

Implicit Temporal Methods for Entropy Stable Spectral Collocation

Mark H. Carpenter¹ David C. Del Rey Fernández²
Chris A. Kennedy³

¹NASA LaRC

²University of Waterloo

³Private Professional Consultant, Palo Alto, CA

Coworkers: And Many Special Thanks

- [NASA](#): Derlaga, Thompson, Jackson, Nielsen
- [Sandia Nat Lab](#): Travis Fisher
- [KAUST](#): Matteo Parsani
- [KAUST / PETSC](#): Licandro Dalcin
- [Old Dominion](#): Nail Yamaleev

The View from 35,007'

The “Whole Package” matters!

Given spatial discretization, important elements of an integrator are:

- The main integration scheme : $\mathbf{U}^n \rightarrow \mathbf{U}^{n+1}$
 - ▶ Implicit: Stepwise & internal A- & L-stability, (E)SDIRK
 - ▶ Explicit: Runge-Kutta
- Peripheral temporal components
 - ▶ Stage-Value-Predictors : $\tilde{\mathbf{U}}^{(j)}$ Starts Nonlinear iteration!!!!
 - ▶ Embedded error estimate : $\hat{\mathbf{U}}^{n+1}$
 - ▶ Controller
 - ▶ Timestep adjustment (track error $\|\epsilon\| = \|\mathbf{U}^{n+1} - \hat{\mathbf{U}}^{n+1}\|$)
 - ▶ Iteration termination (don't oversolve) Söderland
 - ▶ Reevaluate preconditioner Jacobian (only when needed)
- Solver: Linear/Nonlinear
 - ▶ Weakest link for high-order spatial/temporal schemes
 - ▶ Parallel partitioning strategies
 - ▶ Local / Global preconditioner(s)
 - ▶ H-O (DG) operators are horribly conditioned!

Do the “Unthinkable”: Add more Runge-Kutta stages!

- With more stages in main Runge-Kutta Scheme : $\mathbf{U}^{n+1}$ can we
 - ▶ **Lessen burden on implicit linear/nonlinear solver mechanics?**
 - ▶ Build schemes with better internal properties (accuracy/conditioning/error estimates)?
 - ▶ Build better Stage-Value predictors?
 - ▶ More controllability
- Are they more efficient than state-of-art 3rd-order RK integrators?
 - ▶ IMEX3(2)4L[2]SA (KC: ANM 2003)
 - ▶ Vogl, Steyer, Reynolds, Ullrich, Woodward, JAMES (J...Earth Sci), 10.1029/2019MS001700

Motivating Problems of Interest

Reliable Simulation Tools for Unsteady Phenomena

Technical Challenge: Develop robust tools for predicting aerodynamic stall, buffet, flutter and propulsion system performance

- RANS: existing models are reliable for portions of design space
- **RANS: NOT RELIABLE for unsteadiness around perimeter of envelope!** (e.g., Low-speed/High-lift, stall, buffet, flutter, propulsion systems)
- Currently, higher fidelity unsteady simulation tools are necessary (e.g., LES)

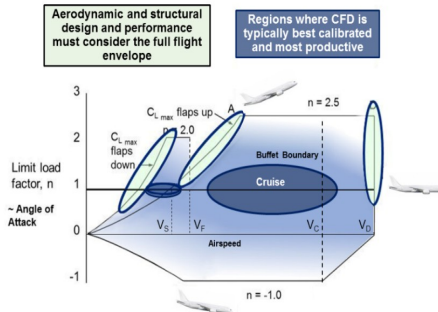


Figure: Load Factor/Airspeed Flight Envelope.

High-Order (H-O) spatial operators and numerical methods

- Entropy Conservative/Stable Spectral Collocation for Navier-Stokes
 - ▶ Enables stronger statements on integral stability of numerical solution (albeit incomplete)

But which temporal scheme?

