

Properties of Phenylethynyl Imide Composites Fabricated Via VARTM

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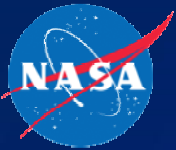
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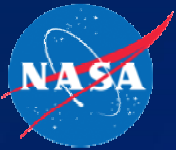
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19th International Conference on Composite Materials
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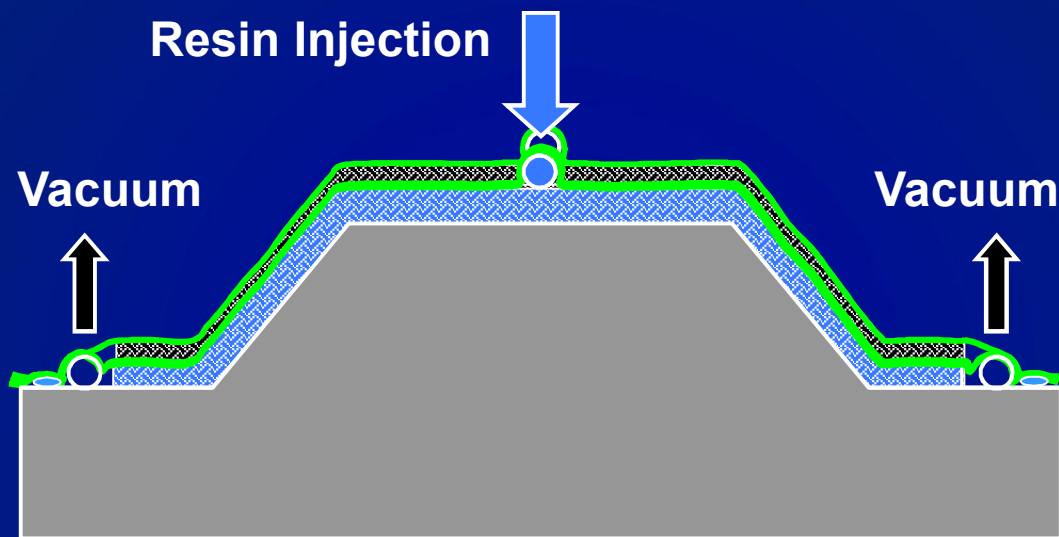


Outline

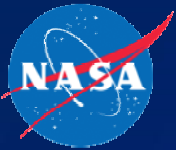
- ❖ Introduction to VARTM Processing
- ❖ Phenylethynyl Terminated Imide (PETI) Resins
- ❖ Past Research and Issues
- ❖ Current HT-VARTM Process
- ❖ Experimental
- ❖ Laminate Fabrication Results
- ❖ Mechanical Properties
- ❖ Structural Sub-Component Demonstration Article
- ❖ Summary



Vacuum Assisted Resin Transfer Molding



- Upper half of metal mold replaced by vacuum bag
- Atmospheric pressure provides both the resin driving force and the preform compaction force
- Distribution medium is used to facilitate resin flow
- Room temperature process



VARTM Characteristics

➤ Strengths

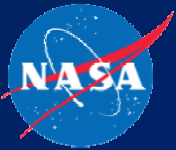
- Reduced cost
- Complex shapes
- Essentially no size limitation

➤ Weaknesses/Possible Issues

- High fiber volume more difficult to obtain than in autoclave
- Void content $< 2\%$ for aerospace applications (like epoxies)?
- Most applications tolerate higher void content than aerospace
- Applicable to high temperature resins?

➤ Controlled Atmospheric Pressure Resin Infusion (CAPRI)

- Patented by The Boeing Company
- SCRIMP variation
- vacuum debulking and reduced pressure difference
- minimizes thickness gradients and resin bleeding

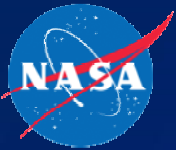


Selection of PETI Resins for HT-VARTM

- Precursor form offers
 - Low melt viscosity, large processing window
 - Long shelf-life
 - No solvent, no unreacted toxic monomers
 - Commercial availability in prepreg solutions and neat resin form (for RTM and RI)

- Cured resin offers
 - High temperature performance (>1000 h at 300 ° C in air)
 - Composite mechanical properties including damage tolerance meet aerospace criteria

- Disadvantages
 - High temperature cure (typically 1-3 h at 371 ° C)
 - High cost (relative to epoxies)
 - Niche market (jet engine applications, recent interest in airframe structure)



