	
2	0.27	0.98	0.17	0.90	0.43
4	0.52	0.89	0.33	0.95	
8	0.80	1.00	0.67	0.98	



## Eigenfunctions of DG applied to diffusion for p=1



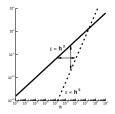
 $x/\Delta x$ 

#### P=0 (piecewise constant) **cannot**:

- represent either mode.
- provide a correction.

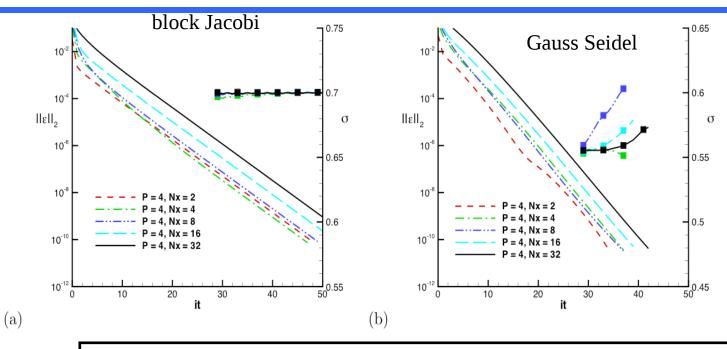
#### P=1-continuous can:

- resolve and correct the low-order mode
- provide reduction in number of variables



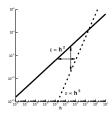
## P-multigrid: Analysis of DG applied to diffusion





Mesh independent convergence (the holy grail!), but with two caveats:

- 1. Careful construction of low-P equations.
- 2. Stop decent before P=0 -> restrict to P=1 continuous

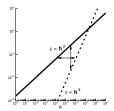


### **Conclusions for DG**



- Rigorous analysis of DG's accuracy properties:
  - insensitive to mesh smoothness,
  - proofs of super-convergence,
  - insight into nature of DG's super-convergence properties.
- Uncovered and remedied several hidden flaws in P-multigrid:
  - requires careful construction of low-order equations,
  - restriction to p=1 continuous basis instead of p=0 discontinuous.

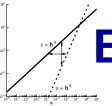
Careful analysis has helped to uncover and resolve several potential pitfalls, and has greatly improved our understanding of the strengths and limitations of the DG method.



#### **Outline**

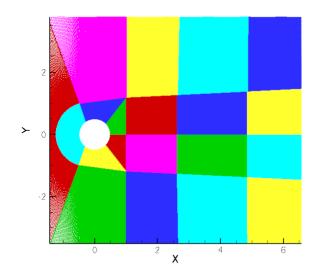


- Why High-Order / History / Impediments
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# **Boundary Closures for ESWENO**Schemes

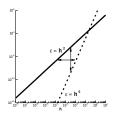




**Travis Fisher** (Purdue University)

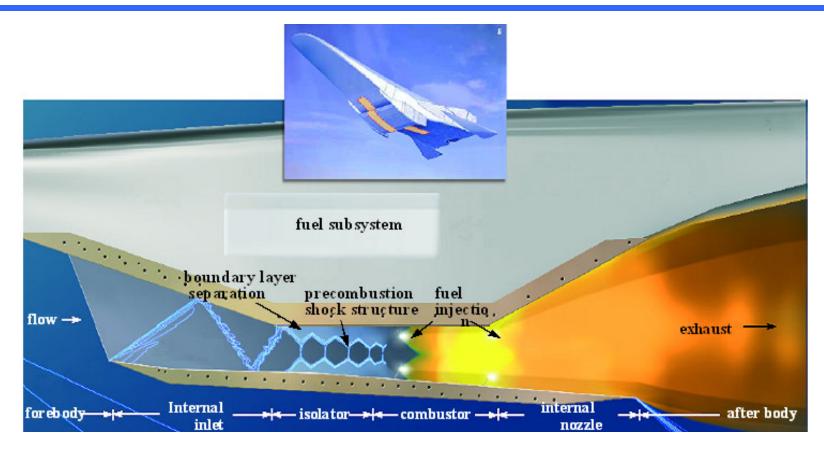
Nail K. Yamaleev (North Carolina A&T State University)

Mark H. Carpenter (NASA Langley Research Center)

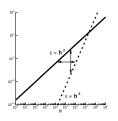


## **Applications with Shocks**





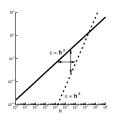
Hypersonic/Supersonic/Subsonic (Combustion, Shock BL interaction, BL stability, transition, turbulence,...)



## Big Picture: Complex Geometries What's the plan?



- HOFD for Complex Geometries: Summation-by-parts HOFD (SBP)
  - High-Order Block Structured Finite-Difference: JCP 111(2), pp 220, 1994. (SSH 98)
  - Multi-Domain HO FD: JCP 148(2), pp. 341, 1999. (SSH 54)
    - Only C\_0 Block Interface Connectivity: Geometric Flexibility
    - Penalty Interface Coupling between Blocks (SAT)
    - SBP-SAT: Design Order Accurate, L<sub>2</sub>-Stability, Conservation
- Plan of Attack: Four Development Phases for ESWENO
  - Stability of WENO schemes on 1D periodic Domains (ESWENO):
     JCP 228(11), pp. 4248, 2009. (SSH 1)
  - Stability of Boundary Closure Operators
    - 4th-Order Central: NASA/TM-2009-216166
    - 6th-order/8th-order central
  - Stability of discontinuity crossing interface
  - Nonlinear Stability proofs



## 6th-order WENO Schemes:



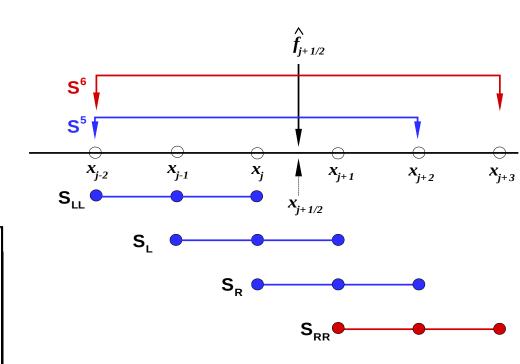
$$U_t + F_x = 0$$

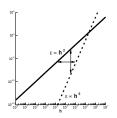
$$\frac{du_{j}}{dt} + \frac{f_{j+1/2}^{W} - f_{j-1/2}^{W}}{\Delta x} = 0$$

