

National Aeronautics and Space Administration

Optimization of the Lunar IceCube Trajectory Using Stochastic Global Search and Multi-Point Shooting



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code 595

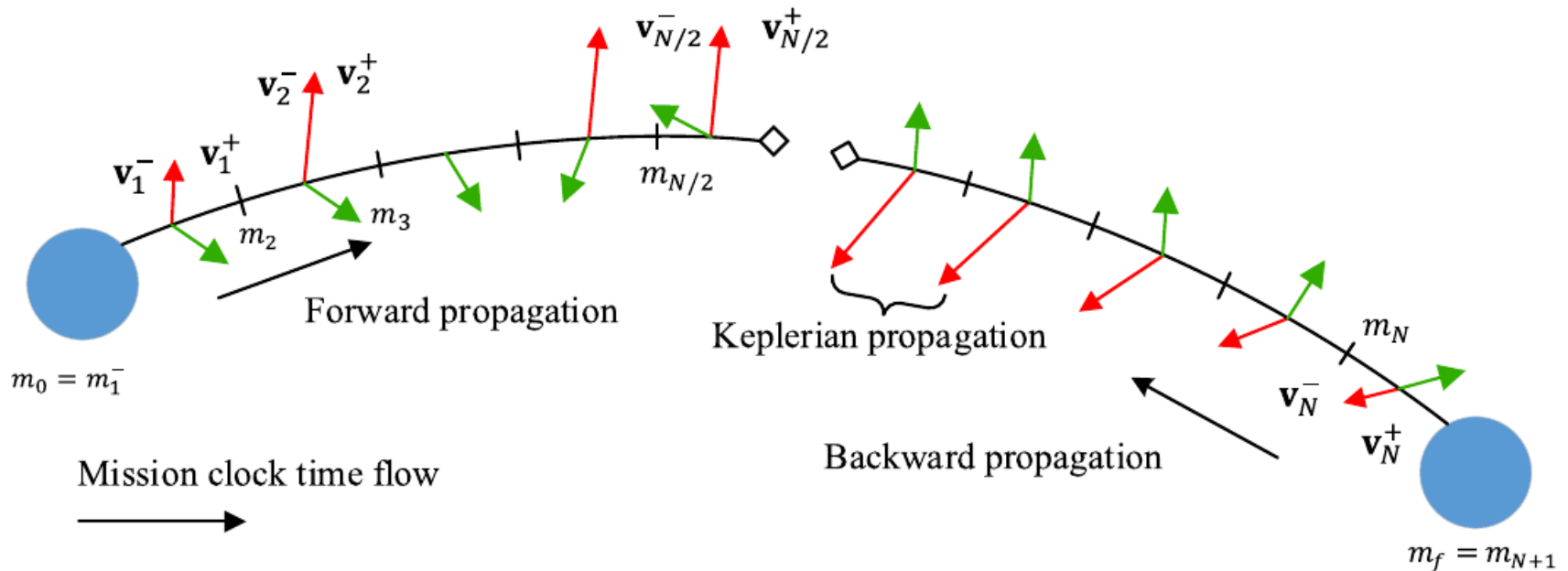
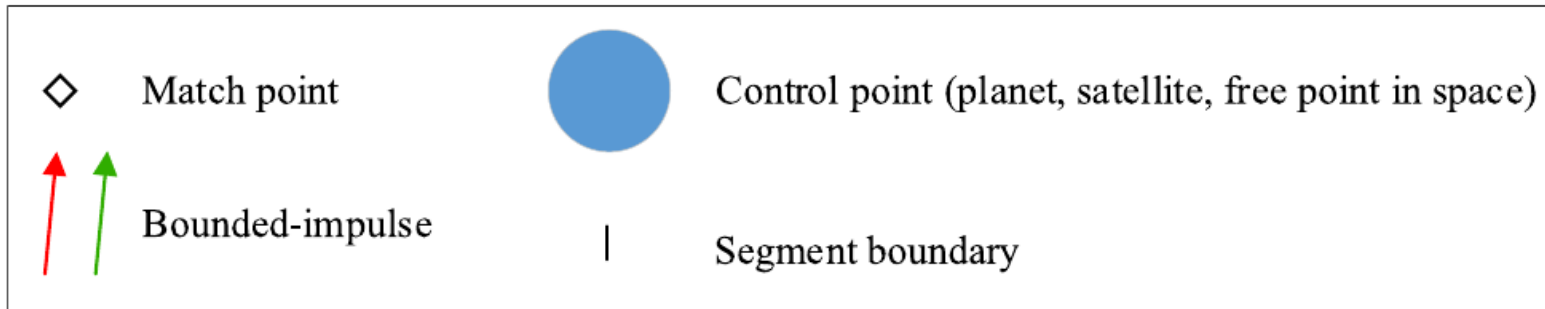


