

# Water Chemistry for Space Exploration

*at NASA's Kennedy Space Center*

National Aeronautics and  
Space Administration



**Luke Roberson, Ph.D**

Senior Principal Investigator

Exploration Research/UB-G

NASA Kennedy Space Center, FL



Presented to the  
**NETSACS**  
February 24, 2022

# The Artemis Program

Artemis is the twin sister of Apollo and goddess of the Moon in Greek mythology. Now, she personifies our path to the Moon as the name of NASA's program to return astronauts to the lunar surface.

When they land, Artemis astronauts will step foot where no human has ever been before: the Moon's South Pole.

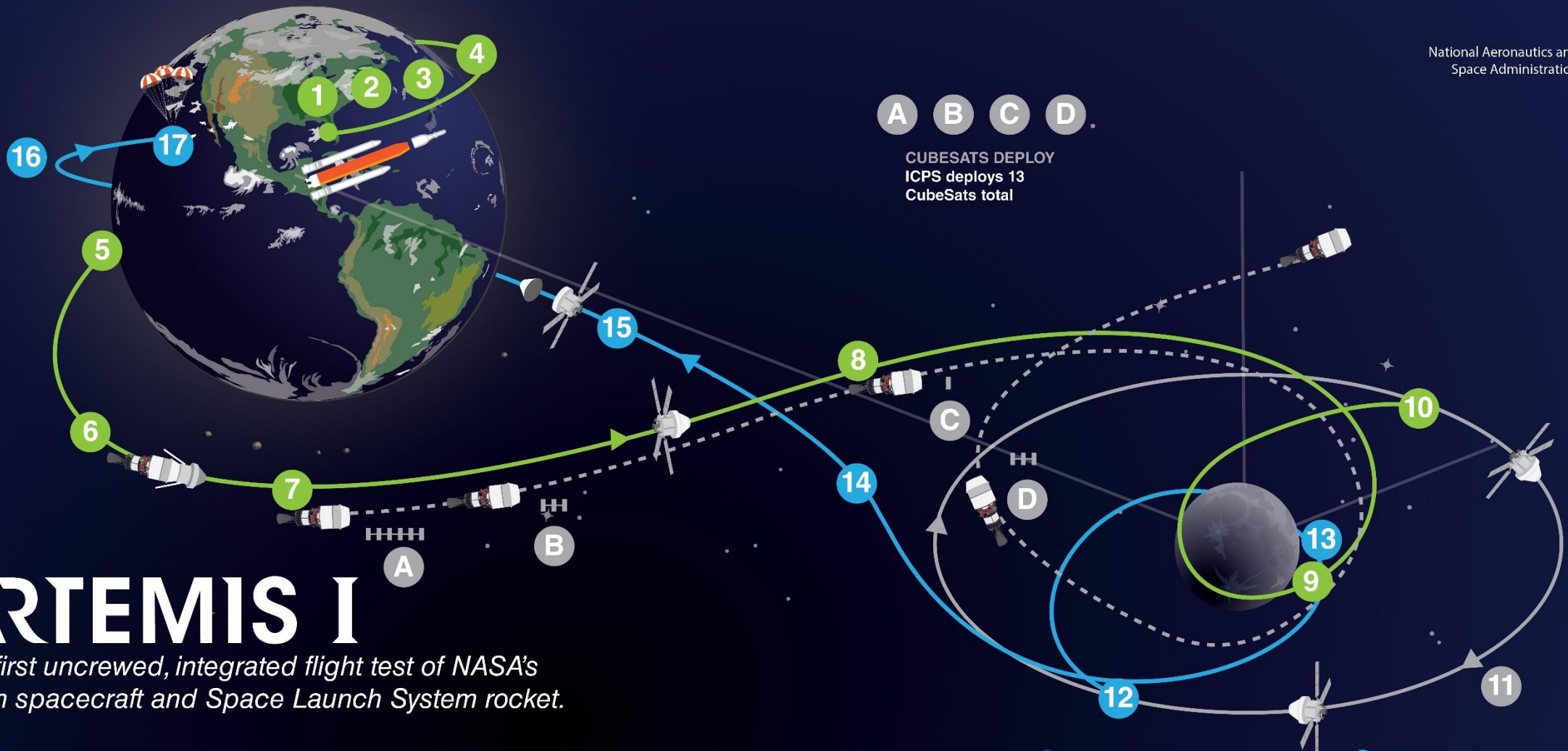
With the horizon goal of sending humans to Mars, Artemis begins the next era of exploration.





A B C D

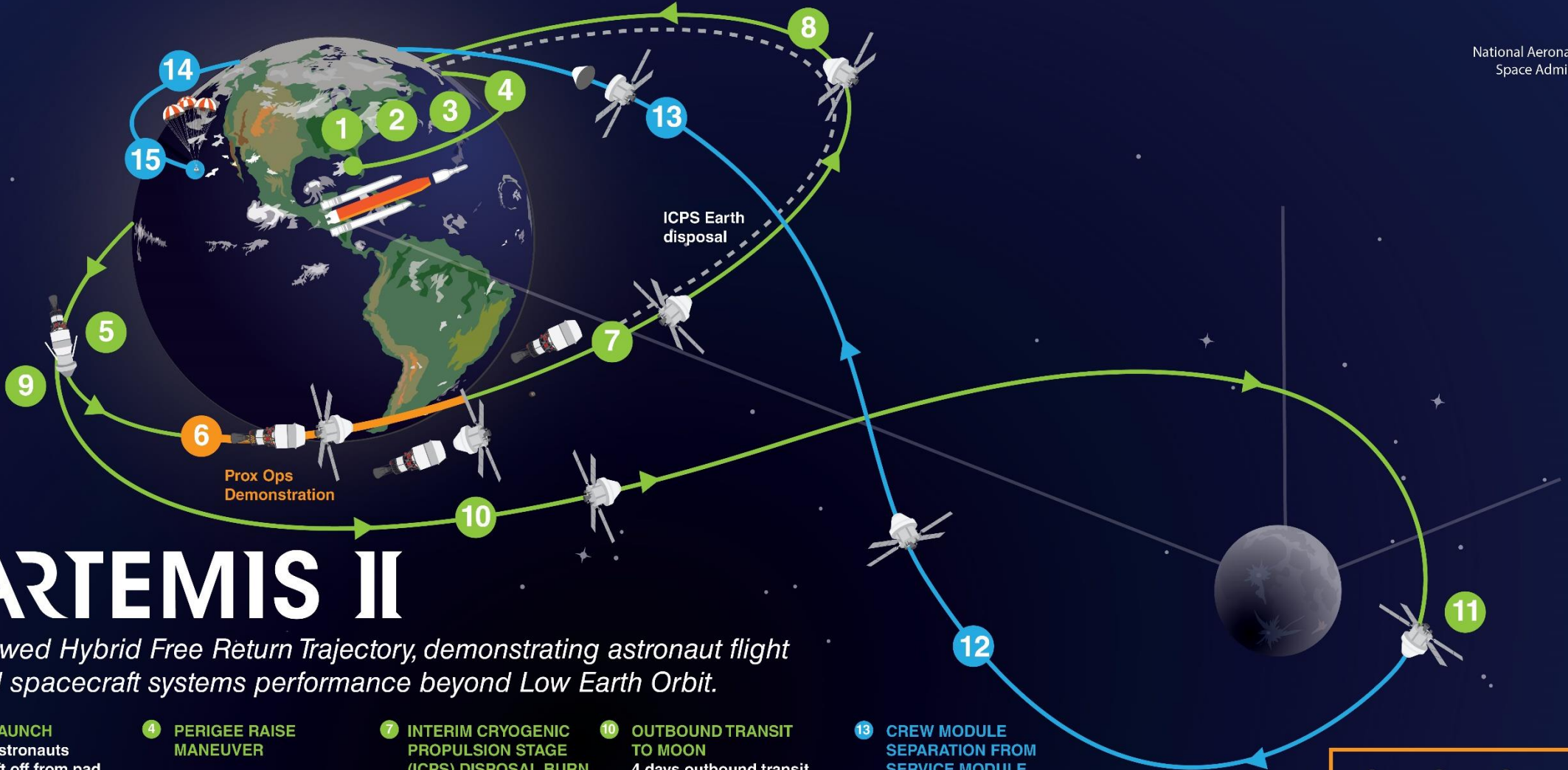
CUBESATS DEPLOY  
ICPS deploys 13  
CubeSats total



# ARTEMIS I

The first uncrewed, integrated flight test of NASA's Orion spacecraft and Space Launch System rocket.

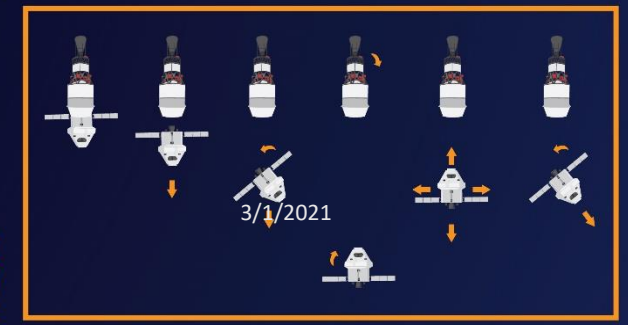
- 1 LAUNCH**  
SLS and Orion lift off from pad 39B at Kennedy Space Center.
- 2 JETTISON ROCKET BOOSTERS, FAIRINGS, AND LAUNCH ABORT SYSTEM**
- 3 CORE STAGE MAIN ENGINE CUT OFF**  
With separation.
- 4 PERIGEE RAISE MANEUVER**
- 5 EARTH ORBIT**  
Systems check with solar panel adjustments.
- 6 TRANS LUNAR INJECTION (TLI) BURN**  
Maneuver lasts for approximately 20 minutes.
- 7 INTERIM CRYOGENIC PROPULSION STAGE (ICPS) SEPARATION AND DISPOSAL**  
The ICPS has committed Orion to TLI.
- 8 OUTBOUND TRAJECTORY CORRECTION (OTC) BURNS**  
As necessary adjust trajectory for lunar flyby to Distant Retrograde Orbit (DRO).
- 9 OUTBOUND POWERED FLYBY (OPF)**  
60 nmi from the Moon; targets DRO insertion.
- 10 LUNAR ORBIT INSERTION**  
Enter Distant Retrograde Orbit for next 6-23 days.
- 11 DISTANT RETROGRADE ORBIT**  
Perform half or one and a half revolutions in the 12 day orbit period 38,000 nmi from the surface of the Moon.
- 12 DRO DEPARTURE**  
Leave DRO and start return to Earth.
- 13 RETURN POWER FLY-BY (RPF)**  
RPF burn prep and return coast to Earth initiated.
- 14 RETURN TRANSIT**  
Return Trajectory Correction (RTC) burns as necessary to aim for Earth's atmosphere; travel time 5-11 days.
- 15 CREW MODULE SEPARATION FROM SERVICE MODULE**
- 16 ENTRY INTERFACE (EI)**  
Enter Earth's atmosphere.
- 17 SPLASHDOWN**  
Pacific Ocean landing within view of the U.S. Navy recovery ship. 3/1/2021



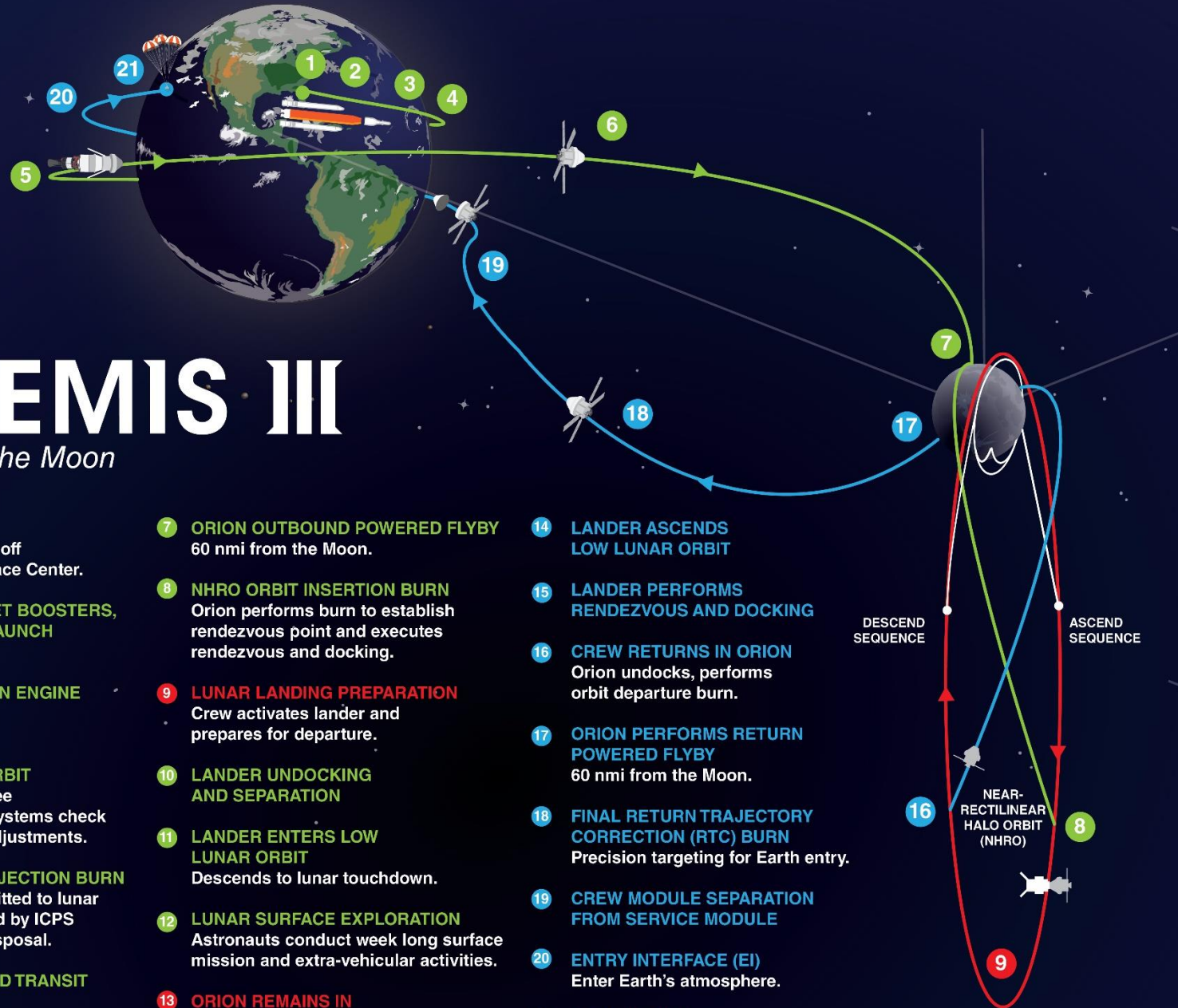
# ARTEMIS II

*Crewed Hybrid Free Return Trajectory, demonstrating astronaut flight and spacecraft systems performance beyond Low Earth Orbit.*

- 1 LAUNCH**  
Astronauts lift off from pad 39B at Kennedy Space Center.
- 2 JETTISON ROCKET BOOSTERS, FAIRINGS, AND LAUNCH ABORT SYSTEM**
- 3 CORE STAGE MAIN ENGINE CUT OFF**  
With separation.
- 4 PERIGEE RAISE MANEUVER**
- 5 APOGEE RAISE BURN TO HIGH EARTH ORBIT**  
Begin 42 hour checkout of spacecraft.
- 6 PROX OPS DEMONSTRATION**  
Orion proximity operations demonstration and manual handling qualities assessment for up to 2 hours.
- 7 INTERIM CRYOGENIC PROPULSION STAGE (ICPS) DISPOSAL BURN**
- 8 HIGH EARTH ORBIT CHECKOUT**  
Life support, exercise, and habitation equipment evaluations.
- 9 TRANS-LUNAR INJECTION (TLI) BY ORION'S MAIN ENGINE**
- 10 OUTBOUND TRANSIT TO MOON**  
4 days outbound transit along free return trajectory.
- 11 LUNAR FLYBY**  
4,000 nmi (mean) lunar farside altitude.
- 12 TRANS-EARTH RETURN**  
Return Trajectory Correction (RTC) burns as necessary to aim for Earth's atmosphere; travel time approximately 4 days.
- 13 CREW MODULE SEPARATION FROM SERVICE MODULE**
- 14 ENTRY INTERFACE (EI)**  
Enter Earth's atmosphere.
- 15 SPLASHDOWN**  
Astronaut and capsule recovery by U.S. Navy ship.