- Stiffness $\rightarrow$ Implicit methods (A-stability, L-stability)
 - ▶ High-aspect ratio boundary elements: e.g., 1000:1:1
 - ▶ Physical scale separation: e.g., vortex generators in BL
 - ▶ Defective grid generation: e.g., sliver cells/elements
 - ▶ Point singularities: sharp trailing edge refinement
 - ▶ Source terms: Chemistry or turbulence models

An Ocean of Integrators: Automated Integration

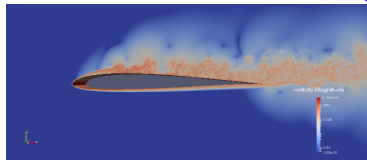
- Multipurpose integrator CFD: Linear Multistep, MultiStage, GLM
 - ▶ External Aero: nonstiff / stiff / separably stiff (IMEX, multirate)
- Unique Algorithmic Constraints
 - ▶ Adaptive Mesh Refinement → Self-starting → Runge-Kutta
 - ▶ Refinement transmission conditions: difficult for HO methods
 - ▶ Multistep transmission condition ???????? Not today!
 - ▶ Implicit RK: Stepwise & internal A- & L-stability → Robustness
 - ▶ Amenable to implicit/explicit partitioning → Efficiency
 - ▶ And the verdict is? IMEX (E)SDIRK Runge-Kutta schemes
- Lessons learned with (E)SDIRK Schemes
 - ▶ H-O accuracy → multiple stages → mandates huge timesteps
 - ▶ Factor 20 larger timestep (relative to BDF2)
 - ▶ Whack-a-mole: Increased burden on linear/nonlinear solver
- Potentially a problem with HO spatial discretizations?

A Complimentary Strategy for H-O solvers

- The main integration scheme : $\mathbf{U}^n \rightarrow \mathbf{U}^{n+1}$
 - ▶ Stepwise & internal A- & L-stability, (E)SDIRK
 - ▶ (E)SDIRK characterized by γ on diagonal of Butcher tableau
 - ▶ $\gamma : \iff (\mathbf{I} + \gamma \Delta t \frac{\partial R}{\partial \mathbf{U}}) \delta \mathbf{U} = R(\mathbf{U})$
 - ▶ Increase stages by one
 - ▶ Minimize γ , $\rightarrow$ better matrix conditioning
 - ▶ Optimize error coefficient to allow larger timesteps
- Peripheral temporal components
 - ▶ Stage-Value-Predictors: $\tilde{\mathbf{U}}^{(j)}$, used to start nonlinear iteration
 - ▶ Study SVP accuracy vs. internal stability tradeoff
 - ▶ Design SVP to be A-stable when possible
 - ▶ Embedded Error estimate : $\hat{\mathbf{U}}^{n+1}$
 - ▶ Controller
 - ▶ Timestep adjustment (track error $\|\epsilon\| = \|\mathbf{U}^{n+1} - \hat{\mathbf{U}}^{n+1}\|$)
 - ▶ Iteration termination (don't oversolve) Söderland
 - ▶ Reevaluate preconditioner Jacobian (only when needed)
- Can additional stage result in better efficiency?