### The 6th-order WENO flux

$$f_{j+1/2}^{W} = \omega_{j+1/2}^{LL} f_{j+1/2}^{LL} + \omega_{j+1/2}^{L} f_{j+1/2}^{L} + \omega_{j+1/2}^{RR} f_{j+1/2}^{RR} + \omega_{j+1/2}^{RR} f_{j+1/2}^{RR}$$

$$\begin{bmatrix} f_{j+1/2}^{LL} \\ f_{j+1/2}^{LL} \\ f_{j+1/2}^{LL} \\ f_{j+1/2}^{LL} \end{bmatrix} = \frac{1}{6} \begin{pmatrix} 2 & -7 & 11 & & & 0 \\ & -1 & 5 & 2 & & \\ & & 2 & 5 & -1 & \\ 0 & & & 11 & -7 & 2 \end{pmatrix} \begin{bmatrix} f_{j-2} \\ f_{j-1} \\ f_{j} \\ f_{j+1} \\ f_{j+1} \\ f_{j+1} \end{bmatrix}$$





## **6th-order WENO Schemes**

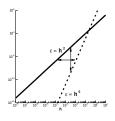


## The weights should satisfy the following properties:

$$0 \le \omega_{j+1/2}^r \le 1$$
,  $\sum_r \omega_{j+1/2}^r = 1$ ,  $\forall j$   
 $\omega_{j+1/2}^r = d_r + O(\Delta x^{p-s+1})$ 

## Conventional weights of Jiang and Shu:

$$\omega_{j+1/2}^{r} = \frac{\alpha_{r}}{\sum_{l} \alpha_{l}}, \quad \alpha_{r} = \frac{d_{r}}{(\varepsilon + \beta_{r})^{2}}, \quad \beta_{r} = \sum_{k=1}^{s-1} \Delta x^{2k-1} \int_{x_{j-1/2}}^{x_{j+1/2}} \left(\frac{d^{k} q_{r}(x)}{dx^{k}}\right)^{2} dx \implies \omega^{r} = d^{r} + O(\Delta x^{2})$$

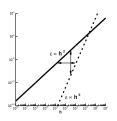


## **Useful References**



#### **ENO/WENO SCHEMES**

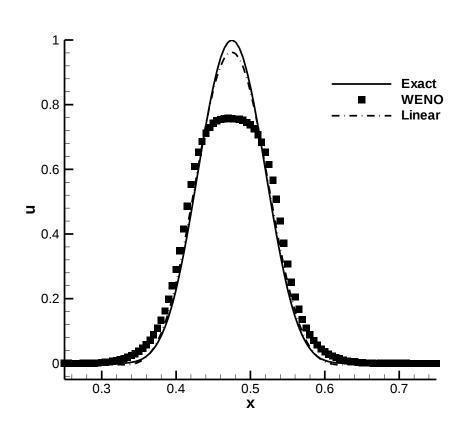
- A. Harten, B. Engquist, S. Osher, and S. Chakravarthy "Uniformly high order essentially non-oscillatory schemes, III," *J. of Computational Physics*, Vol. 71, pp. 231-303, 1987.
  - Adaptively chooses "smoothest" stencil from s candidates.
     Provides  $s^{th}$ -order accuracy.
- X.-D. Liu, S. Osher, and T. Chan
   "Weighted Essentially Non-oscillatory Schemes," *J. of Computational Physics*, Vol. 115, pp. 200-212, (1994).
  - Uses a weighted convex combination of all s candidate stencils. Provides  $(s+1)^{th}$ -order accuracy.
- G. Jiang and C.-W. Shu
   "Efficient implementation of weighted ENO schemes," *J. of Computational Physics*, Vol. 126, pp. 202-228, (1996). (SSH 698)
  - Generalization to the FD formulation. Provides  $(2s-1)^{th}$ -order accuracy on sufficiently fine meshes.

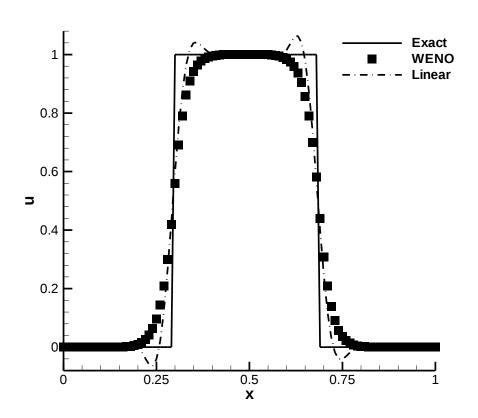


## **Remaining Issues: Accuracy**

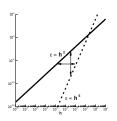


## **Left Achilles Heel of High-Order WENO Schemes**





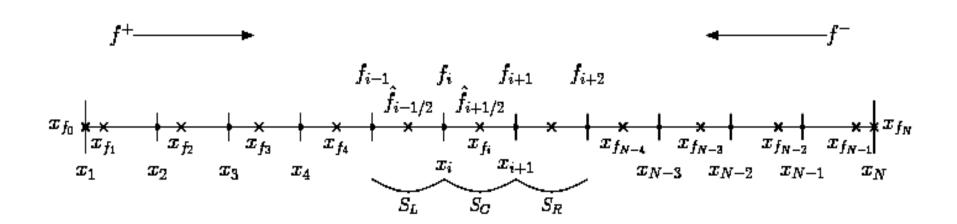
Moving pulse simulation using the 3rd-order WENO scheme



