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LH₂ Storage and Handling Demonstrations Using Active Refrigeration

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Background

- In the 1950's and 1960's DoD and NASA requirements drove the development of large scale LH₂ systems
- Kennedy Space Center has not substantially changed its LH₂ hardware or processes since that time
- Inefficiencies lead to the loss of almost 50% of liquid hydrogen purchased during the shuttle program
- Some technology development work done with densified propellants but never incorporated by NASA

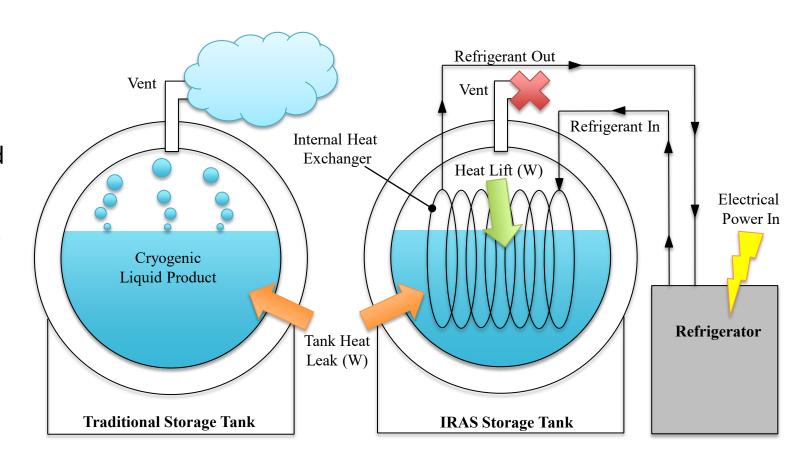




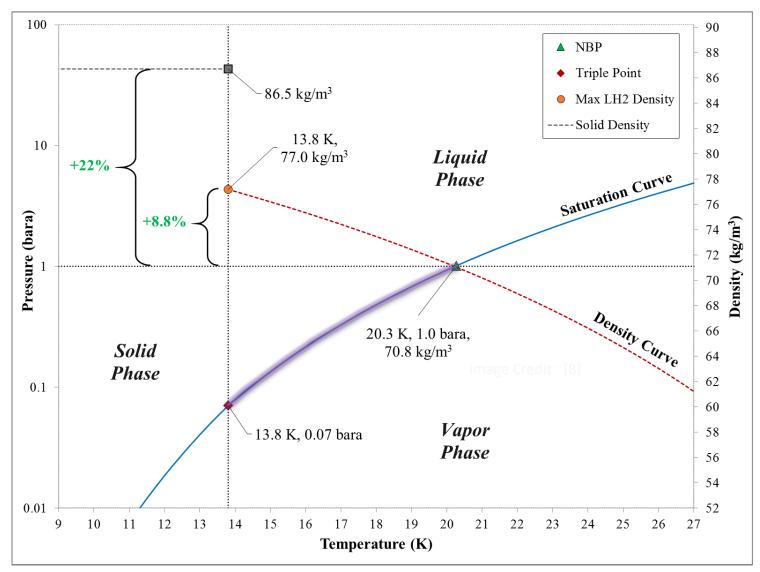


Integrated Refrigeration and Storage (IRAS)

- Interface a cryogenic refrigerator to a liquid hydrogen storage tank via an internal heat exchanger
- Remove energy directly from the liquid to control bulk fluid
- Enables Full Control Storage, including Zero Boil-Off, Densification, and Liquefaction
- NASA and DoE funded small scale LH₂
 IRAS proof of concept demonstration from 2002-2006
- NASA funded IRAS Heat Exchanger characterization tests in 2008-2009



Densification Benefits



Properties of para-hydrogen from RefProp Version 8

- Additional payload to orbit of 4.9% to 17.5% for liquid, up to 26% for slush
- Enables advanced capabilities such as reusability (SpaceX Falcon 9)



 Additional energy storage capacity and enthalpy margin

GODU-LH2 Project

- Ground Operations Demonstration Unit for Liquid Hydrogen (GODU-LH2) project ran from 2012 to 2016
- IRAS tech development and scale-up

Project Goal

Demonstrate cost efficient cryogenic operations using IRAS, on a relevant scale that can be projected onto future Spaceport architectures

