

Rotating Balance Design and Production, GRC-419R

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Collaborators



Langley Research Center

- Devin Burns
- Peter Parker
- Sean Commo
- Donny Harper

Glenn Research Center:

- Mark Woike (retired)
- Jonathon Ponder
- Tony Shook
- Joe Dussling
- Jim Buckley
- Scott Metzger

Modern Machine & Tool

- Dennis Yates
- Yousuf Mohammad
- Kevin Bloxom
- Mark Roth
- Howard Walls
- And Others

Project Overview



Four (x4) new rotating balances are in various stages of design and production

 (x2) Two-component small form factor rotating balances, GRC-517R-A&B for advanced turbofan testing at NASA Glenn Research Center

 (x1) Two-component rotating balance GRC-419R for boundary layer ingestion testing at NASA Glenn Research Center

• (x1) Two-component reduced diameter balance for tail cone thruster testing at NASA Ames Research Center

· (x1) Six-component balance Unducted Single Fan, USF at NASA Glenn Research Center

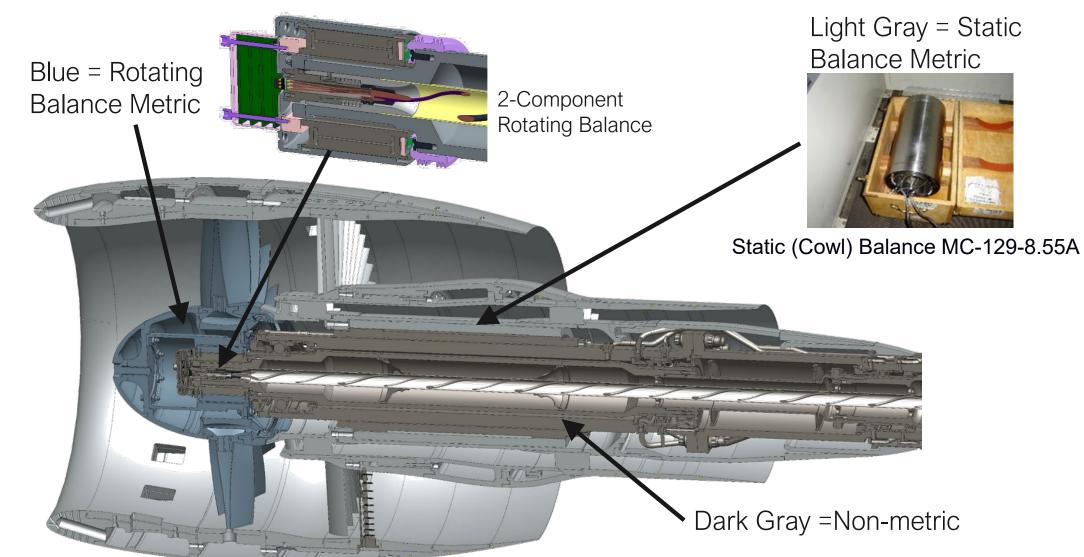
Motivation



- Force balances provide a measure of propulsion performance independent from flow field measurements.
 - System performance (net thrust drag) is only measured with force balances.
 - Fan efficiency measurements based on rotor torque are more accurate than pressure/temperaturebased measurements.
- New two-component rotating balances are being required for the testing of fan concepts being developed under several NASA Aeronautical Research Mission Directorate (ARMD) and Advanced Air Transport Technology (AATT) Projects
 - Higher bypass ratio fans with smaller engine cores (GRC-517Rs)
 - Boundary layer ingestion tolerant fans (GRC-419R)
 - Hybrid electric propulsion concepts (tail cone thruster, etc.)
 - Unducted single fan (GRC-222R)
- Need to interface to new state-of-the-art FPGA telemetry systems that have been developed for fan testing at NASA GRC

