

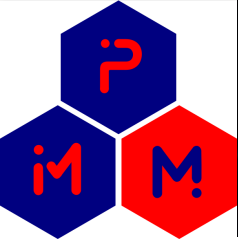


# UPDATES TO THE PREDICTIVE MATERIAL MODELING SOFTWARE TOOLS

A. Borner, J. Meurisse, G. Bellas-Chatzigeorgis, B. Dias, S. F. Izquierdo, F. Semeraro, K. Swaminathan Gopalan, J. Thornton, N. Mansour

**AMA Inc., at NASA Ames Research Center**

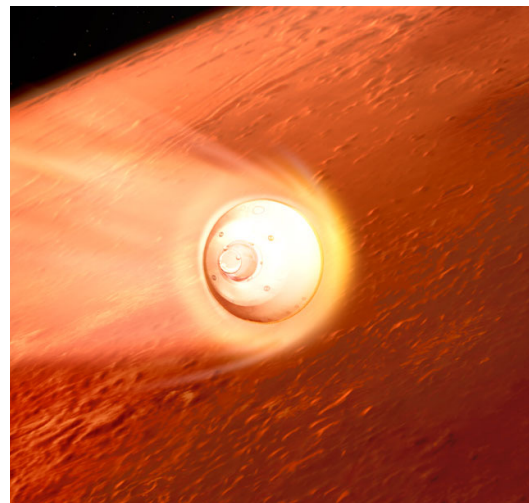
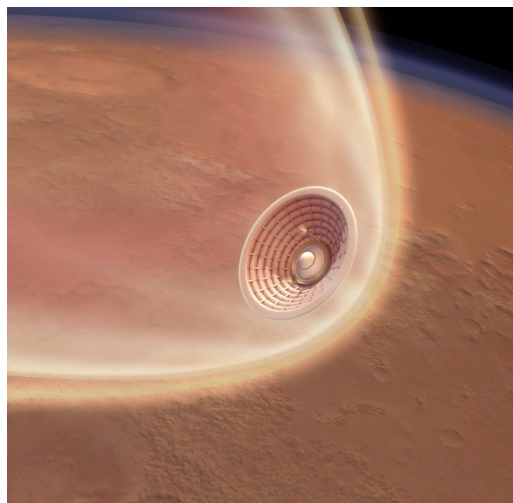




# Thermal Protection Systems (TPS)



## Need for Thermal Protection Systems



## Ground-based Testing

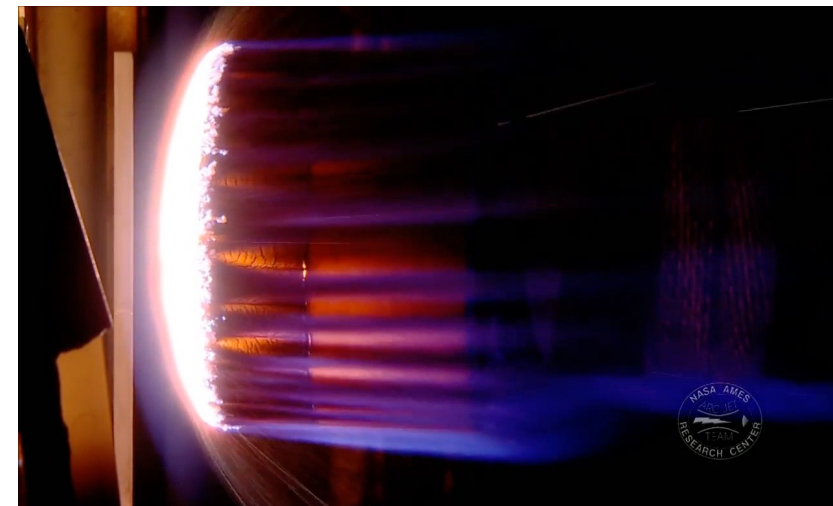
P. Agrawal et. al. 2016.



Virgin PICA Sample



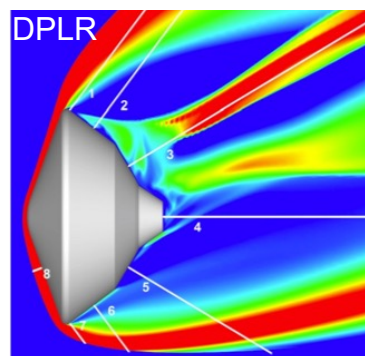
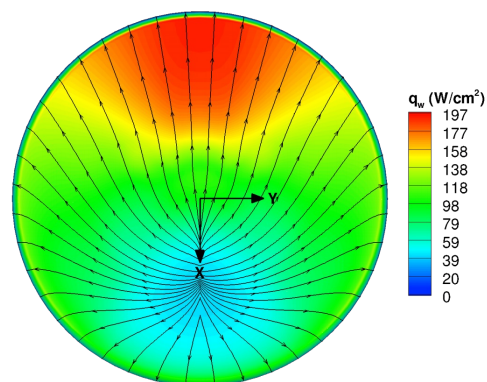
Charred PICA Sample



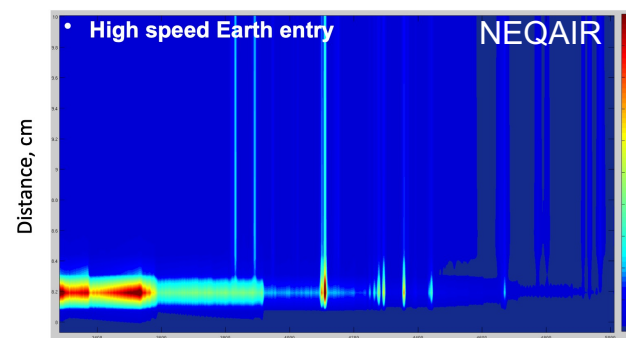
Arc Jet Testing of TPS

## Computational Modeling

### Hypersonic CFD

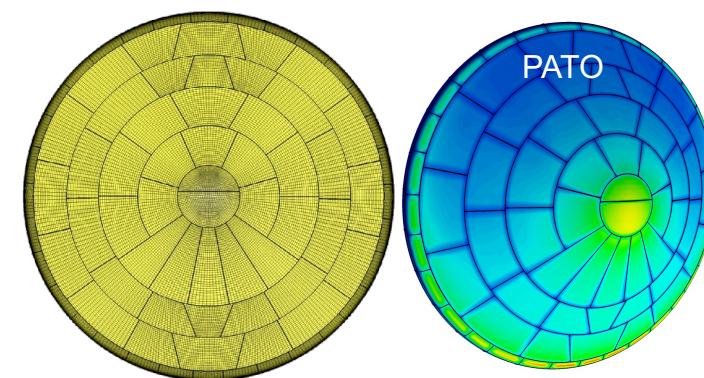


### Radiation Analysis

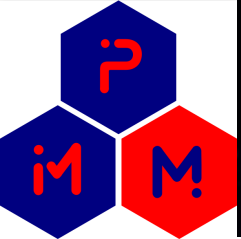


Wavelength, A

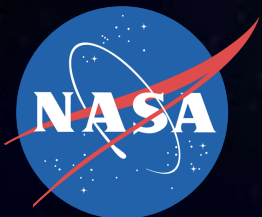
### Material Response



Computational modeling is critical for entry systems



# NASA Modeling Capabilities



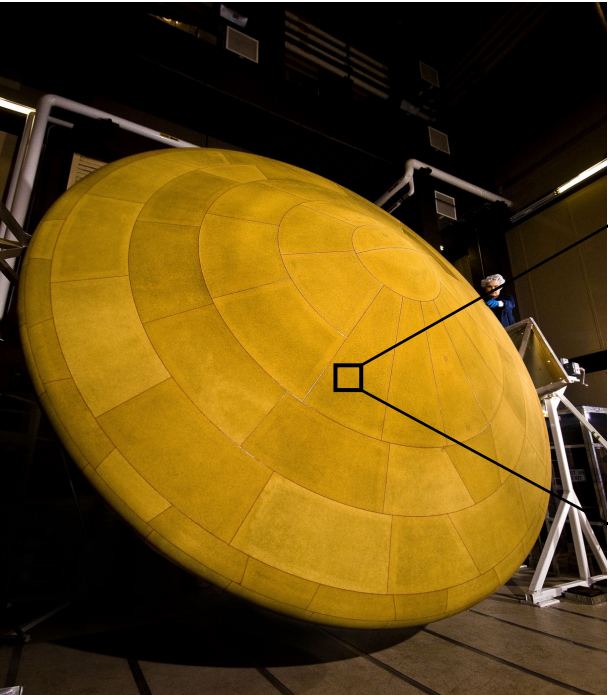
FIAT

CHAR

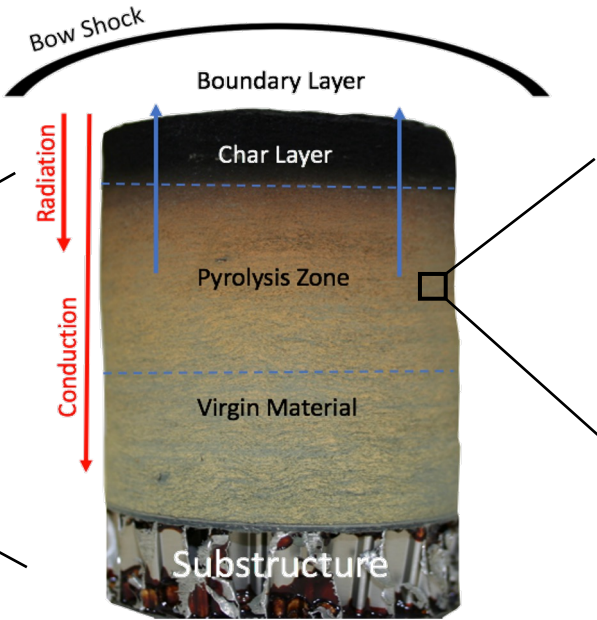
Icarus

PATO

Full scale



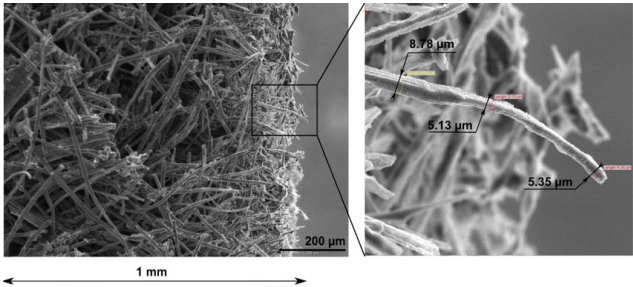
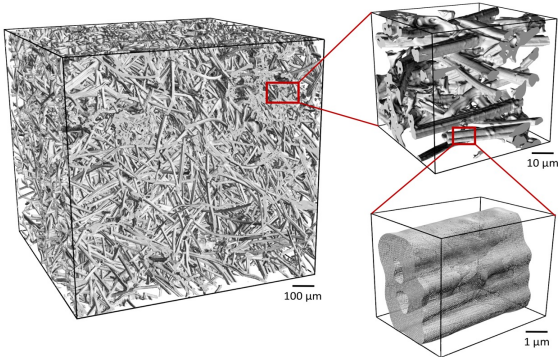
Mars Science Laboratory (MSL) heat shield