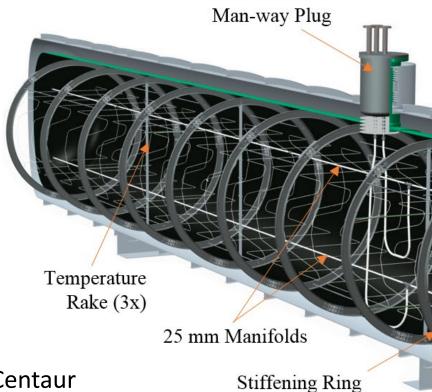
Primary Technical Objectives

- 1. Demonstrate large scale zero loss storage and transfer of LH₂
- 2. Demonstrate hydrogen densification inside the storage tank
- 3. Demonstrate in situ hydrogen liquefaction



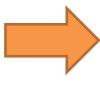
IRAS Tank





Originally constructed in 1991 for Titan-Centaur program

- 33,000 gallons (125, m³) of NBP LH₂ storage
- Modified into an IRAS tank by incorporating an internal HX, stiffening rings, temperature rakes, and man-way feed-through





(9x)

6.4 mm Coil

(40x)

Site Build-Up



September 14th, 2012

Site Build-Up



October 30th, 2014

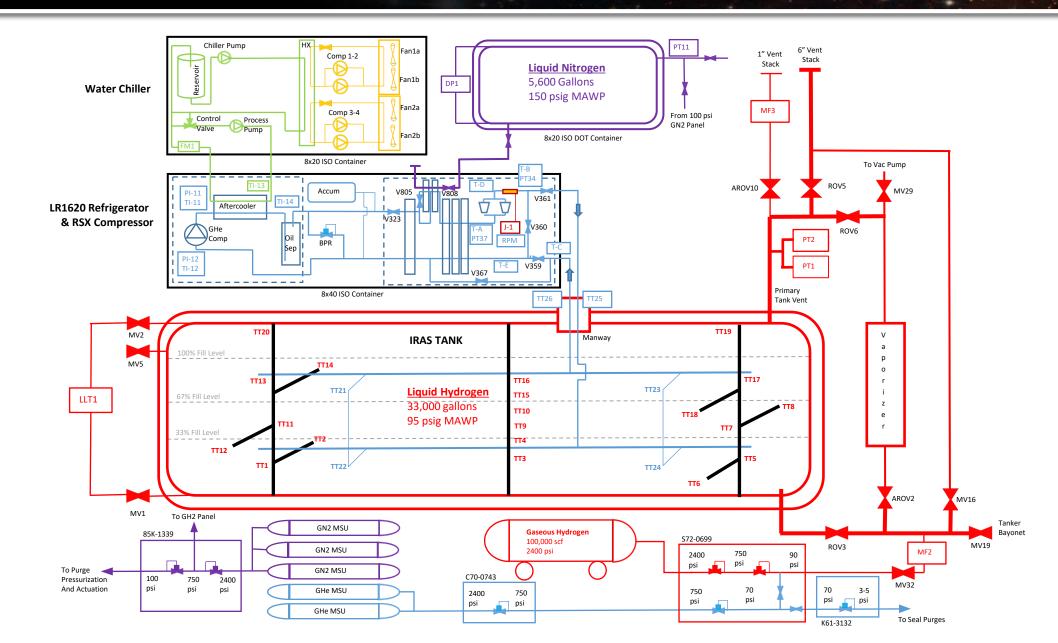
Site Build-Up







GODU-LH2 Functional Diagram



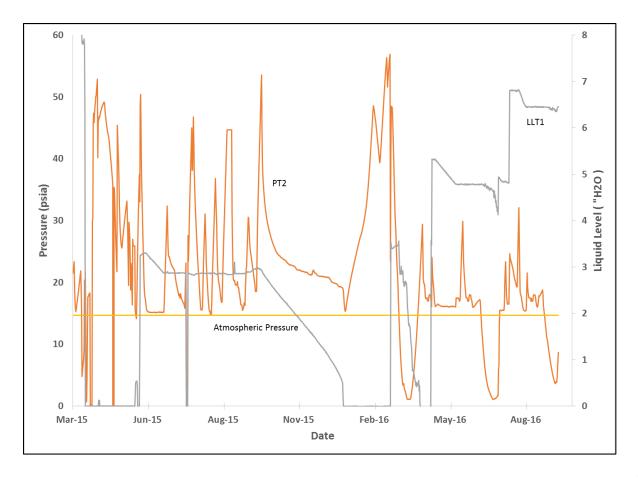
"Bird's-eye View" of GODU-LH2 Site



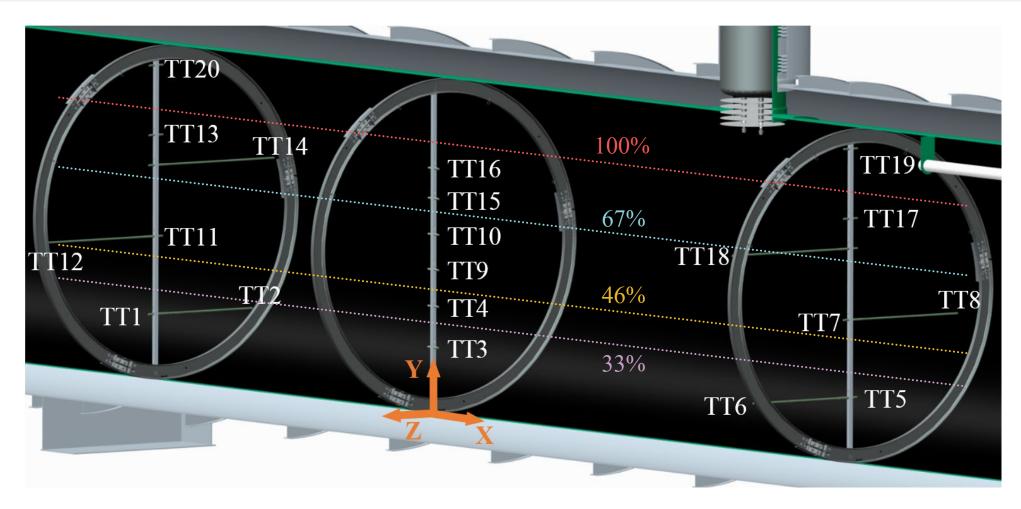
Test Matrix and Timeline

- Completed Test Readiness Review on February 12, 2015
- First tanker offload occurred May 21, 2015