Current Need





Rotating Balance Design Overview



 Spoke wheel design, with both forward and aft beam groups

- Two measurement components:
 - Torque and Thrust

Thick "bulkhead" behind beams

Keyways to transmit torque loads



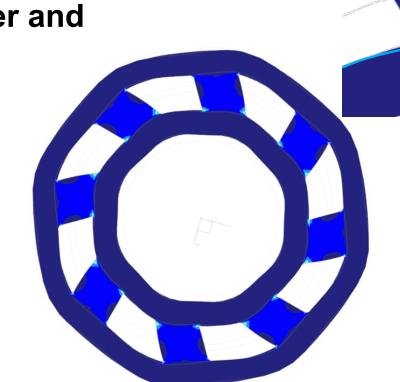
Finite Element Analysis Observations



 Double bending behavior was initially expected due to torque load

 FEA revealed warping behavior of the outer and inner diameters

 Warping of outer diameter reduces double bending behavior of beams



Analytical Model – T-Strap

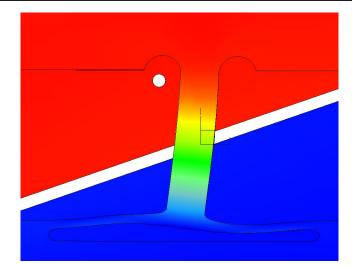


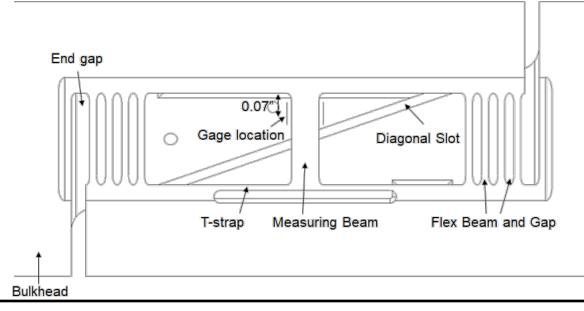
 Behavior of the outer diameter is similar to another known balance feature, the T-Strap

 A T-strap is a design used to reduce normal force interactions on the axial measuring beam



 See NASA/TM-2019-220248 for more details





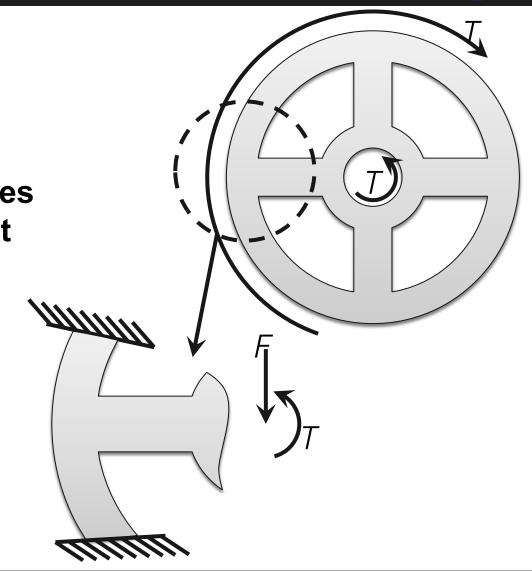
T-Strap Applied to Rotating Balances



 A single spoke of the rotating balance can be viewed as a T-Strap

 Modification to the T-Strap calculations involves reevaluating the derivation to include torque at balance center

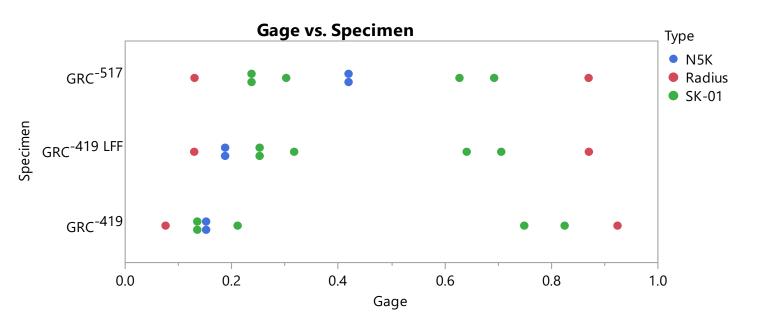
Assumption is made on effective strap width



Test Specimens



- 17-4PH conventionally machined using mill/lathe operations, no EDM
- 8 strain gages placed at various locations with two locations per specimen having replicates





Test Method



 MTS Tension-Torsion machine was used to apply loads to specimens

 Torque was only applied to the specimen, small tension load to engage grippers

 Maximum torque was scaled from balance design loads based on difference in material properties

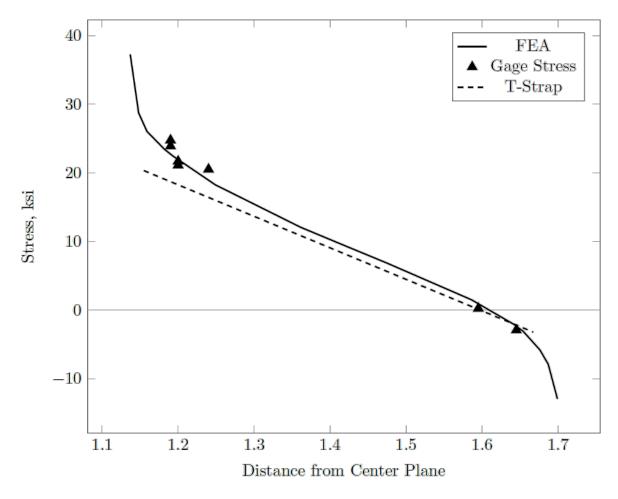


Model Validation Results



- FEA conducted on test specimen
 - Constraints applied dowels
 - Stress measurements taken along beam length for torque
- Gage data taken for maximum torque load applied

