



The NASA Advanced Composite Solar Sail System (ACS3) Flight Demonstration: A Technology Pathfinder for Practical Smallsat Solar Sailing

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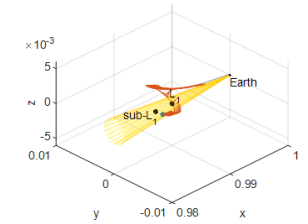
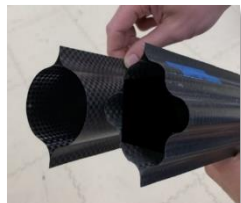
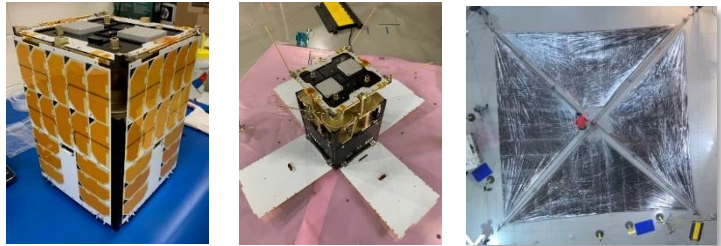
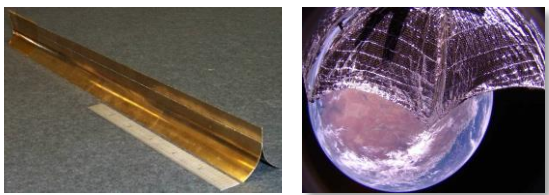
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Overview

- Motivation and Background
 - *Metallic deployable booms and solar sails*
- NASA Deployable Composite Boom (DCB) technology
- The Advanced Composite Solar Sail System (ACS3) technology demonstrator
 - *ACS3 Flight Systems*
 - *ACS3 Mission*
 - *Extensibility to Future Small Sat Solar Sail Missions*
- Summary/Status



LightSail 2 Update: Boom Buckling Anomaly

ref: The Planetary Society, 17 June 2020



*Buckled metallic 'TRAC' boom
between sail quadrants.*

"LightSail 2 near central and eastern Australia

This LightSail 2 image captured on 15 January 2020 shows central and eastern Australia, with north approximately at right. Shadows are visible from the spacecraft's solar panels; the panel shadow at right appears at a different angle than expected. A bent sail deployment boom is also visible in the gap between solar sail panels in the upper-left. Imaging initially identified both of these anomalies; neither has had a significant impact on the mission."

<https://www.planetary.org/blogs/lightsail-2-extended-mission.html>



NASA Deployable Composite Boom Project (DCB)

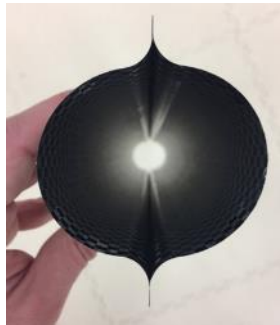
Developing high performance deployable composites for small sat applications

The Deployable Composite Booms project (DCB) was begun in 2017 to advance manufacturing and flight readiness of compact deployable composite booms:

- Sponsor: NASA Space Technology Mission Directorate (STMD) Game Changing Development Program
- Collaboration between NASA and German Aerospace Research Agency (DLR).

Benefits:

- High bending and torsional stiffness.
 - *Closed cross-section can carry high compression loads, minimizing risk of collapse.*
- High packaging efficiency.
 - *Ideal for small volume spacecraft.*
- High thermal stability.
 - *Insensitive to thermal distortions.*
- Low weight.
 - *< 25% mass of comparable metallic booms.*
- Scalable.
 - *DCB/ACS3 7-m boom technology is extensible to 14-m to 16.5-m deployable boom lengths.*



The Advanced Composite Solar Sail project (ACS3) will demonstrate DCB composite boom technology for solar sailing applications.

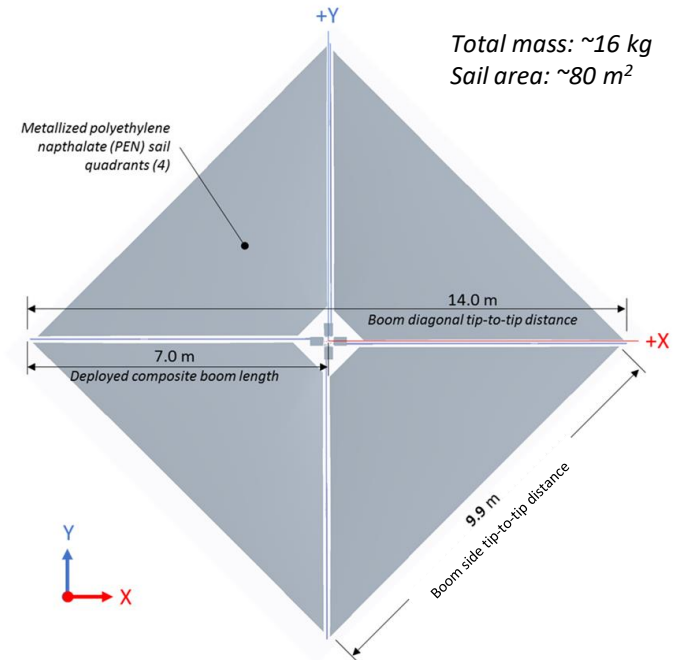
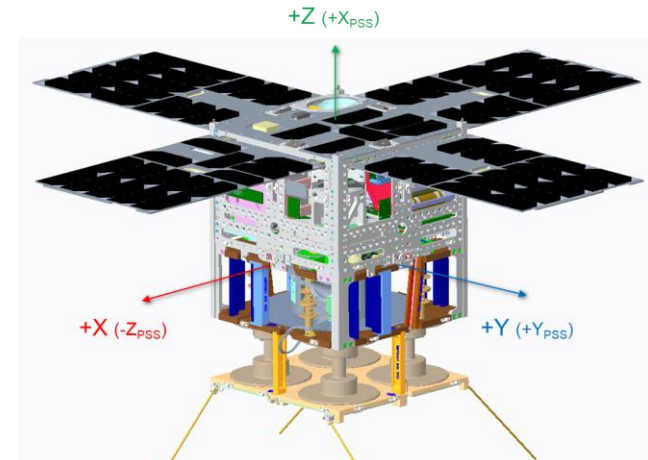


Advanced Composite Solar Sail System Project (ACS3)

LEO Solar Sail Structures Technology Demonstration [20210620]

- **ACS3 is a low-Earth orbit (LEO) flight demonstration of a composites-based small-sat solar sail propulsion system.**
 - Sponsor: STMD Small Spacecraft Technology Program (SSTP)
 - Composite Technology: STMD Game Changing Development Program (GCDP), Deployable Composite Booms Project (DCB)
 - NPR 7120.8 Research & Tech. Demonstration Project
- **Launch: 1 July – 31 December 2022**
 - Orbit: 715 km x 715 km altitude to minimize aerodynamic drag; sun-synchronous orbit preferred; mid-inclination orbit acceptable.
- **Objectives:**
 - Primary: *On-orbit deployment and characterization of a smallsat-class composite solar sail propulsion system.*
 - Secondary: *Demonstrate controlled solar sailing flight (e.g., SMA-raising/lowering) in LEO; Characterize deployed structural dynamics.*
- **Partner roles and responsibilities:**
 - NASA Ames Research Center – ACS3 payload control avionics, sail diagnostic camera system, and payload FSW.
 - NASA Langley Research Center – ACS3 solar sail system payload.
 - NanoAvionics US- 12U Spacecraft bus.
 - Santa Clara University Robotic Systems Lab – CubeSat operations support.
- **ACS3 architecture is scalable to future larger solar sails.**

