



*Materials TDT Talk:
An Evolving Perspective on AM Qual and Cert*

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- B.S. Industrial & Systems Engineering, Auburn University, 2013
- M.S. Human Factors in Aeronautics, Florida Institute of Technology, 2017
- US Army Aviation and Missile Center, Redstone Arsenal
 - Co-op, Production Engineering, 2010-2013
- NAVAIR, Patuxent River Naval Air Station
 - Manufacturing & Quality, 2013-2016
 - Platform Stores Integration, Lead Systems Engineer, 2016-2019
- US Army Aviation and Missile Center, Redstone Arsenal
 - Production Engineering Lead for Improved Turbine Engine Program, 2019-2021
- NASA, Marshall Space Flight Center
 - M&P Lab, Additive Manufacturing & Digital Solutions Team, 2021-present
 - AM Qualification and Certification



Overview



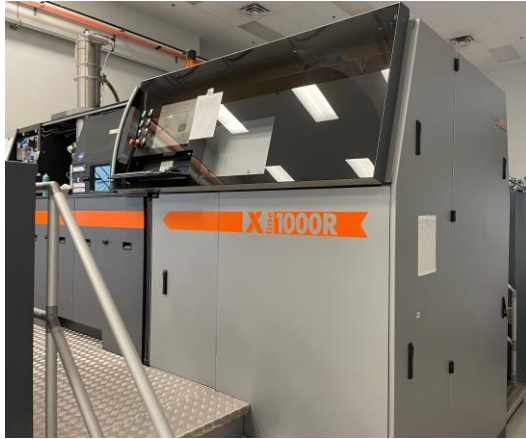
- Capabilities and History of MSFC Additive Lab
- Qualified Material Processes
- Additive Manufacturing Control Plan
- Equipment and Facility Control Plan
- New AM Technology Knowledge Development Project

- The lab has experience with about 30 different systems over the last 30 years
- NASA AM Objectives
 - Decrease production lead time & costs
 - Develop and Maintain Flight Certification Standards
 - Process development and characterization
 - Share knowledge and data in pursuit of smart vendor base & ensure agency remains a smart buyer
 - Design optimized components & test at relevant conditions
 - Appropriate Application
 - High complexity & difficult to manufacture
 - Low production rate
 - Long lead time & high cost





Metallic Additive Manufacturing



Concept Laser Xline 1000R
(Retiring Soon)

In718

Build Volume: 630 x 400 x 500 mm



Image Credit: Velo3D

Velo3D Sapphire
(Summer 2022)

In718

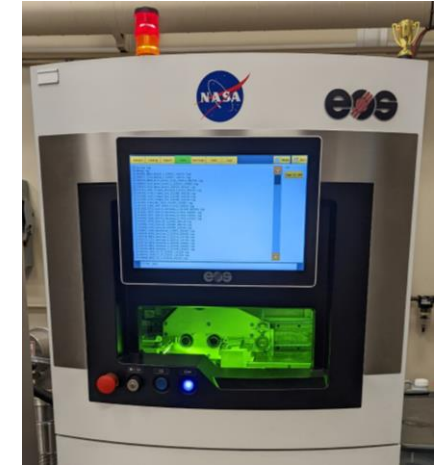
Build Volume: 315mm Ø x 400mm z



EOS M290 (Two Machines)

HR1, JBK75

Build Volume: 250 x 250 x 325 mm



EOS M100

Various Materials

Build Volume: 100 Ø x 95 mm z



Concept Laser Xline 2000R

In718

Build Volume: 400 x 800 x 500 mm



Concept Laser M1

In625/In718

Build Volume: 250 mm³



Concept Laser M2

GRCop-42

Build Volume: 245 x 245 x 350 mm



DM3D Directed Energy Deposition

Various Materials



Polymer Additive Manufacturing



Stereolithography (SLA)

5000 SLA

Somos Watershed 11122 XC

Build Volume: 20 in³

Viper SLA

DMX Somos

Build Volume: 10 in³



Fused Deposition Modeling (FDM)

AON M2+ (Summer 2022)