Agenda

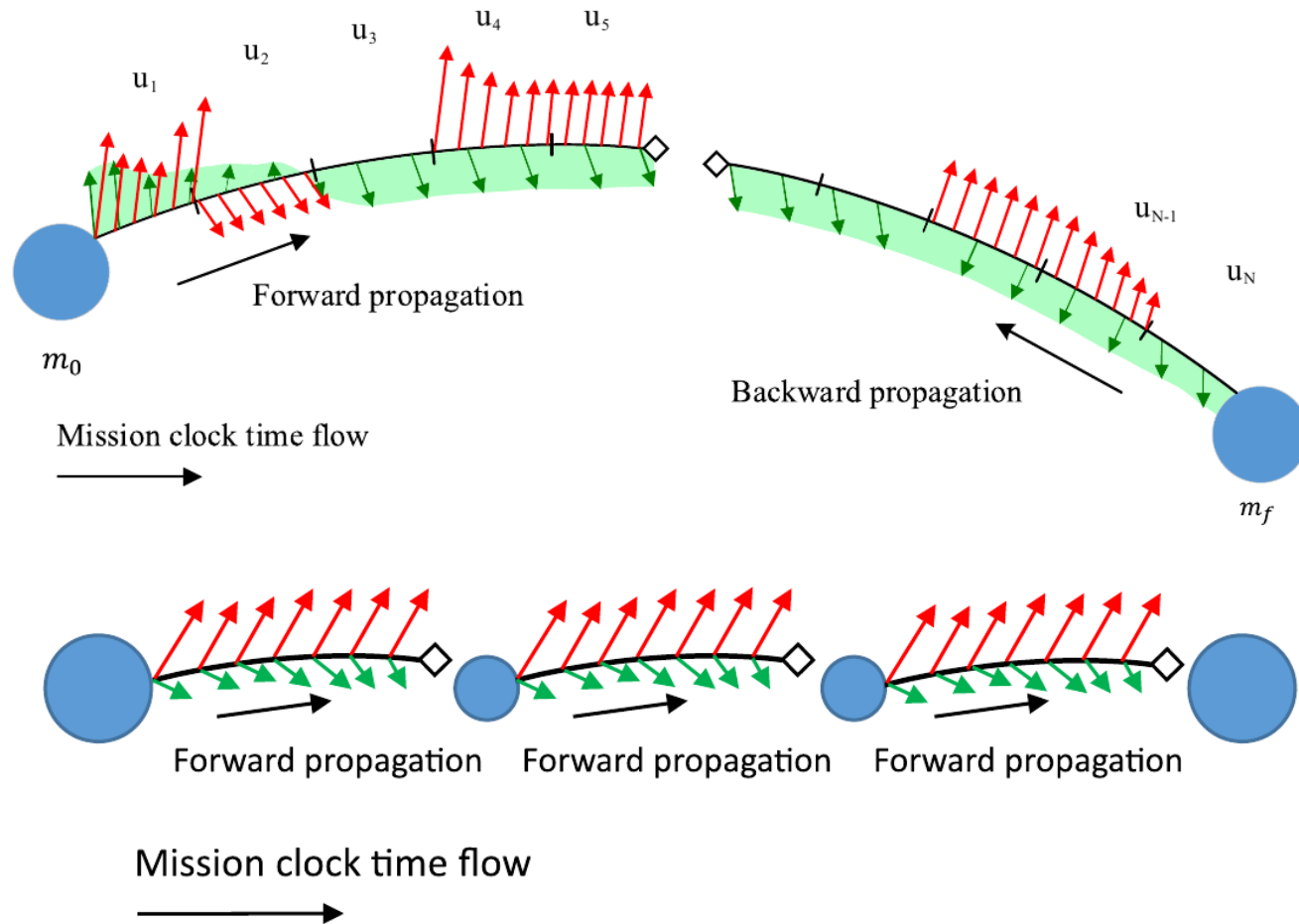
- Physical Model
- Optimization and Global Search
- ConOps and Problem Structure
- Examples