ACS3 12U Spacecraft and Deployed Solar Sail





ACS3: 12U Spacecraft Subsystems

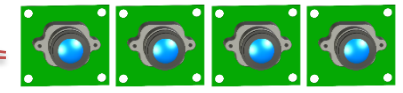
[solar panels, UHF antennas, sail quadrants, and booms omitted]

- NASA ARC
- NASA LaRC
- AST-NanoAvionics

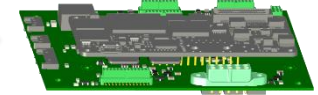
Spacecraft Bus:

- S-band transceiver
- UHF radio
- Flight computer
- ADCS
 - Reaction wheels (4)
 - Magnetorquers (3 x 3-axis)
 - Sun sensors (5)
 - GPS receiver
 - Star tracker
 - IMU
- EPS

Payload Avionics System: (AS)



AS sail diagnostic cameras (4)



AS payload control electronics

- Board included with spacecraft bus avionics stack.
- Controls sail diagnostic cameras and SBS.

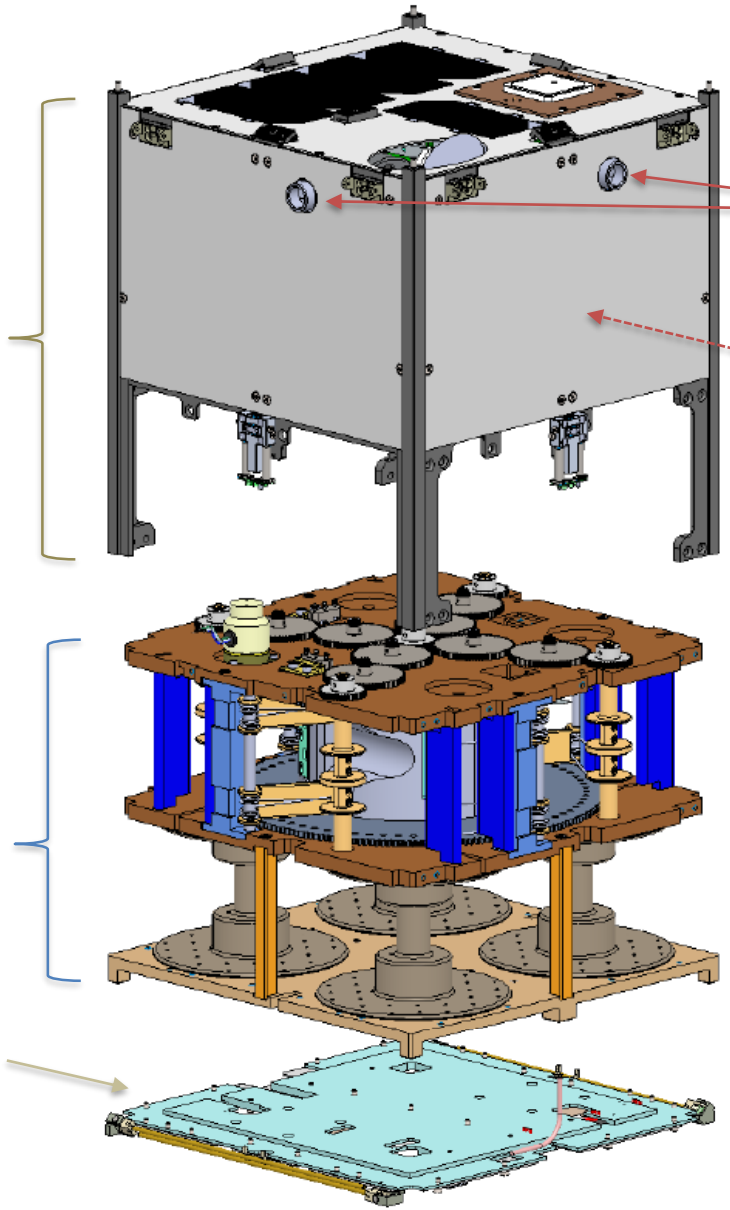
SBS composite boom deployer

SBS solar sail quadrant stowage

Sail-Boom Subsystem: (SBS)

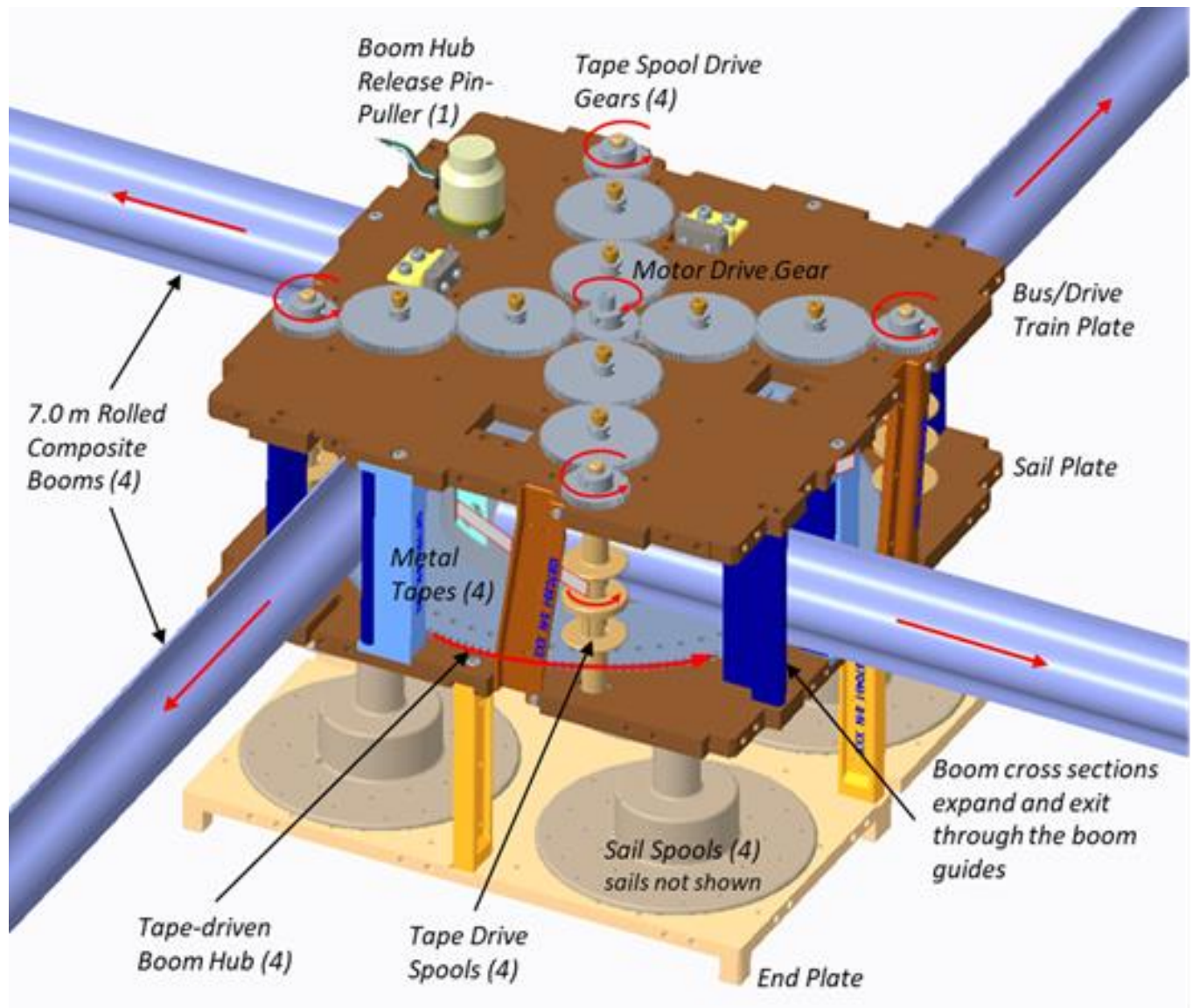
Spacecraft bus “-Z” plate:

- Pop-up UHF antennas (4)
- S-band antenna patches (2)
- Sun sensor (1)





ACS3: Sail-Boom Subsystem Deployer Design

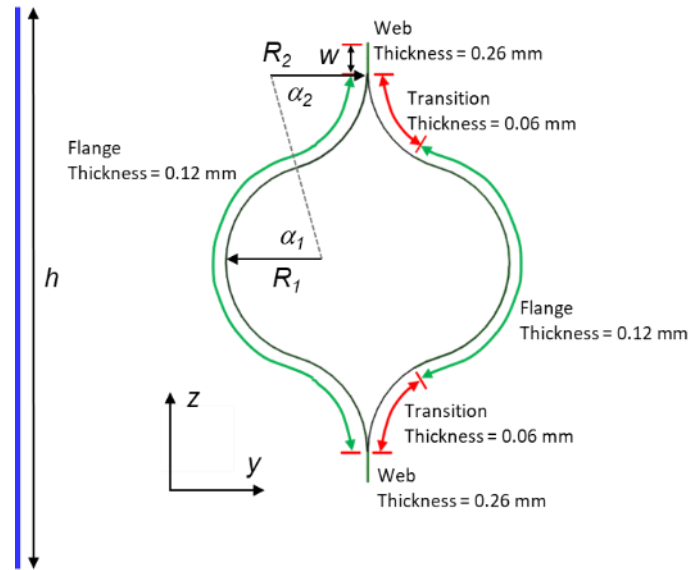


ACS3: Deployable Composite Boom Design



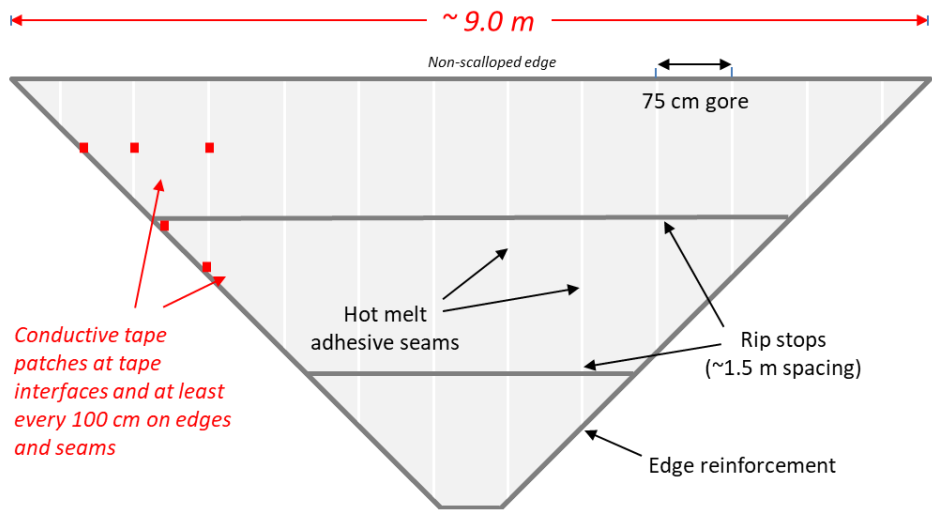
- ACS3 uses **collapsible tubular mast (CTM)** composite booms developed by the NASA STMD/GCDP Deployable Composite Booms (DCB) project.
- ACS3 booms are 7.0 m long.
 - Cross-section geometry:
 - Flattened height, h : 65 ± 0.5 mm
 - Expanded CS geometry:
 - CS width: 33.0 mm, height: 49.9 mm
 - Web height, w : 3.5 - 4 mm
 - Laminate Properties:
 - Web [45PW/0-90PW/45PW]
 - $E_{11} = E_{22} = 5.23e+07$ mN/mm²
 - Flange [45PW/0-90PW]
 - $E_{11} = E_{22} = 3.76e+07$ mN/mm²
 - Transition [45PW]
 - $E_{11} = E_{22} = 1.46e+07$ mN/mm²
 - Optimized for minimum coiling diameter and high deployed stiffness.
 - Minimum safe coiling diameter: 115 mm.
 - ACS3 boom hub diameter: 120 mm.
 - Fabricated by NASA Langley.

Individual Boom Mass: 164 g





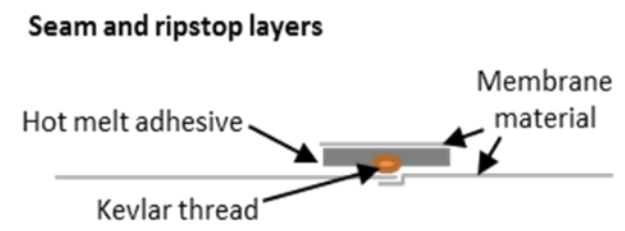
ACS3: Solar Sail Quadrant Design and Fabrication



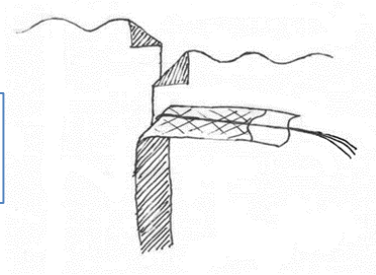
Conductive tape patches at tape interfaces and at least every 100 cm on edges and seams

