

Background – Certification by Analysis



- Aircraft certification is a \$1 Billion to \$2 Billion investment, depending upon whether for a derivative or a new aircraft
 - Meet Regulator requirements as well as performance guarantees to Operators
 - Significant program delays/penalties if requirements/commitments not met
- Certification by Analysis (CbA) could save about \$300 million in an aircraft development program by reducing development through certification flow time while improving the product safety & performance
 - Reduce flight test time during certification
 - Reduce surprises during flight testing that cause a domino effect of rework, delay, & cost
 - Reduce time & cost to market
 - Enable insertion of new technology/design changes with minimal (without) new flight tests
- Industry worldwide identified this as a major need/opportunity and is expanding precompetitive collaboration, including with Regulators and Research Organizations (working together to raise the water level)
- The computational technology/tools required for CbA will also improved design enabling better performing airplanes

Background – CbA for Flight Test Reduction



Challenge - Community Level

- Applicant (e.g. Boeing, Airbus) and Regulator (e.g., FAA, EASA)
 must have sufficient confidence in analysis-based modeling and simulation Means of Compliance
 to reduce flight test without compromise to aviation's gold standard safety bar
- Analysis-based modeling can involve both computational and ground-based physical simulation
- Requires robust, accurate, validated methods tied to relevant data
- Mindset Change
 Flight test to validate airplane characteristics rather than to discover airplane characteristics during certification

NASA Role

- Tools & Methods validated with relevant data at the airplane characteristic and underlying physics levels
 confidence that one gets the right answer for the right reasons, consistently
- Broadly disseminate results to build confidence/trust, but only US access to NASA tools

Background – CbA for Flight Test Reduction



Approach

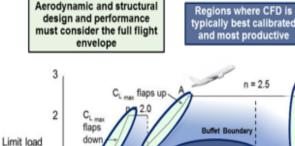
- Partner with US and Global Community
 - Already includes Boeing, Airbus, FAA, EASA, DLR, ATI/UK and many others

NASA leadership capabilities

- Computational Methods
- Experimental Techniques & Facilities
- Open Forum Dissemination of Data/Results
- · Independence from Applicant and Regulator
- Identified Low-Speed Flight Envelope as 1st focus
 - ~2/3 of certification flight test envelope
 - C_{Lmax} is key metric
- Raise the water level broadly, with targeted proprietary evaluation
 - Open/Relevant data for broad advancement/acceptance
 - Proprietary data with existing flight/ground data (limited partnerships)

Results (expected from NASA)

- NASA provides data globally, and validated tools/methods to the US community
- US industry internalizes tools/methods, and successfully implements Means of Compliance with Regulators



factor, n

- Angle of

Attack

Multidisciplinary - Flight Conditions Many Low-Speed Cert Points

n = -1.0

Collaborative Teams within NASA, within the US & Internationally

raise the water level together for the benefit of all



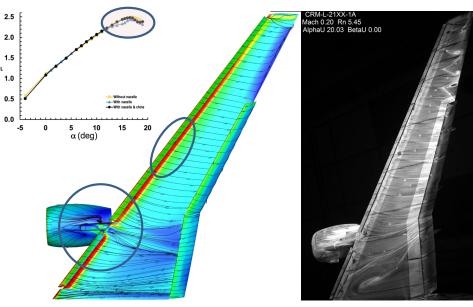
Key Elements Required

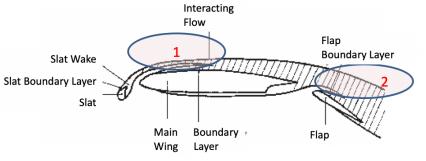
- Integration/coordination
 (across organizations/focus on big picture)
- Computational tools and technologies
- Wind tunnel test and evaluation
- Experimental tools and technologies
- Focused topics (e.g., icing)

Subject matter experts from NASA & partners in each area

- NASA Transformative Tools & Technology Project (TTT)
- NASA Aerosciences Evaluation & Test Capabilities (AETC)
- NASA Advanced Air Transport Technology Project (AATT)

Complex - Challenging - Impactful Problem

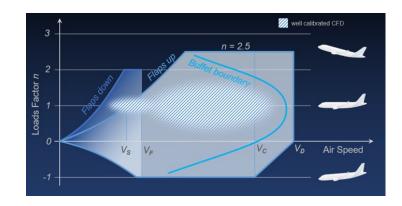




Computational Tools/Technology



- NASA CFD Technical Challenge
 - "Develop and demonstrate computationally efficient, eddy-resolving modeling tools that predict maximum lift coefficient (CL,max) for transport aircraft with the same accuracy as certification flight tests" (Level 1 milestone due 9/30/2025).
 - Can we efficiently model the physics of complex flow separation with todays HPC capability, algorithms, grid resolution, and physical models?