- Refrigerator contamination from October 2015 until March 2016
- Compressed testing from March 2016 until October 2016

Test	Start Date	End Date	
Chilldown	4/9/2015	5/21/2015	
Tanker 1 Offload	5/21/2015	5/29/2015	
Tanker 2 Offload	5/3/2016	5/6/2016	
Tanker 3 Offload	8/3/2016	8/12/2016	
33% Boil Off	5/29/2015	6/19/2015	
66% Boil Off	5/6/2016	5/31/2016	
100% Boil Off	8/14/2016	8/24/2016	
33% ZBO (Press Control)	4/25/2016	5/3/2016	
66% ZBO (Press Control)	6/12/2016	6/21/2016	
100% ZBO (Press Control)	8/25/2016	9/6/2016	
33% ZBO (Temp Control)	6/23/2015	7/13/2015	
66% ZBO (Temp Control)	6/21/2016	6/29/2016	
100% ZBO (Temp Control)	9/6/2016	9/12/2016	
33% ZBO (Duty Cycle)	8/4/2015	8/11/2015	
66% ZBO (Duty Cycle)	6/5/2016	6/13/2016	
100% ZBO (Duty Cycle)	8/12/2016	8/16/2016	
33% Densification	3/24/2016	4/21/2016	
66% Densification	6/29/2016	7/23/2016	
100% Densification	9/12/2016	10/5/2016	
0% Liquefaction	4/9/2015	5/21/2015	
33% Liquefaction	9/23/2015	10/8/2015	
66% Liquefaction	7/22/2016	8/2/2016	