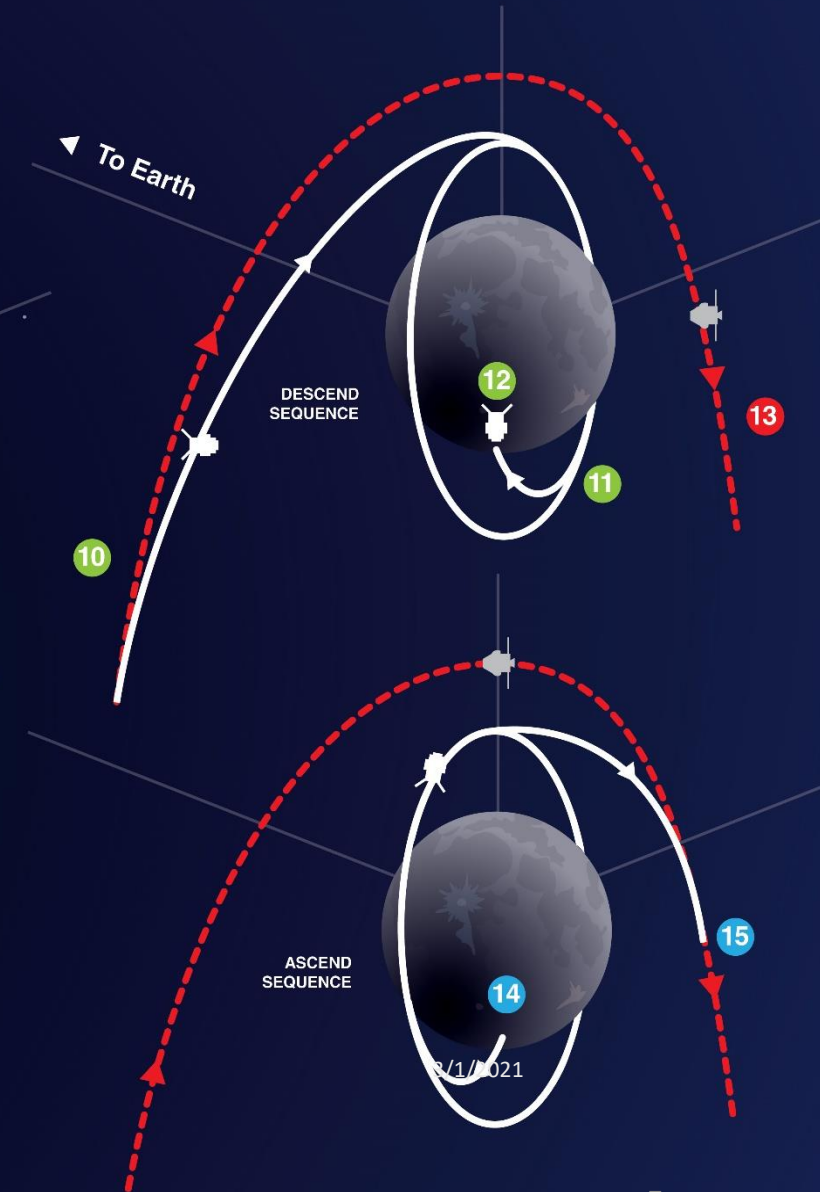
**PROXIMITY OPERATIONS DEMONSTRATION SEQUENCE**



# ARTEMIS III

## Landing on the Moon

- 1 LAUNCH**  
SLS and Orion lift off from Kennedy Space Center.
- 2 JETTISON ROCKET BOOSTERS, FAIRINGS, AND LAUNCH ABORT SYSTEM**
- 3 CORE STAGE MAIN ENGINE CUT OFF**  
With separation.
- 4 ENTER EARTH ORBIT**  
Perform the perigee raise maneuver. Systems check and solar panel adjustments.
- 5 TRANS LUNAR INJECTION BURN**  
Astronauts committed to lunar trajectory, followed by ICPS separation and disposal.
- 6 ORION OUTBOUND TRANSIT TO MOON**  
Requires several outbound trajectory burns.
- 7 ORION OUTBOUND POWERED FLYBY**  
60 nmi from the Moon.
- 8 NHRO ORBIT INSERTION BURN**  
Orion performs burn to establish rendezvous point and executes rendezvous and docking.
- 9 LUNAR LANDING PREPARATION**  
Crew activates lander and prepares for departure.
- 10 LANDER UNDOCKING AND SEPARATION**
- 11 LANDER ENTERS LOW LUNAR ORBIT**  
Descends to lunar touchdown.
- 12 LUNAR SURFACE EXPLORATION**  
Astronauts conduct week long surface mission and extra-vehicular activities.
- 13 ORION REMAINS IN NHRO ORBIT**  
During lunar surface mission.
- 14 LANDER ASCENDS LOW LUNAR ORBIT**
- 15 LANDER PERFORMS RENDEZVOUS AND DOCKING**
- 16 CREW RETURNS IN ORION**  
Orion undocks, performs orbit departure burn.
- 17 ORION PERFORMS RETURN POWERED FLYBY**  
60 nmi from the Moon.
- 18 FINAL RETURN TRAJECTORY CORRECTION (RTC) BURN**  
Precision targeting for Earth entry.
- 19 CREW MODULE SEPARATION FROM SERVICE MODULE**
- 20 ENTRY INTERFACE (EI)**  
Enter Earth's atmosphere.
- 21 SPLASHDOWN**  
Astronaut and capsule recovery by U.S. Navy ship.

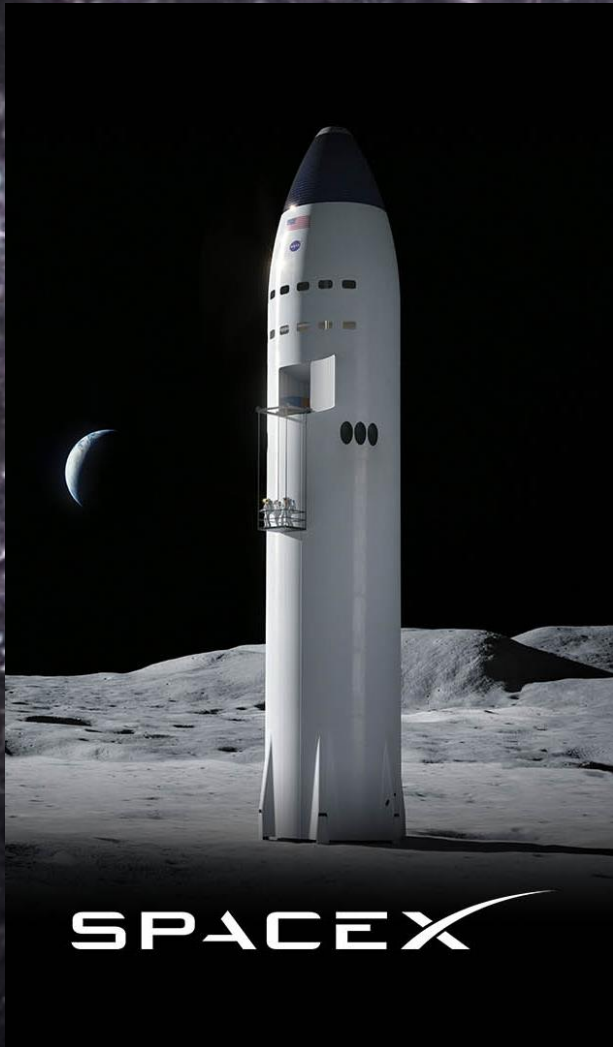


# Gateway International Partners

*Building on ISS partnerships to expand deep space capabilities*



# THE HUMAN LANDING SYSTEM



# Mission Needs Drive Design

## LOW EARTH RETURN

**3 HOURS**

**3,000°F**

**17,500 MPH**

**250 MILES**



## LUNAR RETURN

**3 DAYS**

**5,200°F**

**24,700 MPH**

**240,000 MILES**



## MARS RETURN

**9 MONTHS**

**6,200°F**

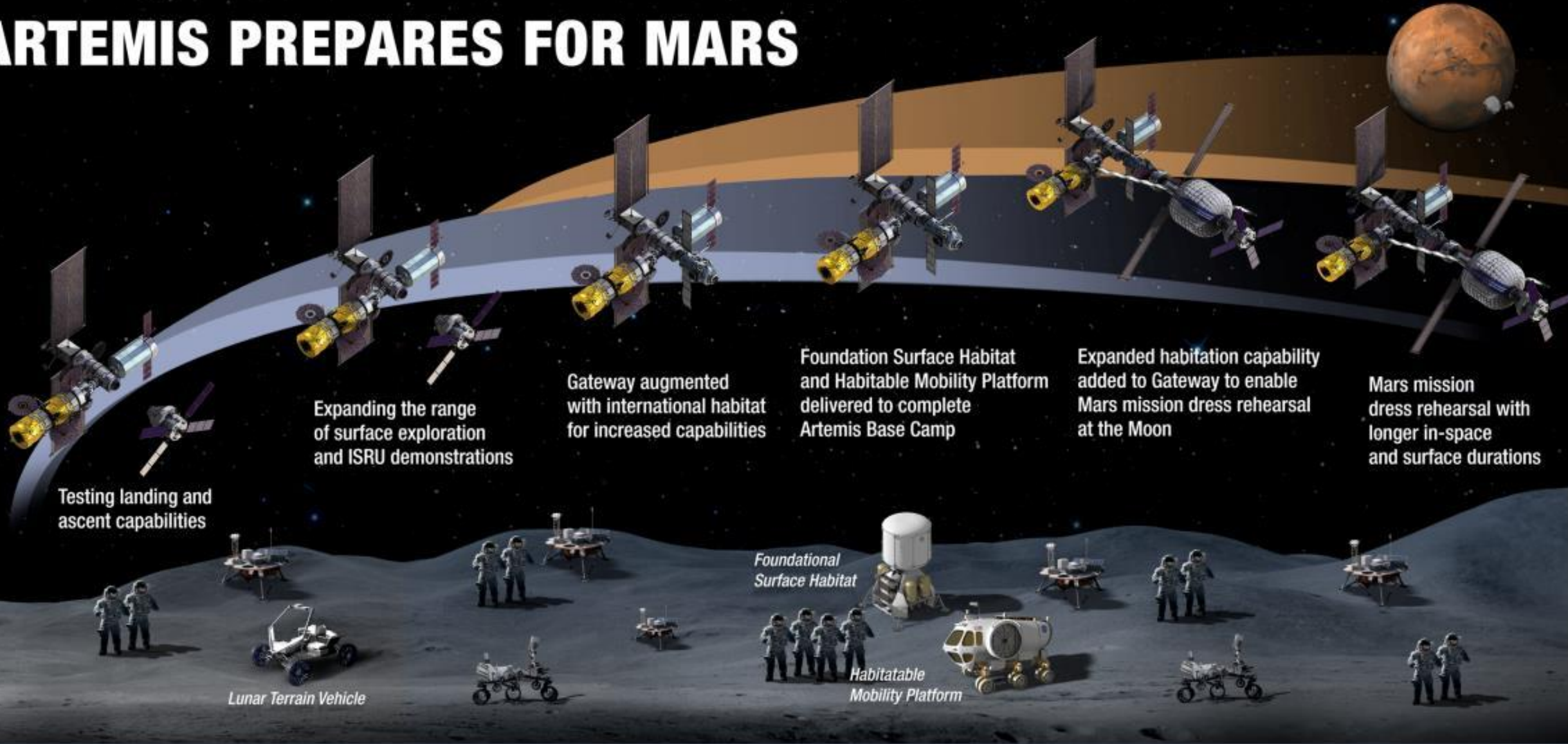
**26,800 MPH**

**39,000,000 MILES**





# ARTEMIS PREPARES FOR MARS



## SUSTAINABLE LUNAR ORBIT STAGING CAPABILITY AND SURFACE EXPLORATION

MULTIPLE SCIENCE AND CARGO PAYLOADS | INTERNATIONAL PARTNERSHIP OPPORTUNITIES | TECHNOLOGY AND OPERATIONS DEMONSTRATIONS FOR MARS