Quadrant Design

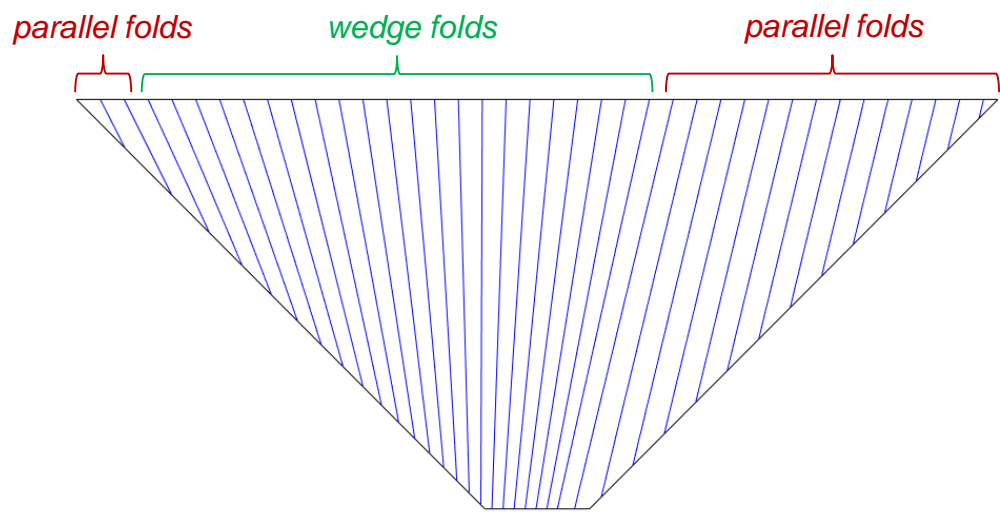
Individual Quadrant Mass (w/ hardware): 85 g