Physical Model

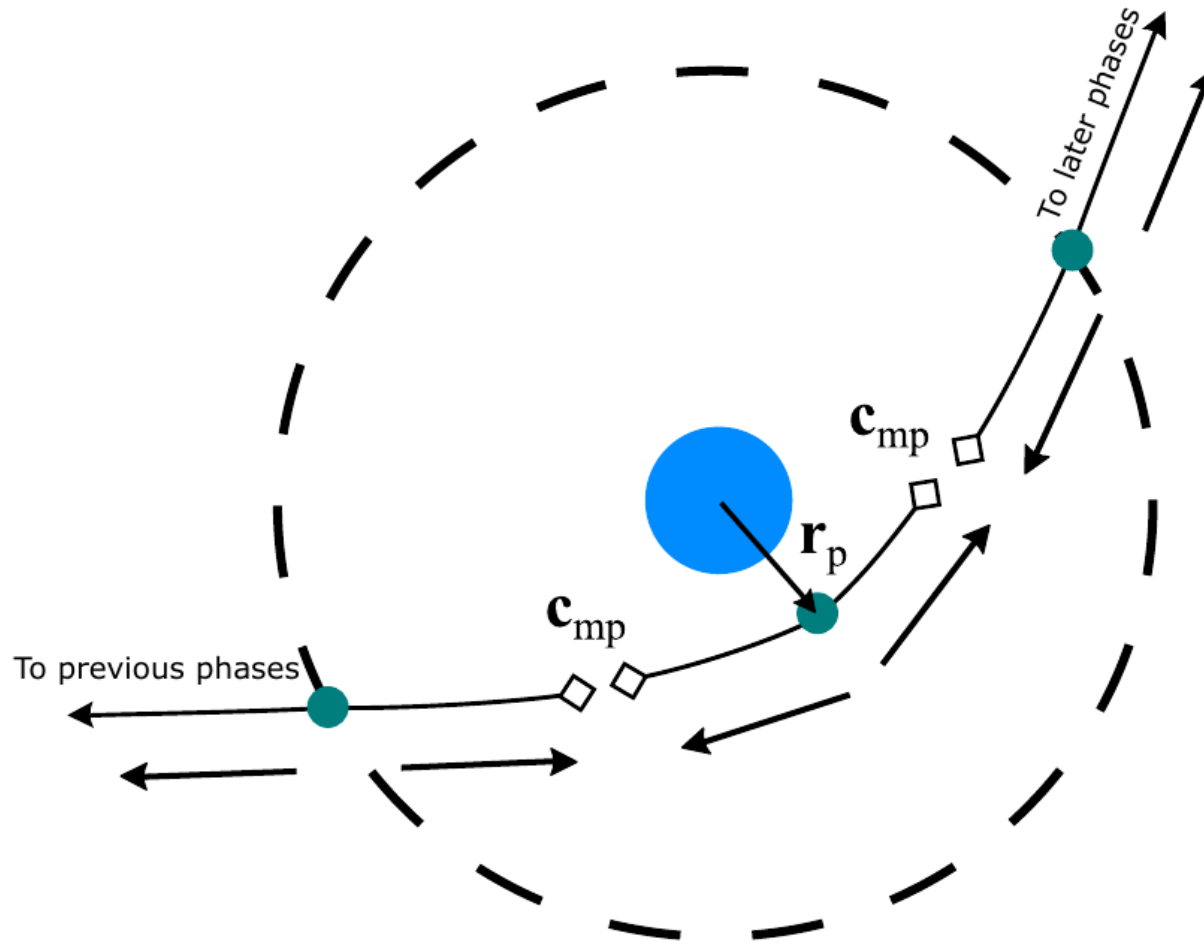
Low-Fidelity Analysis: Sims-Flanagan with Averaged n -Body Gravity



High-Fidelity Optimization via Direct Two-Point or Parallel Shooting



Lunar Gravity Assist



Optimization and Global Search

Inner-Loop Solver: Nonlinear Programming (NLP)

Minimize $f(\mathbf{x})$

Subject to:

$$\mathbf{x}_{lb} \leq \mathbf{x} \leq \mathbf{x}_{ub}$$

$$\mathbf{c}(\mathbf{x}) \leq \mathbf{0}$$

$$\mathbf{A}\mathbf{x} \leq \mathbf{0}$$

where:

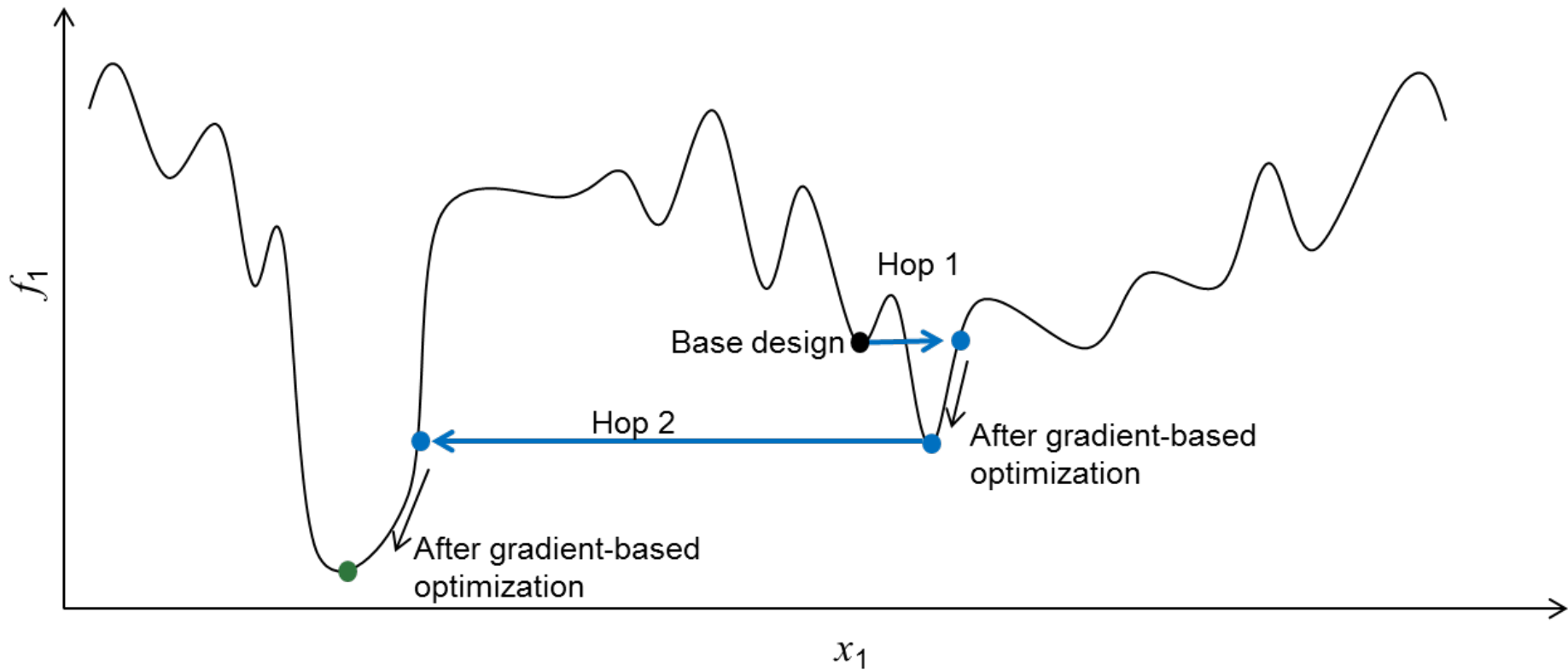
$\mathbf{x}_{lb}, \mathbf{x}_{ub}$ are lower and upper bounds on the decision variables

$\mathbf{c}(\mathbf{x})$ is a vector of nonlinear constraints

$\mathbf{A}\mathbf{x}$ is a vector of linear constraints

- We use the Sparse Nonlinear OPTimizer (SNOPT)
- Analytical expressions are provided for every element of the Jacobian
- But all NLP problem solvers require an initial guess...

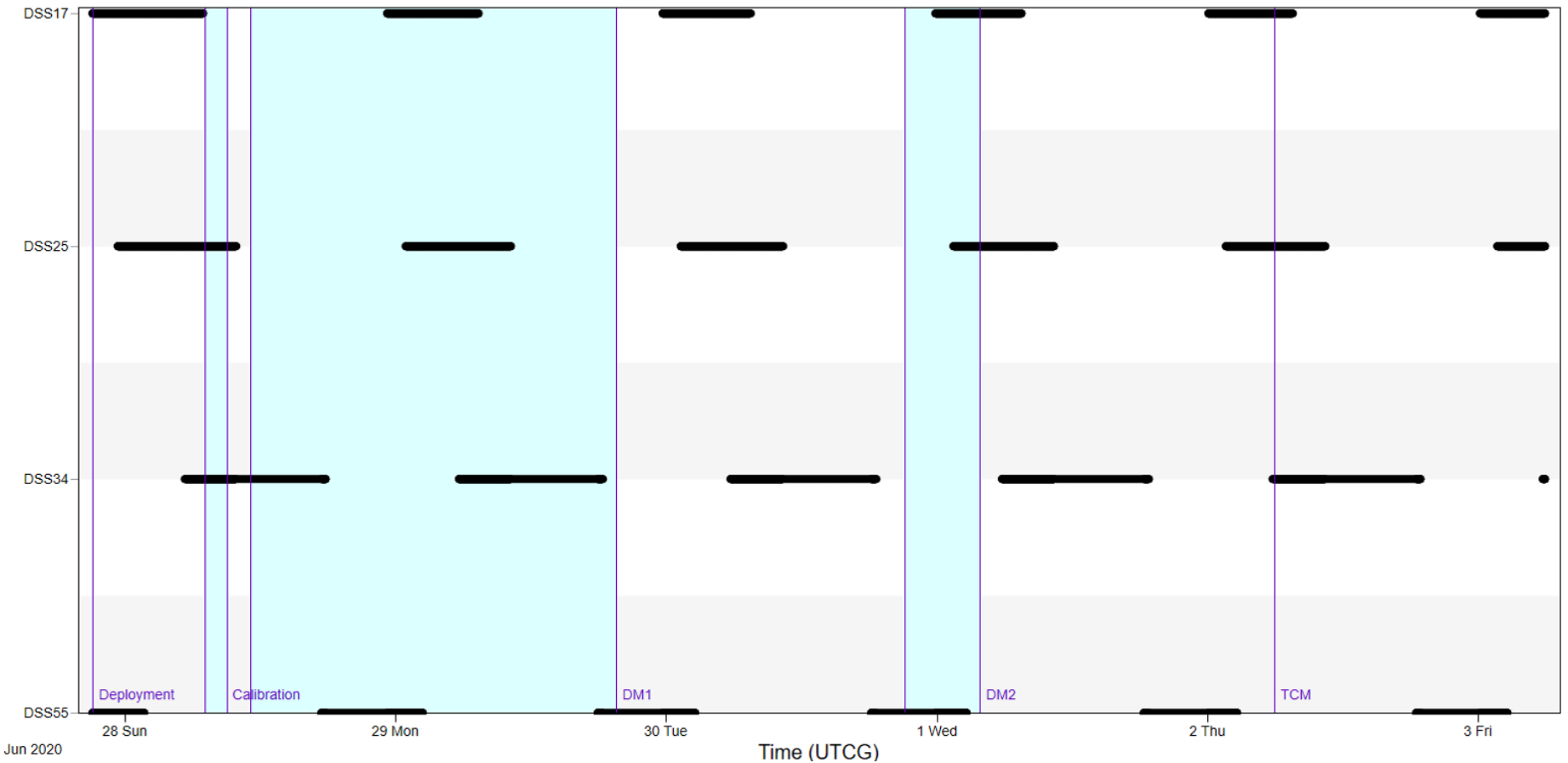
Inner-Loop Solver: Monotonic Basin Hopping (MBH)



