



# 30 Years of Atmospheric Radiometer Instruments and Technology Development

SUNY Albany  
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# Outline

- Technology Readiness Level (TRL)
- Millimeter-wave Imaging Radiometer (MIR)
- Compact Scanning Millimeter-wave Imaging Radiometer (CoSMIR)
- Compact Scanning Submm-wave Imaging Radiometer (CoSSIR)
- IceCube – 884 GHz Ice Cloud Radiometer
- Submillimeter Enceladus Life Fundamentals Instrument (SELFIE)
- Venus Wideband Submillimeter Heterodyne Spectrometer (V-WiSHeS)
- Submillimeter Solar Observation of Lunar Volatile Experiment (SSOLVE)
- Technological Thrusts

# Technology Readiness Level

TRL 9

•Actual system “flight proven” through successful mission operations

TRL 8

•Actual system completed and “flight qualified” through test and demonstration (ground or space)

TRL 7

•System prototype demonstration in a space environment

TRL 6

•System/subsystem model or prototype demonstration in a relevant environment (ground or space)

TRL 5

•Component and/or breadboard validation in relevant environment

TRL 4

•Component and/or breadboard validation in laboratory environment

TRL 3

•Analytical and experimental critical function and/or characteristic proof-of-concept

TRL 2

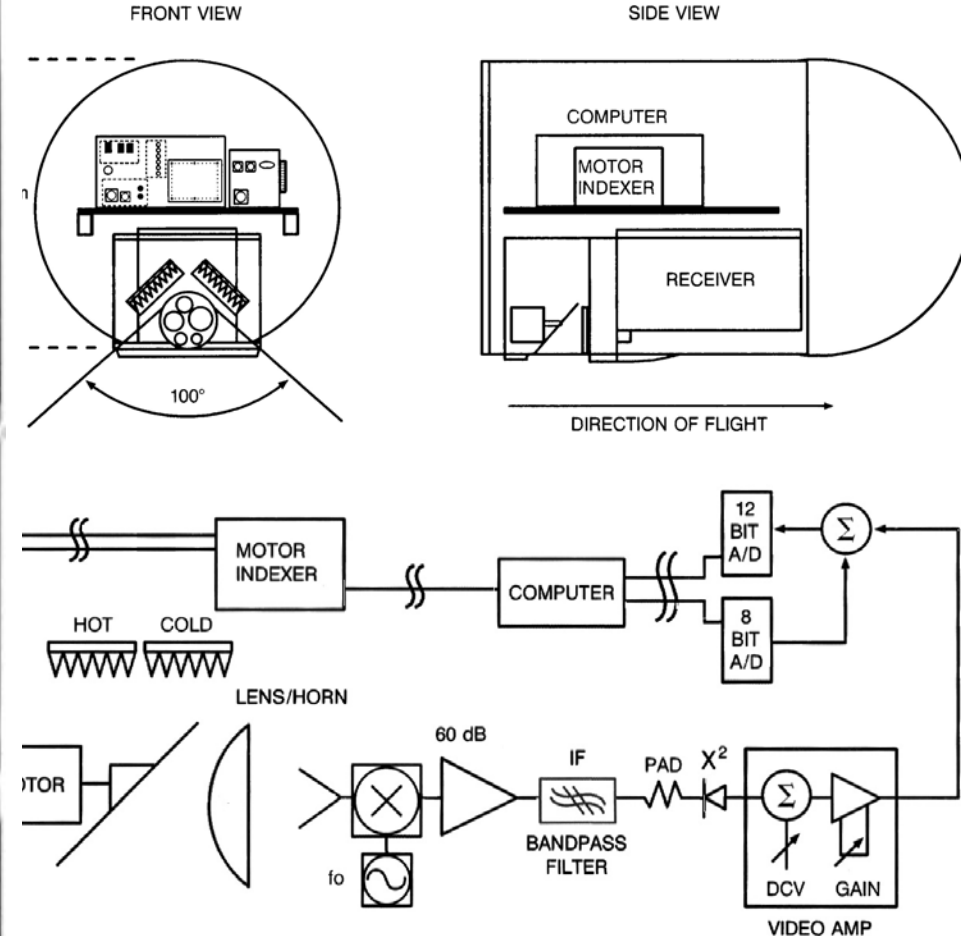
•Technology concept and/or application formulated

TRL 1

•Basic principles observed and reported

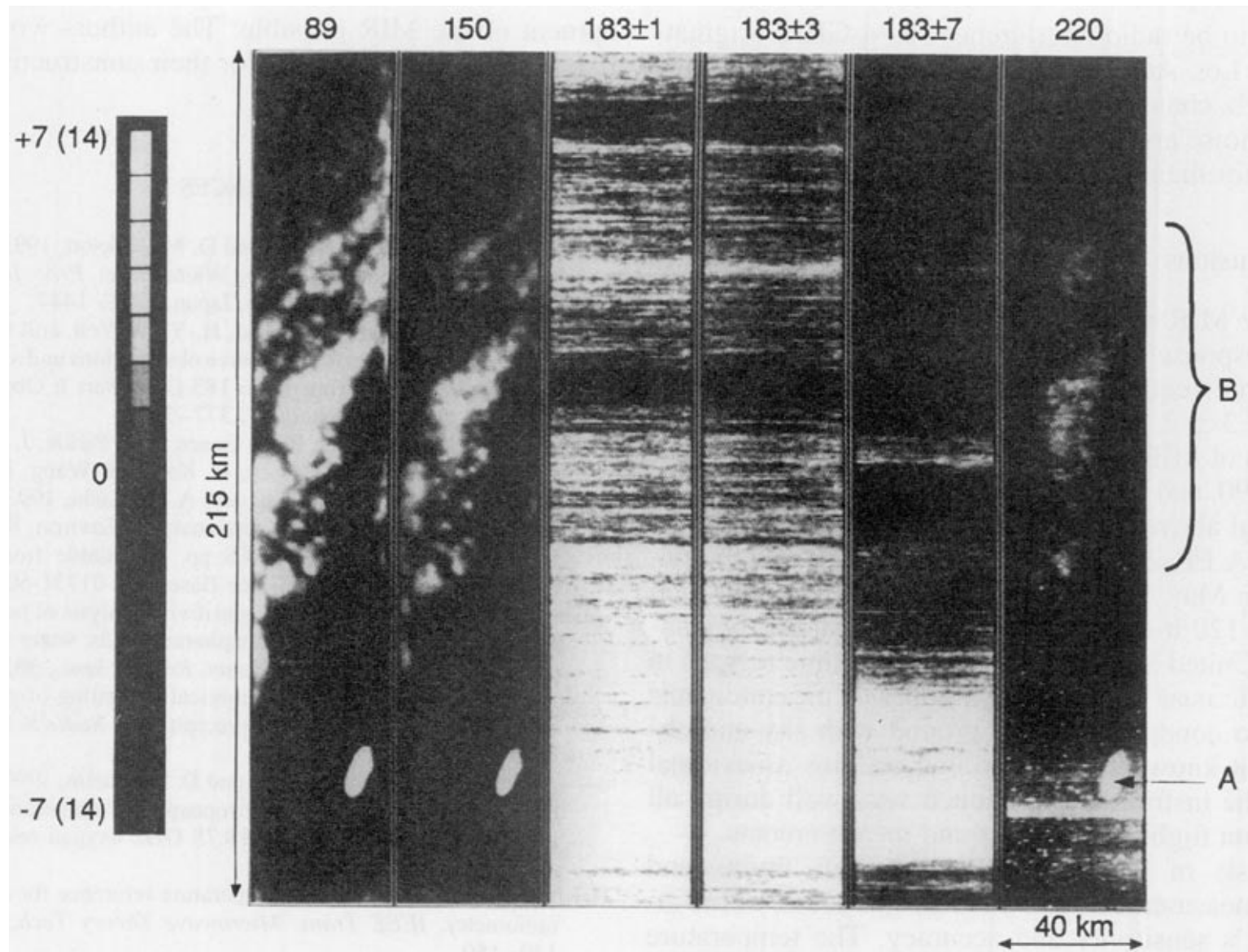
# Millimeter-wave Imaging Radiometer (MIR)

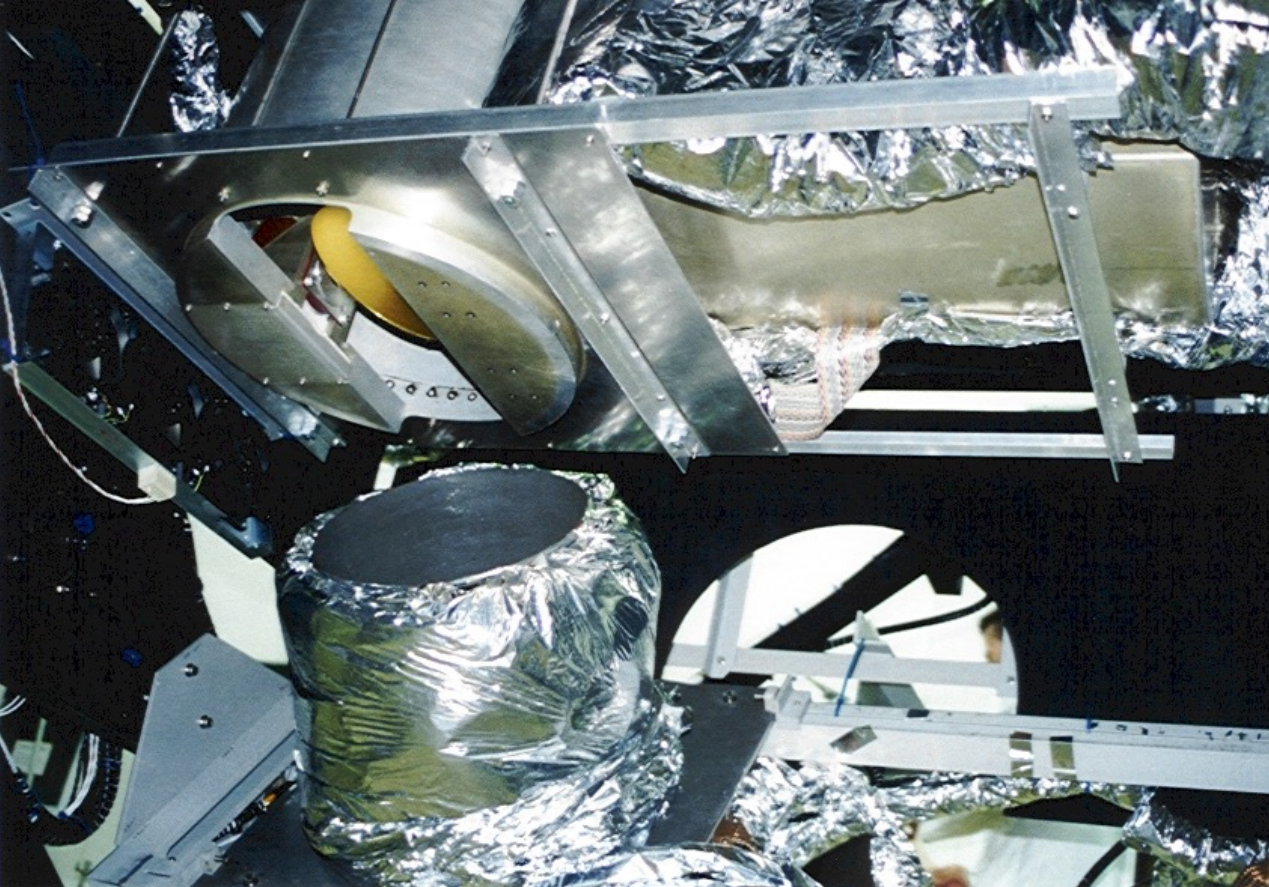
- Frequencies 89 – 340 GHz
- Maiden Flight 1992
- Sorties >150
- Flight Hours > 500
- Retired ~2005



# MIR Imagery

May 1992





# MIR Calibration Studies

# Conical Scanning Millimeter-wave Imaging Radiometer (CoSMIR)

Co-Sponsored by the DoD and NASA for the DMSP SSMIS cal/val program.

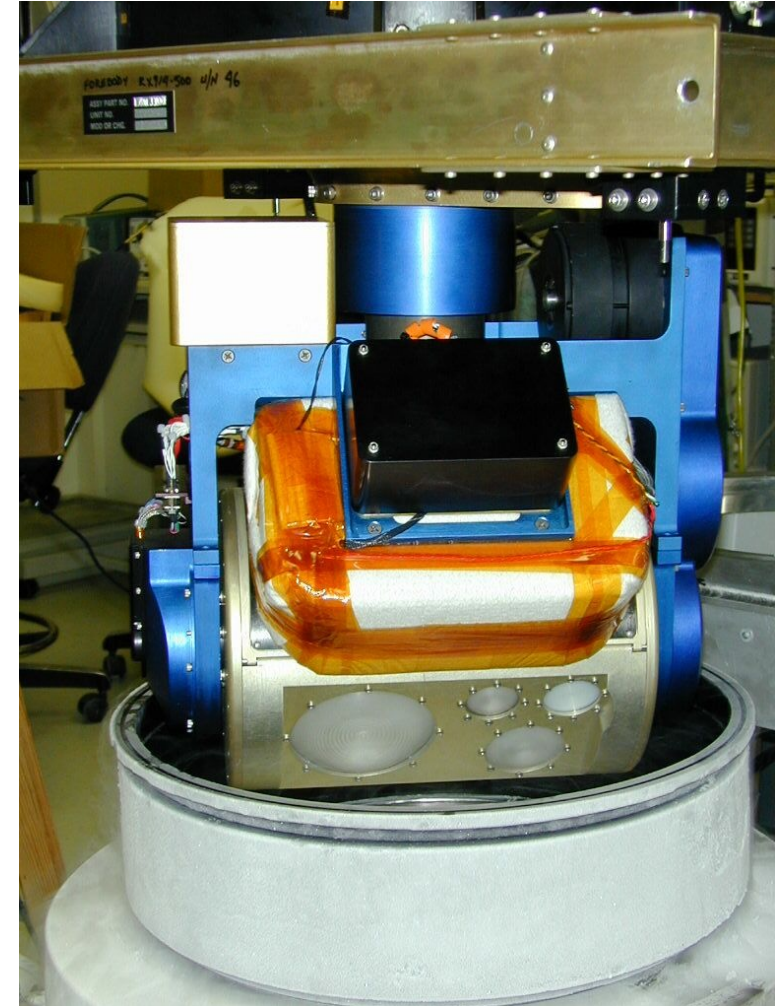
Designed to fly on ER-2 high altitude (~20km) aircraft.

Dual-axis gimbals provide fore and aft conical scan, two cross-track scans, and two calibrations every 10 seconds.

Calibration is achieved by periodic looks (~5s) at hot (330 K) and cold (~250K) close-coupled blackbody references.

