

# On the Moon to Stay

## Challenges Presented to Power Electronics Technology by Sustained Operations on the Lunar Surface

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# Artemis Base Camp Buildup

First lunar surface expedition through Gateway; external robotic system added to Gateway; Lunar Terrain Vehicle delivered to the surface

Sustainable operations with crew landing services; Gateway enhancements with refueling capability, additional communications, and viewing capabilities

Pressurized rover delivered for greater exploration range on the surface; Gateway enables longer missions

Surface habitat delivered, allowing up to four crew on the surface for longer periods of time leveraging extracted resources. Mars mission simulations continue with orbital and surface assets.

Lunar Terrain Vehicle (LTV)

Crew Landing Services

Pressurized Rover

Fission Surface Power

ISRU Pilot Plant

Surface Habitat

## **SUSTAINABLE LUNAR ORBIT STAGING CAPABILITY AND SURFACE EXPLORATION**

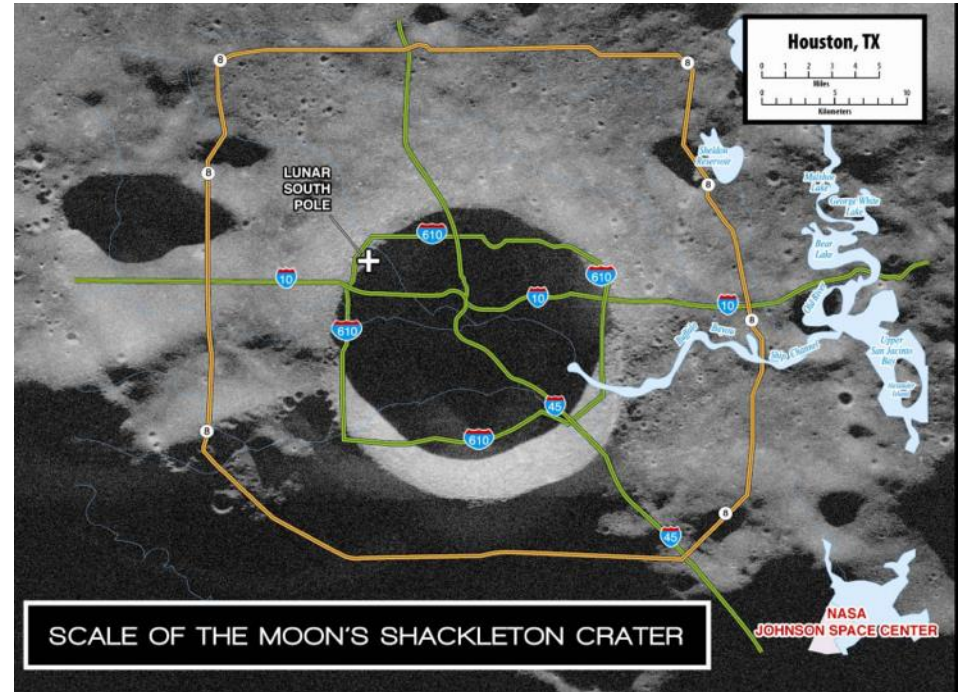
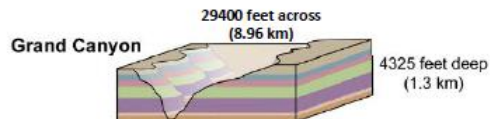
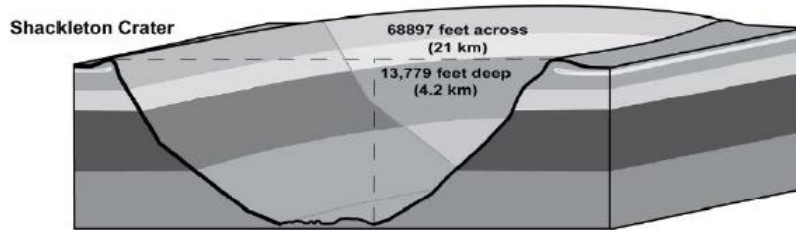
MULTIPLE SCIENCE AND CARGO PAYLOADS | U.S. GOVERNMENT, INDUSTRY, AND INTERNATIONAL PARTNERSHIP OPPORTUNITIES | TECHNOLOGY AND OPERATIONS DEMONSTRATIONS FOR MARS