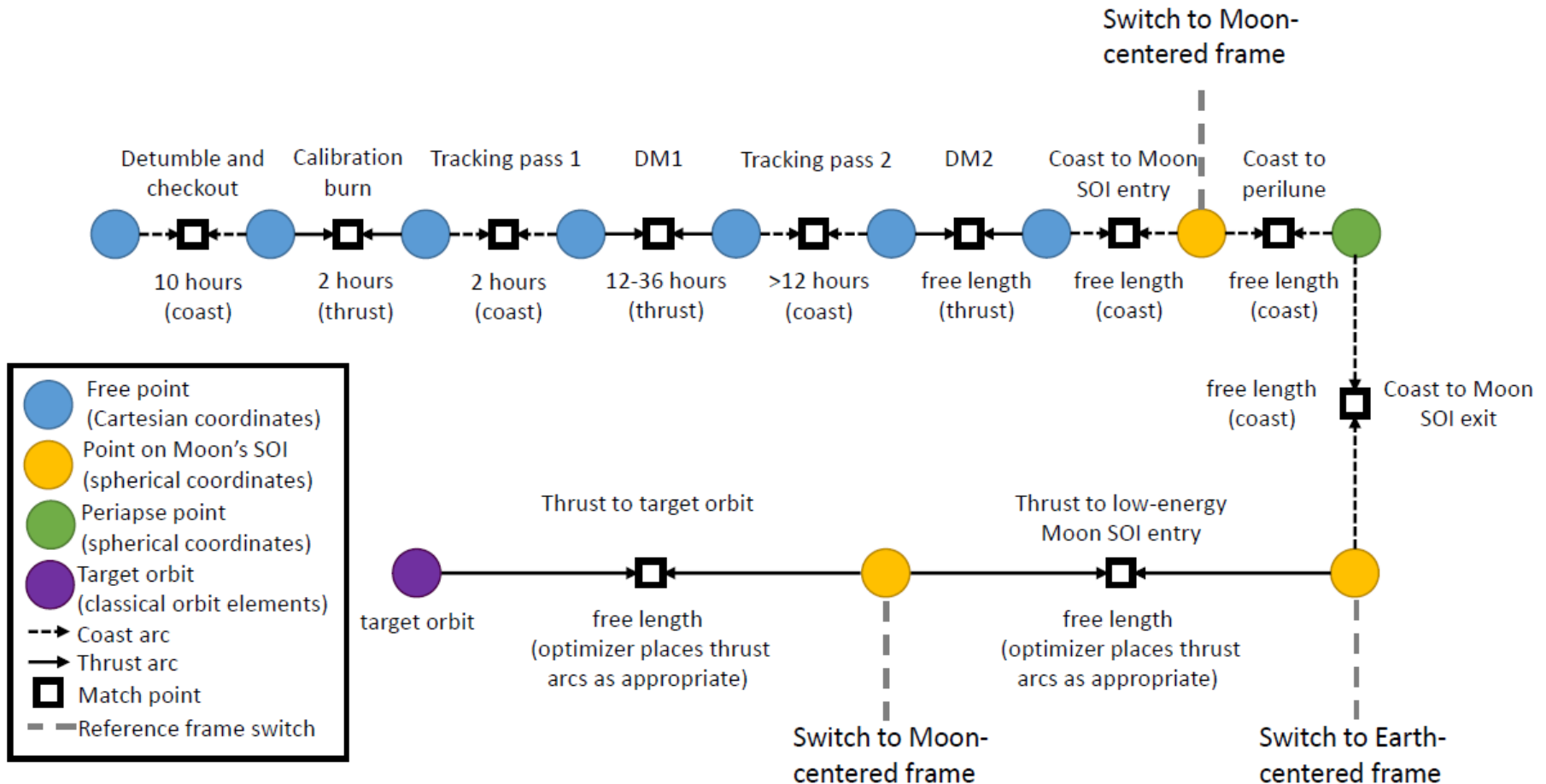
ConOps and Problem Structure

Deployment to LGA

Measurement Times by Tracker