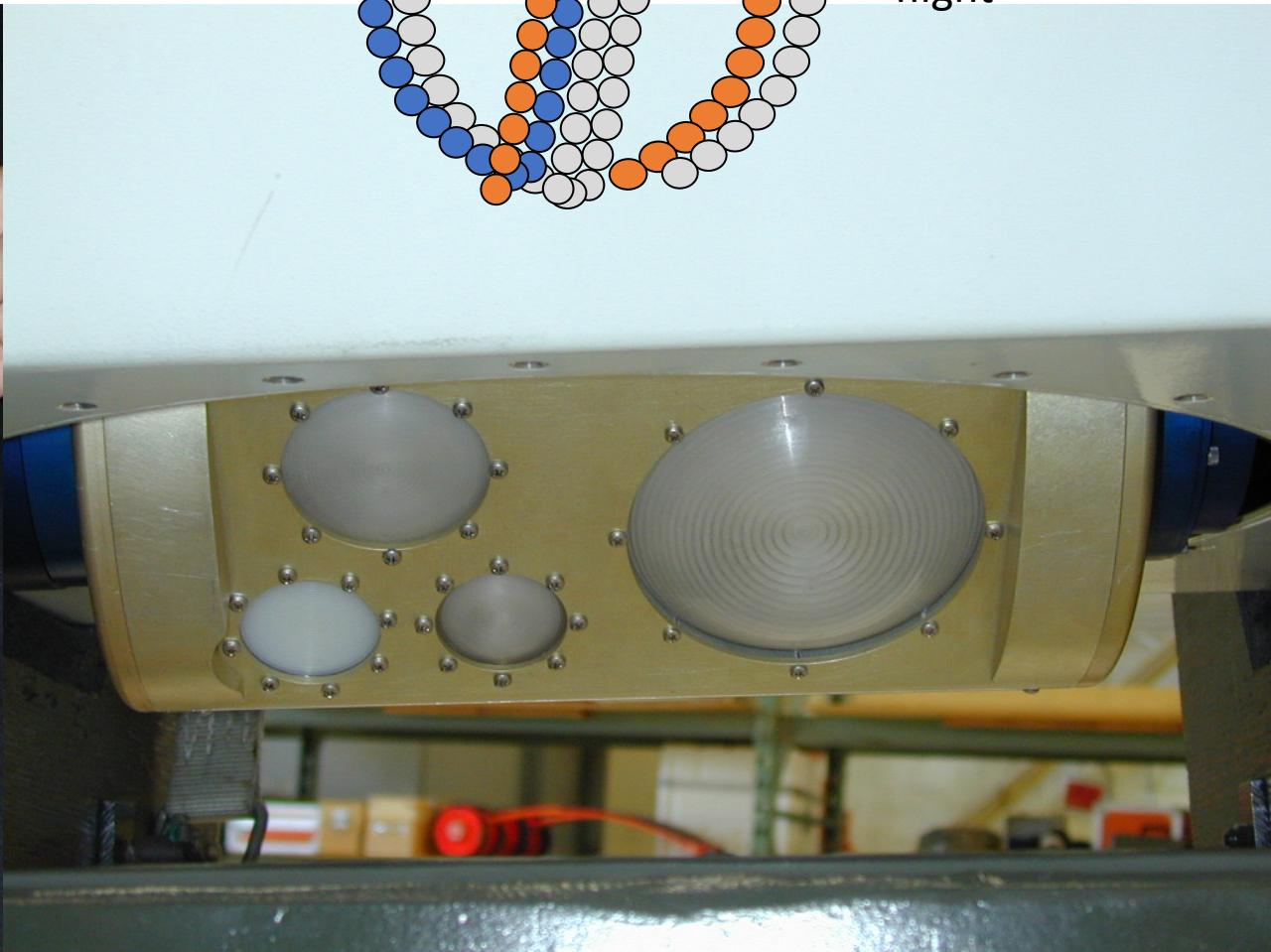
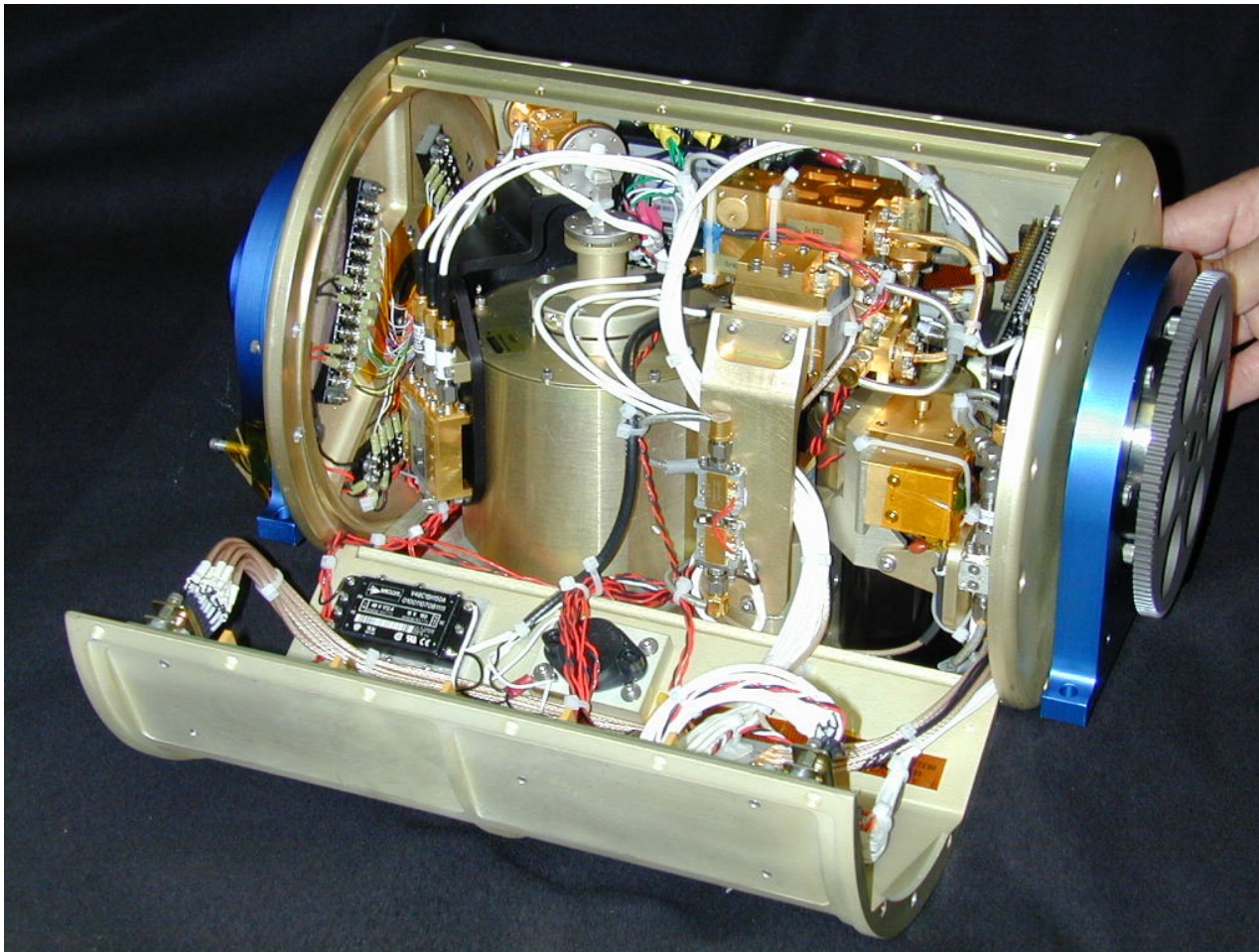
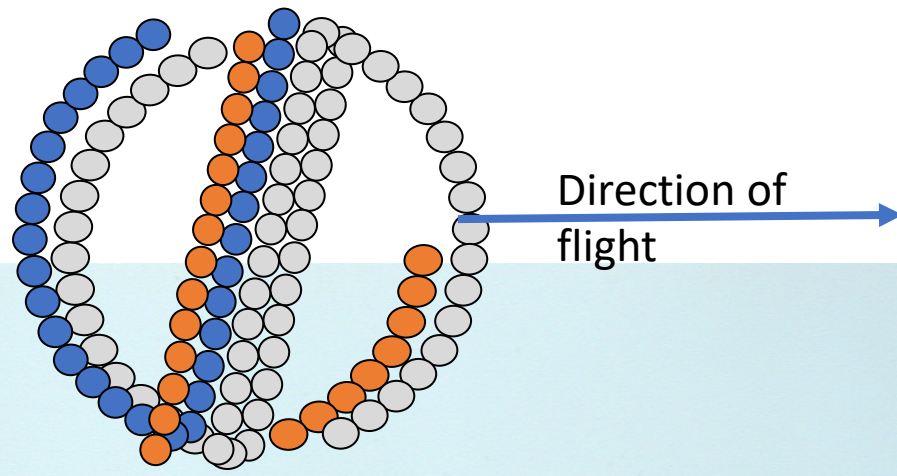
First flight: March 2001

Center Frequency (GHz)	Bandwidth (MHz)	Noise Figure (dB)	NE $\Delta$ T for $\tau = 100$ ms (K)	Beamwidth (degrees)	Polarization
50.3	400	4.8	0.13	< 5	Horizontal
52.8	400	4.8	0.13	< 5	Horizontal
53.596	400	4.8	0.13	< 5	Horizontal
91.655	1500	6.5	0.10	4	Dual-linear
150.0	1500	7.5	0.13	4	Horizontal
183.3 $\pm$ 1	500	8.7	0.30	4	Horizontal
183.3 $\pm$ 3	1000	8.7	0.21	4	Horizontal
183.3 $\pm$ 6.6	1500	8.7	0.17	4	Horizontal



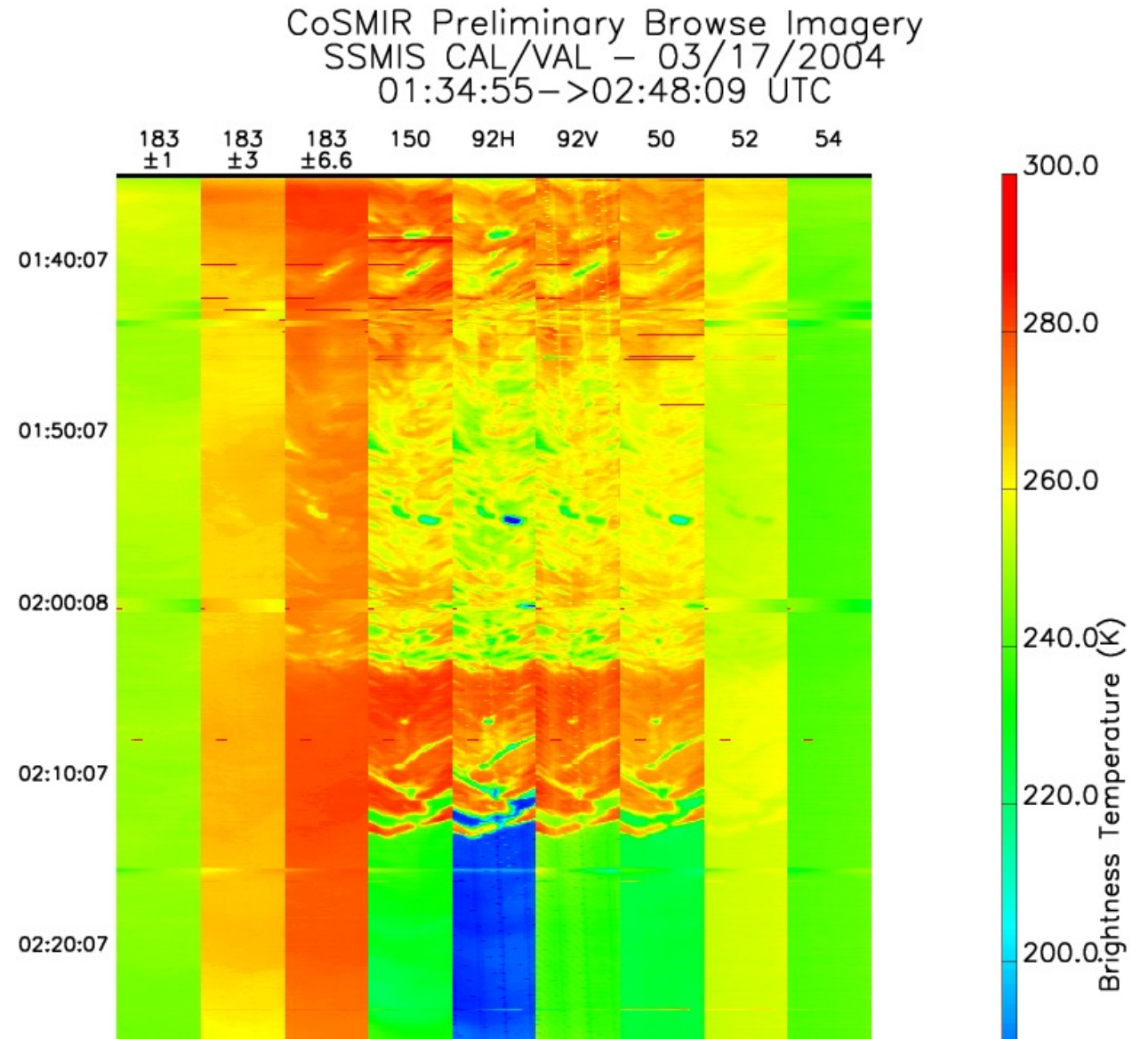
# CoSMIR Scanhead and Scan Profile

Conical/Cross-track Scan Profile



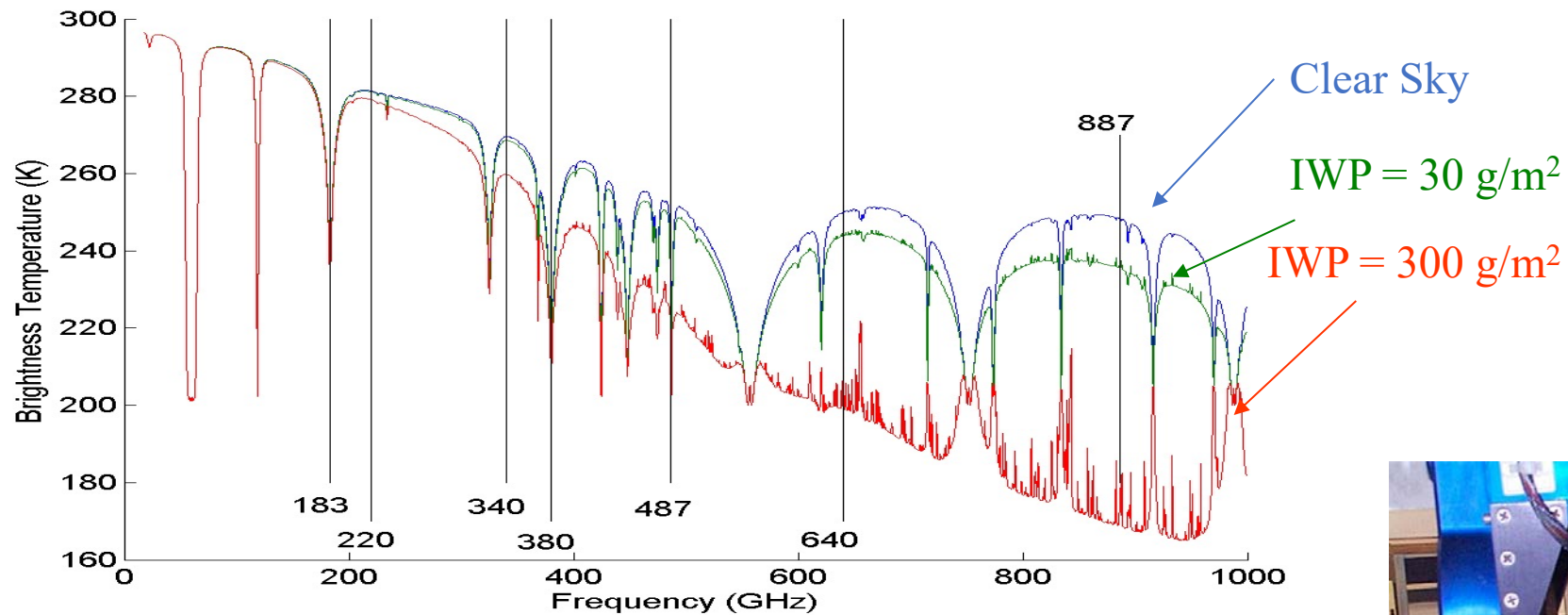
# CoSMIR Data

- An example of CoSMIR 9-channel brightness temperature images over San Francisco Bay area. Data are not geolocated. Some stripes in the images are times when the aircraft making turns.



