

Modeling and Analysis of Stirling Power Convertors

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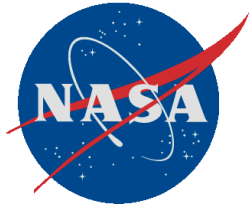
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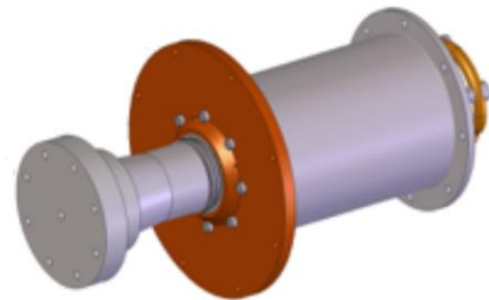
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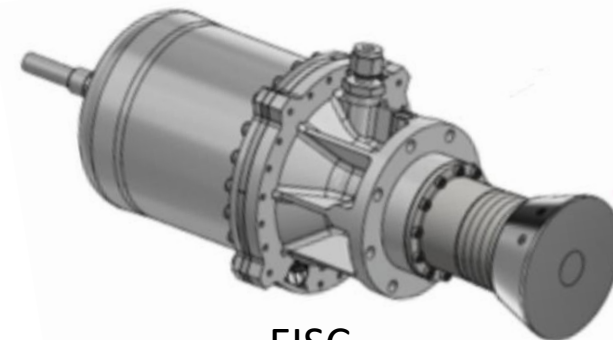


Introduction

- NASA is developing Free-piston Stirling convertors as a steady source of electrical power for future space missions
- NASA Currently has contracts with Sunpower Inc. and American Superconductor (AMSC)
 - Sunpower Inc. is developing the Sunpower Robust Stirling Convertor (SRSC)
 - Both SRSC #1 and #2 delivered to GRC Oct 2020 and under test
 - AMSC is developing the Flexure Isotope Stirling Convertor (FISC)
 - Manufacturer completing testing on FISC pair, anticipate delivery summer 2021
- Both mechanisms used by convertor manufactures provide means of preventing contact in Stirling convertor operation
 - Mechanisms been well proven exceeding 10-14 years of operation at GRC



SRSC

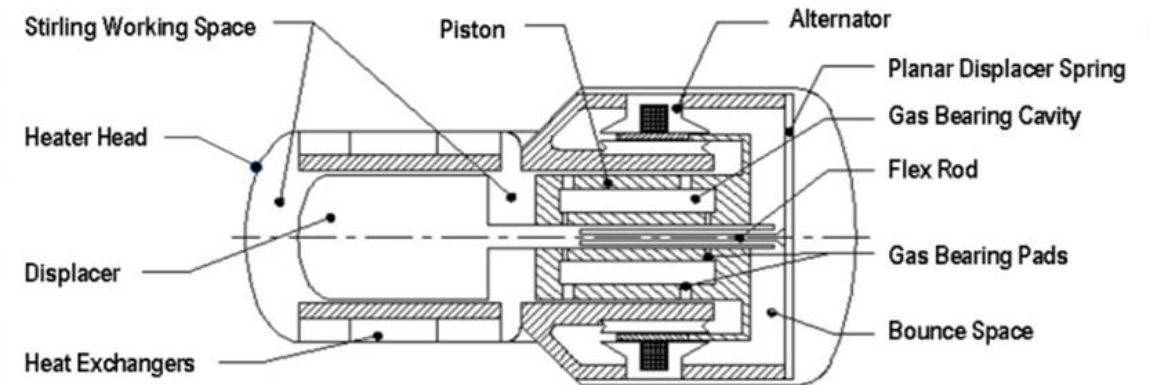


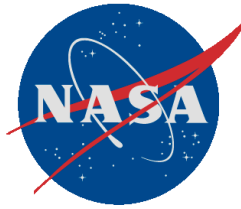
FISC



Introduction

- **Sage** is a one-dimensional object-oriented software package used for modeling and optimizing Stirling convertors
- In this analysis, Sage models are compared against each convertors lab measured performance data and Sankey diagrams are used to illustrate the energy paths through the convertors
- **Ansys Fluent** used to build a 3-D CFD model to examine the Stirling Cycle losses and net heat input of the SRSC convertor
- **Ansys Maxwell** used to create a 3-D axisymmetric model for the SRSC and FISC alternators
- The alternator models calculate terminal voltage, current, electrical power, and efficiency