PEEK, PEKK

Build Volume: 450 x 450 x 565 mm

Stratasys Fortus 900

ABS, Polycarbonate, Ultem 9085

Build Volume: 914 x 609 x 914 mm

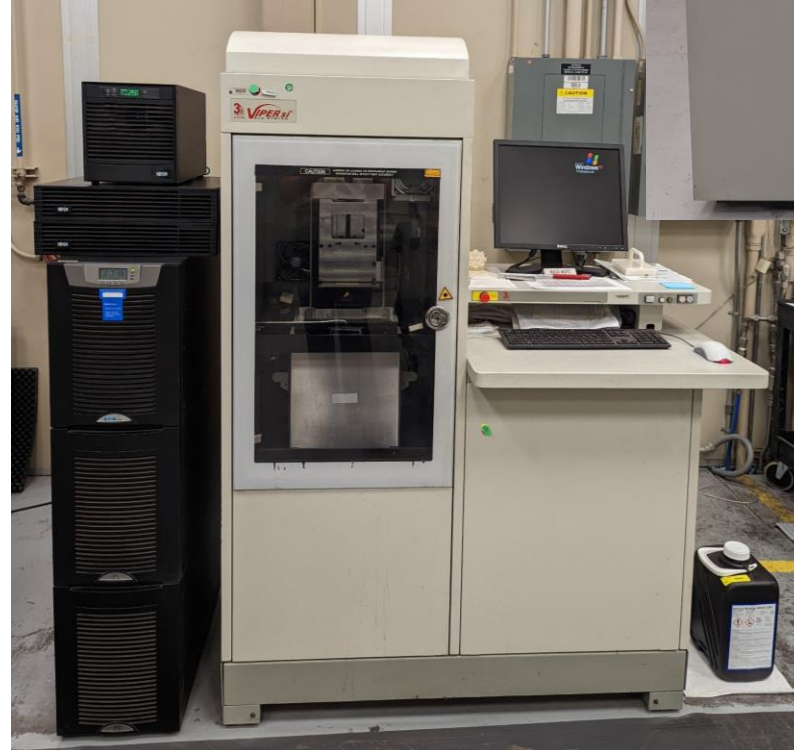
Markforged Mark Two

Carbon Fiber, Glass Fiber, Kevlar

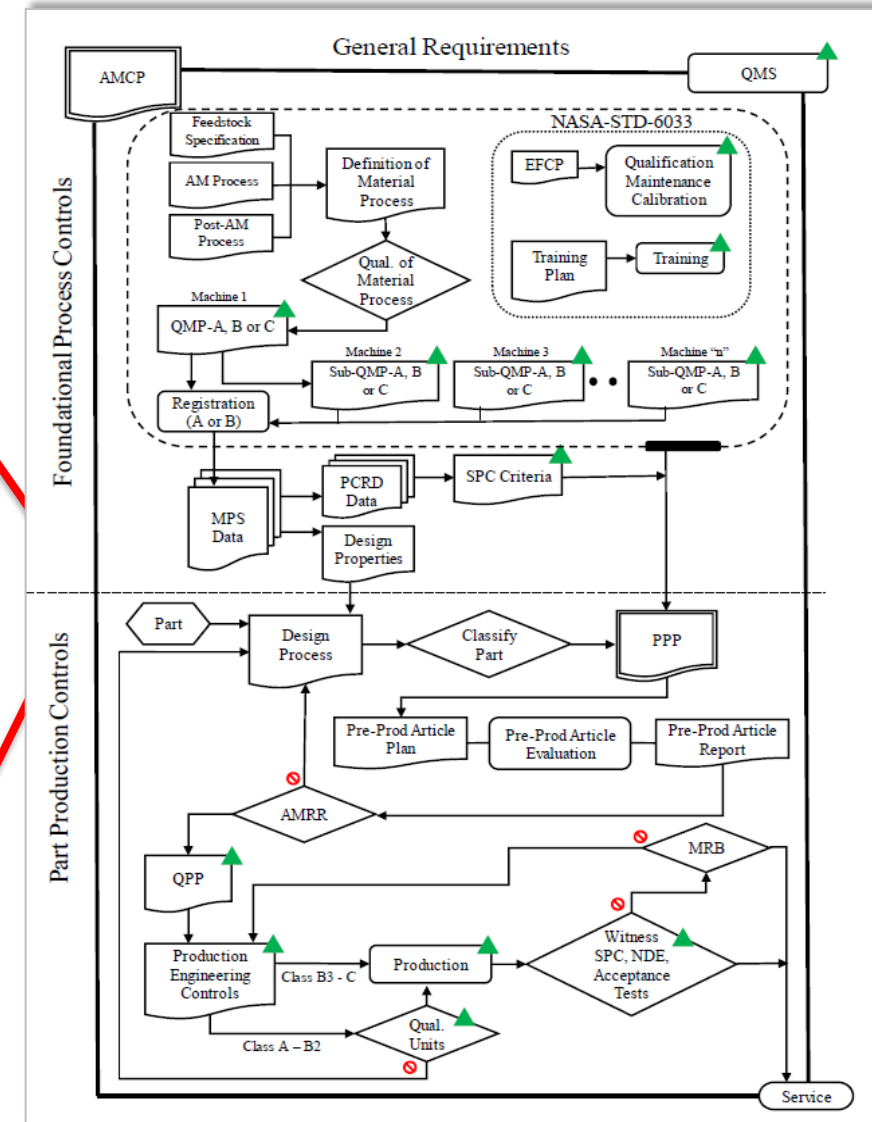
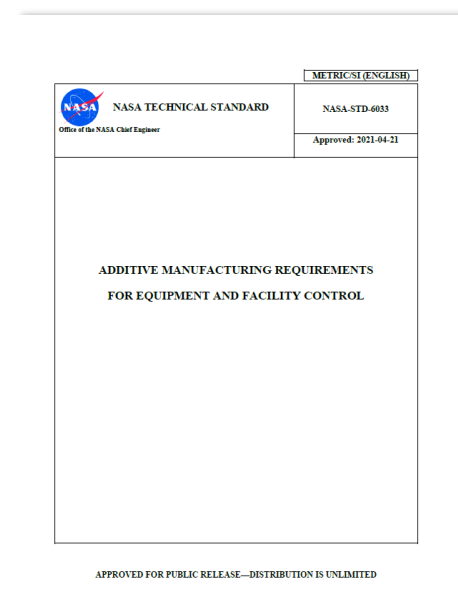
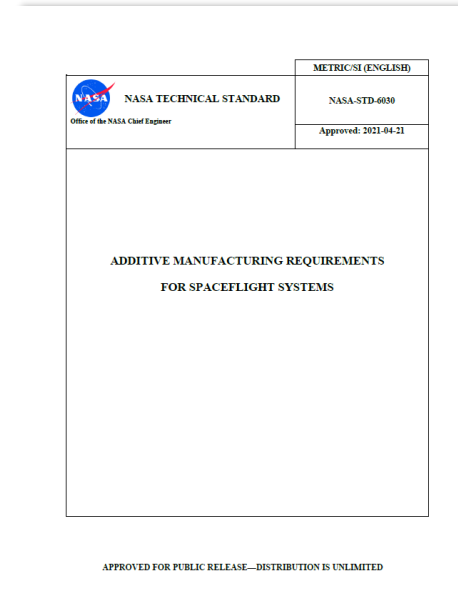
Build Volume: 320 x 132 x 154 mm