Baseline NASA LaRC Mechanics

Applications → “Productionized” Algorithm

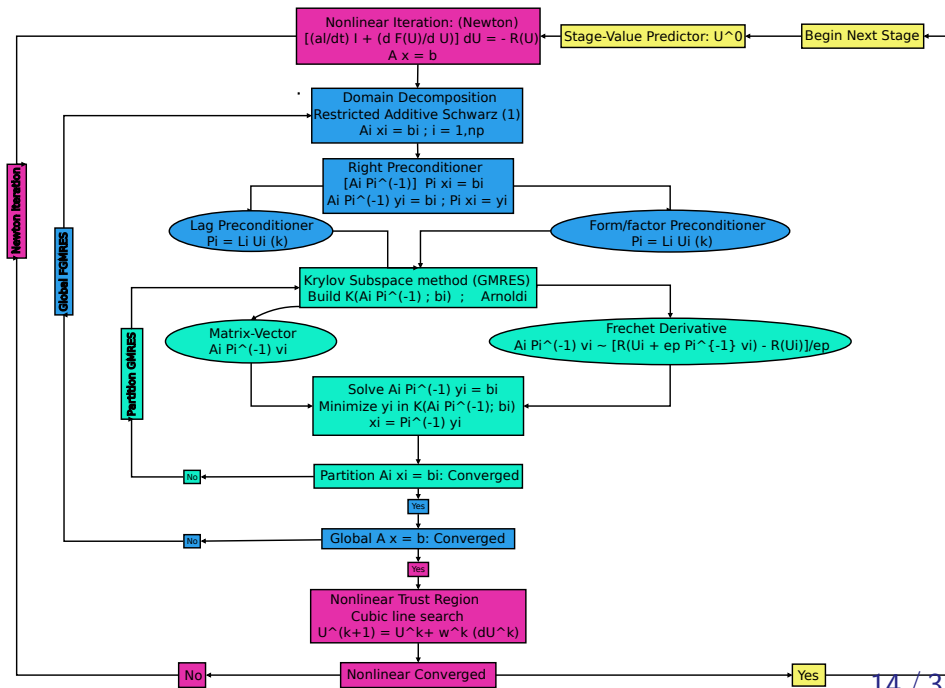


- Spatial Operator: Entropy dissipative Spectral Collocation
 - ▶ Curvilinear Hexahedral Elements: Focus on $P1 - P7$
 - ▶ Conforming / Nonconforming interfaces
 - ▶ H- & P-refinement capability
 - ▶ Several Dissipation Kernels
- Temporal Operator: Implicit-Explicit Runge-Kutta
 - ▶ 3rd- & 4th-order IMEX Runge-Kutta : Stage-order 2
 - ▶ Explicit Singly Diagonally Implicit RK (ESDIRK) Implicit
 - ▶ Stepwise and Internal A- & L-stability
 - ▶ Error estimator / Stage-Value-Predictors (SVP)
- Solver Mechanics: PETSC infrastructure

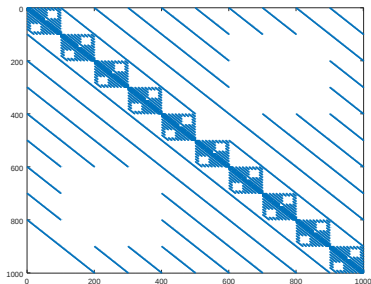
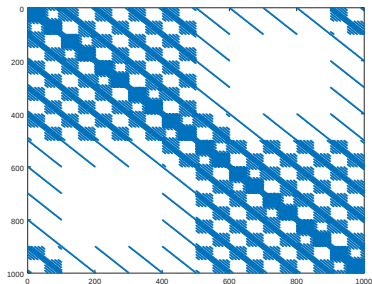
PETSc Solvers: Another talk for another day!

- Conventional Architecture: 1.5GB/process
- Solver Mechanics: PETSC infrastructure
 - ▶ Preconditioned “Reduced-Jacobian”, Newton-Krylov
 - ▶ Classical Dom-Decomp: Additive-Schwarz (1 layer overlap)
 - ▶ Classical parallel “Inner-Outer” Krylov Iteration (Saad '93, Van Der Vorst, Vuik '94)
 - ▶ Inner: Partitions each run preconditioned GMRES
 - ▶ Approximate Preconditioning matrix: Thin-layer NS
 - ▶ Preconditioner: ILU(k), $k=0,1,2$
 - ▶ Stored ILU factors: 10-20 Newton Iterations
 - ▶ Arnoldi subspace $AV^n = \bar{V}^{n+1}\bar{H}$ using Frechet derivatives
 - ▶ Outer: Flexible GMRES
 - ▶ Newton Update
 - ▶ Trust-Region / line search
 - ▶ Stage-value-predictor used at beginning of each RK stage

PETSC: Nonlinear/Linear "Inner-Outer" Solver



Full Navier-Stokes vs. Thin-Layer NS Preconditioner



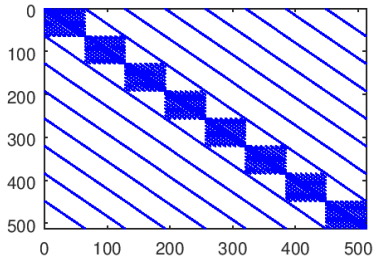
Spy: Full NS VS. Thin-Layer NS. 2x2x2 Periodic. P4 Elements

<i>PolyOrder</i>	<i>NNZ(Full)</i>	<i>MB/Elem</i>	<i>NNZ(Thin)</i>	<i>MB/Elem</i>
1	164	0.0082	68	0.0034
2	954	0.0477	378	0.0189
3	3520	0.1760	1168	0.0584
4	10010	0.5005	2750	0.1375
5	24036	1.2018	5508	0.2754
6	50701	2.5350	10045	0.5022
7	97364	4.8682	16448	0.8224

