

Preparations for Tomographic Background-Oriented Schlieren at the 31-Inch Mach 10 Wind Tunnel

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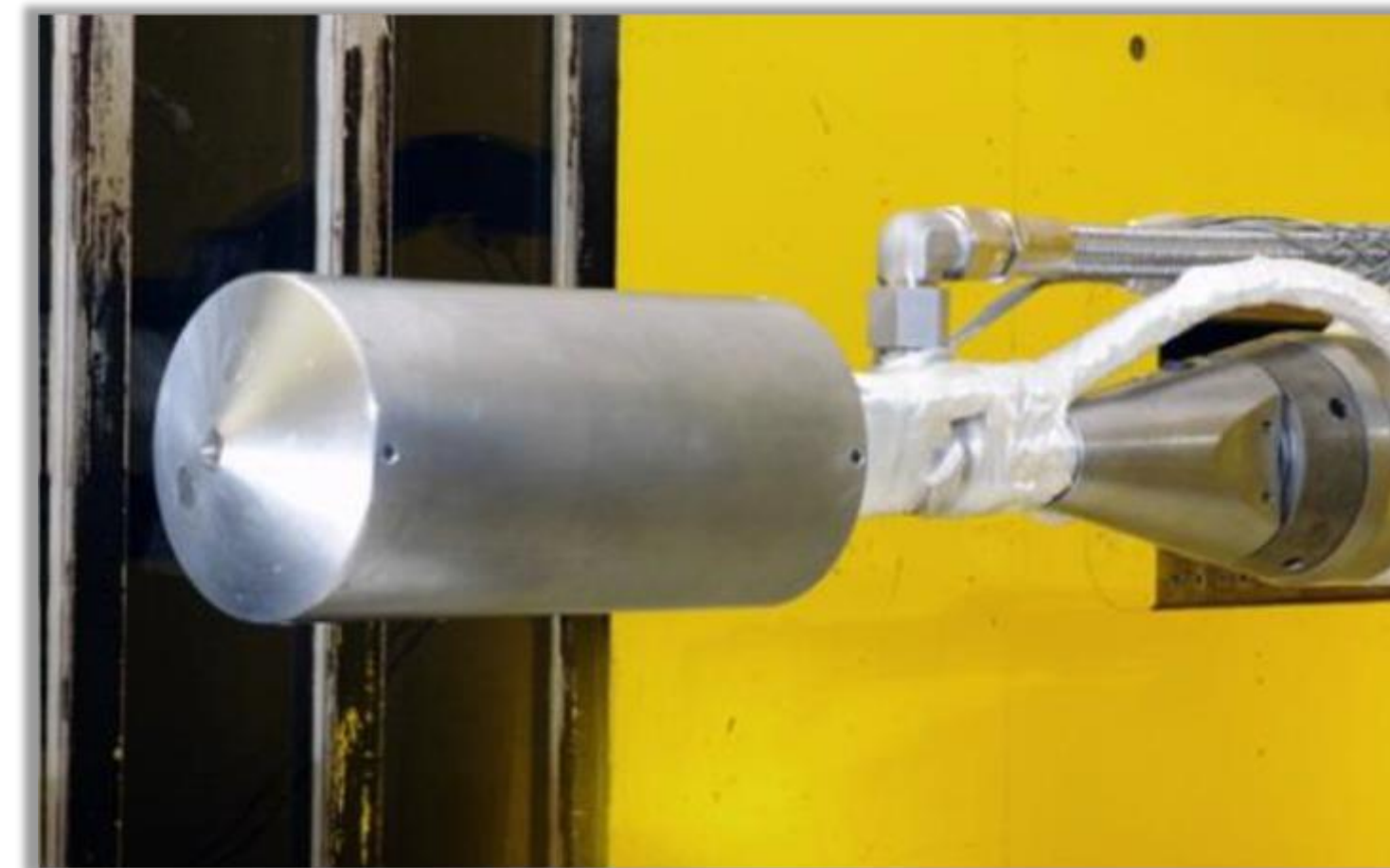
Stephen B. Jones

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Motivation

- Single- and multi-nozzle supersonic retropropulsion (SRP) test
 - NASA Langley Research Center's 31-Inch Mach 10 wind tunnel
 - Compare off-body flow visualization measurements with on-body high-speed pressure transducer measurements
 - Previous test performed with conventional schlieren
 - Demonstrate high-speed tomographic background-oriented schlieren (tomo BOS)

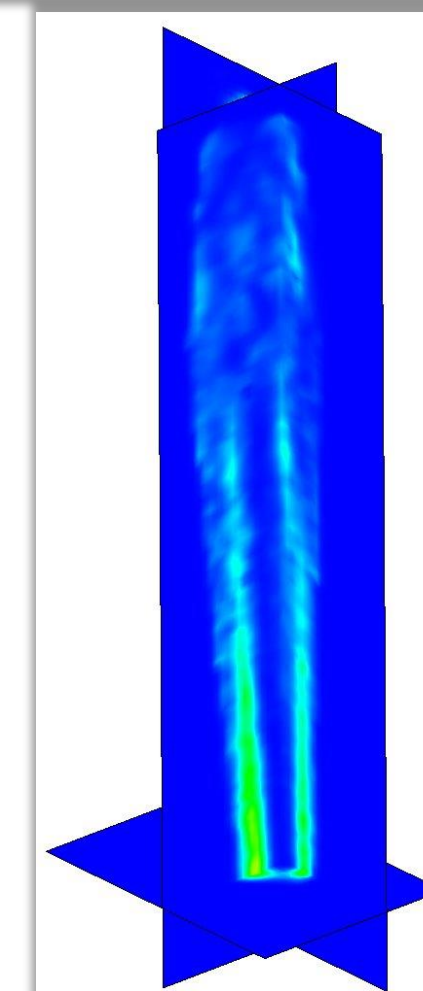
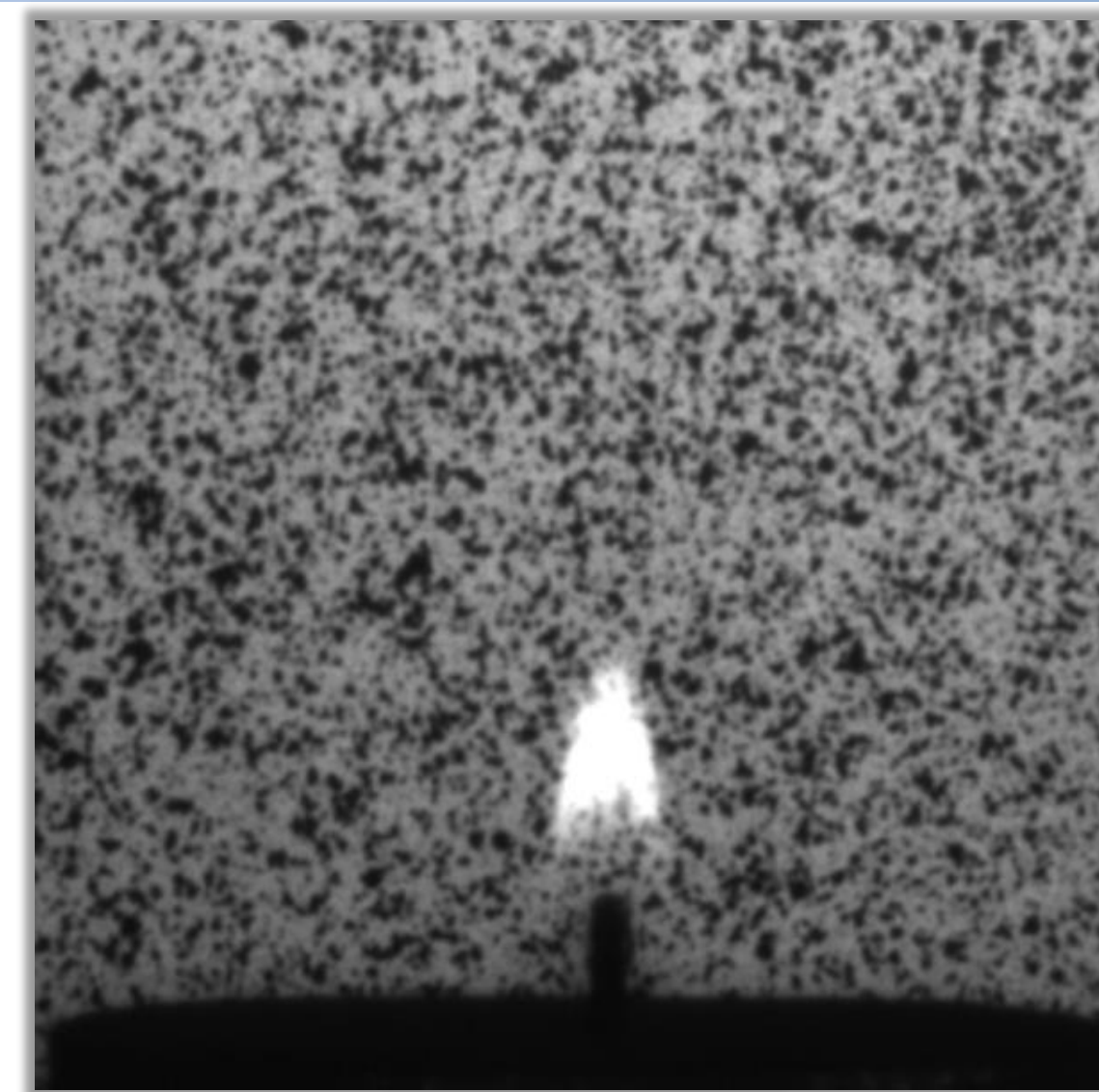
- Background-oriented schlieren
 - Camera images patterned background
 - Refractive index (density) variations result in distortion of pattern
 - Algebraic reconstruction technique uses 2D BOS image data from multiple perspectives to compute density field reconstruction