# Shackleton Crater



- ~20 km in diameter
- ~4 km deep and ~3x deeper and wider than the Grand Canyon at Enfilade Point
- Located at Lunar South Pole
- Rim and Connecting Ridge are primary targets for future lunar landings

SHACKLETON CRATER vs. GRAND CANYON



Artemis Base Camp Zone

The key commodity needed to exploit the Lunar Surface

## Equatorial Illumination Limits

- Cyclical periods of 14 days illuminated, 14 days dark
- Consistent

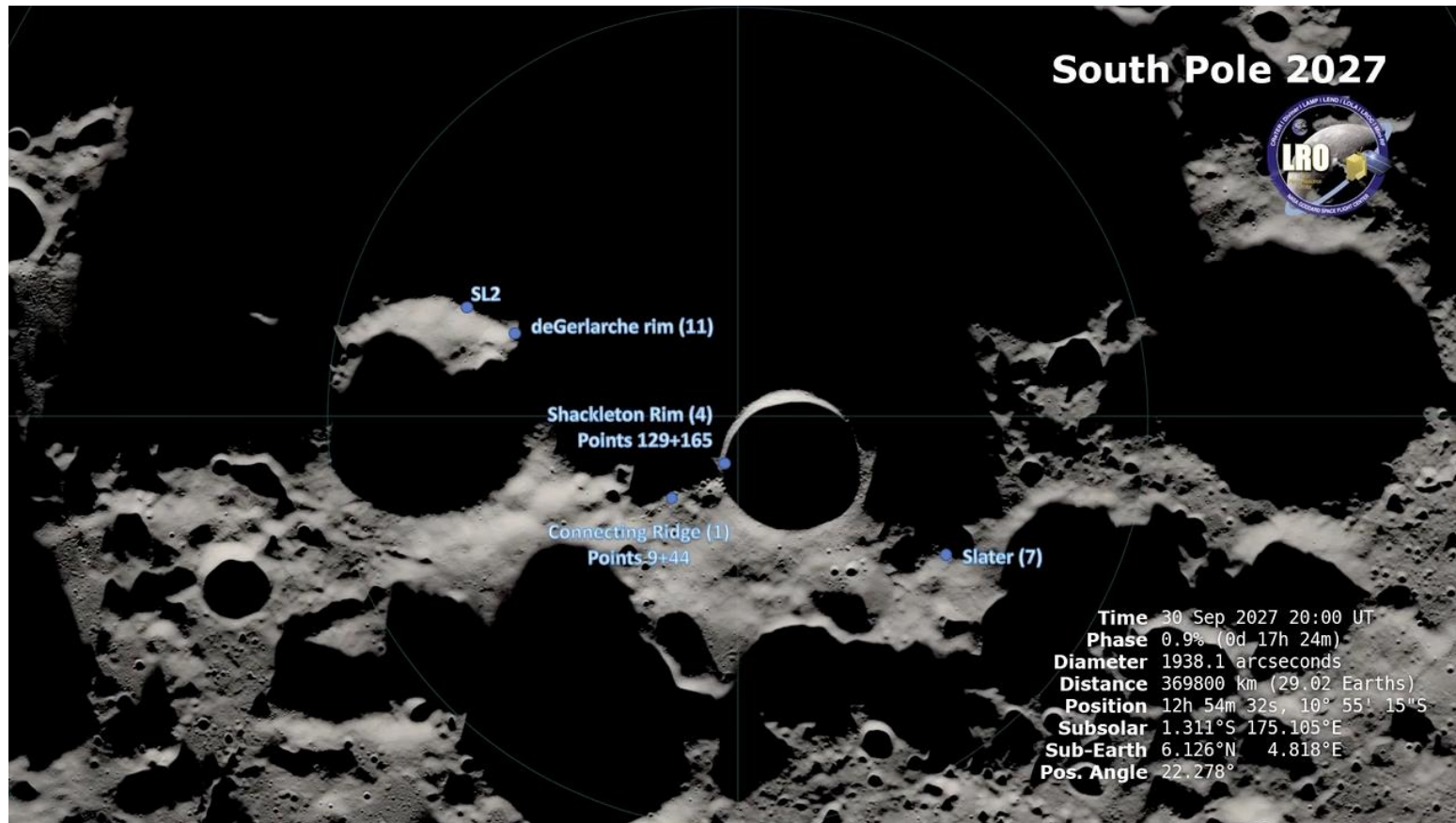
# Illumination

The scarce resource needed to produce power

## Polar Illumination Limits

- Intermittent with up to 100 hours darkness
- Highly dependent on location/elevation