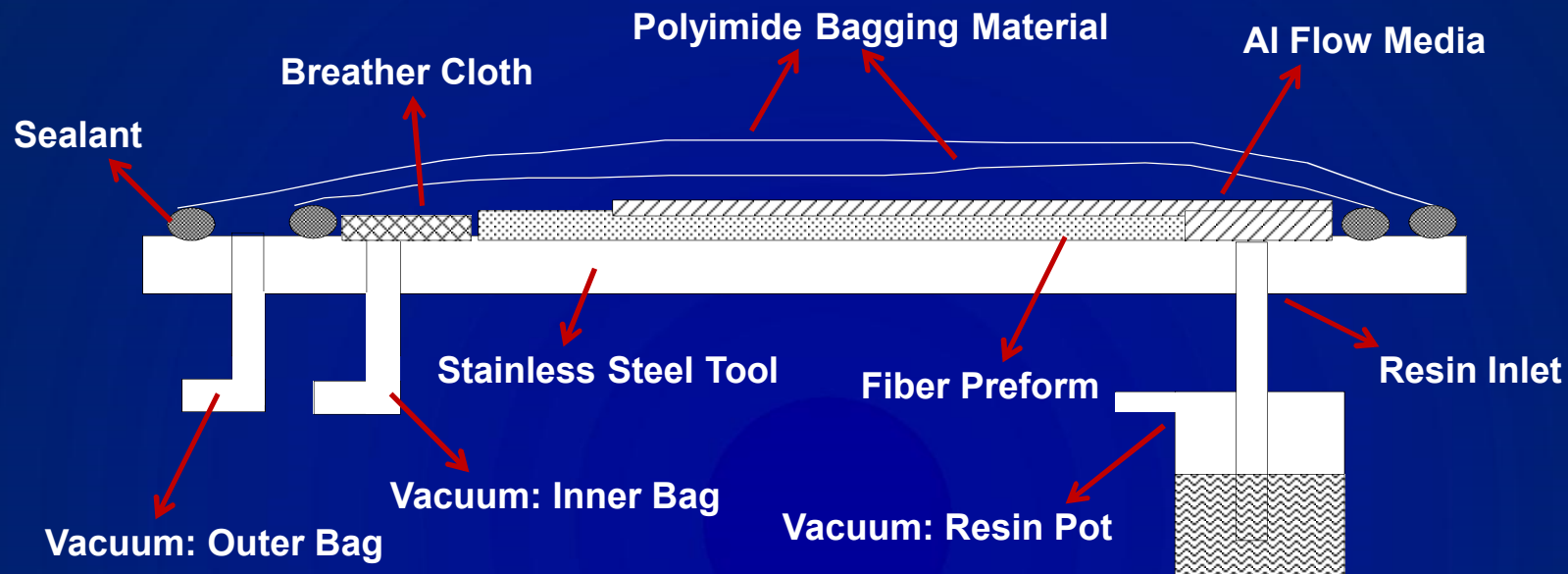
State of the Art

- PETI laminates from same resins (different mol wt) by autoclave (prepreg), RI and RTM, porosity levels of $< 2\%$ routinely achieved
- Difference between RTM and VARTM = pressure (1.4 vs. 0.1 Mpa)
- HT-VARTM laminate porosity by volume reduced from $\sim 10\%$ to as low as 1% by identification of porosity sources and processing modifications.

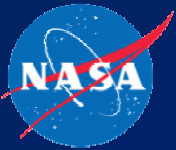
Ref: S. Ghose, et al.; "High Temperature VARTM of Phenylethynyl Terminated Imides," High Performance Polymers, 21 (5), 653 (2009)



Experimental



- Steel plate with 3 holes to provide one resin inlet and two vacuum outlets
- Polyimide bagging material (Thermalimide™)
- Inner bag with several layers of carbon fiber preform, 5 layers of Aluminum (Al) screen distribution media (DM), Release Ease™ fabric - a breather material
- High temperature sealant to seal both an inner bag and an outer bag (for redundancy should a leak occur in the inner bag after infiltration)



Experimental

- Resins: PETI-9: cured $T_g \sim 265^\circ \text{C}$ (Imitec, Inc.),
 PETI-330: cured $T_g \sim 330^\circ \text{C}$ (Ube America, Inc.)

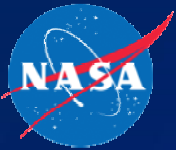
- Fabrics: T650-35-3K-8HS
 IM7-6K-5HS
 IM7-6K uniweave (unidirectional weave)

- Rheology

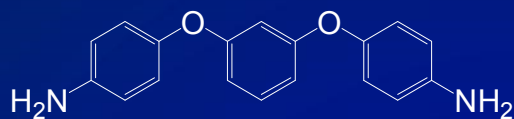
- C-scan

- Acid digestion

- Mechanical properties (unaged and aged, room temperature and elevated temperature)
 - Short Block Compression (SBC)
 - Open Hole Compression (OHC)
 - Compression After Impact (CAI)



LaRC™ PETI-330



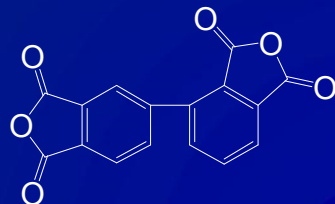
**1,3-bis(4-aminophenoxy)benzene
(1,3,4-APB)**

+



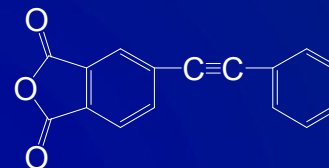
+

1,3-diaminobenzene (m-PD)