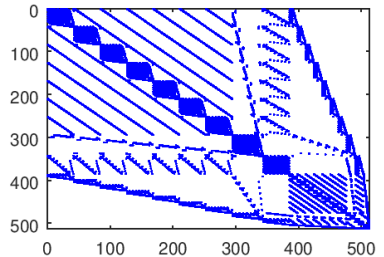
Storage (Mega-Bytes) per element: Full and Thin layer preconditioner

Thin-Layer NS Reordering Strategies

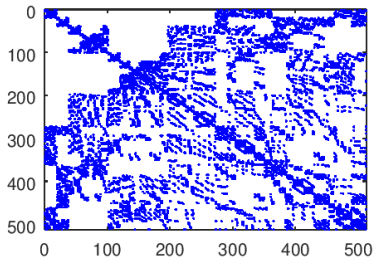
BASE: $\text{n nz}(\text{lu}(\mathbf{A})) = 233528$



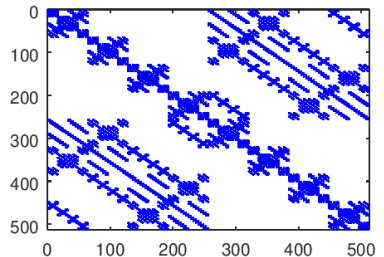
RCM : $\text{n nz}(\text{lu}(\mathbf{B})) = 183314$



AMD : $\text{n nz}(\text{lu}(\mathbf{C})) = 146036$



RND : $\text{n nz}(\text{lu}(\mathbf{D})) = 134528$



Temporal Mechanics

The Case for “Solver Optimized” Schemes: Free Lunch?

- Design “Solver Optimized” ESDIRK IMEX Runge-Kutta schemes
 - ▶ Add additional stage(s): Optimized properties offset cost penalty
- Main design considerations: 1) Accuracy and 2) Condition of solver

- ▶ Accuracy: For $\uparrow \Delta t$, need $\downarrow$ T.E.

Name	4(3)6L[2]SA	4(3)7L[2]SA
s	6	7
T.E. $A^{(5)}$	0.01224	0.000260

- ▶ Conditioning: Diag Coeffs γ : $\iff (\mathbf{I} + \gamma \Delta t \frac{\partial R}{\partial U}) \delta U = R(U)$

- ▶ BDF1: $\gamma = 1$,
- ▶ BDF2: $\gamma = \frac{3}{2}$,
- ▶ ESDIRK3(2)4L[2]SA: $\gamma = 0.43$,
- ▶ ESDIRK4(3)6L[2]SA: $\gamma = \frac{1}{4}$
- ▶ Solver Optimized: ESDIRK4(3)7L[2]SA: $\gamma = \frac{1}{8}$
 NASATM:219173, ANM(136):2019, ANM(146):2019

- Free Lunch! “A wash” for $(\Delta t)_{4(3)6} \approx (2\Delta t)_{4(3)7}$

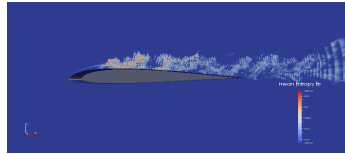
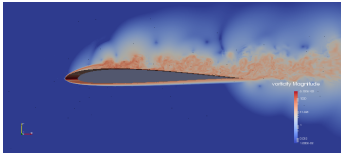
Peripheral Considerations:

Stage-Value-Predictors/Error Estimation/Controllers

- SVP: Long History in RK literature (NASATM:219173)
 - ▶ Hairer & Wanner, Laburta, Higuera & Roldán, Roldán, González-Pinto, Calvo, Olsson & Söderland, ...
 - ▶ Design: delicate balance of accuracy and stability constraints
 - ▶ Number of stages determine accuracy of stages
 - ▶ Internal L-stability of $\mathbf{U}^n \rightarrow \mathbf{U}^{n+1}$
- Error Estimators $\|\epsilon\| = \|\mathbf{U}^{n+1} - \hat{\mathbf{U}}^{n+1}\|$
- Controller (NASATM:219173)
 - ▶ Söderland's PPID
$$(\Delta t)^{[n+1]} = \kappa (\Delta t)^{[n]} \left\{ \frac{\text{tol}}{\|\epsilon^{[n+1]}\|} \right\}^{\alpha} \left\{ \frac{\|\epsilon^{[n+0]}\|}{\text{tol}} \right\}^{\beta} \left\{ \frac{\text{tol}}{\|\epsilon^{[n-1]}\|} \right\}^{\gamma} \left\{ \frac{(\Delta t)^{[n]}}{(\Delta t)^{[n-1]}} \right\}$$
$$\alpha = +\frac{0.30}{\hat{p}}, \beta = -\frac{0.05}{\hat{p}}, \gamma = -\frac{0.25}{\hat{p}}, \text{tol} = \text{user input}$$
 - ▶ tolerance (tol) set to $10^{-3}, 10^{-4}, 10^{-5}$
 - ▶ Used for all tests with a-priori NO TUNING