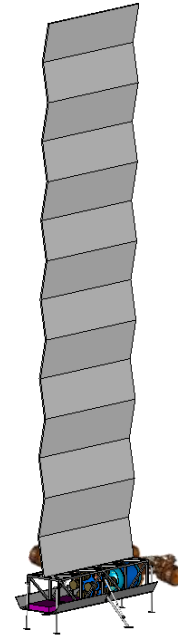


## Building Blocks: Generation



Photovoltaic Arrays

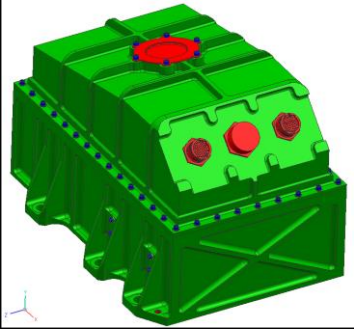
50 kW<sub>e</sub> increment, vertical deployment  
*~200 VDC output*



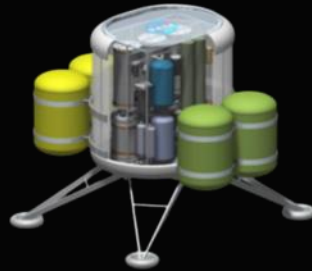
Fission Power

40 kW<sub>e</sub> increment  
*~240 VAC/50 Hz output*

## Building Blocks Energy Storage and Power Transmission



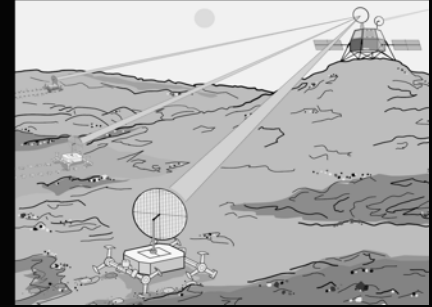
Secondary Batteries  
50 kW<sub>e</sub>h increment  
~30-100 VDC output