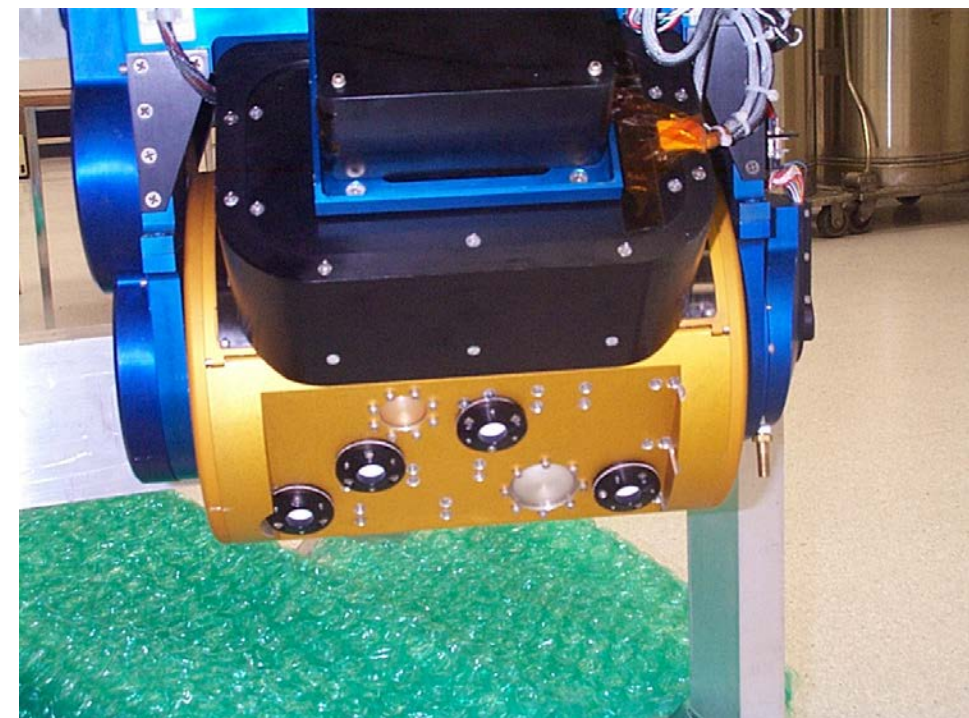
# CoSSIR's 15 Channels Response to Cirrus

Response of Upwelling Radiation to the Presence of Cirrus Clouds

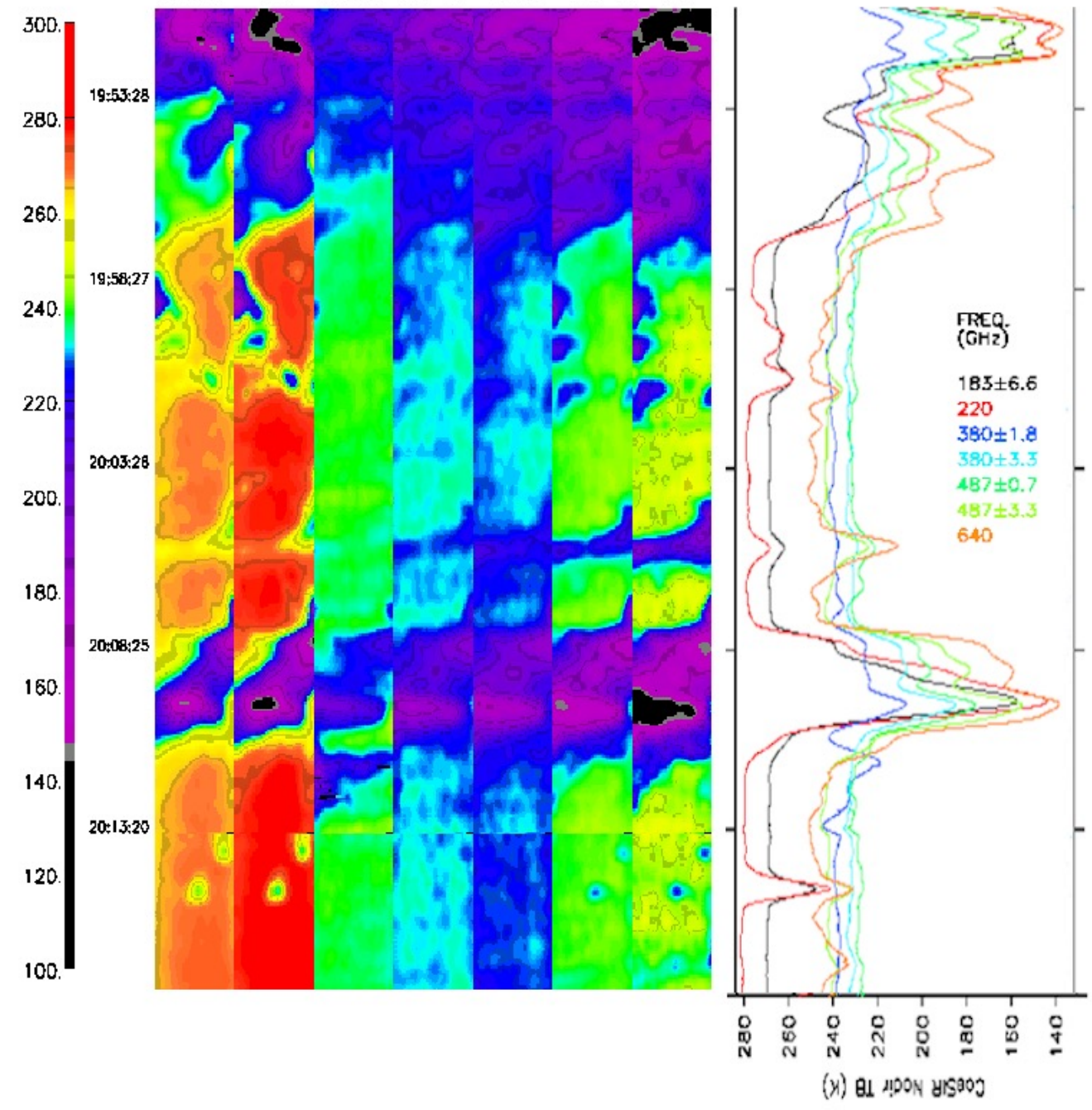


Frequency (GHz)	Channels (GHz)	Polarization
183	$\pm 1, \pm 3, \pm 6.6$	H
220	$\pm 3$	H
380	$\pm 0.8, \pm 1.8, \pm 3.3, \pm 6.2$	H
487	$\pm 0.7, \pm 1.2, \pm 3.3$	V & H
640	$\pm 4$	H
874	$\pm 6$	H

CoSSIR Scanhead

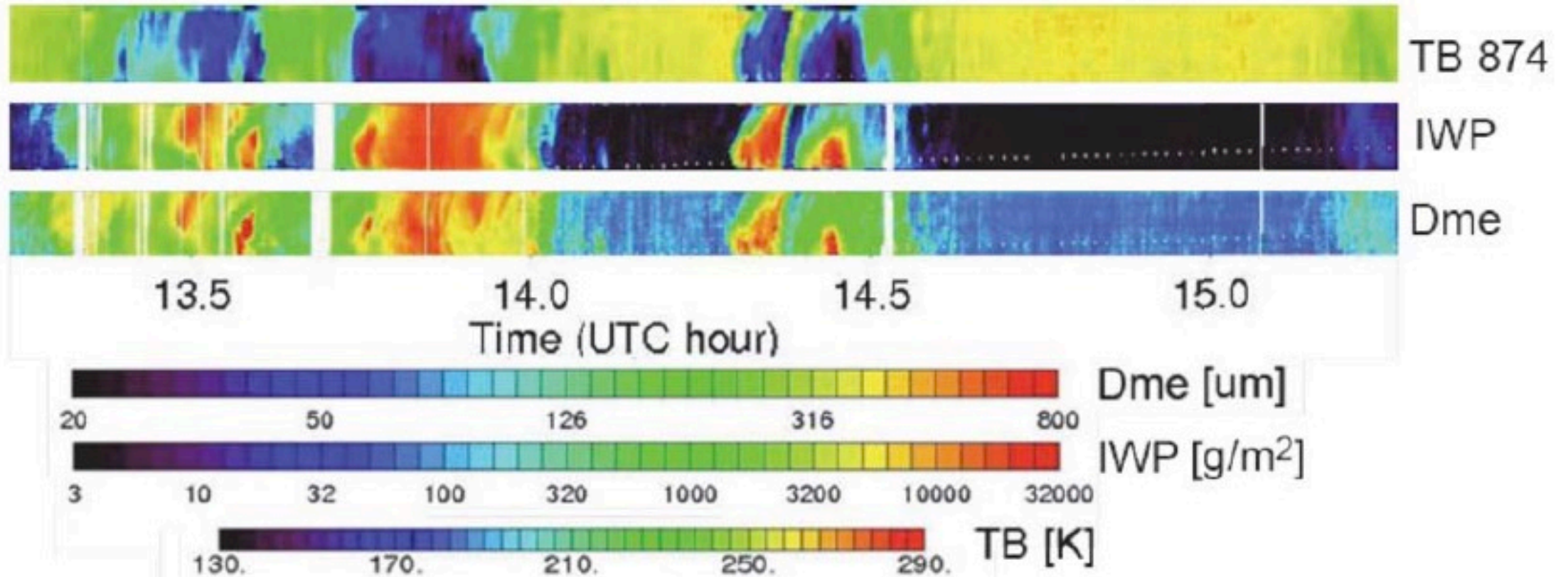


183±6.6 220 380±1.8 380±3.3 487±0.7 487±3.3 640



# CoSSIR Airborne Data

# CoSSIR's 874 GHz Brightness Temps & Retrieved Ice Cloud Parameters





National Aeronautics and Space Administration

Goddard Space Flight Center

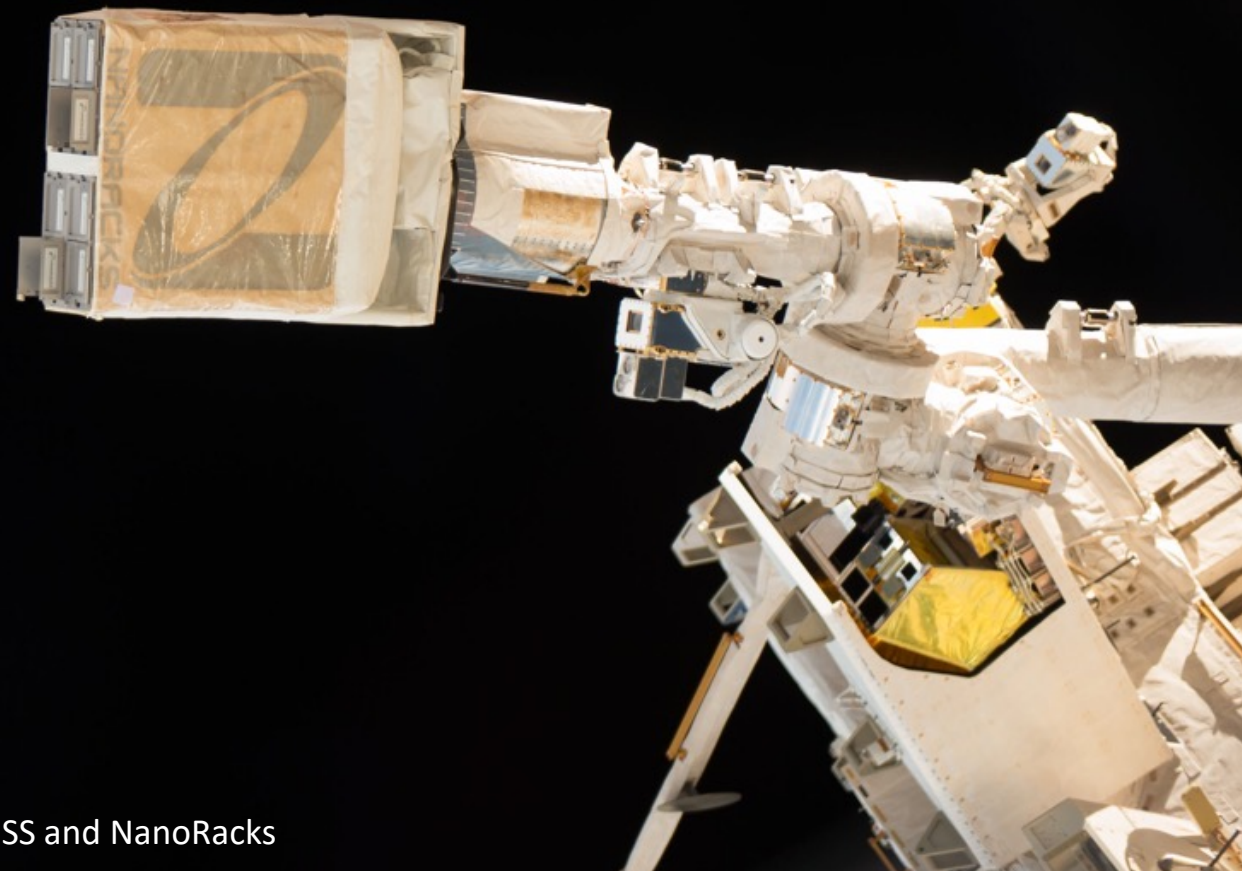
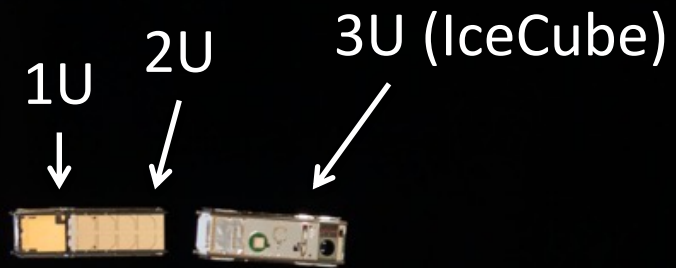