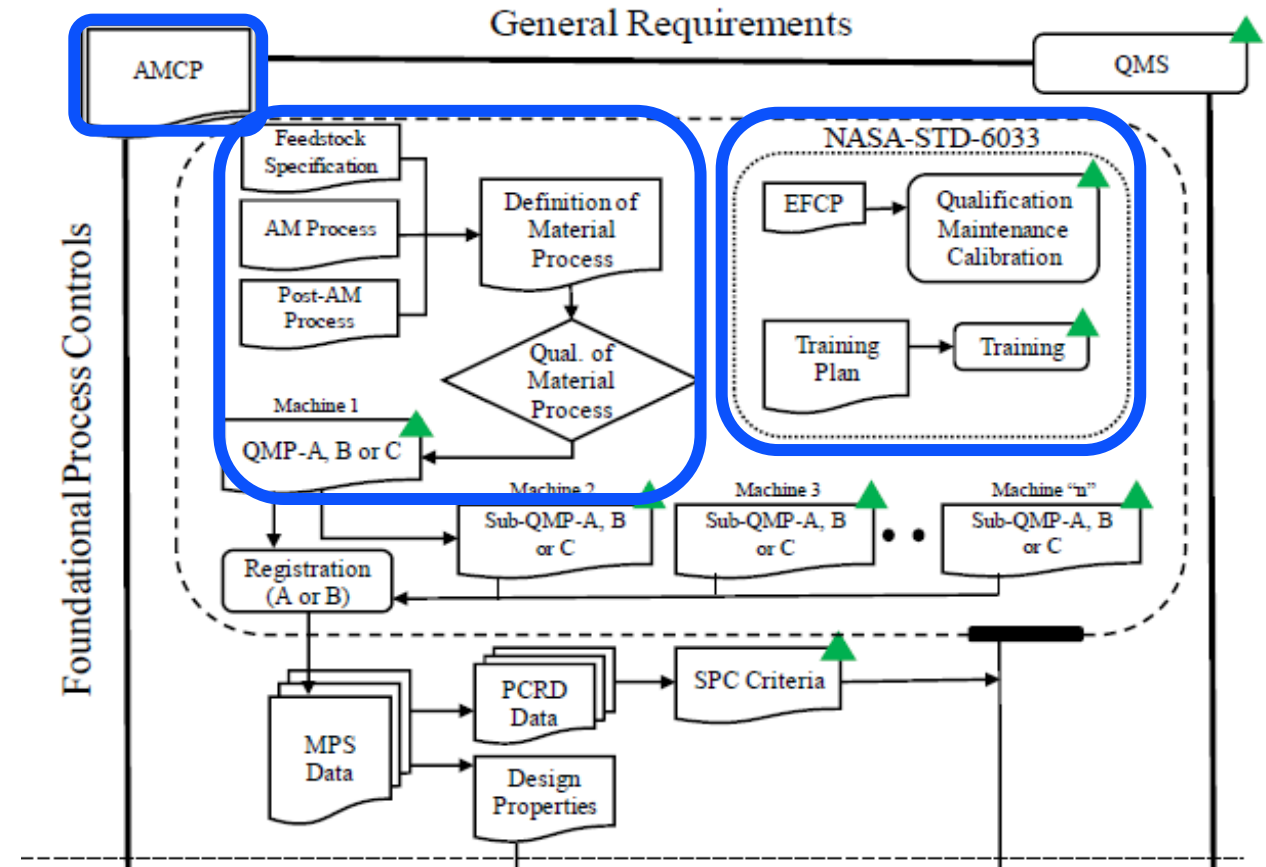
Image Credit: AON3D



- Why are qualification and certification important?
 - Ensure repeatability of process and part
 - Central to safe, flightworthy spaceflight hardware
- Walking the talk in lab
 - Stress test the requirements documents
 - Provide quality products to lab customers
- Adapt requirements to evolving technology
- Will be teaching class on NASA Approach to AM Q&C Methodology at end of August at The Aircraft Airworthiness & Sustainment Conference



- Central to NASA-STD-6030 is qualifying a material process for each different machine and material combination
 - What are you building with?
 - How does your machine operate?
 - Post processing – what evolves your material state?
- Will review completed QMPs for lab capabilities
- QMPs for each machine/material combination in the lab are in development
 - Both metallic and polymer processes/materials
- Plans for future work also include development of an Additive Manufacturing Control Plan and Equipment and Facility Control Plan



From NASA-STD-6030




Stratasys Fortus 900 Ultem 9085 QMP



FDM Qualified Material Process Record

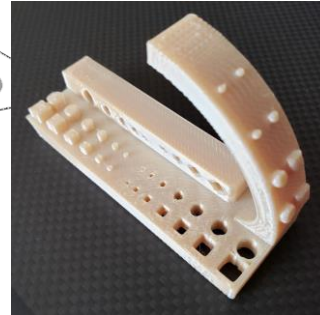
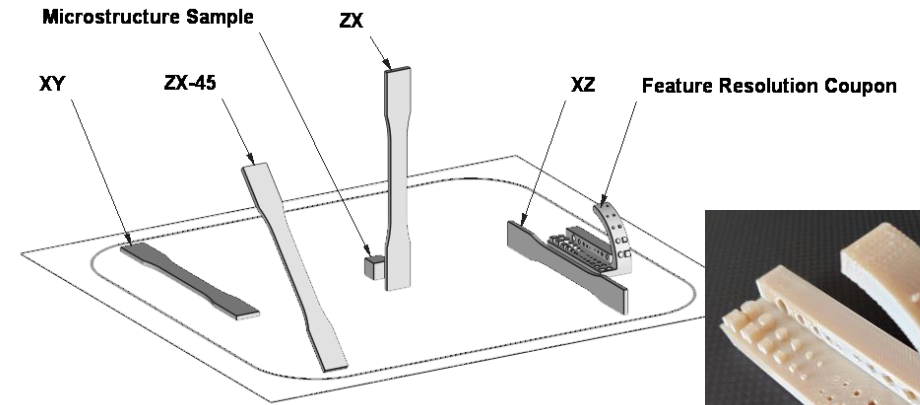
QMP Title:	QMP-MSFC-900mc-ULTEM (PEI) 9085
QMP Record Number:	QMP-MSFC-900mc-ULTEM (PEI) 9085_REV0
Check as applicable:	<input checked="" type="checkbox"/> Master QMP <input type="checkbox"/> QMP, based upon Master QMP: <input type="checkbox"/> Customized QMP (Customized FDM Visual Inspection Typical Acceptable Anomalies & Unacceptable Defects Section)
General Description:	Fused Deposition Modeling (FDM) ULTEM (PEI) 9085
RESTRICTIONS ON USE:	QMP-C applicable only to parts classified as Class-C or Exempt per NASA-STD-6030
QMP Approval Statement:	All necessary data for qualification of this material and traceability to the requirements of NASA-STD-6030 Class C has been reviewed, judged acceptable, and archived.
CEO Approval:	Brian West Date:

FDM Material Process Definition

Filament Feedstock			
Feedstock Specification:	STRATASYS ULTEM™ 9085 Certified Grade (CG)		
	 MDS_FDM_ULTEM90_85_0921.apdf		
Material traceability:	ULTEM™ 9085 CG MFR Certificates of Analysis for both raw material and filament are supplied, documenting test results and identification to match filament manufacturing lot number to raw material lot numbers providing traceability from printed part per MSFC-SPEC-555 requirements		
FDM Process Controls			
Machine ID:	900mc	Model/Model:	FORTUS 900mc
Serial Number:	L0444	Configuration Date:	5/3/2022
Slicer Software:	Insight 14.11	System Software:	3.32.3700.0
<i>System/Slicer software updates shall be evaluated by the MSFC Additive Manufacturing team in order to fully understand the changes and impacts to the FDM process. If changes are deemed to be significant or not fully understood, a new Qualification Build shall be required to confirm test data remains in family and the FDM process performs as expected. For changes deemed insignificant, written justification shall be provided in order to document that a thorough review and impact assessment has been performed. This activity shall be logged and archived in a software revision log as a companion file to the approved QMP.</i>			
Extruder Tip & Life:	T16 392 in³ Maximum model material volume consumption		
Dew point limit:	N/A		
Build Plan/Toolpath File	.CMB		
Part Interior Style	Solid		
Contour Style	Single Contour		
Contour width	0.020 inch		

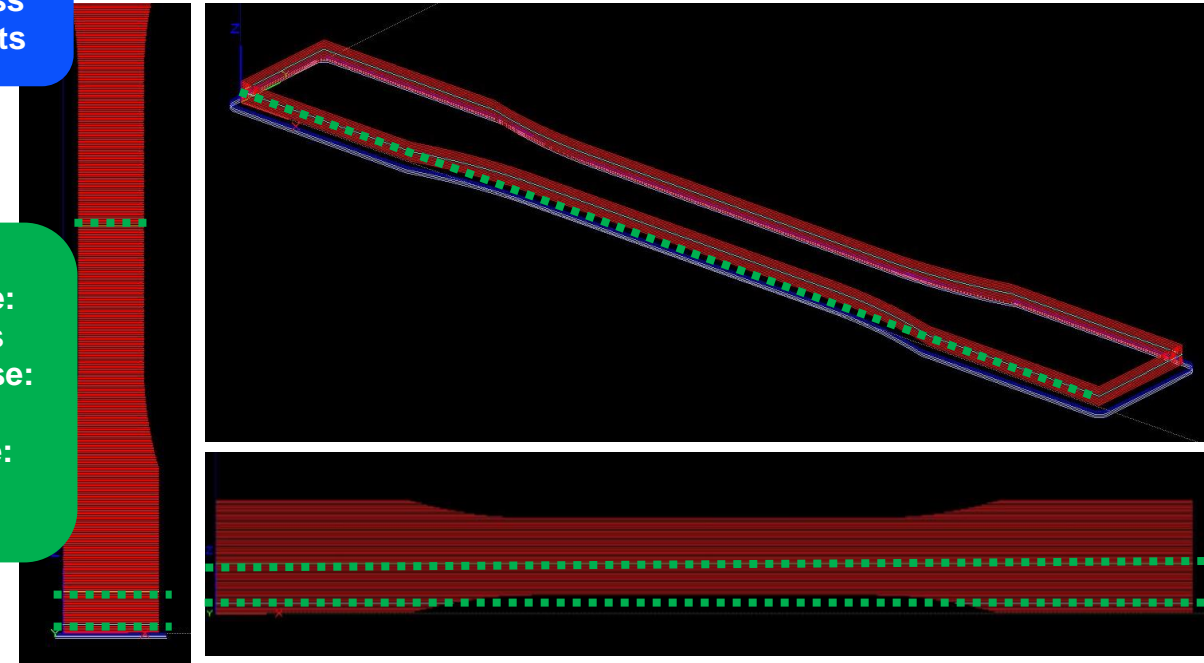
Other Topics

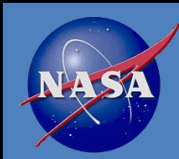
- Inspection Guide
- Tensile Data for Nominal and Thin Wall Parameters
- Comparison to NCAMP Reference Tensile Data



Qualifying Process Restarts

Short pause:
Ten minutes
Medium pause:
One hour
Long pause:
24 hours





Concept Laser M2 GRCop-42 QMP



L-PBF Material Process Record

Title: MP BlueOrigin_ACO_M2_GRCop42	
Record Number: MP BlueOrigin_ACO_M2_GRCop42	
General Description: Powder bed fusion GRCop42	
Approval Statement: This document was developed as part of an Advanced Collaboration Opportunity between NASA Marshall Space Flight Center (MSFC) and Blue Origin. It draws comparisons to MSFC GRCop-42 data and NASA-STD-6030 requirements. It should not, however, be interpreted as a complete qualification to NASA-STD-6030 requirements.	
CEO Review: Brian West	Date: 5/18/2022