# Humans on Mars

*Pushing the Boundaries  
of Current Possibilities*



## Go

*Rapid, safe, & efficient  
space transportation*

## Land

*Expanded access to  
diverse surface  
destinations*

## Live

*Sustainable living  
and working farther  
from Earth*

## Explore

*Transformative missions  
and discoveries*

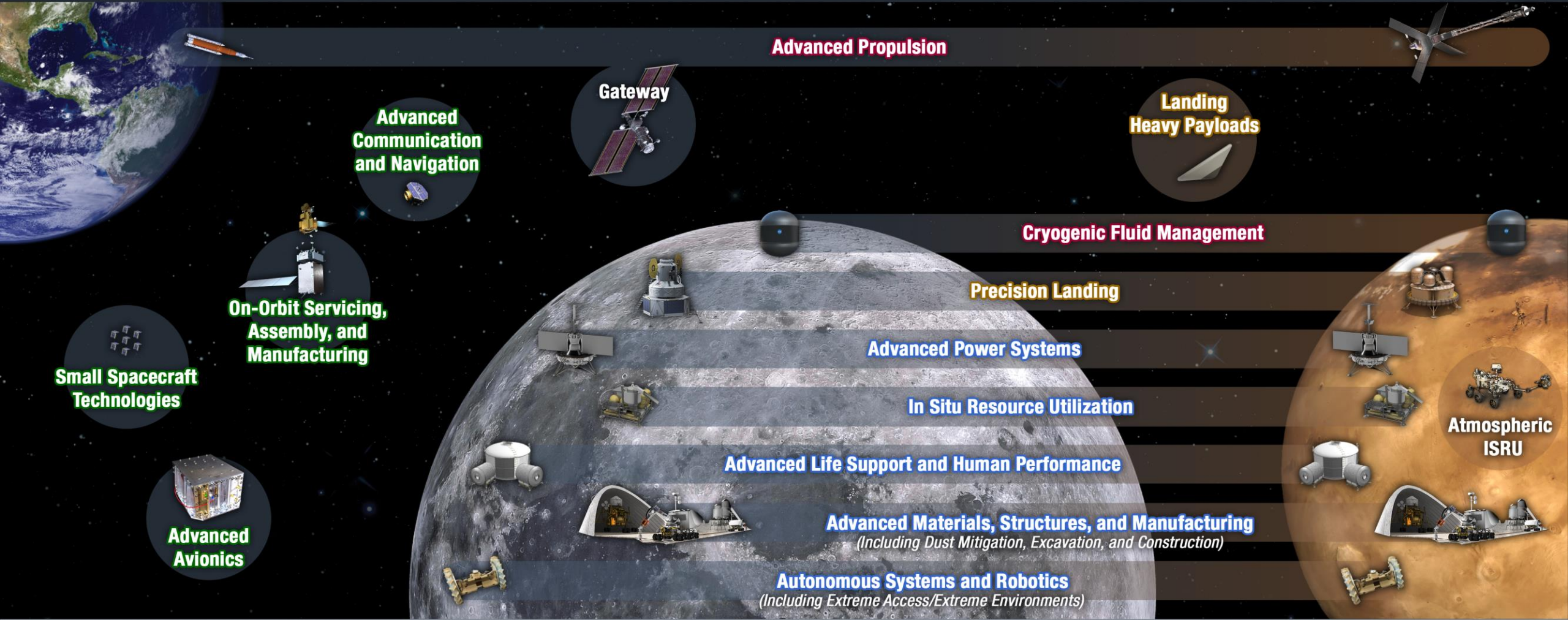
# TECHNOLOGY DRIVES EXPLORATION

**Rapid, Safe, and Efficient  
Space Transportation**

**Expanded Access to Diverse  
Surface Destinations**

**Sustainable Living and Working  
Farther from Earth**

**Transformative Missions  
and Discoveries**



**Advanced Propulsion**

**Gateway**

**Advanced  
Communication  
and Navigation**

**Landing  
Heavy Payloads**

**On-Orbit Servicing,  
Assembly, and  
Manufacturing**

**Small Spacecraft  
Technologies**

**Advanced  
Avionics**

**Cryogenic Fluid Management**

**Precision Landing**

**Advanced Power Systems**

**In Situ Resource Utilization**

**Atmospheric  
ISRU**

**Advanced Life Support and Human Performance**

**Advanced Materials, Structures, and Manufacturing**  
*(Including Dust Mitigation, Excavation, and Construction)*

**Autonomous Systems and Robotics**  
*(Including Extreme Access/Extreme Environments)*

2020

**GO | LAND | LIVE | EXPLORE**

203X

### Mars Ascent Vehicle

A landing pad made out of 3-D printed regolith will keep the MAV from blasting a big hole with its rockets. The MAV will not have ascent fuel onboard when it arrives. By reacting carbon dioxide and hydrogen, methane can be made to fuel the MAV back off the Martian surface.



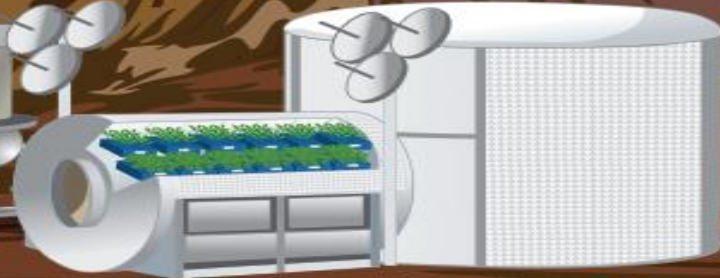
### Processor

In a reactor, water will be extracted from regolith and combined with carbon dioxide to make drinking water, breathing air, and propellants like oxygen and methane.



### Plant Habitat

Water that has been processed from the Martian surface, along with the proper nutrient blend, can be used for growing plants for astronauts to eat. Plants also purify water and produce oxygen from respired carbon dioxide.



### Cryogenic Storage

Once the propellants have been extracted from the resources they must be safely stored as high-density cryogenic liquids for future use.



### Human Habitat

Oxygen extracted from the soil and atmosphere can be used for breathable air and shields made from regolith or water may be used to help protect against radiation.



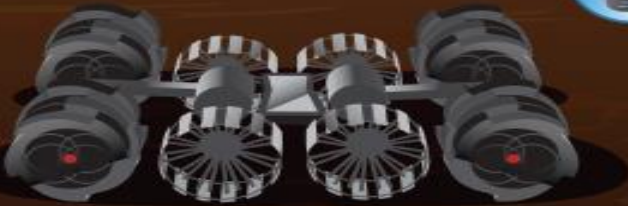
### Prospector

The prospector will drill to find resources buried in the Martian soil, or regolith.



### Miner

A robot will mine the regolith to obtain the resources locked inside.



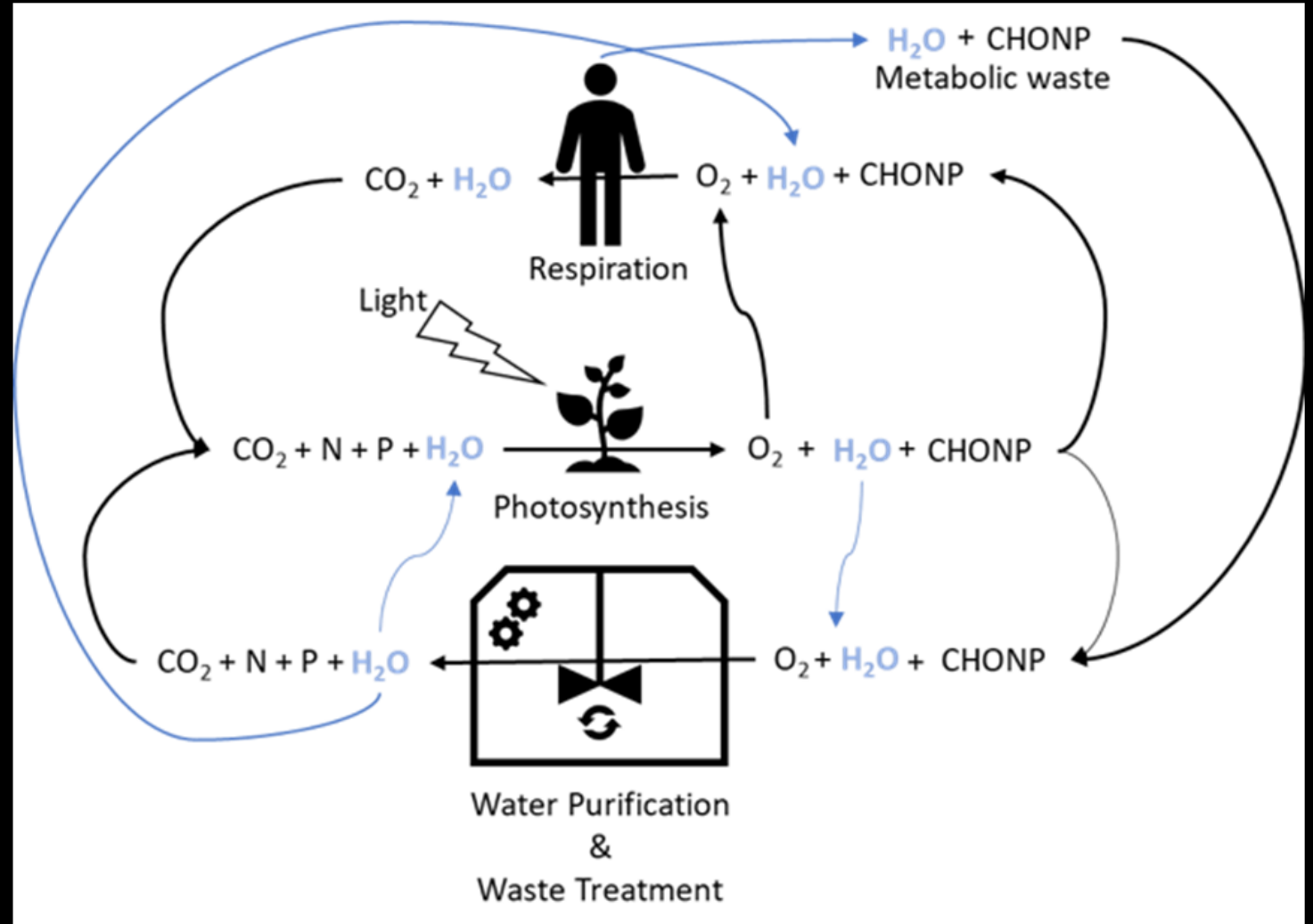
# 2020 NASA Technology Taxonomy (TX)

## ECLSS Driven Challenges

	Goals	Challenges
TX 6.1.2 Water Recovery and Management	Increase <b>re-useable</b> water recovered from sources	Production of water with <b>minimal expendables</b> and maintenance <b>Tolerance to dormancy</b> Recovery of water from <b>complex</b> mixture of inorganic and <b>organic sources</b>
TX 6.1.3 Waste Management	Enable the utilization of <b>solid</b> and <b>liquid</b> metabolic wastes and <b>trash</b>	<b>Effective</b> separation and <b>treatment</b> of metabolic liquid and <b>solid waste</b>
TX 6.3.5 Food Production, Processing, and Preservation	Reduce food resupply requirement	<b>Sustainable food growth</b> , processing, and preparation

# ECLSS Requirements

- For a 30-month mission, a single Crew Member (CM) will require:
  - 2250 kg water
  - 1359 kg food
- And generate:
  - 5678 kg total waste
    - 1612 kg metabolic waste
- Exceeds \$16M/CM at a payload cost of \$10,000/lb (~\$4,535/kg)

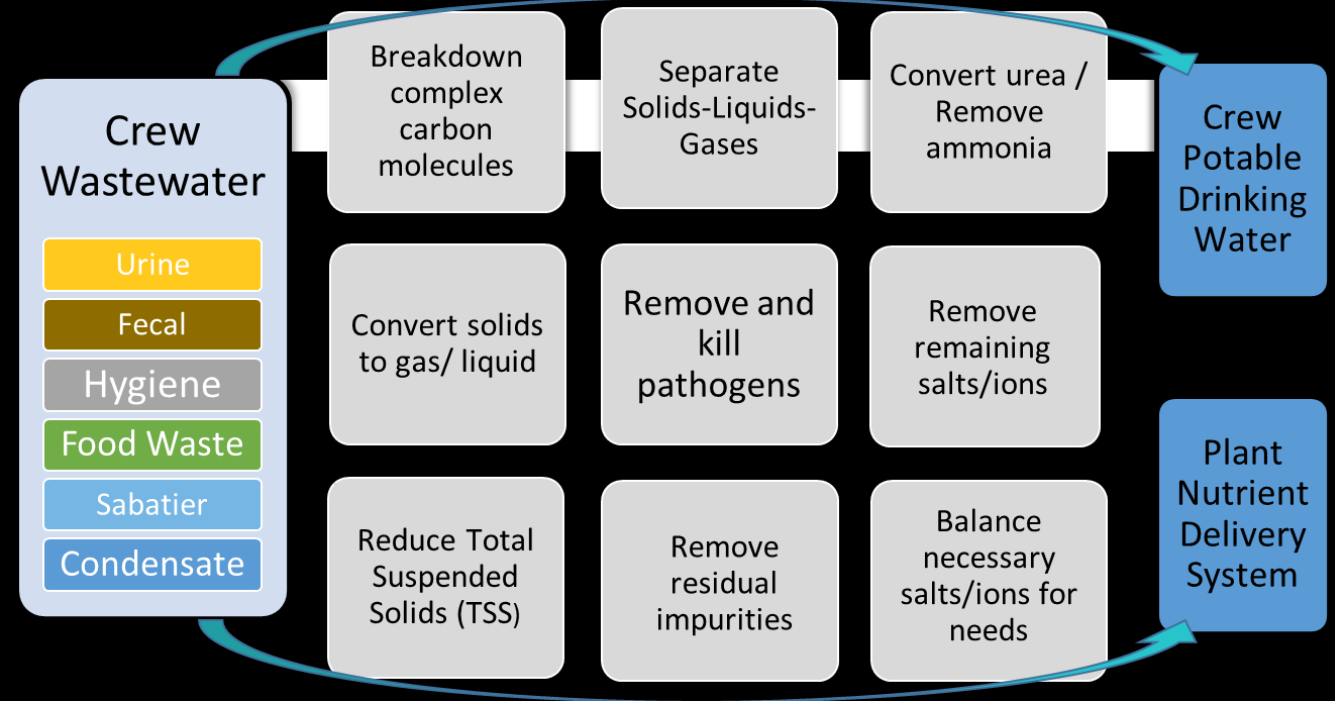


Adapted from Wheeler, R. M. (2003). CARBON BALANCE IN BIOREGENERATIVE LIFE SUPPORT SYSTEMS: Some Effects of System Closure, Waste Management, and Crop Harvest Index. *Adv. Space Res.*, 31(1), 7.