# IceCube: CubeSat Demonstration of Commercial 883-GHz Cloud Radiometer

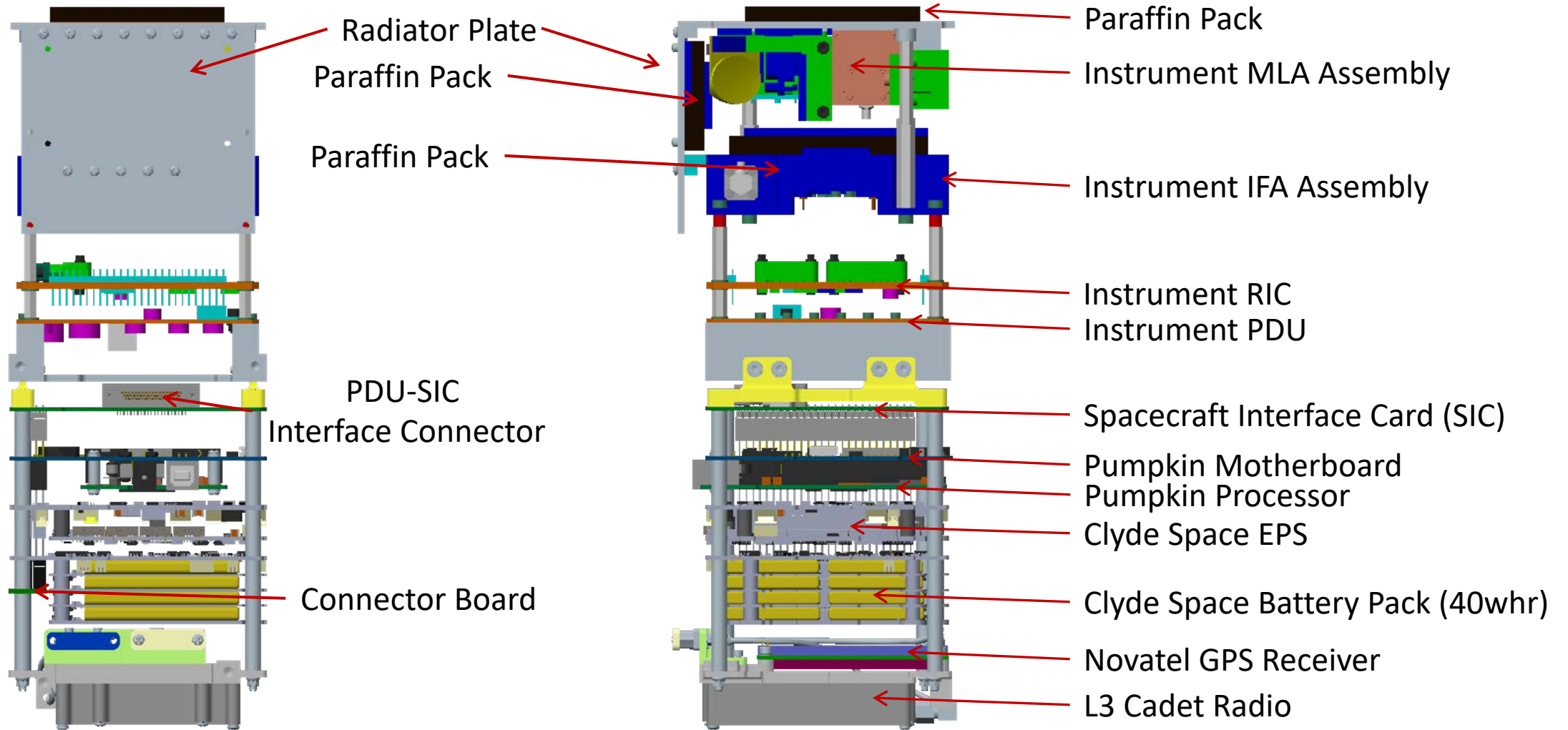
Dong Wu, Tom Johnson, Negar Ehsan, Jaime Esper,  
Brian Abresch, Jeff Piepmeier, Paul Racette, and the  
IceCube Team

NASA Goddard Space Flight Center

# IceCube Released from ISS on May 16, 2017

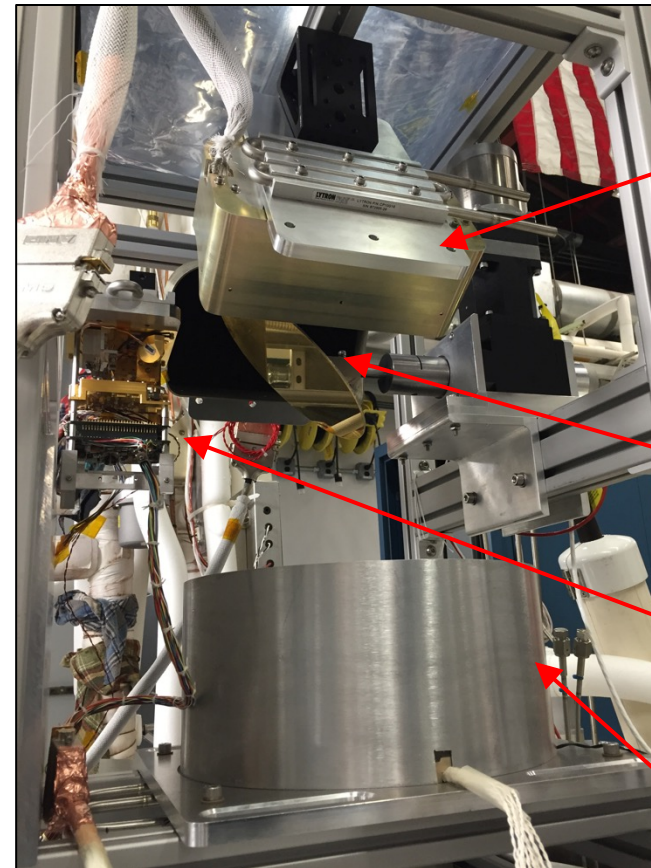
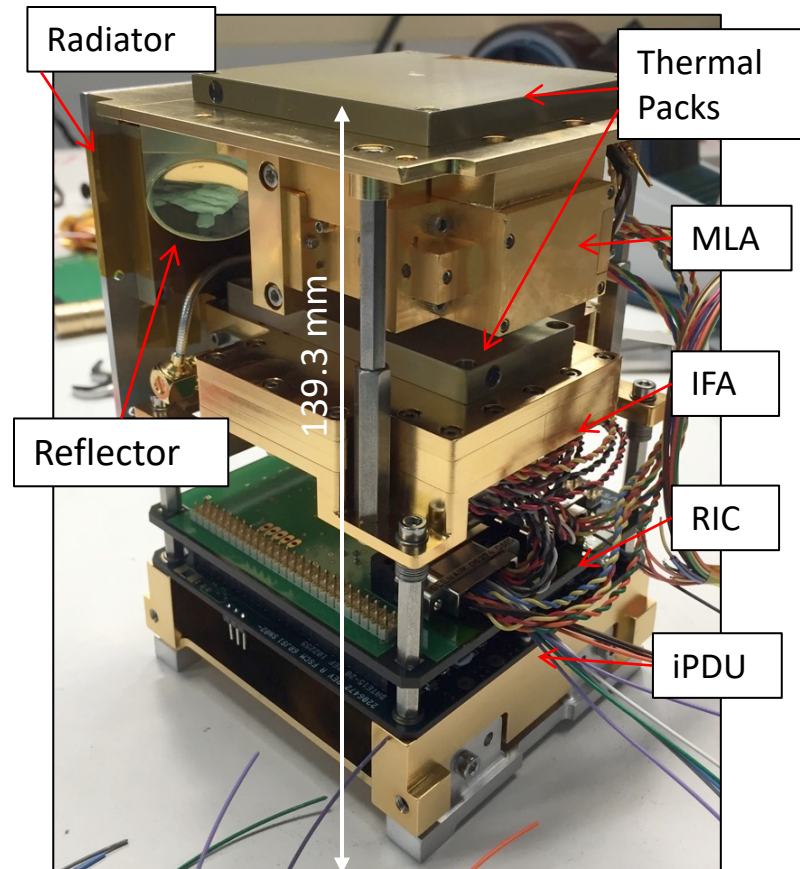


# IceCube 3U Observatory



# Instrument Integration & Test

Flight Model IceCube Instrument



IceCube Calibration Fixture  
(57 cm x 57 cm x 91 cm)

Sub-MMW  
blackbody cold  
target (100 K) & hot  
target (300 K) for  
external calibration;  
tolerance  $\pm 1$  K

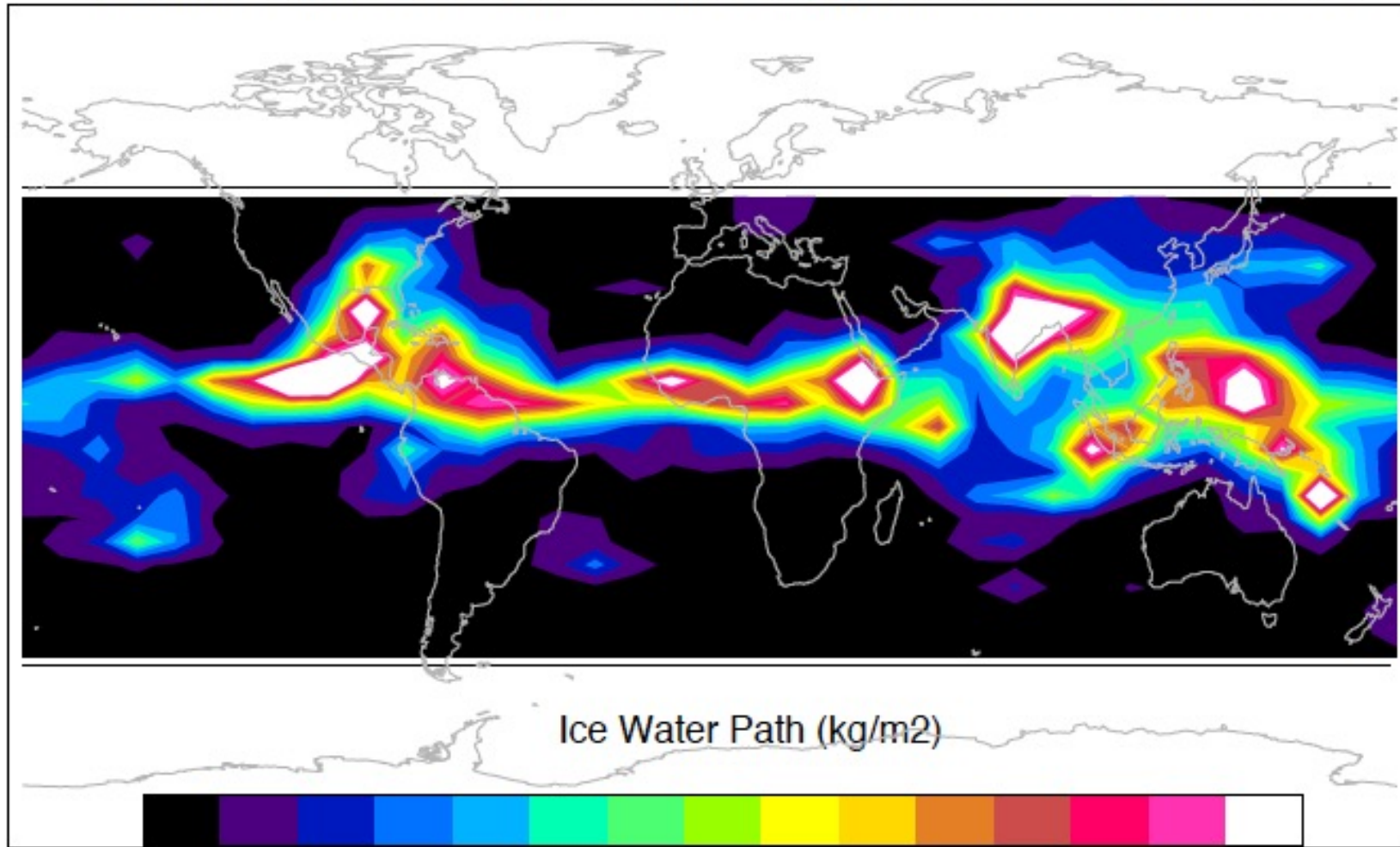
45° offset rotating  
mirror to alternate  
observed  
scene/target

Instrument

Sub-MMW  
blackbody target  
(variable 100-300  
K); tolerance  $\pm 1$  K

# First 883-GHz Cloud Ice Map

IceCube Cirrus Ice Water Path During 20170620 to 20170702



0.0 0.2 0.4 0.6 0.8 1.0 1.2 1.4 1.6 1.8 2.0 2.2 2.4 2.6 2.8



# Picture this SELFI:

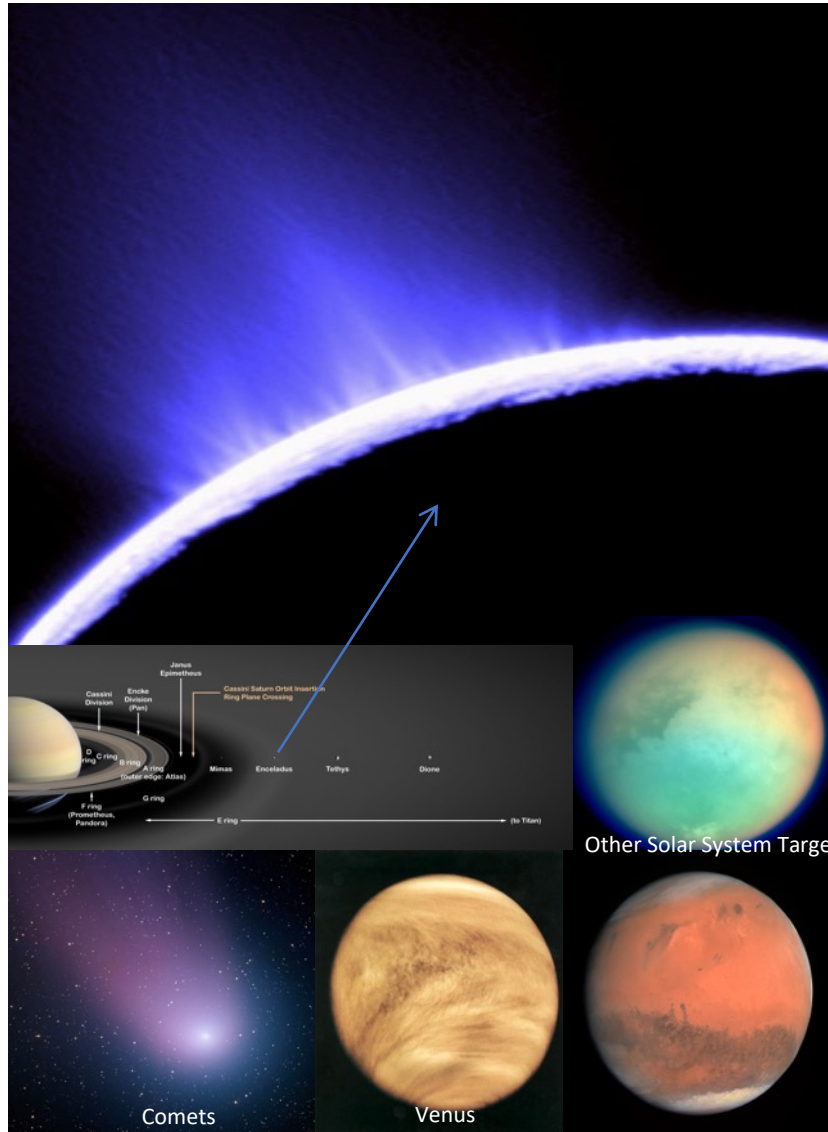
Submillimeter Enceladus Life Fundamentals Instrument

Principal Investigator

Dr. Gordon Chin

Code 693, Goddard Space Flight Center

# SEIFI Science Objectives



SEIFI observes nearly simultaneously 14 molecular species that are important in the context of life and habitability (five of **CHNOPS** elements necessary for life) of the Enceladus' subsurface ocean.