The Test Case and Preliminary Testing

TEST CASE: SD7003 Airfoil



- High-order workshop III, very popular test case
- $Re_{ch} = 60K$, $Pr = 0.72$, $M = 0.20$, $AOA = 8.0^\circ$
- Hex meshes (DOF): 14502 (3M), 55008 (11M), 116016 (25M)
 - ▶ Eyeball initial mesh $\rightarrow$ Spatial error estimator $\rightarrow$ Adjust mesh
- Stiffness: Inadequate GMSH skills

PolyOrd	$(\Delta t)_{max}$
P1	$9.0e-6$
P2	$1.5e-6$
P3	$4.8e-7$
P4	$2.0e-7$
P5	$9.7e-8$
P6	$5.2e-8$

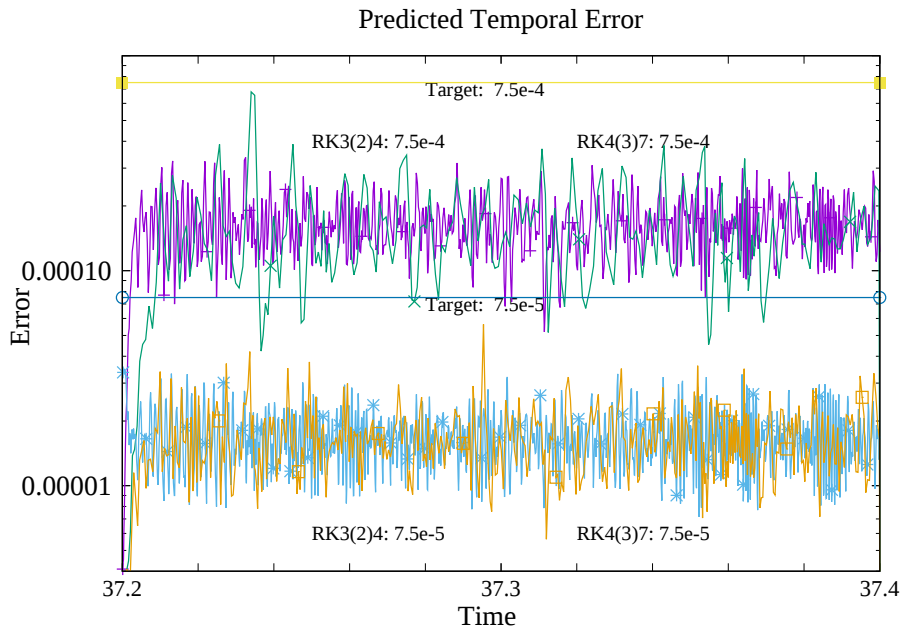
Maximum explicit timestep

Preliminary Test I: Efficacy of Controller

Can controller maintain the specified error tolerance for all schemes?
Critical capability for impartial comparison between schemes.

- SD7003: 116016 Hexahedral elements run at P3
- Tol = $7.5e-5$, $7.5e-4$
- Test at Engineering error tolerances: e.g., tol = $7.5e-4$ / timestep
HUGE ERROR (3-4 significant digits in lift and/or Drag)
- Test ONLY IMPLICIT operator from each IMEX pair
- Schemes:
 - ▶ IMEX3(2)4L[2]SA (KC: ANM 2003)
 - ▶ IMEX4RK(3)7L[2]SA (KC: ANM 2019a)
 - ▶ ESDIRK4(3)7L[2]SA (KC: ANM 2019b)