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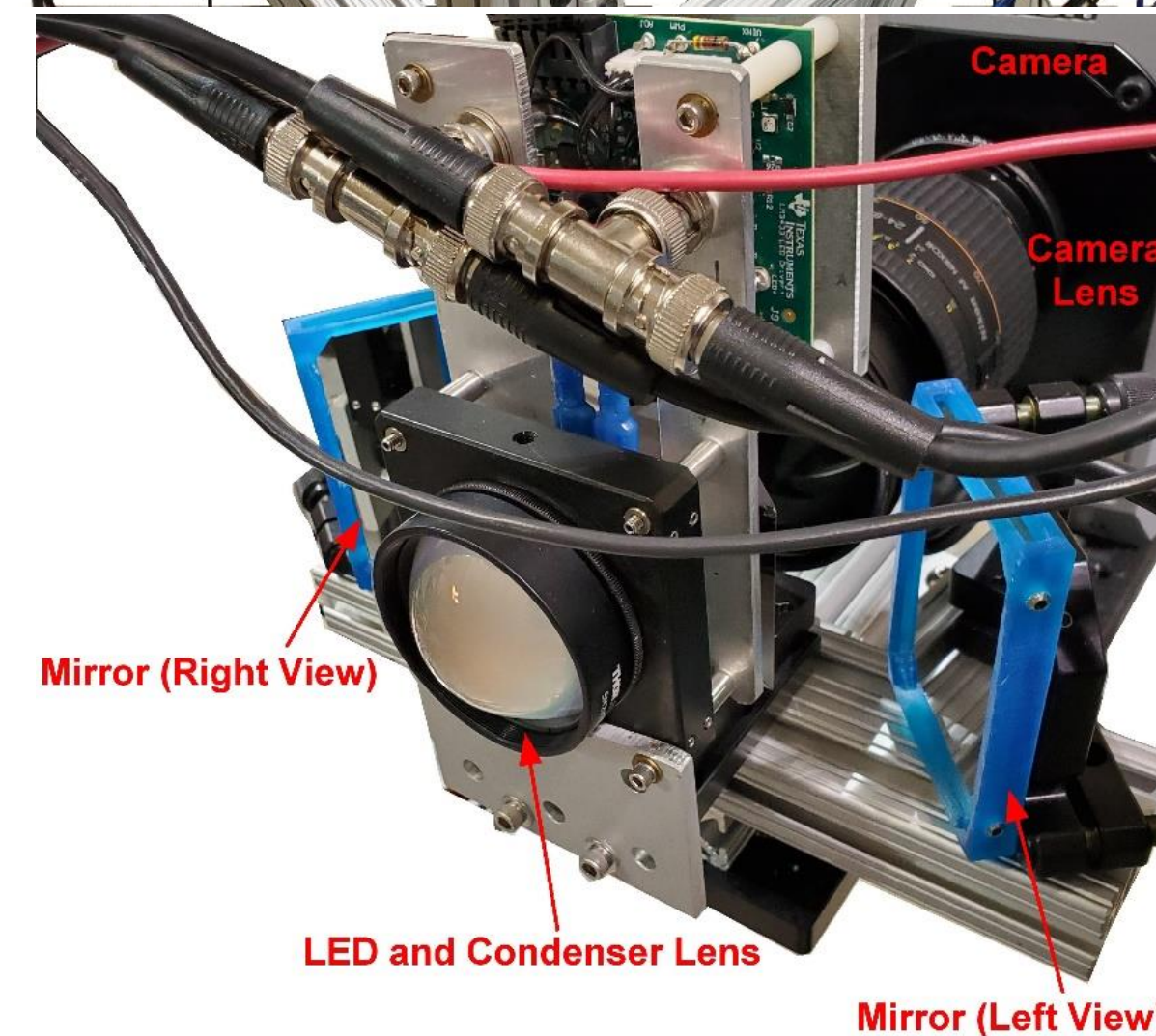
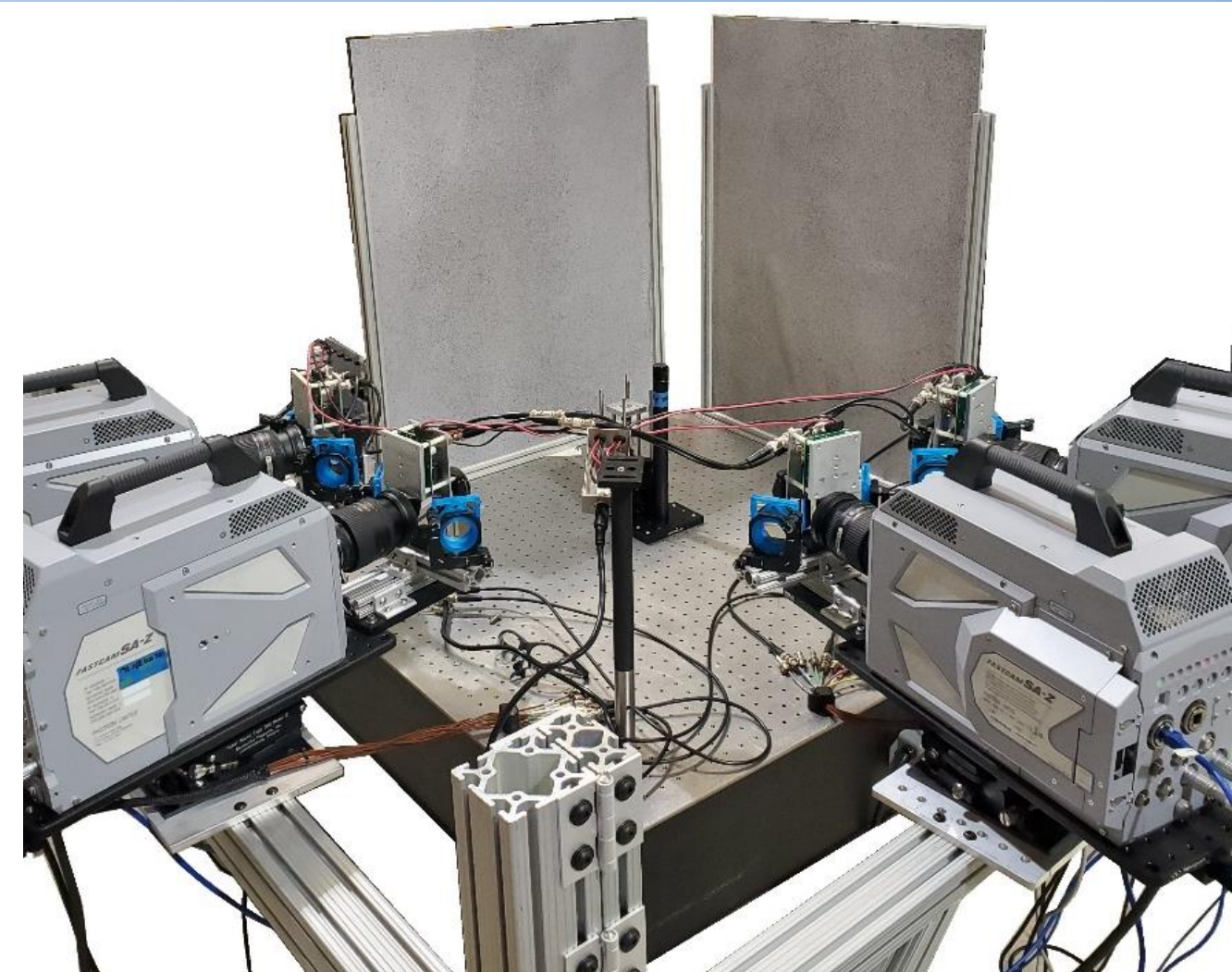
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