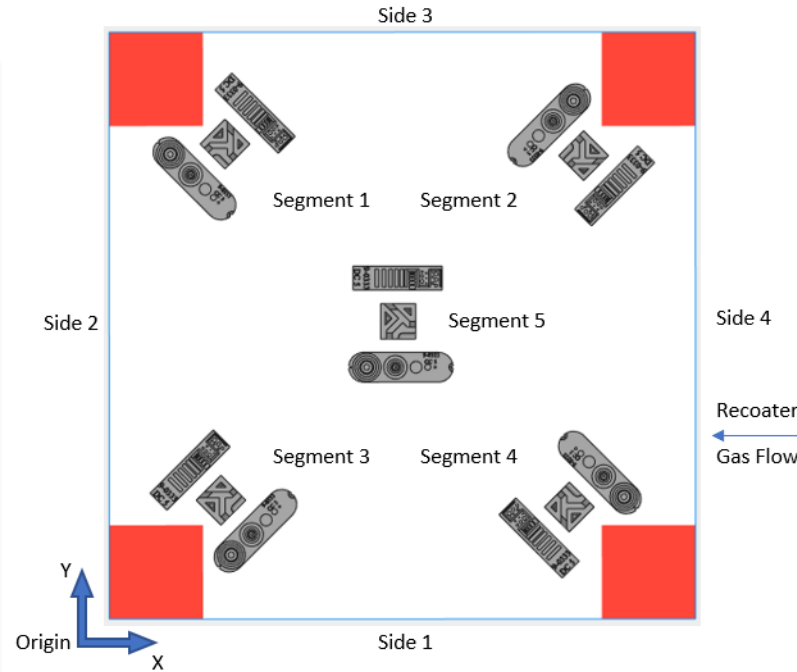
L-PBF Metallurgical Process Definition

Powder Feedstock	
Feedstock Specification:	PAC GRCop42 Lot#: AMPGR42NASAAM21030
Reuse protocol:	No Reuse

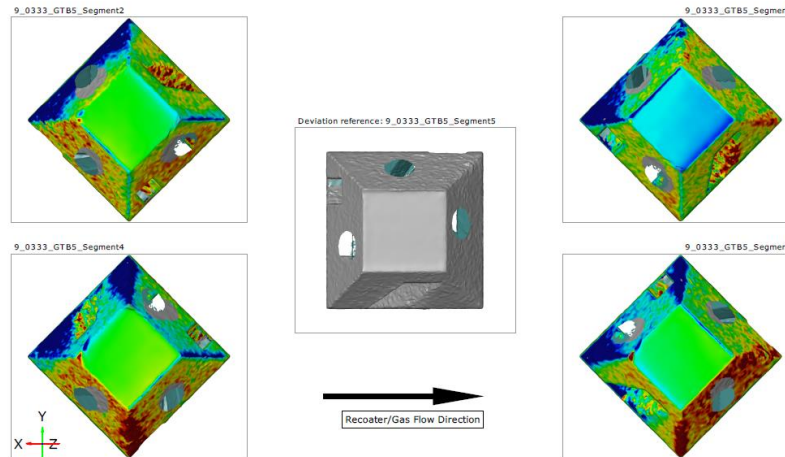
Powder Specification and Certifications

Powder procured IAW *GRCop-42 Alloy Gas Atomized Powder Specification, Rev 1-29, issued 7/22/2020*
 Powder CoC available upon request.

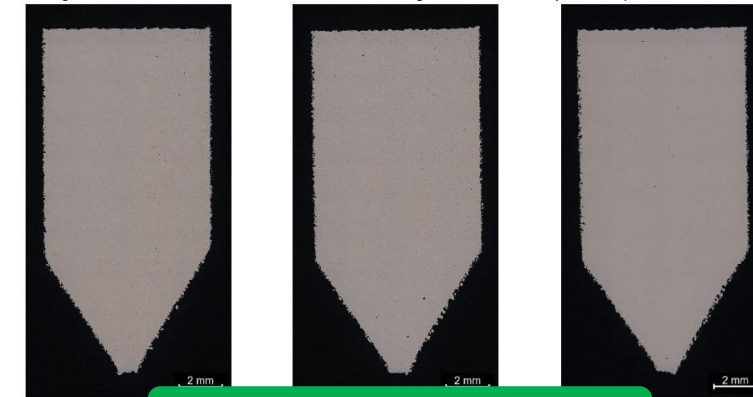
Fusion Process Controls			
Machine ID:	Concept Laser	Model/Model:	M2
Serial Number:	M2-2021-02-02	Configuration Date:	5/20/21 - 9/9/21
Slicer Software:	Materialize Magics	Software Version:	25.02
Recoater Configuration:	Steel Blade		
Build platform material:	Stainless steel, In 718 coating		
Preheat temperature:	None		
Nominal dosing range:	Variable		
Purge Gas composition:	Argon		
Argon gas flow:	60mm/s		
Oxygen limit:	Not directly controlled, but typically in ~0.01-0.05% range		
Humidity and Temperature controls:	Not measured/controlled		
Parameter File:	GRCop42_Material_Parameter	Hash:	N/A
Layer thickness:	30µm		
Other:	N/A		



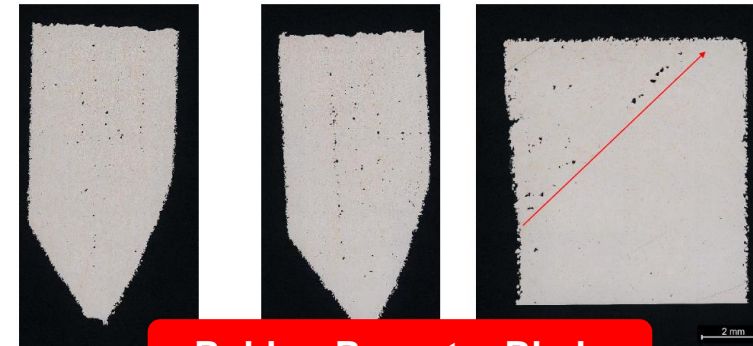
3D Deviation - Bottom View of Part Layout



Alignment By Geometric Elements
 Blue Origin ACO Geometry Test Block V01 (5) 9_0333_GT85_Sec1-5 Version 2.2 Customer: Brian West, EM42 Length unit: in 104/105



Steel Recoater Blade



Rubber Recoater Blade

Mechanical Testing

1. Tensile
2. Tensile, Thin Walls
3. High Cycle Fatigue
4. Low Cycle Fatigue
5. Fracture Toughness
6. Fatigue Crack Growth Rate

As-built Microstructure Evaluation:

Plane	Z and XY Planes
Magnification	100x

XY plane

Lack of Fusion Defects in Core

Z plane

Contour Defects at Edges

Heat-treated Microstructure Evaluation:

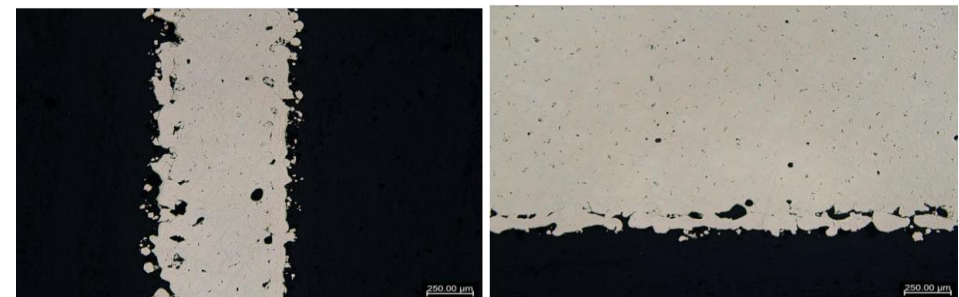
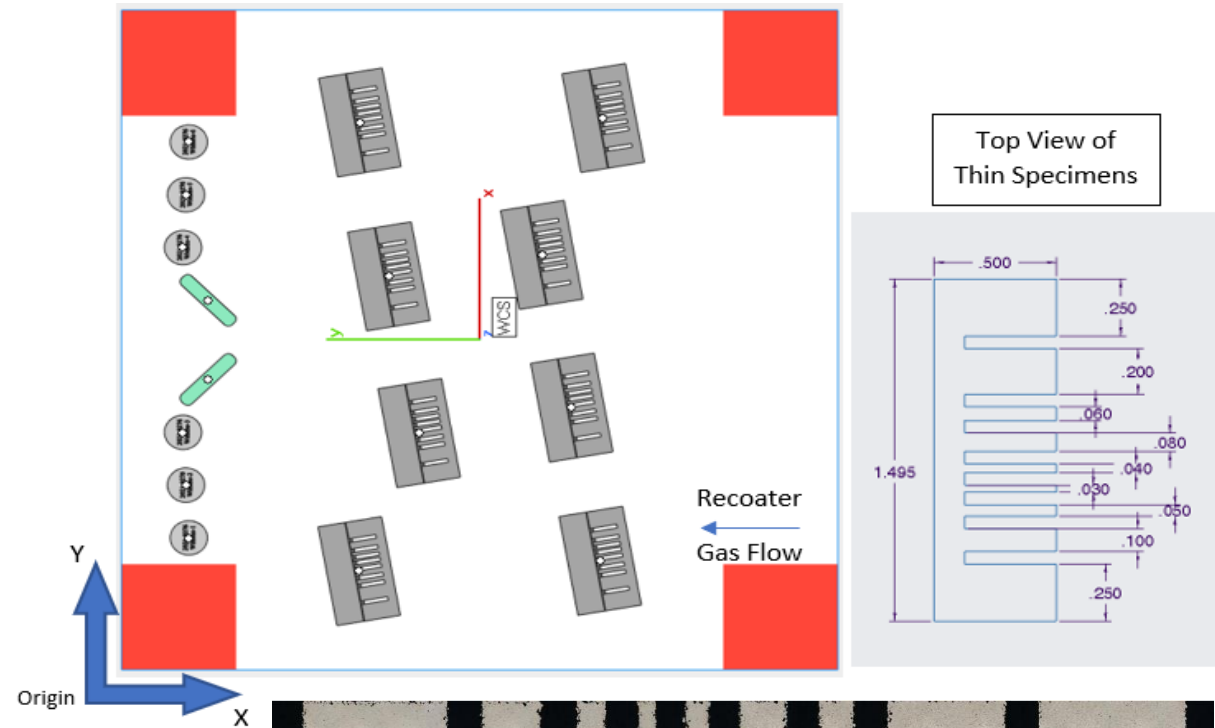
Plane	Z Plane
Magnification	100x

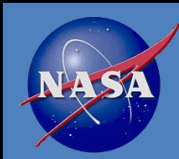
Z plane

No core defects noted

Z plane

Contour Defects at Edges





Additive Manufacturing Control Plan



MSFC Technical Document		
EM42		
Additive Manufacturing Control Plan	EM40-AMCP-001	Revision [Draft]
	Document Date:	Page 1 of 6



ADDITIVE MANUFACTURING CONTROL PLAN (AMCP)

EM40 – NONMETALLIC MATERIALS & ADVANCED MANUFACTURING DIVISION

Will define and govern lab policy and procedures for controlling our internal processes and serve as an AM team member handbook

Highlights of topic areas shown on the right

MSFC Technical Document		
EM42		
Additive Manufacturing Control Plan	EM40-AMCP-001	Re
	Document Date:	

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 - 6.8. End Item Data Package (EIDP) Information

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Equipment & Facility Control Plan



MSFC Technical Document		
EM42		
Equipment & Facility Control Plan	EM40-EFCP-001	Revision [Draft]
	Document Date:	Page 1 of 6



George C. Marshall Space Flight Center
Marshall Space Flight Center, Alabama 35812

EQUIPMENT AND FACILITY CONTROL PLAN (EFCP)

EM40 – NONMETALLIC MATERIALS & ADVANCED MANUFACTURING DIVISION

Will define and govern AM lab equipment and facility, including machine qualification, maintenance, and calibration as well as operator training

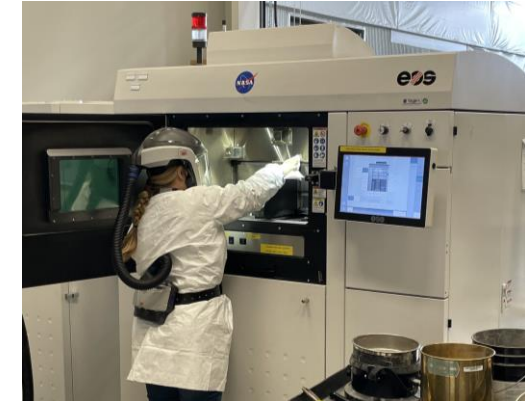
Highlights of topic areas shown on the right