Flight membrane material:
2.115 μm Al/PEN/Cr



Seam detail



Quadrant folding scheme



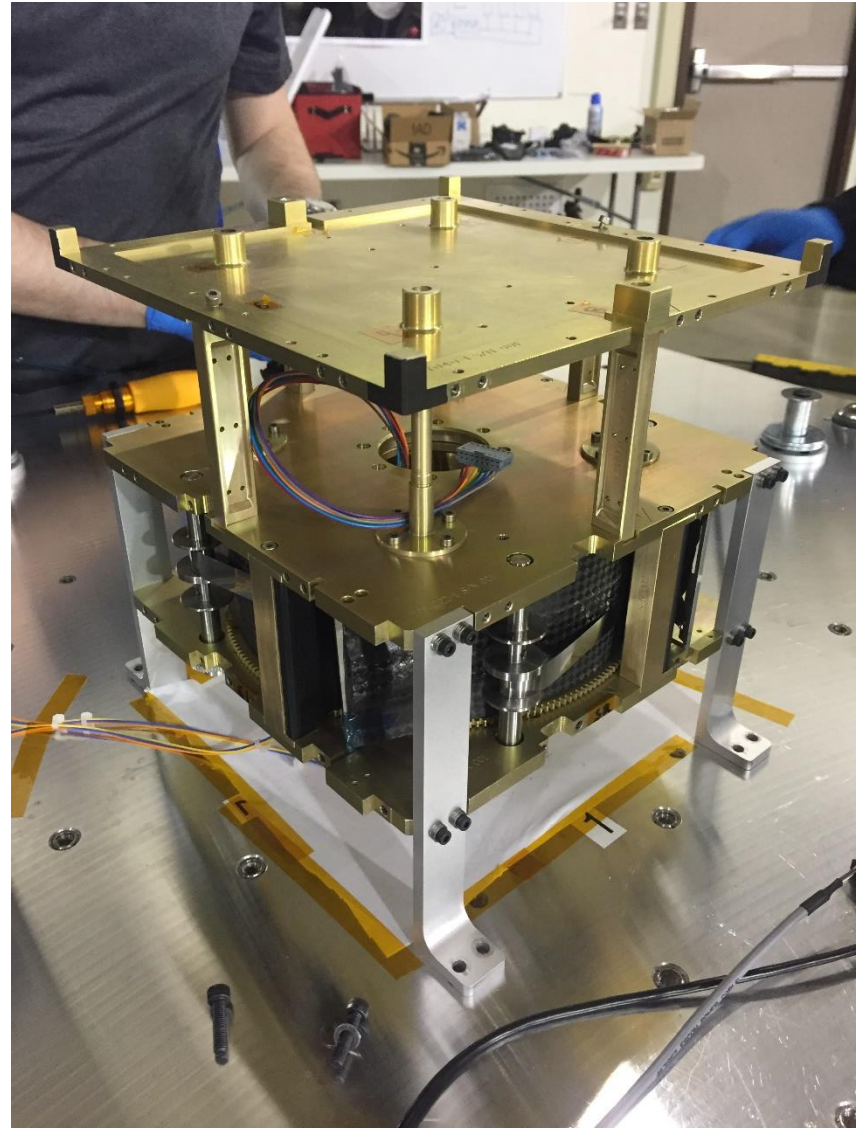
Quadrant spooled for stowing



ACS3: Boom Packaging



Boom reeling and packaging MGSE

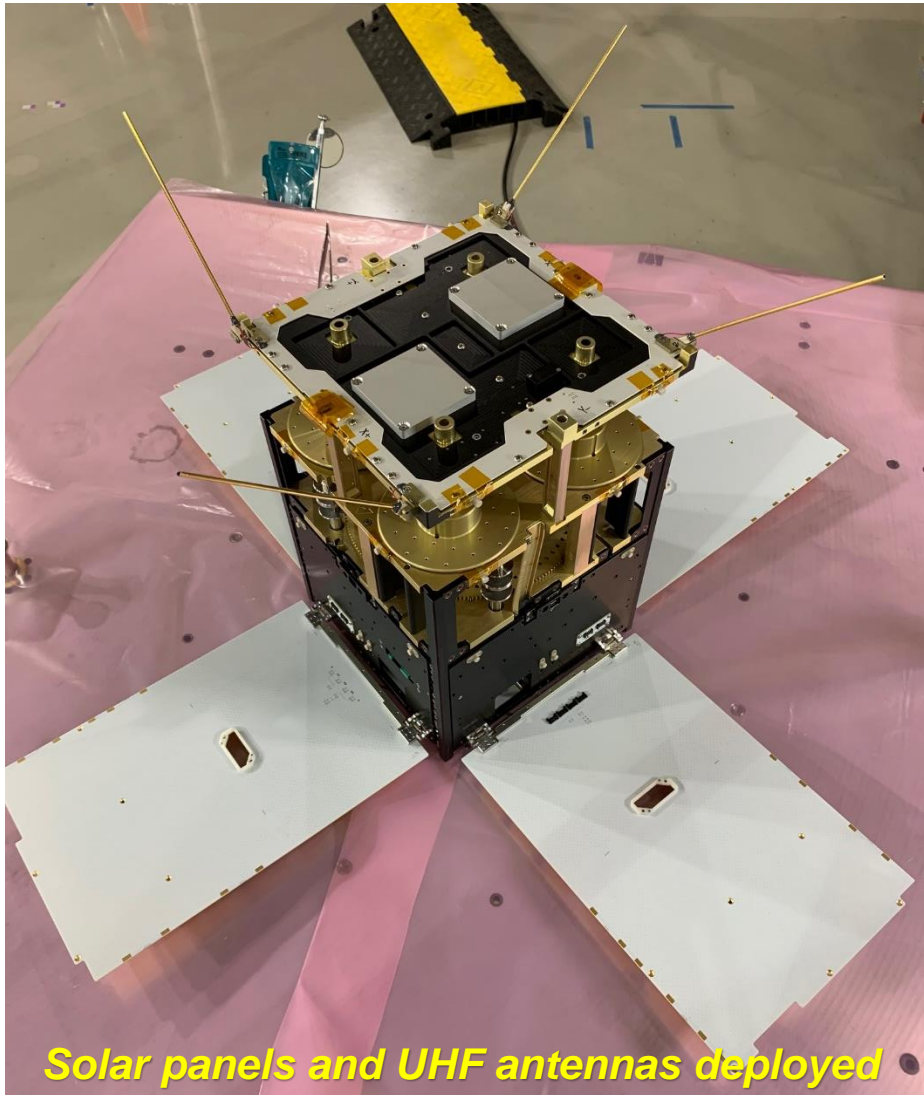


SBS Flight Unit with Development Booms



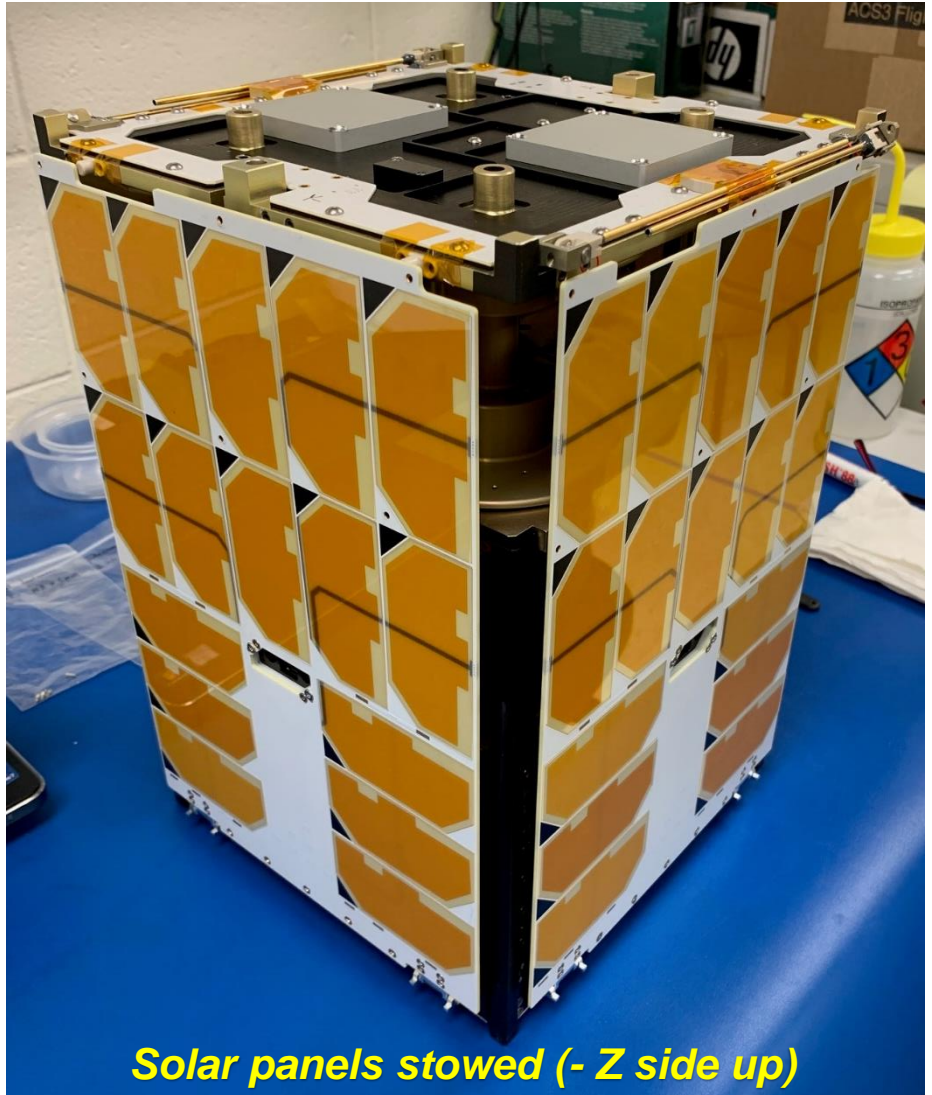
ACS3: SBS Flight Unit w/ Bus Structural Model – Solar Panels Stowed/Deployed –Z side up

Pre-Sail Deployment Configuration



Solar panels and UHF antennas deployed

Launch Configuration



Solar panels stowed (- Z side up)

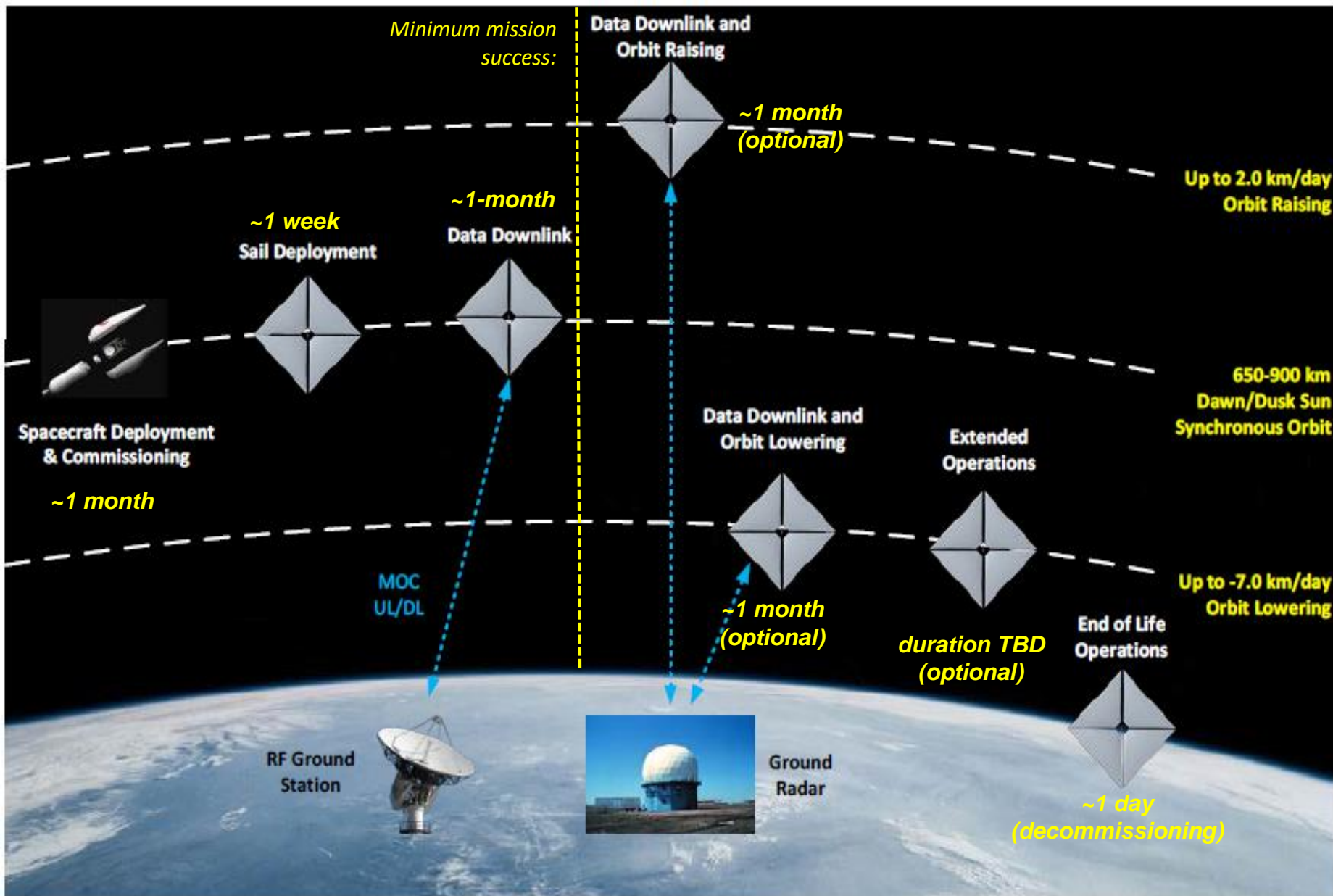
SBS Prototype Deployment Testing w/ Development Sails