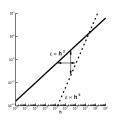
## **Remaining Issues: Boundaries**



## Right Achilles Heel of High-Order WENO Schemes



Grabbing data outside the domain: 4th-order WENO scheme

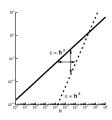


## **Needed Improvements for WENO**



## Improvements needed by conventional WENO:

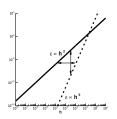
- The WENO schemes are too dissipative as compared with the corresponding target linear schemes
  - Typically based on upwind operators.
  - •The order of the conventional WENO schemes deteriorates from (2s-1) to s at critical points.
- For some steady state problems, the residual cannot be driven to the machine zero.
- No stability proofs are available for discontinuous solutions of hyperbolic systems.
- Boundary stencils and extension to complex geometry



## **Energy Stable WENO**Accomplishments to date (ROYOT)



- Energy Stable WENO (ESWENO)
  - A WENO-type scheme that is stable in the L\_2 norm by construction for continuous and discontinuous solutions
- Developed New weights
  - Faster convergence to the underlying linear scheme
  - Improved shock-capturing capabilities
- Boundary Closures for 4th-order ESWENO
  - Design order accurate (3-4-3)
     Full stencil biasing up to boundaries (almost)
     L\_2 Stability
- Maintains original properties
  - Design order of accuracy for smooth solutions/extrema
  - Conservative (Lax-Wendroff theorem)
  - Essentially nonoscillatory solutions



## **Energy Estimates**



## **Continuous problem**

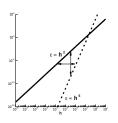
$$\frac{\partial u}{\partial t} + \frac{\partial f}{\partial x} = \sum_{n=1}^{N} (-1)^{n-1} \frac{\partial^n}{\partial x^n} \left( \mu_n(u) \frac{\partial^n u}{\partial x^n} \right) \quad f = au, \quad a > 0, \quad \mu_n \ge 0 \quad (1)$$

Multiplying Eq. (1) by u and integrating over [0,1] yields

$$\frac{1}{2} \frac{d}{dt} \|u\|_{l_2}^2 + \frac{1}{2} a u^2 \Big|_{0}^{1} = \sum_{n=1}^{N} \int_{0}^{1} (-1)^{n-1} u \frac{\partial^n}{\partial x^n} \left( \mu_n(u) \frac{\partial^n u}{\partial x^n} \right) dx$$

Integrating RHS by parts yields

$$\frac{d}{dt}||u||_{l_2}^2 + au^2|_0^1 = -2\sum_{n=1}^N \int_0^1 \mu(u) \left(\frac{\partial^n u}{\partial x^n}\right)^2 dx \le 0$$



## **Energy Estimates (cont.)**



## **Discrete problem**

$$\frac{\partial \mathbf{u}}{\partial t} + P^{-1}Q\mathbf{f} = -\sum_{n=0}^{S} P^{-1}D_1^n \Lambda_n [D_1^n]^T \mathbf{f}$$
 (2)

Multiplying Eq. (2) by  $\mathbf{u}^T \mathbf{p}$  and adding it to its transpose yields

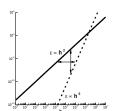
$$\frac{d}{dt} \|u\|_P^2 + a\mathbf{u}^T (Q + Q^T)\mathbf{u} = -a\sum_{n=0}^S ([D_1^n]^T \mathbf{u})^T (\Lambda_n + \Lambda_n^T) D_1^n]^T \mathbf{u}$$

#### What if:

$$P = P^{T} ; \psi^{T} P \psi > 0 ; Q + Q^{T} = Diag[-1,0,\cdots,0,1] ; \psi^{T} (\Lambda + \Lambda^{T}) \psi \geq 0$$
$$\frac{\partial f}{\partial x} = P^{-1} Q f + \sum_{n=0}^{S} P^{-1} D_{1}^{n} \Lambda_{n} [D_{1}^{n}]^{T} f + O(\Delta x)^{2s}$$

#### Then

$$\frac{d}{dt} \|\mathbf{u}\|_{P}^{2} + a \mathbf{u}^{2}|_{0}^{1} = -2 \sum_{n=0}^{N} \left( [D_{1}^{n}]^{T} \mathbf{u} \right)^{T} \Lambda_{n} \left( [D_{1}^{n}]^{T} \mathbf{u} \right) \leq 0$$



# **6th-order WENO Schemes: Are They Stable?**



Can WENO schemes be represented as  $\frac{\partial \mathbf{u}}{\partial t} + P^{-1}Q\mathbf{f} = -\sum_{n=0}^{S} P^{-1}D_1^n \Lambda_n [D_1^n]^T \mathbf{f}$ ?

The 6th-order WENO operator: 
$$[D_{W}\mathbf{f}]_{j} = \frac{f_{j+1/2}^{W} - f_{j-1/2}^{W}}{\Delta x}$$

$$D_{W} = \frac{1}{2}(D_{W} - D_{W}^{T}) + \frac{1}{2}(D_{W} + D_{W}^{T}) = P^{-1}Q + P^{-1}D_{W}^{sym}$$

Is  $D_w^{sym}$  Stable? Find a constructive stability test????