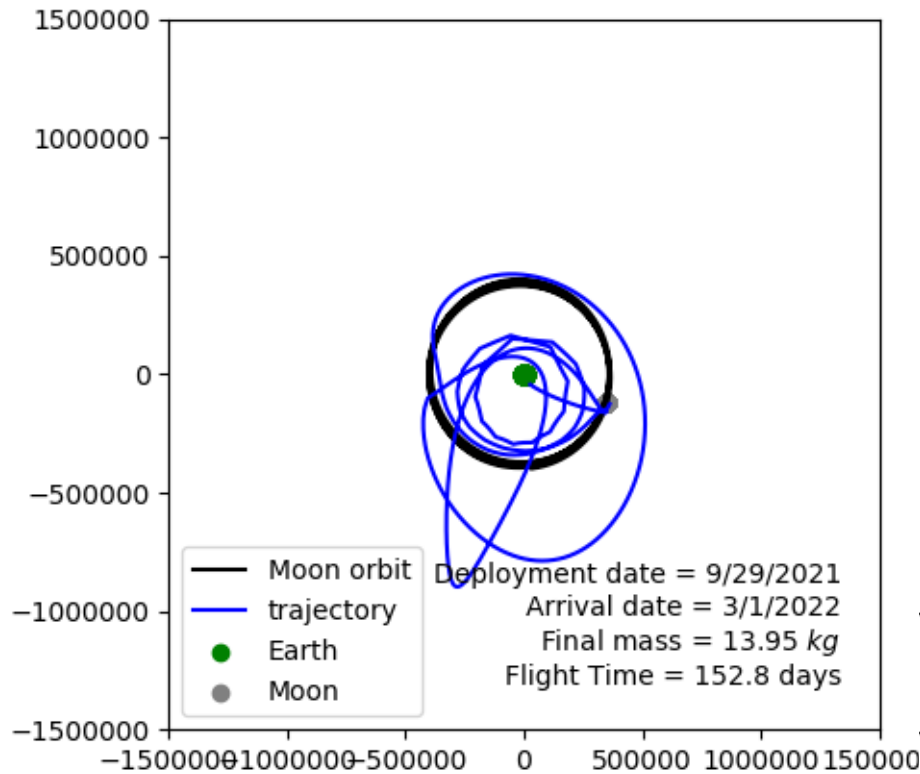


Lunar IceCube as an Optimization Problem



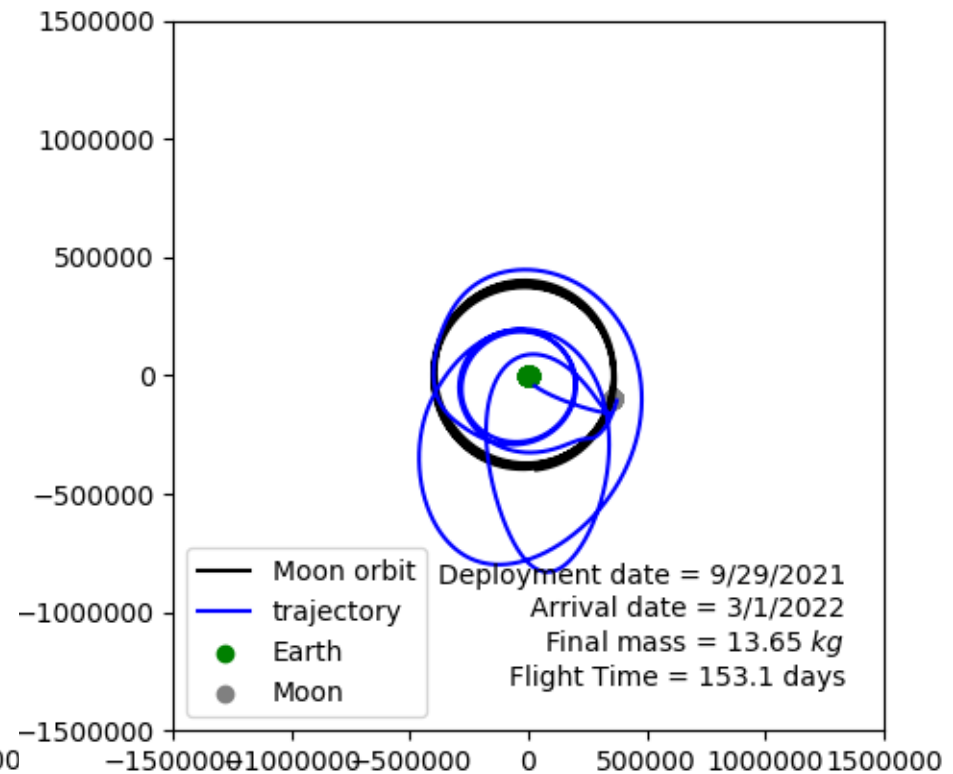
Examples