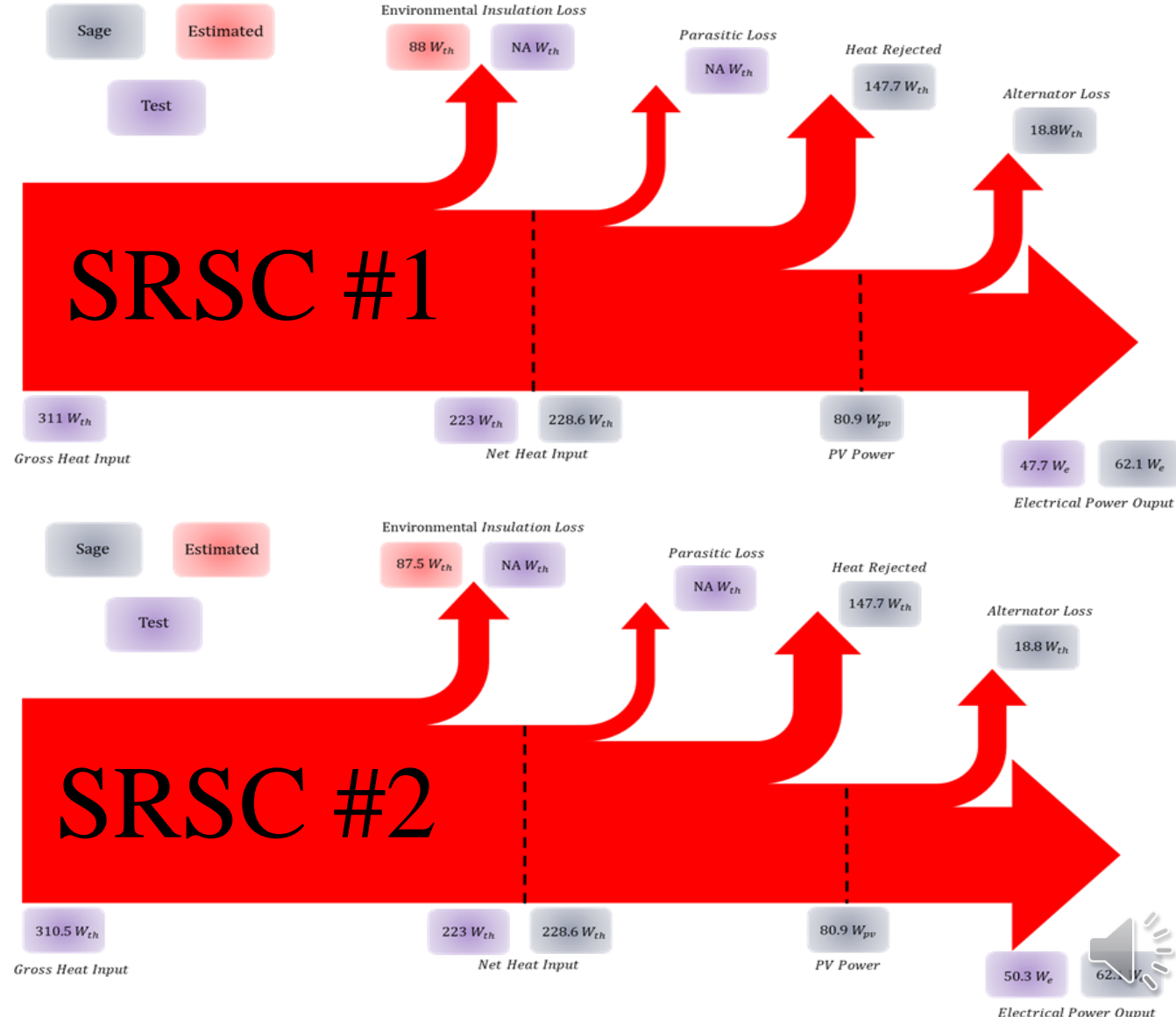


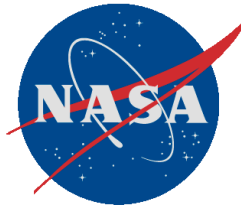


Current SRSC Experimental and Sage Data

Parameter	Definition	Sage Input Value	SRSC #1 Measured Values (08.31.20)	SRSC #2 Measured Values (05.21.20)	Unit
X_p	Piston Amp.	4.20	3.77	3.36	mm
f	Operating Frequency	96	99.6	99.6	Hz
P_{charge}	Charge Pressure (abs.)	3.8e6 (551.14)	4.238e6 (614.7)	4.238e6 (614.7)	Pa (psi)
T_h	Hot End Temperature	720	700.3	699.3	°C
T_k	Cold End Temperature	100	100.3	99.6	°C

Parameter	Definition	Sage Output Value	SRSC #1 Measured Values (08.31.20)	SRSC #2 Measured Values (05.21.20)	Unit
X_d	Displacer Amp.	3.15	3.01	2.97	mm
ϕ_{diff}	Phase Difference	53.92	55.7	58.4	deg
Q_{in}	Net Heat Input	228.56	223	223	Watt
Q_{rej}	Net Heat Rejected	147.68	-	-	Watt
W_{pv}	Piston PV Power	80.88	-	-	Watt
W_{alt}	Electrical Power Out	62.11	47.7	50.3	Watt
η_{conv}	Convertor Efficiency	27.17	21.4	22.6	%