Inner Tank Instrumentation



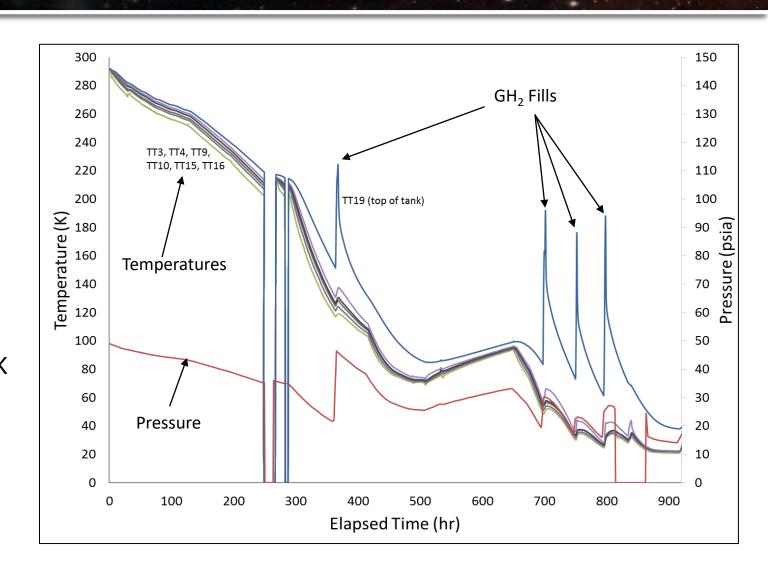
Accuracies

Diodes: ±0.5 K from 450 K to 25 K, and ±0.1 K from 25 K to 1.5 K

Pressure Transducers: ±6.89 kPa (1% of full scale)

Zero-Loss Tank Chilldown Test Results

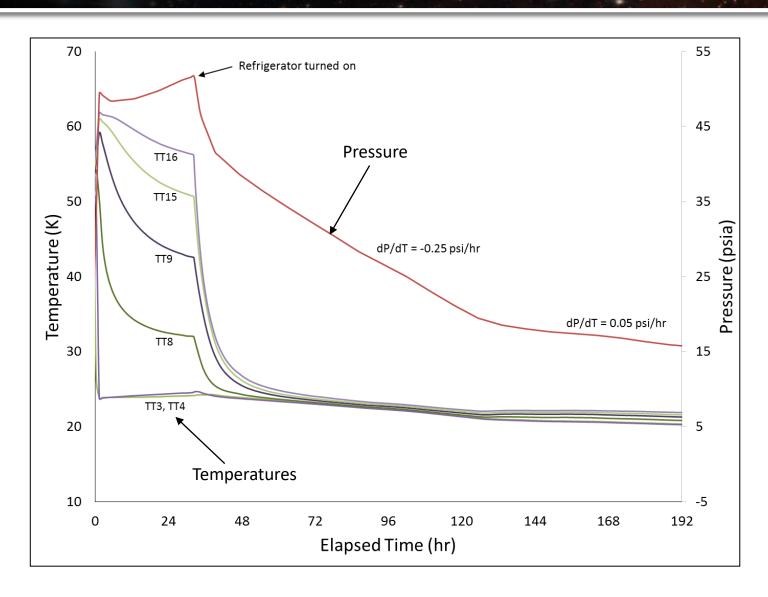
- Initial Conditions
 - 99.95% GH₂ at 300 K and 40 psia.
- Lock up tank and turn on refrigerator at T-0.
- Add GH₂ as tank pressure decreases
- Final Conditions
 - Tank near isothermal at 20.8K 22.4 K and 14.7 psia
 - Saturated vapor with condensation on HX tubing
- Multiple lessons learned would decrease total timeline in the future



Conclusion: IRAS enables zero-loss chilldown of a large cryogenic vessel

Zero-Loss Tanker Offload Test Results

- Based on STS Program data, 13% of purchased LH₂ is lost due to transport and offload inefficiency
- Heat from transport and line chilldown can be removed by refrigerator, allowing no loss offload
- Zero-loss tanker offloads were achieved at 33%, 67%, and 100% fill levels

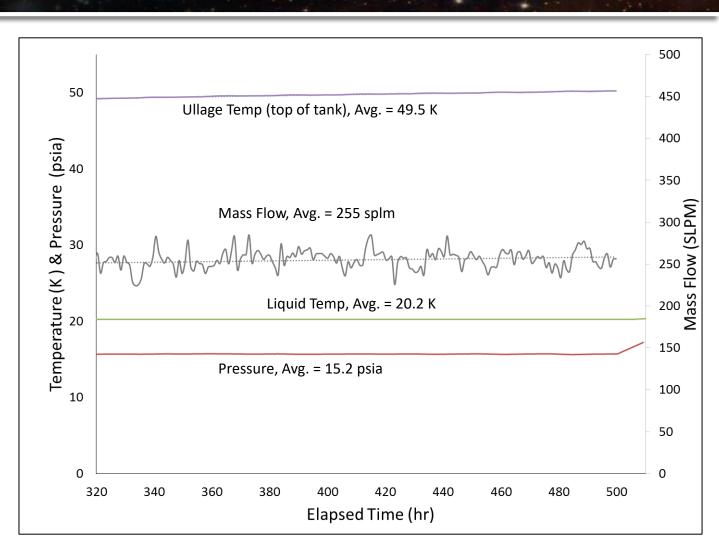


Boil-off Heat Leak Test Results

- Boil off testing to quantify heat leak was conducted at 3 fill levels
- Vented thru control valve and mass flow meter
- Pre-test analysis estimated 300 W