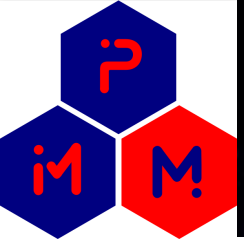
PuMA



Microscale

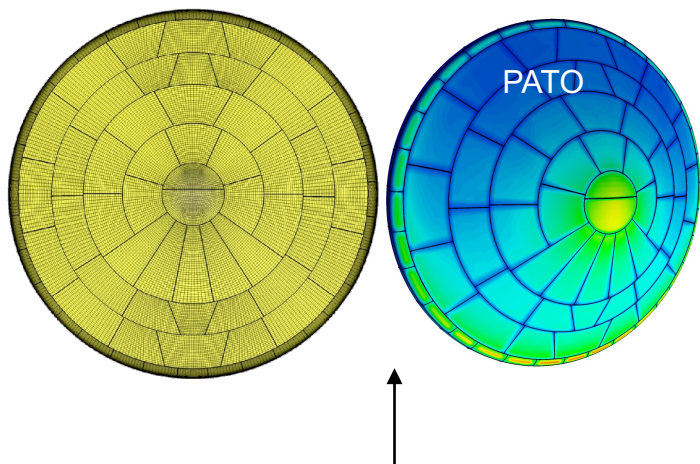


Carbon-fiber microstructure

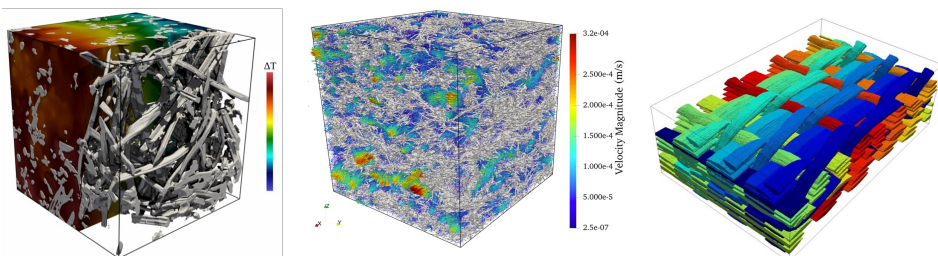


# Importance of Microstructure Analysis

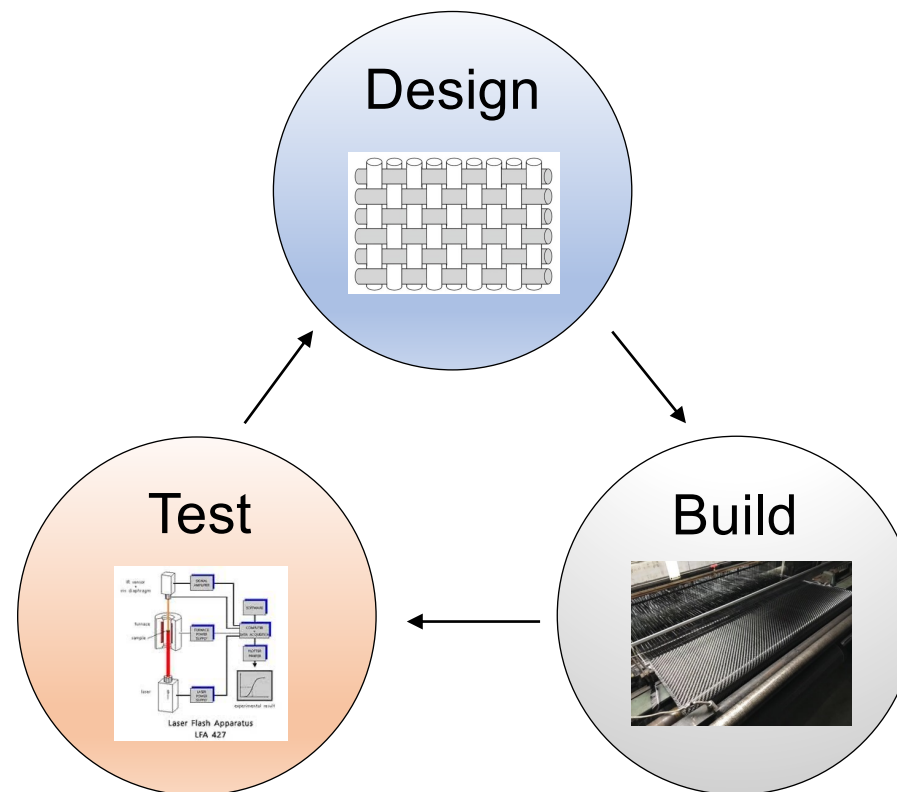
## Improved Material Response



*Microscale materials modeling*

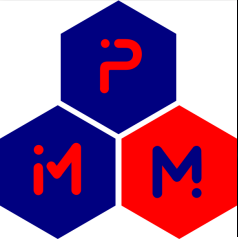


## Improved Material Design

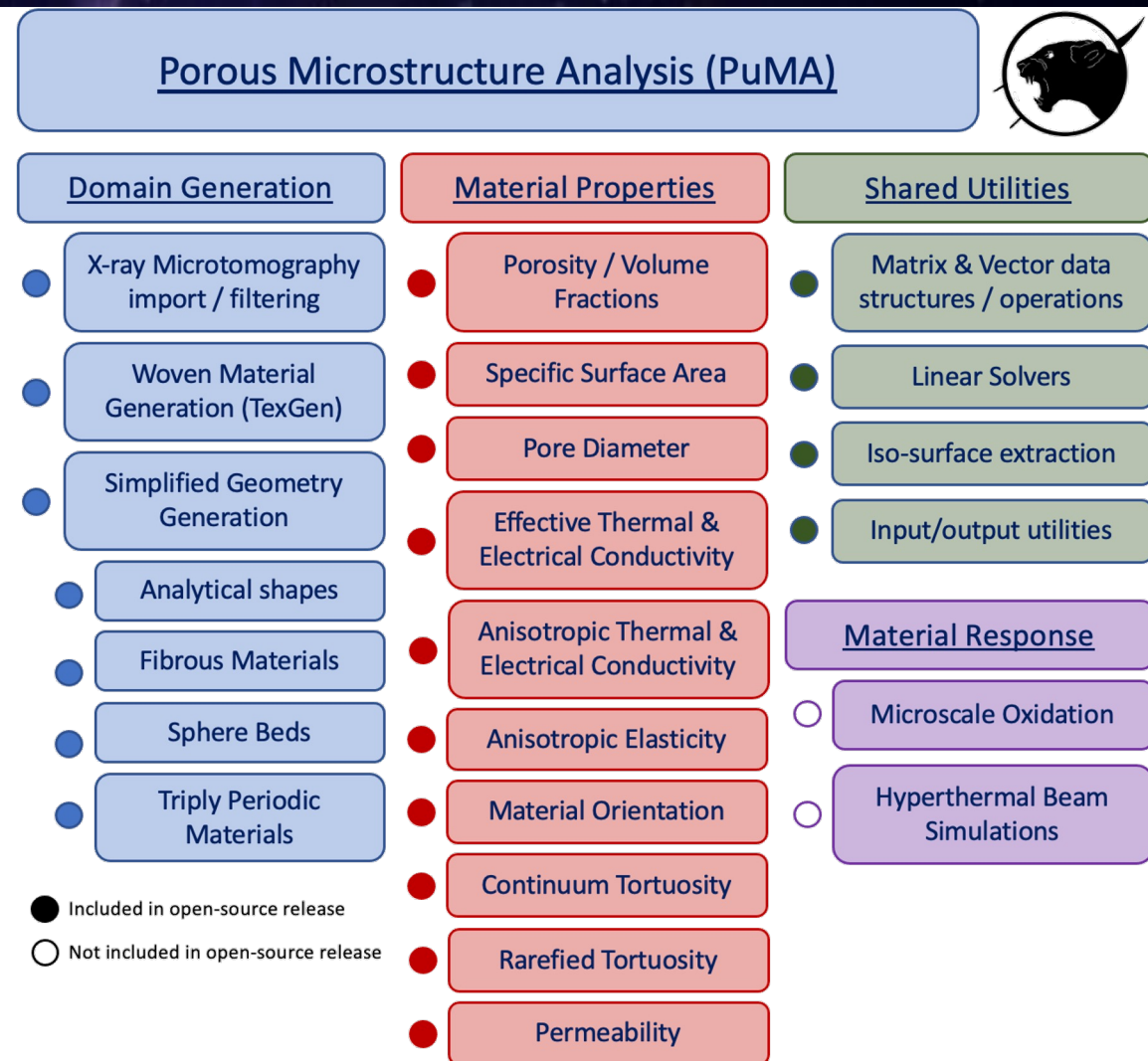
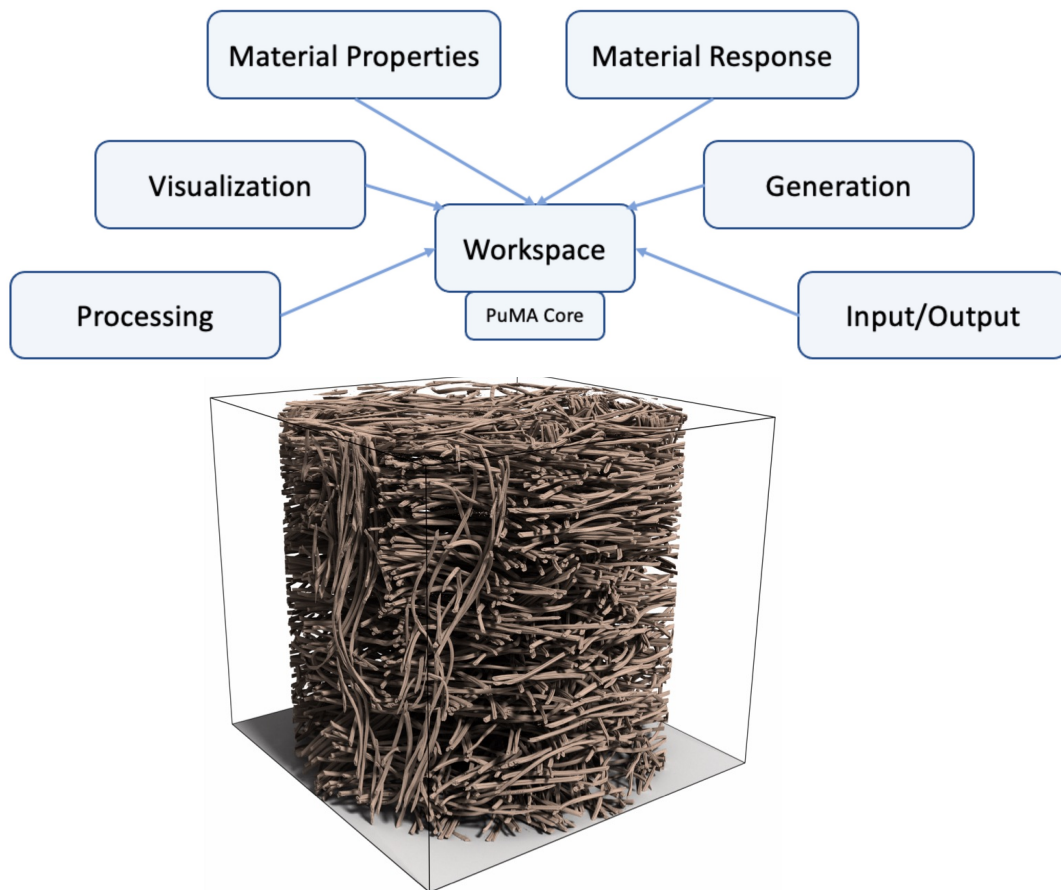
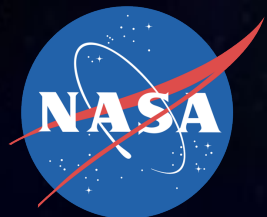


*Current material design processes are costly and slow*

**Microscale modeling improves NASA's predictive modeling and can improve material design**



# The PuMA software



**Installation:** `conda install -c conda-forge puma`

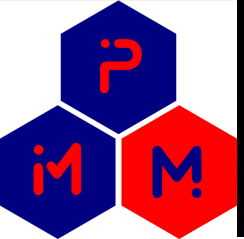
**Open-source repository:** <https://github.com/nasa/puma>

**Documentation:** <https://puma-nasa.readthedocs.io>

**Video tutorials:** [PuMA YouTube channel](#)

**Main developers:** [federico.semeraro@nasa.gov](mailto:federico.semeraro@nasa.gov), [joseph.c.ferguson@nasa.gov](mailto:joseph.c.ferguson@nasa.gov)

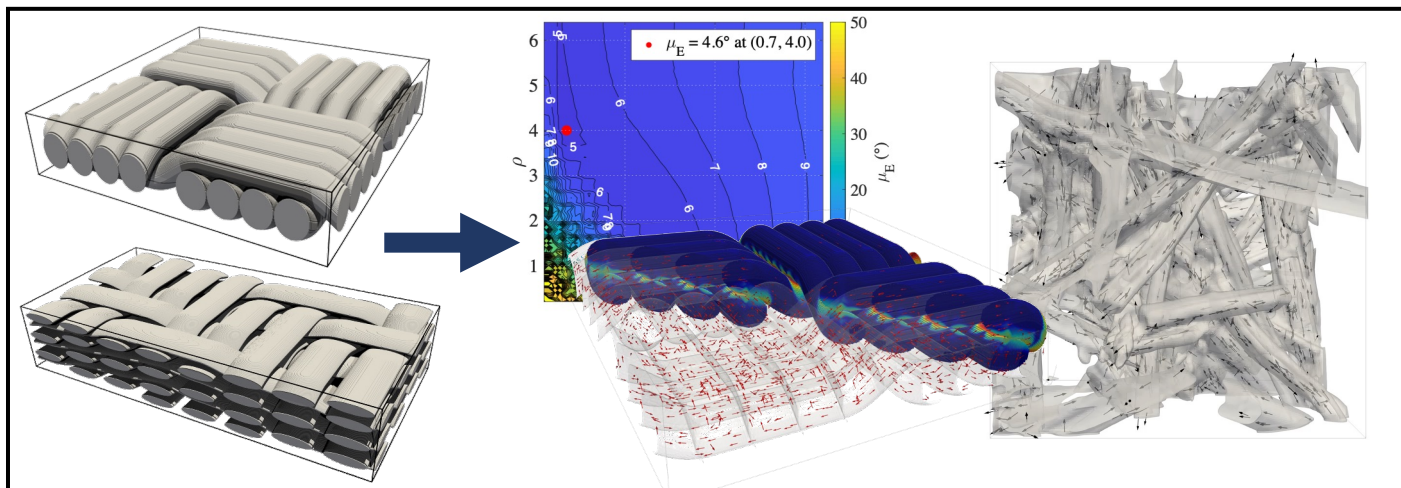
**PuMA is an all-in-one toolbox for microstructure analysis and design**