- Assess plume spatial/temporal compositional variability;
- $\text{H}_2\text{O}$ ,  $\text{HDO}$ , and  $\delta^{18}\text{O}$ , and  $\delta^{17}\text{O}$  evidence of ocean evolution;
- $\text{H}_2\text{O}_2$ , and  $\text{O}_3$  oxidation state of the sub surface ocean;
- $\text{CO}$ ,  $\text{NH}_3$ ,  $\text{CH}_3\text{OH}$ ,  $\text{HCN}$  and  $\text{HNC}$  are biologically relevant species;
- $\text{SO}_2$  and  $\text{H}_2\text{S}$  links to pre-biotic molecules and volcanoes;
- $\text{NaCl}$  provides salinity level of the subsurface ocean and the source of salt in the Saturn ring system.

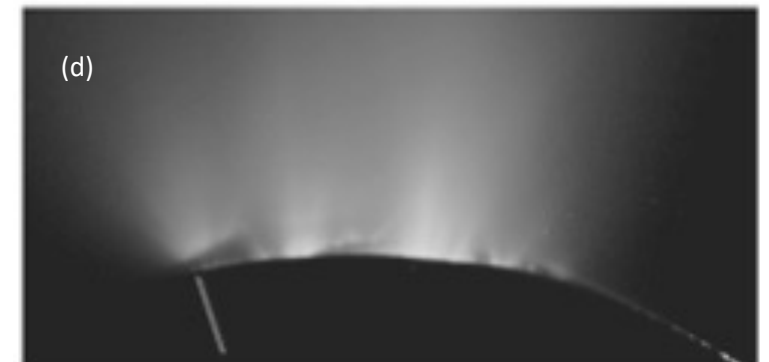
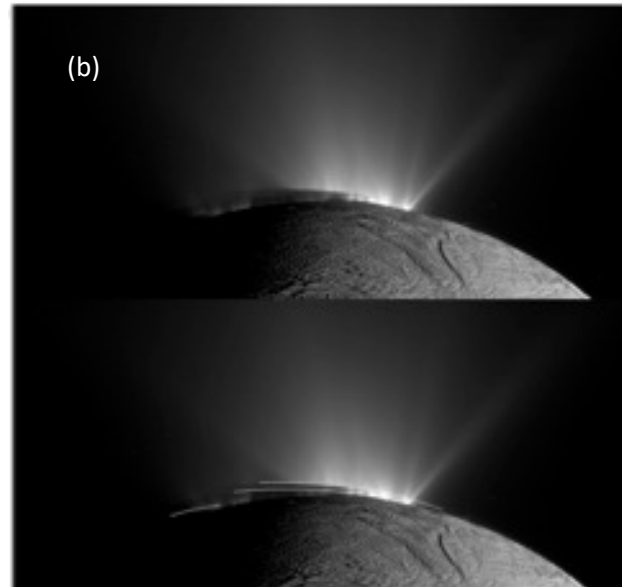
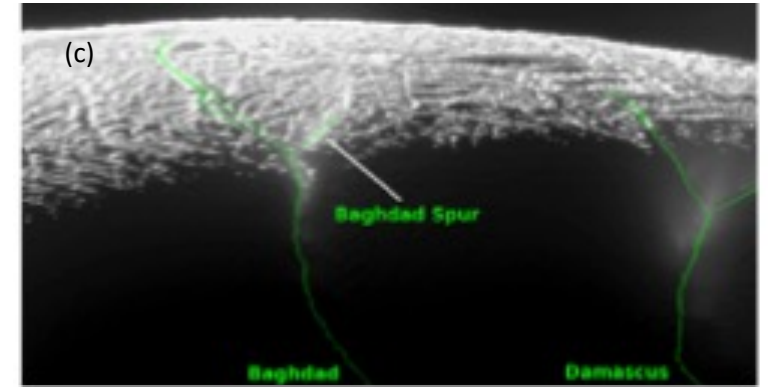
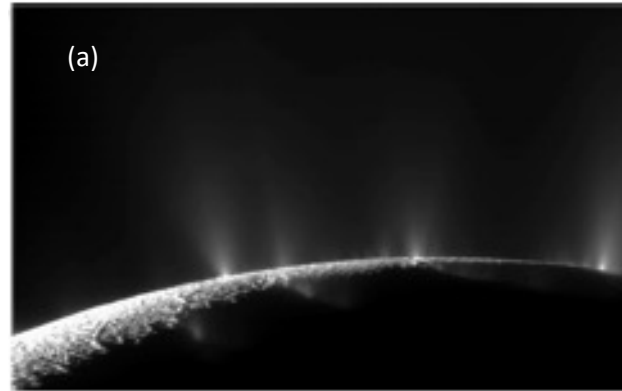
Continuum observations measure surface temperature from 30 - 250 K with 0.1 K resolution.

- Correlation of plume activity with surface temperature.

# Enceladus Plume Observations

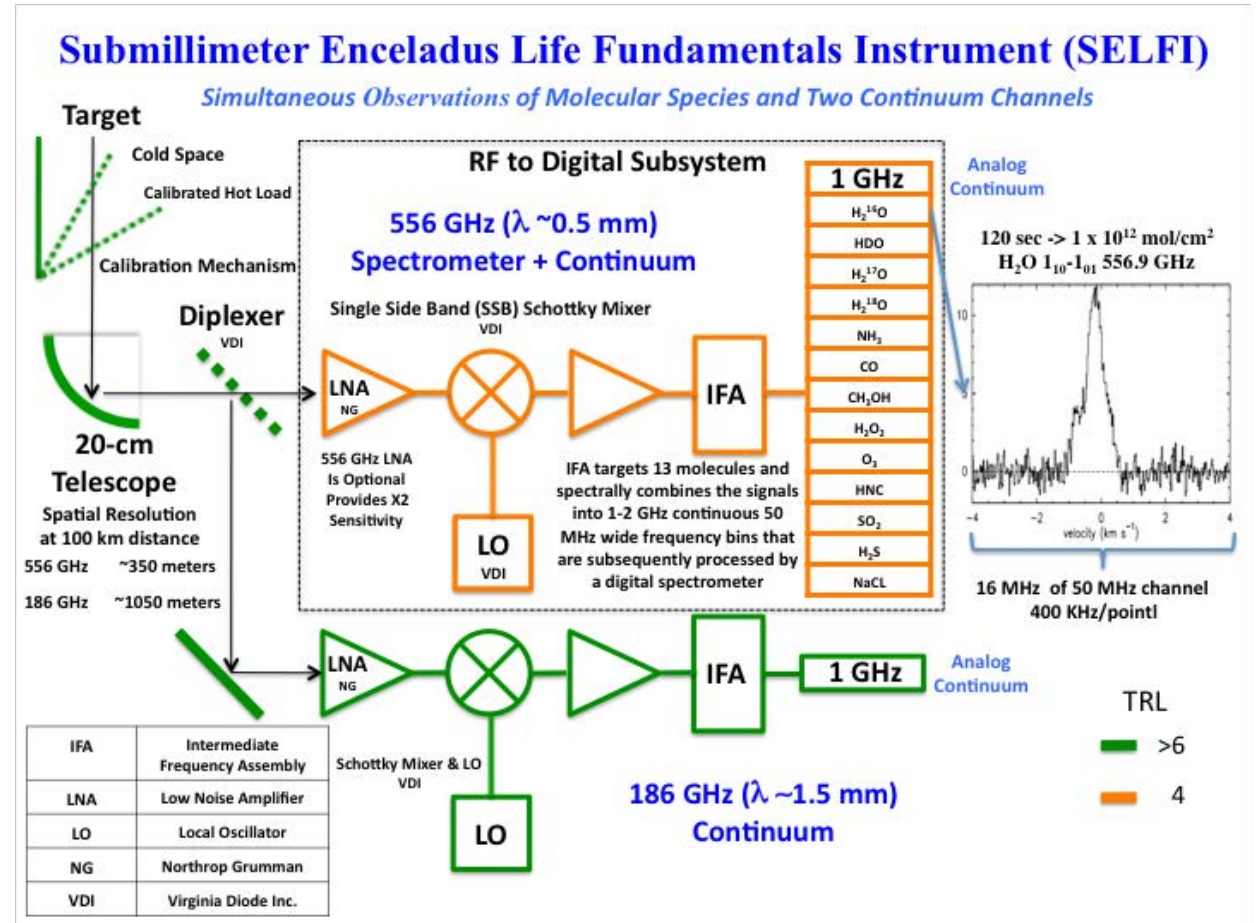
Enceladus' plumes are spatially (a) and temporally (b) variable, arising from fractures called Tiger Stripes (c), and may be curtains of eruptions whose density inhomogeneity gives an observational appearance of individual plumes (d).

From Porco, DeNino, and Nimmo  
Astronomical Journal 248, No. 3, 2014.



# SELEFI Technical Objectives

- Advance the TRL of RF Low Noise Amplifier (RF LNA)
  - Optimize Noise Figure & Gain
- Advance TRL of Mixer & LO Assembly (MLA)
  - Partnership with Virginia Diodes, Inc. via Phase 3 SBIR
- Design, build and test Microwave Assembly (MWA)
  - In-house build leveraging COTS where possible
- Design, build and test Digital Spectrometer Assembly (DSA)
- Raise TRL to 6 of SELFI RF-to-Digital Spectrometer through functional and environmental testing
  - Reduce risk of flight instrument development

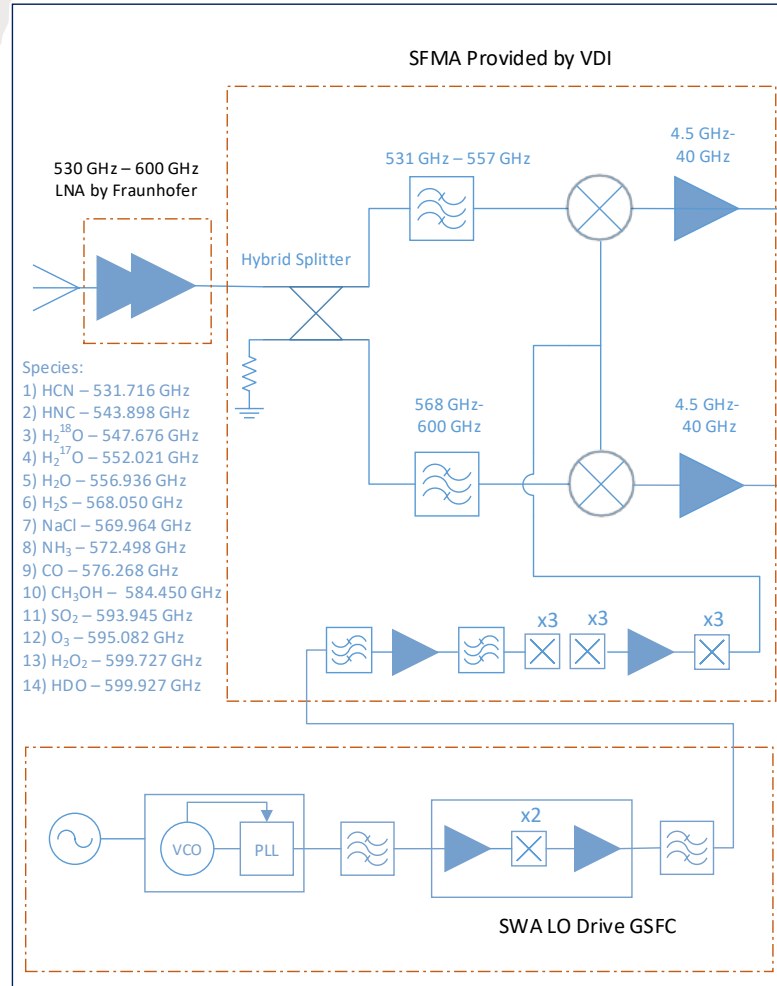


Advance our technical capabilities, reduce technical risks and gain the knowledge necessary to build an instrument to make submm spectroscopic measurements of Enceladus plumes