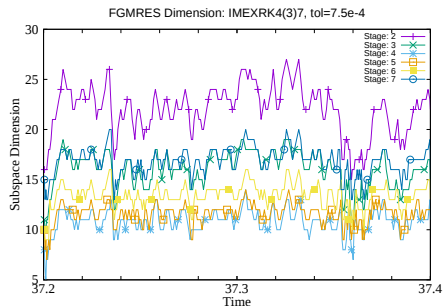
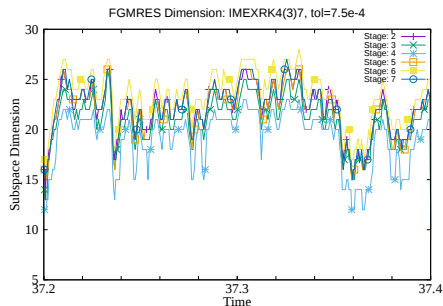
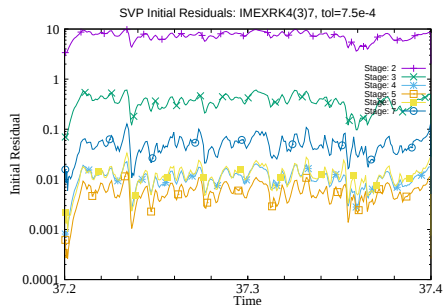
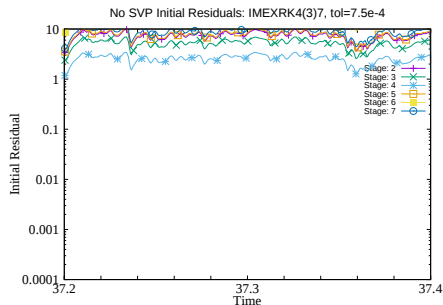
Efficacy of Controller: Maintaining Target Error



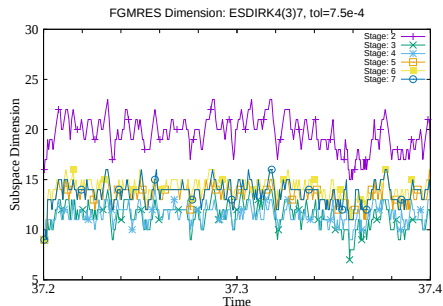
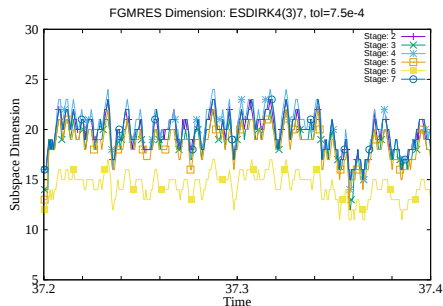
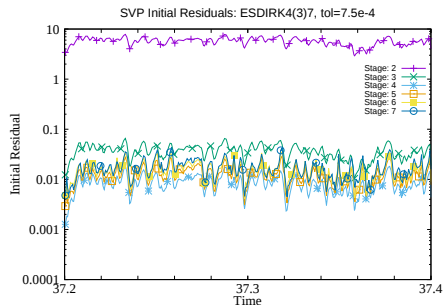
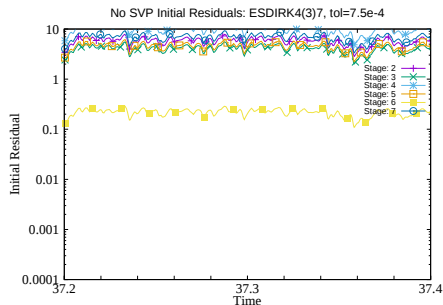
Do SVP reduce time-to-solution of all schemes?

- SD7003: 116016 Hexahedral elements run at P3
- $\text{tol} = 7.5\text{e-}4$
- Engineering error tolerance: $\text{tol} = 7.5\text{e-}4 / \text{timestep}$
HUGE ERROR (3 significant digits in lift and/or Drag)
- Testing ONLY IMPLICIT portion of each IMEX pair
- Schemes used in Testing
 - ▶ IMEXRK4(3)7L[2]SA (Solver Optimized)
 - ▶ ESDIRK4(3)7L[2]SA (Solver Optimized)

Convergence WITHOUT and WITH SVP: IMEXRK4(3)7



Convergence WITHOUT and WITH SVP: ESDIRK4(3)7



Efficacy of Controller and SVP: The Verdict is?

- Controller drives all schemes as advertised!
- Always run with the Stage-Value-Predictor!