 CFD has been calibrated only in relatively small regions of the operating envelope where the external flow is well modeled by current RANS methods

Euler Equations (inviscid)

Reynolds-averaged Navier-Stokes: "RANS" (turbulence modeled) Large-Eddy Simulations: "LES" (large scales resolved) Direct Numerical Simulations: "DNS" (all scales resolved)

Increasing physics, increasing cost

Integrated Test and Evaluation



- Responsible for planning, integrating requirements and conducting validation focused WT tests including incorporating WT corrections using high fidelity CFD tools
- Define Reference Configuration: CRM-HL (open/relevant) (NASA/Boeing)
 - Based on the original, broadly used CRM for high speed
 - Present representative flow physics of interest for CFD methods development & validation, improved testing and data acquisition methods, high lift technology development and high lift aerodynamics knowledge enhancement
 - Provide reference performance metrics for a conventional high lift system
 - Be easy to replicate across a wide user base
 - AIAA-2020-2771 Lacy, D., et al., "Definition of Initial Landing and Takeoff Reference Configurations for the High Lift Common Research Model (CRM-HL)"
 - Final refinement nearing completion



Initial 10%-scale CRM-HL in 14x22 at Langley

Experimental Tools and Technology



Advanced measurement initiatives

(beyond standard measurements)

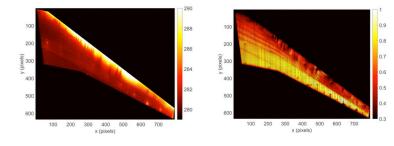
- "get the right answer for the right reasons"
- On-body to high Reynolds number in cryogenic environment
 - Model deformation
 - Surface flow visualization
 - Skin friction, transition, flow separation
- Off-body to high Reynolds number in cryogenic environment
 - FLEET (Femtosecond laser electronic excitation tagging)
 - · High Speed Velocimetry including PIV
 - New Measurement Techniques being investigated
 - Stereo PTV, Stereo FLEET, Filtered Rayleigh Scattering

Wind Tunnel test environment

- Reynolds number to full scale
- Semi-span vs full-span
- Model support

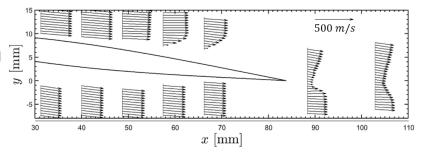
· Test efficiency initiatives

Remote control actuation, for example



TSP image

Extracted skin friction lines



Mean Velocity - FLEET

CbA Integrated Roadmap

Regulatory, other research

CY 2018 2019 2020 2021 2023 2024 2025 2022 CQbA = Certification NASA Integration Qualification by Analysis **Project** MoC = Means of Compliance **AIAA CFD2030 Integration Committee** Timeline for near and mid-term community impact CRM = Common Research AIAA CQbA Community of Interest International community engagement & partnership HL = High Lift HS = High Speed **CQbA NRA** Regulatory participation SS = semi-span FS = full-span TTT **Prediction Workshops HLPW** = High Lift Prediction Workshop Data Fusion/Analytics/UQ (TBD) JFPW = Juncture Flow Prediction Workshop Computational NASA Flow Solvers/Models for US TTT NRA = NASA Research **Tools & Technology** Announcement results broadly disseminated Physics, Numerics, Efficiency CP = Challenge Problem Integrated UW = University of Washington **Test & Evaluation** SBSE = Smooth Body Experimental Validation is Critical – flight Re is key Separation Experiment **Multiple** High Lift Common Research Model Open/Relevant CRM-HL Ground Data for Community NTF TTT + AETC + AATT with US + International Partners ETW NASA Juncture Flow Model (JF) Unit Boeing Speed Bump (SBSE) Q5m **Experimental** TTT 14x22 Tools & Technology Advanced Measurements for Critical Validation WT Test Environment O TDT Advanced Measurements Advanced Test Efficiency for Speed **AETC** Advanced Manufacturing (AM) 0.3m Cryo Rapid Testing (Active Systems) Opportunities for Test/Improve/Test Cycle on Techniques UW 3x3 Icing TTT + AETC with US + International Partners Notre Dame **Ice Shape Development** exploring icing, acoustics, aeroelastics specialty focus **CFD Tools Icing Tools**