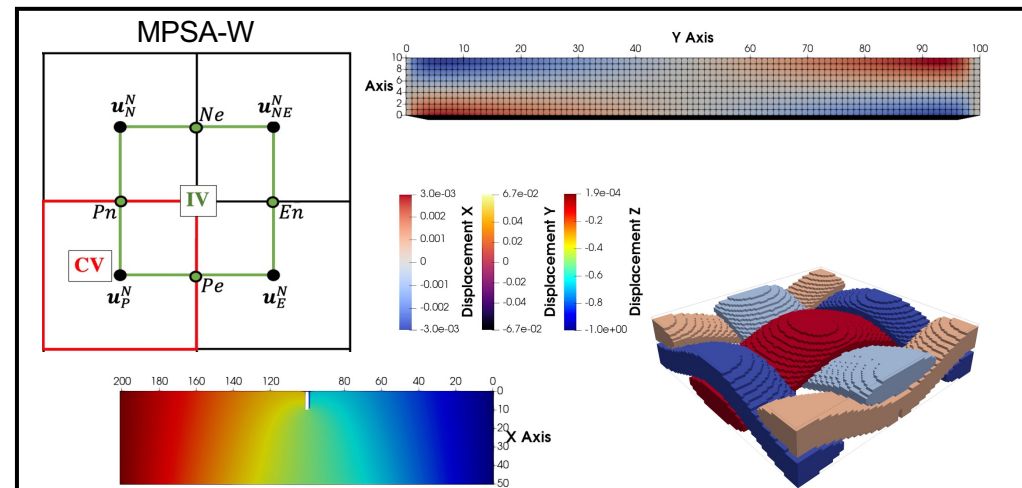
# Overview of Material Properties Computations



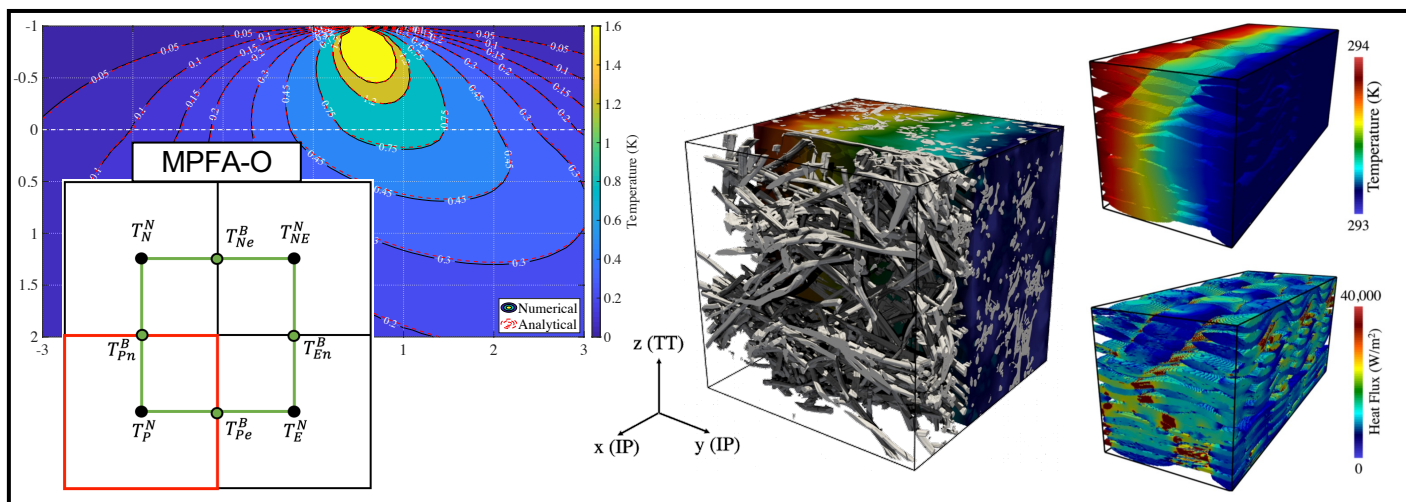
## Part 1: Estimation of local orientation. *Computational Materials Science* (2020)



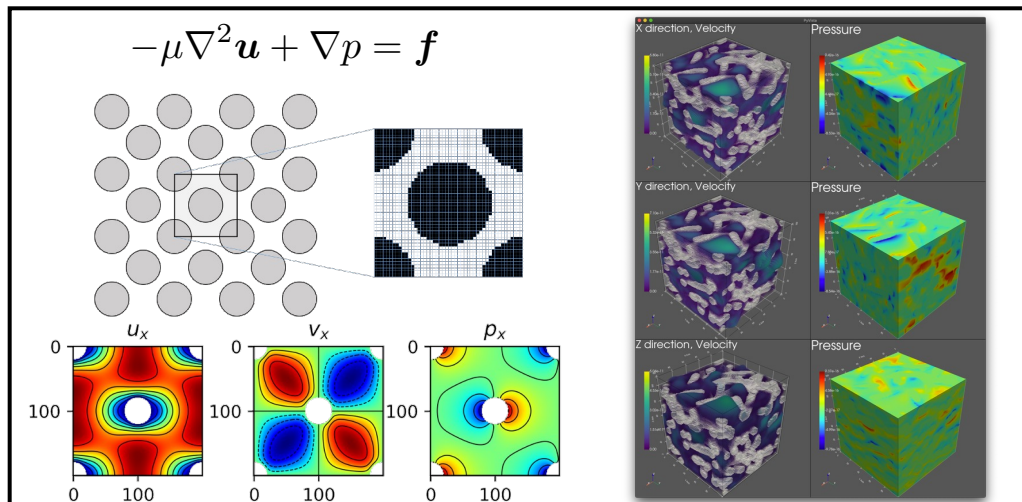
## Computation of effective elasticity

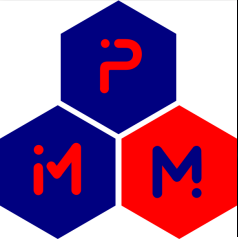


## Part 2: Computation of effective conductivity. *Computational Materials Science* (2021)

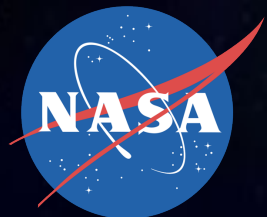


## Computation of effective permeability





# PATO



## OpenFOAM

|                 |                   |
|-----------------|-------------------|
| Finite Volume   | PETSc             |
| I/O management  | Numerical schemes |
| Massive MPI     | Fluid solvers     |
| Moving geometry | Chemistry         |
| Basic mesh gen. | Thermo/Transp.    |

## PATO: material response

*PATOx executable*

*libPATOx library*

*Equilibrium chemistry*

*Finite-rate chemistry*

*Volume Ablation*

*Pyrolysis*

*Pure conduction*

*1D/2D/3D mapping*

*Multi-material*

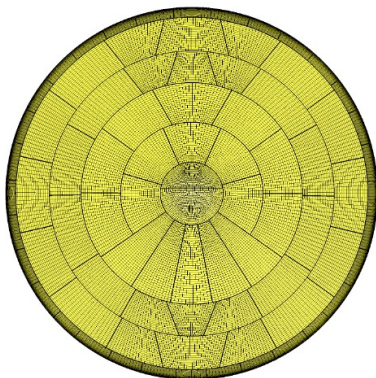
*Fluid coupling*

Complex mesh  
generation

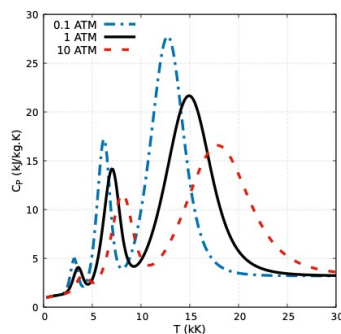
Thermo/Transport/  
Chemistry

Optimization/  
UQ/SA/PE

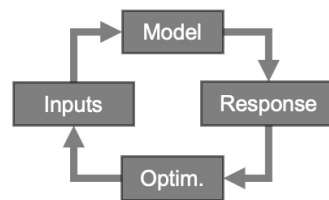
Post-processing  
1D/2D/3D



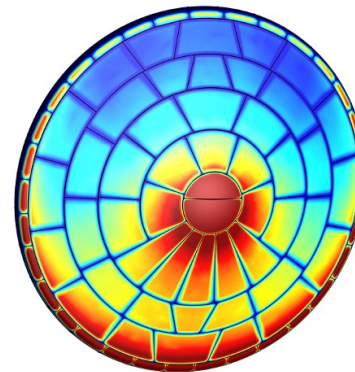
**POINTWISE/GMSH**



**MUTATION++**



**DAKOTA**



**FV/TEC**



## More info on PATO

### Open-source installation

```
conda config --add channels conda-forge
conda config --add channels pato.devel
conda config --set channel_priority strict
conda create -y --name pato pato
```

### PATO website

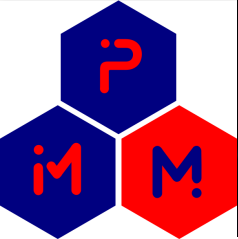
<http://pato.ac>

### Documentation

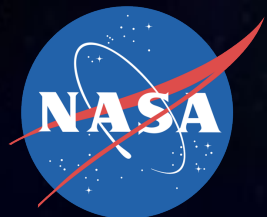
<https://pato.ac/index.php/documentation>

### Main developer

[jeremie.b.meurisse@nasa.gov](mailto:jeremie.b.meurisse@nasa.gov)

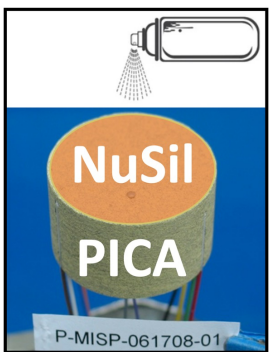


# PICA-NuSil Material Response

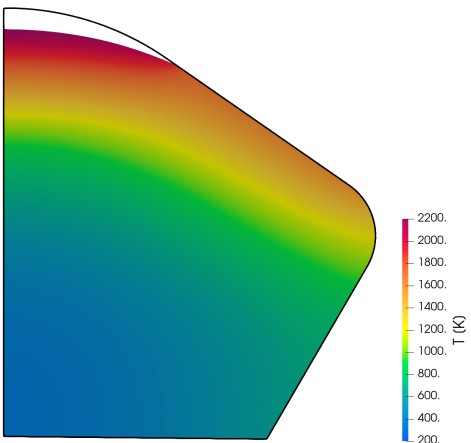


## NuSIL CV-1144-0

Controlled volatility RTV silicone oxygen protective overcoat  
[https://www.avantorsciences.com/assetsvc/asset/en\\_US/id/29017290/contents](https://www.avantorsciences.com/assetsvc/asset/en_US/id/29017290/contents)



**NuSil**, a **silicone-based overcoat**, was sprayed onto the **MSL** and **Mars 2020** heatshields including their in-depth temperature instruments (**MISP**) to mitigate the spread of phenolic dust from **PICA**. A novel material response model for **PICA-NuSil** was implemented in **PATO** using results from **dedicated experiments**.