Deployment Test #6, 1x tensioning springs * Deployment time: 26 minutes *





ACS3: Technology Demonstration Concept of Operations





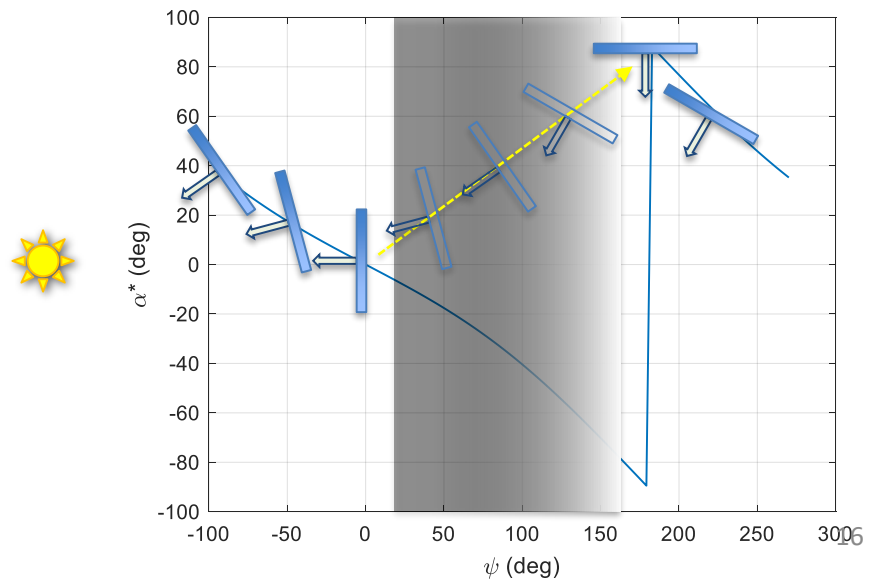
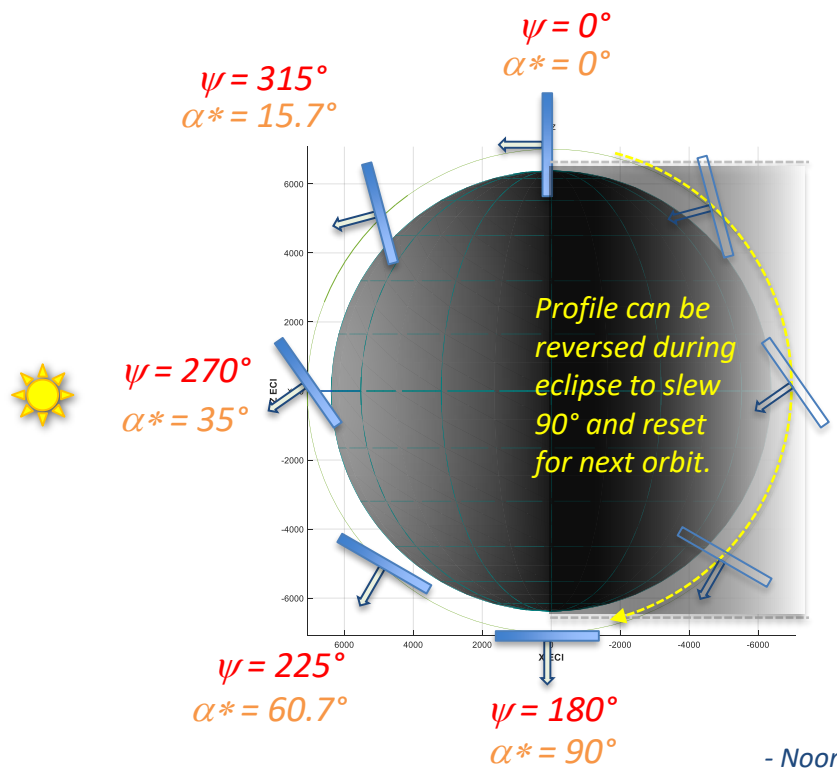
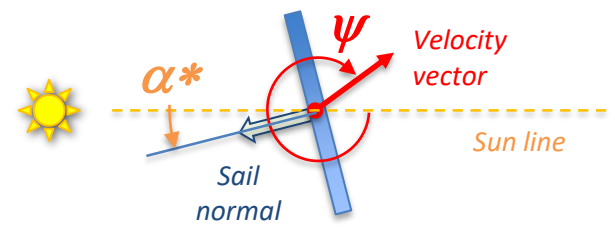
SMA Raising/Lowering Sail Steering Profile [ref: McInnes, 1999]

- Locally optimal steering law for maximum energy gain/loss each orbit.
 - Sail oriented at all times to maximize solar radiation thrust component in direction of flight.
 - For lowering, thrust component opposite direction of flight is maximized.

Locally optimal sail pitch angle.

Angle of velocity vector with respect to sun line.

$$\alpha^* = \frac{1}{2} \left[\psi - \sin^{-1} \left(\frac{\sin \psi}{3} \right) \right]$$