$$D_W^{sym} = P^{-1} M (A + A^T) M^T$$

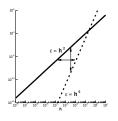
$$D_W^{sym} = P^{-1} \left( D_1^3 \Lambda_3^W [D_1^3]^T + D_1^2 \Lambda_2^W [D_1^2]^T + D_1^1 \Lambda_1^W [D_1^1]^T + D_1^0 \Lambda_0^W [D_0^3]^T \right)$$

$$\Lambda_3^W = \frac{1}{6} diag \left[ \omega_{j+5/2}^{LL} - \omega_{j+5/2}^{RR} \right]$$

$$\Lambda_{2}^{W} = \frac{1}{12} diag \left[ \omega_{j+3/2}^{LL} - 4\omega_{j+5/2}^{LL} + \omega_{j+3/2}^{L} - \omega_{j+1/2}^{R} + 4\omega_{j-1/2}^{RR} - \omega_{j+1/2}^{RR} \right]$$

$$\Lambda_{1}^{W} = \frac{1}{12} diag \left[ 3\omega_{j+1/2}^{LL} - 5\omega_{j+3/2}^{LL} + 2\omega_{j+5/2}^{LL} + \omega_{j+1/2}^{L} - \omega_{j+3/2}^{L} + \omega_{j-1/2}^{R} - \omega_{j+1/2}^{R} - 2\omega_{j-3/2}^{RR} + 5\omega_{j-1/2}^{RR} - 3\omega_{j+1/2}^{RR} \right]$$

$$\Lambda_0^{\!\scriptscriptstyle W} = \frac{1}{2} \operatorname{diag} \! \left[ - \omega_{j-1/2}^{\scriptscriptstyle LL} - \omega_{j-1/2}^{\scriptscriptstyle L} - \omega_{j-1/2}^{\scriptscriptstyle R} - \omega_{j-1/2}^{\scriptscriptstyle RR} + \omega_{j+1/2}^{\scriptscriptstyle LL} + \omega_{j+1/2}^{\scriptscriptstyle L} + \omega_{j+1/2}^{\scriptscriptstyle RR} + \omega_{j+1/2}^{\scriptscriptstyle RR} \right] = 0$$

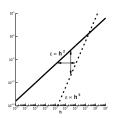


## **6th-order WENO Schemes**



## Conventional WENO schemes:

- No energy estimate/stability proof (that we could find)
- The WENO dissipation operator is **not positive semidefinite**, because for unresolved solutions  $\Lambda_n^W < 0$ ,  $n = \overline{1,3}$ !
- For discontinuous or unresolved solutions, the WENO artificial dissipation operator  $-P^{-1}\sum_{n=1}^{3}D_1^n\Lambda_n^w[D_1^n]^T$  could have positive eigenvalues, thus indicating that the scheme may become locally unstable.



## 6th-order ESWENO Schemes



## **Stability**

ESWENO scheme has **all the terms** of the 6th-order WENO scheme and also includes **an additional artificial dissipation term** 

$$\frac{\partial \mathbf{u}}{\partial t} + P^{-1}Q\mathbf{f} = -\sum_{n=0}^{3} P^{-1}D_1^n \Lambda_n^{ES} [D_1^n]^T \mathbf{f}$$

Central operator:  $P = \Delta xI$ ,  $Q = PD_W^{skew}$ 

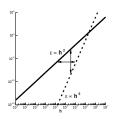
Additional artificial dissipation operator:  $\overline{D}_{ad} = -P^{-1} \sum_{n=1}^{3} D_1^n \overline{\Lambda}_n [D_1^n]^T$ 

$$\left(\overline{\lambda}_{n}\right)_{j,j} = \frac{1}{2} \left[ \sqrt{\left(\lambda_{n}^{W}\right)_{j,j}^{2} + \delta_{n}^{2}} - \left(\lambda_{n}^{W}\right)_{j,j} \right]$$

$$D^{ES} = D^W + \overline{D}_{ad}$$

$$\left(\lambda_n^{ES}\right)_{j,j} = \left(\lambda_n^W\right)_{j,j} + \left(\overline{\lambda_n}\right)_{j,j} = \frac{1}{2} \left[\sqrt{\left(\lambda_n^W\right)_{j,j}^2 + \delta_n^2} + \left(\lambda_n^W\right)_{j,j}\right] > 0 \text{ for any } u_j \text{ and } \forall j$$

CFD Development Peer Review: Jan 2010



## **6th-order ESWENO Schemes**



### The weights should satisfy the following properties:

$$0 \le \omega_{j+1/2}^r \le 1$$
,  $\sum_r \omega_{j+1/2}^r = 1$ ,  $\forall j$   
 $\omega_{j+1/2}^r = d_r + O(\Delta x^{p-s+1})$ 

## Conventional weights of Jiang and Shu:

$$\omega_{j+1/2}^{r} = \frac{\alpha_{r}}{\sum_{k=1}^{r} \alpha_{k}}, \quad \alpha_{r} = \frac{d_{r}}{(\varepsilon + \beta_{r})^{2}}, \quad \beta_{r} = \sum_{k=1}^{s-1} \Delta x^{2k-1} \int_{x_{j-1/2}}^{x_{j+1/2}} \left(\frac{d^{k} q_{r}(x)}{dx^{k}}\right)^{2} dx \implies \omega^{r} = d^{r} + O(\Delta x^{2})$$

## New weights with Smooth Consistency

$$\omega_{j+1/2}^{r} = \frac{\alpha_{r}}{\sum_{k} \alpha_{k}}, \quad \alpha_{r} = d_{r} \left( 1 + \frac{\tau_{5}}{\varepsilon + \beta_{r}} \right) \quad r = \{LL, L, R, RR\}$$

$$\tau_{5} = \begin{cases} \left( -f_{j-2} + 5f_{j-1} - 10f_{j} + 10f_{j+1} - 5f_{j+2} + f_{j+3} \right)^{2}, \text{ for } \varphi \neq 0 & \Rightarrow \quad \omega^{r} = d^{r} + O(\Delta x^{8}) \\ \left( f_{j-2} - 4f_{j-1} + 6f_{j} - 4f_{j+1} + f_{j+2} \right)^{2}, \text{ for } \varphi = 0 & \Rightarrow \quad \omega^{r} = d^{r} + O(\Delta x^{6}) \end{cases}$$