# Next-generation ECLSS

- Harvest and recycle not just water, but all elements (particularly C, H, O, N, P, K, S, Ca, Mg)
- Treat all organic matter as a resource – not waste
- Integrate water recycling, plant production, atmospheric revitalization, and waste management
- Need to be effective, reliable, robust, resilient, and safe
- Be bioregenerative – no consumables or resupply

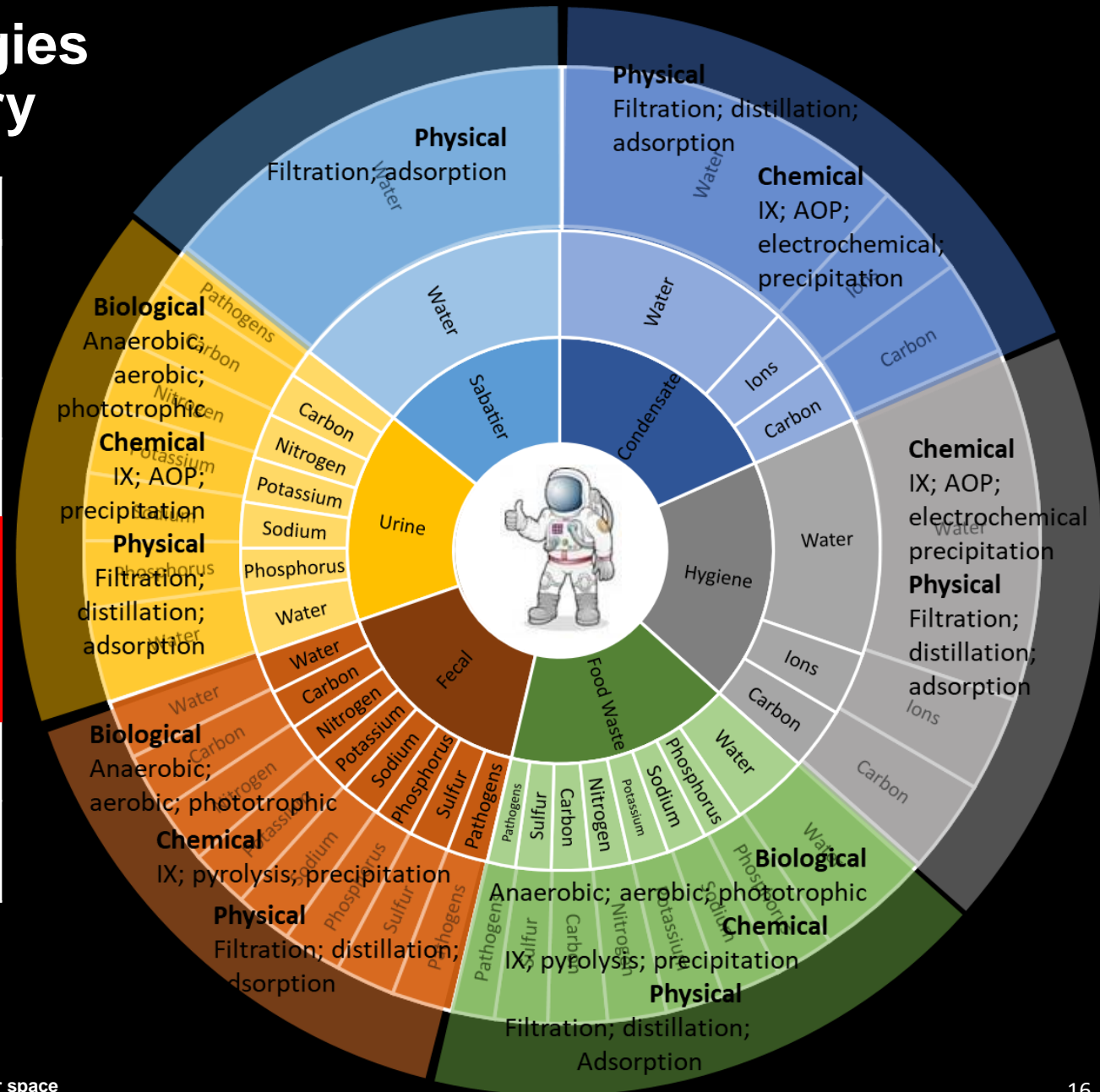
## Functional Flow Diagram



**What are the enabling technologies?**

# Primary Wastewater Streams: Constituents and potential technologies for treatment and resource recovery

Process Stream	Description	Constituents by type										Constituents by size		
		C	N	P	K	S	Na	Ca	Mg	Other ions	Pathogens	Particulate	Colloidal	Dissolved
<b>Food Waste</b>	High content of complex particulate OM, fibers, COD	4	3	3	2	3	2	2	2	2	2	4	4	3
<b>Fecal</b>	High content of complex particulate OM, fibers, COD, pathogens	4	3	3	2	3	2	2	2	2	4	4	4	3
<b>Urine</b>	Mod COD, organic N, urea, ammonium, phosphate, other salts	3	4	4	4	2	3	2	2	3	2	0	2	4
<b>Hygiene</b>	Low COD, constituents from skin and body secretions, environmental contaminants	2	2	1	1	1	2	1	1	2	2	2	2	3
<b>Humidity</b>	Mostly pure condensate, with contaminants from air and surfaces	1	1	0	1	0	1	0	0	1	1	0	0	2
<b>Sabatier</b>	Pure, potentially aggressive, can use for dilution, regeneration, backwash	0	0	0	0	0	0	0	0	1	1	0	0	0

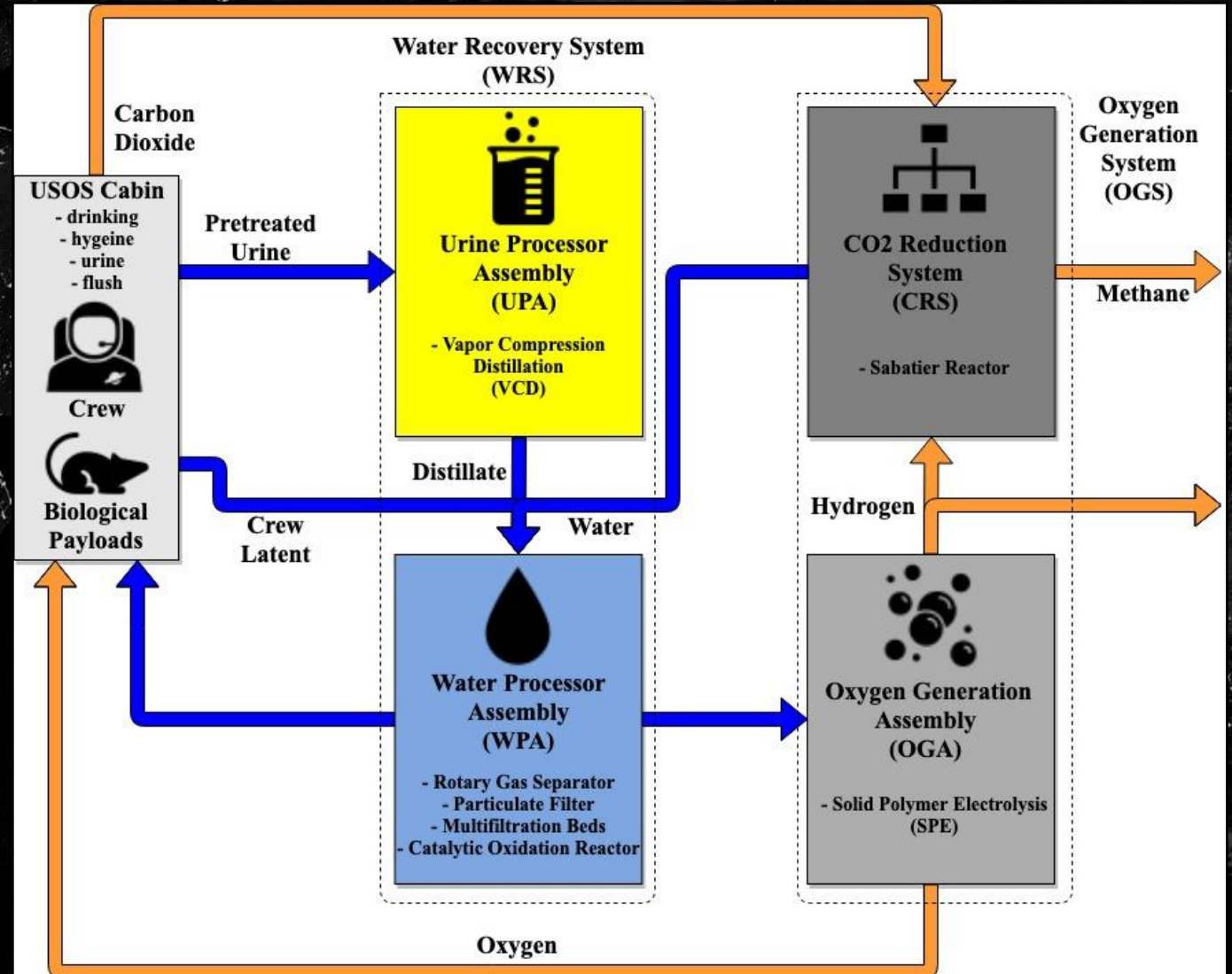


**Notes:** First two rows denote streams currently not addressed by technologies on ISS  
 Relative concentration: Very High (4) High (3), Medium (2), Low (1), Trace (0)  
 COD = chemical oxygen demand



# Current ECLSS

- Water Recovery System (WRS)
  - Urine Processor Assembly (UPA)
  - Water Processor Assembly (WPA)
  - Brine Processor Assembly (BPA)



# New ECLSS Technologies on ISS

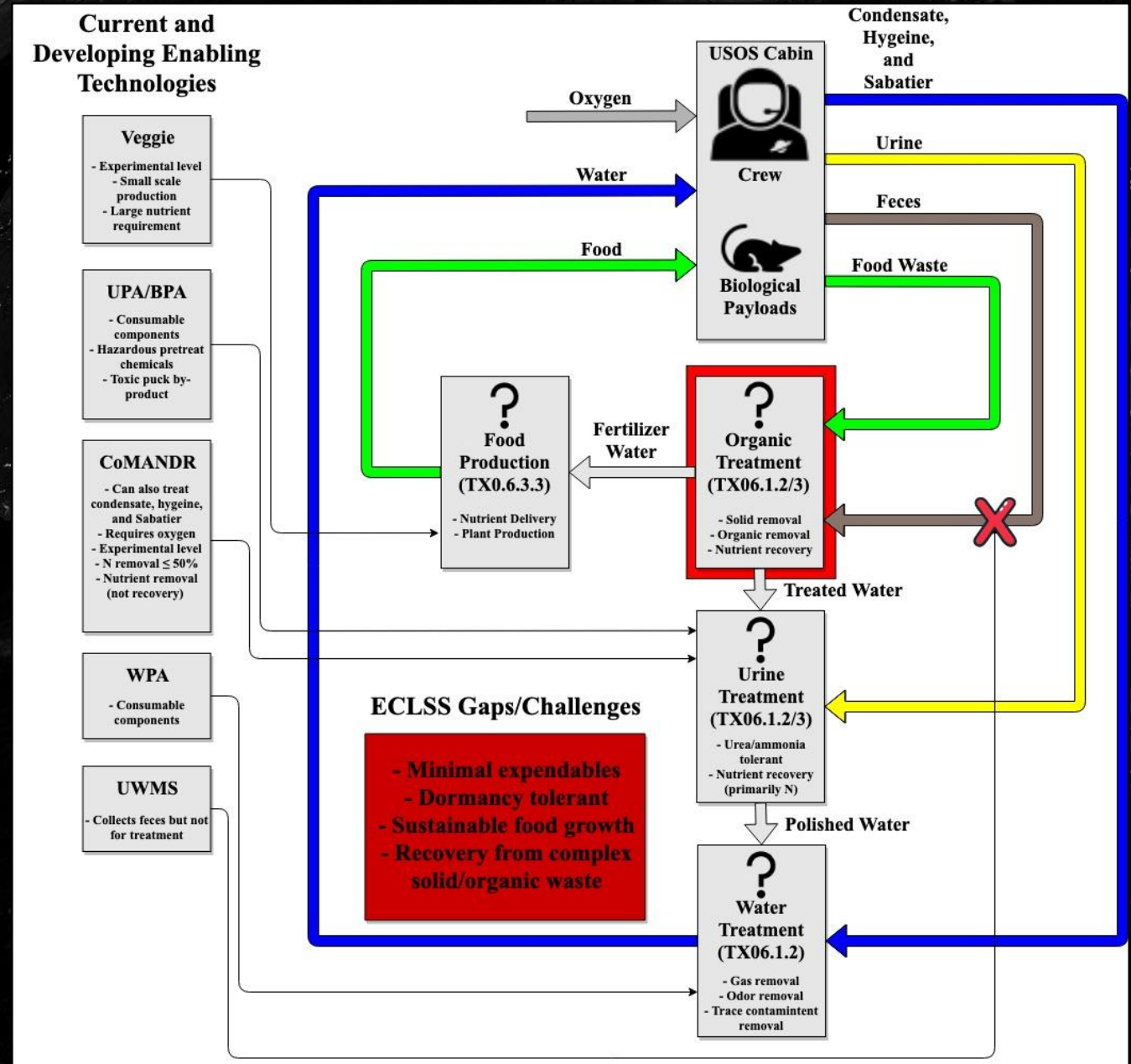
## Universal Waste Management System (UWMS)

- Fecal waste collected and sealed in hydrophobic bag
- Not designed for fecal waste to be treated for recovery
- Consumable activated carbon filter to remove odor



# Developing Future ECLSS

- Current and developing technologies available to enable addressing these gaps
- EXCEPT concentrated waste (i.e., fecal/food)
- Benefits
  - Increased water recovery
  - Enables sustainable food production
  - Decreases requirement for waste storage



# WATER: SUSTAINING LIFE ON THE MOON

Sustainably exploring the Moon will require a safe habitat for the crew. To stay on the Moon, new technology is needed to simulate Earth's environment that will reliably regenerate water, air, and food. Smart water recycling within the habitat will treat wastewater, provide necessary fertilizers and water for food crops, and create safe drinking water for our explorers. Proving these technologies will then take us to Mars and beyond.