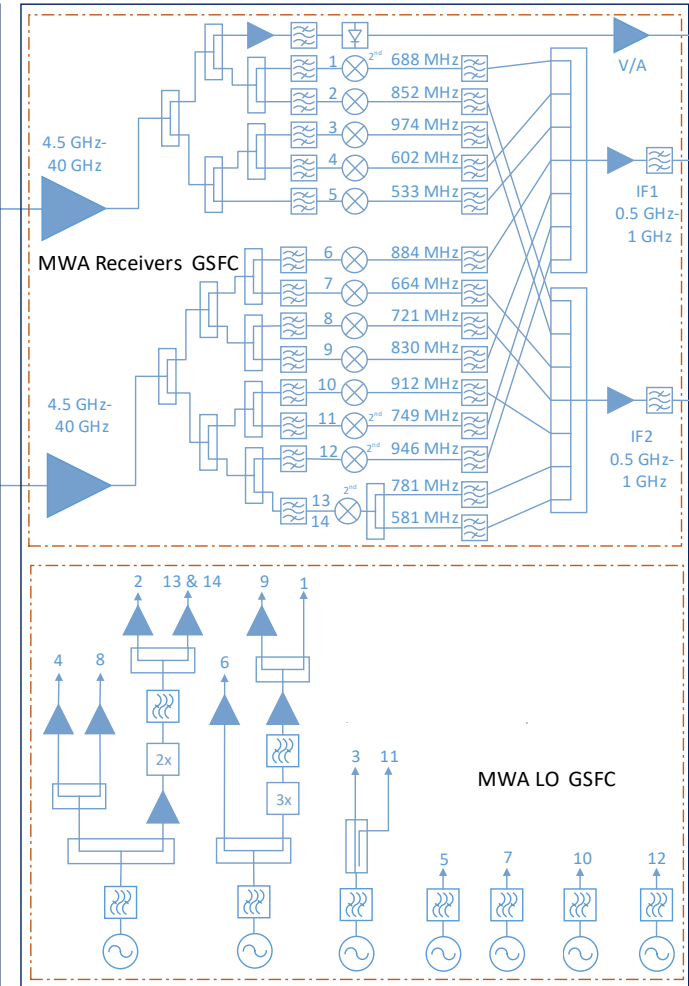
# RF System Architecture

- Cascaded RF LNA has > 20 dB gain and < 12 dB NF
- Hybrid Coupler and dual-mixer downconverts upper/lower sidebands into two 35 GHz IF bands
- SWA LO Drive provides frequency switching for baseline subtraction
- Microwave Assembly downconverts 14 species into two 500 MHz bandwidths

Submm-Wave Assembly

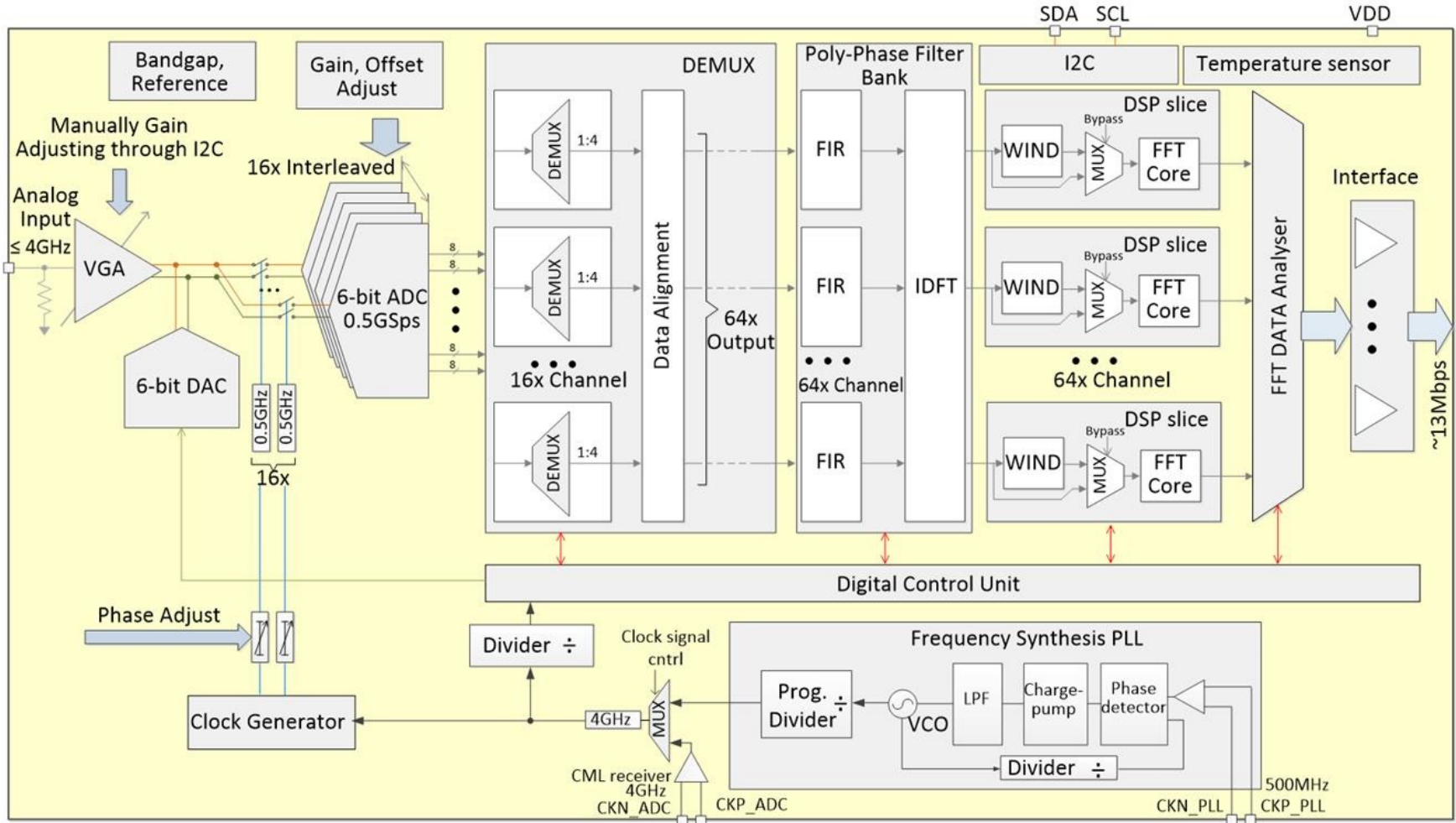


Microwave Assembly

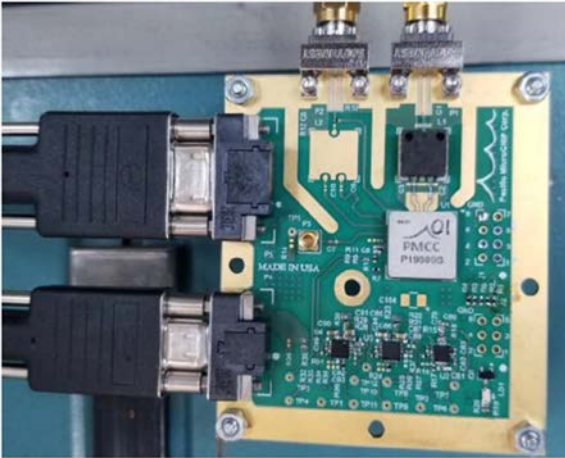


# Enabling Technology: Low-Power Digital Spectrometers

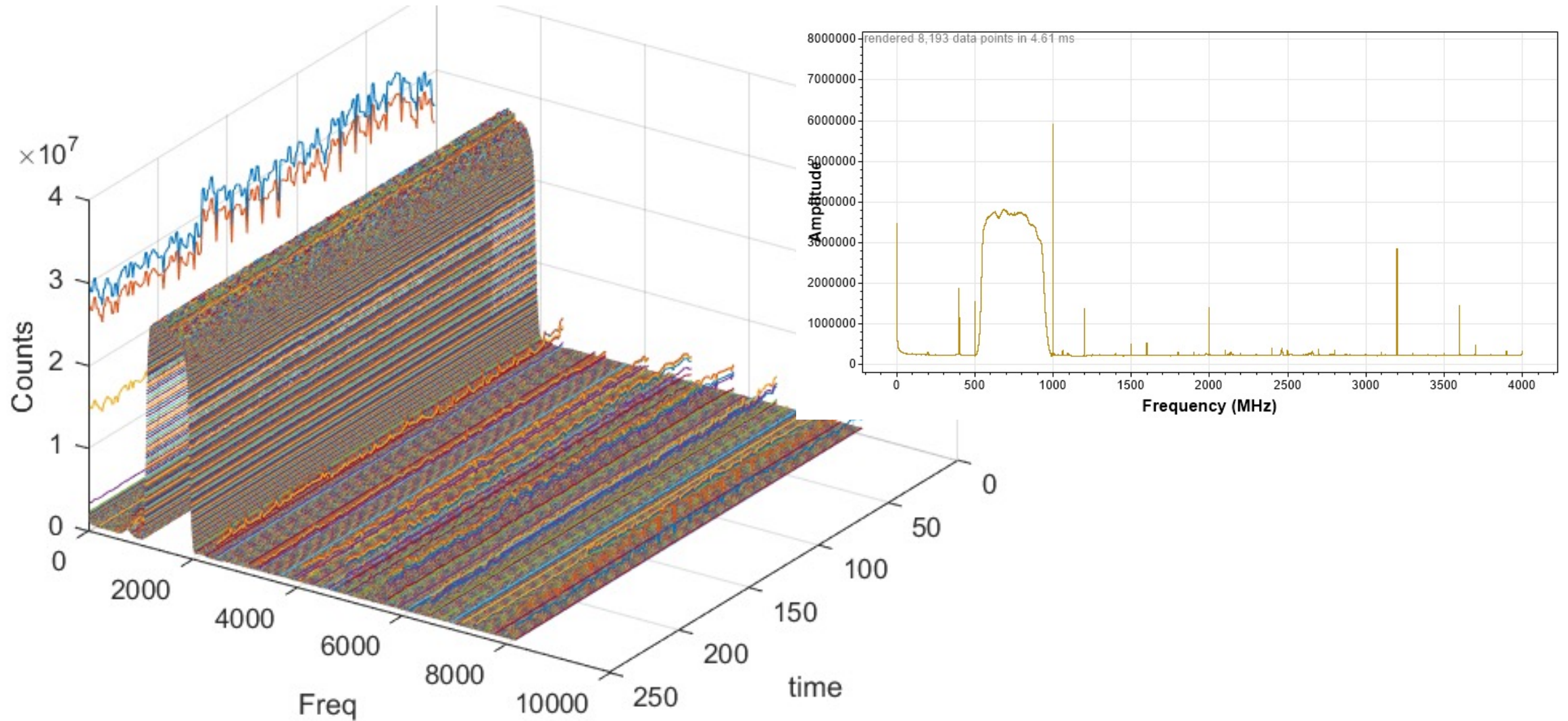
## Radiation Tolerant 4GHz Bandwidth 8k Channel Spectrometer ASIC



**Pacific MicroCHIP**



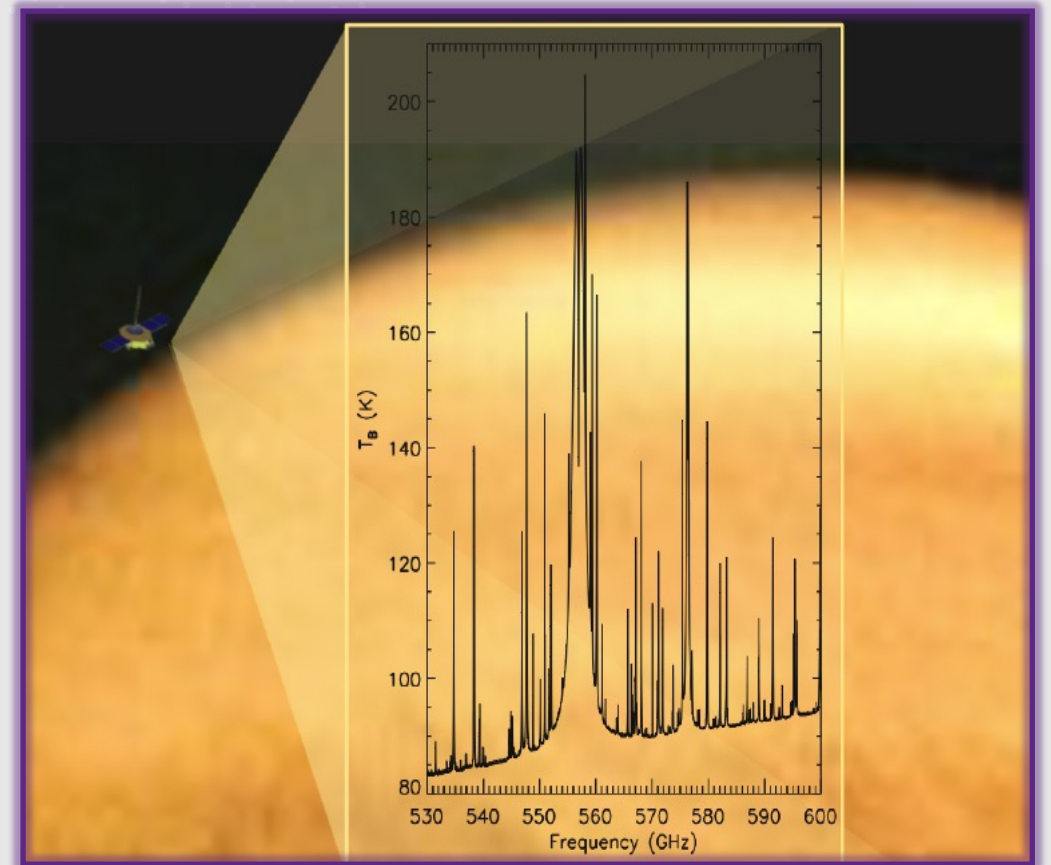
# Digital Spectroscopy - Time Series Spectra



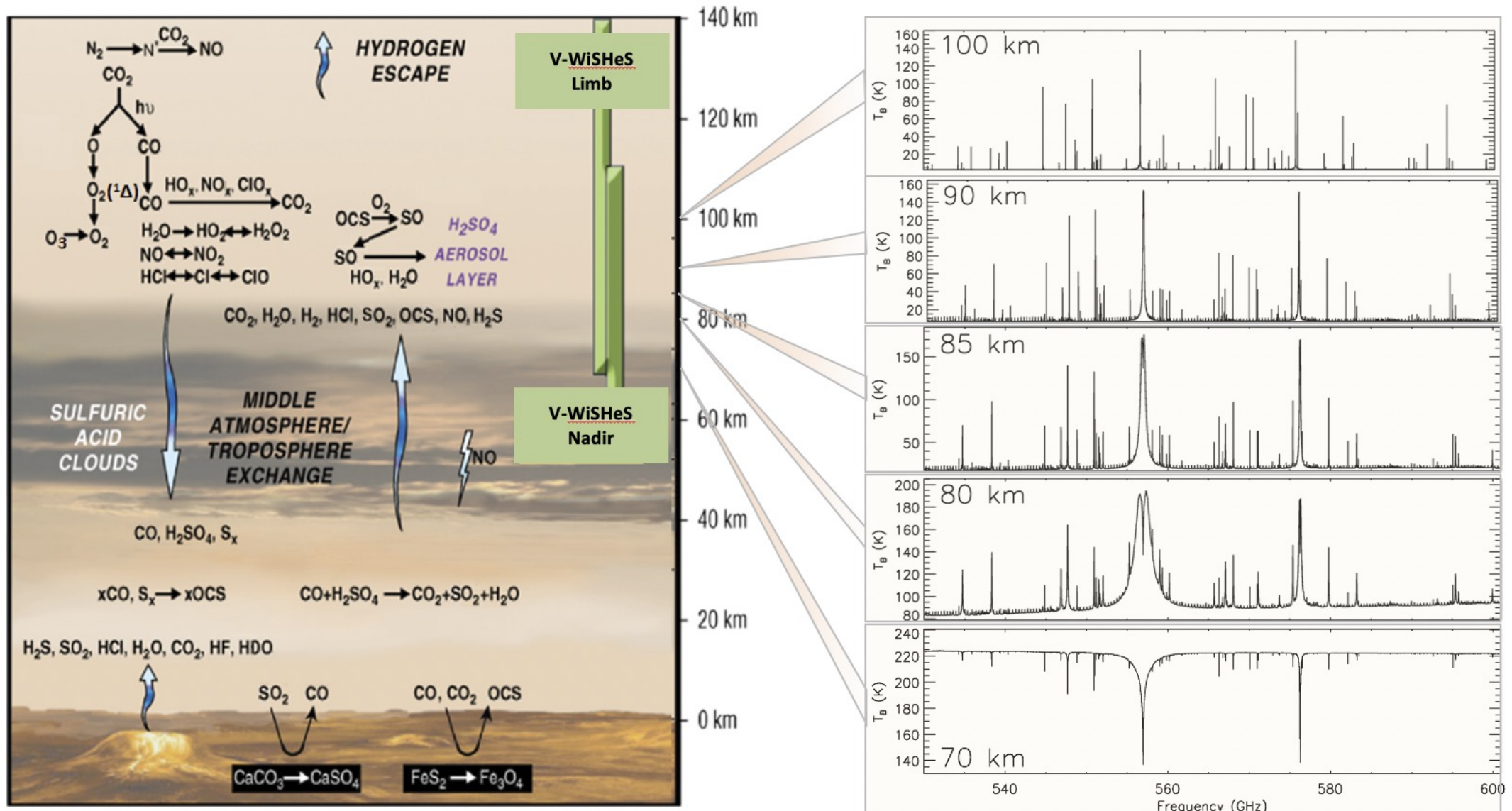
# *V-WiSHeS: Venus Wideband Submillimeter Heterodyne Spectrometer*

