

Planning for Airborne Schlieren Imaging of X-59 QueSST

James T Heineck

NASA Ames Research Center

Edward A. Haering Jr., Daniel W. Banks, Troy Robillos

NASA Armstrong Flight Research Center

Nathanial T. Smith

Metis Technology Solutions, Inc.

SciTech Forum and Exhibition, San Diego, CA and Online, January 3-7, 2021

Published by the American Institute of Aeronautics and Astronautics, Inc.

Planning for Airborne Schlieren Imaging of X-59 QueSST

Agenda:

- Introduction to the Task

 - X59 Low Boom Flight Demonstrator mission (LBFD)*

 - Data requirements for schlieren imaging*

 - Three potential methods*

- Testing for the vibration environment

- Testing for the pressure/temperature environment

- Current state of hardware and software

- Summation

Motivation

- NASA Commercial Supersonic Technology / FDC
 - Desire for a schlieren system for full scale aircraft in flight to visualize shockwaves generated by supersonic aircraft
 - Primary use for X-59 Quiet Supersonic Technology demonstrator
 - Validate/refine shock modeling for low boom airframe design
 - MUST image downward propagating shocks



Photo Credit: Lockheed Martin

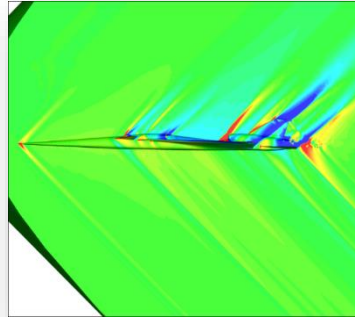
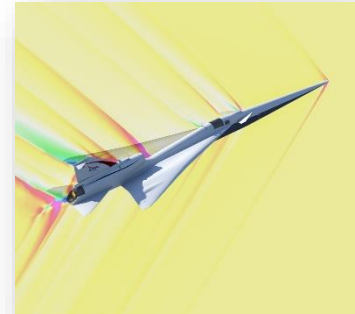


Photo Credit: NASA/James Jensen



Credit: Lockheed Martin and NASA

Three In-Flight Schlieren Methods

AirBOS, Air-to-air Background Oriented Schlieren

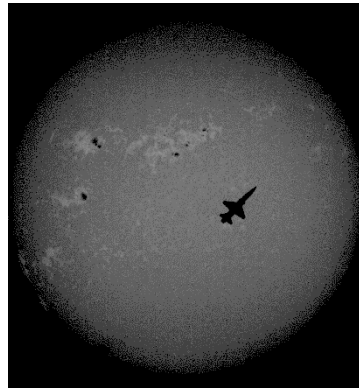
- Downward looking at desert bushes or ocean sun glint
- Wide view, very detailed, easier flying
- Can fly in formation, or from slow observer

Profile view requires aggressive maneuvers to get lower side shocks



- BOSCO, BOS using Celestial Objects
 - Spectral imaging gives visual texture to the sun
 - Ground-to-air currently
 - Potential for lower side shocks

0.5 deg field of view

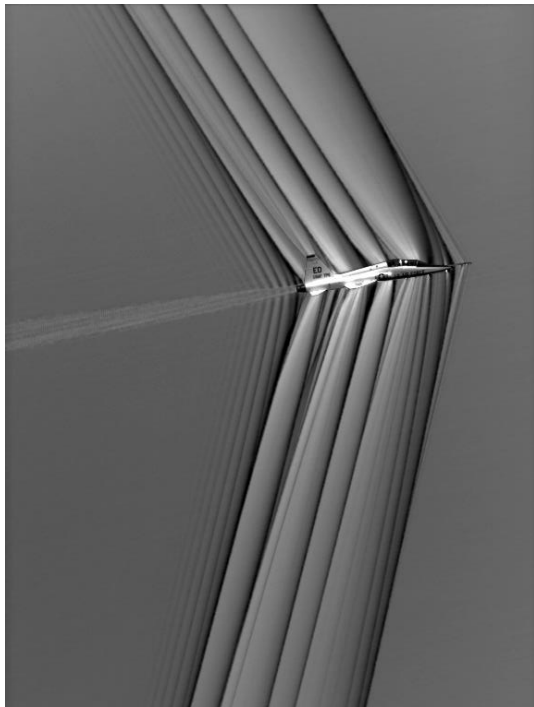


- Sun-edge Distortion
 - Capture the perimeter of Sun being distorted with TR imaging
 - Extract from each frame 10-20 pixel rings, align sequence
 - Same flight planning as BOSCO, more forgiving