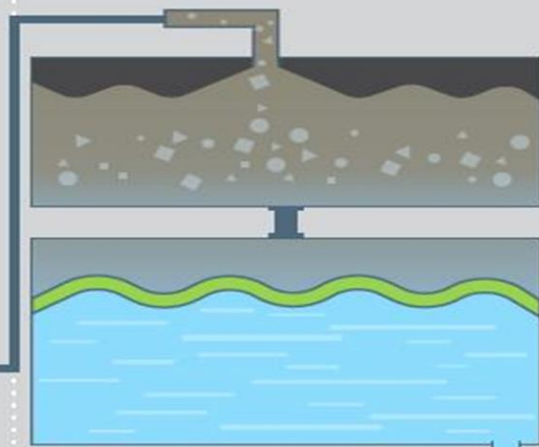
1

## Waste Water Collection

C N P K H<sub>2</sub>O

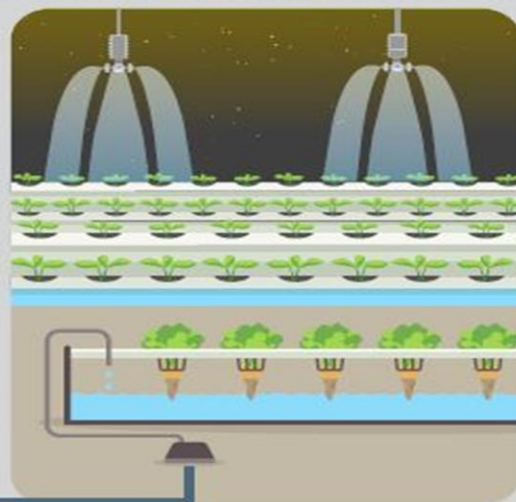
2

## Waste Water Treatment

N P K H<sub>2</sub>O

3

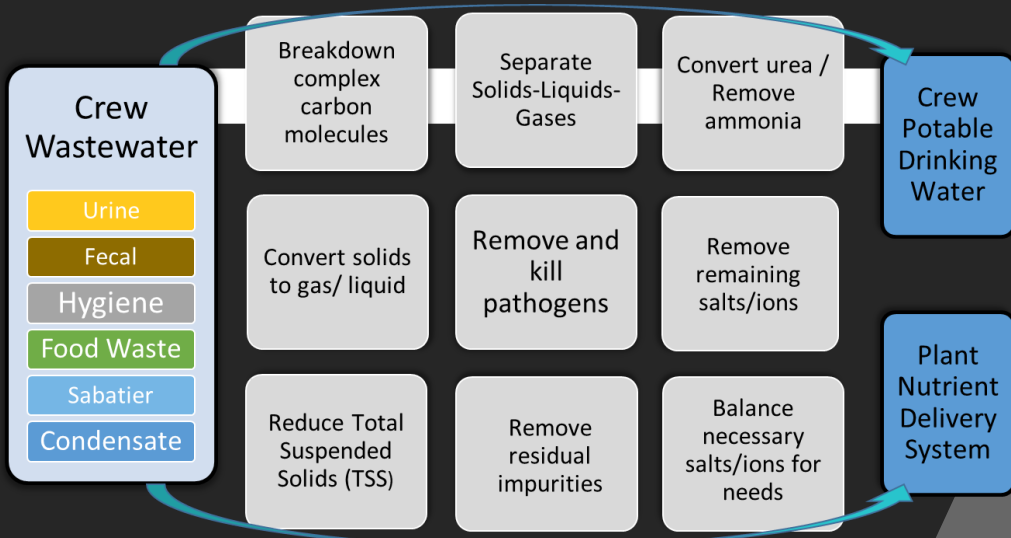
## Water & Fertilizer for Plants

N P K H<sub>2</sub>O

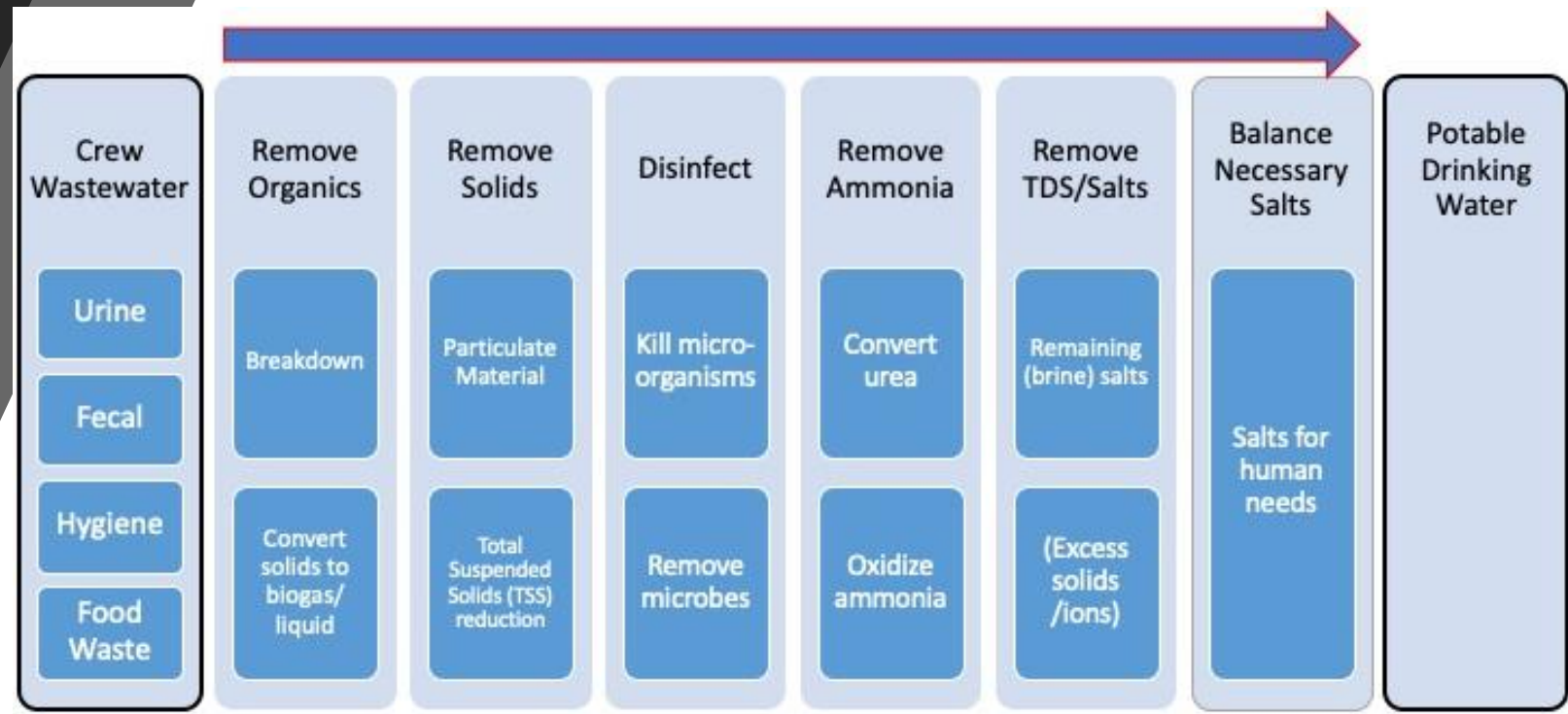
4

## Clean Water & Food for Humans

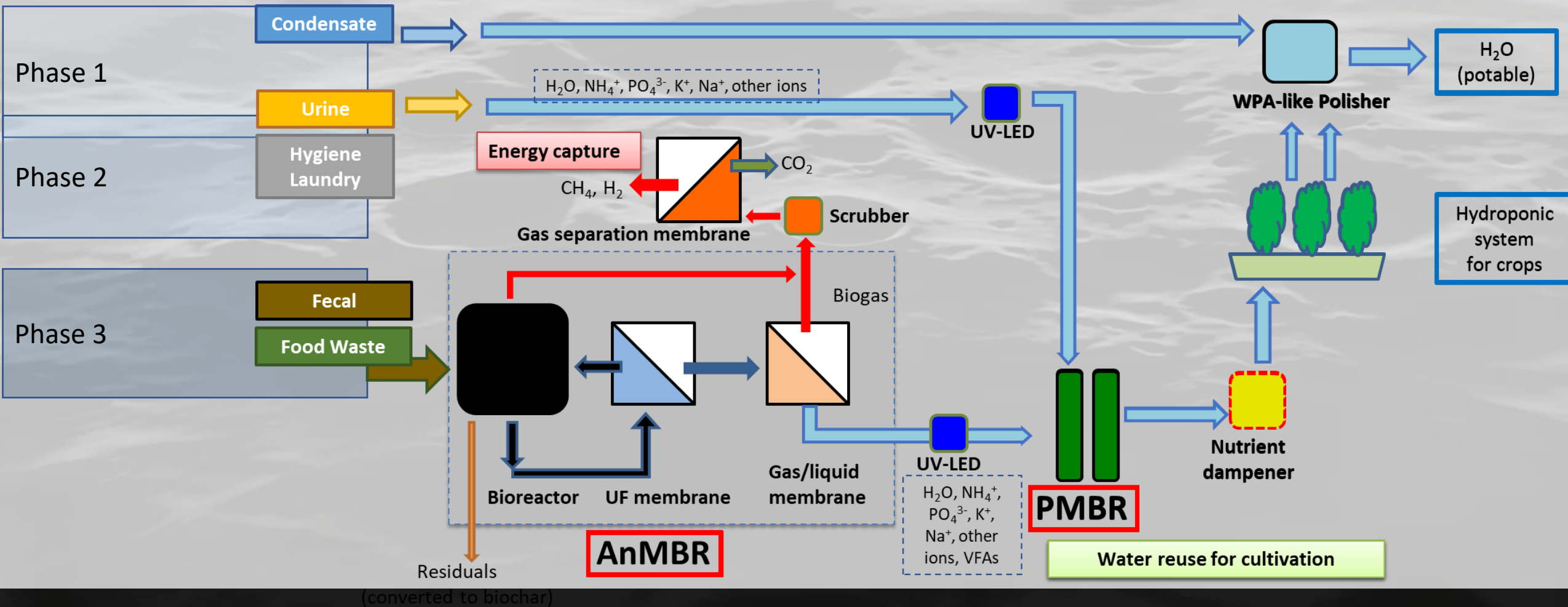
H<sub>2</sub>O



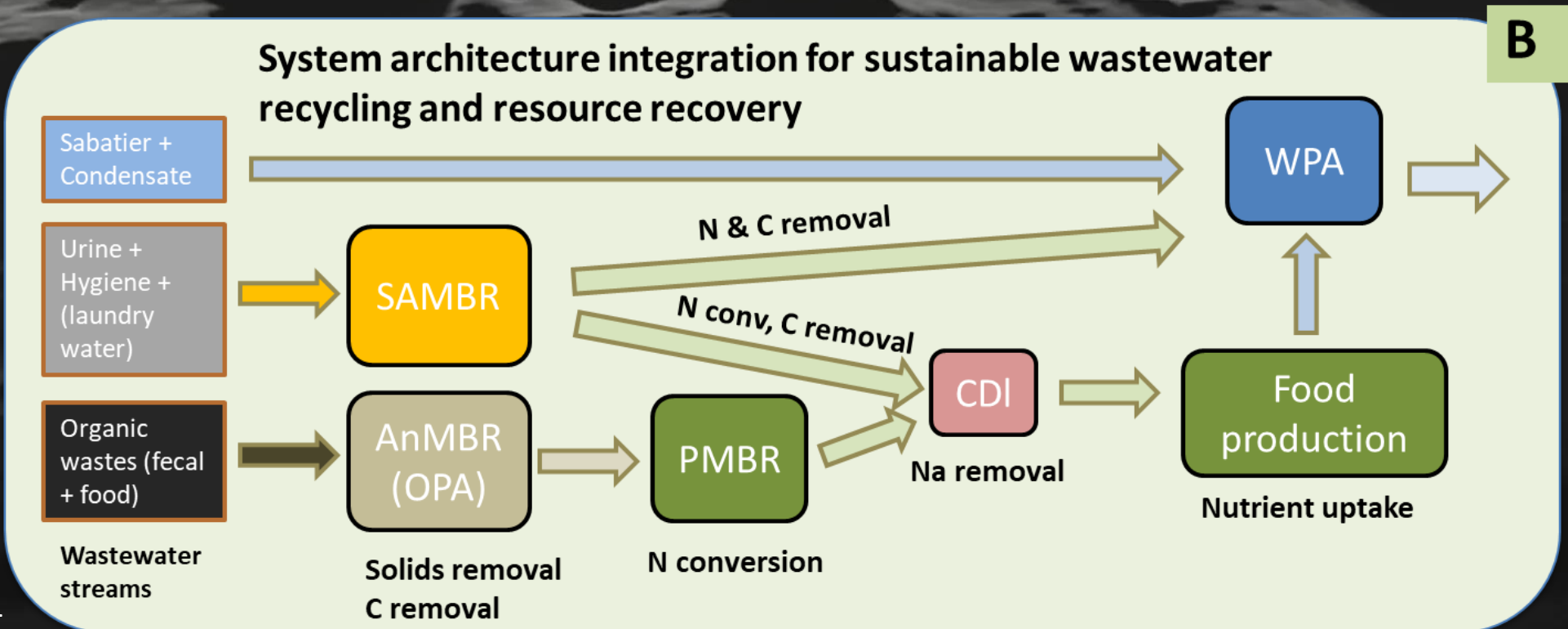
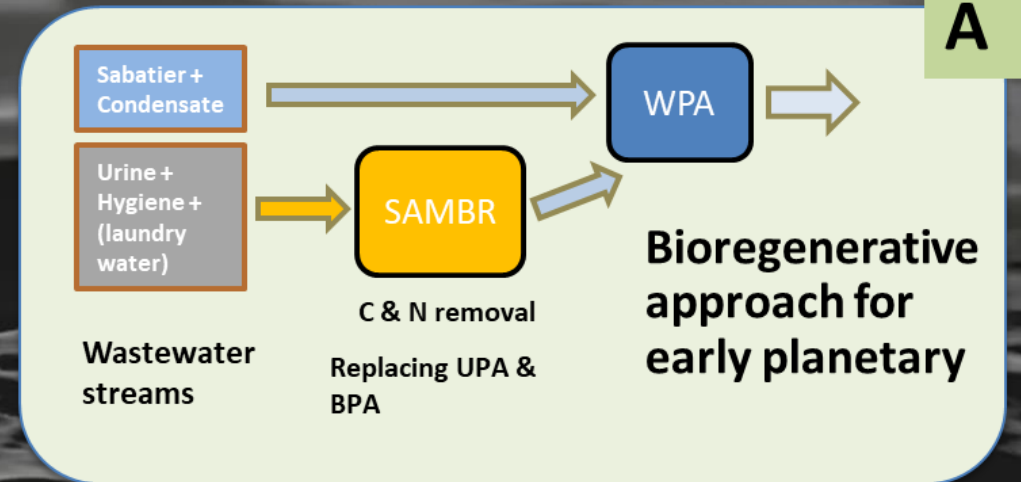
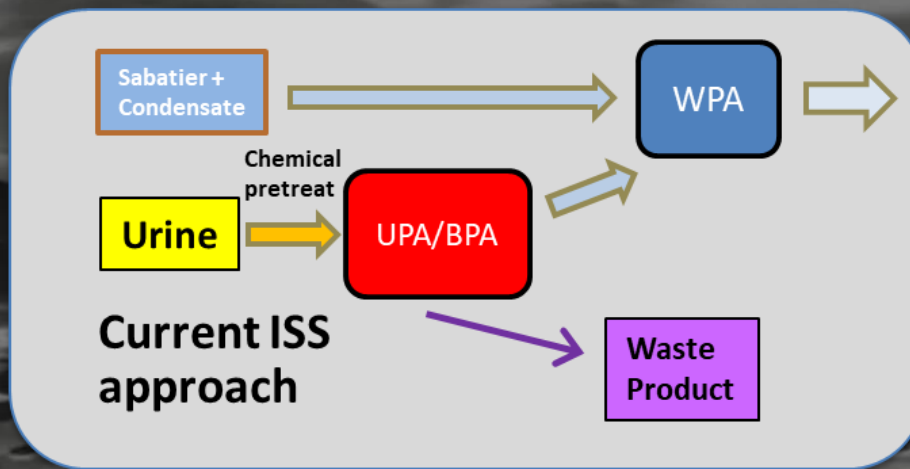
# Functional Flow Block Diagram (FFBD) for EPB Treatment