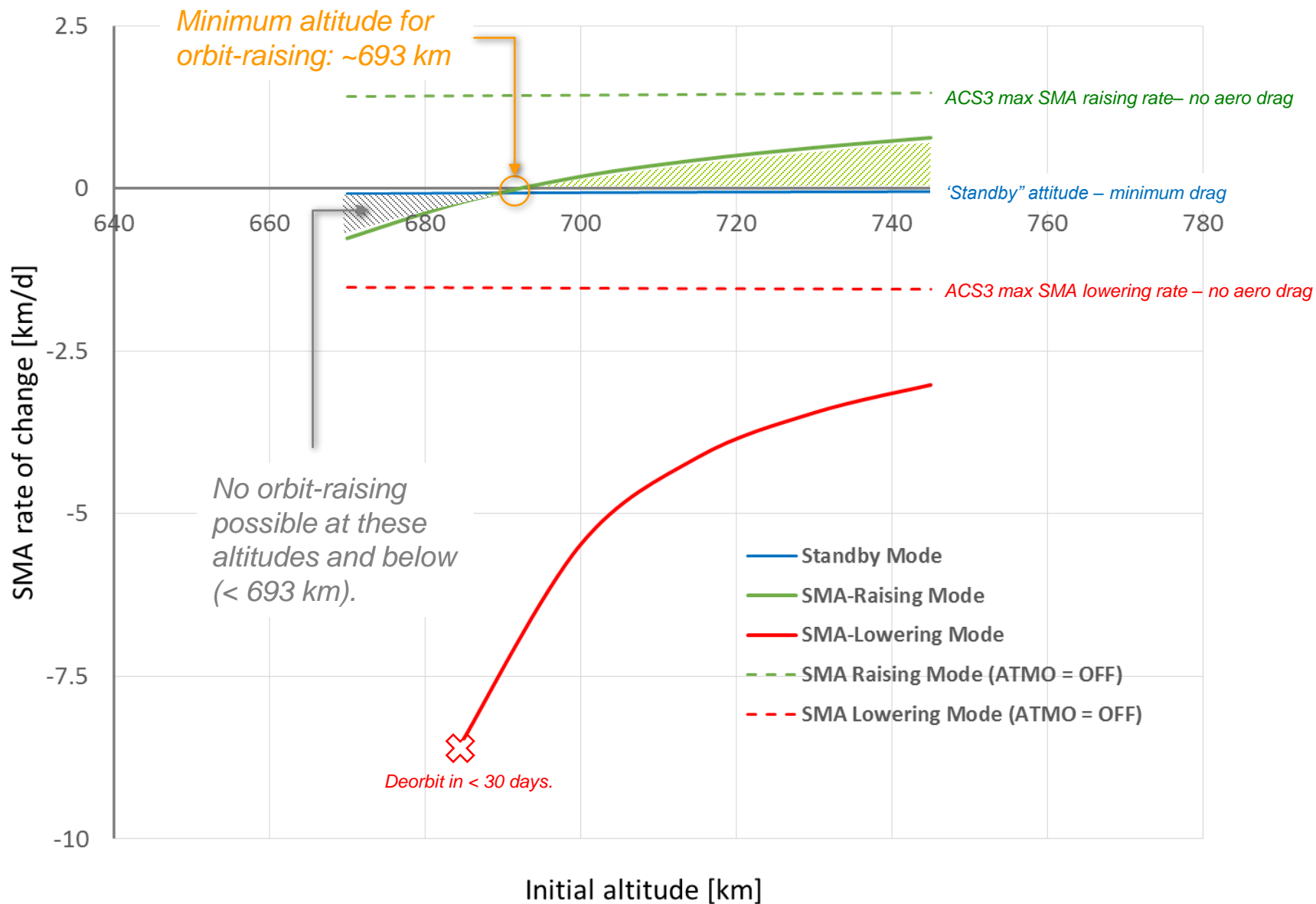


- Noon-midnight SSO example shown. -

ACS3: 670-745 km, $i = 45$ deg: Altitude Effect on SMA Raising/Lowering



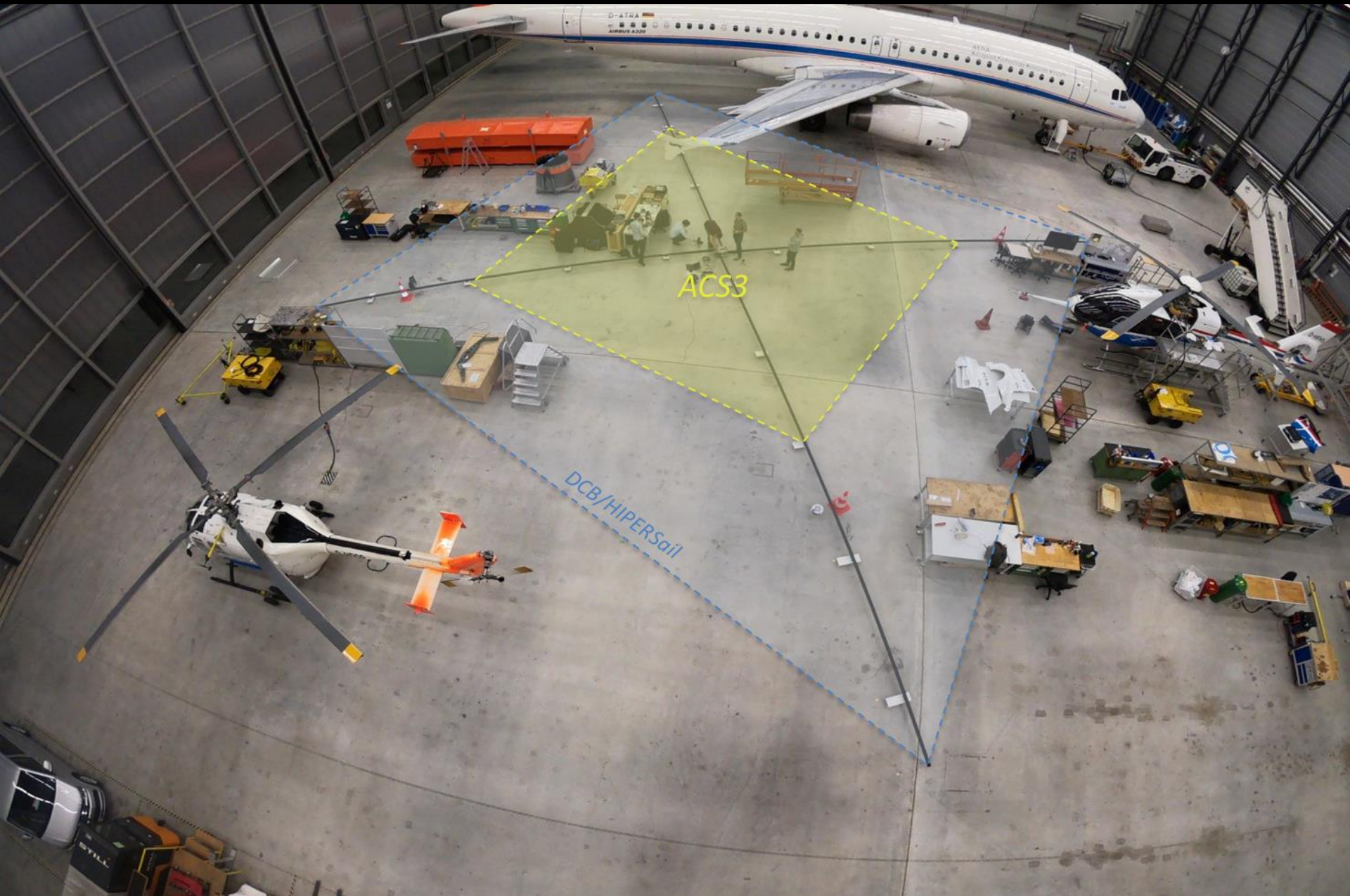
[1 August 2022 deployment, $RAAN_0 = 130$ deg; 'SMA Raising'; ACS3 v.18]





Extensibility to Future Solar Sail Applications

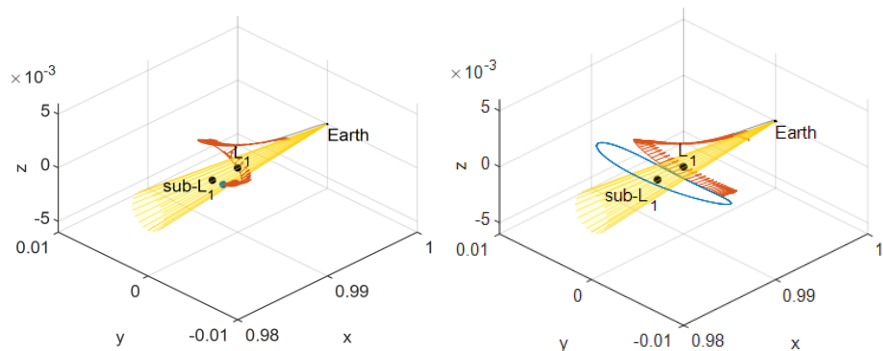
DCB 16.5-meter boom deployment testing at DLR ca. 11/12/2019