Regenerative Fuel Cells  
1 MW<sub>e</sub>h increment  
~30-100 VDC output

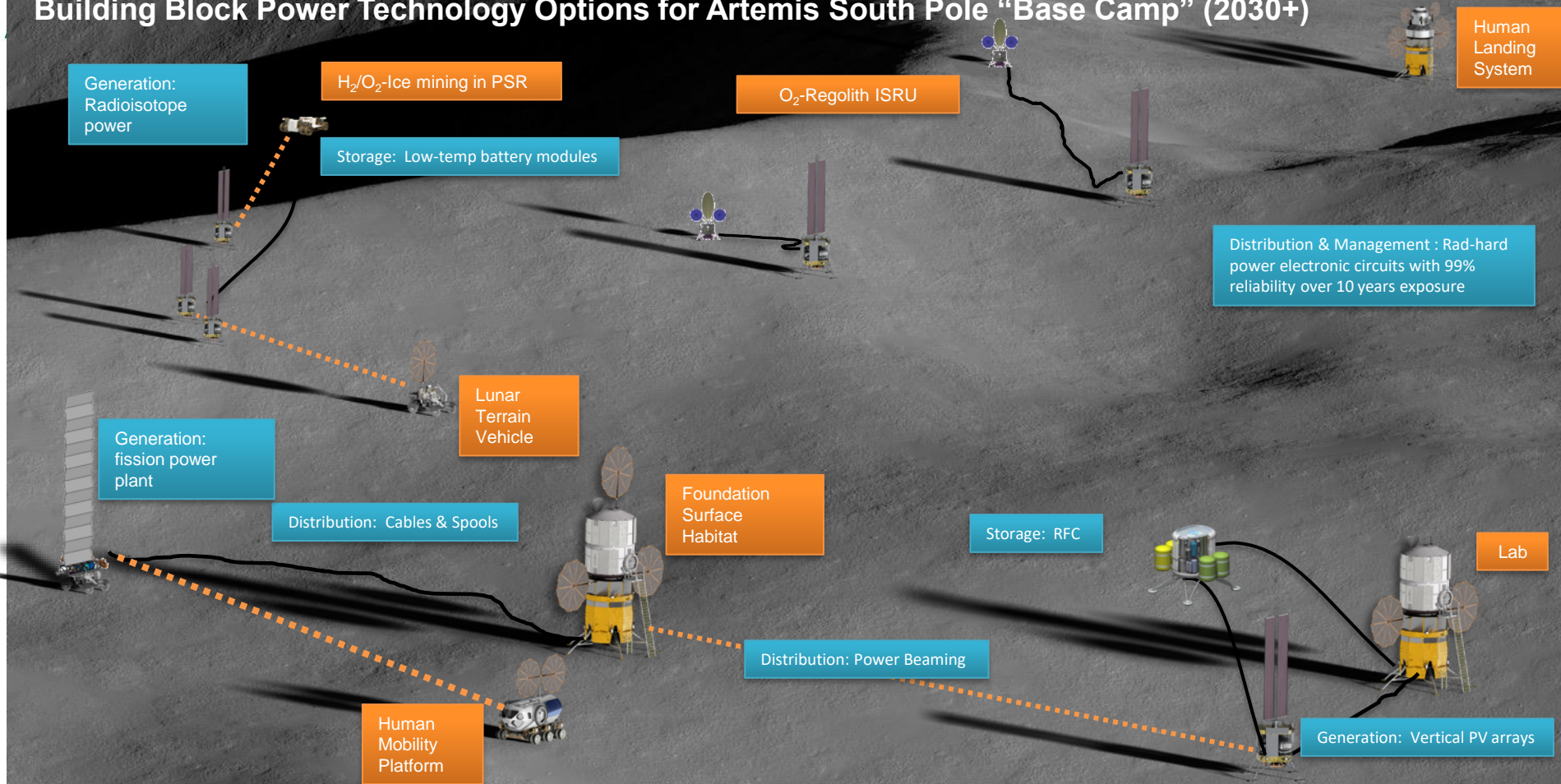


Cable/Reel  