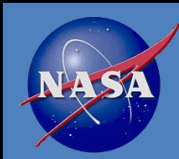
MSFC Technical Document		
EM42		
Equipment & Facility Control Plan	EM40-EFCP-001	Revision [Draft]
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 - 4.1.3. Powder Feedstock Lot Control Requirements
 - 4.1.4. Cleaning Procedures for Removal of Residual Feedstock
 - 4.1.5. Contamination and Foreign Object Debris Control
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 - 4.2.1. Computer Security
 - 4.2.2. Records Retention
 - 4.2.3. Sensitive Data
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 - 4.4.2. In Situ Monitoring
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 - 4.4.7. Calibration
 - 4.4.8. Calibration Schedules
 - 4.4.9. Optical System Calibration
 - 4.4.10. Calibration Intervals
 - 4.4.11. Calibration State
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 - 4.5.1. AM Machine Qualification Status for Production
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 - 4.5.3. Reestablishing Qualification
 - 4.6. Operator Certification
 - 4.6.1. Training Program

- 4.1. Feedstock Management
- 4.2. Digital Thread
 - 4.2.2. Records Retention
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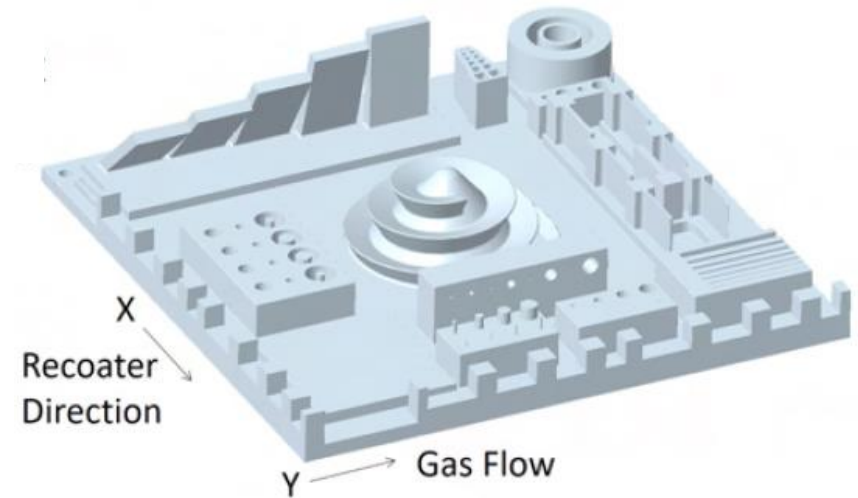
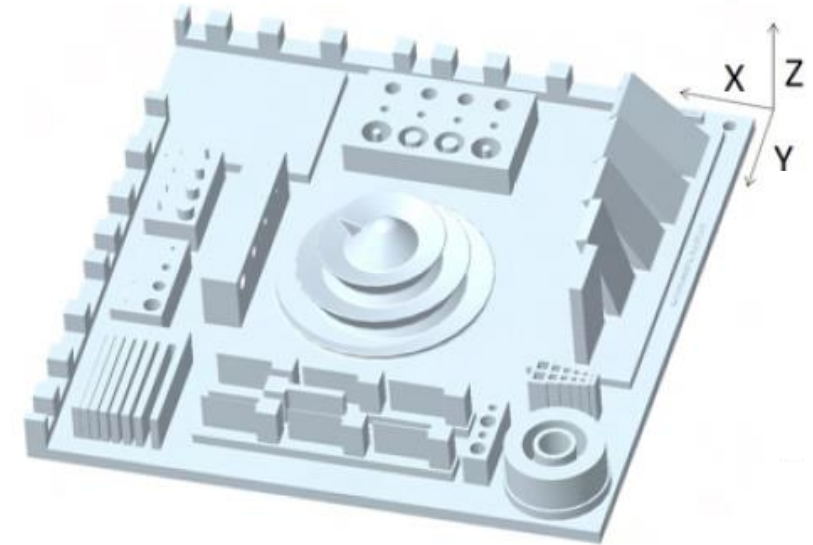


What's Next?

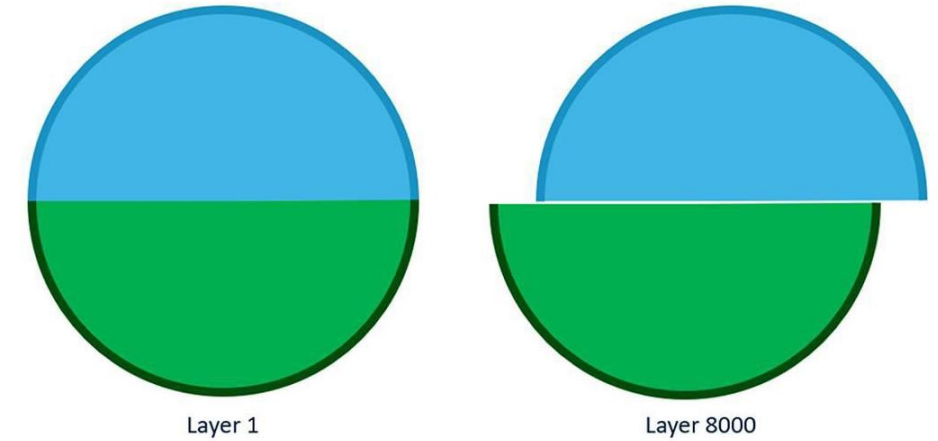
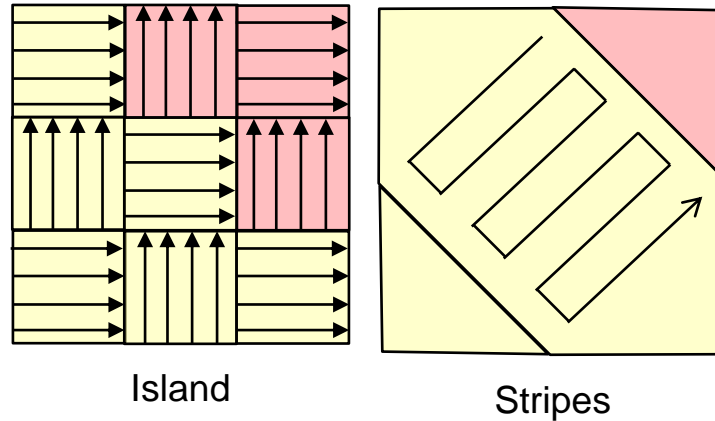