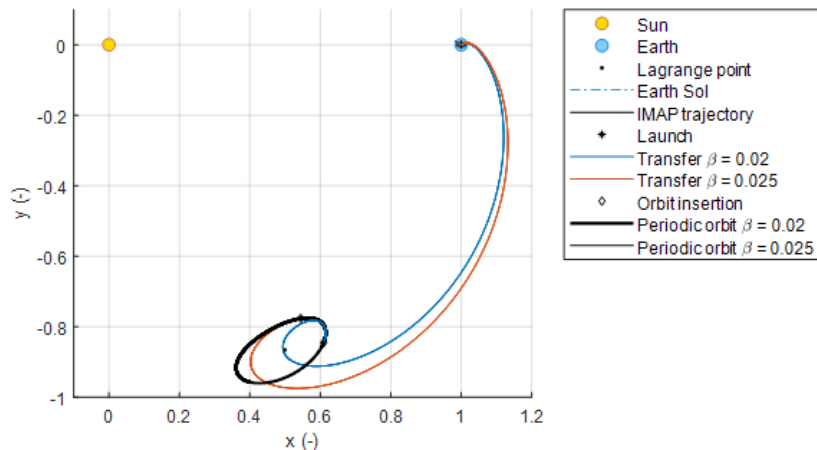


DCB/ACS3 Solar Sail Mission Applications $\beta = 0.02-0.025$

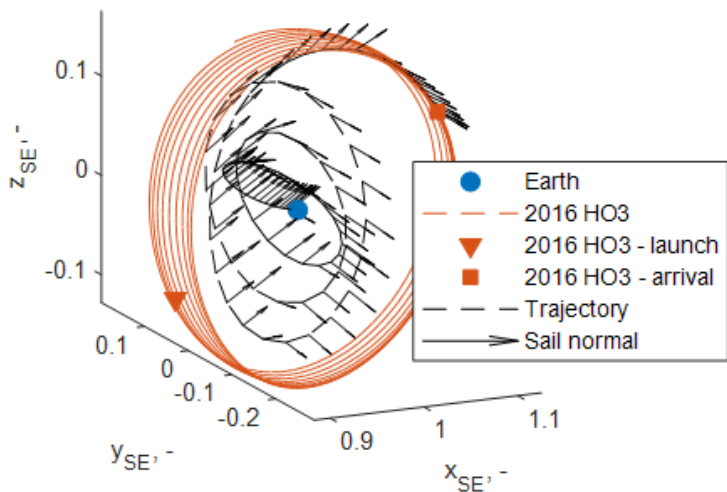
Sun-Earth Sub-L1 Space Weather EW



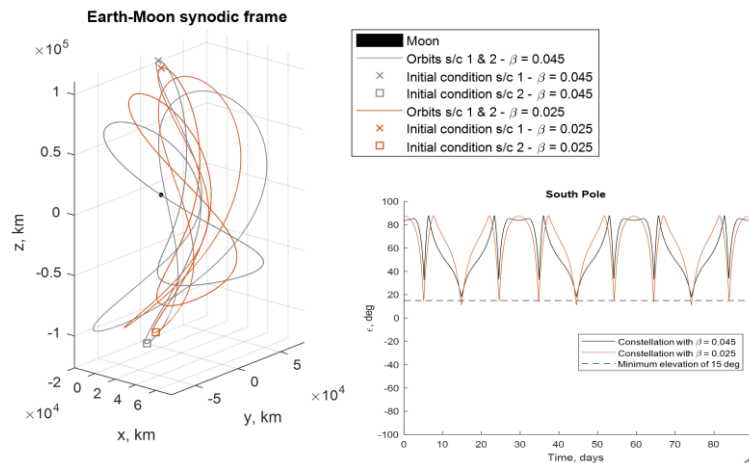
Sun-Earth L1-L5 Transfers



NEA Planetary Science

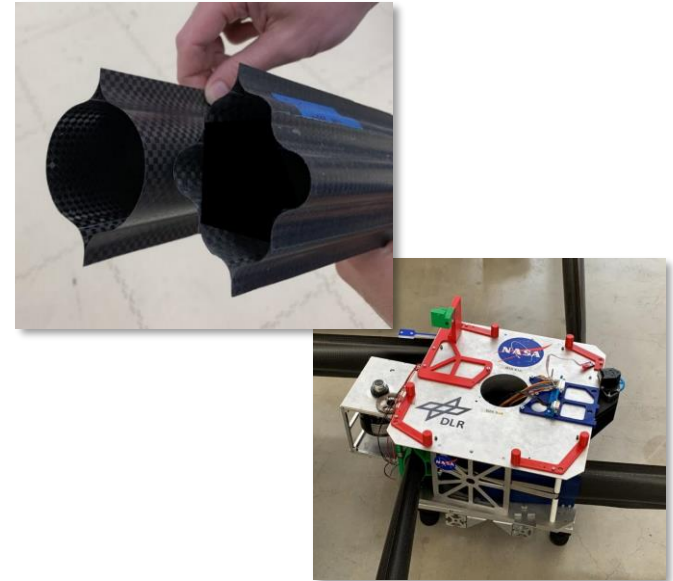


Lunar South Pole Comm Relays



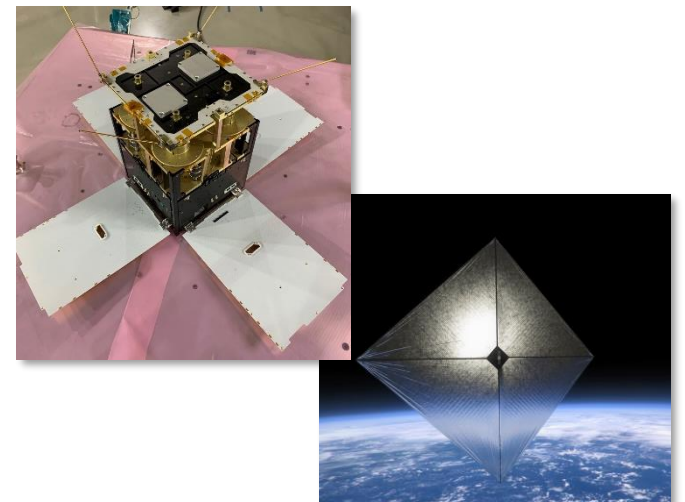
- **Deployable Composite Booms Project (DCB):**

- *DCB-1: successfully developed DCB design and manufacturing capabilities to 16.5-m scales.*
- *DCB-2: follow-on project approved for FY22 start.*
- ***DCB-2 will develop larger-scale deployable composite booms and manufacturing capabilities.***



- **Advanced Composite Solar Sail Project (ACS3):**

- *80 m² sub-scale solar sail flight demonstration of DCB-1 technology in LEO.*
- *Solar sail payload and 12U spacecraft Assembly, Integration and Testing underway.*
- ***ACS3 launch window: July-December 2022.***



ACS3 and DCB: Questions and Information

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