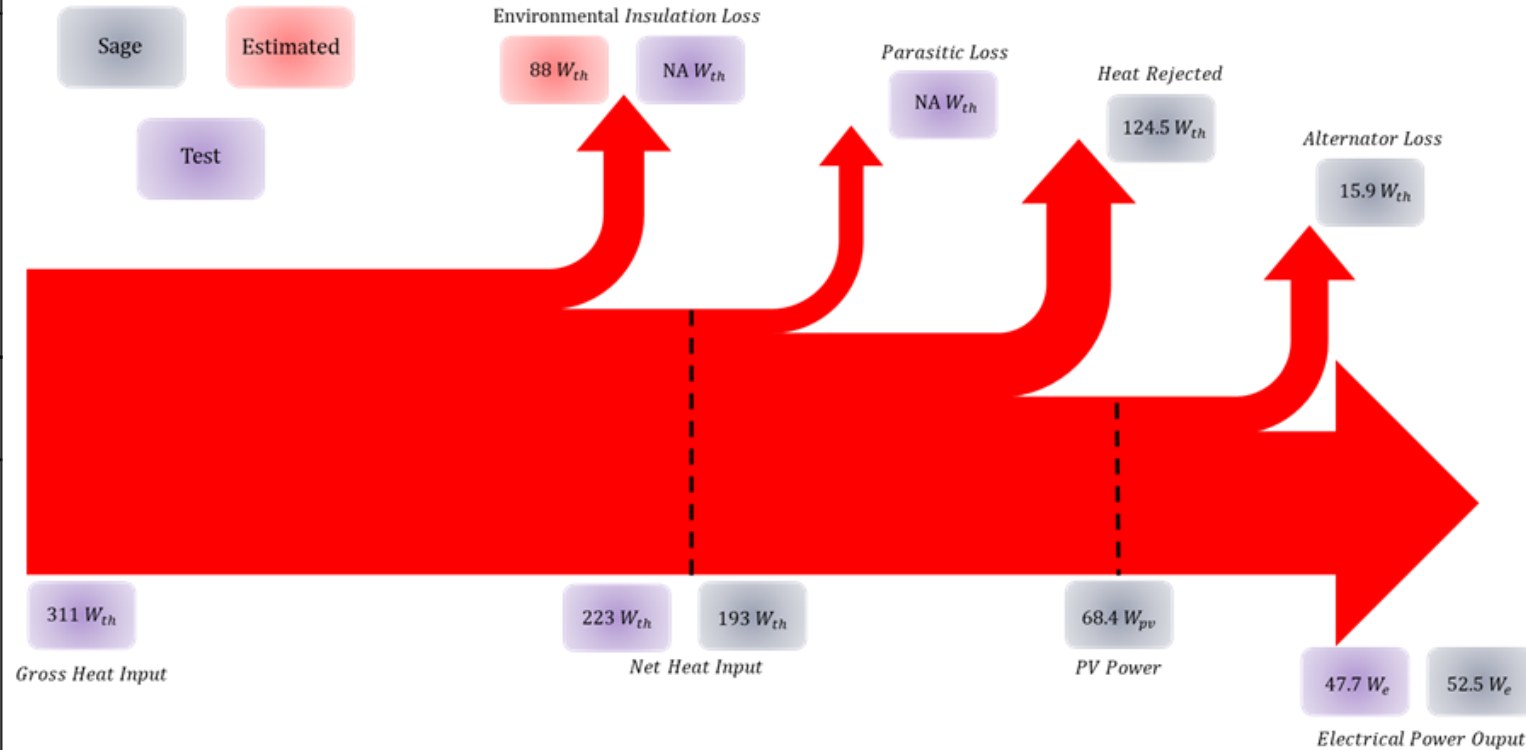


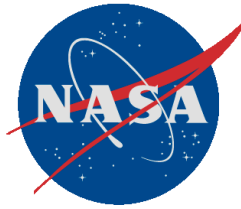


SRSC #1 Calibrated Sage Model

Parameter	Definition	Sage Input	SRSC #1 Measured Values (08.31.20)	Unit
X_p	Piston Amp.	3.77	3.77	mm
f	Operating Frequency	99.6	99.6	Hz
P_{charge}	Charge Pressure (abs.)	4.238e6 (614.7)	4.238e6 (614.7)	Pa (psi)
T_h	Hot End Temperature	700	700.3	°C
T_k	Cold End Temperature	100	100.3	°C

Parameter	Definition	Sage Output	SRSC #1 Measured Values (08.31.20)	Unit	Error (%)
X_d	Displacer Amp.	2.97	3.01	mm	1.4
ϕ_{diff}	Phase Difference	43.6	55.7	deg	21.7
Q_{in}	Net Heat Input	193.0	223	Watt	13.5
Q_{rej}	Net Heat Rejected	124.5	-	Watt	-
W_{pv}	Piston PV Power	68.4	-	Watt	-
W_{alt}	Electrical Power Out	52.5	47.7	Watt	10.1
η_{conv}	Convertor Efficiency	27.2	21.4	%	27.2