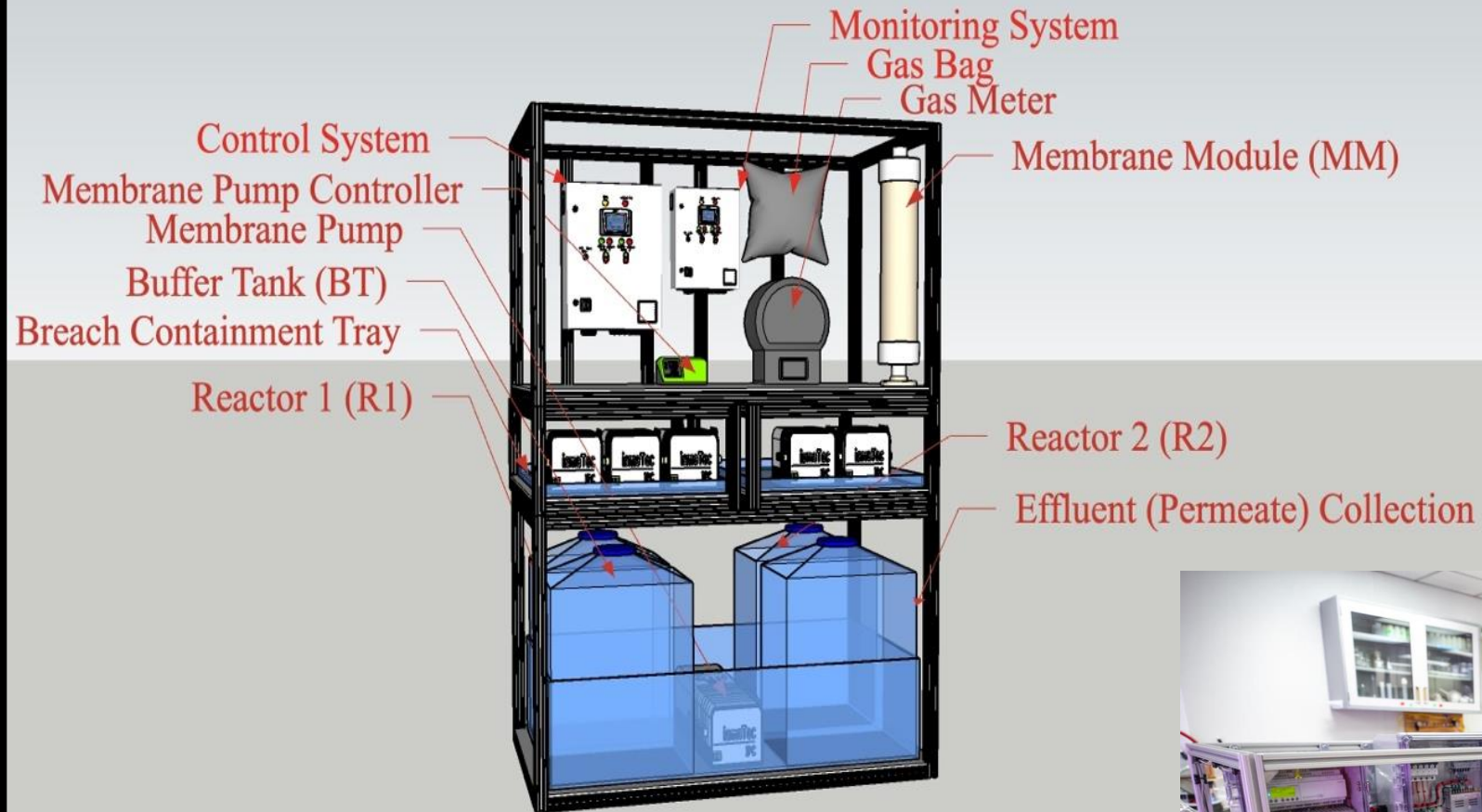
# System Architecture Design



# System Architecture Approach for EPB

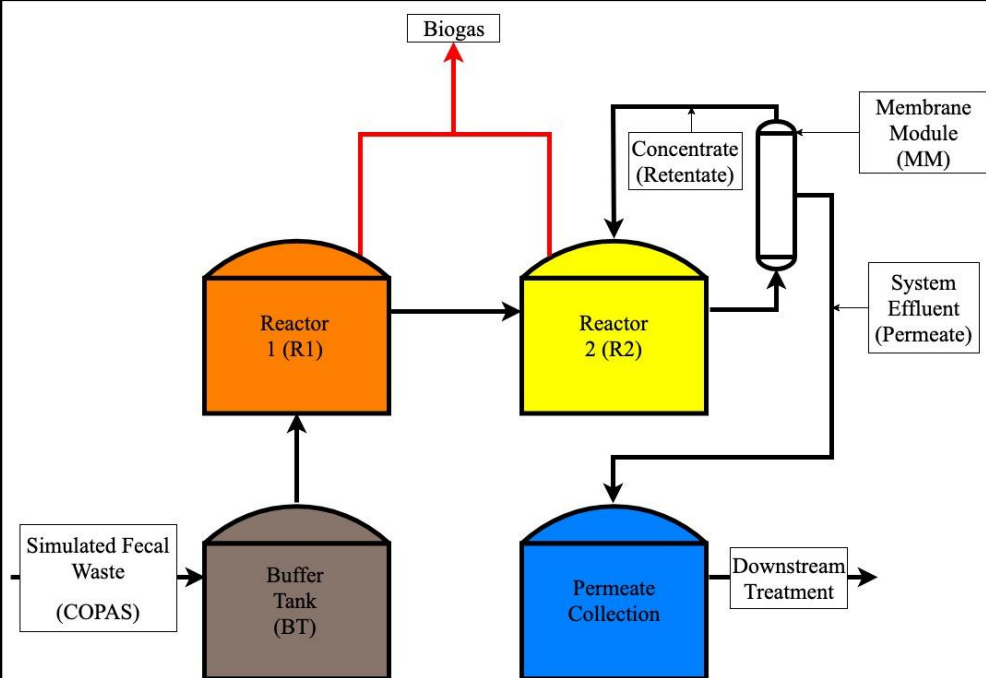


# AnMBR Design and Fabrication

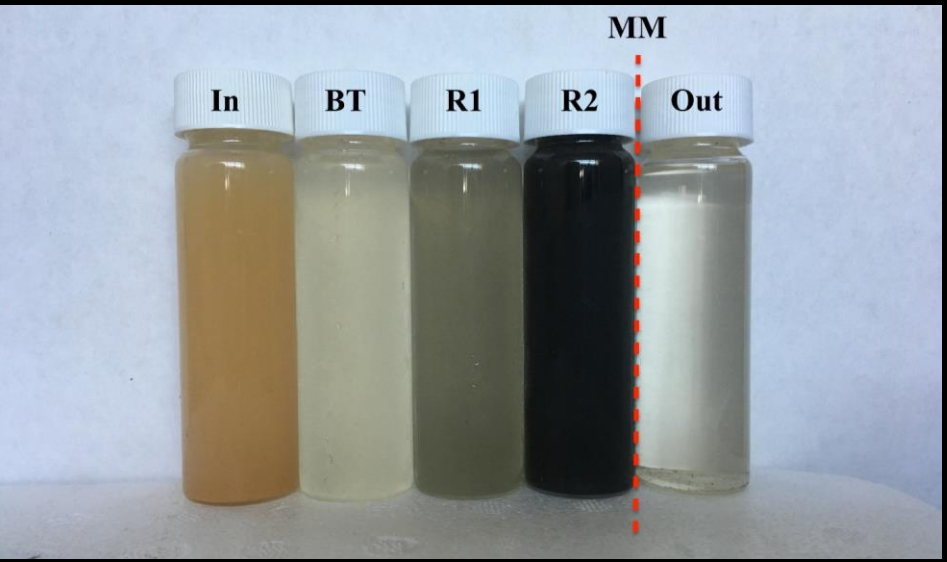




# AnMBR Experimental Plan

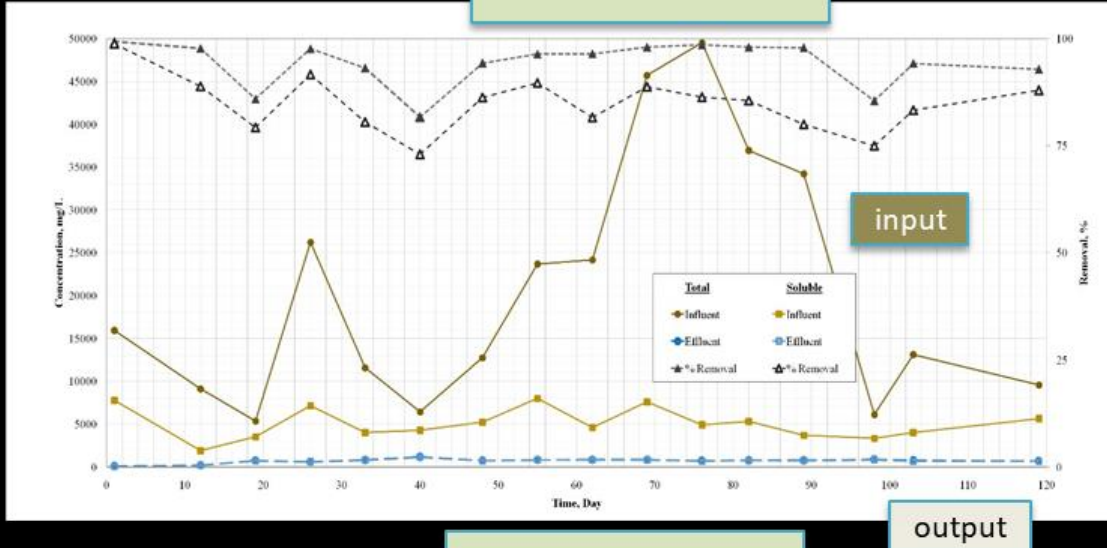


Waste Stream	Stage A	Stage B	Stage C	EPB Fecal and Flush
Solids (g/L)	1% (10)	3% (30)	5% (50)	5% (50)
Carbon (mg/L)	4761	14282	23803	21266
Nitrogen (mg-N/L)	566	1698	2830	2373
Phosphorous (mg/L)	140	420	700	791
COD (mg/L)	11902	35705	59508	N/A
COD Relative to Typical Municipal Wastewater	23x	71x	119x	
Organic Loading (OLR) (g-COD/L-d)	0.53	1.28	2.64	



# AnMBR 1 Status Update from USF

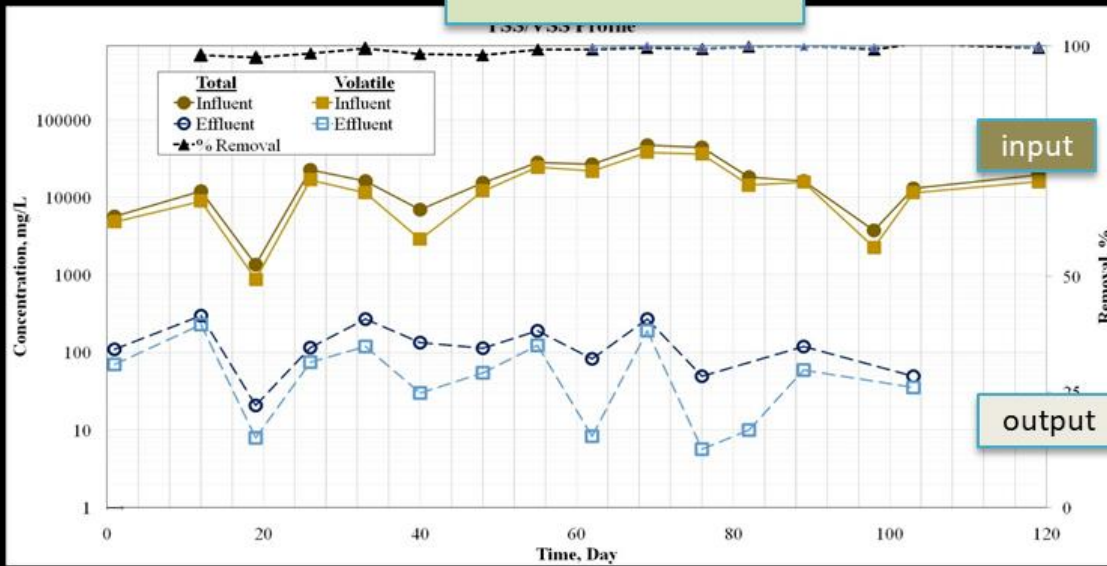
## COD



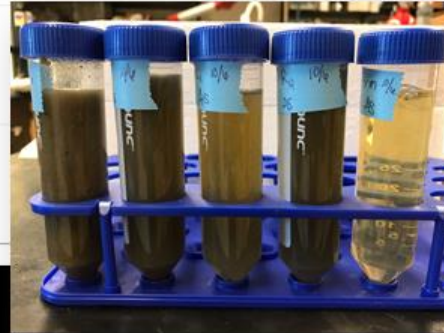
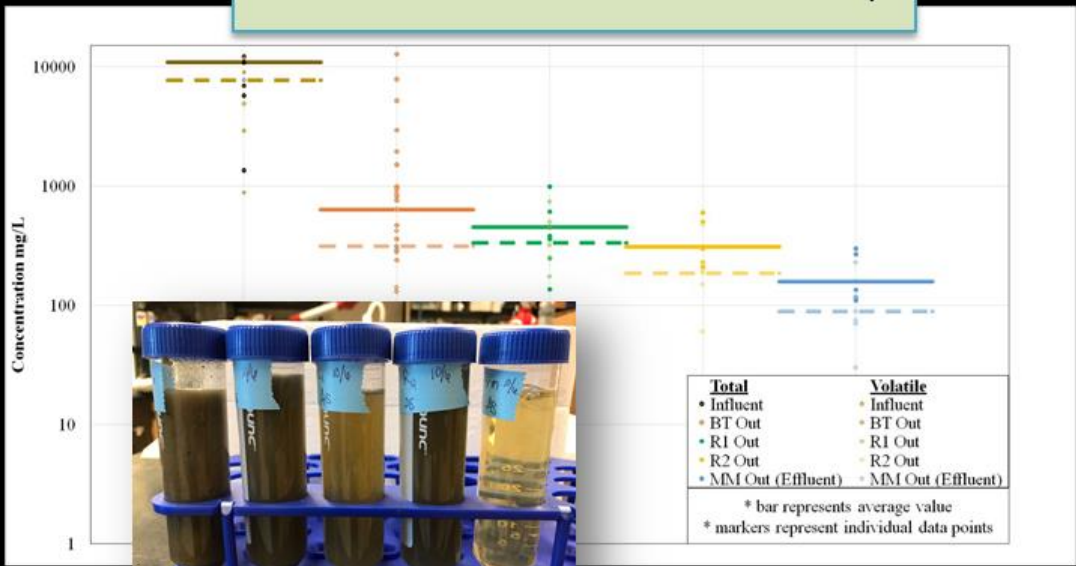
## COD – reduction at each treatment step