Temperature and shape change at 30 sec



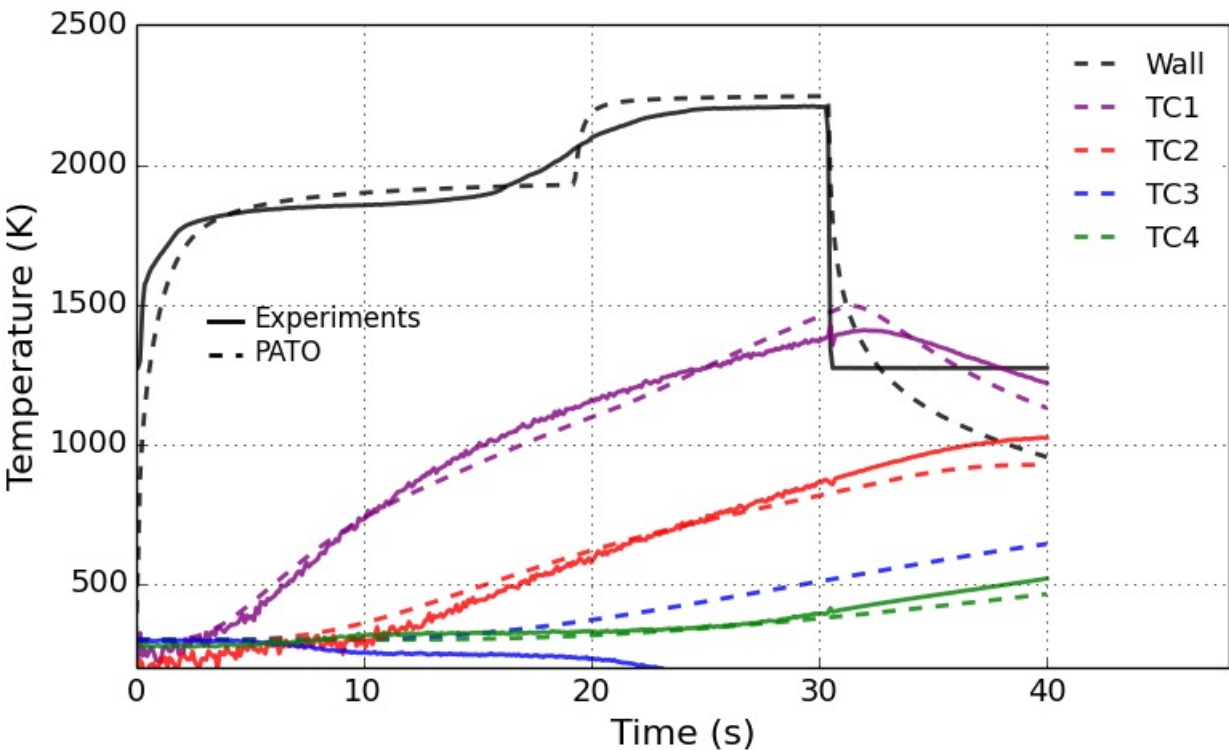
HYMETS PS02 at 28 sec

B. Bessire *et. al.* 2019

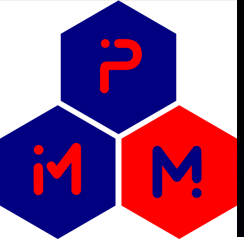
## HyMETS Test Campaign from March 2019

B. Bessire *et. al.* 2019

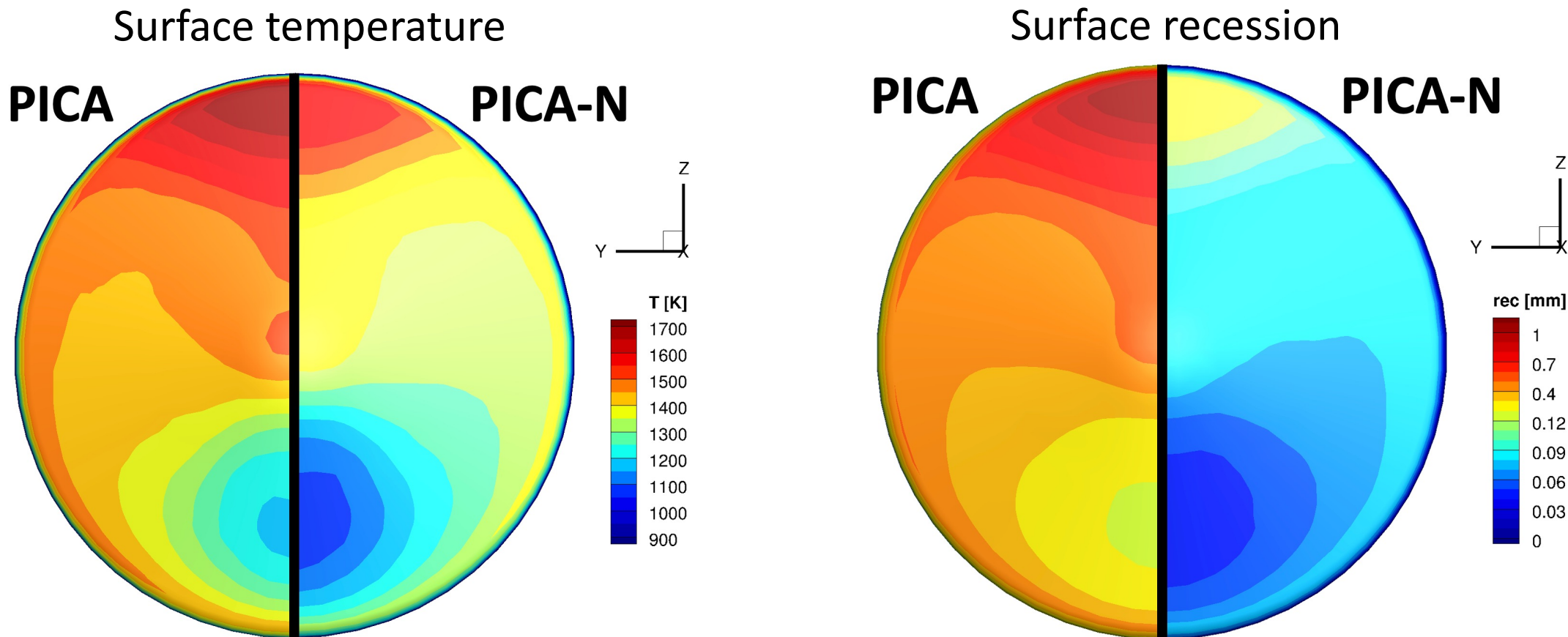
| Material   | Model | Atmosphere | Heat flux [W/cm <sup>2</sup> ] | Pressure [kPa] |
|------------|-------|------------|--------------------------------|----------------|
| PICA-NuSil | PS02  | Air        | 140                            | 5.6            |



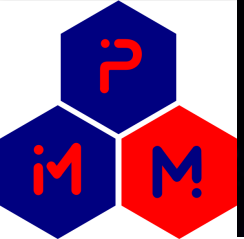
Evolution in time of the temperature



# Comparison of PICA and PICA-N Models



**3D material response of the MSL heatshield using PICA and PICA-NuSil as silica at 80 sec after EI for a fully turbulent environment computed in DPLR. The estimation of the NuSil layer thickness on top of PICA was  $200\ \mu\text{m}$ . The PICA-N model gave lower surface temperature and recession results than the PICA model. According to the PICA-N model, the NuSil coating **still fully covered** the MSL heatshield.**



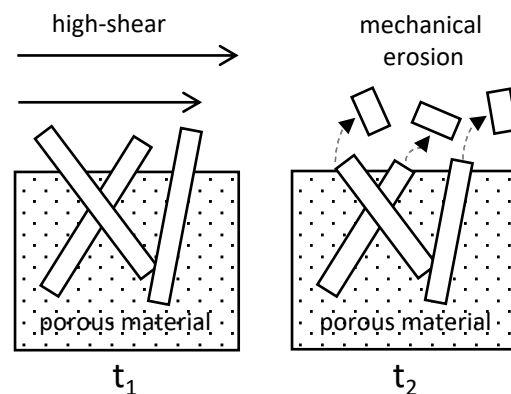
# Mechanical Erosion Model in PATO



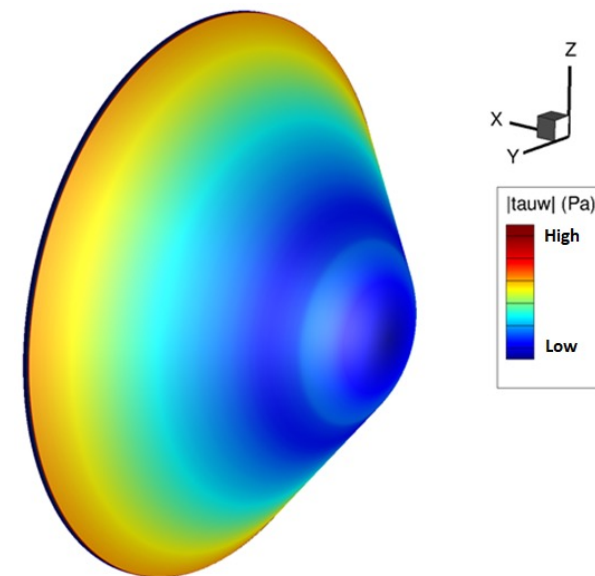
Mechanical erosion may lead to additional mass removal of TPS material during atmospheric entry, increasing the total surface recession.

Main mechanisms identified:

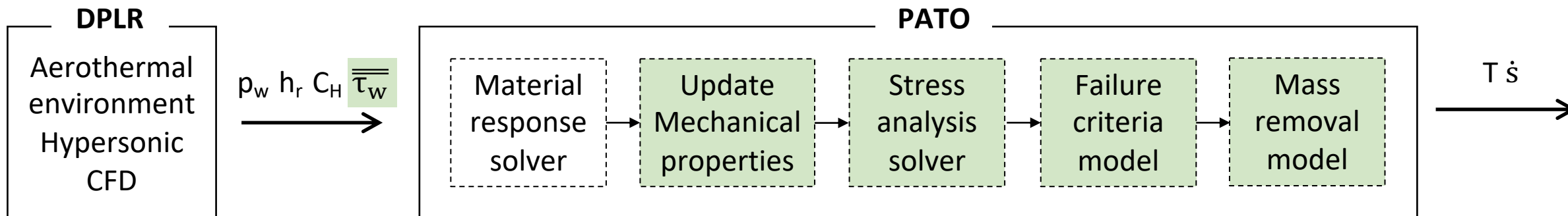
- Shear stress induced by the flow
- Normal stress induced by pyrolysis gas
- Thermal stress induced by the material's temperature field

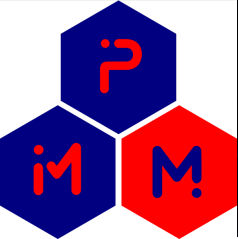


*Schematic of the shear induced mass removal*

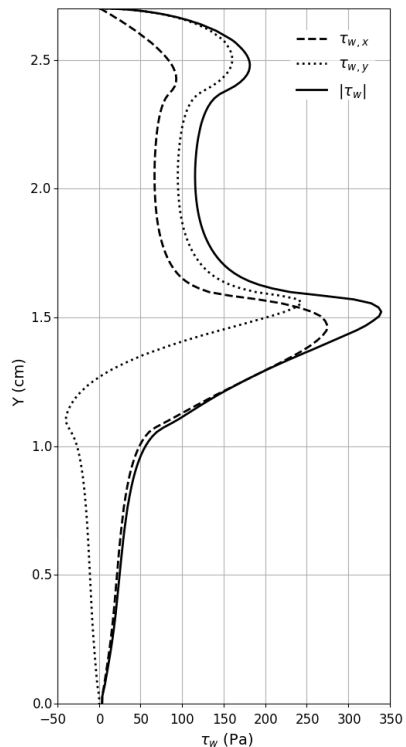


*Shear stress for MSR-EEV entry*

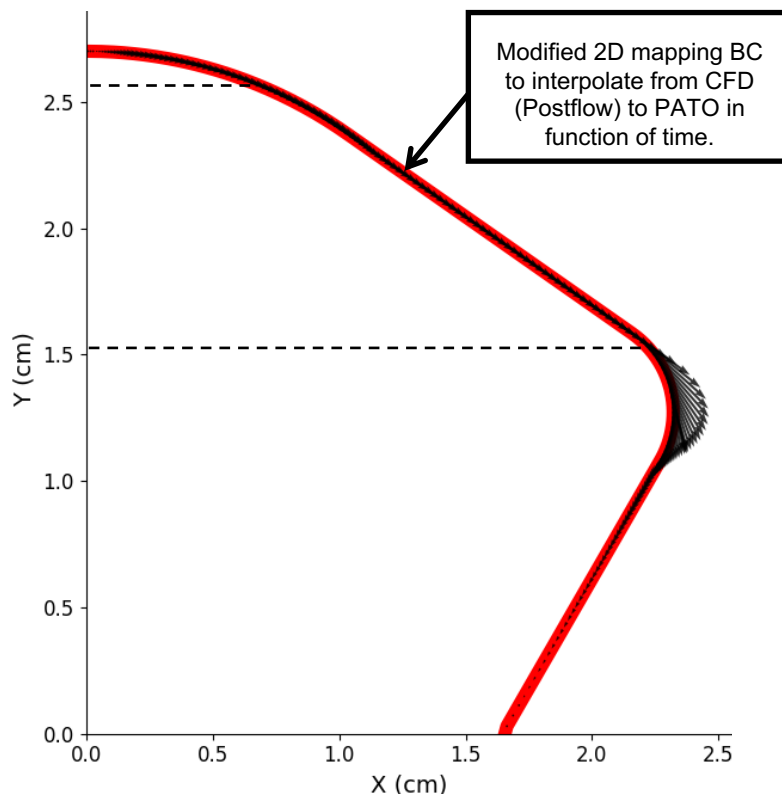




# Mechanical Erosion Model in PATO



Components of the wall shear stress in function of the Y axis



Direction and magnitude of the wall shear stress

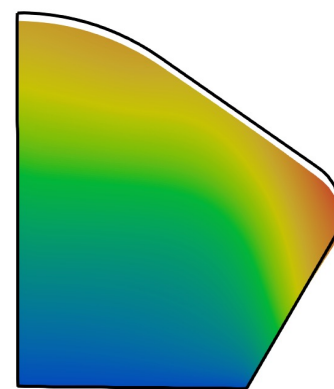
## Theoretical failure criteria

$$\max \left( \frac{\sigma_{xx}}{F_{tu,xx}}, \left| \frac{\sigma_{xx}}{F_{cu,xx}} \right|, \frac{\sigma_{yy}}{F_{tu,yy}}, \left| \frac{\sigma_{yy}}{F_{cu,yy}} \right|, \left| \frac{\sigma_{xy}}{F_{su,xy}} \right| \right) > 1$$