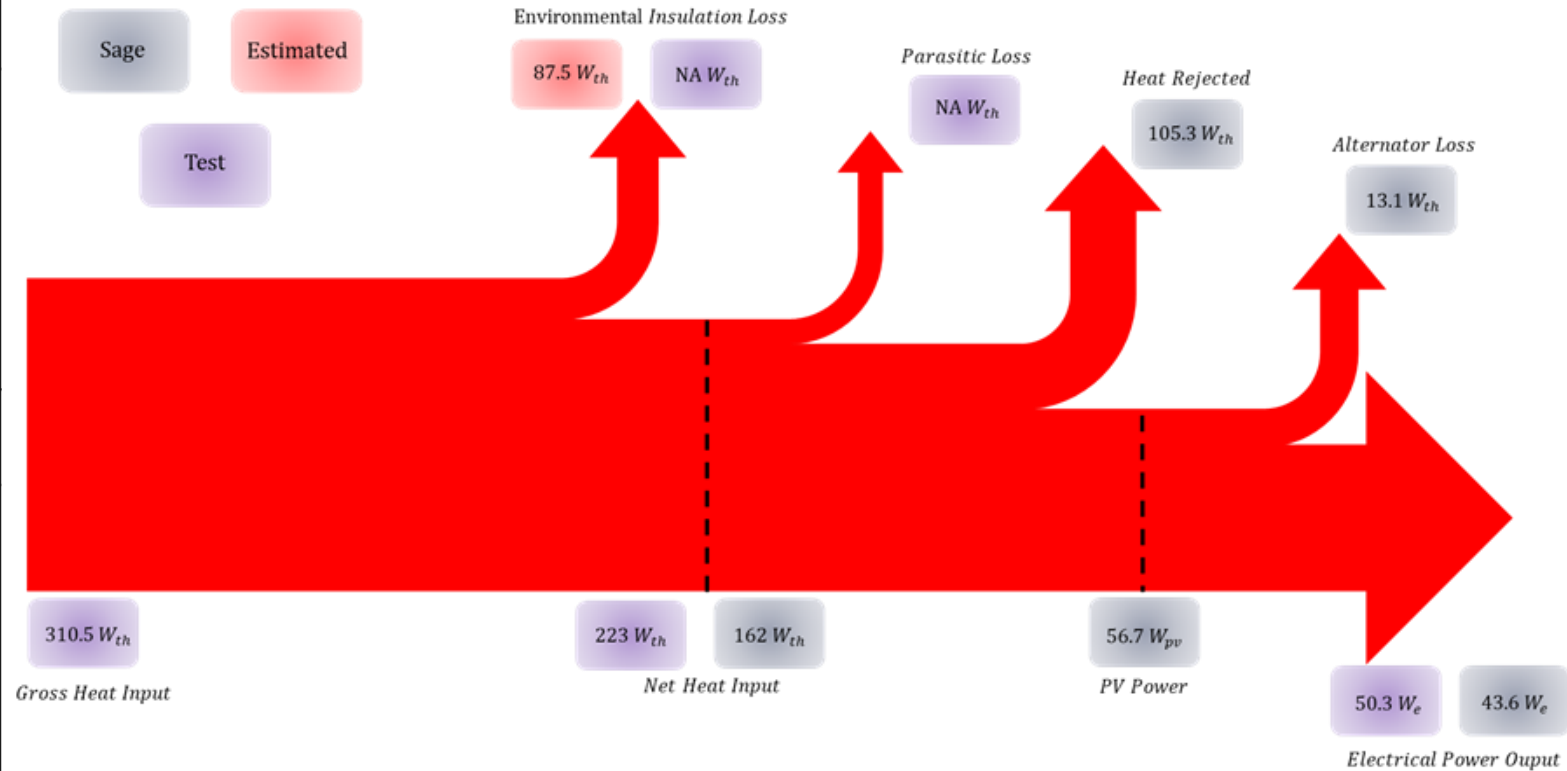


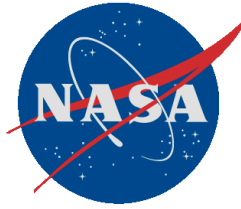


SRSC #2 Calibrated Sage Model

Parameter	Definition	Sage Input	SRSC #2 Measured Values (05.21.20)	Unit
X_p	Piston Amp.	3.36	3.36	mm
f	Operating Frequency	99.6	99.6	Hz
P_{charge}	Charge Pressure (abs.)	4.238e6 (614.7)	4.238e6 (614.7)	Pa (psi)
T_h	Hot End Temperature	700	699.3	°C
T_k	Cold End Temperature	100	99.6	°C

Parameter	Definition	Sage Output	SRSC #2 Measured Values (05.21.20)	Unit	Error (%)
X_d	Displacer Amp.	2.68	2.97	mm	9.8
ϕ_{diff}	Phase Difference	43.8	58.4	deg	24.9
Q_{in}	Net Heat Input	162.0	223	Watt	27.4
Q_{rej}	Net Heat Rejected	105.3	-	Watt	-
W_{pv}	Piston PV Power	56.7	-	Watt	-
W_{alt}	Electrical Power Out	43.6	50.3	Watt	13.4
η_{conv}	Converter Efficiency	26.9	22.6	%	19.0

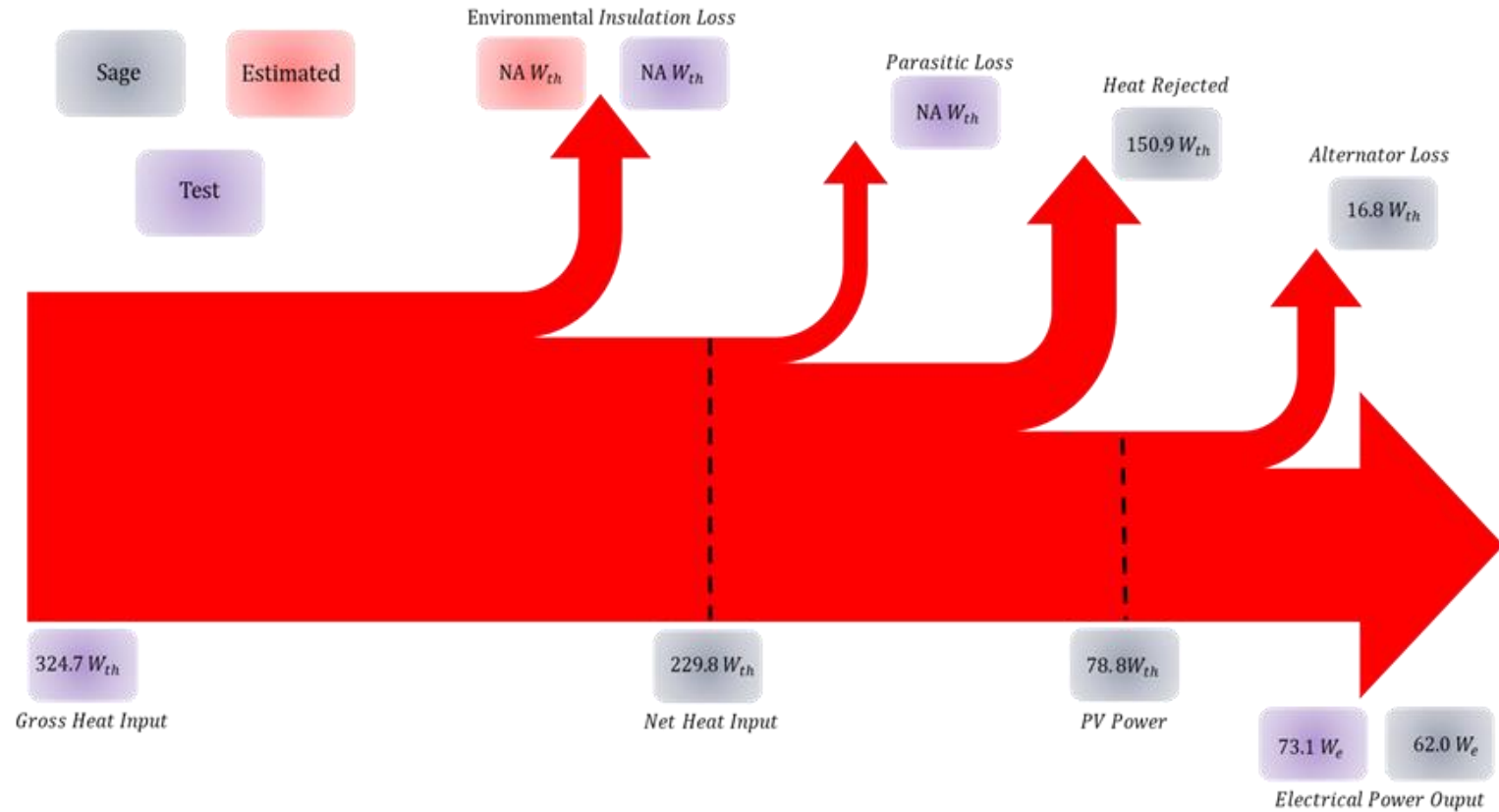


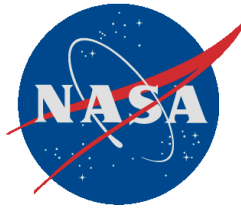


Current FISC Experimental and Sage Data

Parameters	Definition	Sage Input	Measured Values (02.08.21 Results)	Unit
X_p	Piston Amp.	5.55	5.86	mm
f	Operating Frequency	81	82.9	Hz
P_{charge}	Charge Pressure (abs.)	2.5e6 (362.59)	2.48e6 (359.74)	Pa (psi)
T_h	Hot End Temperature	613.8	652.4	°C
T_k	Cold End Temperature	100.1	104.6	°C