## Solids

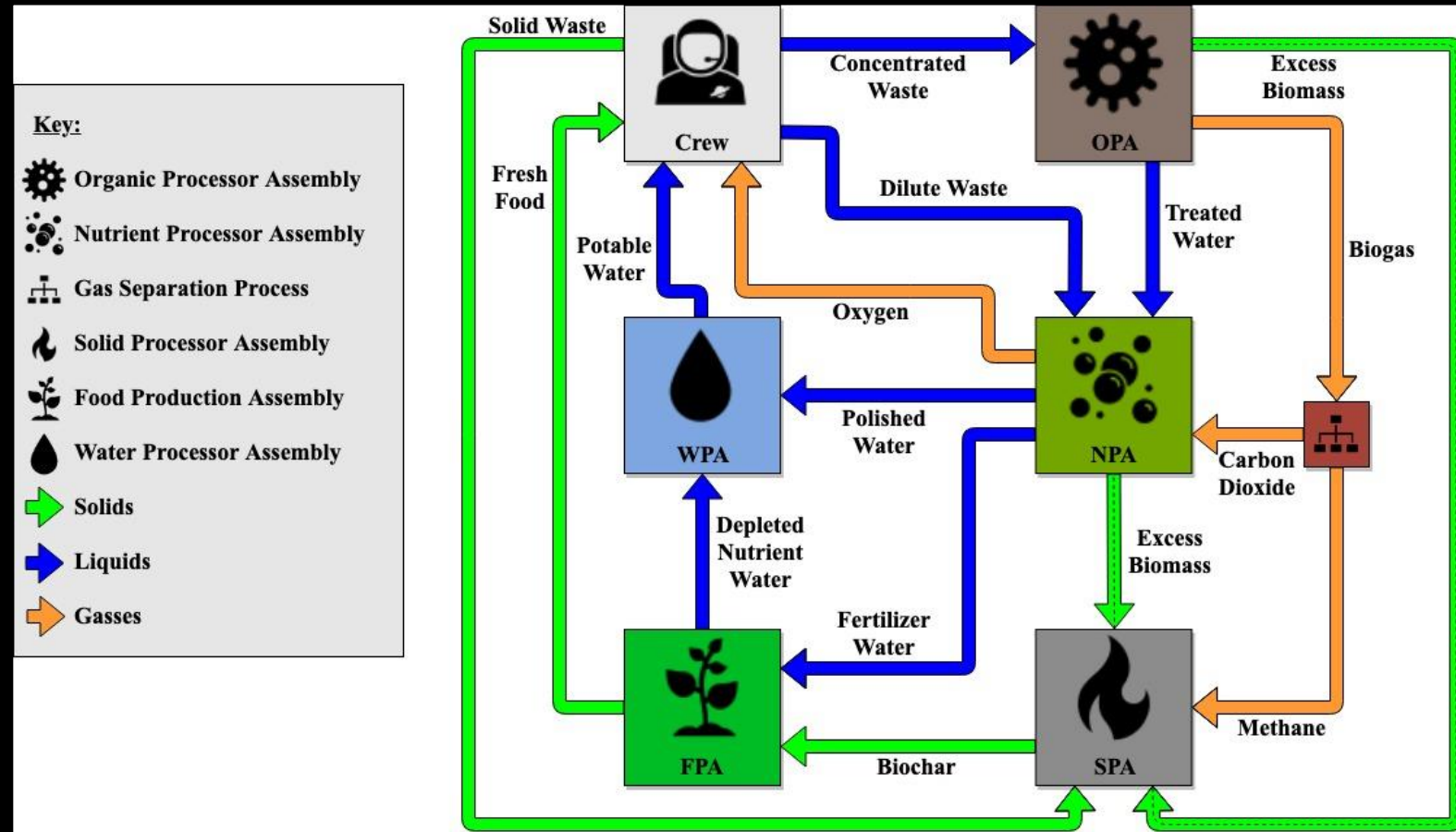


## Solids – reduction at each treatment step

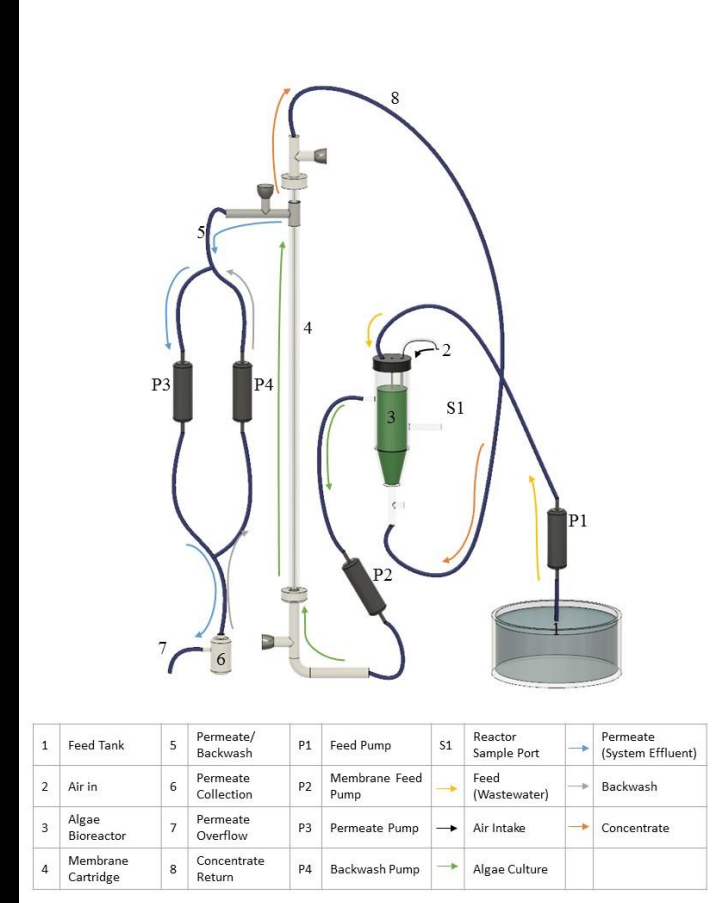
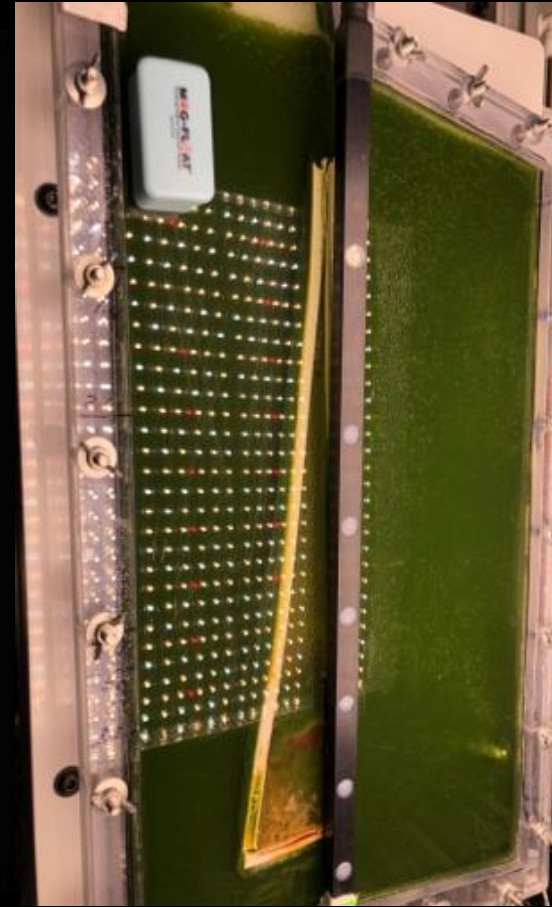
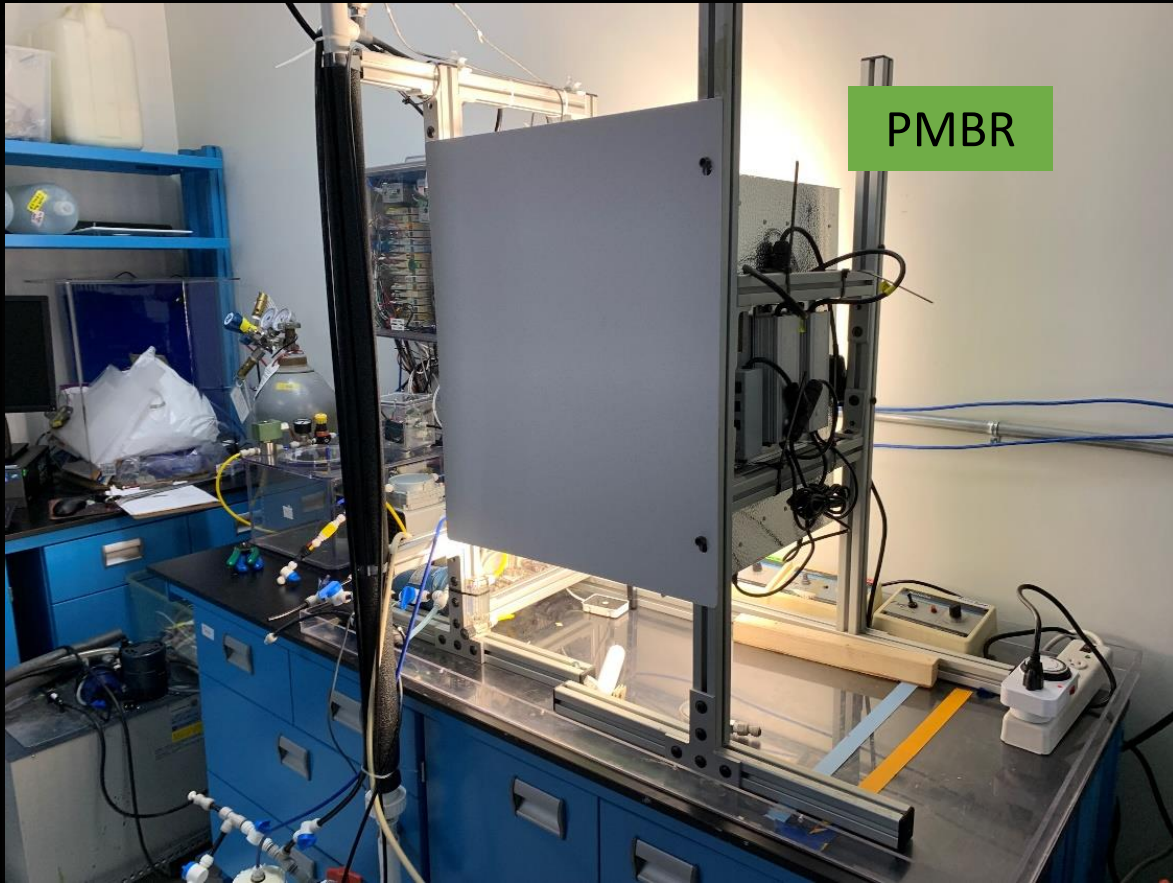


# AnMBR Improvements and Future Work

- Operating at full strength waste (5% solids)
- Expand to treat food waste
- Operate with real waste
- Enhanced pretreatment in BT
- Microbial static loop in permeate collection
- Integrate with complimentary systems for comprehensive sustainable architecture