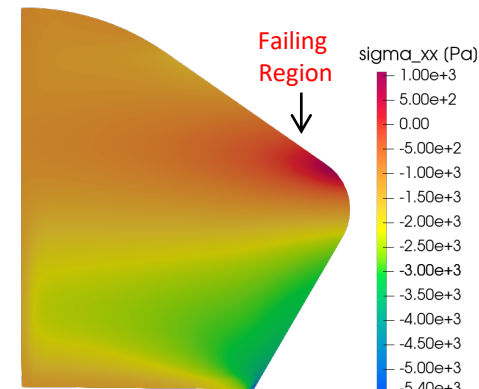
The recession rate and the mass loss are computed as follows:

$$M_{\text{loss}} = \rho A \dot{s} \Delta t \quad \dot{s} = \frac{L}{\Delta t}$$

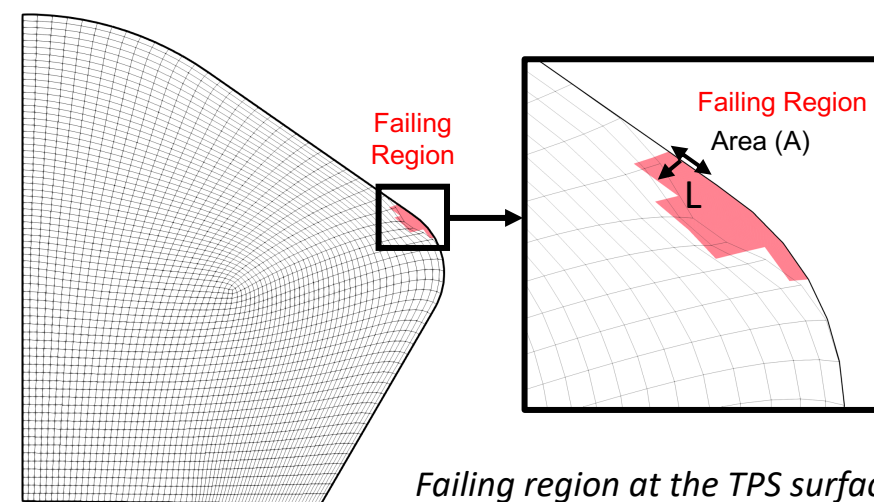
**Ultimate Strength**  
 tu: Traction  
 cu: Compression  
 su: Shear



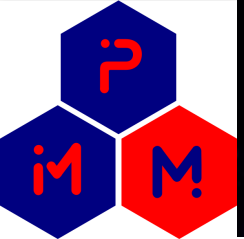
Displacement due to shear stress with a scale factor ( $\times 10^5$ )



Component X of the stress tensor and theoretical failure criteria



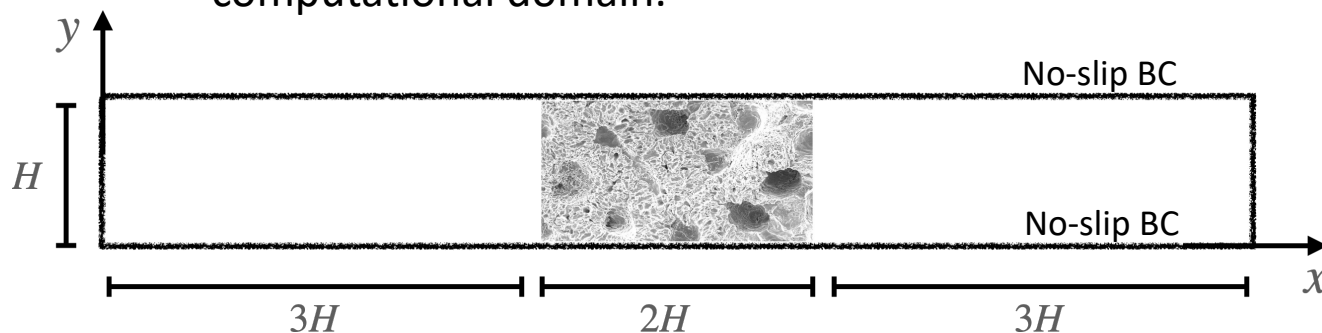
Failing region at the TPS surface



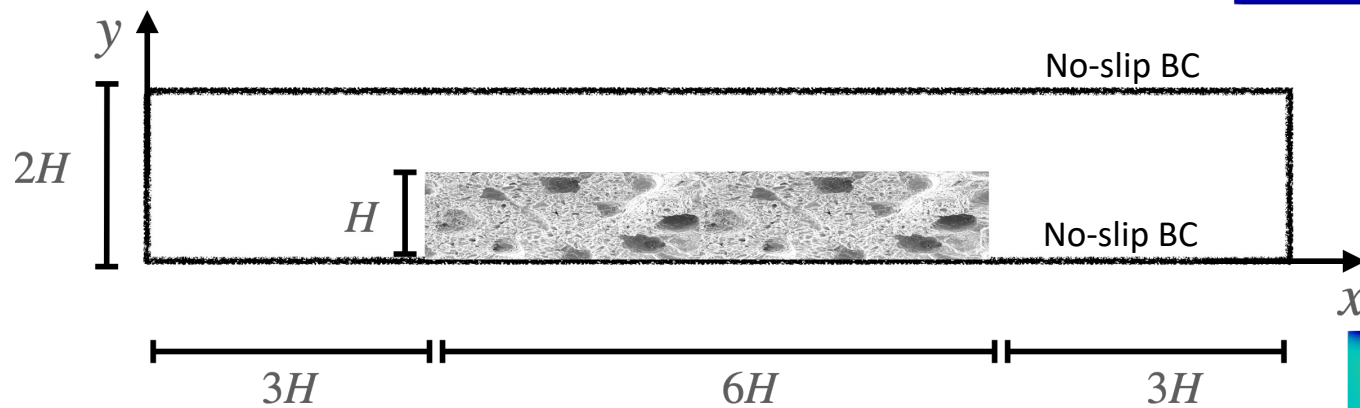
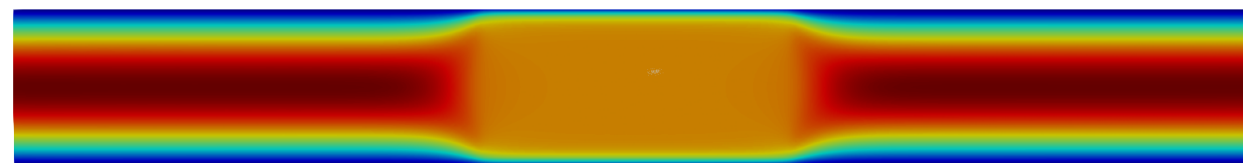
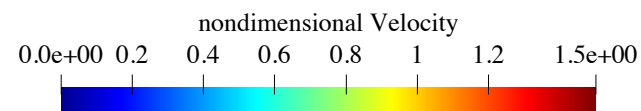
# On-going Effort on the Universal Solver Development



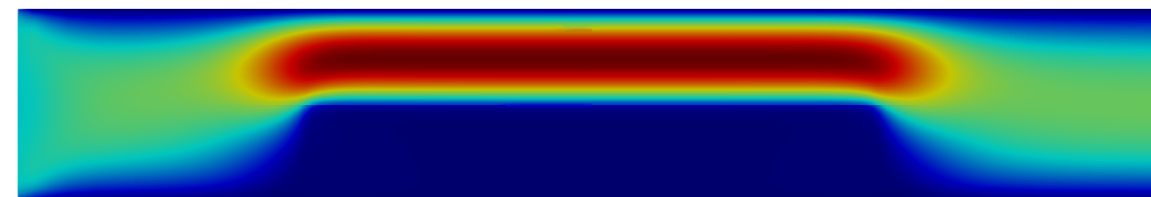
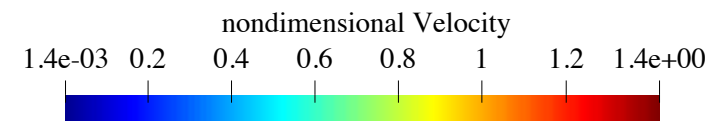
- The governing equations for the pure fluid and porous medium are coupled and solved in the same computational domain.

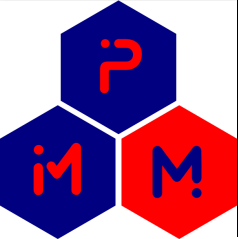


*Porous Plug Flow from Betchen*



*Beavers & Joseph problem*



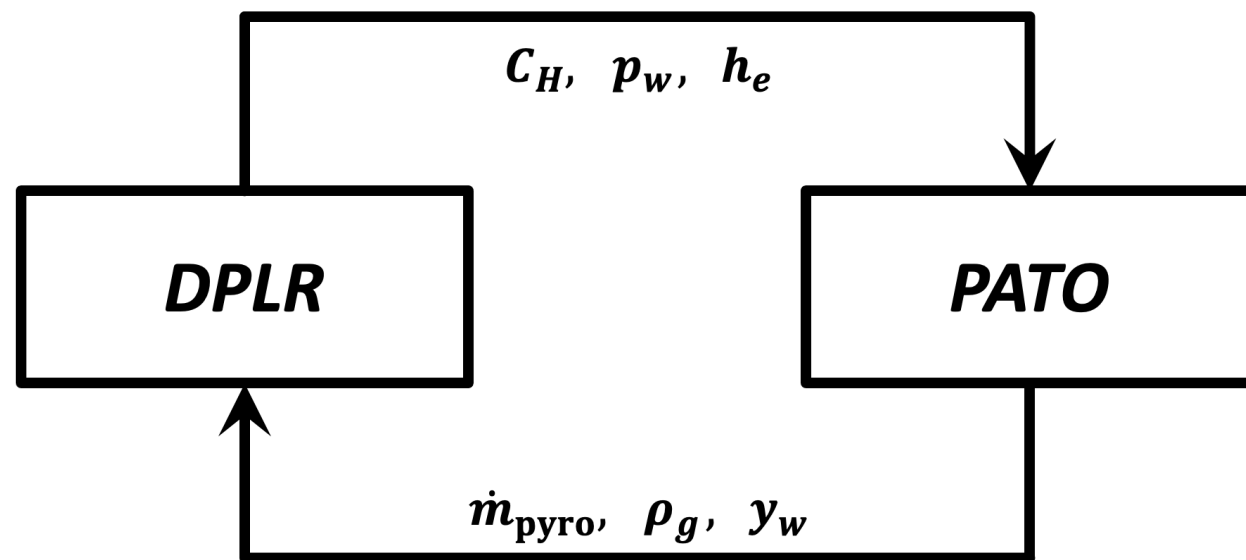


# Loose Coupling Between Environment and Material Response

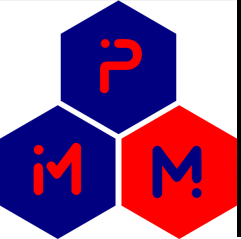


Coupling via pyrolysis gas blowing at the heatshield surface:

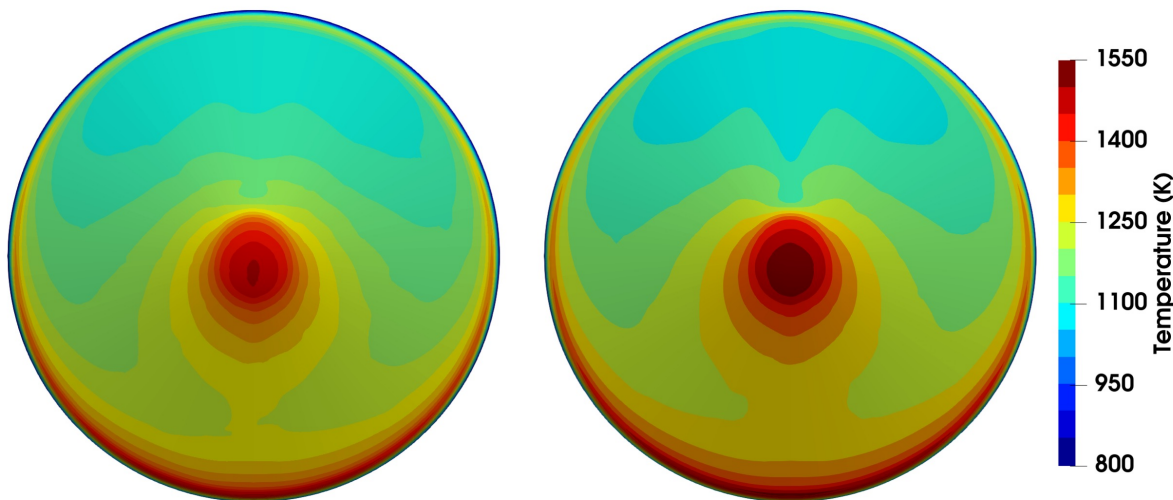
- Blowing gases computed in PATO and given to DPLR for use in a blowing boundary condition.
- Pressure, heat transfer coefficient and boundary layer edge enthalpy computed in DPLR and given to PATO.