**asymmetric biphenyl
dianhydride (a-BPDA)**

+

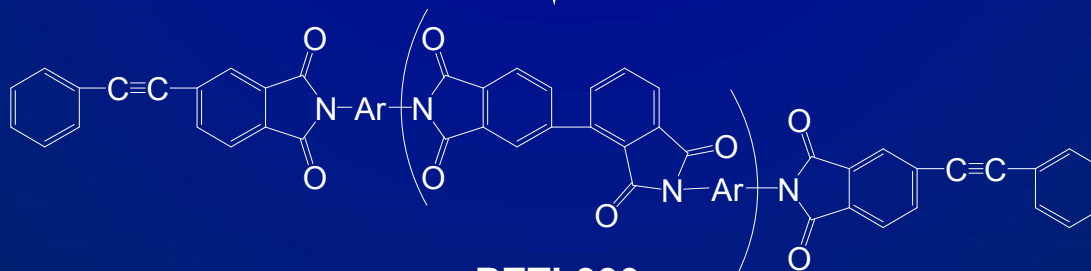


**phenylethynyl phthalic
anhydride (PEPA)**

nitrogen
NMP

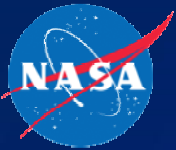
POLYAMIDE ACID

-H₂O

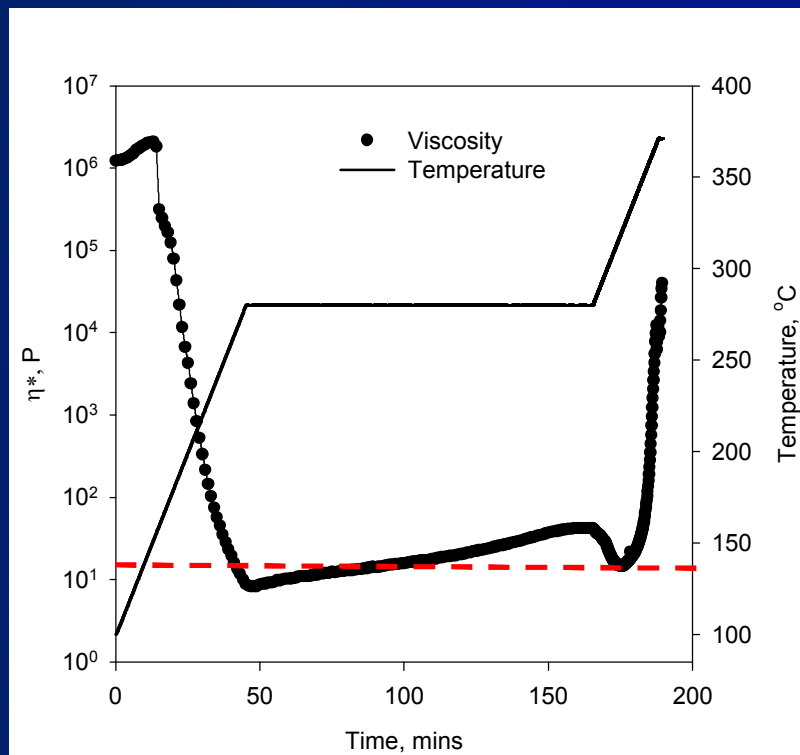


PETI-330

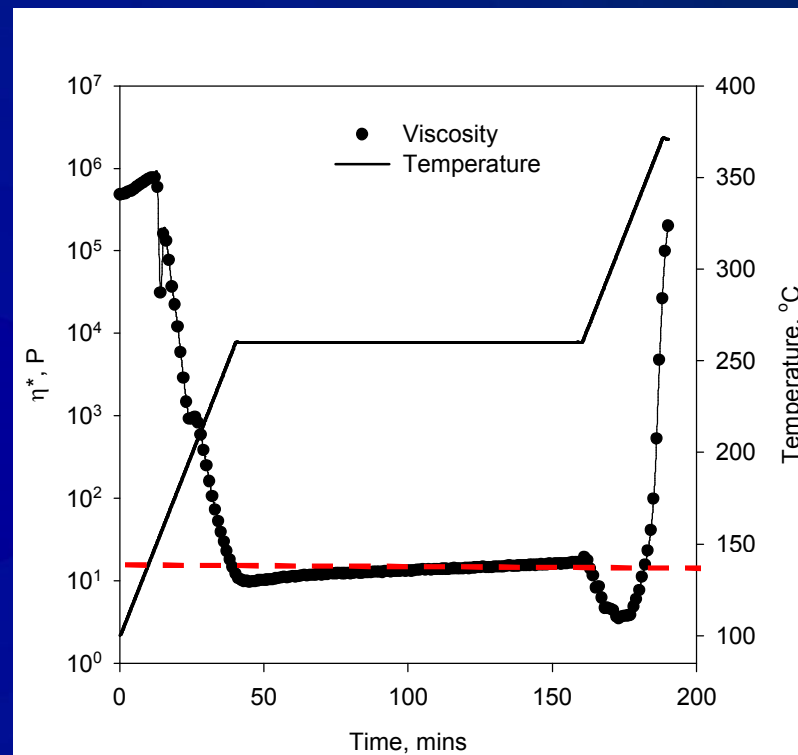
Where Ar = 1,3,4-APB and m-PD; 50:50 ratio, M_n ~1290 g/mol



Rheology Profile of PETI-330



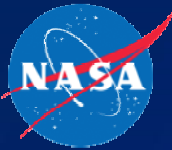
2 h hold at 280 °C



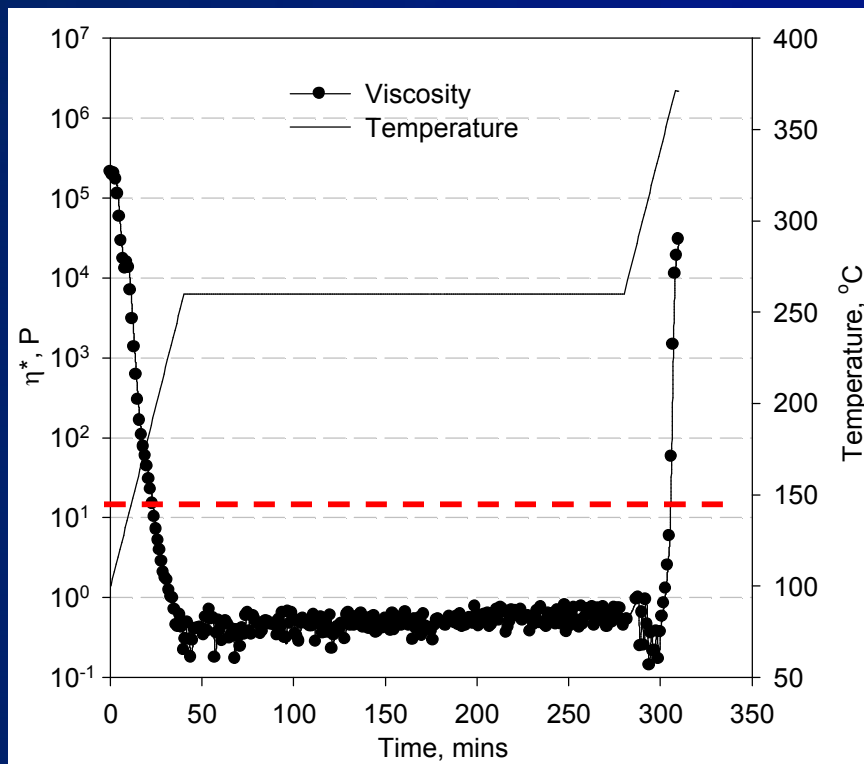
2 h hold at 260 °C

Larger processing window at 260 °C for PETI-330

Rheology conditions: 100 – 371 °C at 4 °C/min; 2 h hold at infusion temperature, frequency of 10 rad/sec, autostrain