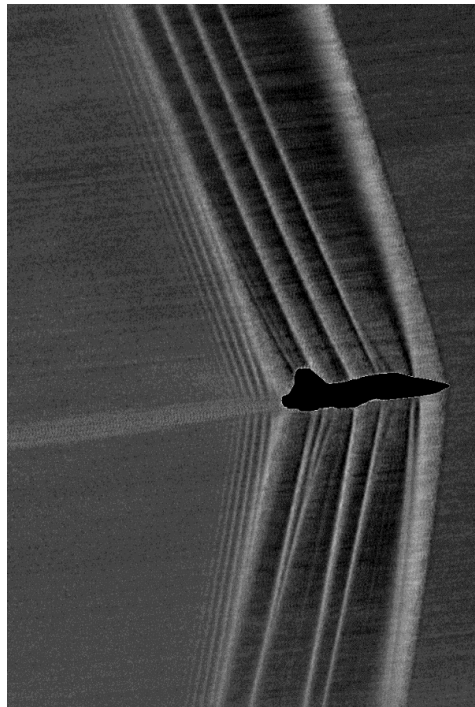
Least amount of detail of the shocks, 0.5 deg fov



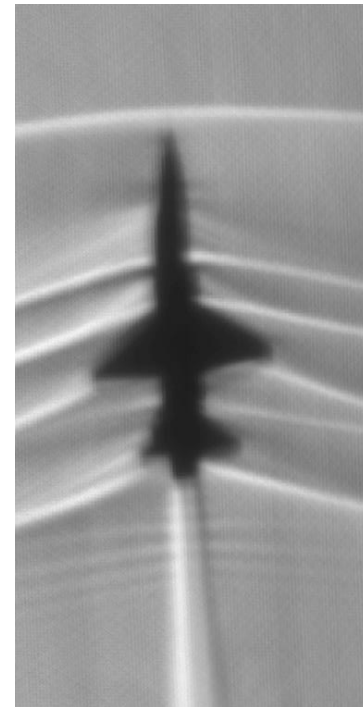
Examples of the Three Methods of a T-38



AirBOS



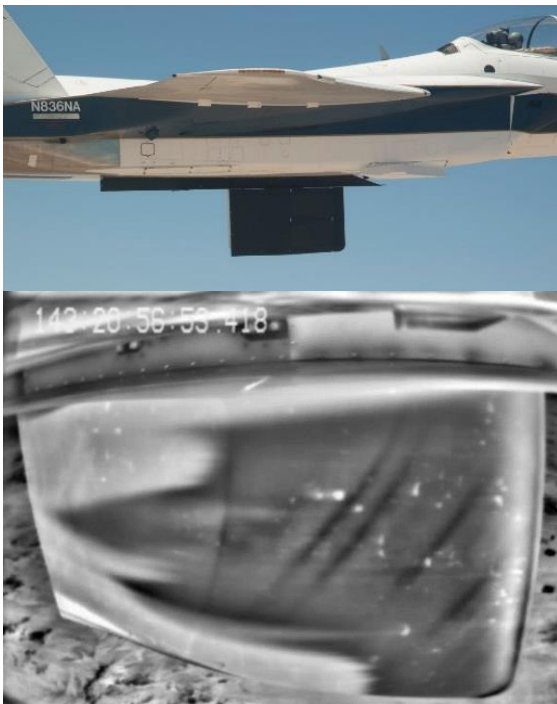
AirBOSCO with $H\alpha$



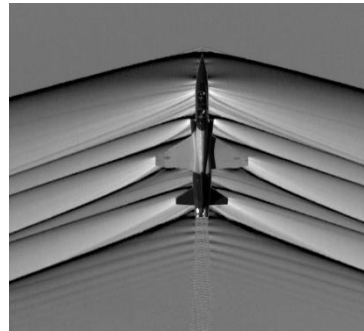
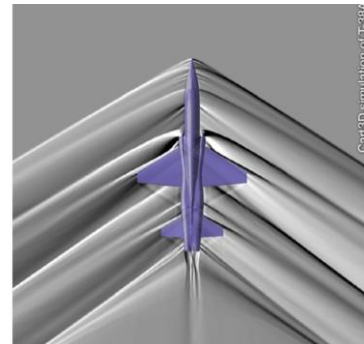
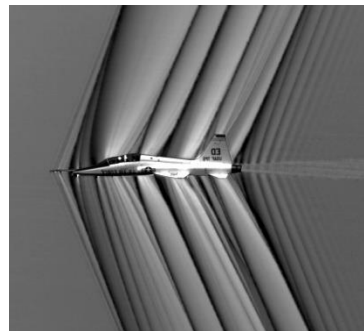
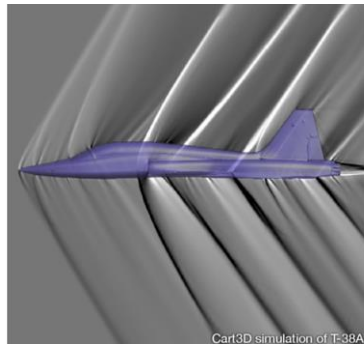
Sun Edge Distortion

CFD Validation: Primary driver for schlieren data

Evidence CFD not correctly predicting all shocks