*proposed

Regulatory compliance

version: 17 December 2020

Regulatory compliance

CRM-HL Ecosystem Overview

involving anyone who is willing and able to contribute





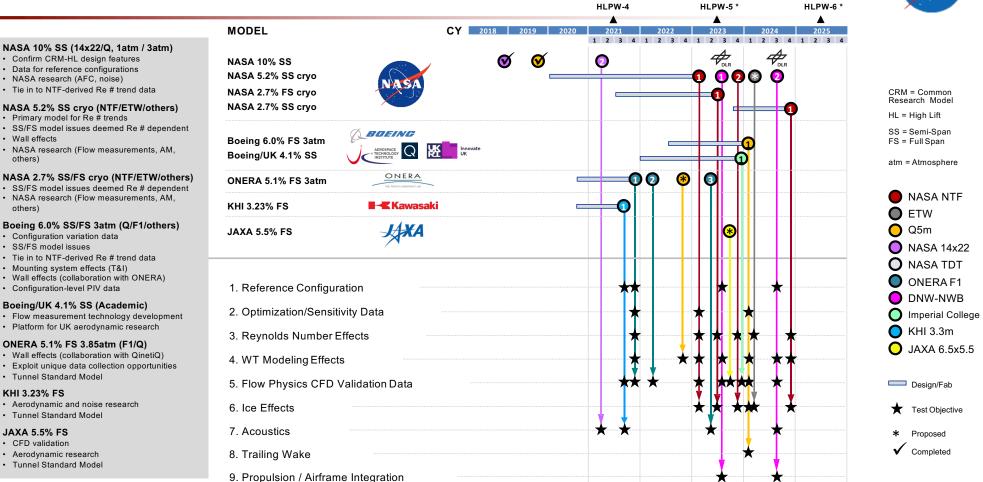
- Develop a "community-sourced" set of resources or "Ecosystem" related to the CRM-HL
 - Common controlled geometry
 - Wind tunnel models (potentially designed for use in multiple facilities)
 - Wind tunnel datasets (e.g., force & moment, noise, on & off body flowfield)
 - CFD grids (with and w/out tunnel & support system)
 - CFD results
- Enable it to continue to grow over time (add new geometry elements, configurations, models, datasets, etc.)



CRM-HL Ecosystem Development Plan

Version: 5 January 2021





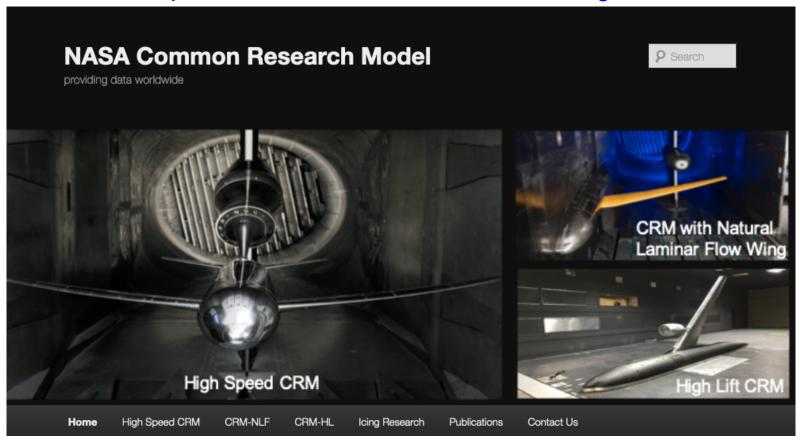
Concluding Remarks



- Collaboration has started using CRM-HL as precompetitive approach/mechanism for sharing
- Draft 5-year roadmap has been developed it is a living document
- Initial wind tunnel model and tests completed
- Next set of model and test requirements being finalized
 - Coordination ongoing between CFD and test communities
 - Some activities funded, others in proposal stage, others TBD
- The value is in partnering and leveraging investment for mutual benefit
 - Opportunities in CFD Tools/Tech, Test Tools/Technology, and advance airplane technology
- Join in the fun CRM-HL Ecosystem Meetings typically at AIAA Aviation and SciTech Forums
 - Contact me and I'll put you in touch with someone who can help richard.a.wahls@nasa.gov



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