**PI: Carrie Anderson**

NASA's Goddard Space Flight Center

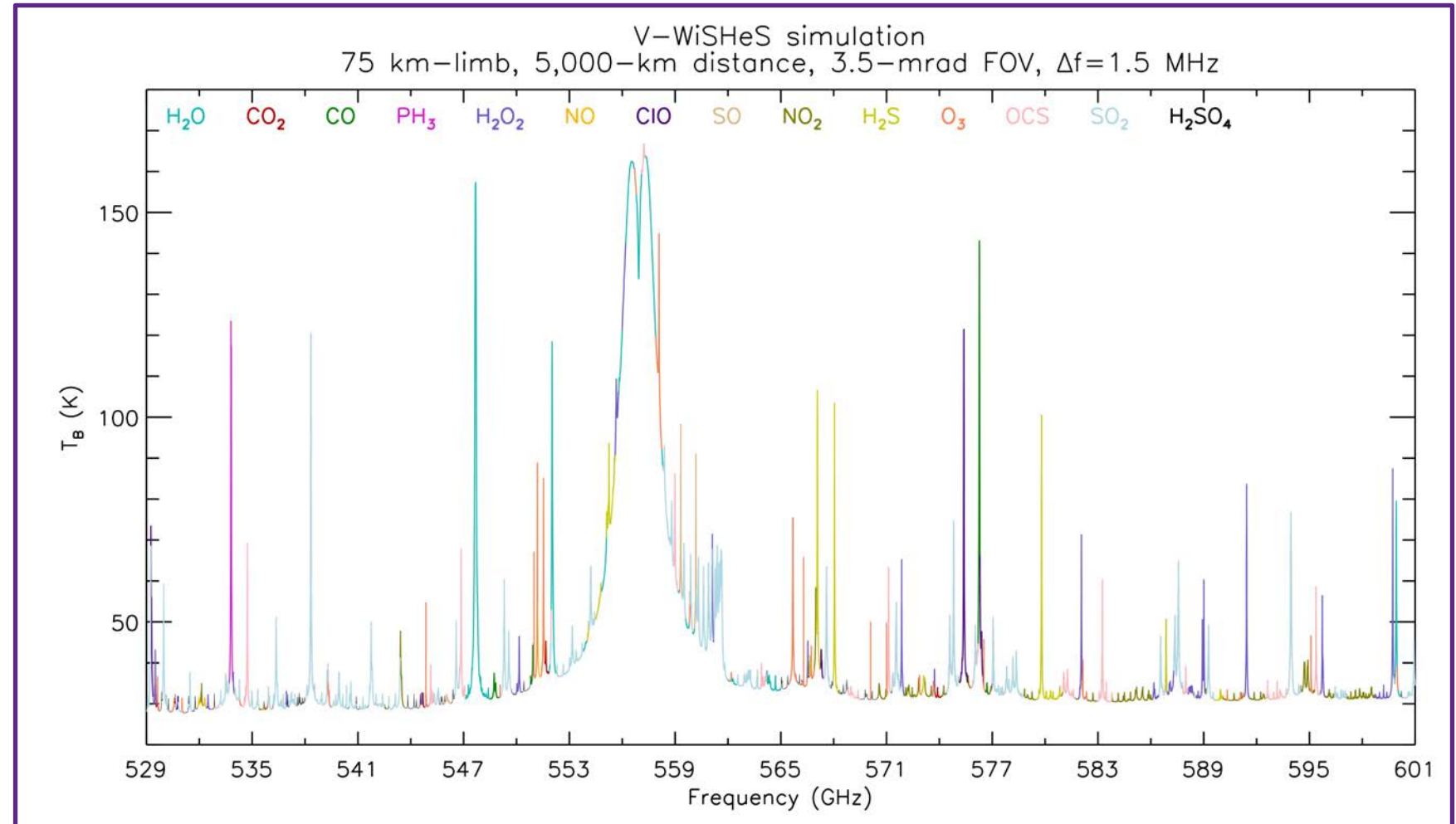


# Venus Atmospheric emission at different tangent heights



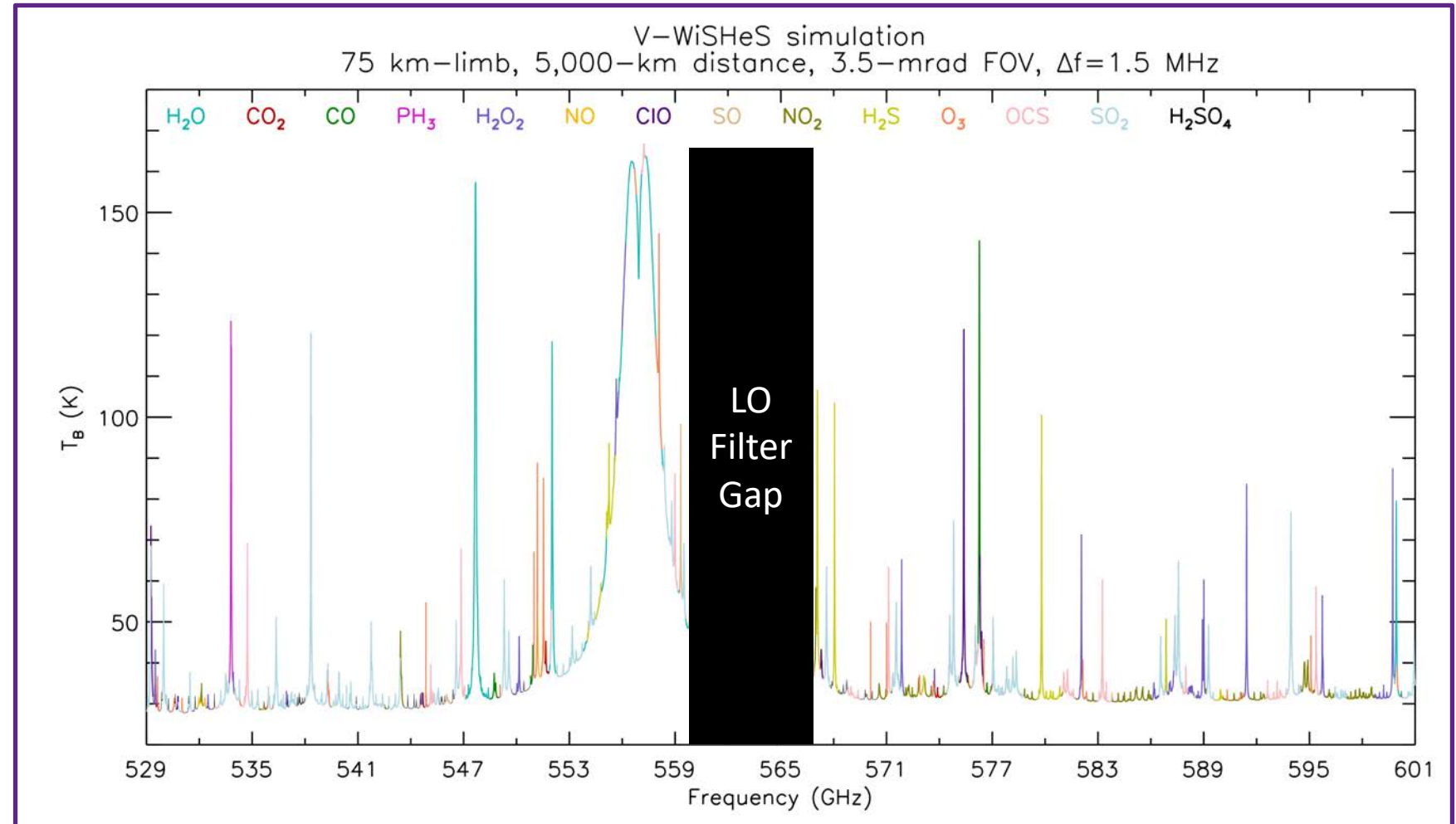
# Science Summary

- V-WiSHeS is optimized to measure trace gases and continuum opacities in Venus' middle atmosphere, enabling spatial and temporal profile measurements of wind, temperature, aerosol opacity, and concurrent isotopic abundances of  $\text{H}_2\text{O}$ ,  $\text{H}_2^{17}\text{O}$ ,  $\text{H}_2^{18}\text{O}$ ,  $\text{HDO}$ ,  $\text{CO}^{18}\text{O}$ ,  $\text{CO}^{17}\text{O}$ ,  $^{13}\text{CO}^{18}\text{O}$ ,  $^{13}\text{CO}^{17}\text{O}$ ,  $\text{CO}$ ,  $^{13}\text{CO}$ ,  $\text{C}^{18}\text{O}$ ,  $\text{O}^{18}\text{O}$ ,  $\text{O}_3$ ,  $\text{OO}^{17}\text{O}$ ,  $\text{OO}^{18}\text{O}$ ,  $\text{H}_2\text{S}$ ,  $\text{H}_2^{34}\text{S}$ ,  $\text{H}_2^{33}\text{S}$ ,  $\text{H}_2\text{SO}_4$ ,  $\text{ClO}$ ,  $^{37}\text{ClO}$ ,  $\text{H}_2\text{O}_2$ ,  $\text{SO}$ ,  $\text{SO}_2$ ,  $^{34}\text{SO}_2$ ,  $\text{OCS}$ ,  $\text{O}^{13}\text{CS}$ ,  $\text{OC}^{34}\text{S}$ ,  $\text{OC}^{33}\text{S}$ ,  $\text{NO}$ , and  $\text{NO}_2$ .