Parameter	Definition	Sage Output	Measured Values (02.08.21)	Unit
X_d	Displacer Amp.	2.64	2.87	mm
ϕ_{diff}	Phase Difference	74.7	72.4	deg
Q_{in}	Net Heat Input	229.8	-	Watt
Q_{rej}	Net Heat Rejected	150.9	-	Watt
W_{pv}	Piston PV Power	78.8	-	Watt
W_{alt}	Electrical Power Out	62.0	73.1	Watt
η_{conv}	Convertor Efficiency	27.0	25.4	%

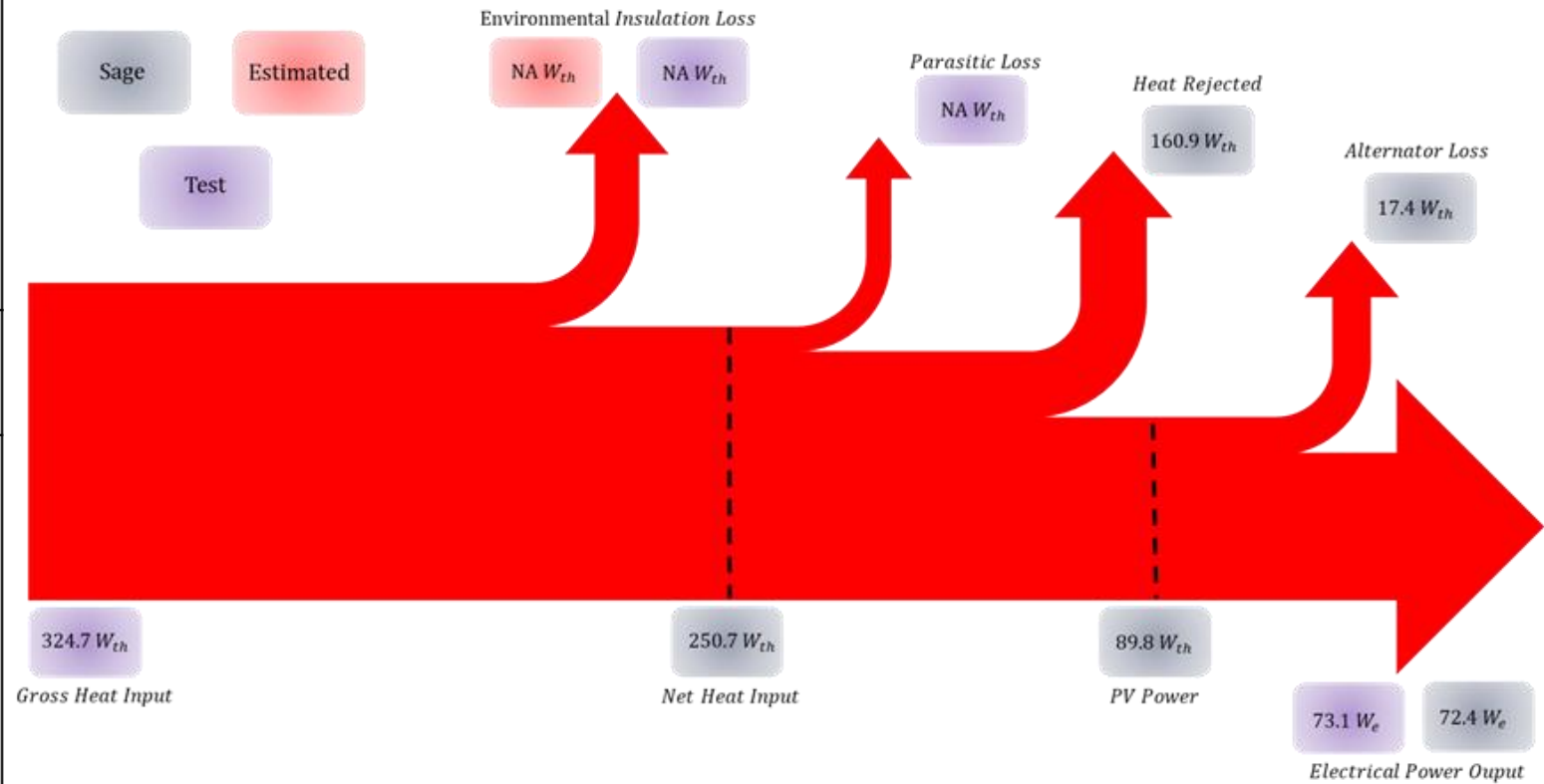


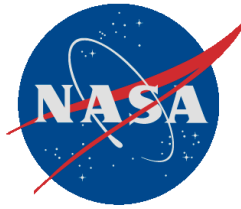


FISC Calibrated Sage Model

Parameters	Definition	Sage Input	Measured Values (02.08.21 Results)	Unit
X_p	Piston Amp.	5.86	5.86	mm
f	Operating Frequency	82.9	82.9	Hz
P_{charge}	Charge Pressure (abs.)	2.48e6 (359.74)	2.48e6 (359.74)	Pa (psi)
T_h	Hot End Temperature	652.4	652.4	°C
T_k	Cold End Temperature	104.6	104.6	°C