- Advanced Exploration Systems established an agency-wide team for AM Certification Support Team (AACT)
 - Develop “Smart Buyer” NASA workforce to ensure sustainable agency-wide AM certification
 - Advocate for critical AM tech advancement and capabilities across the agency, and cross-agency risk reduction
 - Provide centralized leadership for AM technical integration across the agency
- New AM Tech Knowledge Development Project
 - Focused on large-format, multi-laser technologies
 - SLM NXG XII 600, 12 lasers, Build Volume: 600 mm³
 - Additive Industries MetalFAB1, 4 lasers Build Volume: 420 mm² x 400 mm z
 - AMCM M4K, 4 lasers, Build Volume: 450 mm² x 1000 mm z
 - Velo3D Sapphire Machine Lease at MSFC
 - Working with external partners for technical exchanges and metallurgical evaluation of new multi-laser powder bed fusion systems
 - Continually monitoring AM community for other areas of technology to research
 - Model-based tools
 - Generative design
 - Computational materials

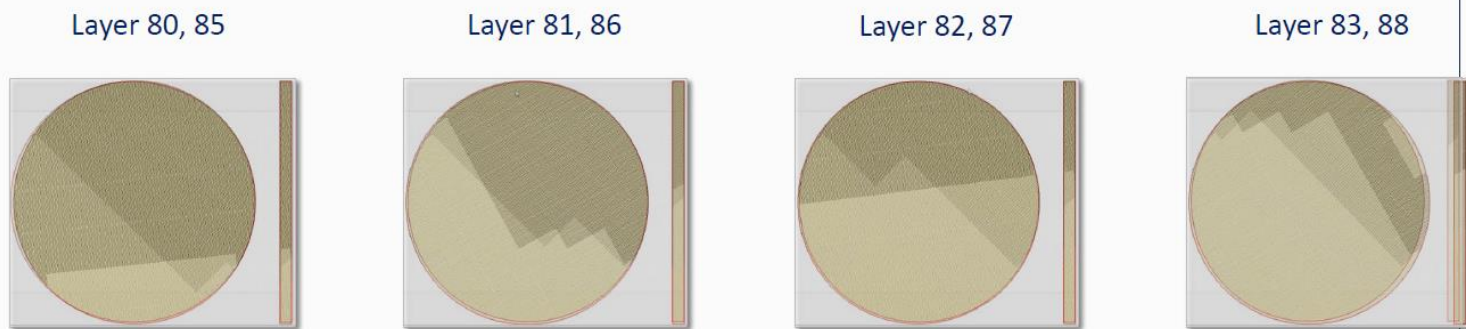
Part Name/ Number	Quantity	Details
Challenge part [Credit: P. Gradl et. al., NASA]	2	One shall be single laser (if possible), other shall be multi-laser*
Tensile specimens, ASTM E8/E8M	15	3 vertical, room temp 3 horizontal, room temp 3 vertical, high temp 3 horizontal, high temp 3 vertical, stitched*
High Cycle Fatigue (HCF) ASTM E466	9	3 vertical 3 horizontal 3 vertical, stitched*
High Cycle Fatigue (HCF) ASTM E466	3	3 vertical, as-built surfaces
Low Cycle Fatigue (LCF) ASTM E606/E606M	12	3 vertical, high strain 3 vertical, middle strain 3 vertical, low strain 3 vertical, stitched*
Promoted combustion rods	8	
Metallography bars	4	Remove one per build prior to stress relief
Laser quality specimens	One per laser source	



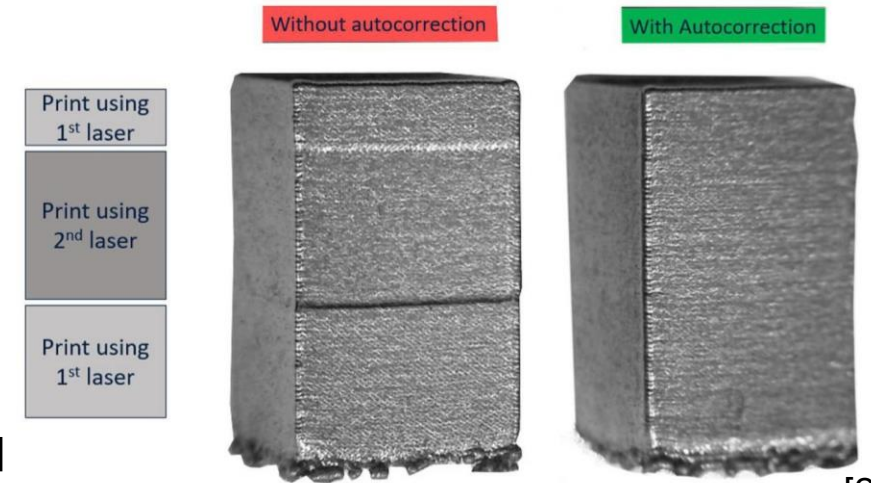
- Scan Strategies
- Laser Overlaps in Contour vs. Infill
- Laser Assignment
- Laser Alignment and Calibration
- Condensate or spatter interference with laser focus or intensity



[2]



[1]



[2]

References

[1] Godfrey, Donald. SLM Solutions. Validation of Multi-Laser Printing Technology for Additive Manufacturing.

[2] Murphree, Zach. Velo3D. Why Pre-Build Calibration is Critical to Part Quality for Metal Additive Manufacturing. Quality Magazine. 2020-09-25. http://digital.bnppmedia.com/publication/frame.php?i=674814&p=&pn=&ver=html5&view=articleBrowser&article_id=3772591



Acknowledgements



I'd like to thank the following teammates for their help and contributions to the projects discussed today:

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Tony Jones
Will Tilson