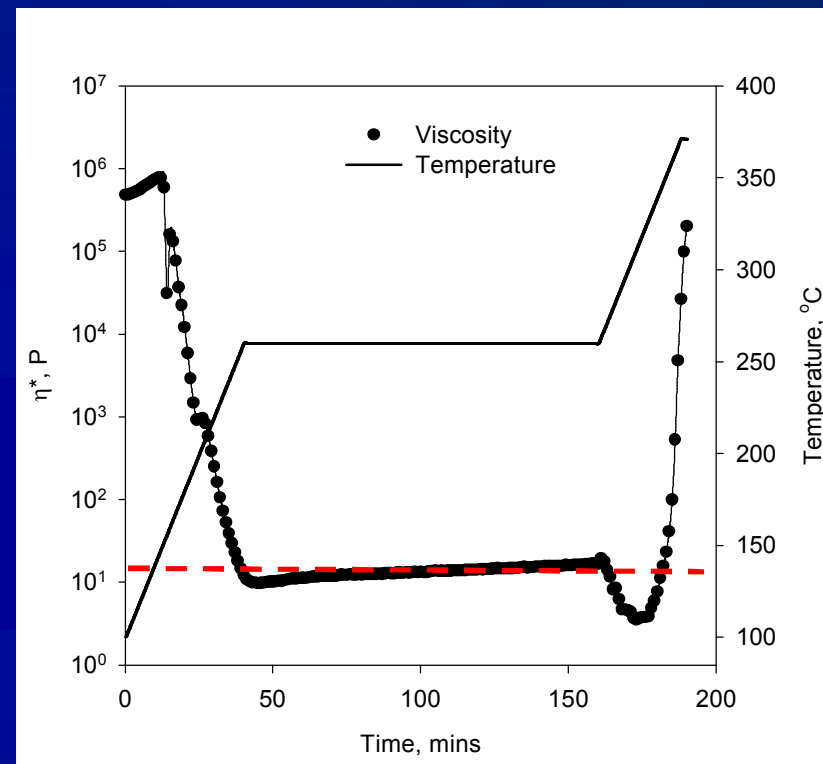


Rheology Profile of PETI-9 / PETI-300



PETI-9

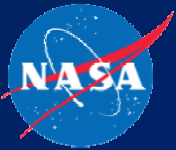
4 h hold at 260 °C



PETI-330

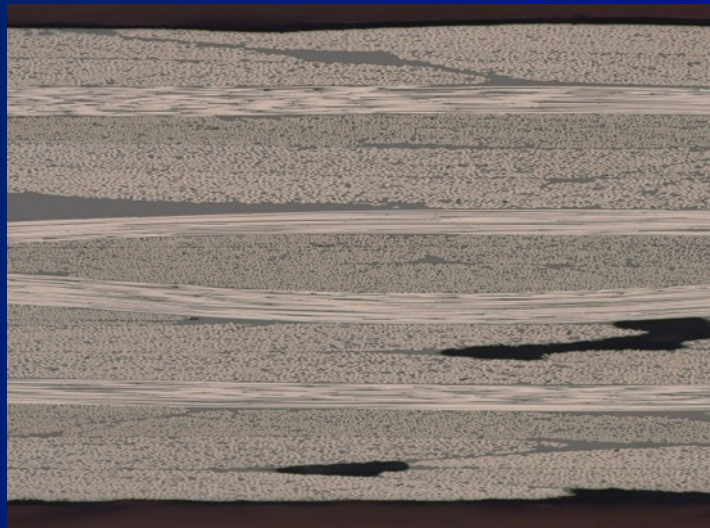
2 h hold at 260 °C

Rheology conditions: 100 – 371 °C at 4 °C/min; 4 h hold at infusion temperature, frequency of 10 rad/sec, autostrain



Typical PETI Panels Quality

PETI-9/T650

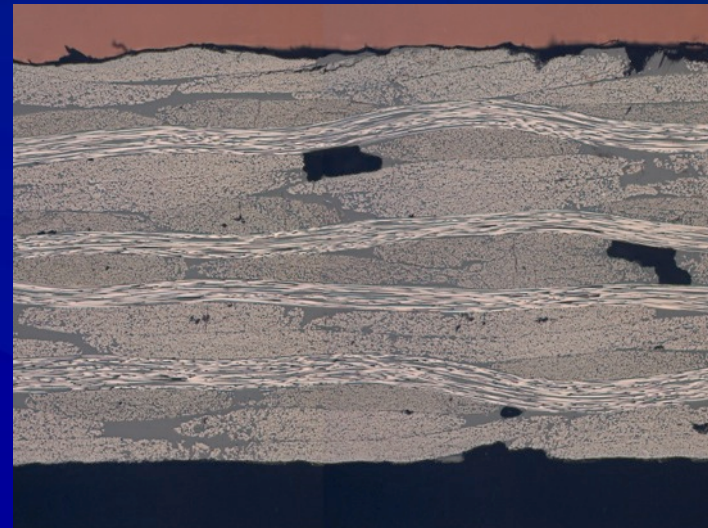


Void – 3.6 %,
Fiber volume – 63 %

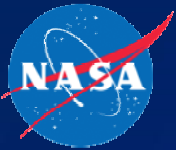
Quasi lay-up: OHC $[\pm 45/(0/90)/\pm 45/(0/90)]_s$ (Biaxial Fabric)
OHC $[-45/0/+45/90]_{2s}$ (Uniweave Fabric)
SBC, CAI $[\pm 45/(0/90)/\pm 45/(0/90)]_{2s}$ (Biaxial Fabric)

- ❖ Heat treatment of C-fibers at 400 ° C
- ❖ Infusion at 260 ° C
- ❖ Staged cure cycle

PETI-330/IM7



Void – 2.9 %,
Fiber volume – 57 %



Fiber and Void Volume Contents of PETI Laminates

Test	Material: Fabric / Resin					
	IM7-5HS/ PETI-9	T650-8HS/ PETI-9	IM7-Uni/ PETI-9	IM7-5HS/ PETI-330	T650-8HS/ PETI-330	IM7-Uni/ PETI-330
SBC RT	57.2 / 3.9	59.6 / 4.4	--	53.4 / 3.9	58.3 / 4.5	--
SBC ET	57.2 / 3.9	59.6 / 4.4	--	53.4 / 3.9	58.3 / 4.5	--
SBC Aged	57.2 / 3.9	59.6 / 4.4	--	53.4 / 3.9	58.3 / 4.5	--
OHC RT	58.7 / 2.7	63.7 / 3.6	57.4 / 2.7	57.5 / 2.9	62.4 / 3.2	56.5 / 1.7
OHC ET	57.6 / 2.4	60.0 / 3.7	57.5 / 3.1	53.5 / 4.2	63.1 / 3.4	53.5 / 3.8
OHC Aged	58.2 / 2.6	57.5 / 2.9	61.9 / 3.7	55.7 / 3.6	62.4 / 3.2	55.7 / 1.7
CAI RT	56.8 / 2.9	61.7 / 4.2	--	55.9 / 2.9	60.1 / 3.8	--
CAI ET	56.8 / 2.9	60.8 / 4.2	--	55.9 / 2.9	60.1 / 3.8	--
Fiber Volume % / Void Volume %						