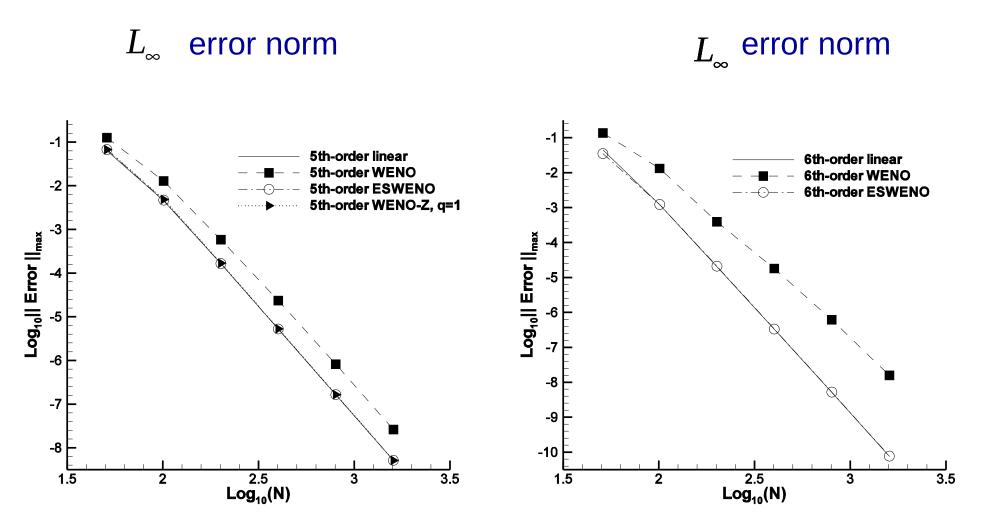
# $\mathbb{E} = h^{\frac{1}{2}}$ $\mathbb{E} = h^{\frac{1}{2}}$ $\mathbb{E} = h^{\frac{1}{2}}$ $\mathbb{E} = h^{\frac{1}{2}}$ $\mathbb{E} = h^{\frac{1}{2}}$

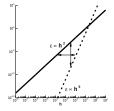
## **Numerical Results**

# NASA

## **Linear Wave Equation**

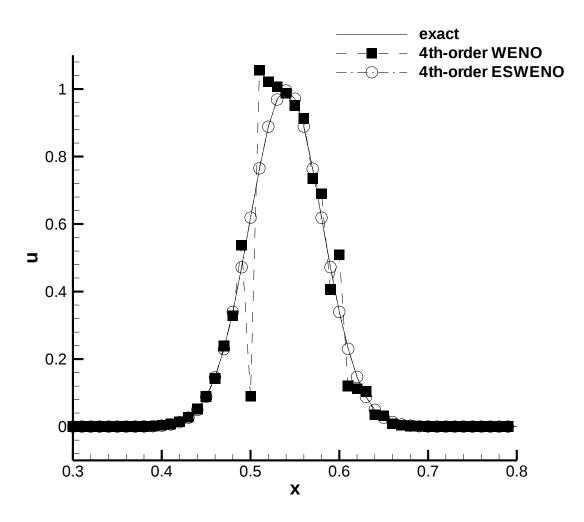
The 5th- and 6th-order WENO and ESWENO schemes (Gaussian pulse)



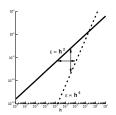


#### 4th-order WENO Schemes





Solution obtained with the 4<sup>th</sup>-order WENO scheme for the linear wave equation with initial condition  $u_0(x) = e^{-300(x-0.5)^2}$ 



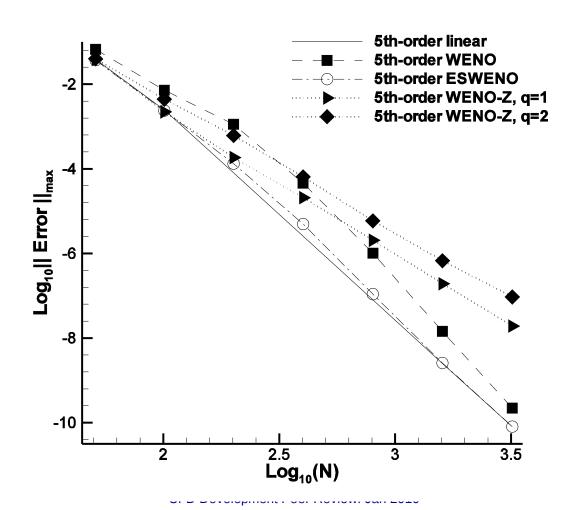
#### **Numerical Results (cont.)**

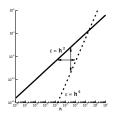


#### **Linear Wave Equation**

The 5th-order WENO and ESWENO schemes (solution with critical points)

$$L_{\infty}$$
 error norm

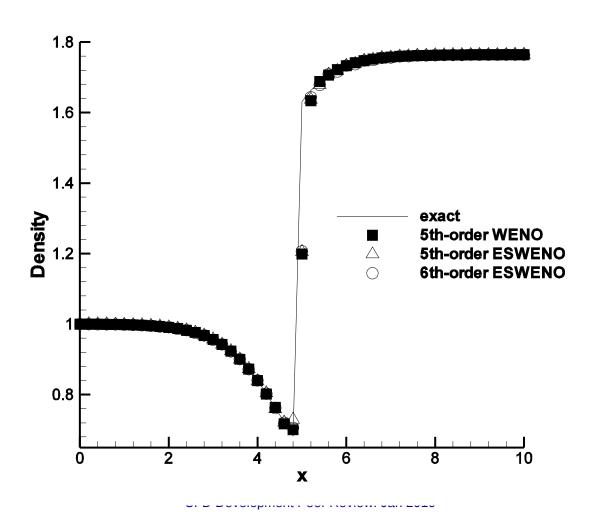


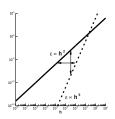


## Numerical Results (cont.) The Steady Quasi-1-D Euler Equations

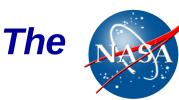


Quasi-1-D nozzle problem,  $5^{th}$  and  $6^{th}$ -order WENO and ESWENO schemes (Mach=1.5, J=51 grid points)