# PMBR Design and Fabrication



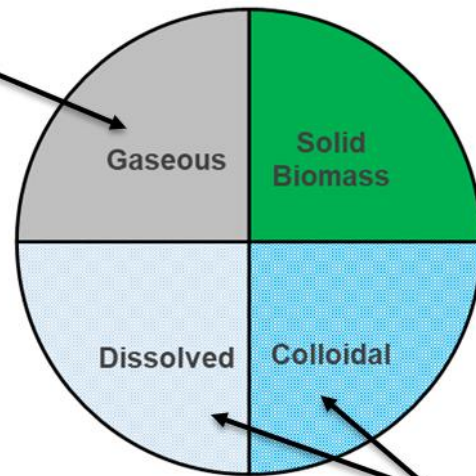
# PMBR Nitrogen Balance



Daniella, Riley, and Jason in front of OPA 2 and PMBR

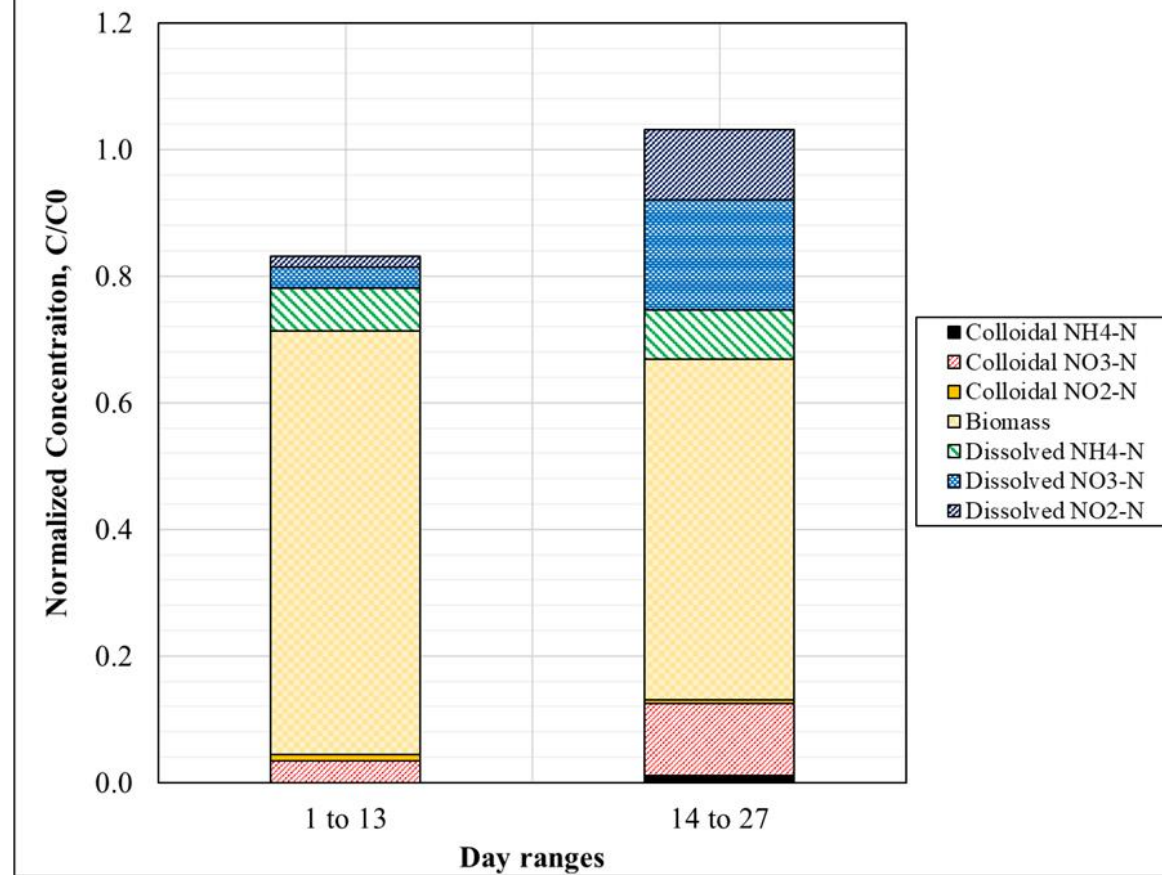
## Phases of Nitrogen in PMBR

At pH < 8, gaseous nitrogen is assumed negligible



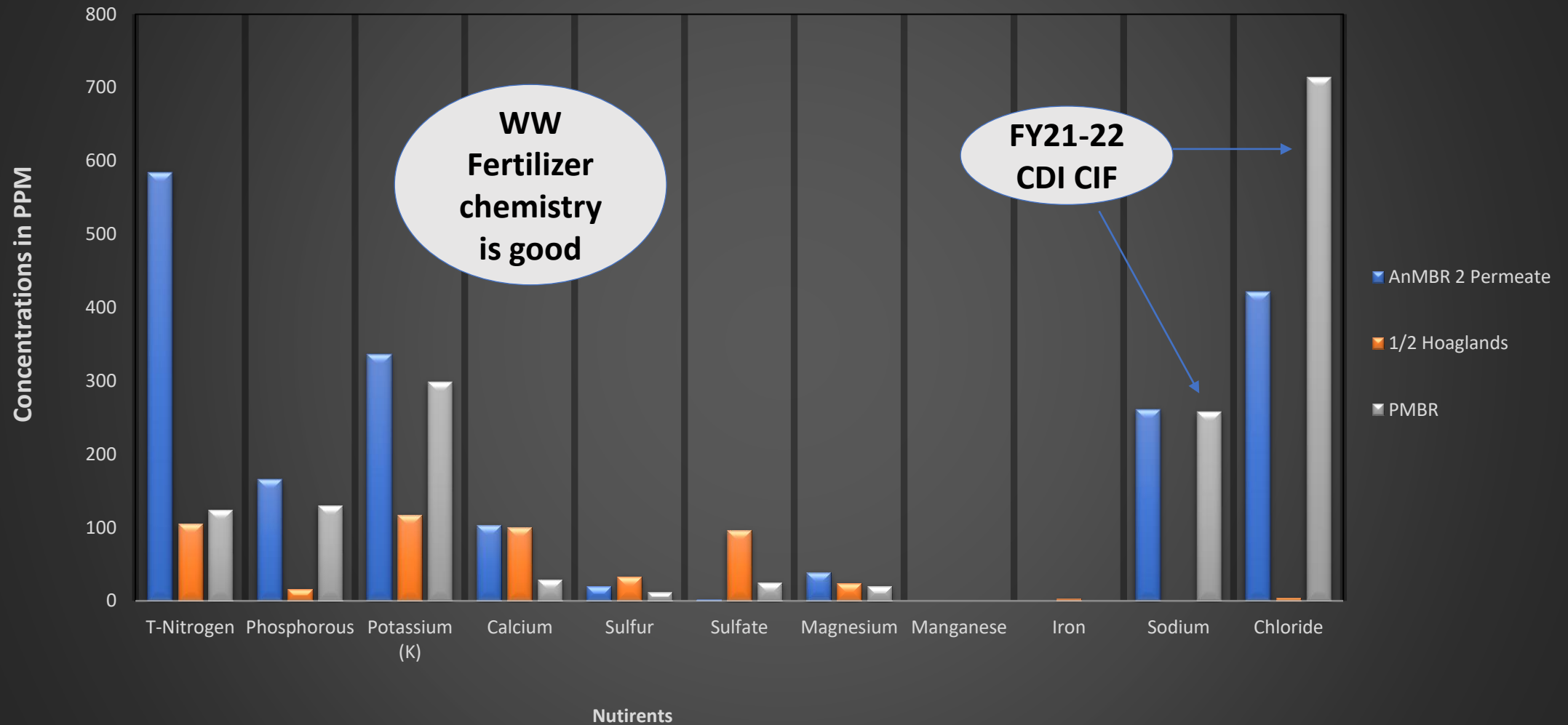
Liquid phase in PMBR

## Nitrogen Mass Balance



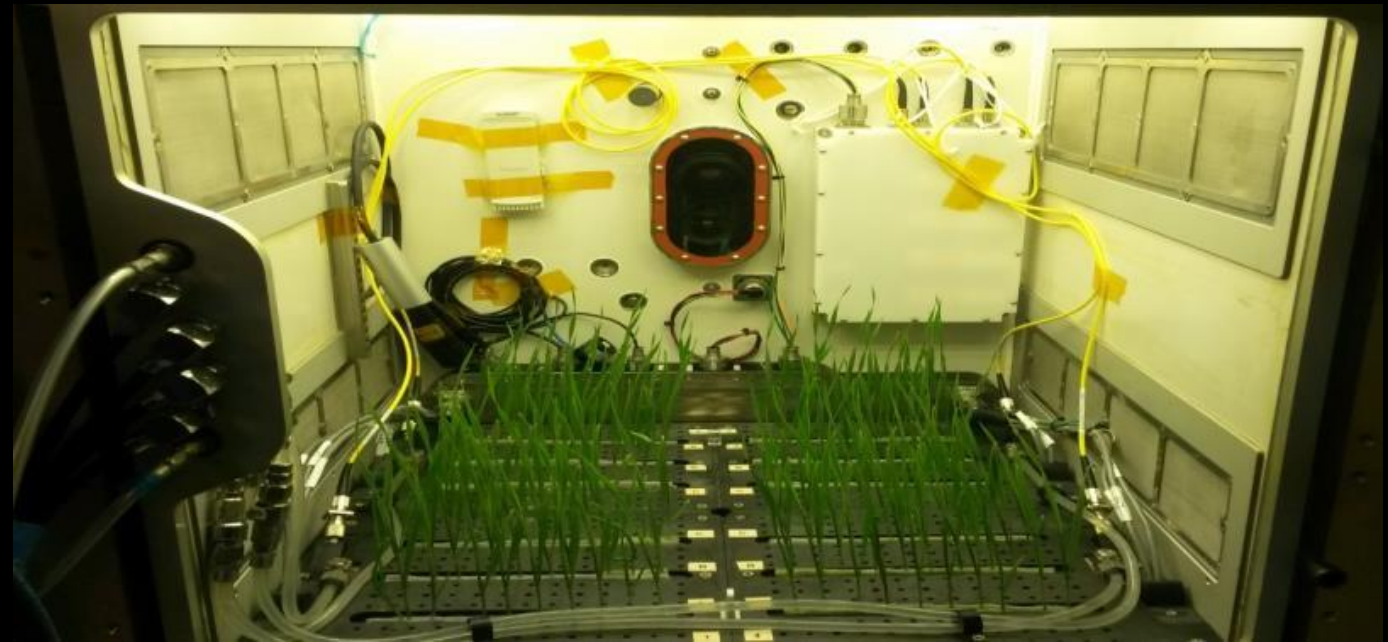
# Wastewater as Fertilizer

## Nutrient Comparison - AnMBR2 v PMBR v Hoaglands



# Food Production Challenges

Sustainable technologies: Veggie/APH at TRL5, OHALO TRL 2  
Cultivation resources (fertilizer, water, seeds) depend on resupply



# Developing Future ECLSS

- Current and developing technologies available to enable addressing these gaps
  - EXCEPT concentrated waste (i.e., fecal/food)
- Benefits
  - Increased water recovery
  - Enables sustainable food production
  - Decreases requirement for waste storage

## Current and Developing Enabling Technologies

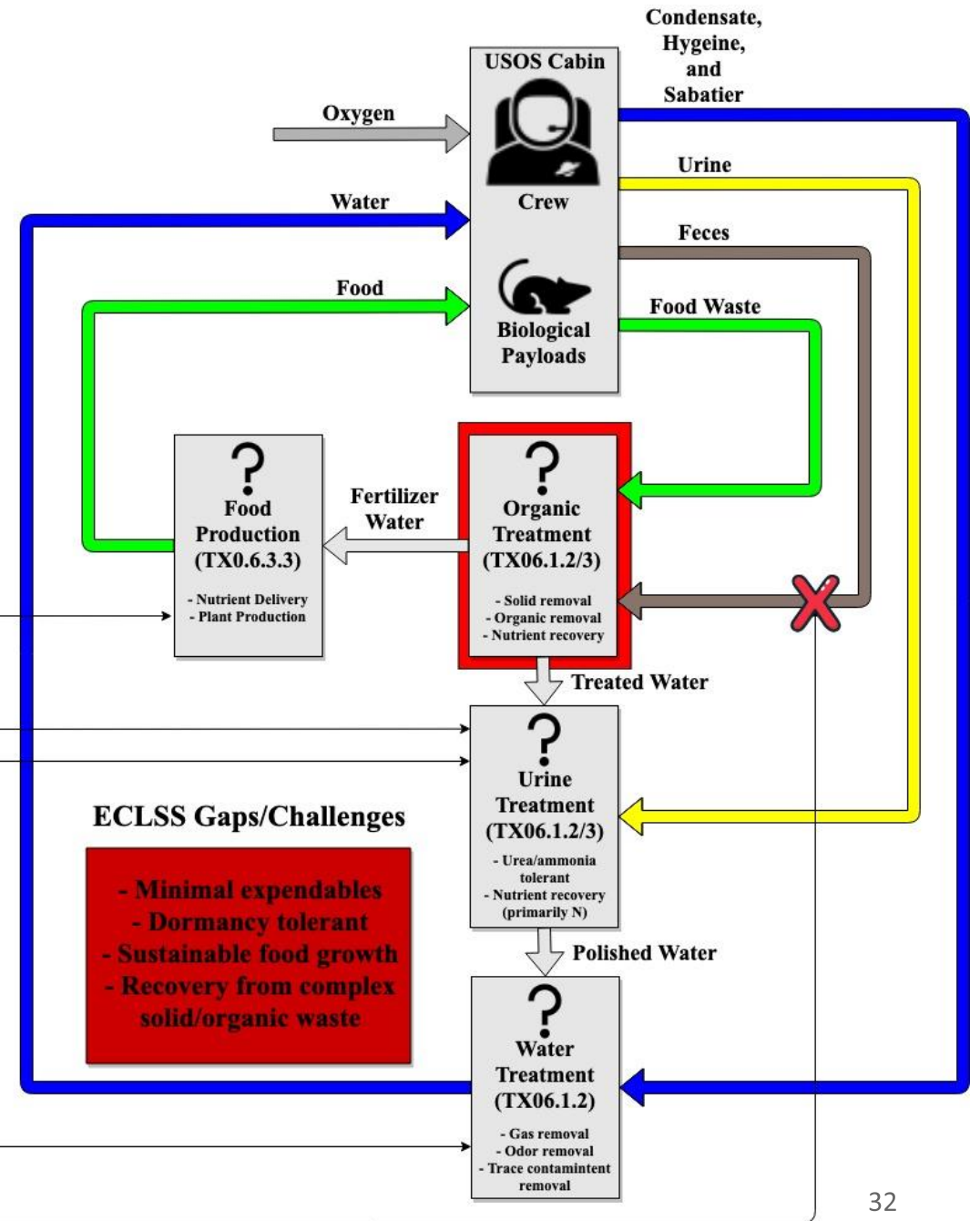
**Veggie**  
 - Experimental level  
 - Small scale production  
 - Large nutrient requirement

**UPA/BPA**  
 - Consumable components  
 - Hazardous pretreat chemicals  
 - Toxic puck by-product

**CoMANDR**  
 - Can also treat condensate, hygiene, and Sabatier  
 - Requires oxygen  
 - Experimental level  
 - N removal  $\leq 50\%$   
 - Nutrient removal (not recovery)

**WPA**  
 - Consumable components

**UWMS**  
 - Collects feces but not for treatment







Collaborations & Partnerships  
help us explore

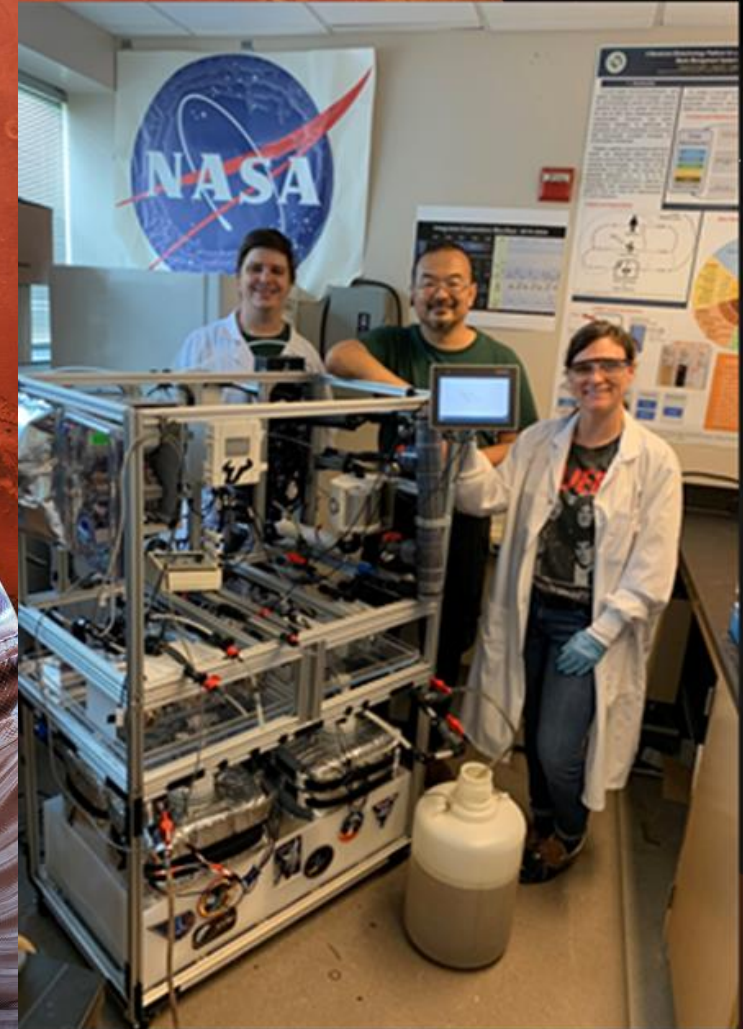
**OPPORTUNITIES  
FOR YOU!!!**

[www.intern.nasa.gov](http://www.intern.nasa.gov)

[www.usajobs.gov](http://www.usajobs.gov)

[www.nspires.nasa.gov](http://www.nspires.nasa.gov)

[www.KSCpartnerships.ksc.nasa.gov](http://www.KSCpartnerships.ksc.nasa.gov)



# Acknowledgements



**QUESTIONS?**