<i>Scheme</i>	<i>WITHOUT SVP</i>		<i>WITH SVP</i>		<i>COST</i>
	<i>CPUTime</i>	<i>Steps</i>	<i>CPUTime</i>	<i>Steps</i>	<i>Ratio</i>
<i>IMEXRK4(3)7</i>	52990	245	34577	245	0.65
<i>ESDIRK4(3)7</i>	48997	311	33799	314	0.68

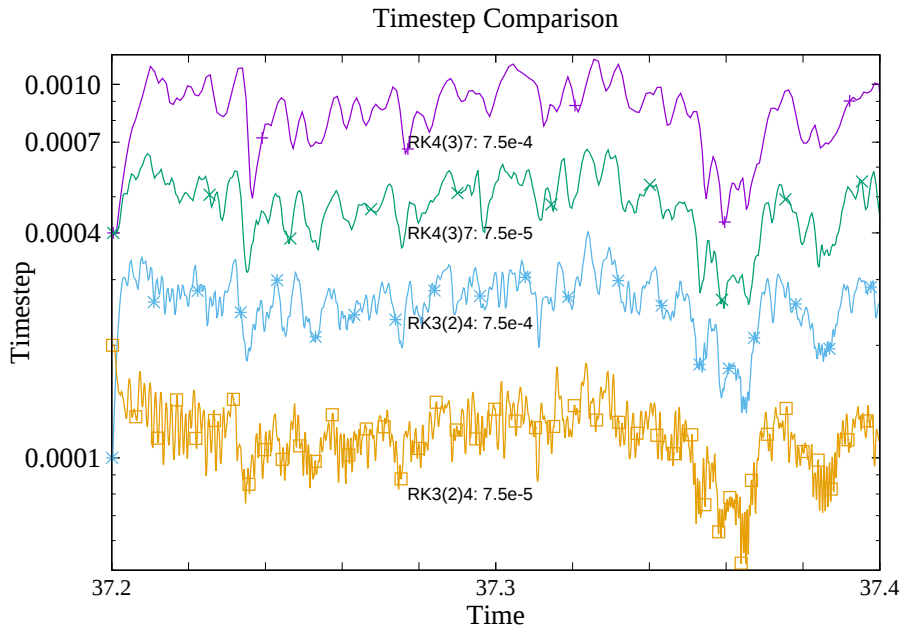
- Results consistent with those seen in past work
- Rapid convergence USUALLY correlates with initial residual reduction
- Multiple step SVP for variable time-stepping: i.e., Stage 2,3
- More work to be done to optimize A-stability / accuracy of SVP

Efficacy of Solver Optimized Schemes

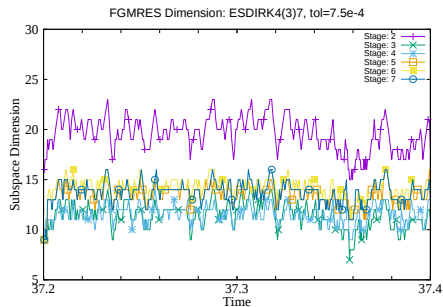
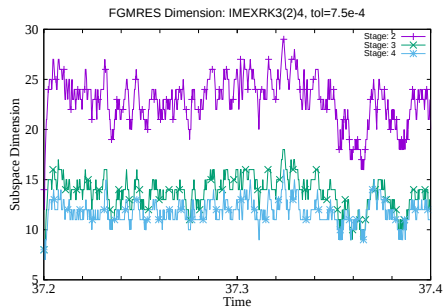
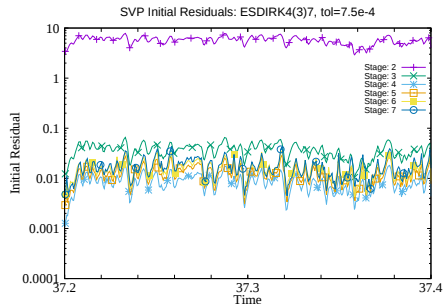
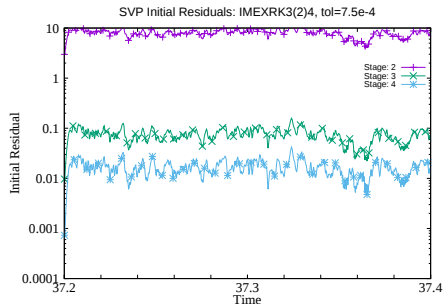
Which scheme is most efficient at large scale?