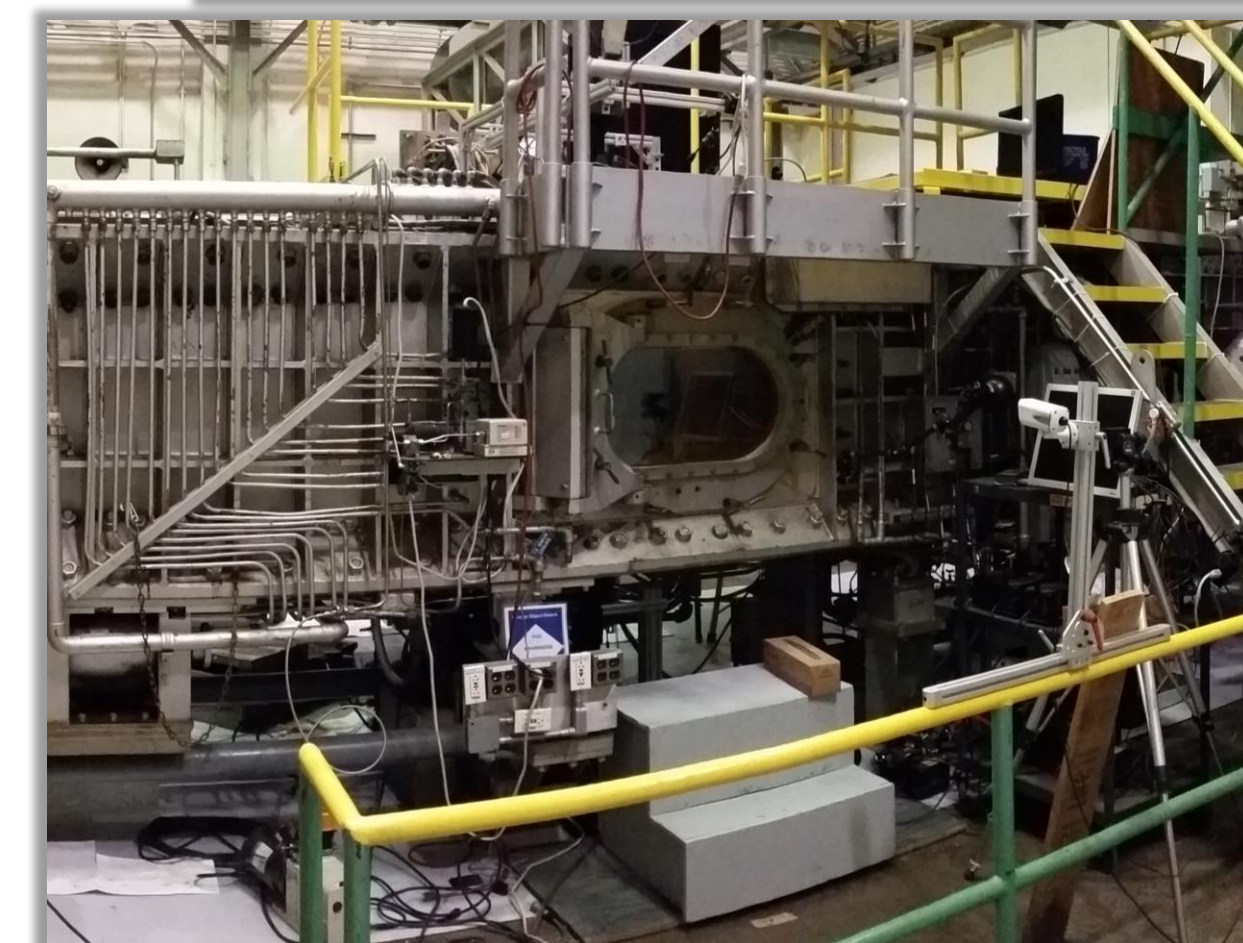
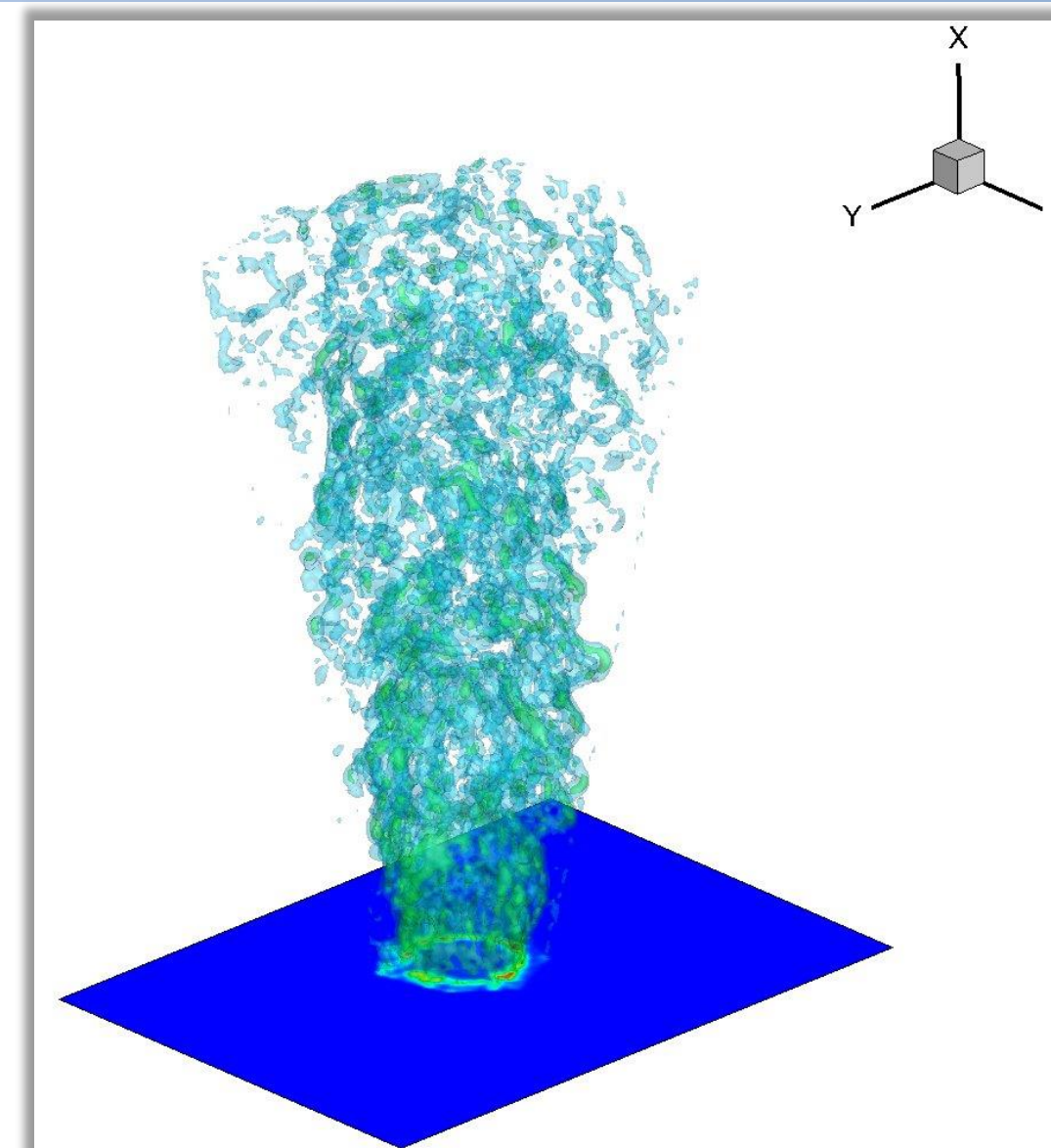
Preparations for Test Campaign at 31-Inch Mach 10 Wind Tunnel