Unidentified shock impingement on
laminar flow test article



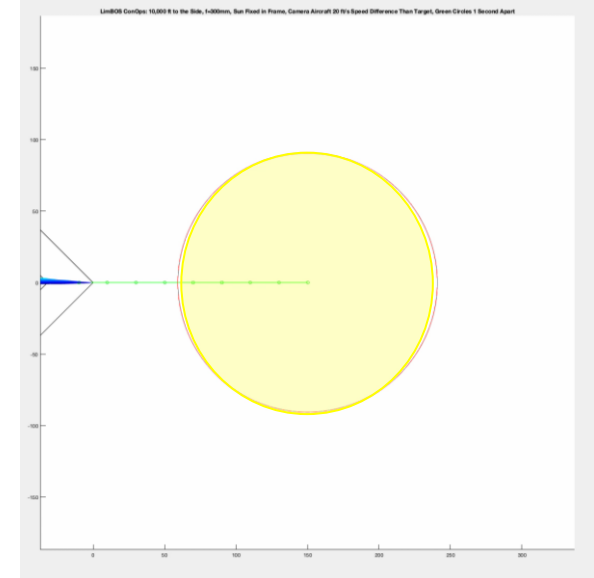
CFD v AirBOS results on T-38

Air-to-Air Formation: Imaging Pod on the NASA F-15

- Side-view of X-59 is priority 1, Sun imaging method
- Use of imaging pod to fly in formation with X59@ ~55k+ feet (~23km)
- Pod takes three cameras, ranges 180° h,
- $\pm 35^\circ$ v, mounts on either wing
- New program/electronics to uplink X59 to F15 for formation flight coordination – ALIGNS
- Programmed to track sun
- **High vibration and unconditioned environment**