PETI-9 Aged for 1000 hr at 177° C

PETI-330 Aged for 1000 hr at 288° C



TGA/MS: Source of Volatiles

Heating Ramp

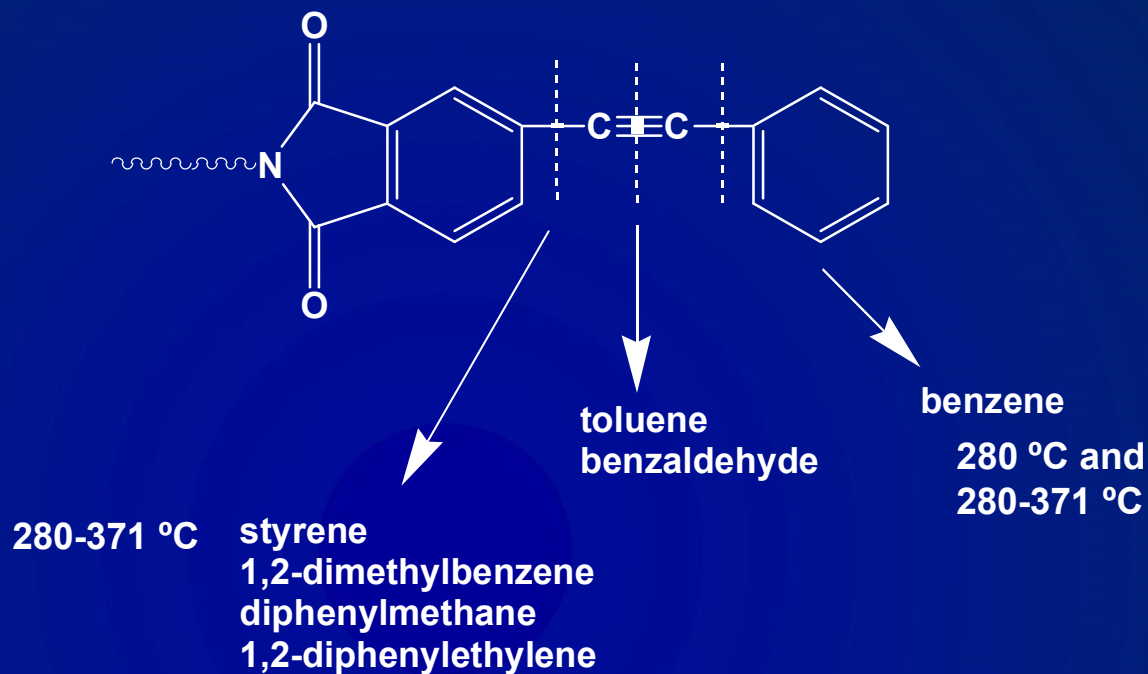
Heat to 280 °C

Hold 1 h

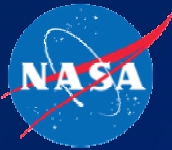
Heat to 371 °C

Rate 10 °C /min

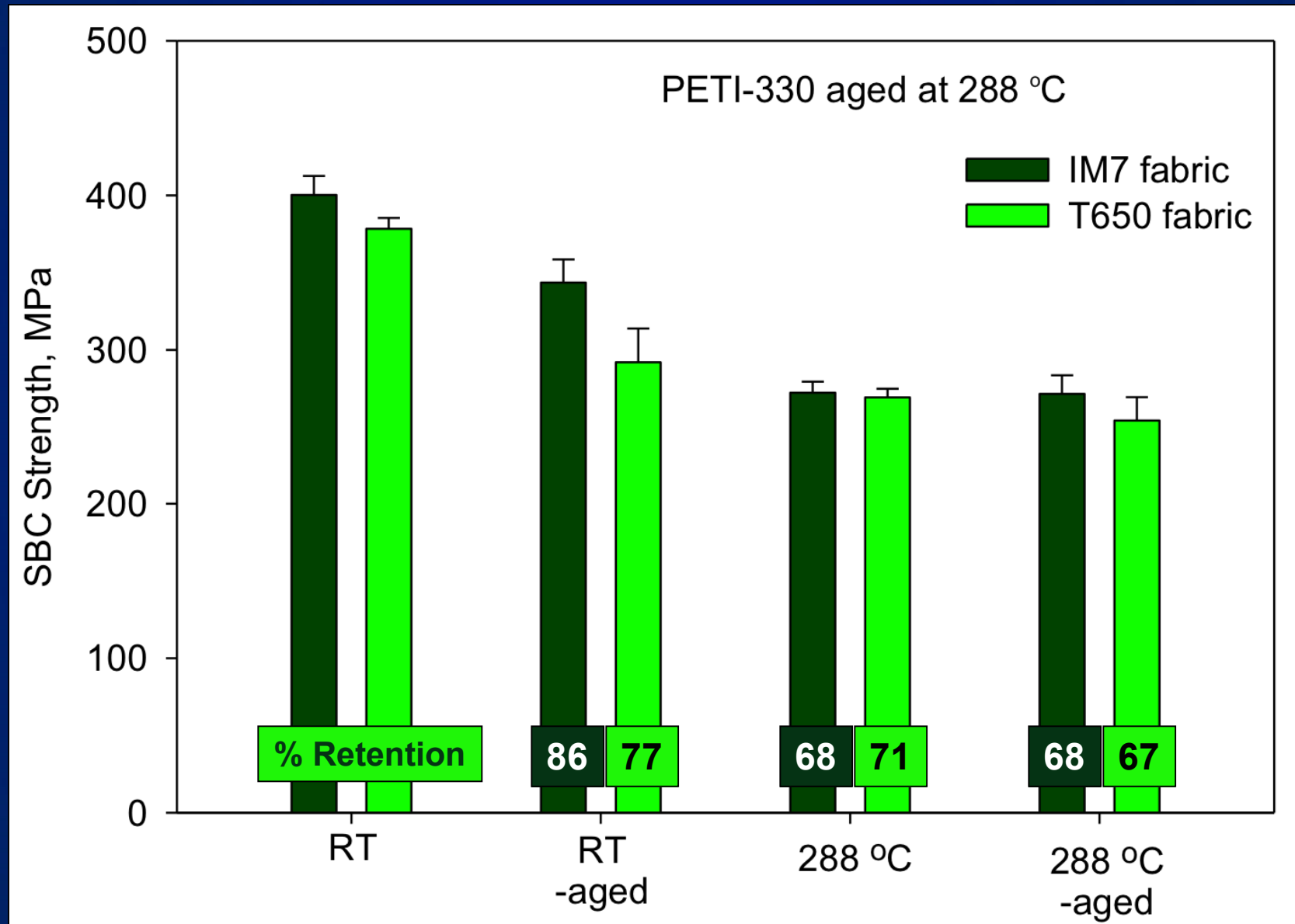
Atmosphere N₂

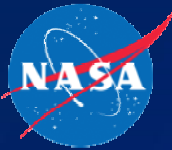


- Heating ramp designed to simulate laminate fabrication cycle (N₂ versus vacuum)
- Benzene, toluene and benzaldehyde detected during both heating ramps
- Other species listed evolved only during second heating ramp
- Total weight loss was 0.5-1.0 %