- Previous high-speed tomo BOS laboratory testing
 - Obtained tomo BOS data with four high-speed Photron SA-Z cameras
 - Frame splitters in front of each camera provided two unique perspective views of the helium jet
 - Illumination provided by single off-axis pulsed LED
 - Camera placement mimicked optical restrictions of top and side wind tunnel windows
- Wind tunnel test requires more extensive preparation
 - Camera placement and alignment
 - Illumination method
 - Pattern placement and survivability
 - Calibration procedure



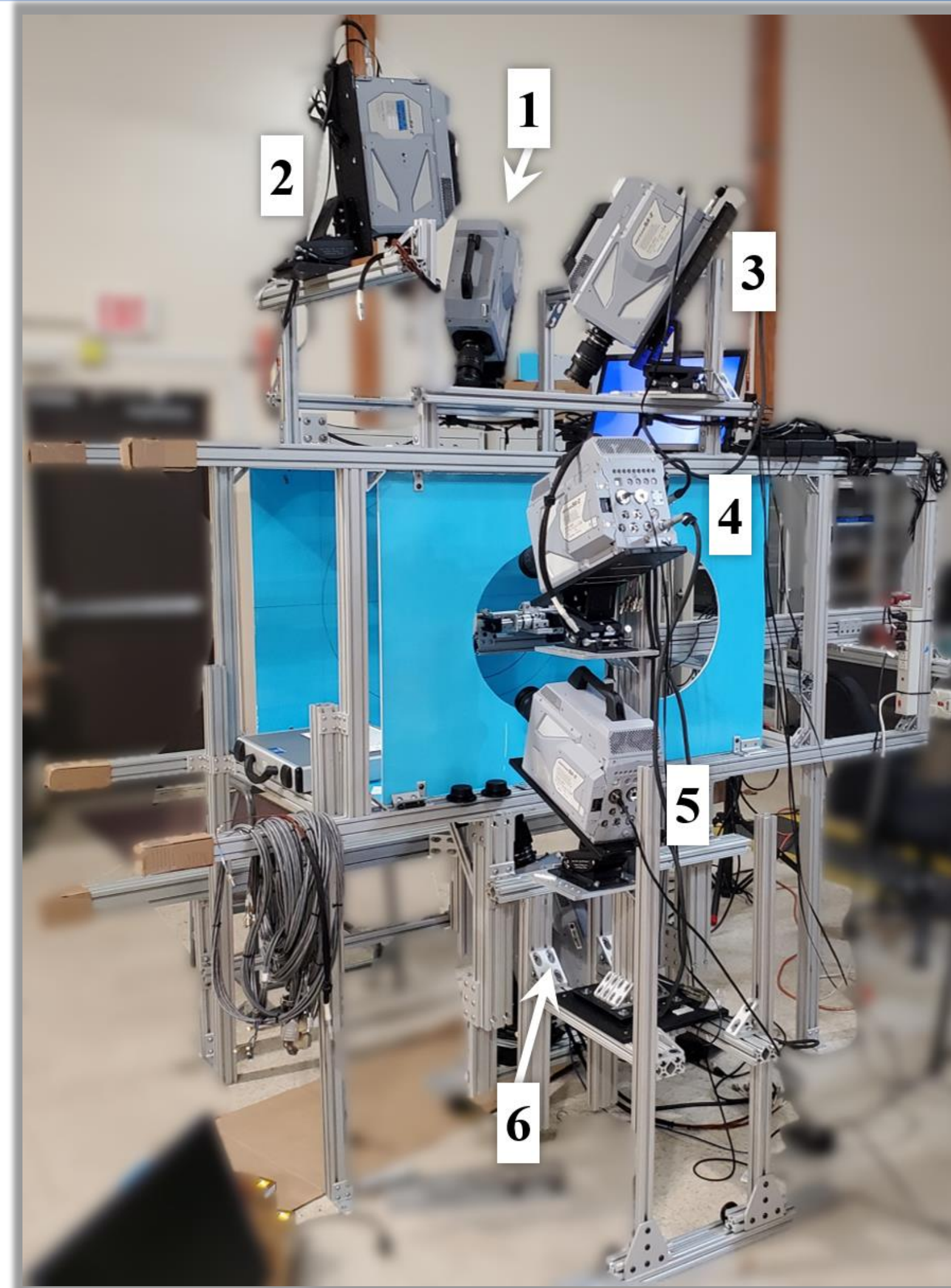
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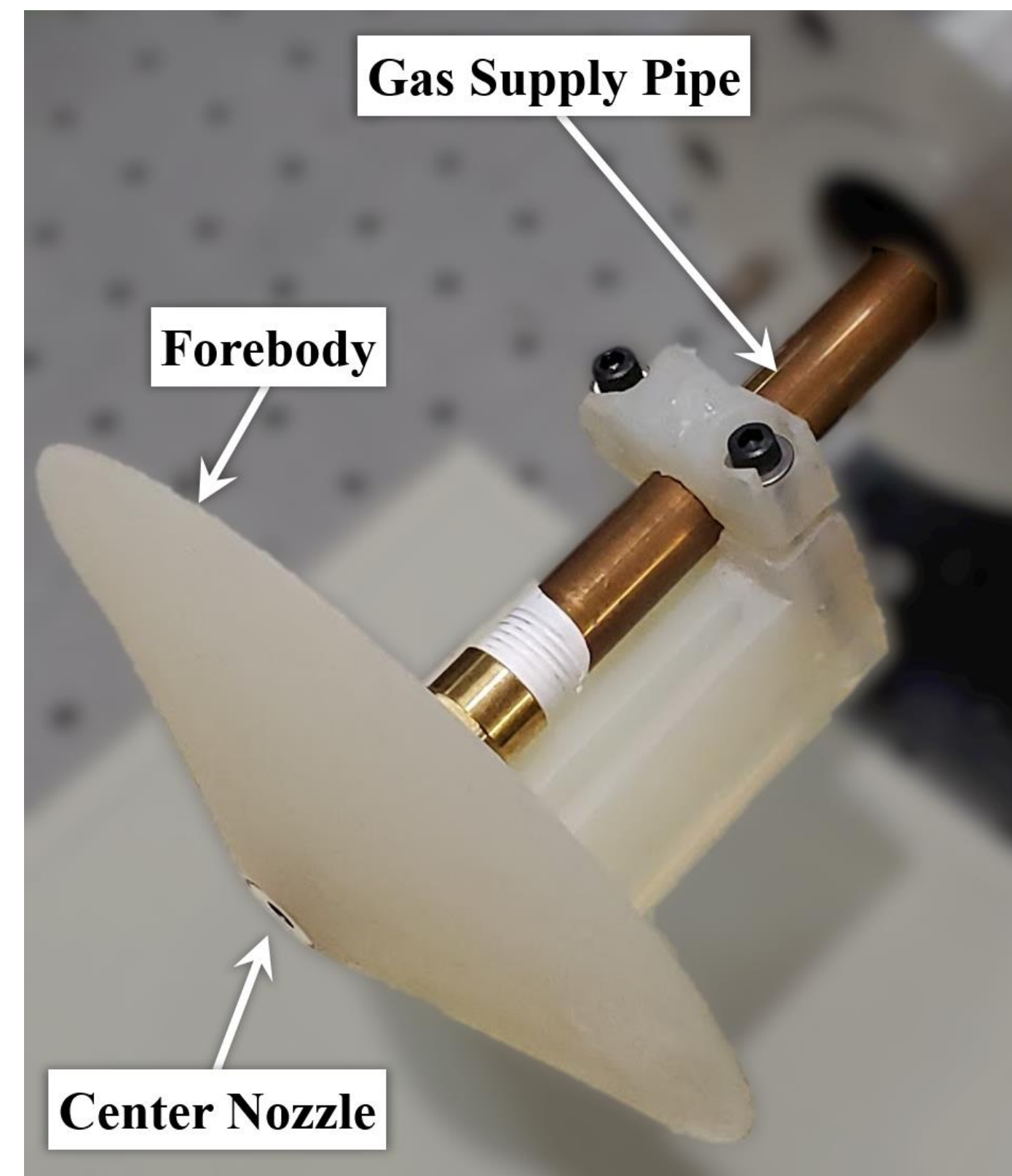
31-Inch Mach 10 Wind Tunnel Full-Scale Mock-Up