The planned pod for imaging the X-59 QueSST



Pattern for imaging the downward shocks

Environmental Testing: Image blur due to vibration

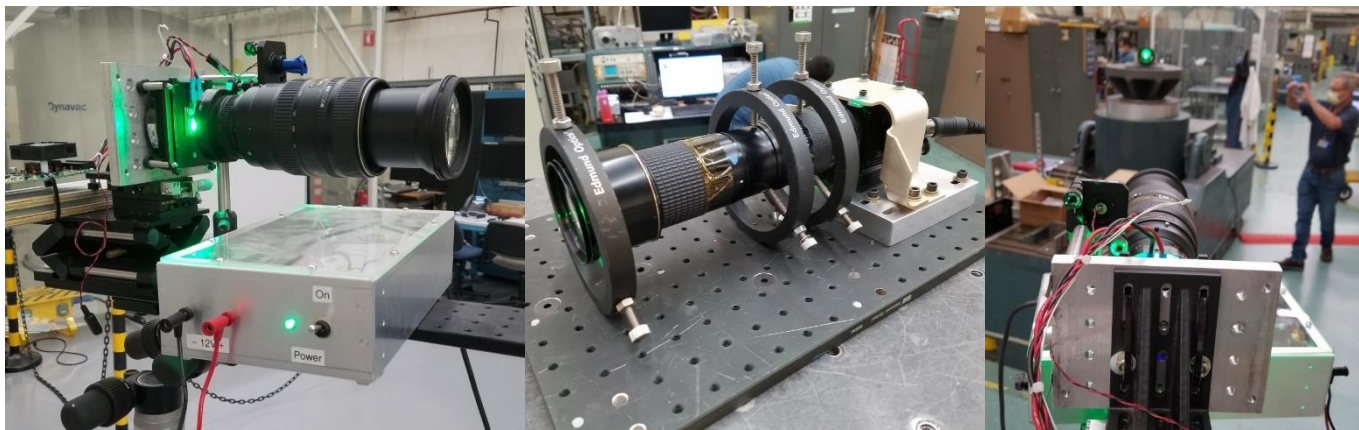
3-axis accelerometers for in-flight vibrate of turret, programmed vibrate table to match
IDT O10 camera/lens combos tested on vibrate table at measured levels

A high-intensity LED was collimated and re-imaged on flight camera

Analysis showed optimal integration time for Sun-edge optics

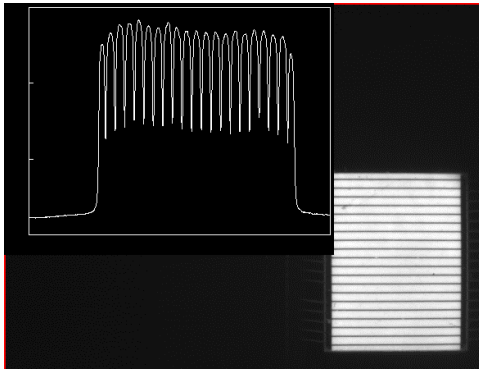
AirBOSCO optics with etalon required 200 microsecond integration time:

Significant image blur eliminated technique

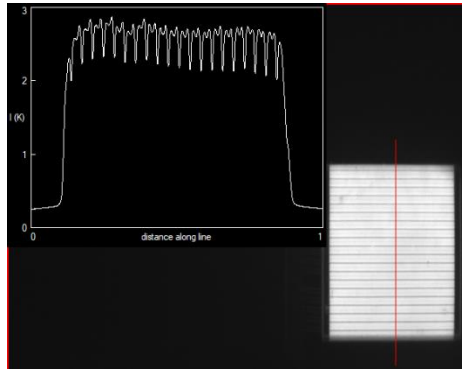


LED and 400 mm lens, stabilized Sun-edge camera / lens, projection alignment

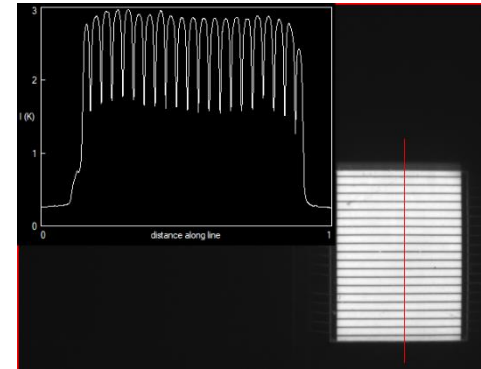
Profile plots of imaged LED



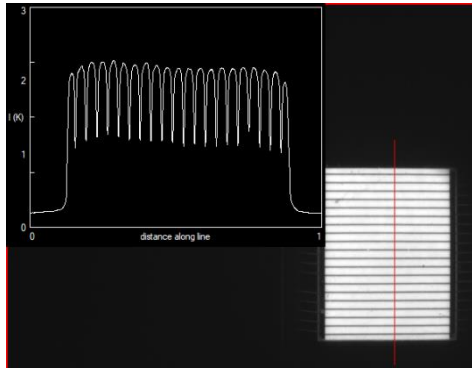
Reference



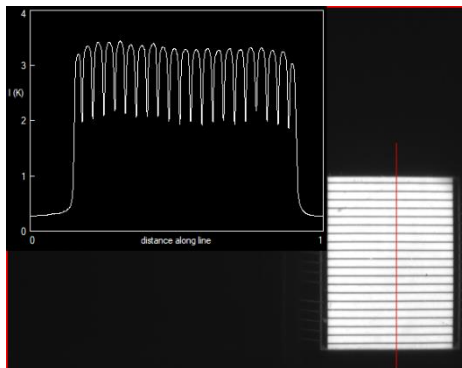
1 microsec



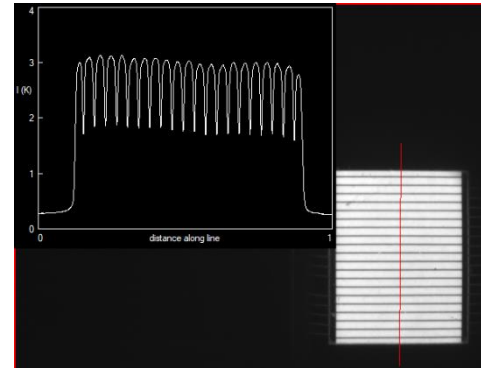
10 microsec



25 microsec

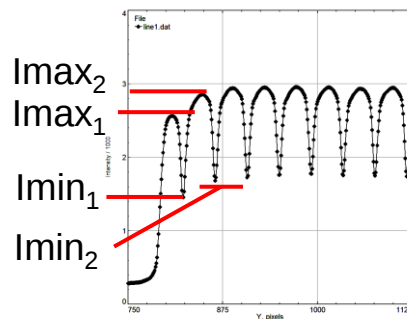
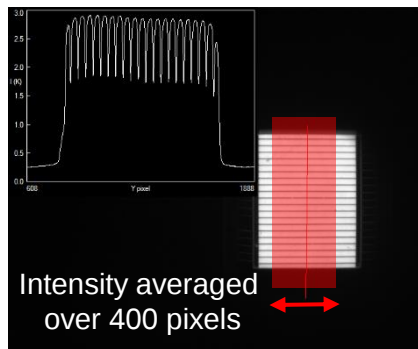


50 microsec

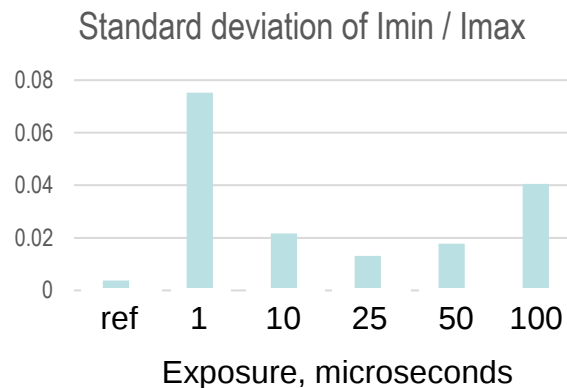
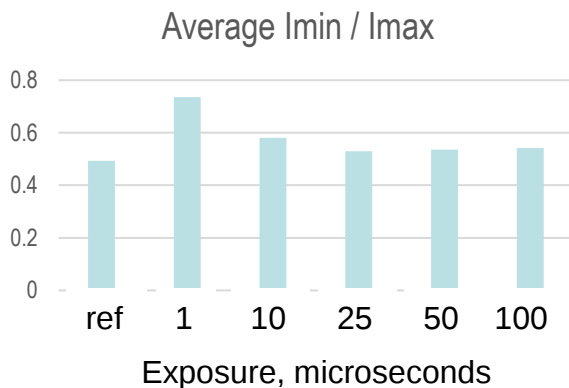