10 kW<sub>e</sub> increment  
1000-1500 VDC



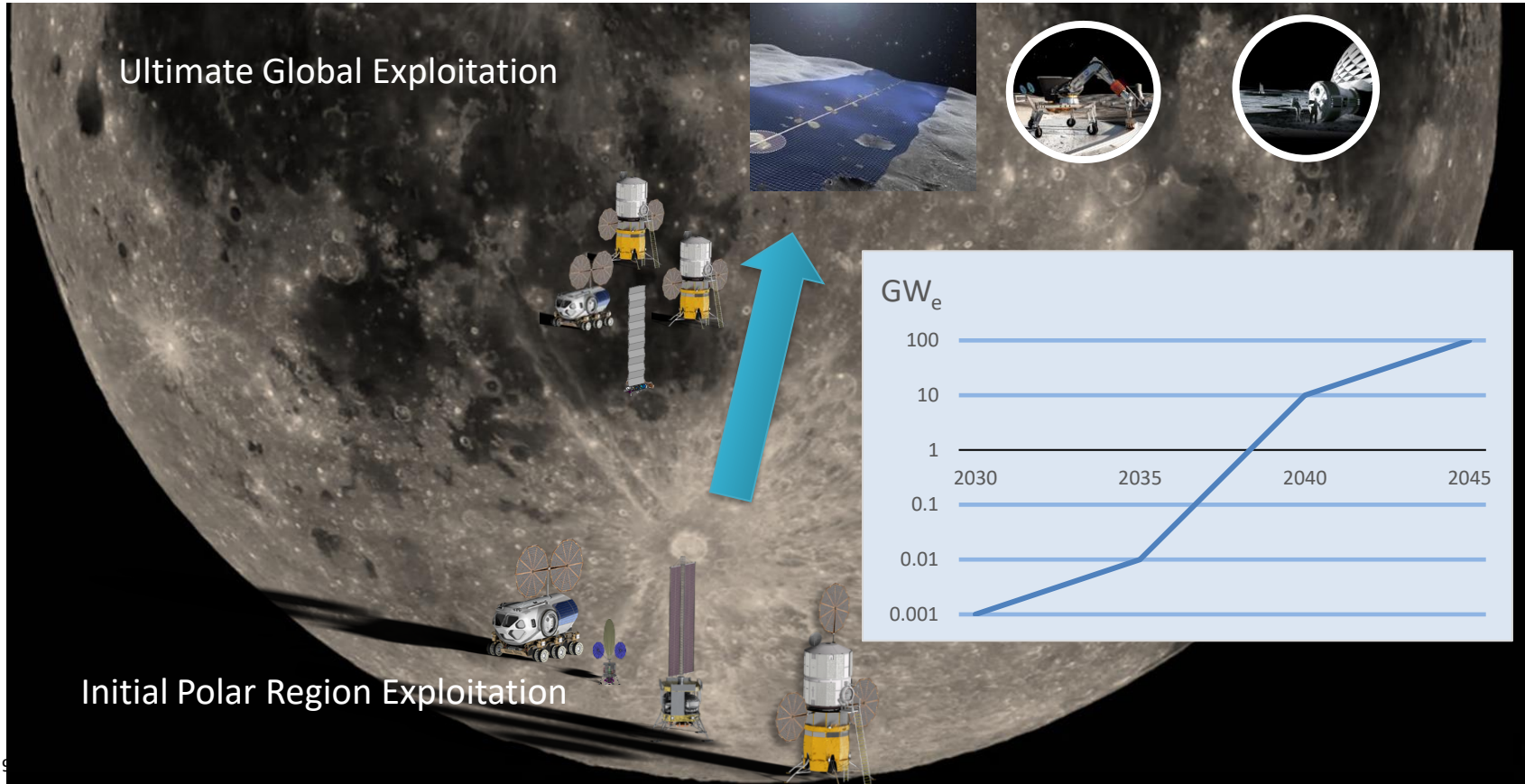
Power Beaming  
10 kW<sub>e</sub> increment