- Outer structure made of 80/20 t-slotted profiles
- Walls made of Alumalite paneling
- Window cutouts sized and placed exactly as in the facility test section
- Six cameras mounted on Huber Type D 300 pitch/roll/yaw stages for fine alignment
- Mock-up of wind tunnel model mounted to an 80/20 sting assembly that allowed adjustment in streamwise and spanwise directions using linear bearings



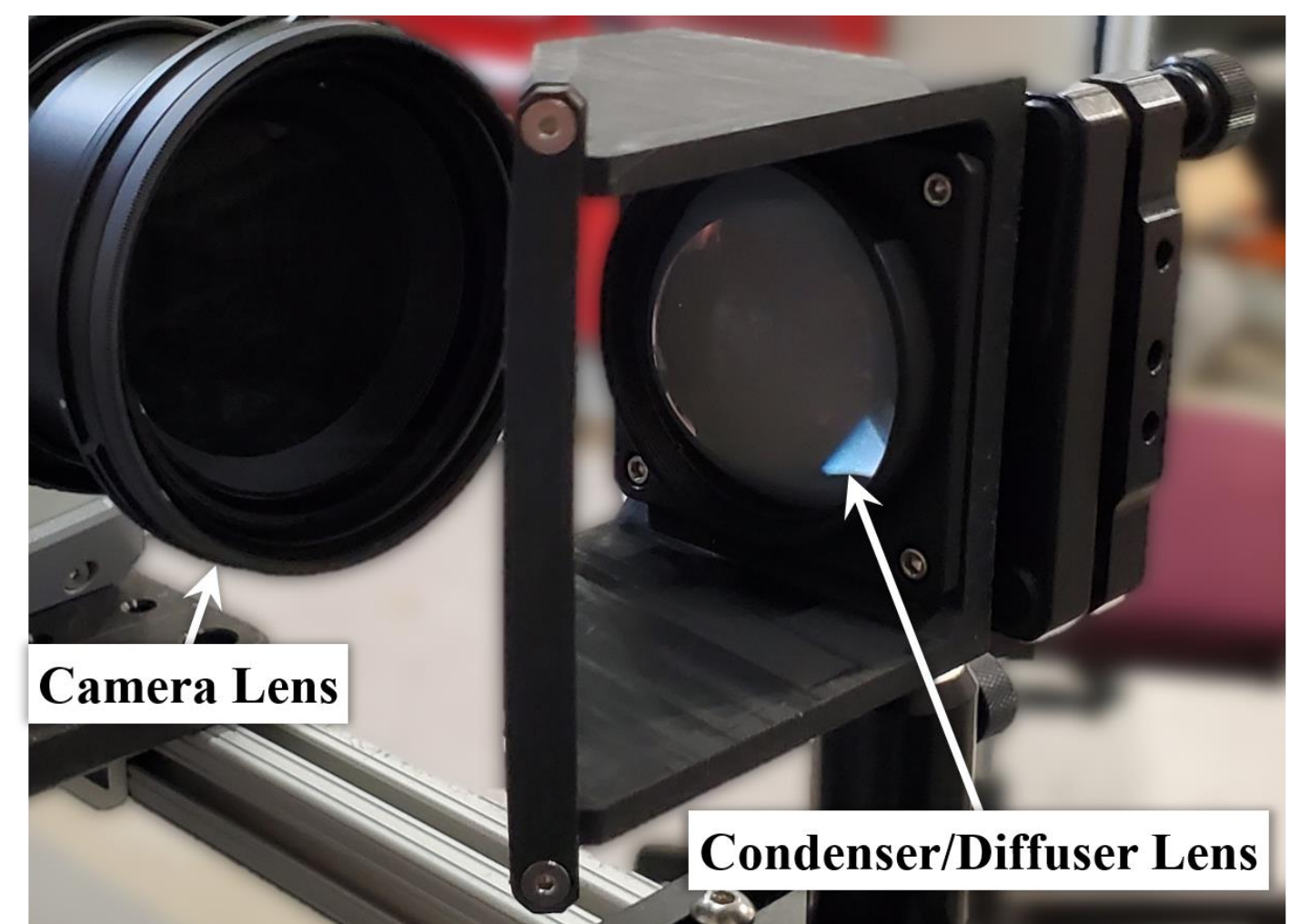
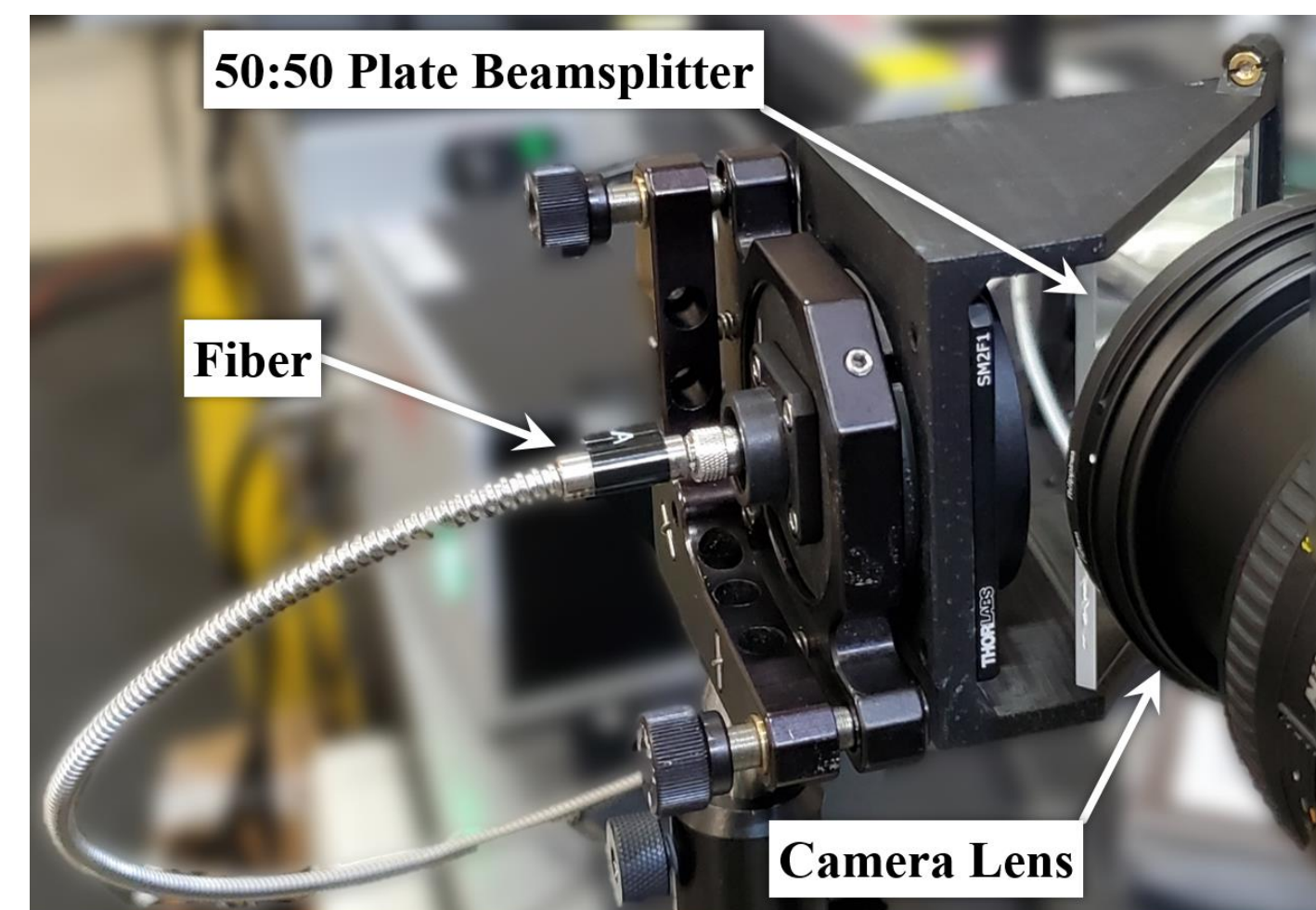
3D-Printed Model of SRP Forebody