100 microsec

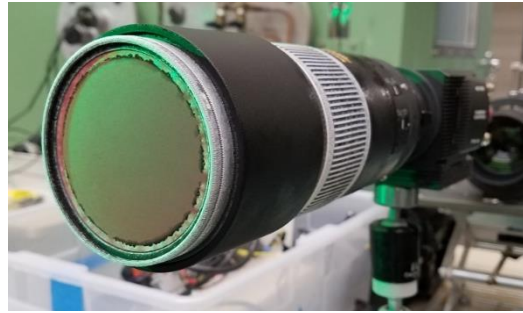
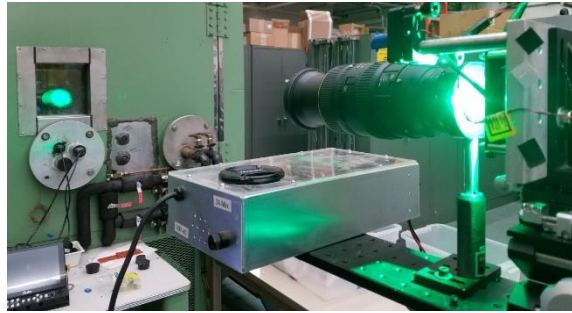
Optimized Exposure: Ghosting v. Motion Blur



$$(Imin / Imax)_{ave} = \frac{1}{N_{cycles}} \sum_{i=1}^{N_{cycles}} \left(\frac{Imin}{Imax} \right)_i$$



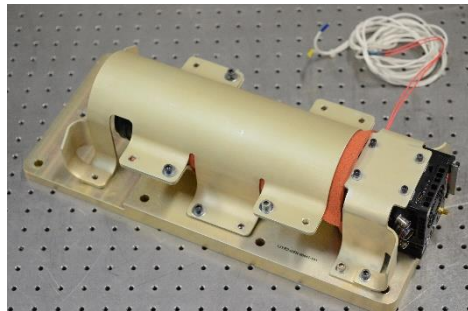
Environmental Testing: Altitude Effects



Focus shifted by ΔCte , air density
Sun-edge config (300 mm +2x TC)
Same LED method, into chamber
Images acquired with pad heater,
PID controller set 20 C +/- 3 C



Lens with pad heater, insulation



Clam-shell and camera cage design



Ref (sea-level)

60,000 ft, 20° C

60,000, -30° C

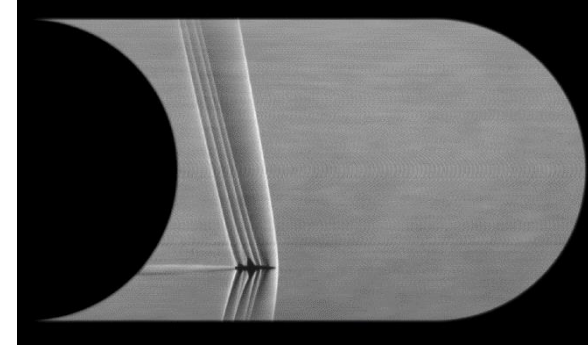
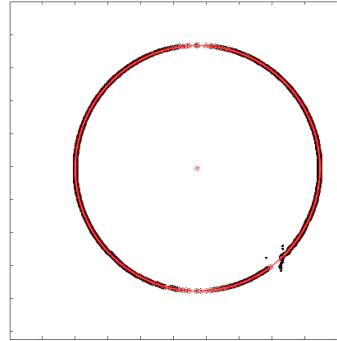
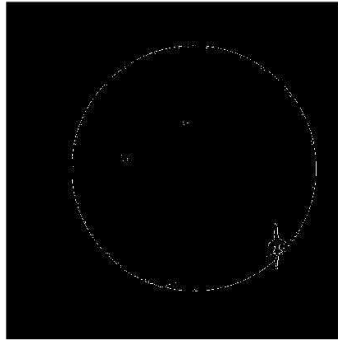
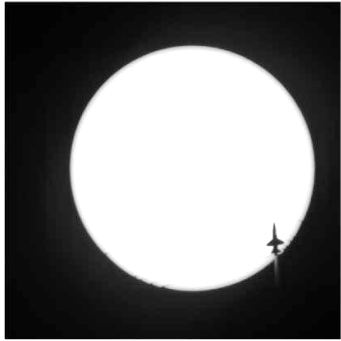
Images of LED during altitude testing show focus stability