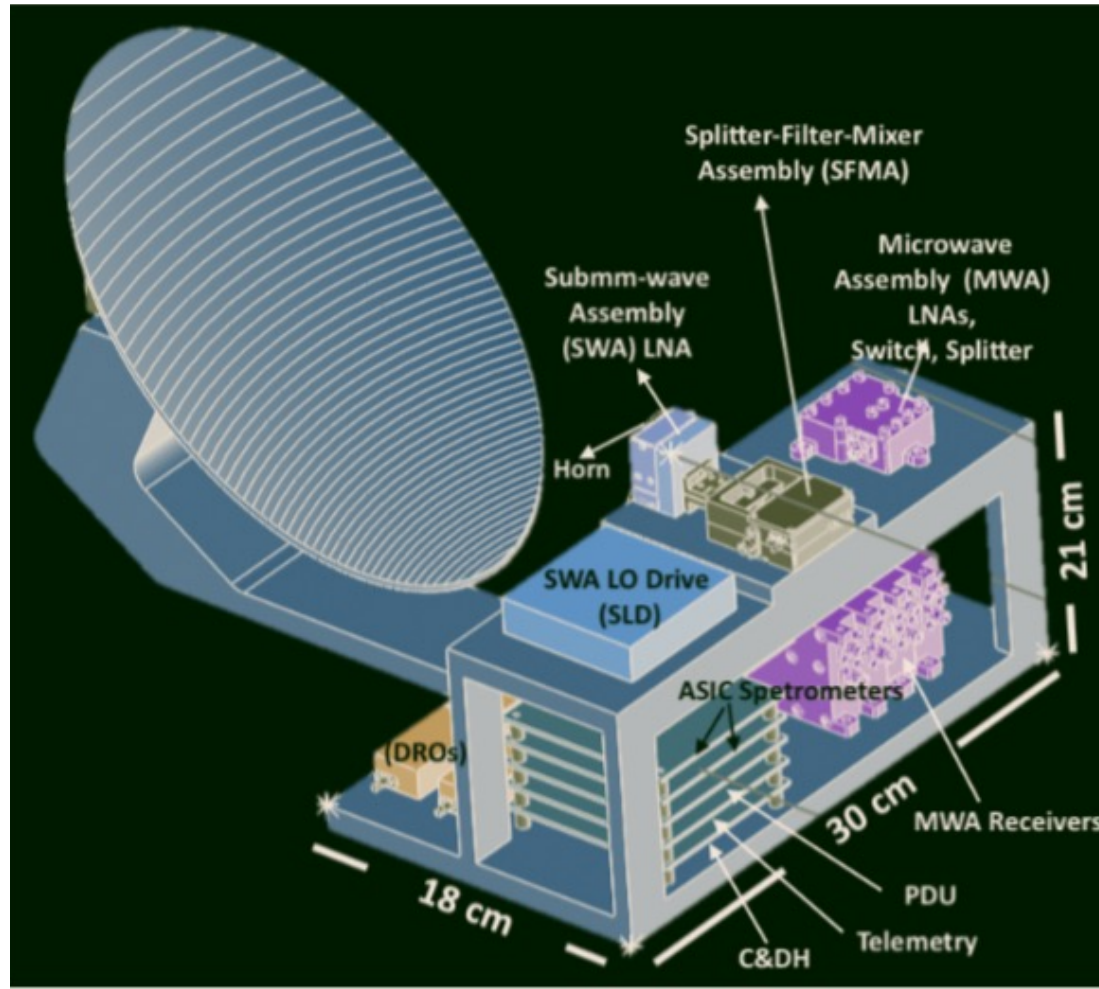


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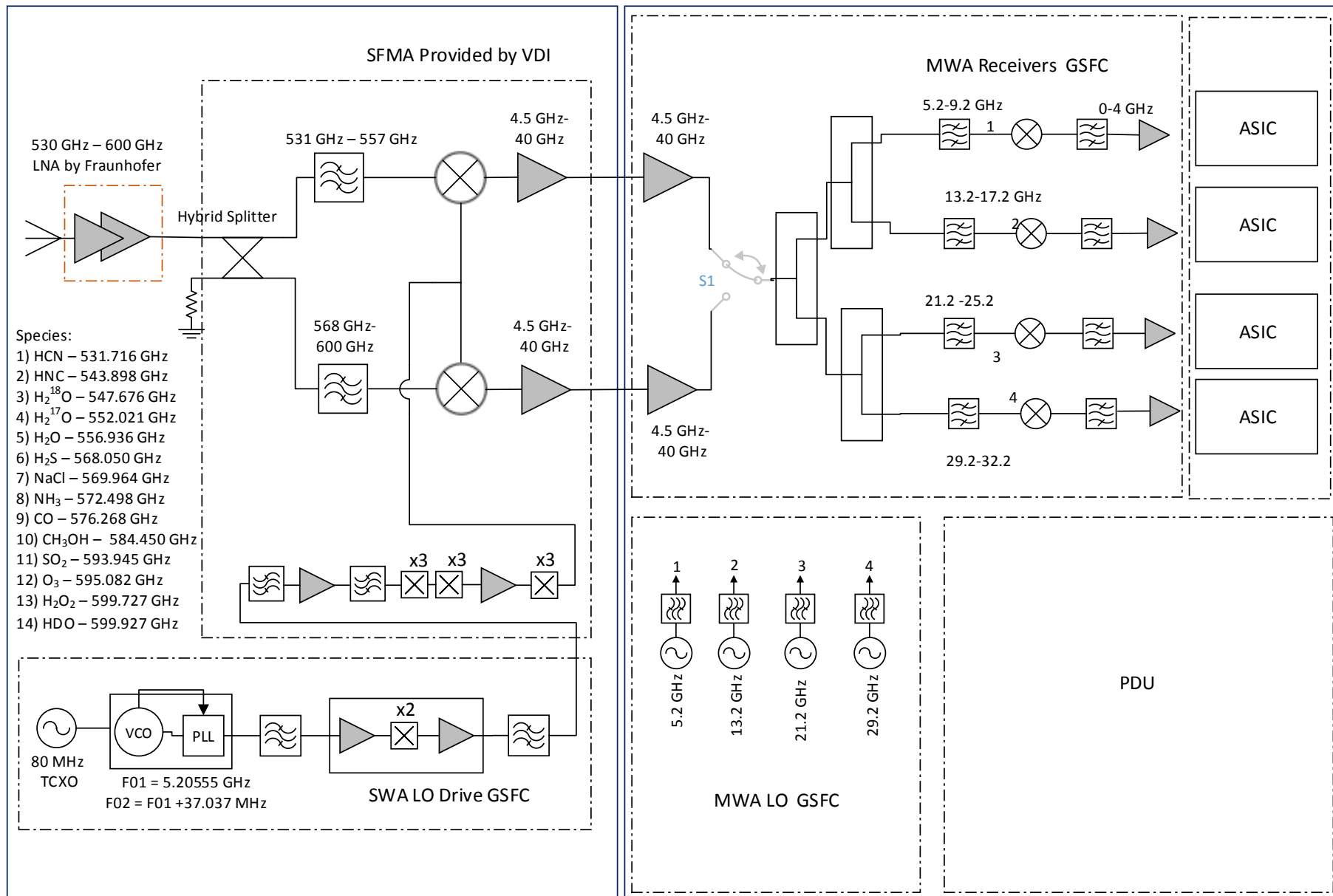
# V-WISHES Architecture & Key Parameters



Total Power	25 W
Mass	12 kg
Spectral Coverage	529 – 600 GHz
Measured Bandwidth	64 GHz
Spectral Resolution	500 kHz
Digital Spectrometer	ASICs with PFB
Receiver Noise Figure	$\leq 12$ dB
Aperture Size	21 cm
Field of View	3.5 mrad
Field of Regard	$> 180$ degrees

# V-WISHES Block Diagram

Submm-Wave Assembly



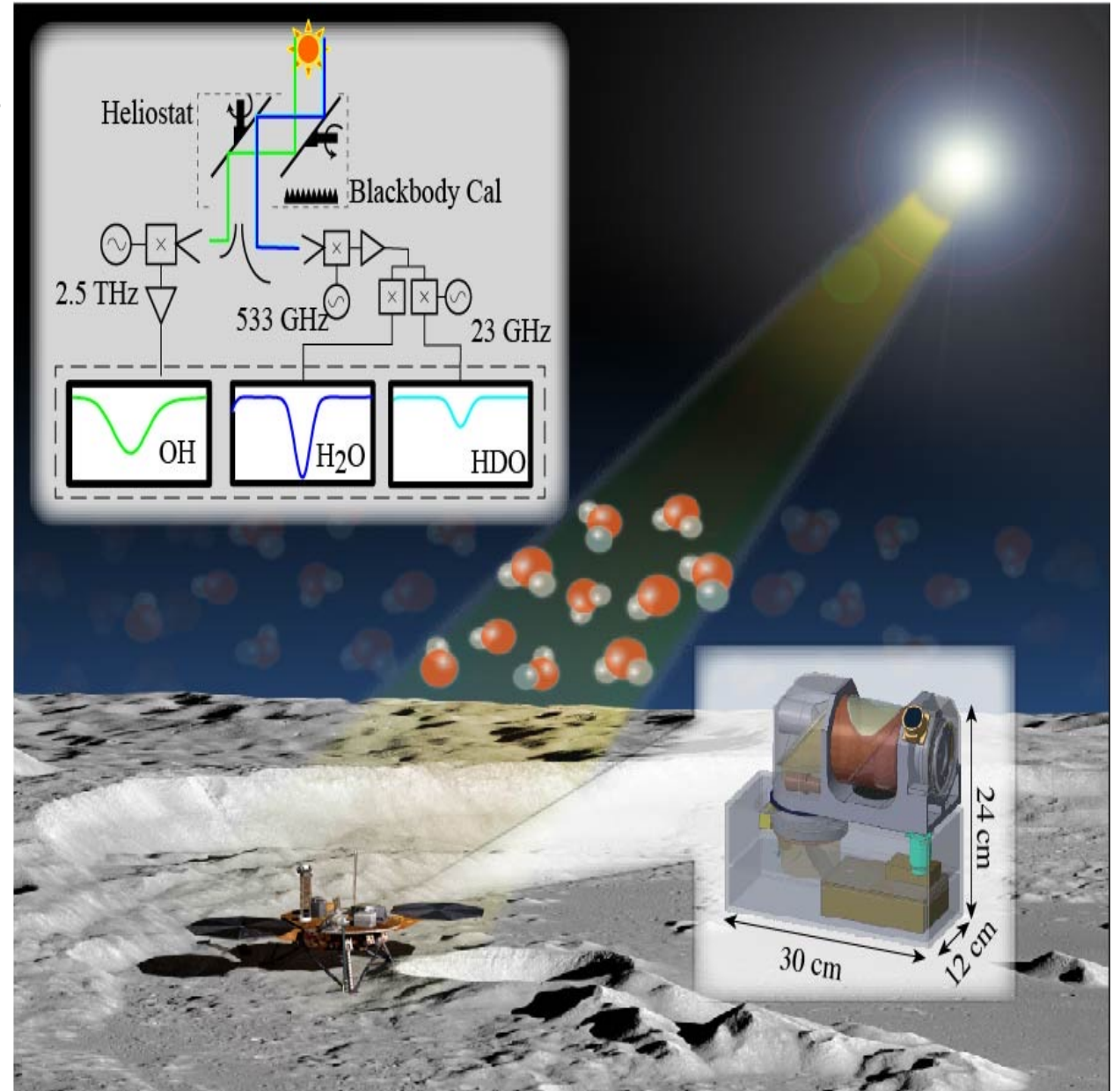


*Solving the mystery of water  
in the Moon's exosphere.*

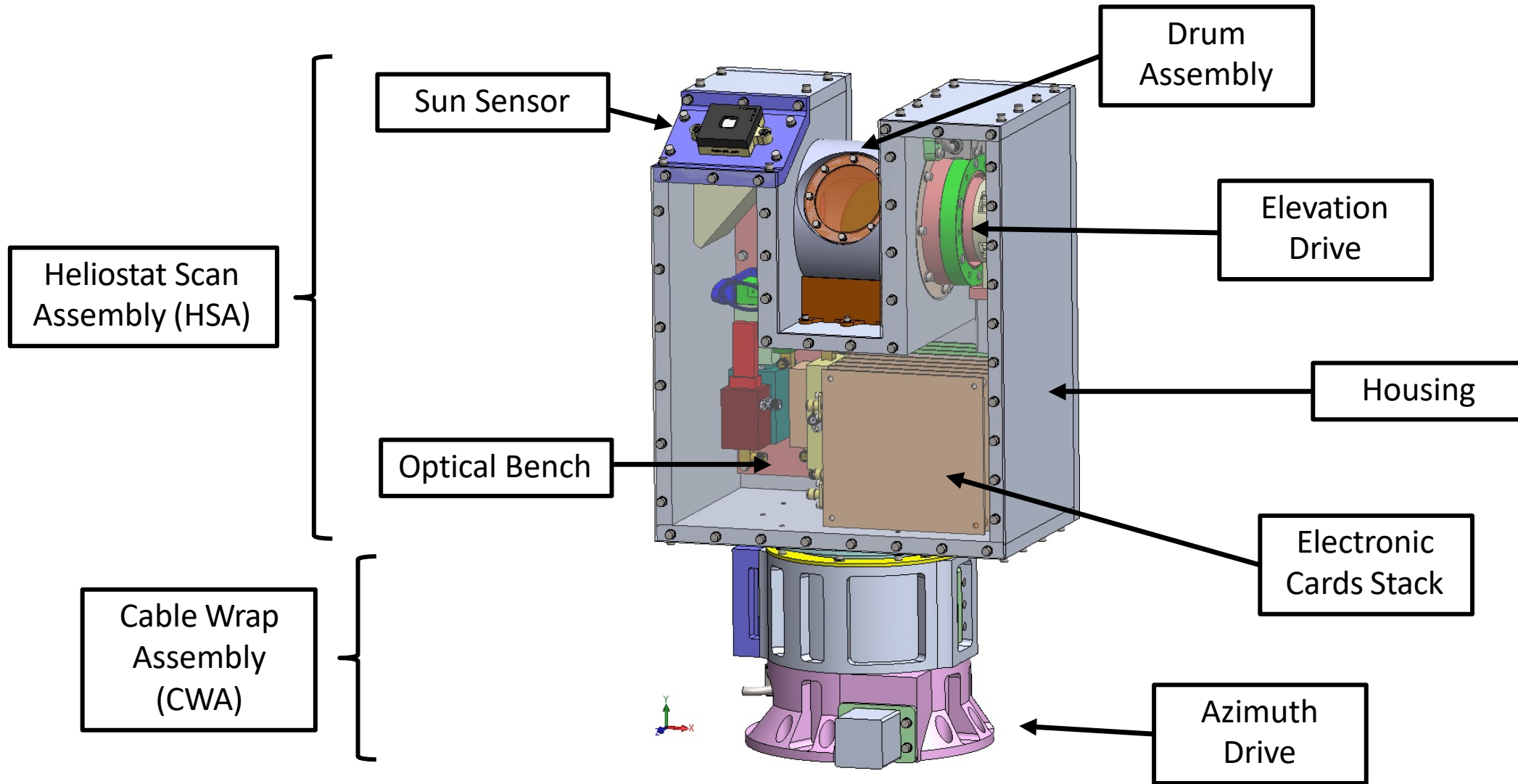
PI: Dr. Tim Livengood/University of Maryland

# SSOLVE Science Measurements

- Measure spectral absorption of water species against Sun as backlight.
  - Absolute species ID
  - Additional physics from line shape, frequency
- Heliostat to track Sun, point to other targets
  - Calibrate internal water against calibration source.
  - Calibrate local water against neighboring surface.

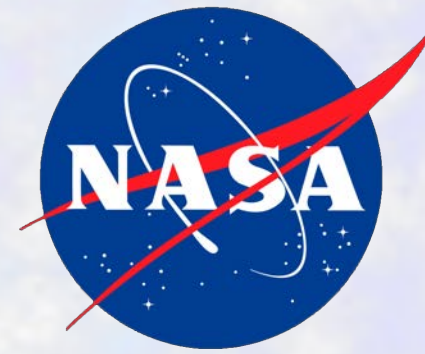


# SSOLVE Instrument Diagram



# SSOLVE TRL Assessment

SSOLVE Technology Entry TRL and Exit TRL					
		Entry TRL		Exit TRL	
	Technology	TRL Description	TRL#	TRL Description	TRL#
1	550 GHz Receiver (B1-2) : extended temperature	<ul style="list-style-type: none"> <li>Laboratory measurements of comparable receiver</li> <li>Operation spans 10-35°C</li> </ul>	5	<ul style="list-style-type: none"> <li>Mixer &amp; LO meet requirements over operational temperature range, 10 – 70 C, TRL – 5</li> <li>Thermal vacuum and vibration test</li> </ul>	6
2	2.5 THz receiver (B3-4)	<ul style="list-style-type: none"> <li>Planar Schottky diode mixer, 1.25 THz Local Oscillator</li> <li>LO multiplier chain</li> </ul>	4	<ul style="list-style-type: none"> <li>NF and conversion loss test over 10 – 70 C – TRL5</li> <li>Thermal vacuum and vibration test</li> </ul>	6
3	Blackbody target	<ul style="list-style-type: none"> <li>Absorbing tiles have good reflectivity from 500 GHz to over 1 THz</li> </ul>	5	<ul style="list-style-type: none"> <li>Characterize reflectivity at SSOLVE frequencies</li> <li>Used for calibrating integrated instrument</li> </ul>	6
4	HelioStat Scan Assembly	<ul style="list-style-type: none"> <li>Design based on available spaceflight components</li> </ul>	4	<ul style="list-style-type: none"> <li>Functional test and optics design validation</li> <li>Thermal vacuum and vibration test</li> </ul>	6
5	SSOLVE Instrument	<ul style="list-style-type: none"> <li>Laboratory testing of 2.5 THz and 550 GHz receivers</li> </ul>	4	<ul style="list-style-type: none"> <li>Demonstration of integrated SSOLVE science measurement in relevant environment.</li> </ul>	6



# Technological Thrusts

- Antenna Technologies
  - Deployable apertures
  - Reflectarrays, active phased arrays
- Size, mass, power and cost (SWaP-C)
  - ASICs
  - Highly integrated receivers, MMICs
  - Advanced manufacturing (e.g. 3D printing, polystrata)
- Performance
  - Spectral coverage
  - Calibration
  - Smart Sensors



Thank You!