$$\dot{\mathbf{Q}} = \dot{\mathbf{m}} * \{\mathbf{h}_{fg} + (\mathbf{h}_{ullage} - \mathbf{h}_{sat,vapor})\} [\mathbf{W}]$$

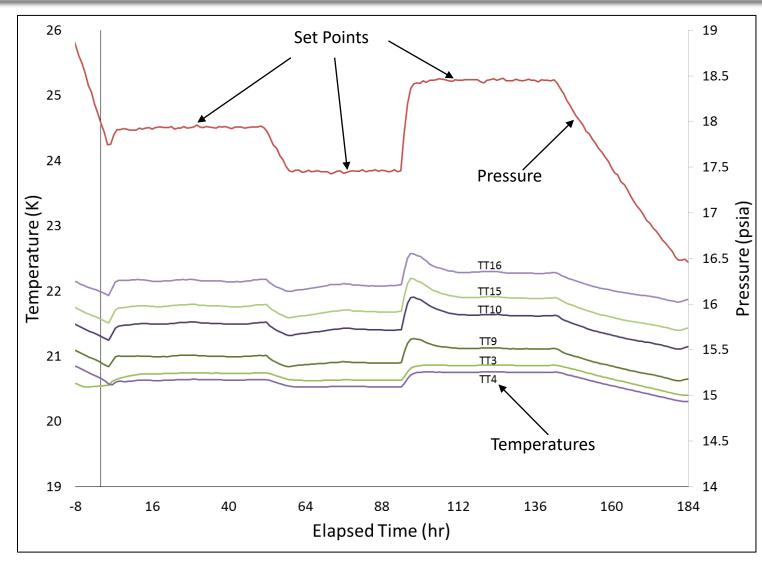
Level	MF3	PT2	TT19	Q_L	Q_V	Q
%	(slm)	(psia)	(K)	(W)	(W)	(W)
33	255	15.2	49.5	170	120	290
67	295	16.6	41.3	196	100	296
100	351	15.9	34.5	234	81	315



Conclusion: Tank heat leak was quantified at three fill levels and agreed closely with pretest estimates

ZBO Pressure Control Test Results

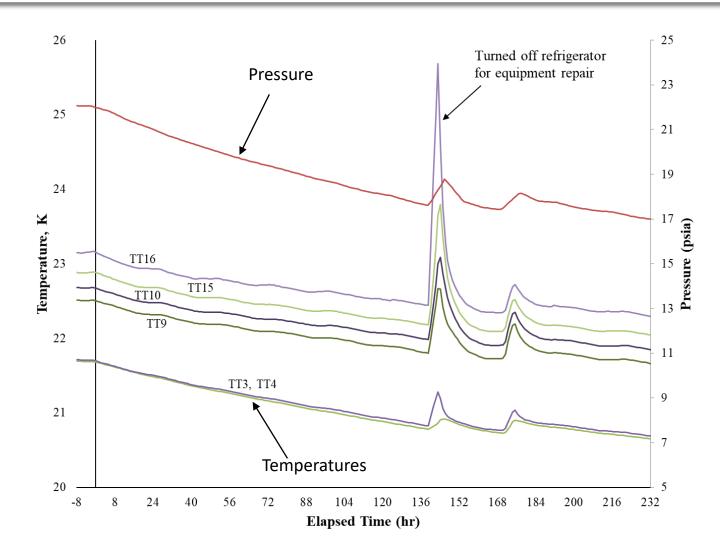
- GODU-LH2 software controlled refrigerator to achieve and maintain IRAS tank pressure set-point.
- No LN₂ pre-cooling used
- Approach set points from above and below
- Pressure stability +/- 0.5%
 for all three fill levels
- Near isothermal temperature profile following saturation line



Conclusion: IRAS using tank pressure control achieves ZBO and provides complete control over the state of the fluid