**Objective is to utilize this coupling in computing MSL and Mars 2020 material responses**

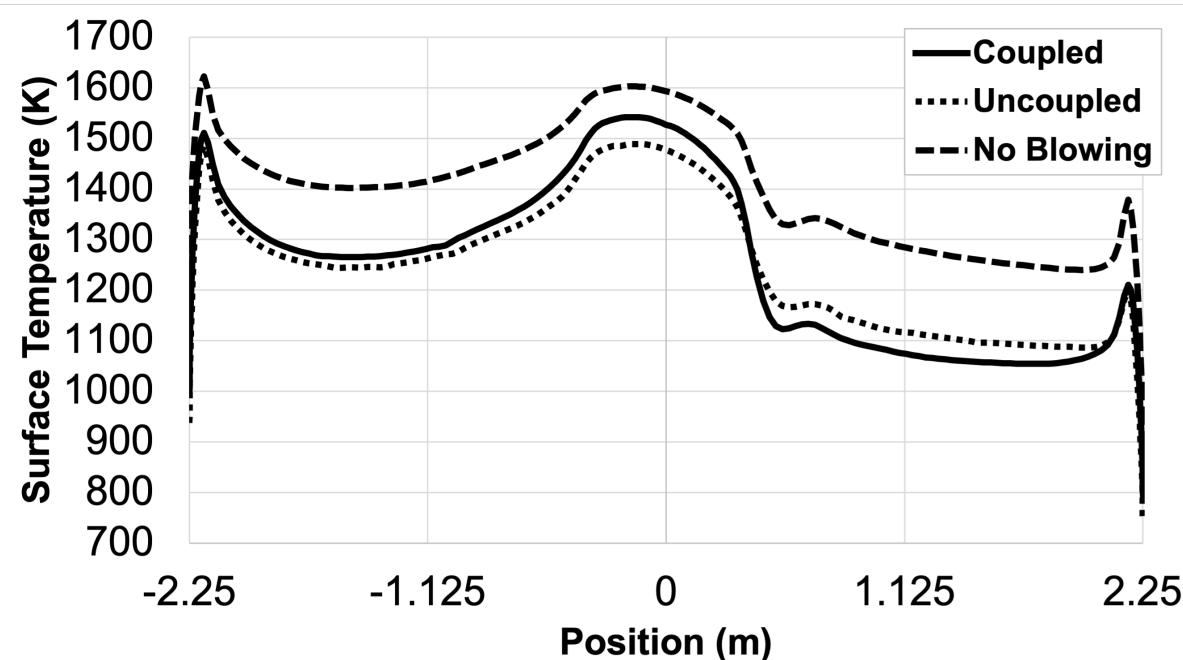
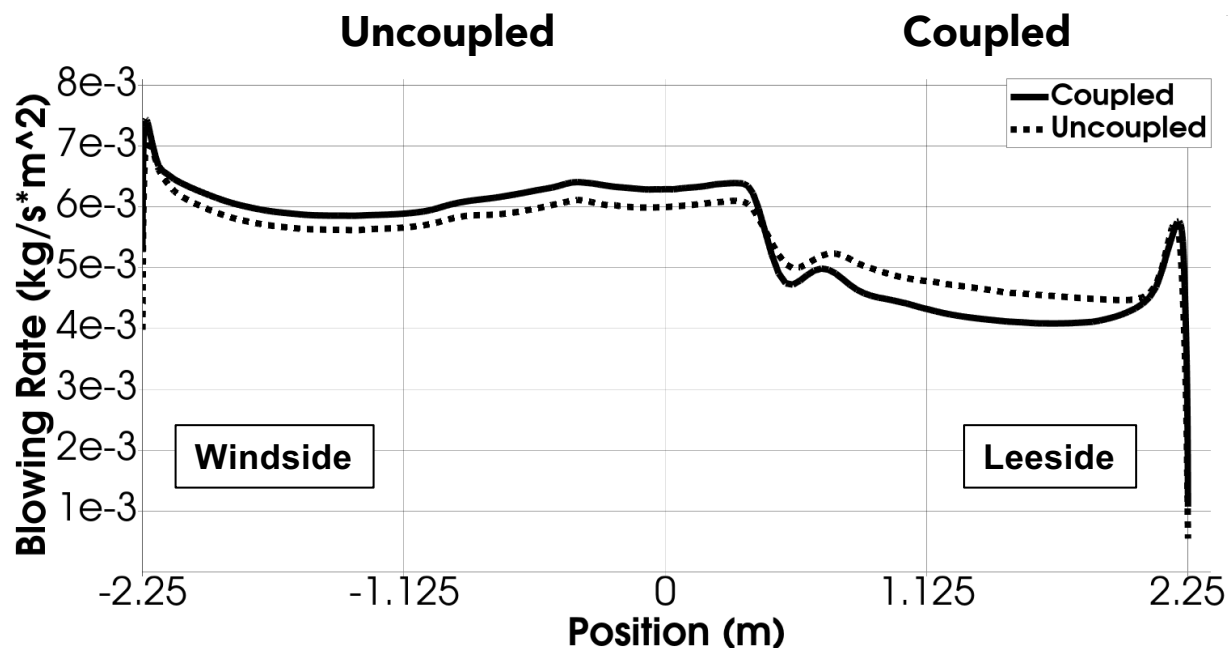


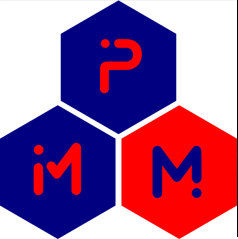
# Heatshield Surface at 65s



Contour plots of surface temperature for the coupled and uncoupled simulations at 65s

(Below) Centerline plots of pyrolysis mass flux at the surface (left) and temperature (right). The surface temperature without blowing was computed directly by DPLR.





# 3D Unsteady Simulations of Arc Heater Plasma Flow

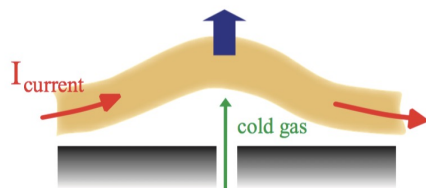


## ARChES = ARC Heater Simulator

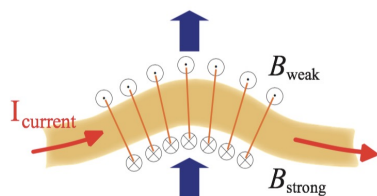
3D unsteady plasma flow analysis tool  
Coupled MHD and radiation models using OpenFOAM

Main developer

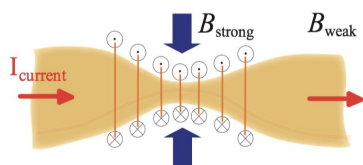
[jeremie.b.meurisse@nasa.gov](mailto:jeremie.b.meurisse@nasa.gov)



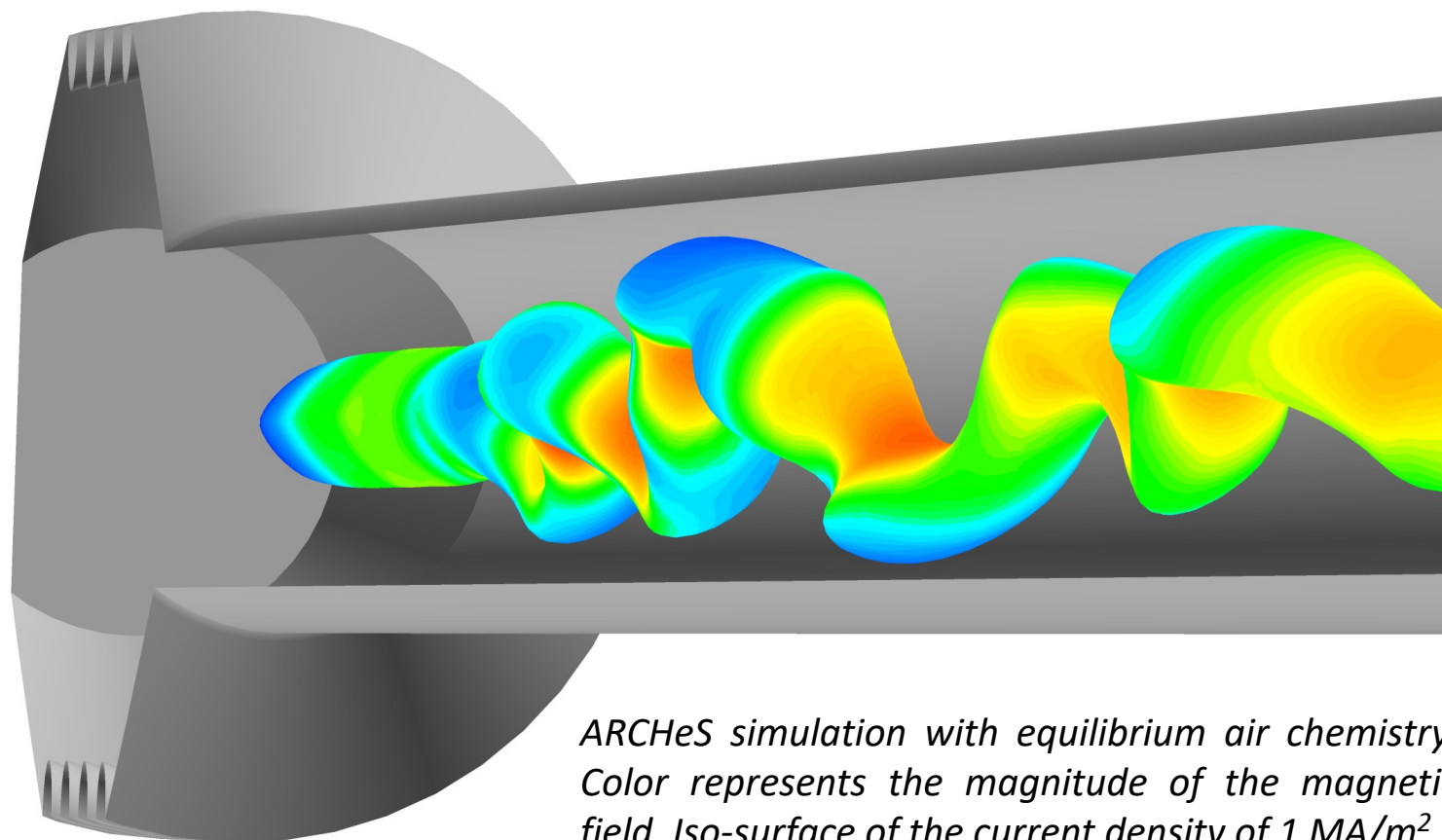
Thermal instability



Kink instability

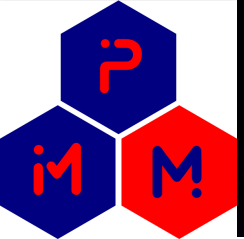


Sausage instability



ARChES simulation with equilibrium air chemistry.  
Color represents the magnitude of the magnetic field. Iso-surface of the current density of  $1 \text{ MA/m}^2$ .