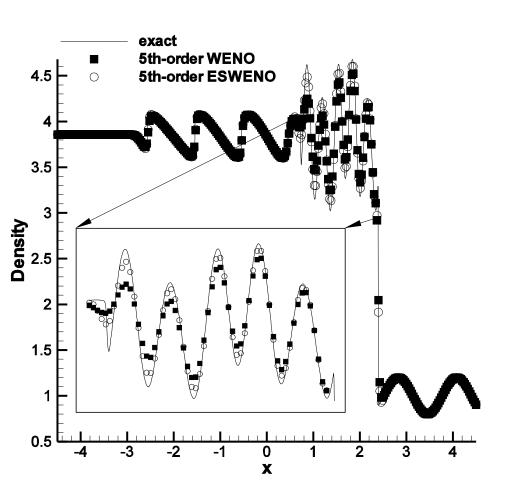


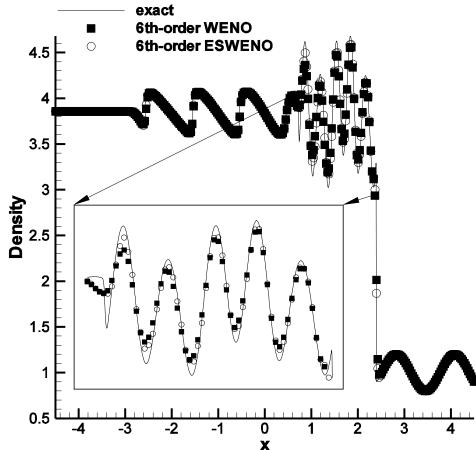


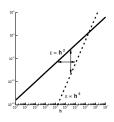
## Numerical Results (cont.) Unsteady 1-D Euler Equations



Shock/acoustic wave interaction problem, 5<sup>th</sup> and 6<sup>th</sup>-order WENO and ESWENO schemes (*J*=300 grid cells)







#### **Boundary Closures: The Plan?**

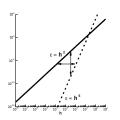


#### Goals

- Conventional WENO stencil biasing to boundary
   4th-order: only volume next to wall has reduced bias
- Use Centered SBP scheme as WENO target operator
  - Resolved flow then operator reduces to SBP operator
  - Prove stability/accuracy of target operator

#### Plan

- Derive general "4th-order" SBP discrete matrix (with BCs)
- Express WENO stencil biasing mechanics in matrix form
- Equate the two forms (matrices) and see what happens

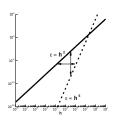


# **Boundary Closures: The Plan Summation-By-Parts (3-4-3)**



- Boundary stencils of  $O(\Delta x^{p-1}) \Rightarrow O(\Delta x^p)$  Global accuracy
  - B. Gustafsson, Math. Comp. 29, 130, 1975
- Summation-by-parts operators
  - H.-O. Kreiss, G. Scherer, Math. Aspects FE PDEs 1974
  - B. Strand, J. Comp. Phy., 110, 1, 1994. For example

$$P = \begin{pmatrix} \frac{1}{4} & \frac{19}{144} & \frac{-11}{216} & \frac{-7}{216} & 0 & 0 \\ \frac{19}{144} & \frac{65}{54} & \frac{5}{108} & \frac{23}{216} & 0 & 0 \\ \frac{-11}{216} & \frac{5}{108} & \frac{91}{108} & \frac{-7}{432} & 0 & 0 \\ \frac{-7}{216} & \frac{23}{216} & \frac{-7}{432} & \frac{55}{54} & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 1 \end{pmatrix}; \quad Q = \begin{pmatrix} \frac{-1}{2} & \frac{593}{864} & \frac{-37}{216} & \frac{-13}{864} & 0 & 0 \\ \frac{-593}{864} & 0 & \frac{191}{288} & \frac{5}{216} & 0 & 0 \\ \frac{37}{216} & \frac{-191}{288} & 0 & \frac{497}{864} & \frac{-1}{12} & 0 \\ \frac{13}{864} & \frac{-5}{216} & \frac{-497}{864} & 0 & \frac{2}{3} & \frac{-1}{12} \\ 0 & 0 & \frac{1}{12} & \frac{-2}{3} & 0 & \frac{2}{3} \\ 0 & 0 & 0 & \frac{1}{12} & \frac{-2}{3} & 0 \end{pmatrix}$$



## **Boundary Closures: SBP (3-4-3) General Parameterization**



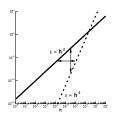
• Pentadiagonal 4th-order:  $Q_i = Pentadiagonal\left[\frac{1}{12}, \frac{-8}{12}, 0, \frac{8}{12}, \frac{-1}{12}\right];$ 

$$P = \Delta x egin{pmatrix} P_0 & 0 & 0 \ 0 & I & 0 \ 0 & 0 & P_0^{PT} \end{pmatrix} \;\; ; \quad Q = egin{pmatrix} Q_0 & Q_d & 0 \ -Q_d & Q_i & Q_d \ 0 & -Q_d^T & -Q_0^{PT} \end{pmatrix}$$

$$P_{0} = \begin{pmatrix} p_{11} & p_{12} & p_{13} & p_{14} \\ p_{12} & p_{22} & p_{23} & p_{24} \\ p_{13} & p_{23} & p_{33} & p_{34} \\ p_{14} & p_{24} & p_{34} & p_{44} \end{pmatrix} ; Q_{0} = \begin{pmatrix} \frac{-1}{2} & q_{12} & q_{13} & q_{14} \\ q_{12} & 0 & p_{23} & q_{24} \\ q_{13} & q_{23} & 0 & q_{34} \\ q_{14} & q_{24} & q_{34} & 0 \end{pmatrix} ; Q_{d} = \begin{pmatrix} \cdots & 0 & 0 & 0 \\ 0 & \cdots & 0 & 0 \\ -1 & 12 & 0 & \cdots & 0 \\ \frac{8}{12} & -1 & 0 & \cdots \end{pmatrix} ;$$