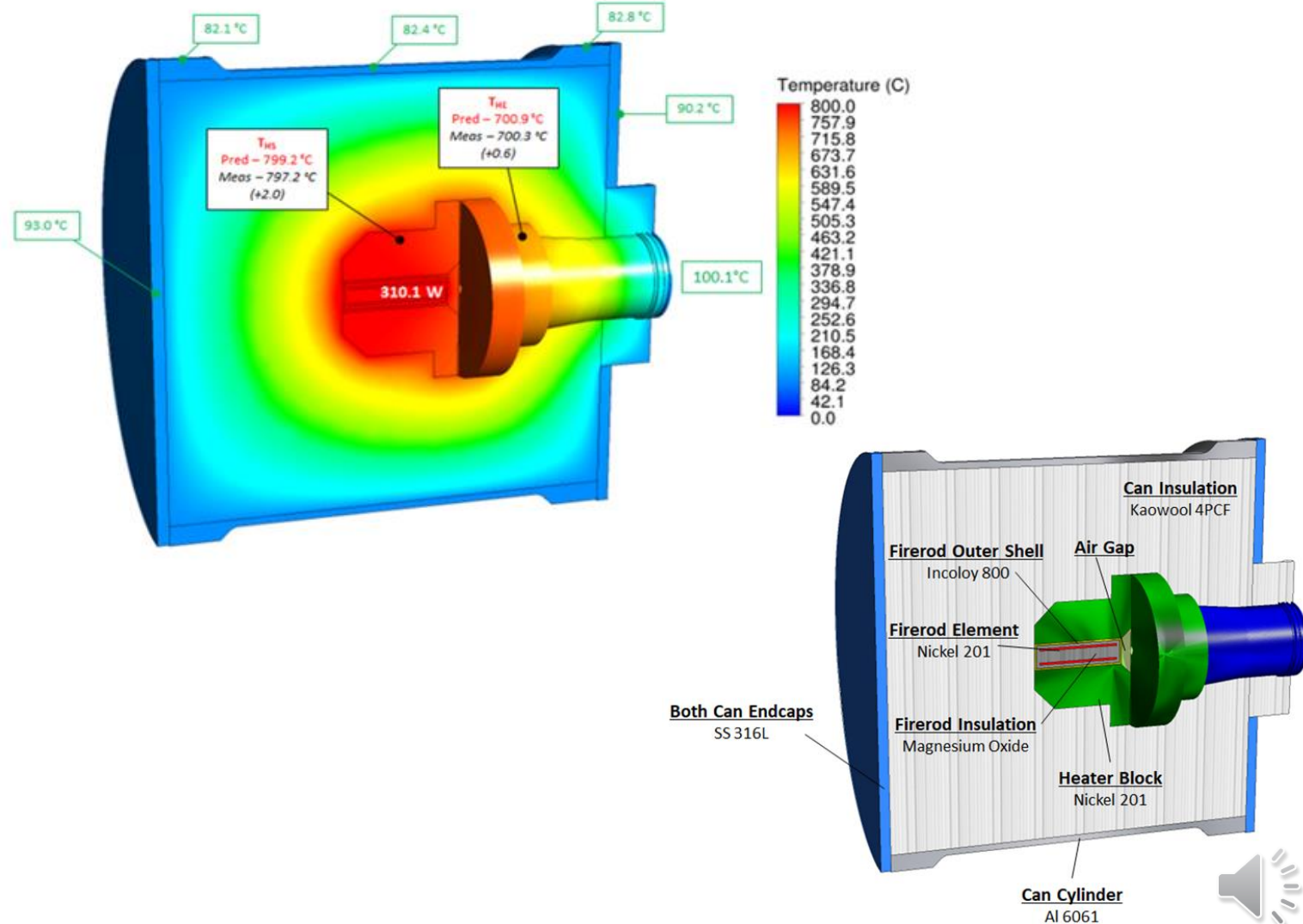
Parameter	Definition	Sage Output	Measured Values (02.08.21 Results)	Unit	Error (%)
X_d	Displacer Amp.	2.84	2.87	mm	1.1
ϕ_{diff}	Phase Difference	64.0	72.4	deg	11.6
Q_{in}	Net Heat Input	250.7	-	Watt	-
Q_{rej}	Net Heat Rejected	160.9	-	Watt	-
W_{pv}	Piston PV Power	89.8	-	Watt	-
W_{alt}	Electrical Power Out	72.4	73.1	Watt	1.0
η_{conv}	Convertor Efficiency	28.9	25.4	%	13.7

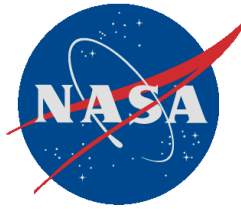




SRSC-Stirling 3D CFD Model Analysis

- The Goal is to use ANSYS's Fluent CFD tools to understand what is going on inside the convertor
 - Since it is difficult to measure some parameters such as Q_{net} and parasitic losses
- Lab gathered temperature boundary conditions, geometry, heat inputs, and material properties are used to determine the net heat input, parasitic losses, and convertor efficiency
- Hence, after heat input and temperature data is collected, Fluent is run in a steady-state mode using a pressure-based solver with the ideal gas law applied to the working fluid





Stirling 3D CFD Model Analysis

Units for values are Watts

- When extracting variables from the converged solution, results indicated that the extracted heat, cold-end losses out of the CSAF, and environmental losses are Q_{esd} , Q_{ce} , and Q_{env}
- These losses can be used to predict parasitic losses and net heat input

Parameters	SRSC # 1	SRSC #2	ANSYS Fluent
Q_{htr}	311.0	310.5	310.1
Q_{ce}	-	-	24.2
Q_{env}	88	87.5	89.4
$Q_{parasitic}$	-	-	113.6
Q_{in}	223	223	220.7