**Stable arc next to the electrode chambers  
Instabilities arise in the constrictor**



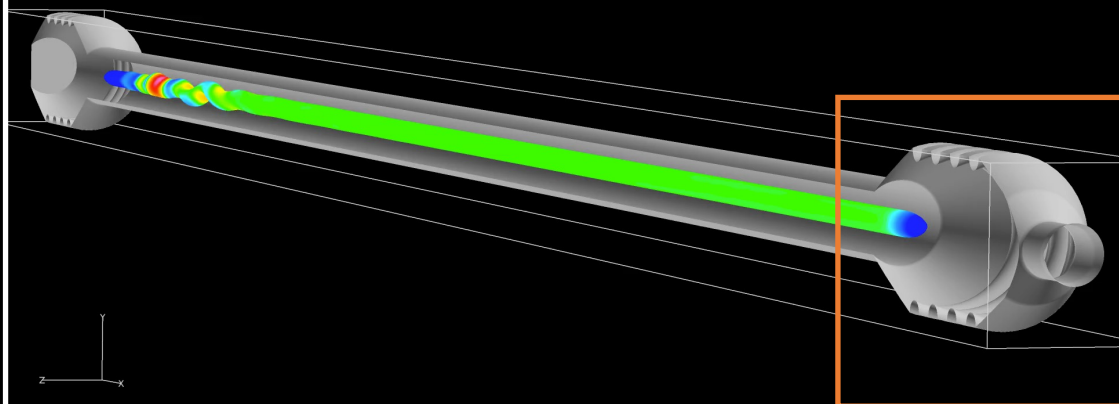
# 3D Unsteady Simulations of Arc Heater Plasma Flow



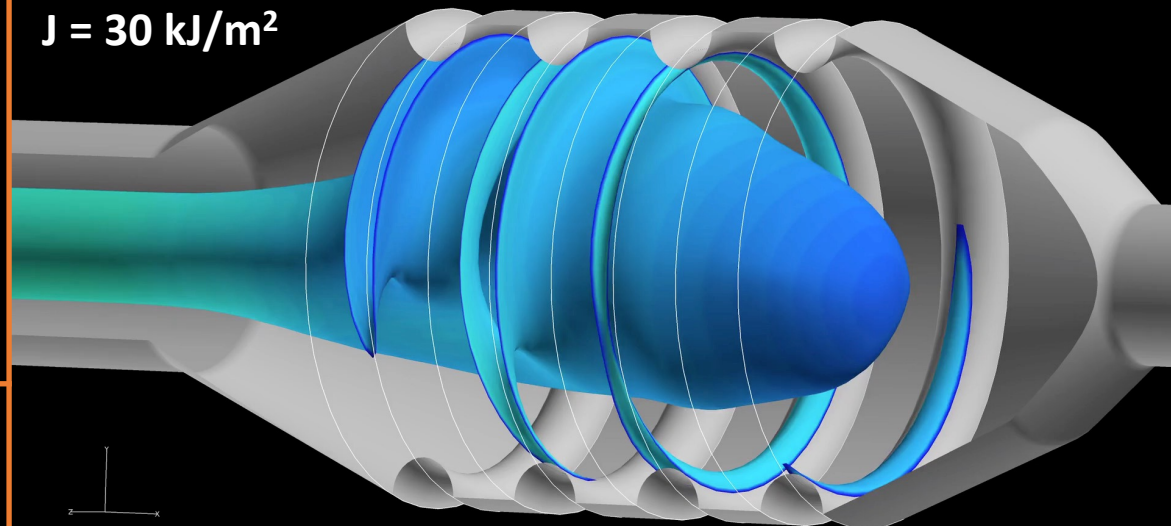
Arc heater simulated with ARChES (MHD + radiation)



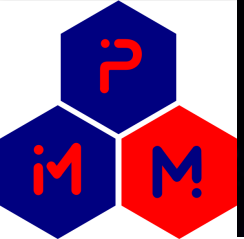
$J = 1 \text{ MJ/m}^2$



$J = 30 \text{ kJ/m}^2$



The location and frequency of the arc attachment at the cathodes are found to be mainly driven by the upstream arc instabilities

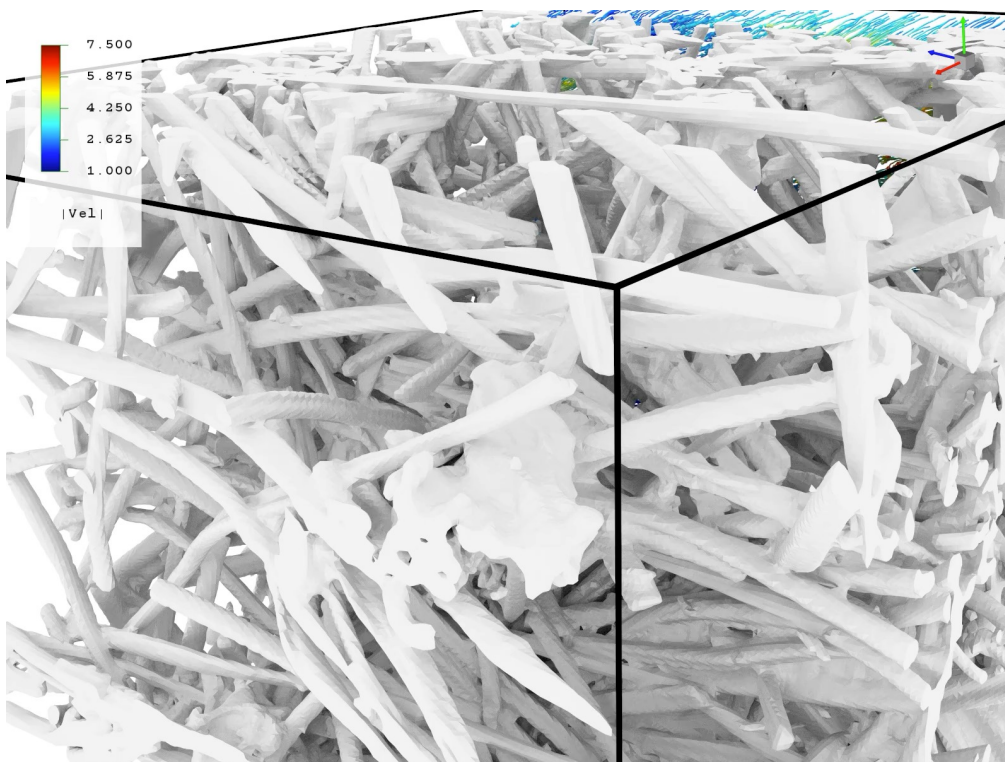


# SPARTA: an Exascale DSMC Code

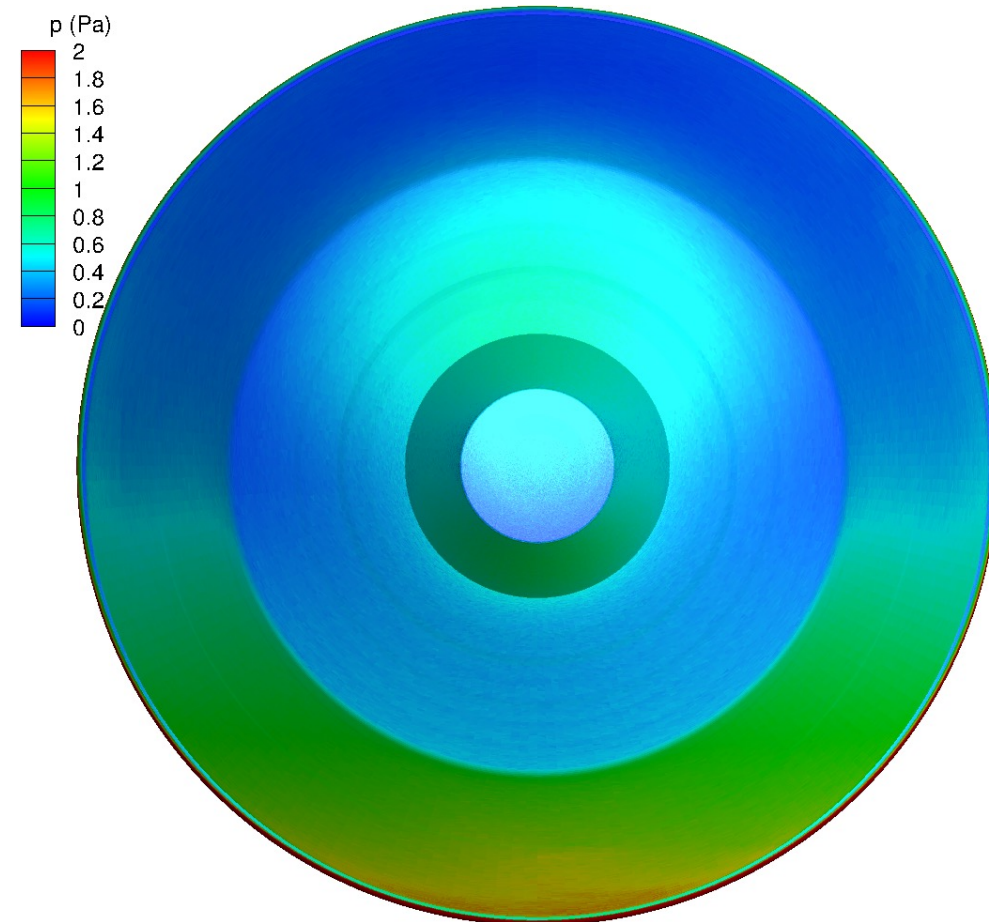


## SPARTA = Stochastic **PAR**allel Rarefied-gas Time-accurate **AN**alyzer

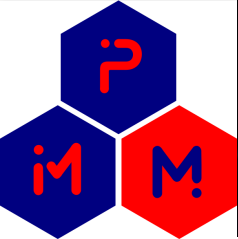
- Open-source DSMC solver initiated at SNL, currently co-developed by Sandia and NASA with multiple users internationally and domestically: <https://sparta.sandia.gov/>
- Can be run from single processor to petascale.
- Has been used to run the largest (we think!) DSMC models up-to-date, with up to **100 billion particles** and **billions of grid cells**. Can also handle **billions of surface elements**



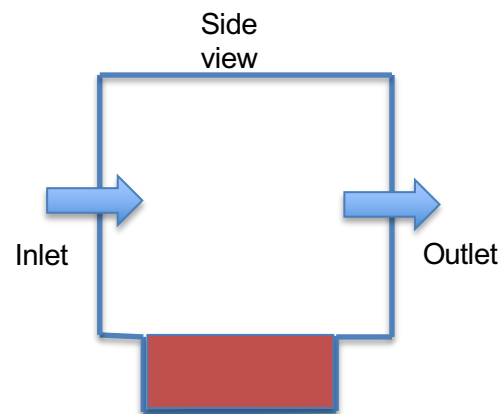
*Flow through a  
FiberForm sample*



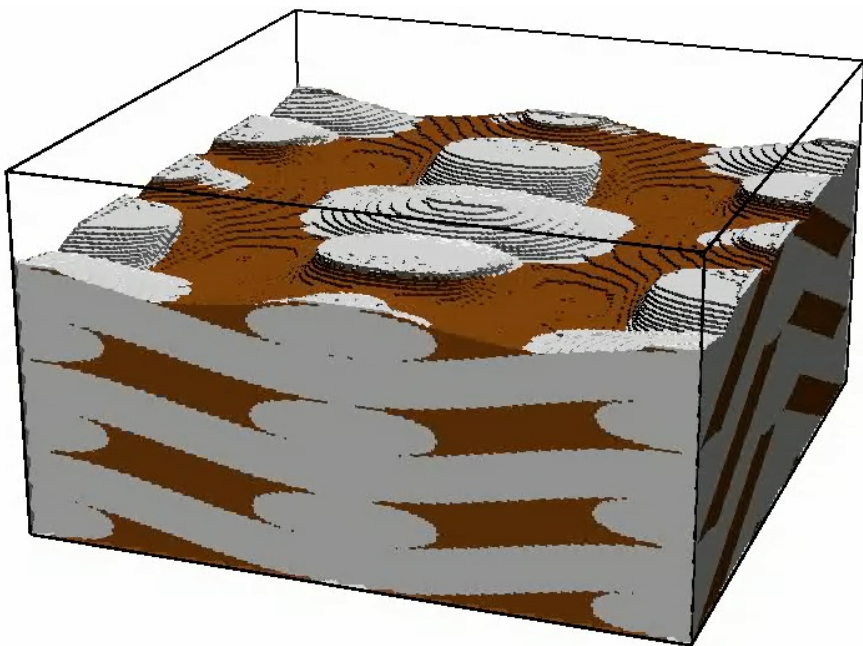
*Mars2020 backshell pressure at  $t = 39s$  after EI*