 General (3-4-3) two parameter solution after satisfying constraints on matrix properties and accuracy conditions

$$D = P^{-1}Q \; ; \; P = P^{T} \; ; \; \psi^{T}P\psi > 0 \; ; \; Q + Q^{T} = Diag[-1,0,\cdots,0,1] \; ;$$
 
$$P_{3-4-3} = P_{3-4-3}(\alpha_{1},\alpha_{2}) \; ; \; Q_{3-4-3} = Q_{3-4-3}(\alpha_{1},\alpha_{2})$$



### **WENO Stencil Biasing Mechanics**

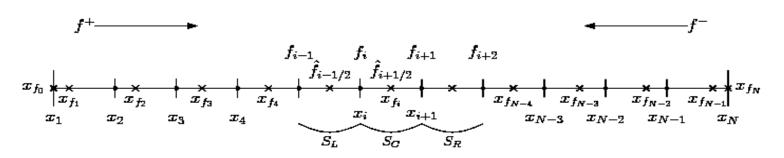


$$U_t + F_x = 0$$

$$\frac{du_{j}(t)}{dt} + \frac{f_{j+1/2}^{ES} - f_{j-1/2}^{ES}}{\Delta x} = 0$$

Complementary Grids: Solution grid and Flux grid

$$\mathbf{x}_N = [x_{1,} x_{2,} \cdots, x_{N-1}, x_N]^T$$
;  $\bar{\mathbf{x}} = [\bar{x}_{0,} \bar{x}_{1,} \cdots, \bar{x}_{N-1}, \bar{x}_N]^T$ 



$$f_{j+1/2}^{ES} = \bar{\omega}_{j+1/2}^{L} \bar{f}_{J+1/2}^{L} + \bar{\omega}_{j+1/2}^{C} \bar{f}_{J+1/2}^{C} + \bar{\omega}_{j+1/2}^{R} \bar{f}_{J+1/2}^{R}$$

$$\bar{f}^{L} = I_{fs}^{L} f \; ; \; \bar{f}^{C} = I_{fs}^{C} f \; ; \; \bar{f}^{R} = I_{fs}^{R} f$$

### Deriving SBP(3-4-3) Target Operator



$$U_t + F_x = 0$$

SBP matrix form:

$$D = P^{-1}Q$$
;  $P = P^{T}$ ;  $\psi^{T}P\psi > 0$ ;  $Q + Q^{T} = Diag[-1,0,\dots,0,1]$ ;

WENO Stencil Biasing matrix form:

$$\frac{\boldsymbol{f}_{j+1/2}^{ES} - \boldsymbol{f}_{j-1/2}^{ES}}{\Delta x} = [\delta \bar{\boldsymbol{x}}]^{-1} D_{1} \bar{\boldsymbol{f}}^{ES} \qquad D_{1} = \begin{bmatrix} -1 & 1 & 0 & \cdots & 0 & 0 \\ 0 & -1 & 1 & 0 & \cdots & 0 \\ 0 & 0 & \ddots & \ddots & 0 & 0 \\ 0 & \cdots & 0 & -1 & 1 & 0 \\ 0 & 0 & \cdots & 0 & -1 & 1 & 0 \end{bmatrix}$$

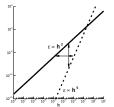
$$f_{j+1/2}^{ES} = \bar{\omega}_{j+1/2}^{L} \bar{\boldsymbol{f}}_{J+1/2}^{L} + \bar{\omega}_{j+1/2}^{C} \bar{\boldsymbol{f}}_{J+1/2}^{C} + \bar{\omega}_{j+1/2}^{R} \bar{\boldsymbol{f}}_{J+1/2}^{R}$$

$$\bar{\boldsymbol{f}}^{L} = I_{fs}^{L} \boldsymbol{f} \; ; \; \bar{\boldsymbol{f}}^{C} = I_{fs}^{C} \boldsymbol{f} \; ; \; \bar{\boldsymbol{f}}^{R} = I_{fs}^{R} \boldsymbol{f}$$

Equating the two forms yields

$$[\delta \bar{x}]^{-1} D_1 \bar{f}^{ES} = [\delta \bar{x}]^{-1} D_1 [\bar{\omega}_t^L I_{fs}^L + \bar{\omega}_t^C I_{fs}^C + \bar{\omega}_t^R I_{fs}^R] f = P^{-1} Q f$$

Solve HUGE Matrix Equation and you're done!



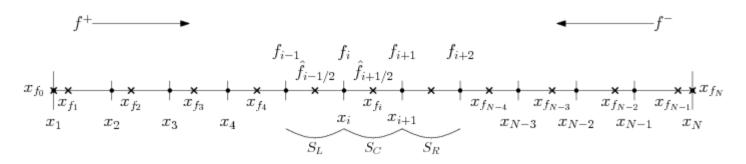
# Deriving SBP(3-4-3) Target Operator $U_t + F_x = 0$



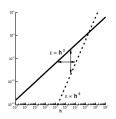
$$U_t + F_x = 0$$

$$[\delta \bar{x}]^{-1} D_1 \bar{f}^{ES} = [\delta \bar{x}]^{-1} D_1 [\bar{\omega}_t^L I_{fs}^L + \bar{\omega}_t^C I_{fs}^C + \bar{\omega}_t^R I_{fs}^R] f = P^{-1} Q f$$

- Mathematica could not find solution for uniform meshes
  - Inconsistent constraints on system
  - Flux mesh must be consistent with P-norm used in proof
- **Equating two matrices (requires??) nonuniform flux** points



$$\bar{x} = \Delta x \left[ 0, \frac{43}{144}, \frac{244}{144}, \frac{349}{144}, \frac{7}{2}, \dots, n-1 - \frac{349}{144}, n-1 - \frac{244}{144}, n-1 - \frac{43}{144}, n-1 \right]$$



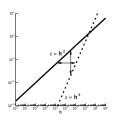
#### Numerical Results: Finite Domain Eigenvalues WENO vs. ESWENO



Eigenvalues for finite domain 4th-order case (3-4-3).