- Aid in alignment of cameras and provide high-speed jet for testing tomo BOS system performance
- Did not want to damage the actual model, which has been instrumented with a significant number of high-speed pressure transducers
- 70° sphere-cone forebody with same dimensions as full wind tunnel model
- Needed to replicate the exact dimensions of the model forebody, which would be in the field-of-view of each camera
- Embedded 60° cone spray center nozzle that allowed for high-pressure gas supply (helium, compressed air) to be connected for BOS imaging purposes



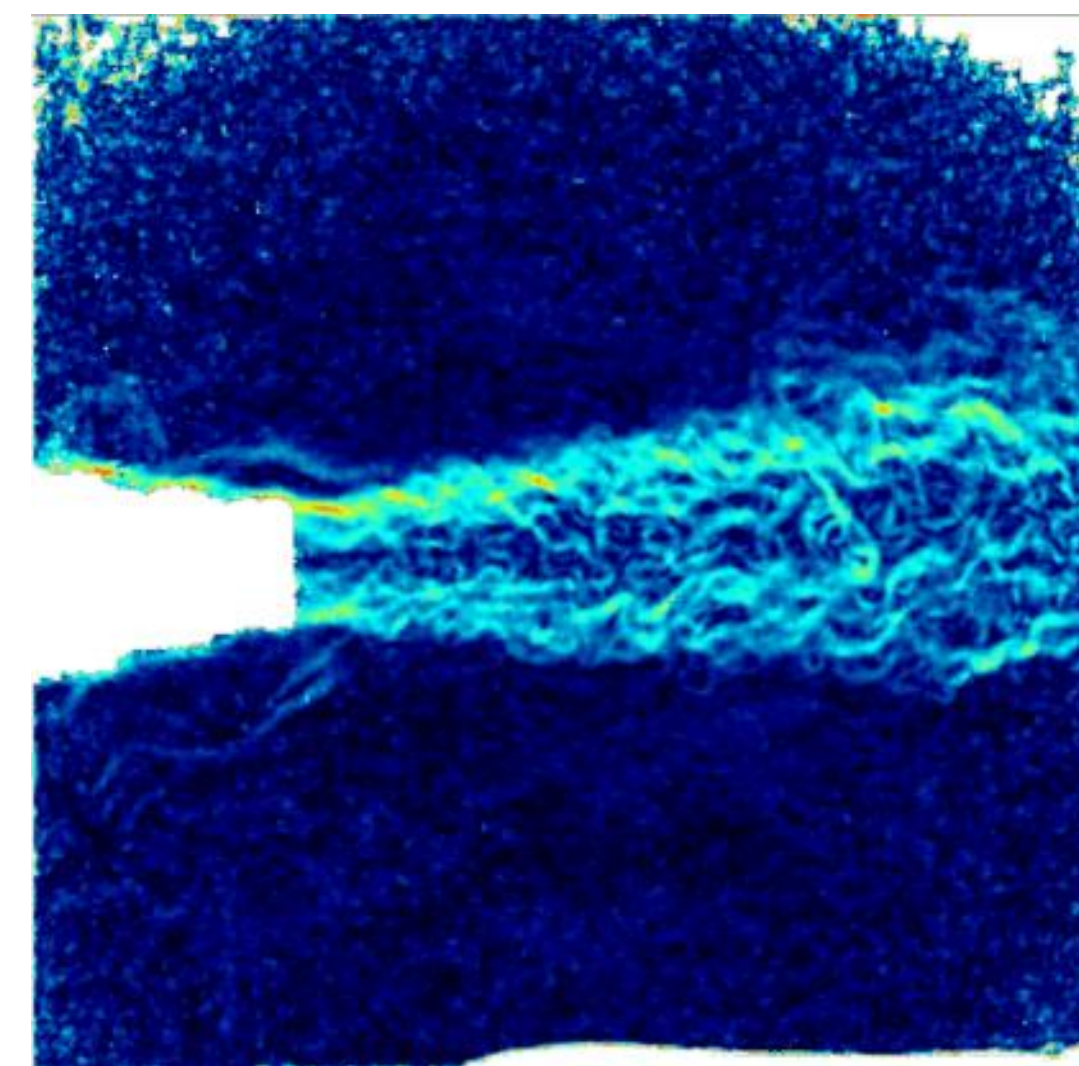
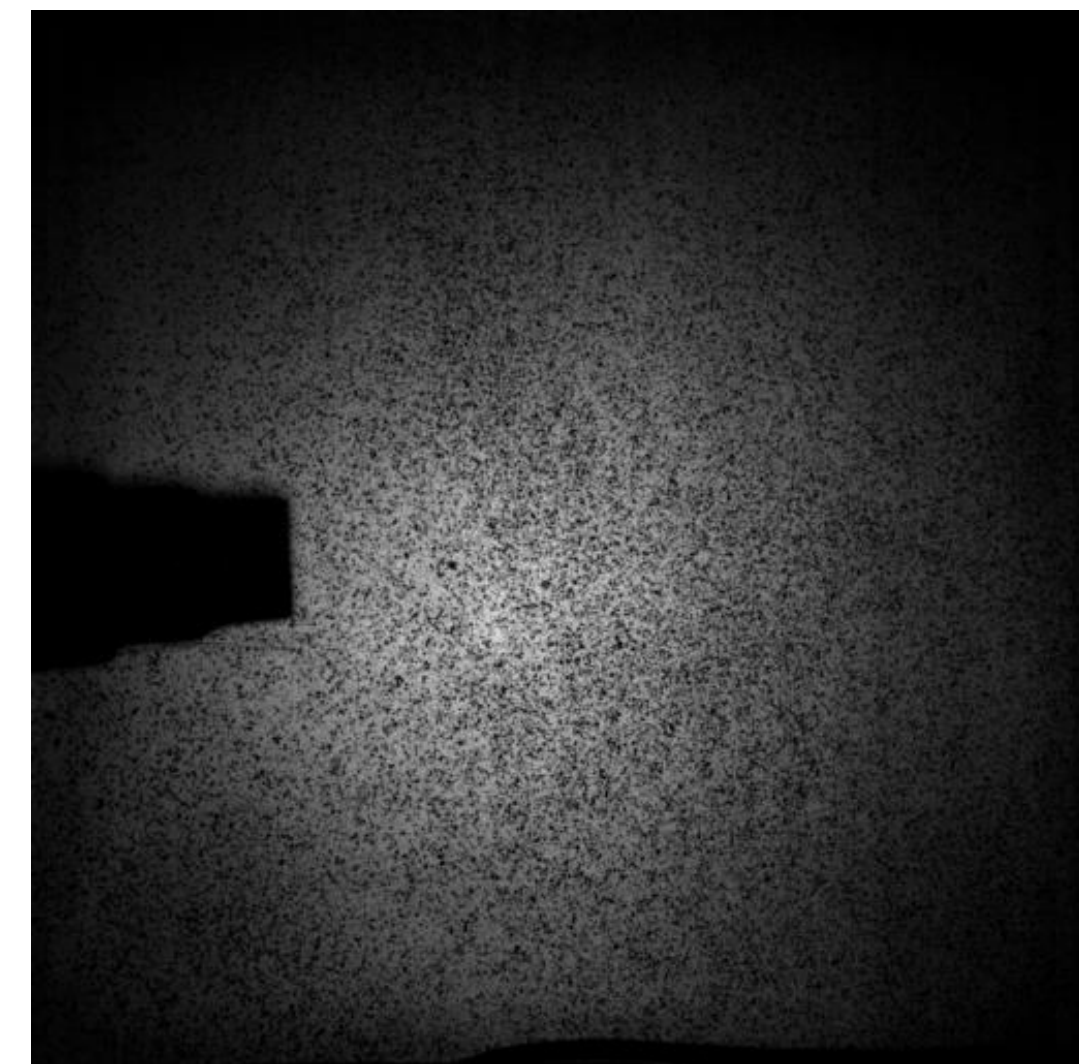
Fiber-Based Illumination Modules

- Original LED-based illumination modules used did not provide on-axis illumination, which gives maximum gain from retroreflective background material
- This required longer pulse width to sufficiently illuminate background, limiting the framing rate to 10-20 kHz
- Intensity of LED was also limited
- Constructed 3D-printed module that allowed us to couple high-intensity pulsed laser light onto the camera's optical axis using a 50:50 plate beamsplitter
- Used a custom 7-fiber bundle that coupled to a Cavitar Cavilux HF pulsed laser light source operating at 640 nm
- Demonstrated sufficient illumination intensity and uniformity using this approach