# Building Block Power Technology Options for Artemis South Pole “Base Camp” (2030+)





Power demand will incrementally increase as surface exploitation grows from initial Artemis installations at the South Pole to intensive industrial and settlement development across the Lunar surface



# Polar Building Block Power Technologies Can Bootstrap Generation, Storage and Distribution at Lower Latitudes (2040+)

- Landing pads and protective structures
- 100's to 1000's metric tons of regolith-based feedstock for construction projects
- 10's to 100's metric tons of metals, plastics and binders

Generation: Vertical PV arrays

Storage: Low temperature battery modules

Distribution: Cables & Spools

Generation: Radioisotope power

Storage: RFC

Distribution: Power Beaming

Generation: Fission power plant

Fission Power drives equipment to print photovoltaic generation, electrochemical, storage, and thermal storage from regolith

Human Landing System

Labs

Surface Habitats

Human Mobility Platforms

Lunar Terrain Vehicles

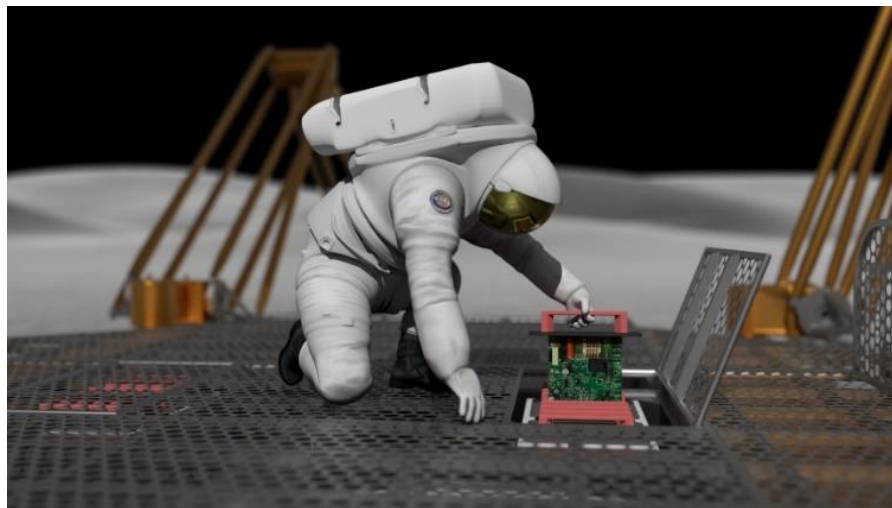


# APEC<sup>®</sup> Envisioned Future Capabilities



## Power Electronics:

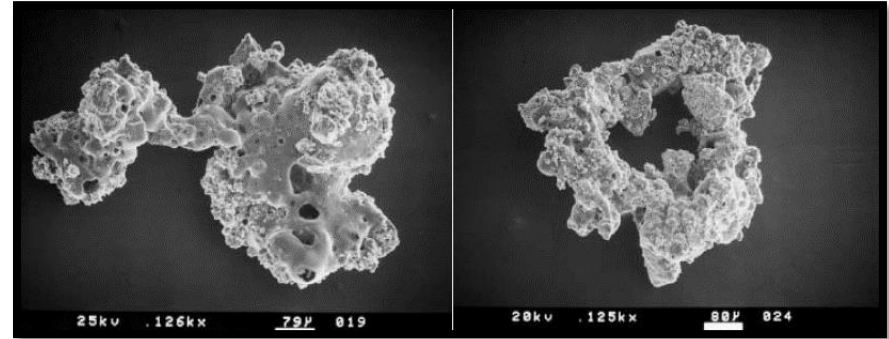
Bring to TRL 6 by 2030 a suite of power management, control, and regulation circuits & software operating at up to 1000 V and at maximum specific power and which are maintainable in the Lunar dust environment and 0.99 reliable for 10 years in the relevant Lunar radiation and thermal environments and in the Lunar hard vacuum and Mars atmosphere environments.



## Lunar Surface Temperatures

-173 C to 130 C

(-250 C in Permanently Shadowed Regions)

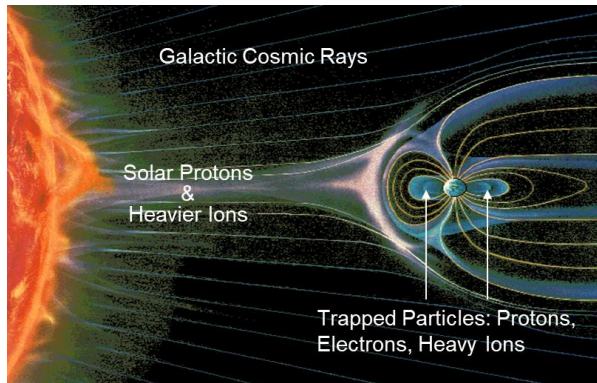


Lunar regolith (i.e. lunar dust) is angular, abrasive, irregular in shape, small in size, and adheres to surfaces