- Head-to-Head comparison
- SD7003: 116016 Hexahedral elements run at P3 and P4
- Test at Engineering error tolerances: e.g., $\text{tol} = 7.5\text{e-}5$ and $7.5\text{e-}4$ / timestep
- Test ONLY IMPLICIT operator from each IMEX pair
- Schemes:
 - ▶ IMEX3(2)4L[2]SA vs. IMEXRK4(3)7L[2]SA (KC: ANM 2003, 2019a)
 - ▶ IMEX3(2)4L[2]SA vs. ESDIRK4(3)7L[2]SA (KC: ANM 2003, 2019b)
 - ▶ IMEX4(3)6L[2]SA (KC: ANM, 2003)
 - ▶ IMEX5(4)8L[2]SA (KC: ANM, 2019b)

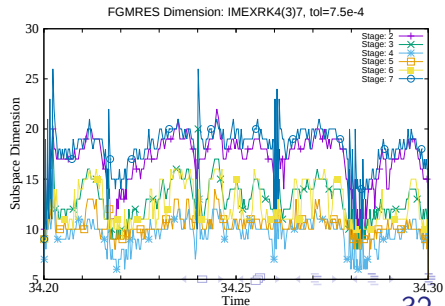
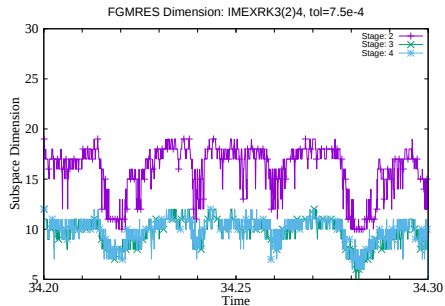
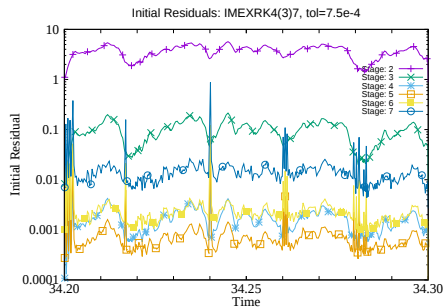
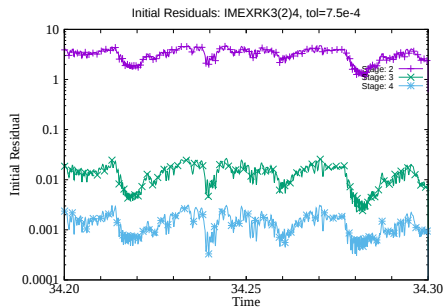
Physical Timestep Used By Schemes



Convergence P3: IMEX324 vs. ESDIRK437



Convergence P4: IMEX324 vs. IMEXRK437



Efficacy of Solver Optimized Schemes: The Verdict is?

- Optimized schemes [RK4(3)7] are twice efficient than RK3(2)4
- Optimized schemes are more 1) accurate, 2) robust → 3) efficient

	P3: Tol=7.5e-5		P3: Tol=7.5e-4		P4: Tol=7.5e-4	
Method	CPUTime	Steps	CPUTime	Steps	CPUTime	Steps
IMEXRK3(2)4	61085	1792	68782	789	96694	1294
IMEXRK4(3)7	30076	434	34577	245	53318	328
ESDIRK4(3)7	35096	563	33799	314	61497	410
ESDIRK4(3)6	49439	664	64654	364		
ESDIRK5(4)8	42348	504	42523	297		

Discussion: Temporal Integrators

- Solver optimized schemes are the clear winner in these comparisons!
 - ▶ Comparisons were performed at fixed accuracy. (Not dt_{max})
- Additional stage: 1) more accurate ($\downarrow 10^2$) with 2) smaller γ ($\frac{1}{4} \rightarrow \frac{1}{8}$)
- Increased accuracy allows huge timesteps ($\uparrow 4$) relative to lower-order schemes.
- First cut for SVP is encouraging. Not yet where I want to be. (FGMRES 10-15/stage. $\rightarrow$, 3-5/stage)
- SVP and optimized γ enables equivalent iteration counts between 3rd- and 4th-order schemes
- Controllers achieved specified accuracy. Current focus: optimality

Thank You For Your Interest

Entropy Stability: an exciting eight years
More work needed on solvers and integrators

Thanks Nail for organizing Minisymposium
Hope you've enjoyed the talks!

Many thanks to all collaborators
and M. Malik (RCA-TTT) for support.