$$U_t + F_x = 0 \; ; \; F = aU$$

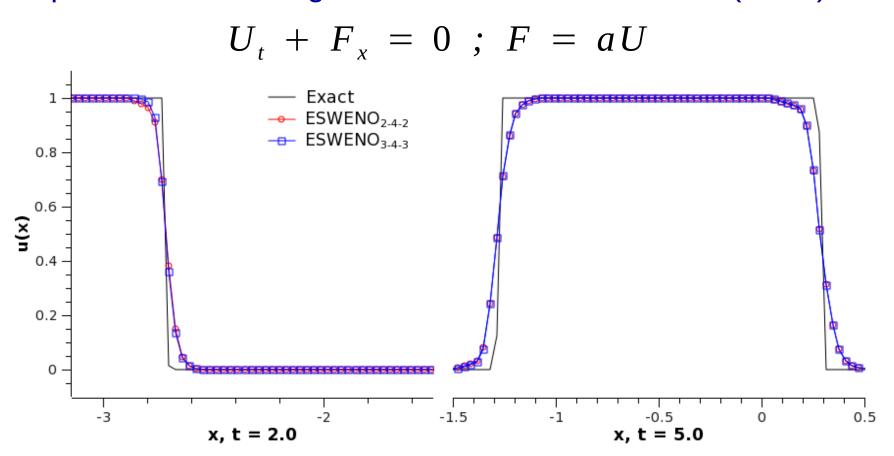
$$0.001 - 0.005 -$$

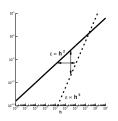


#### Numerical Results: Finite Domain Eigenvalues WENO vs. ESWENO



Square Wave coming into domain. ESWENO 4th:(3-4-3)

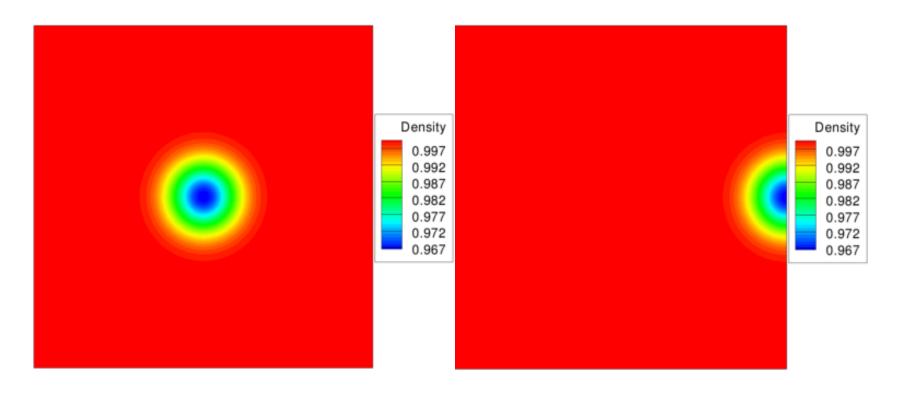


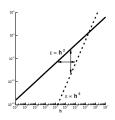


#### Numerical Results: Finite Domain Euler isotropic vortex: ESWENO 4th(3-4-3)



	Linear Block Norm		$ESWENO_{3-4-3}$	
Number of Cells	L <sub>2</sub> Error	$L_2$ Rate	L <sub>2</sub> Error	L <sub>2</sub> Rate
$50 \times 50$	2.49E-05	-	5.32E-05	-
$100 \times 100$	1.64E-06	3.92	2.25E-06	4.57
$200 \times 200$	1.04E-07	3.98	1.08E-07	4.37
$400 \times 400$	6.57E-09	3.99	6.62E-09	4.04

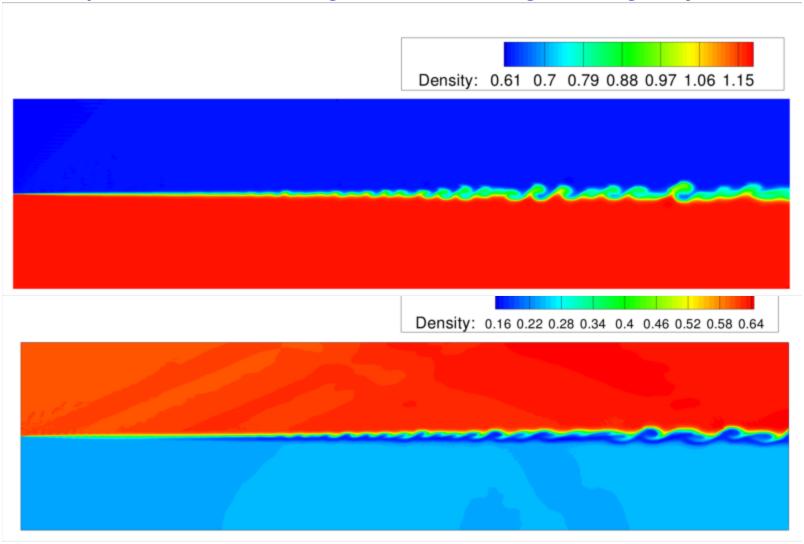


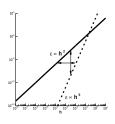


# Numerical Results: Finite Domain 4th-order *ESWENO* for Chemistry



Compare Non-reacting and Reacting Mixing Layers

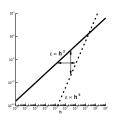




#### **Conclusions**



- The systematic methodology for constructing fourth-order finite domain Energy Stable WENO schemes is developed.
- We prove that for hyperbolic systems, the finite domain ESWENO scheme is stable in the energy norm for both continuous and discontinuous solutions.
- The eigenvalues of the finite domain ESWENO dissipation operator is located in the left-half plane.
- Based on the rigorous truncation error analysis, the new weight functions are developed, which drastically improve the accuracy of the ESWENO scheme and provide excellent shockcapturing capabilities
- Numerical experiments show that the new finite domain ESWENO scheme with the new weights outperform the conventional WENO schemes in terms of accuracy.



### **Questions/Comments**



- Historical Perspective?
- Discontinuous Galerkin?
- ESWENO?