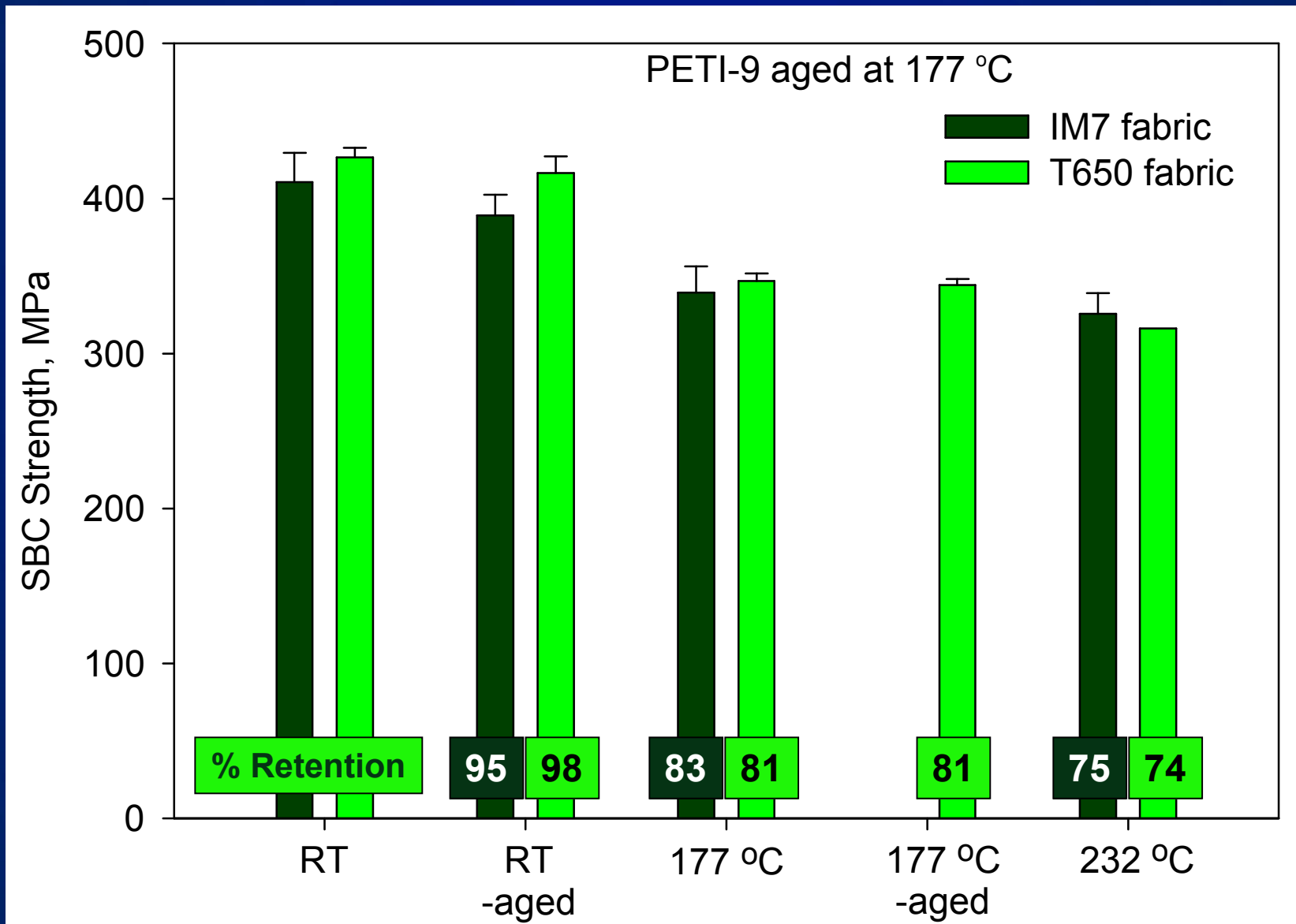


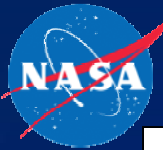
Short Block Compression Properties of PETI-330