Low-Fidelity vs High-Fidelity Point Solution



Low-Fidelity

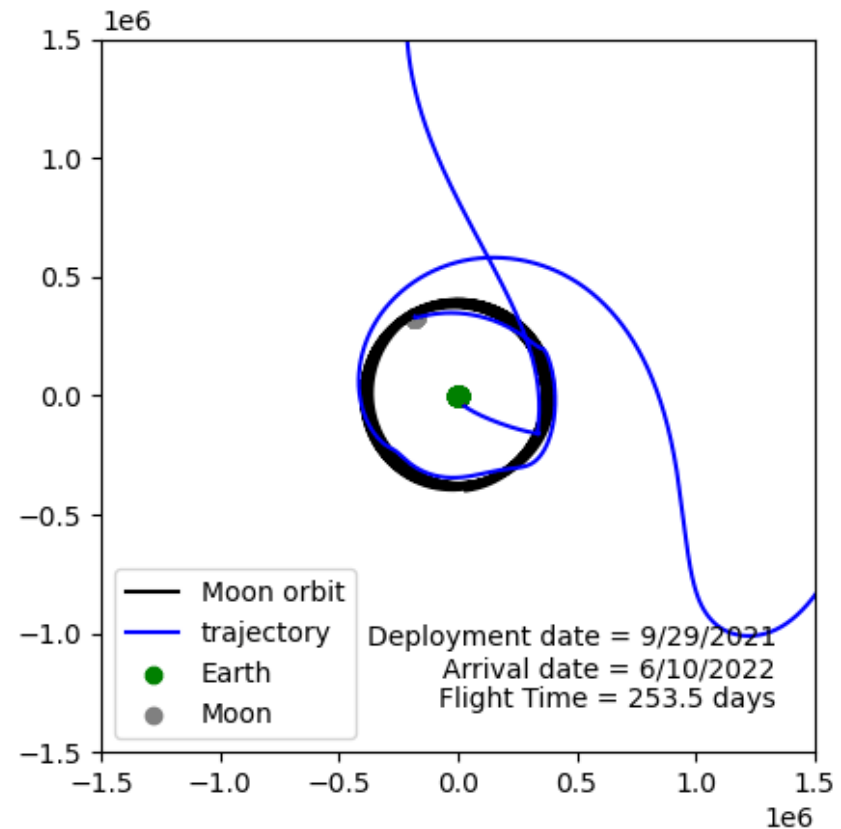
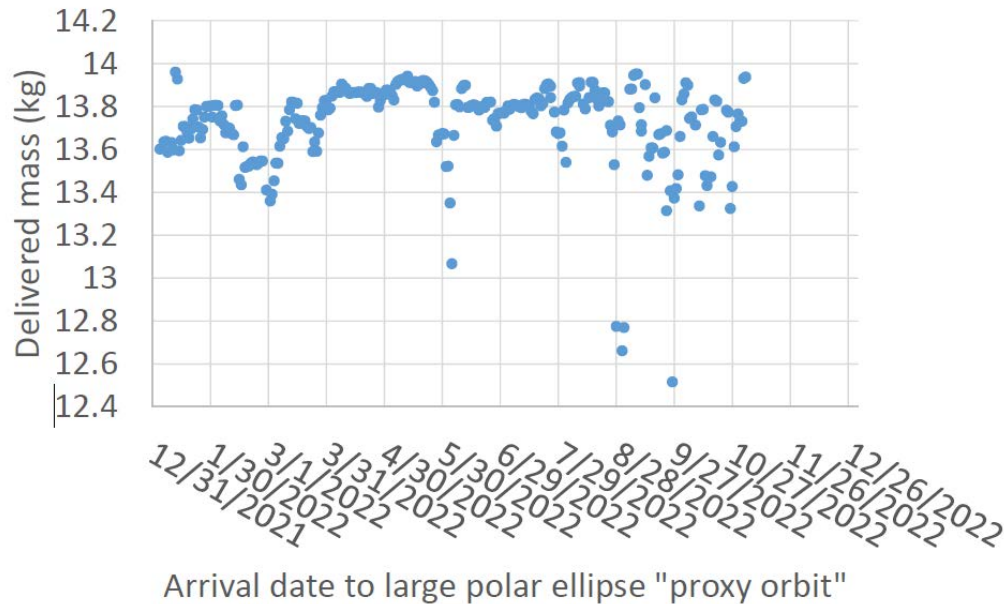
(GSE frame)