Hand-held Sun-edge Imaging Effort

- A backup to pod mounted system
- Battery powered IDT O10 camera
 - Mission Mode: preprogrammed settings from simple trigger input from photographer
- 300 mm Reflex lens with 2x teleconverter
- External HDSDI monitor
- Gunsight for quick alignment to Sun
- Operate from rear seat in cockpit of chase aircraft



Sun-Edge Image Processing



RANSAC method: raw image, to detected perimeter, to multiple 3-point locations that fall in the model of the Sun diameter

Sequence of aligned “rings”

Random Sample Consensus (RANSAC) method: filters outliers caused by disturbances

Select three points from locus set to define arbitrary circle x_c , y_c , R

Classify each data point as inlier or outlier, based on measured undistorted Sun image

Perform many trials, then select solution with largest percentage of inliers

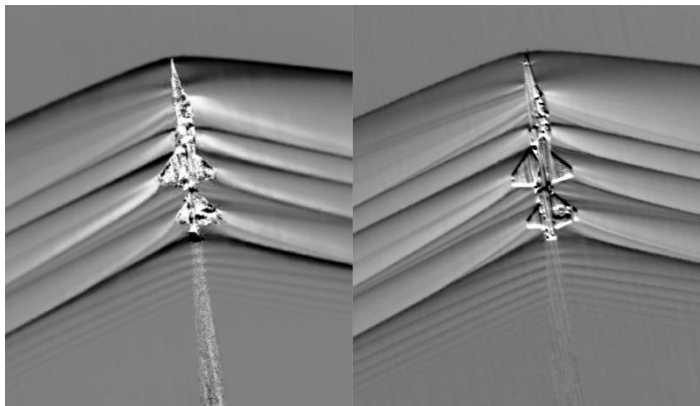
Final least squares refinement with inliers

Extract perimeter of sun with distortions, line up the “rings” to create the schlieren image.

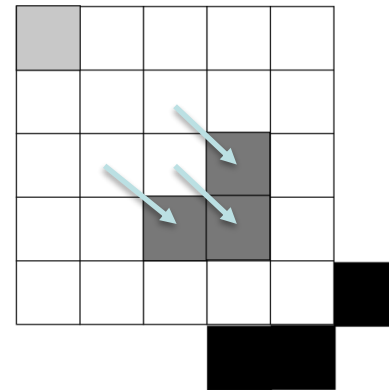
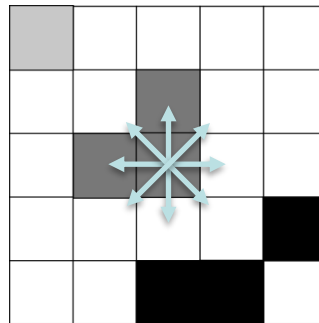
AirBOS Image Processing

Image Processing with Optical Flow

- Similarly uses a reference background image and compares to a data image
- Mathematically measures the apparent motion due to brightness changes
 - Assumes the illumination level is constant between frames
 - The distortion caused by the shockwaves makes the bushes “move” a few pixels
 - The “motion” is detected at each pixel with a partial differential equation



Cross correlation v Optical Flow



Summary

- Spent many years developing useable flight test techniques to image shock waves for X59 QueSST
- Some apparent flow physics anomalies between actual aircraft and predictions (CFD) have been observed.
- Sun-edge distortion imaging will enable capturing downward shocks
- Image processing of BOS flight images has evolved from early cross-correlation methods to the present optical flow methods.
- Hardware and software development have been targeted to support Low Boom Flight Demo (X-59).
 - Currently working on an air-to-air system for imaging pod to be mounted on F-15 to chase future X-59 LBFD.
 - Evaluating a hand-held camera using the Sun-edge technique for operating from the cockpit
- Extensive environmental testing revealed unforeseen issues, which are being mitigated
- New software for both flight planning/guidance and image processing for schlieren data developed



AMERICAN INSTITUTE OF
AERONAUTICS AND ASTRONAUTICS