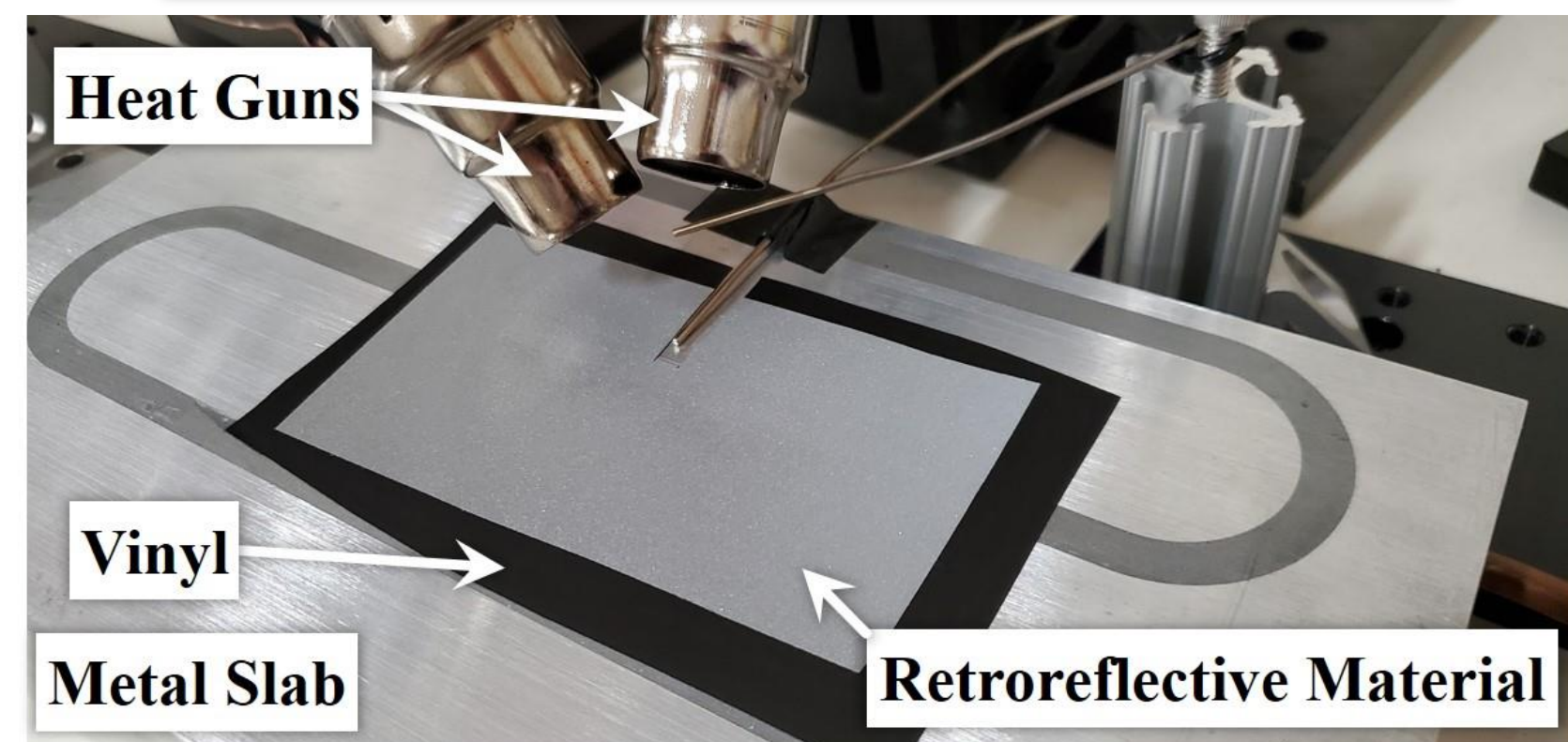
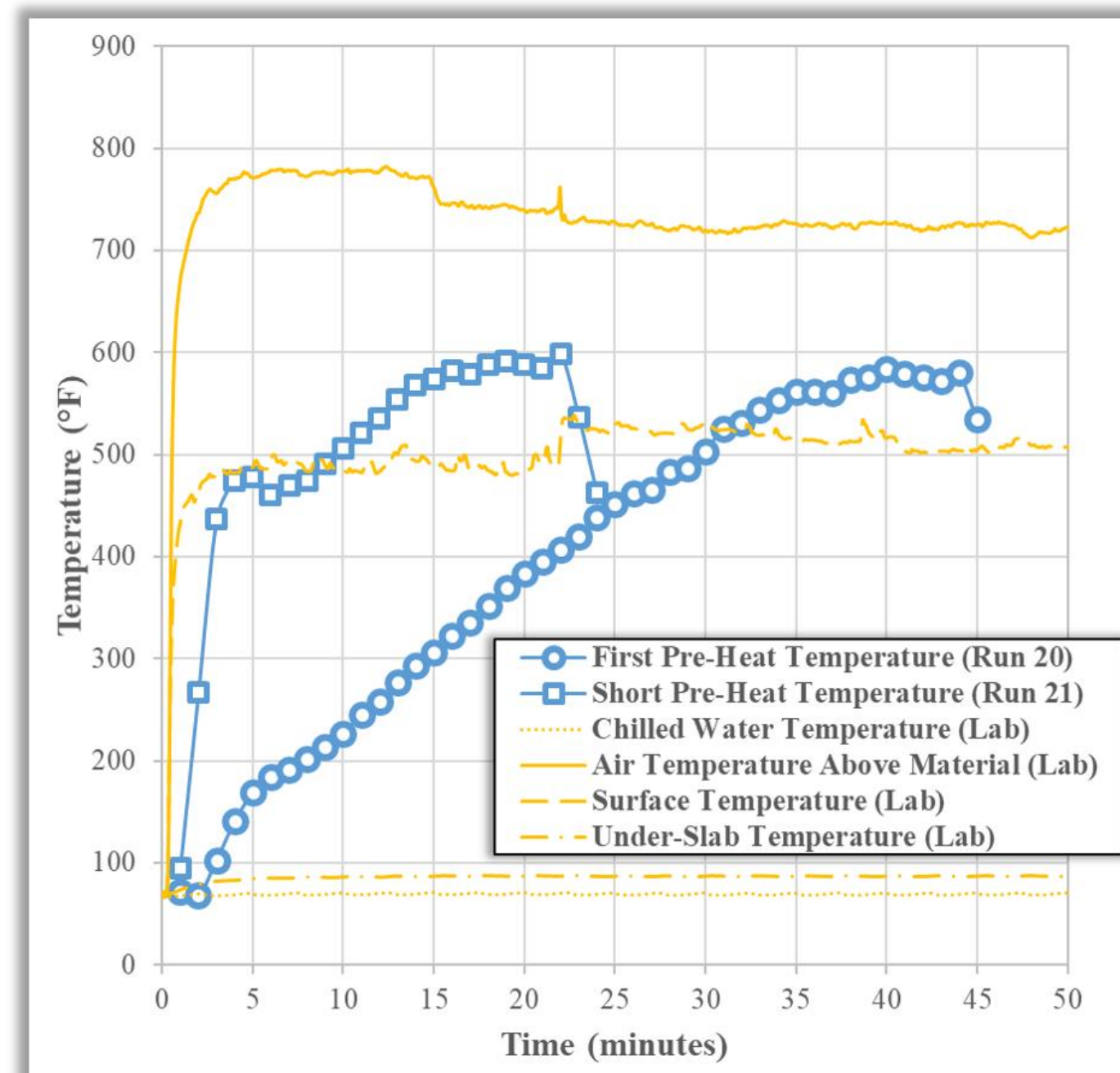
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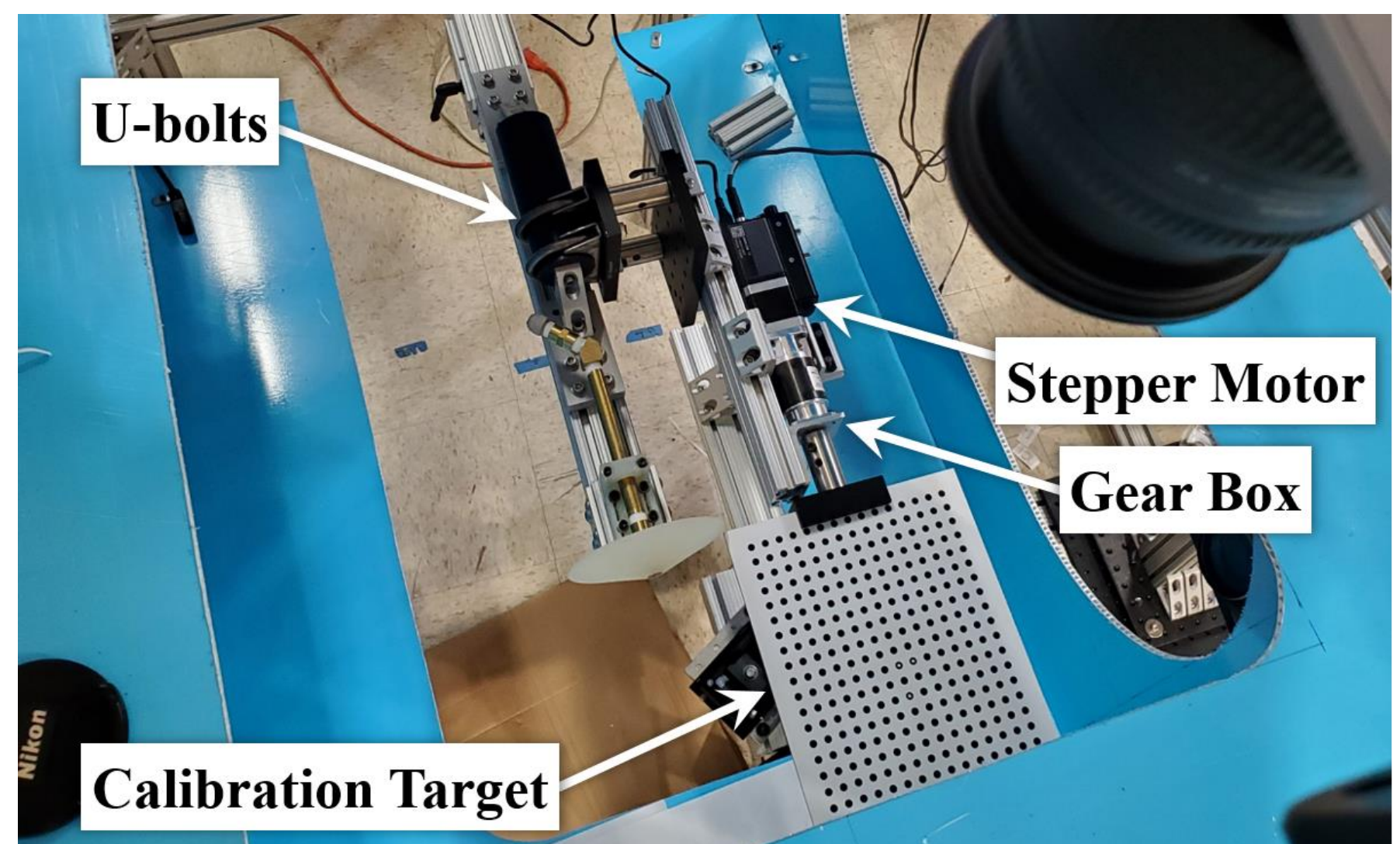
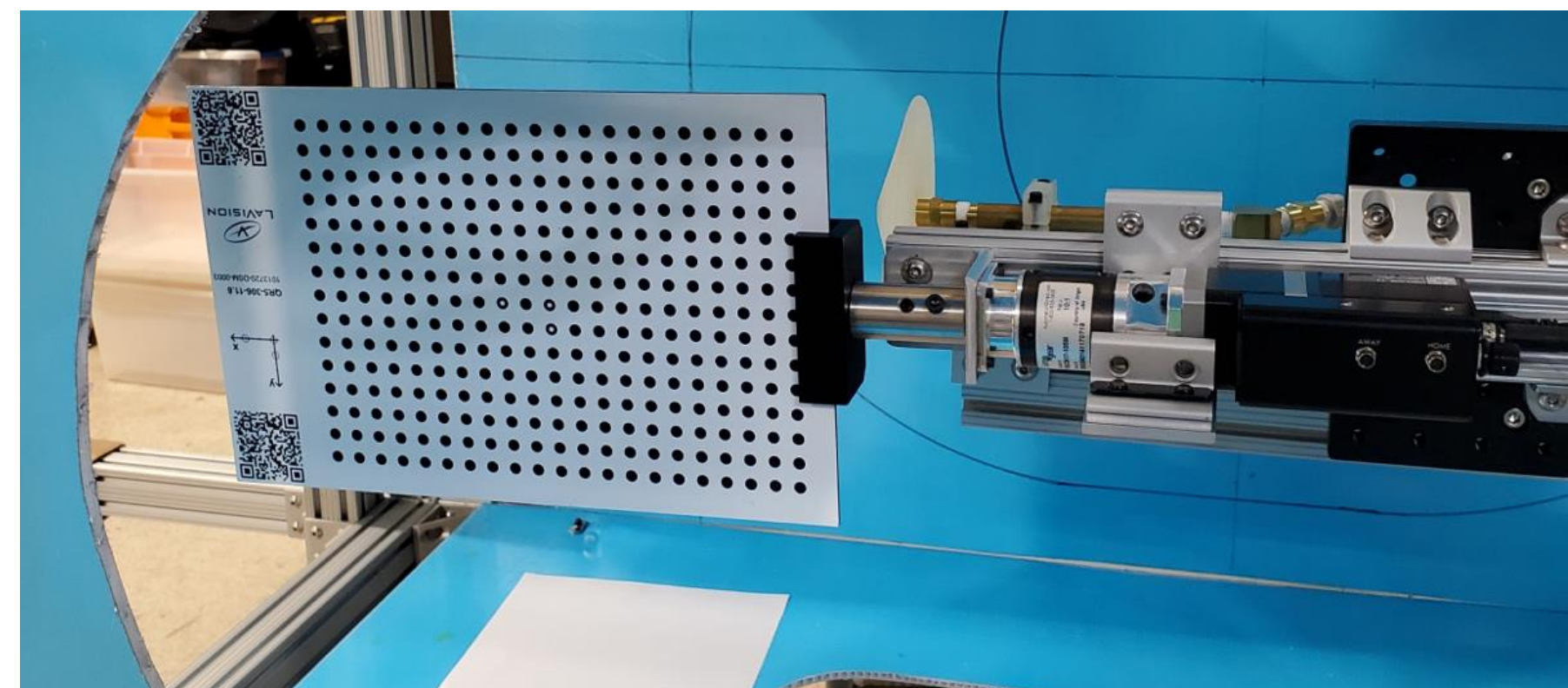
Survivability of Retroreflective Background