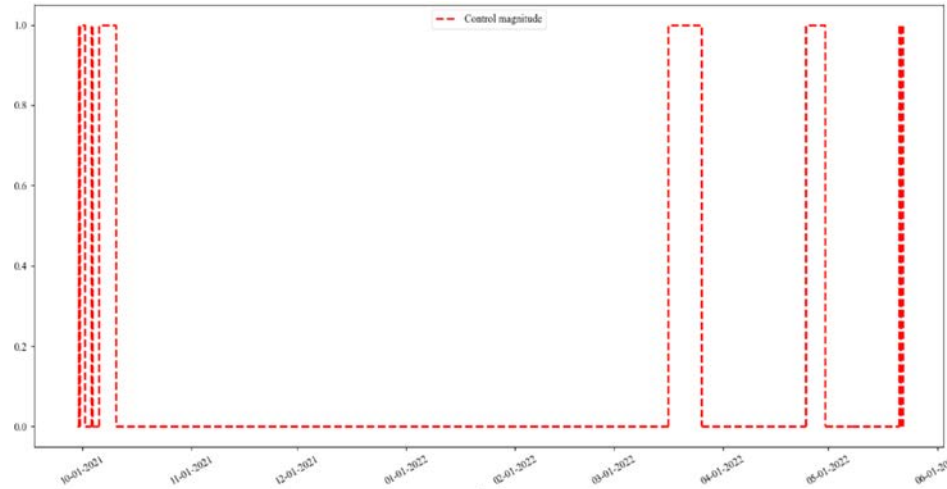
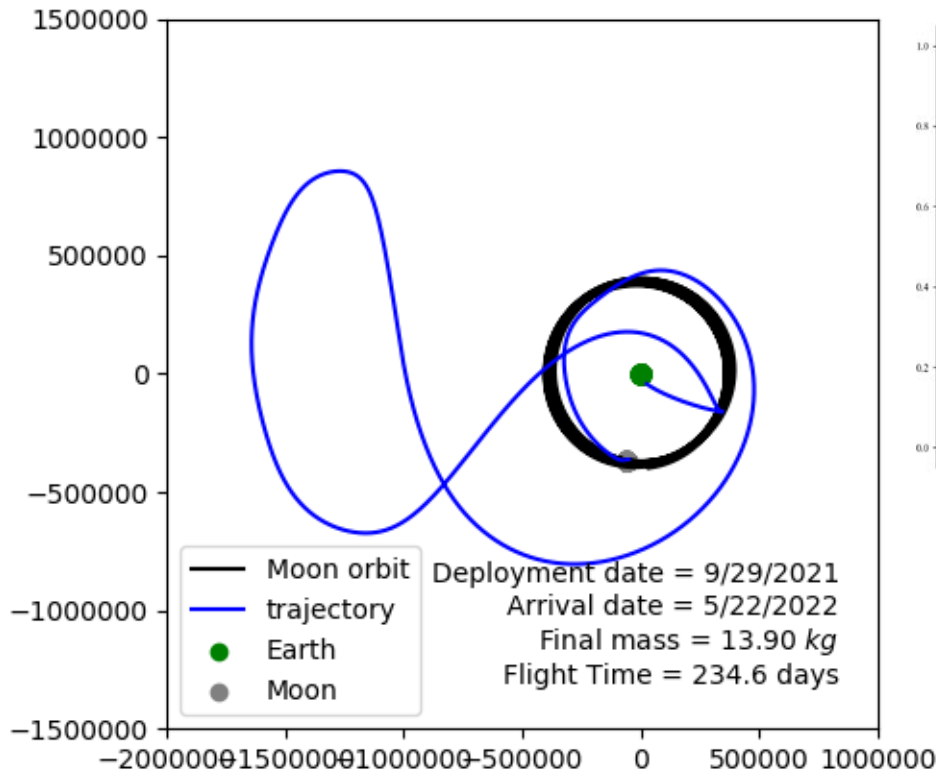
High-Fidelity

High-Fidelity Arrival Date Trade Study

Delivered mass vs arrival date for September 29th, 2021 launch



May 22nd Arrival Case, High-Fidelity



Control Magnitude Profile

Conclusions

- Multi-point direct shooting and monotonic basin hopping are promising candidates for the design of Lunar IceCube and similar missions.
- This was an experiment – the techniques used here were developed for interplanetary missions and this is our first attempt at using them for a cislunar design.
- The methods presented here allow us to quickly and flexibly re-design Lunar IceCube in response to changes in mission requirements and deployment conditions.
- This capability will be very important during operations as we re-plan as we fly to take into account error from orbit determination and maneuver execution.

Thank You

EMTG is available open-source at
<https://github.com/nasa/emtg/>