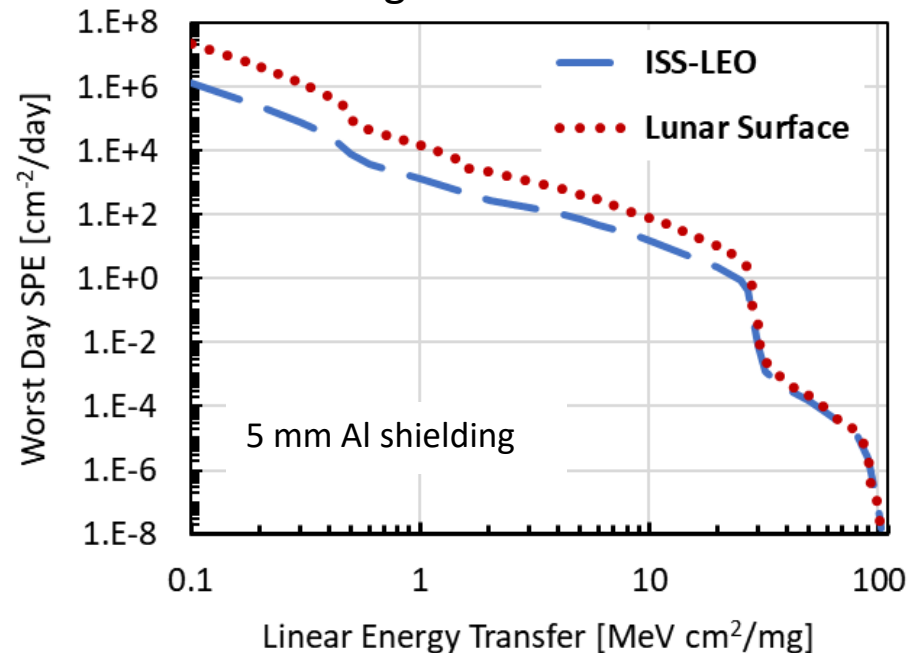


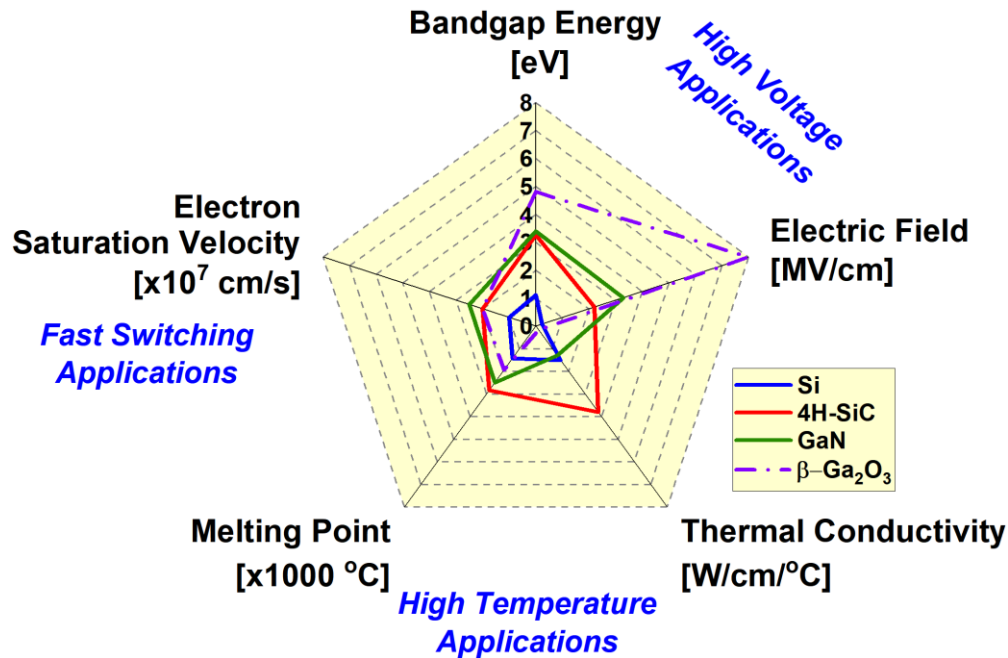
## Total Ionizing Dose

Orbit	~1-year TID (2.5 mm Al) krad(Si)
Jovian	1000
GEO	150
Lunar Orbit	4
Lunar Surface	2
ISS LEO	2
Earth Surface	< 0.001