ZBO Temperature Control Test Results

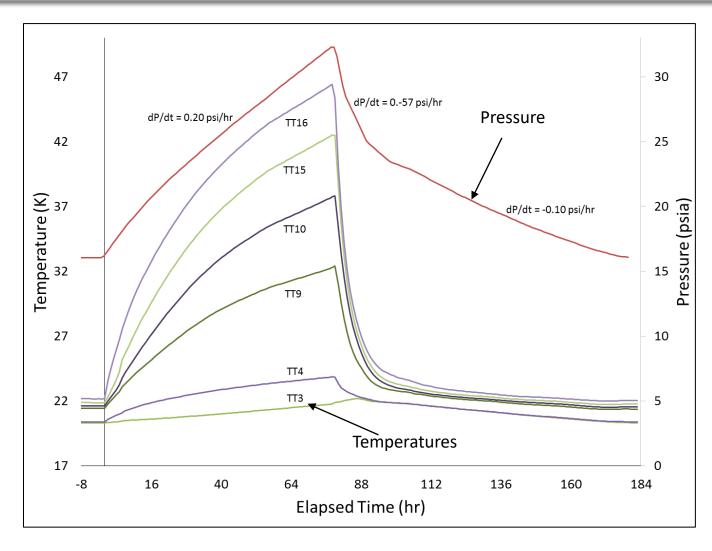
- Linde software controlled refrigerator to achieve and maintain constant helium supply temperature.
- No LN₂ pre-cooling used
- Helium supply temperature response fast and accurate
- But LH₂ takes long time period to reach equilibrium state



Conclusion: IRAS using supply temperature control achieves ZBO but takes a long time to reach LH₂ equilibrium state

ZBO Duty Cycle Test Results

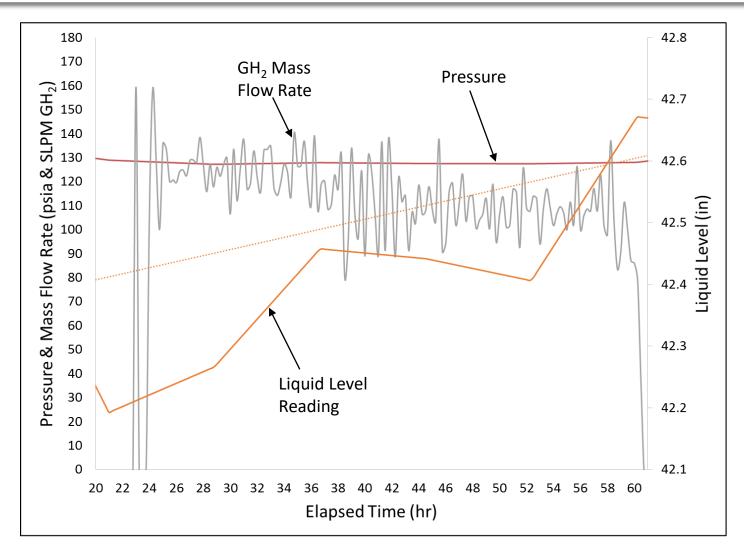
- ZBO achieved in batch processes by turning on and off the refrigerator as required
- Testing was both accidental and purposeful
- Minimum electrical cost but depends on multiple start/stop cycles of cryogenic equipment
- Duty cycle varied from 1.13
 (33%) to 1.16 (67%) to 3.6
 (100%) on/off with no LN₂
 precooling



Conclusion: IRAS using duty cycling of the refrigerator achieves ZBO with minimal energy but provides no control of LH₂ state