- Camera layout will require patterned retroreflective material to be placed on interior surfaces of wind tunnel
- For ease of application/removal, 3M Scotchlite 7610 is mounted onto 3M 1080 vinyl wrap material, which is then mounted on tunnel wall
- 31-Inch Mach 10 is pre-heated at beginning of day and between each run, but test section temperature during this procedure was unknown
- Performed measurements of static temperature in test section during a pre-heat
- Performed laboratory test with background material mounted to water-cooled metal slab subjected to high temperature air to simulate pre-heat
- Determined that material is highly likely to survive pre-heat



Remotely Controlled Calibration System

- No physical access to test section once cameras have been mounted around tunnel and aligned
- Designed remotely controlled calibration system consisting of a stepper motor, gear box, and calibration target
- System mounts to 2-inch-diameter SRP model using two U-bolts
- Model will be injected manually such that rotation axis of calibration system is aligned to tunnel centerline
- Calibration target is rotated through all camera views for volumetric calibration



Conclusions

- Discussed preparations for supersonic retropropulsion (SRP) test that will be conducted in 31-Inch Mach 10 wind tunnel at NASA Langley Research Center
 - Using a six-camera high-speed tomographic background-oriented schlieren (tomo BOS) system to visualize jet flow
 - Constructed full-scale mock-up of wind tunnel facility to optimize camera placement and alignment
 - Fabricated 3D-printed SRP forebody model with center nozzle to aid in alignment and provide high-speed jet for testing out tomo BOS system
 - Split pulsed laser source into seven fibers using 7-fiber bundle and coupled light onto camera axis with a 50:50 plate beamsplitter
 - Performed tests to ensure retroreflective backgrounds would survive test section pre-heat
 - Designed remotely-controlled calibration system
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- Support provided by Transformational Tools and Technologies (TTT) and Aerodynamic Evaluation and Test Capabilities (AETC) projects