$$Q_{esd} = 196.5 \text{ W}$$

$$Q_{ce} = 24.2 \text{ W}$$

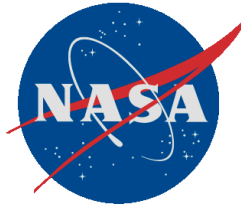
$$Q_{env} = 89.4 \text{ W}$$



$$\text{Parasitic Losses} = Q_{ce} + Q_{env} = 113.6 \text{ W}$$

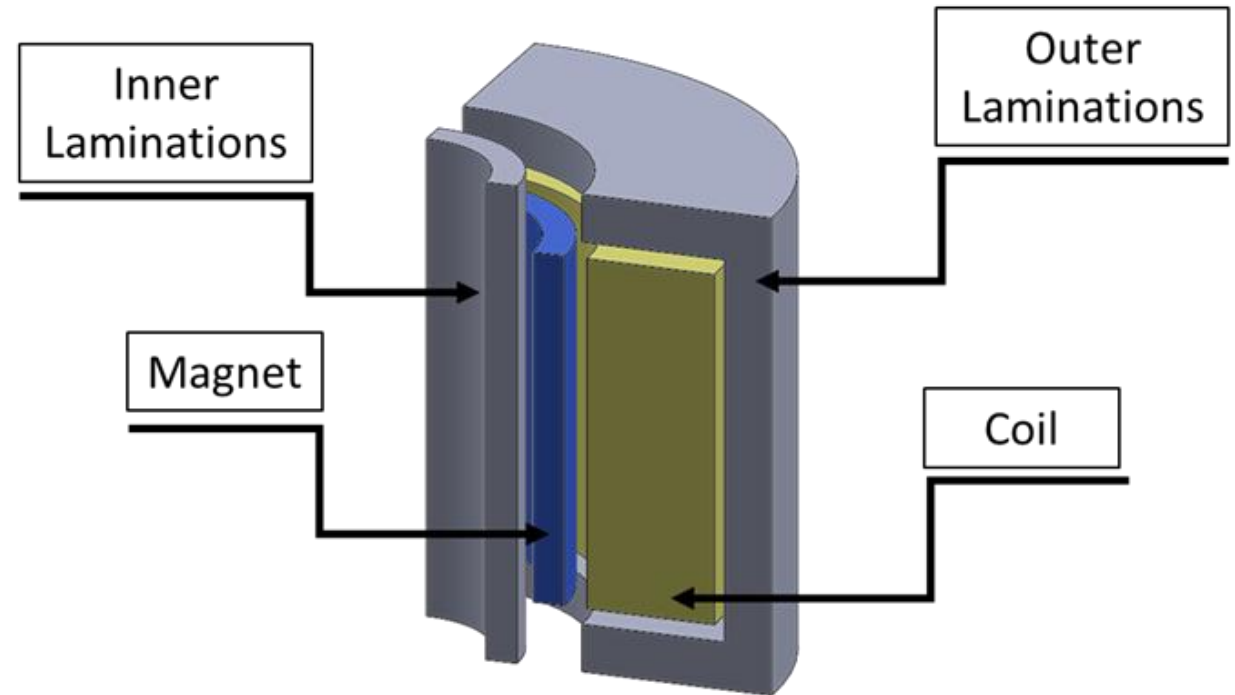
$$\text{Net Heat Input} = Q_{ce} + Q_{esd} = 220.7 \text{ W}$$

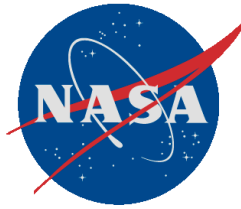




Linear Alternator Modeling

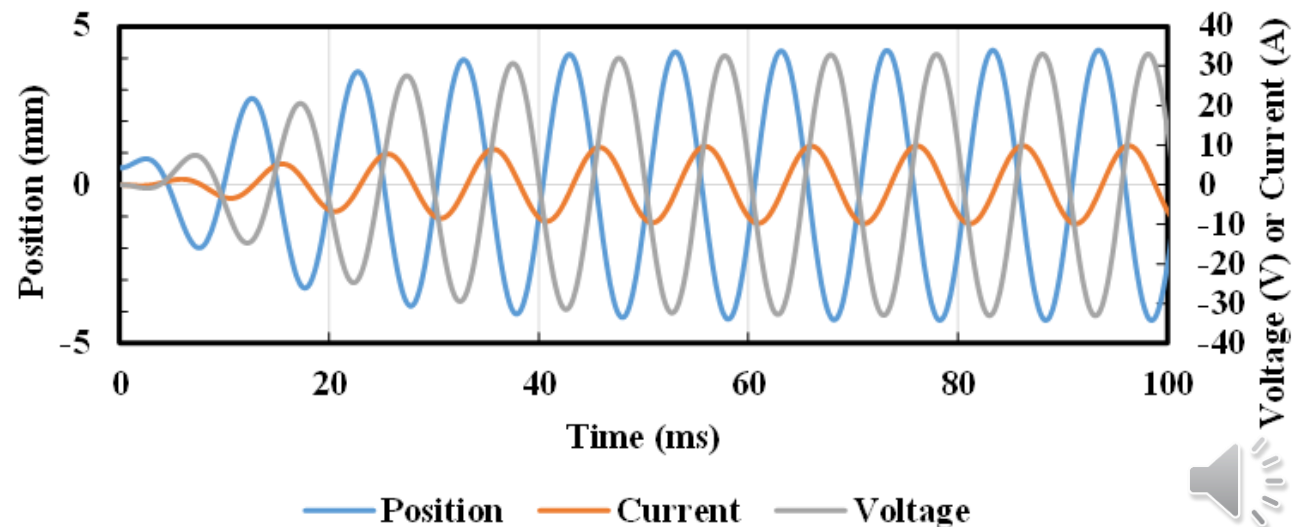
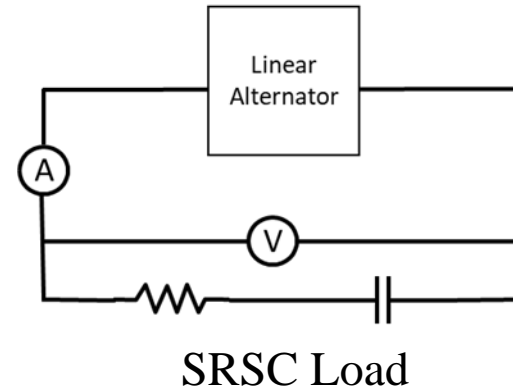
- Transient three-dimensional (3D) axisymmetric Maxwell models were prepared for each of the SRSC and FISC linear alternators
- The key alternator components included for each model are: 1) magnet(s), 2) magnet can (not shown), 3) inner laminations, 4) outer laminations, and 5) coil
- Both alternators are of the moving magnet type, and are made of similar materials

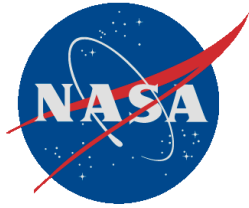




SRSC : LA Performance Verification

- An external load circuit was added to the SRSC LA
 - The values for the load resistance and tuning capacitor were obtained from Sunpower, and reflect the actual load that was used during checkout testing
- In the transient model, the mover was gradually ramped-up to full stroke operation of 8.5 mm to maintain computational stability
 - The model obtained steady-state operation at about 10 cycles
 - The voltage and current values were calculated at the locations indicated as V and A respectively (In the SRSC Load circuit)