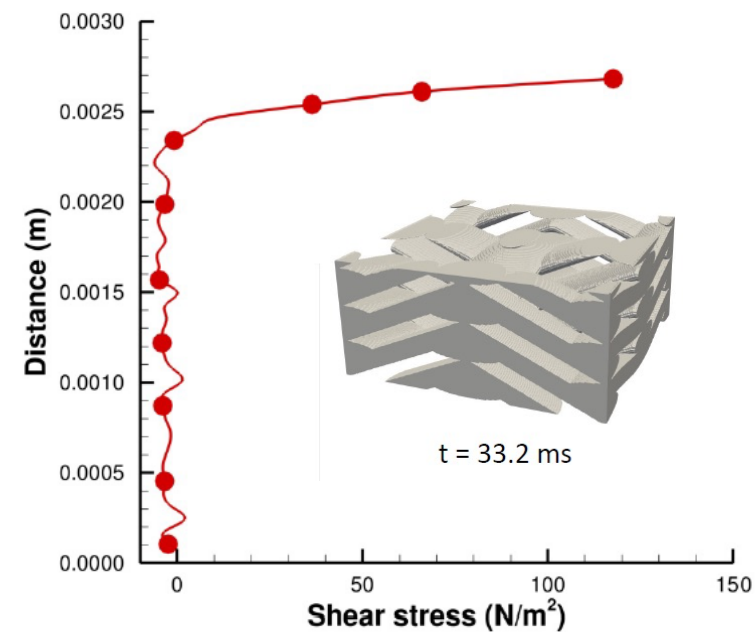
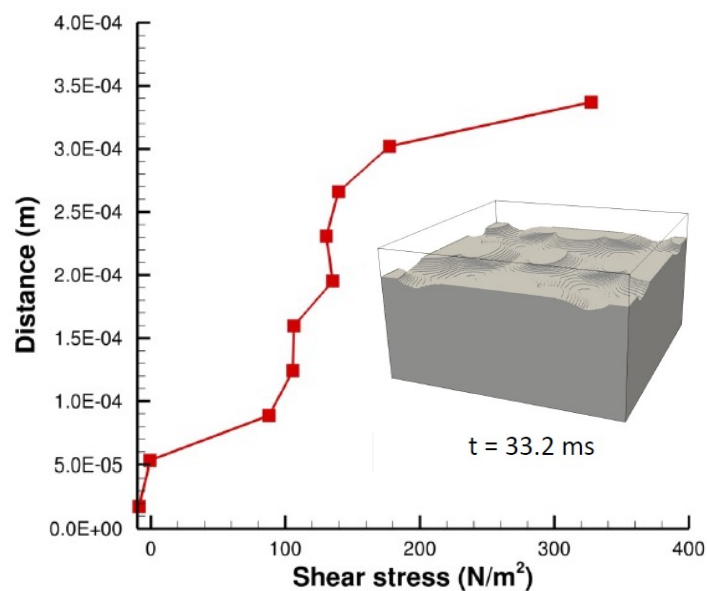
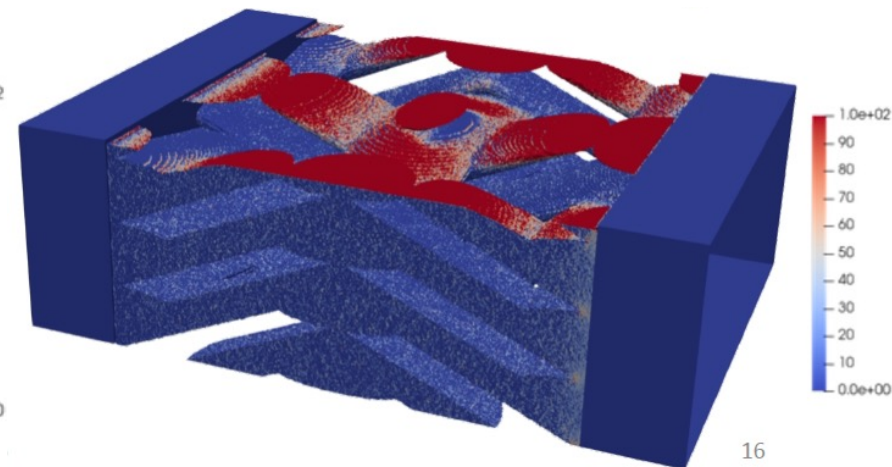
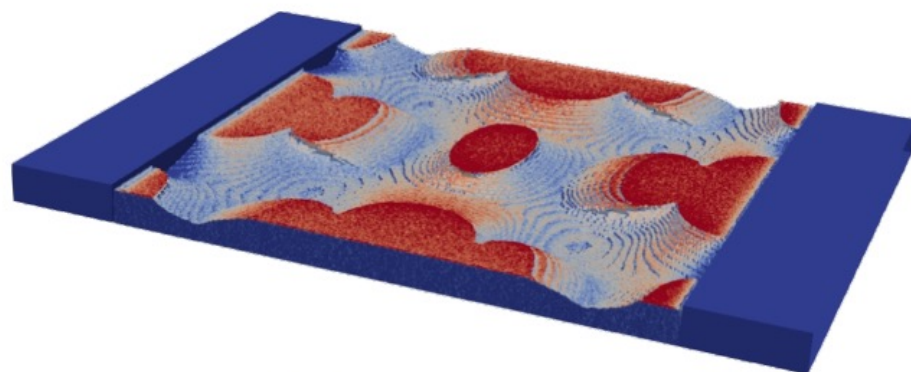
# Computation of Shear Stress on Woven TPS

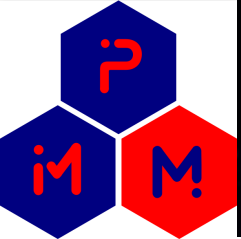


Woven TPS

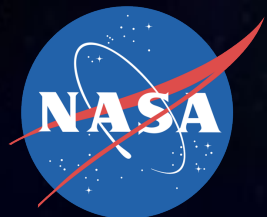


Shear stress ( $\text{N/m}^2$ )

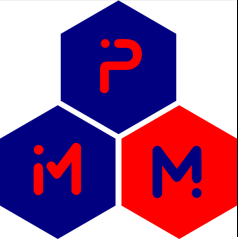




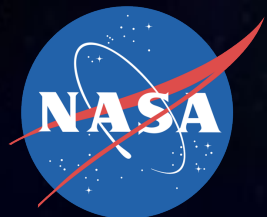
Questions?



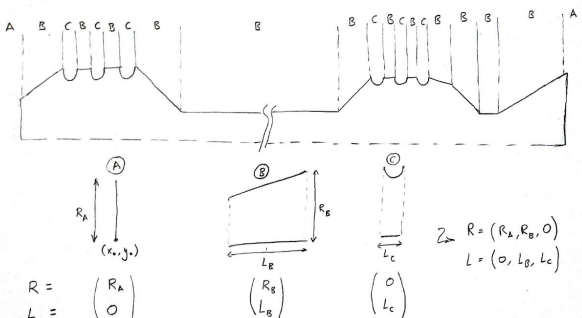
Thank you!



# ARChES



## Geometry concept



## OpenFOAM

Finite Volume

I/O management

Massive MPI

Moving geometry

Basic mesh gen.

Mixture tables

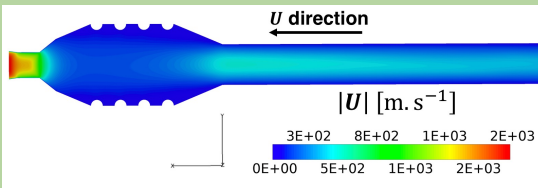
Chemistry

Thermo/Transp.

Turbulence

Compressible flow

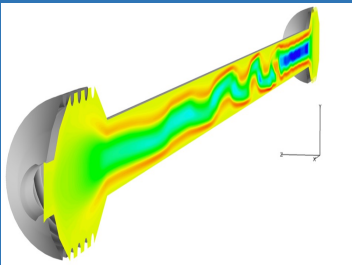
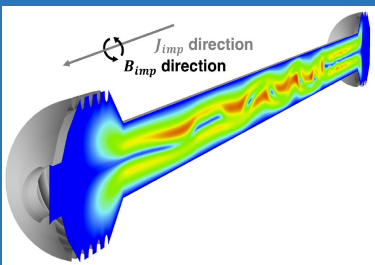
Radiation



## ARChES

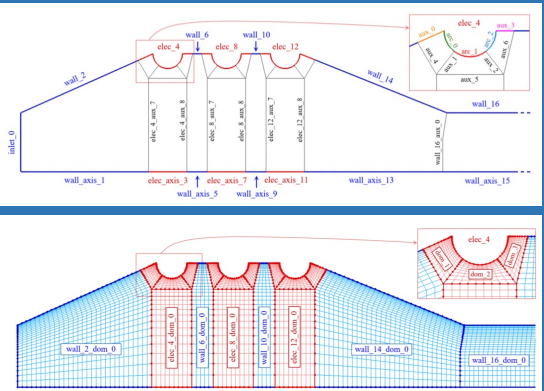
MHD + ballast BC

3D radiative transfer

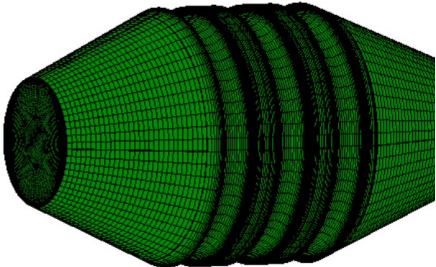


## Lynx

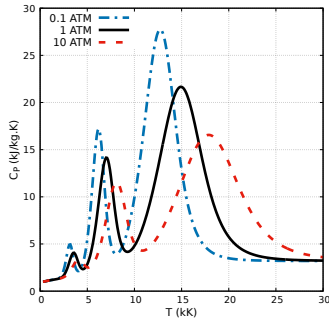
Automatic mesh inputs



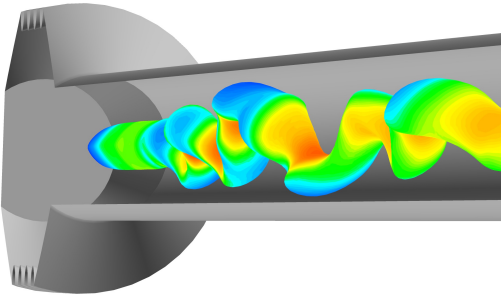
Complex mesh generation

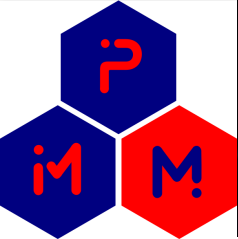


Thermo/Transport/Chemistry

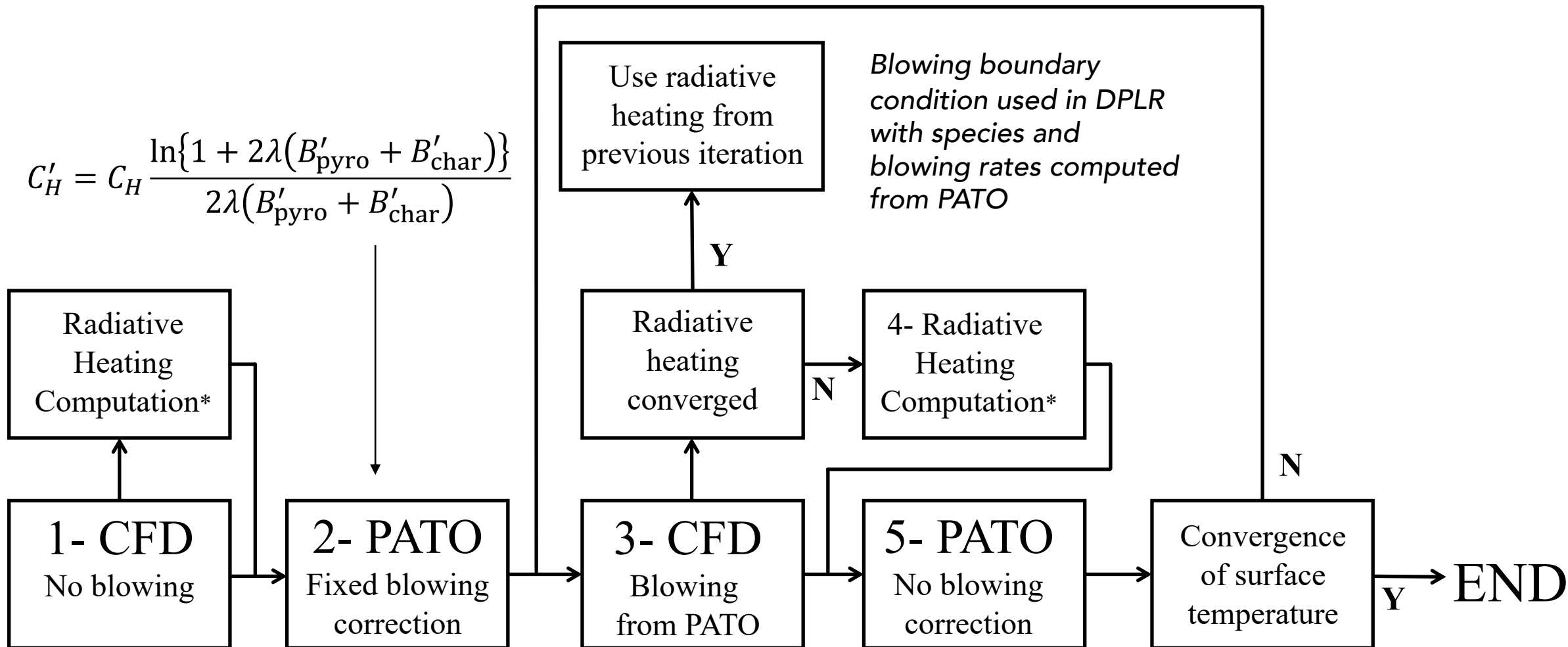


Post-processing 1D/2D/3D





# Coupling Methodology



\*Using NEQAIR: E. Whiting et al. (1996) NASA RP-1389.