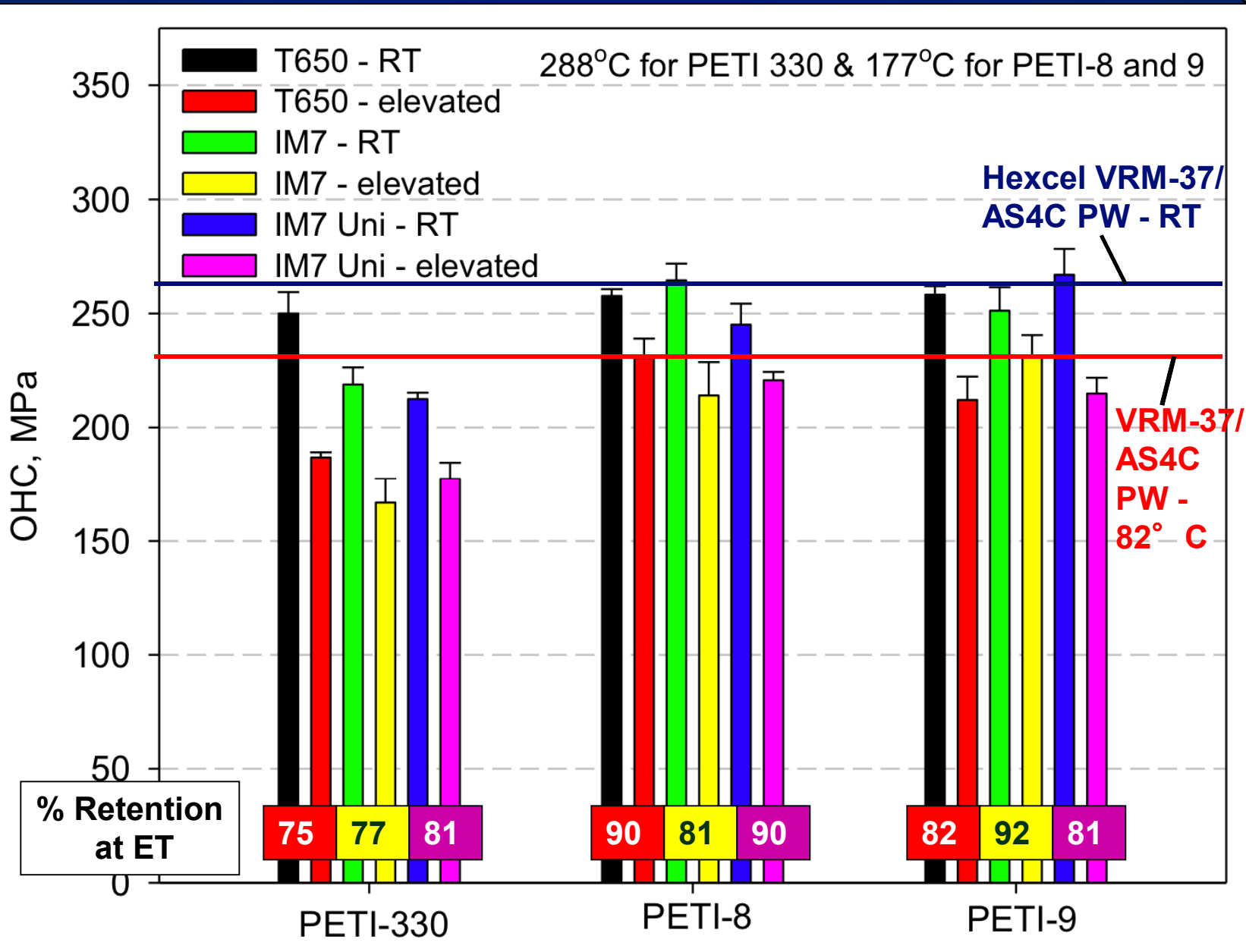


Short Block Compression Properties of PETI-9

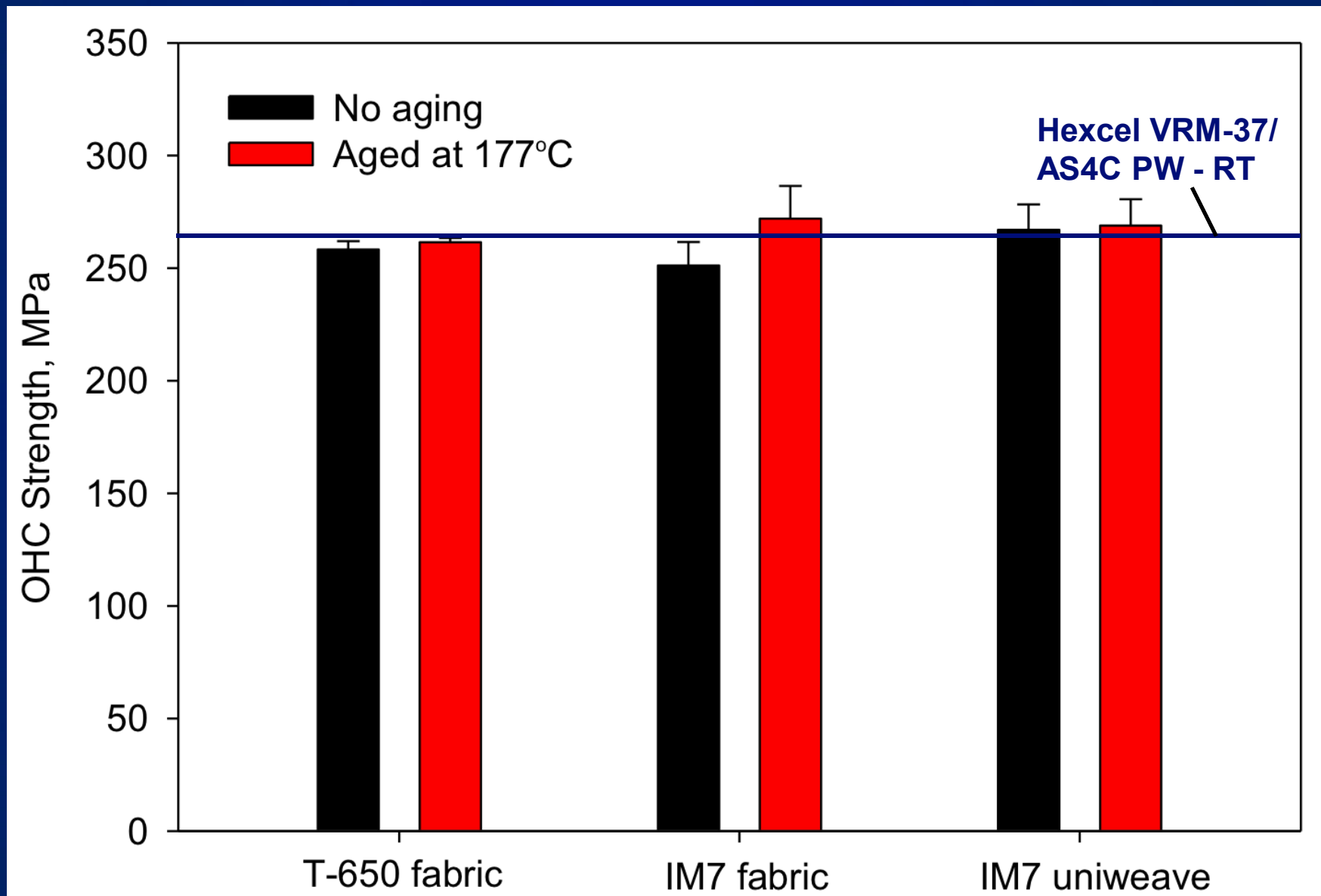


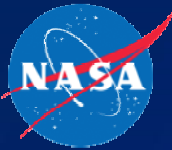


Open Hole Compression Properties

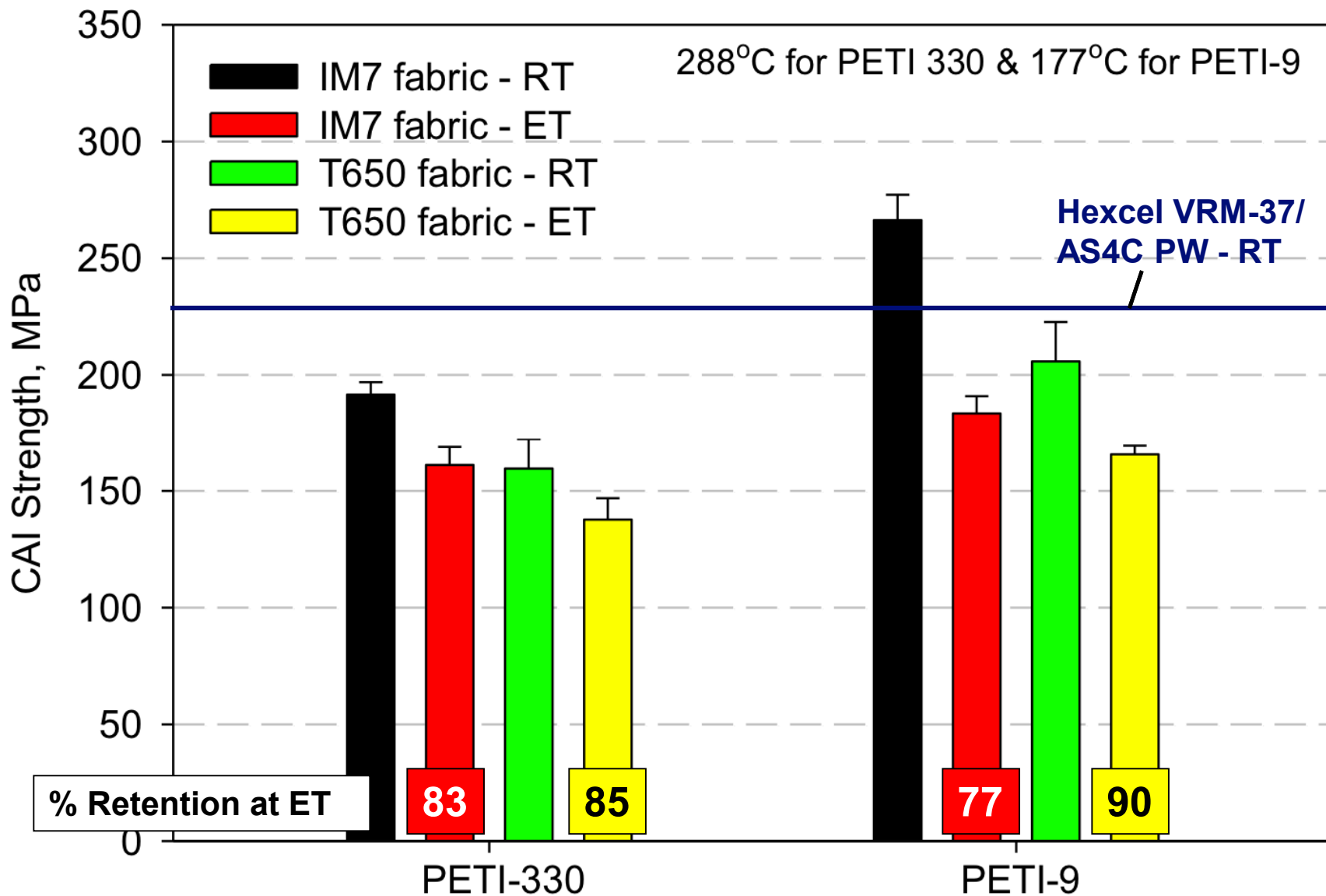


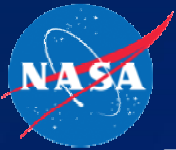
OHC Properties of PETI-9 Aging Effects





Compression After Impact Properties



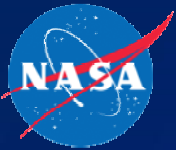


Compression After Impact

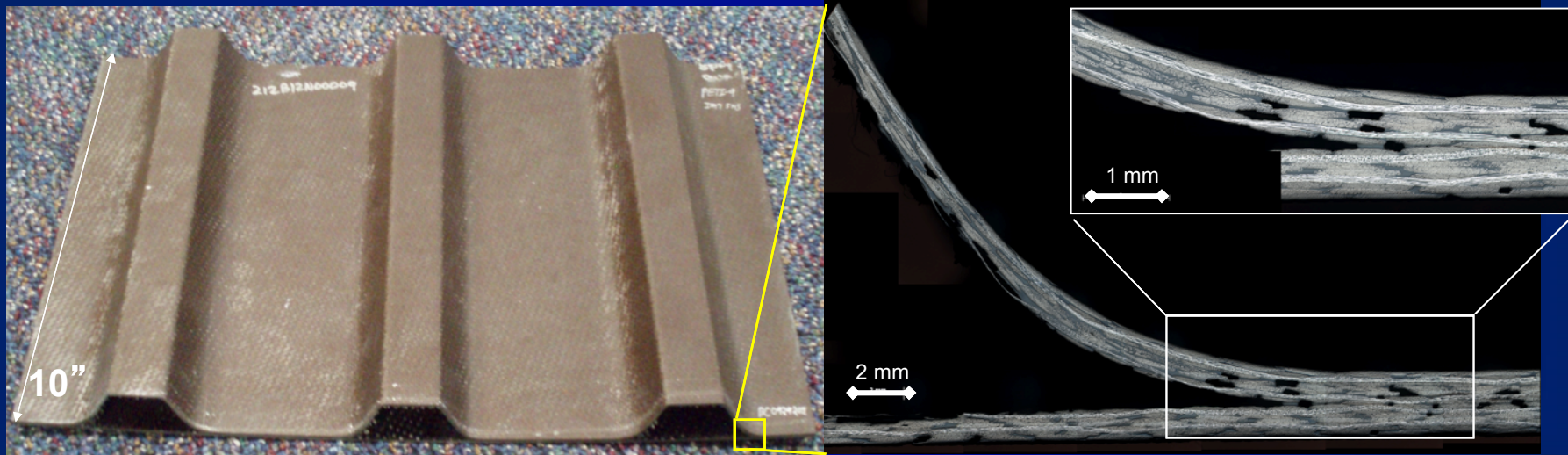
Test condition: dropped weight of 5.3691 kg and an indenter with a 1.59 cm diameter hemi-spherical tip

Material	CAI Test Temp., °C	Impact Energy, J	Drop Height, cm	Max. Force, N	Dent Depth, mm	Damage Area*, cm ²
PETI-330 / IM7 5HS	RT	399.54	63.24	25.41	0.62	3.16
PETI-330 / IM7 5HS	288	385.97	61.09	25.20	0.51	4.23
PETI-330 / T650 8HS	RT	448.04	70.91	28.26	0.96	8.70
PETI-330 / T650 8HS	288	460.61	72.90	27.51	0.45	8.56
PETI-9 / IM7 5HS	RT	375.98	59.51	25.38	0.45	1.92
PETI-9 / IM7 5HS	177	384.04	60.78	25.20	0.58	2.38
PETI-9 / T650 8HS	RT	447.82	70.88	27.63	1.05	3.90
PETI-9 / T650 8HS	177	451.00	71.38	25.94	1.06	3.14

*approximated from C-scan images post impact



Structural Sub-Component: IM7/PETI-9 HT-VARTM Hat Stiffened Panel

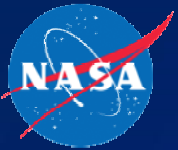


**3.6% Void Volume Fraction/
54% Fiber Volume Fraction**



Conclusions

- LaRC developed polyimide resins with appropriate viscosity/ processing windows have been developed for HT-VARTM and HT-RTM
- With suitable hardware, infusion temperature, and cure cycle adjustments, LaRC™ PETI-330 and PETI-9 can be successfully processed by HT-VARTM
- HT-VARTM void contents of 1.3 to 4% are the lowest reported
- A database of compressive, OHC and CAI properties was developed on two resins and three types of carbon fabrics
- HT-VARTM composites demonstrated 67-100% retention of properties at elevated temperature (177 ° C for PETI-9 and 288 ° C for PETI-330) as well as after aging for 1000 h at those same temperatures
- Fabrication of a prototype part demonstrated that HT-VARTM has the potential to be a viable out-of-autoclave process for these polyimide systems



Acknowledgments



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- Janice Y. Smith, Kathleen Devol, Christopher W. Wright, and Gregory T. Shanks of NASA LaRC for mechanical properties