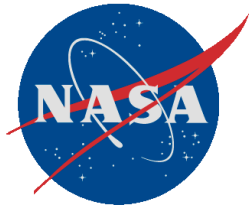


SRSC : LA Performance Verification

- Test data was acquired during the October 11, 2019 convertor checkout testing at Sunpower
- The test data indicated that the engine was operated at 3 Hz above the design frequency of 96 Hz
 - Model frequency was adjusted to match the test operating conditions
- The agreement between the predicted and measured data was good except for the terminal voltage
 - The model did not account for any voltage drops between the coil and convertor output terminals

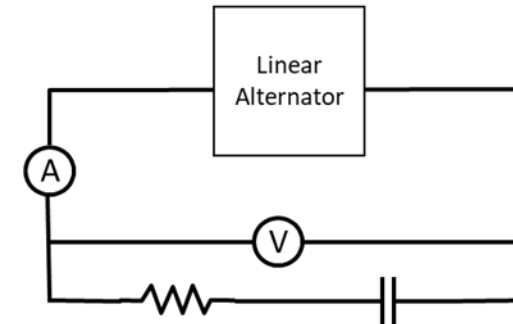
Parameter	GRC Prediction	Test Data	% Error
Piston Amplitude, mm	4.25	4.25	-
Frequency, Hz	99	99	-
Terminal Voltage, V_{rms}	23.3	16.8	38.7 %
Current, A_{rms}	6.98	6.5	7.4 %
Power, W	64.3	63.4	1.4 %
Efficiency, %	80.0	-	-
Total Losses	16.1	-	-



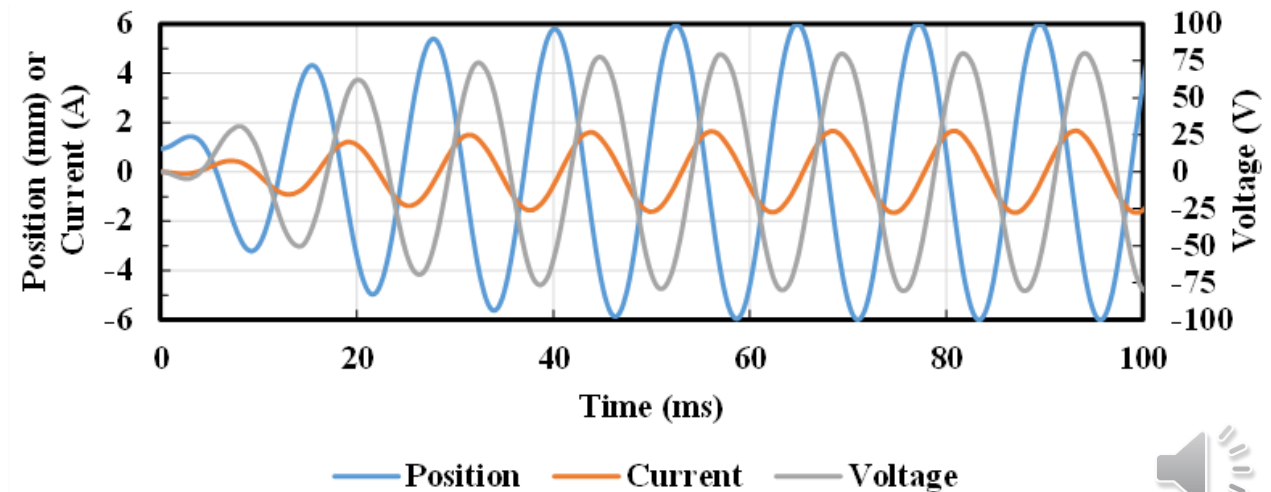


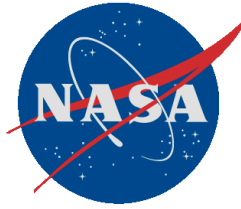
FISC : LA Performance Verification

- A similar load circuit to that of the SRSC was used in the FISC
 - The values for the load resistance and tuning capacitor were obtained from AMSC, and reflect the actual load that was used during checkout testing
- In the transient model, the mover was gradually ramped-up to full stroke operation of 12 mm to maintain computational stability
 - The model obtained steady-state operation at about 10 cycles
 - The voltage and current values were calculated at the locations indicated as V and A respectively (In the FISC Load circuit)



FISC Load



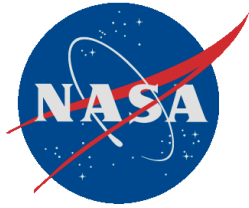


FISC : LA Performance Verification

- Test data was acquired during the October 3, 2019 convertor checkout testing at AMSC
- The test data indicated that the engine was operated at 4 Hz above the design frequency of 81 Hz
 - Model frequency was adjusted to match the test operating conditions
- The generated predictions for the FISC alternator resulted to be in good agreement with the experimental data

Parameter	GRC Prediction	AMSC Test Data (10/3/2019)	% Error
Piston Amplitude, mm	6.0	6.0	-
Frequency, Hz	77	77	-
Terminal Voltage, V_{rms}	54.3	-	-
Current, A_{rms}	1.56	1.54	1.3%
Power, W	67.1	68.6	2.2%
Efficiency, %	92.6	88	5.2%
Total Losses	5.4	-	-





Conclusion

- Sage models were received from both Sunpower and AMSC to interrogate and validate against nominal convertor performance data
 - A difference was noticed between the as received test data and the predictive models.
 - Due to this difference, models were calibrated respectively
 - As a result of errors between test data and the calibrated models further testing, model calibration, and model optimization will be performed

- ANSYS Fluent was used to collect useful data from, SRSC convertors, that is difficult to measure during performance testing
 - Lab temperatures measurements were collected from the cylinder can, CSAF, hot end of the convertor, and heater block surfaces and used in the CFD model
 - Results matched well with in-lab test (Such as : Net Heat Input, and Q_{env})

- ANSYS Maxwell was used to model both SRSC and FISC
 - Model predictions were in good agreement with the test data for both convertor types
 - Except for the SRSC terminal voltage between the model and experimental value were not in good agreement
 - More parameters can be compared in the future as more test data becomes available.