Liquefaction Test Results

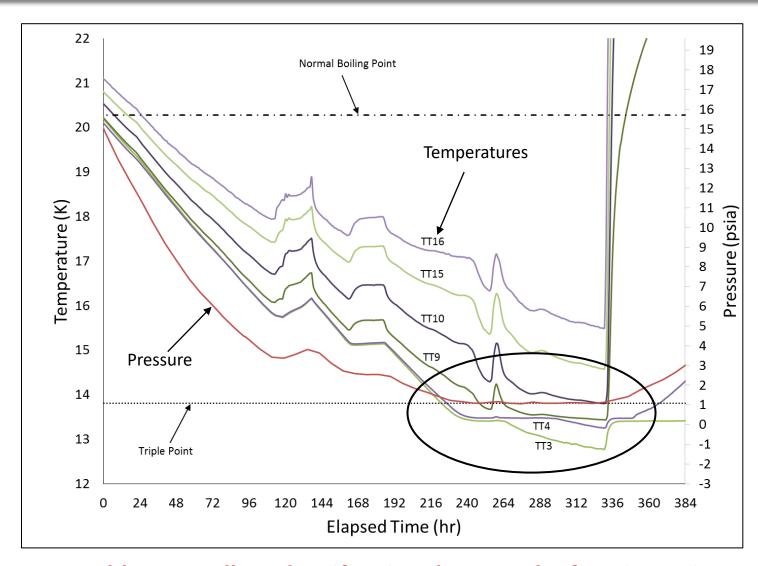
- GH₂ was controlled using a mass flow controller until the tank pressure remained constant.
- <u>NOT</u> optimized for liquefaction. GH₂ was fed in at ambient temperature.
- Using LN₂ pre-cooling, roughly 78 gal of LH₂ was produced during the test.



Conclusion: Hydrogen liquefaction was achieved using IRAS, though the current system was not optimized for yield

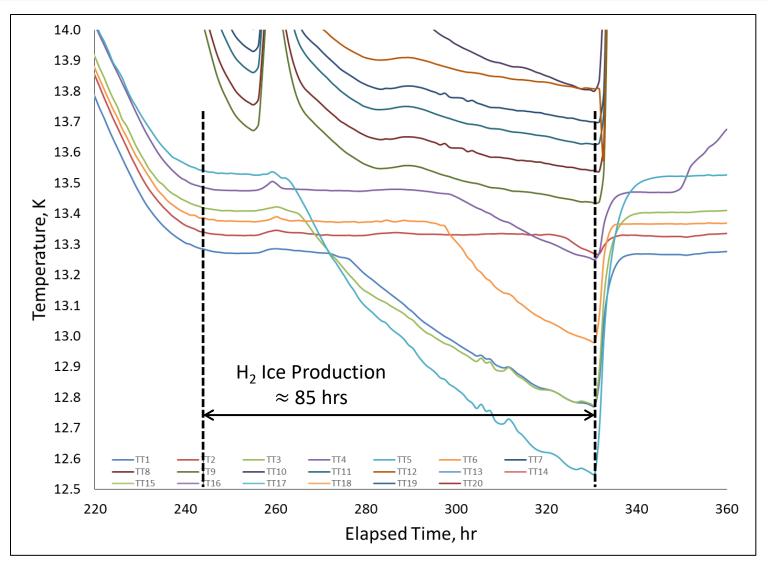
Densification Test Results

- System performance <u>exceeded</u> <u>expectations</u>! Min temp was expected to be ≈15 K
- Fridge ran with LN₂ precooling, and densified 13,000 gallons of LH₂ for 14 days.
- LH₂ cooled <u>below the triple point</u>.
 Minimum temp recorded was
 12.6 K (-437°F)
- Estimated that 3,700 lb of hydrogen ice was formed during the course of testing; or about 5,100 gal



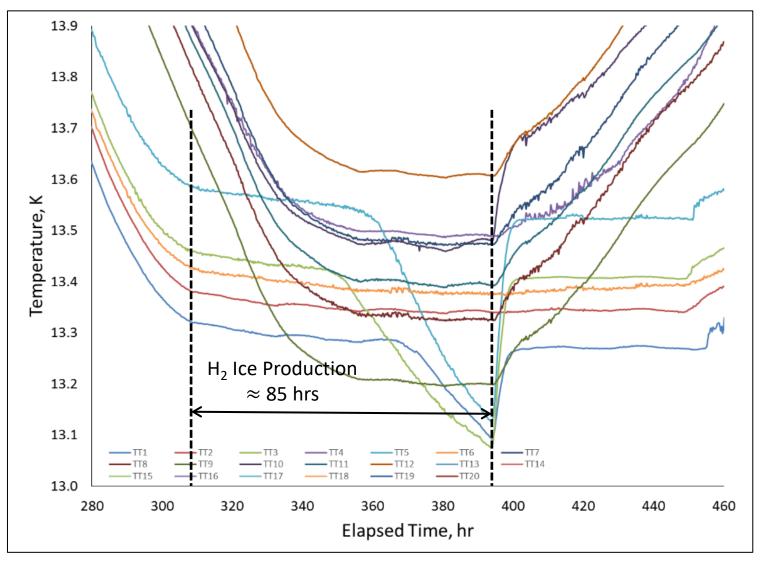
Conclusion: IRAS enables propellant densification down to the freezing point