 Stress gradient illustrated large reduction of double bending due to torque

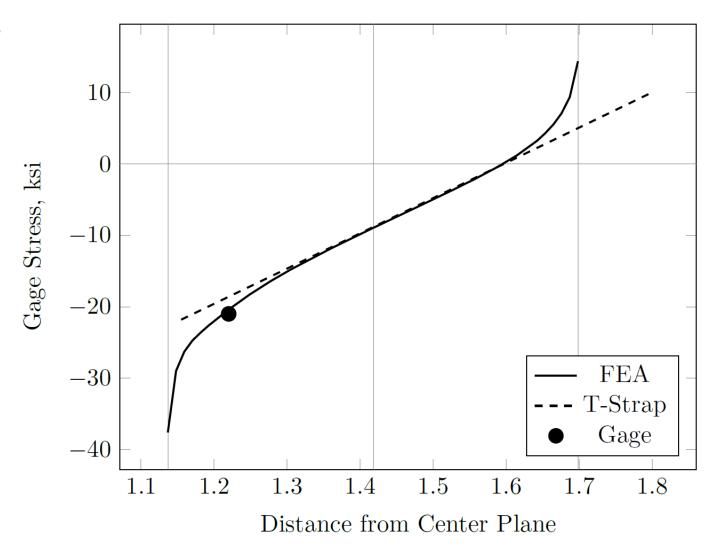


GRC-419R Test Specimen

Model Fit on GRC-419R Calibration Data



- FEA conducted on full-span balance
 - Constraints applied to keys and mating surfaces to replicate fit
 - Stress measurements taken along beam length for torque
- Gage data taken for maximum torque load applied during calibration



Calibration Summary



Broken into multiple phases:

- Phase 1: Gaged and wired
- Phase 2: Temperature Loadings
- Phase 3: PCBs Integrated
- Outputs at full scale loads:

	Output, μV/V	
	Primary	Secondary
Torque	1,685	1,756
Thrust	914	928

No interactions to out-of-plane forces and moments



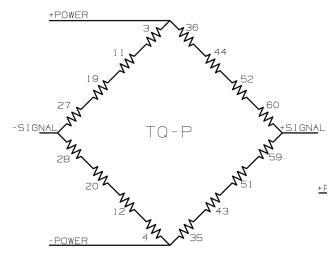
Balance Gaging and Integration

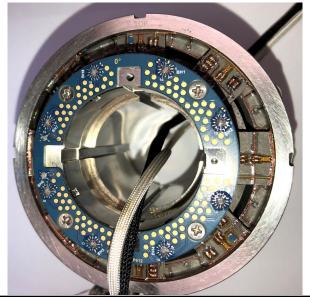


 Gaging of this balance had to follow strict requirements of the telemetry system

 Wiring and terminal placements were predefined to ease PCB integration

 Communication between Telemetry Team, Balance Team, and Fabrication was critical





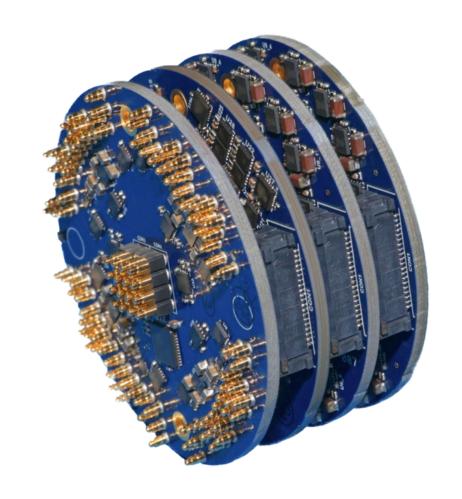
Telemetry System



 Utilizes a custom telemetry system to condition and acquire balance signals and fan blade strain gauges

GRC-419R and GRC-517R use a common telemetry system





Temperature Model



 Linear temperature correction to the sensitivity constant:

$$SC_{@T} = SC_{@RT}(1 - K(T_{op} - T_{ref}))$$

 All hardware pre-heated in oven and moved to calibration stand

 Temperature corrected residuals were within 0.2%F.S.





Summary



- GRC-419R project required high level of integration between all team members
 - Complex model integration
 - Embedded electronics interfacing
 - Multiple mating interfaces
- Modifications to analytical methods allowed for prediction of beam stresses for rapid design space exploration
 - Validated analytical modes and FEA models with fast/low-cost testing
- Successfully calibrated and temperature corrected balance to below 0.25% F.S.