## Single Event Effects





Graphic: courtesy J.-M. Lauenstein, NASA/GSFC

*“Achilles’ Heel” of Wide Band Gap Materials:  
Single Event Effects(SEE)*

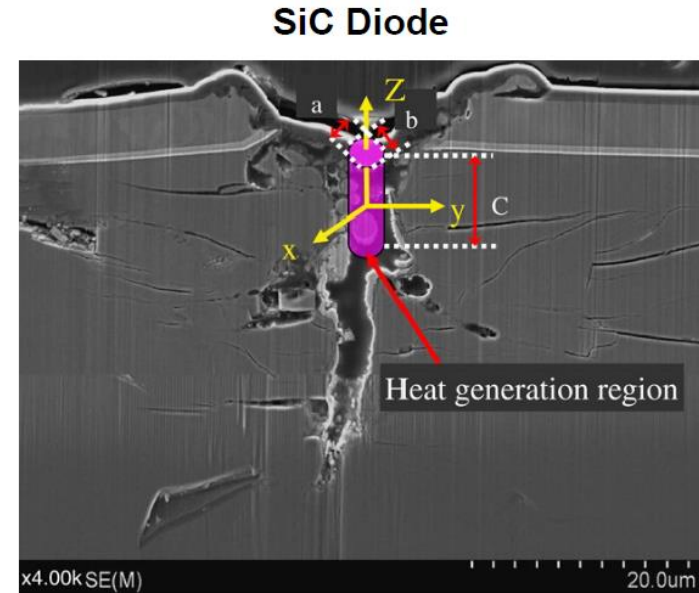
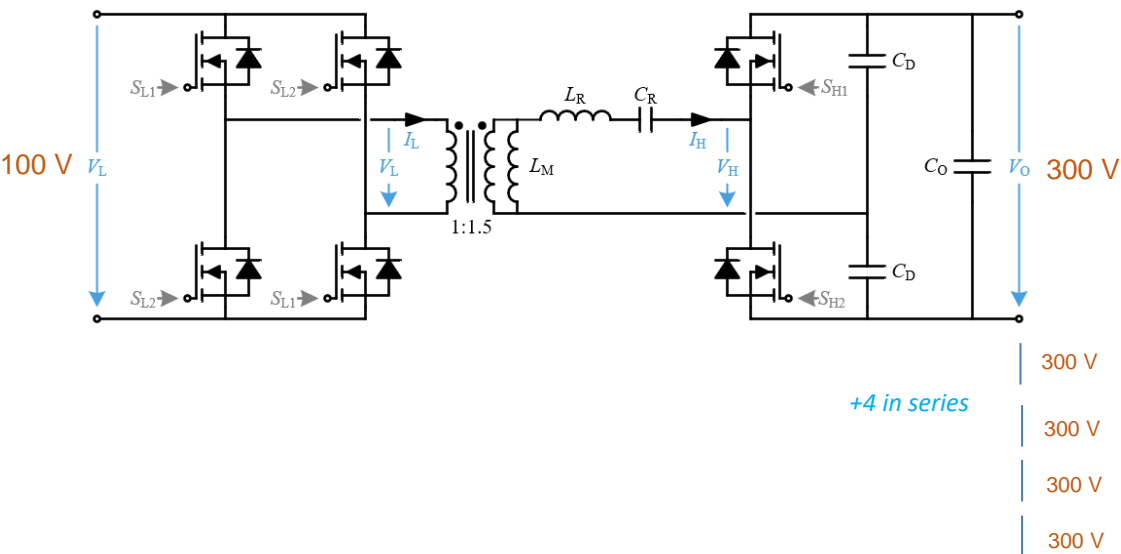


Image: Shoji, et al., © (2014) The Japan Society of Applied Physics, used with permission.

# APEC<sup>®</sup> Power Management Circuit Topology



## Typical Staged Boost Converter Module



## Breadboard PCB



*Barchowsky et al, NASA JPL, IEEE Aerospace Conf. 2022*

# Challenge

Bring to TRL 6 by 2030 a suite of power management, control, and regulation circuits & software operating at up to 1000 V and at maximum specific power and which are maintainable in the Lunar dust environment and 0.99 reliable for 10 years in the relevant Lunar radiation and thermal environments and in the Lunar hard vacuum and Mars atmosphere environments.

## NASA Proposal Opportunities:

Announcement Links:

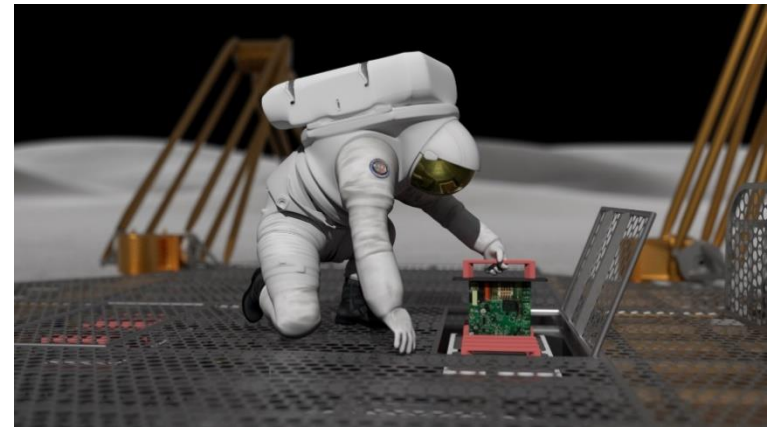
2022 STMD Tipping Point Announcement:

<https://tinyurl.com/2022TippingPoint>

2022 STMD Announcement of Collaboration Opportunity (ACO):

<https://tinyurl.com/2022ACO>

Next Key Date: Mini Proposals Due: 3/31/2022







[www.nasa.gov/spacetech](http://www.nasa.gov/spacetech)