Solid Hydrogen Production



46% LH₂ Fill Level Test

Solid Hydrogen Production



67% LH₂ Fill Level Test

Conclusions

- The GODU-LH2 system successfully met all test objectives at the 33%, 46%, 67%, and 100% tank fill levels
- Complete control over the state of the fluid has been demonstrated using Integrated Refrigeration and Storage (IRAS)
 - First large-scale demonstration of Full Control Storage of LH₂
 - Almost any desired point within the liquid phase envelop can essentially be "dialed in" and maintained indefinitely
- System can also be used to produce densified/slush hydrogen in large quantities

Current Status of the System

- Refrigeration system consolidated into a single 40' shipping container
- IRAS tank and fridge currently installed at Test Stand 300 at NASA-MSFC in Alabama for an upcoming densification loading test







Thank you for your attention!

Questions?



Image References

- 1. Krenn, A., et. al., 2015, The safe removal of frozen air from the annulus of an LH2 storage tank, *Advances in Cryogenic Engineering, AIP Conference Proceedings*, doi:10.1088/1757-899X/101/1/012031
- 2. http://www.apolloarchive.com/apollo_gallery.html
- 3. https://www.nasa.gov/mission-pages/shuttle/flyout/launchpadflyout.html
- 4. https://www.teslarati.com/spacex-nears-third-falcon-heavy-launch/

Technical References

- Kennedy Engineering Academy #144: Final Results of the Ground Operations Demonstration Unit for Liquid Hydrogen (GODU-LH2) Project, Kennedy Space Center, FL, March 28th, 2017 (https://ntrs.nasa.gov/citations/20170002713)
- 2. IRAS Maximizes Cryogenic Process Efficiency—Notardonato, Fesmire, & Swanger, Cold Facts magazine, Vol 33, Number 3, June 2017 (https://cryogenicsociety.org/35879/news/iras-maximizes-cryogenic-process-efficiency/)
- 3. New Technology at NASA Offers Full Control Of Cryogenic Liquids—Cold Facts magazine, Vol 35, Number 1, February 2019 (https://cryogenicsociety.org/36886/news/new-technology-at-nasa-offers-full-control-of-cryogenic-liquids/)
- 4. ASME Section VIII Recertification of a 33,000 Gallon Vacuum-Jacketed LH2 Storage Vessel for Densified Hydrogen Testing at NASA Kennedy Space Center, Swanger A. M., Notardonato W. U., and Jumper K. M., Proceeding of the ASME Pressure Vessels and Piping Conference, Massachusetts, Boston, PVP2015-45625, (2015)
- 5. Modification of a Liquid Hydrogen Tank for Integrated Refrigeration and Storage, Swanger A. M., Jumper, K. M., Fesmire J. E., and Notardonato W. U., Advances in Cryogenic Engineering, IOP Conf. Series: Materials Science and Engineering 101 (2015) 012080 doi:10.1088/1757-899X/101/1/012080
- 6. Integrated Refrigeration and Storage for Advanced Liquid Hydrogen Operations, Swanger A. M., Notardonato W. U., Johnson W. L., and Tomsik T. M., Cryocooler 19: Proceedings of the 2016 International Cryocooler Conference, San Diego, California, 513-522 (2016)
- 7. Large scale production of densified hydrogen to the triple point and below, Swanger A. M., Notardonato W. U., Fesmire J. E., Jumper K. M., Johnson W. L., and Tomsik T. M., Advances in Cryogenic Engineering, IOP Conf. Series: Materials Science and Engineering 101 (2017)
- 8. Zero boil-off for large scale liquid hydrogen tank using integrated refrigeration and storage, Notardonato W. U., Swanger A. M., Fesmire J. E., Jumper K. M., Johnson W. L., and Tomsik T. M., Advances in Cryogenic Engineering, IOP Conf. Series: Materials Science and Engineering 101, (2017)
- 9. Final test results for the ground operations demonstration unit for liquid hydrogen, Notardonato W. U., Swanger A. M., Fesmire J. E., Jumper K. M., Johnson W. L., and Tomsik T.M., Cryogenics, Volume 88, (2017), Pages 147-155, ISSN 0011-2275, https://doi.org/10.1016/j.cryogenics.2017.10.008.