

TC3 Overview

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Welcome!

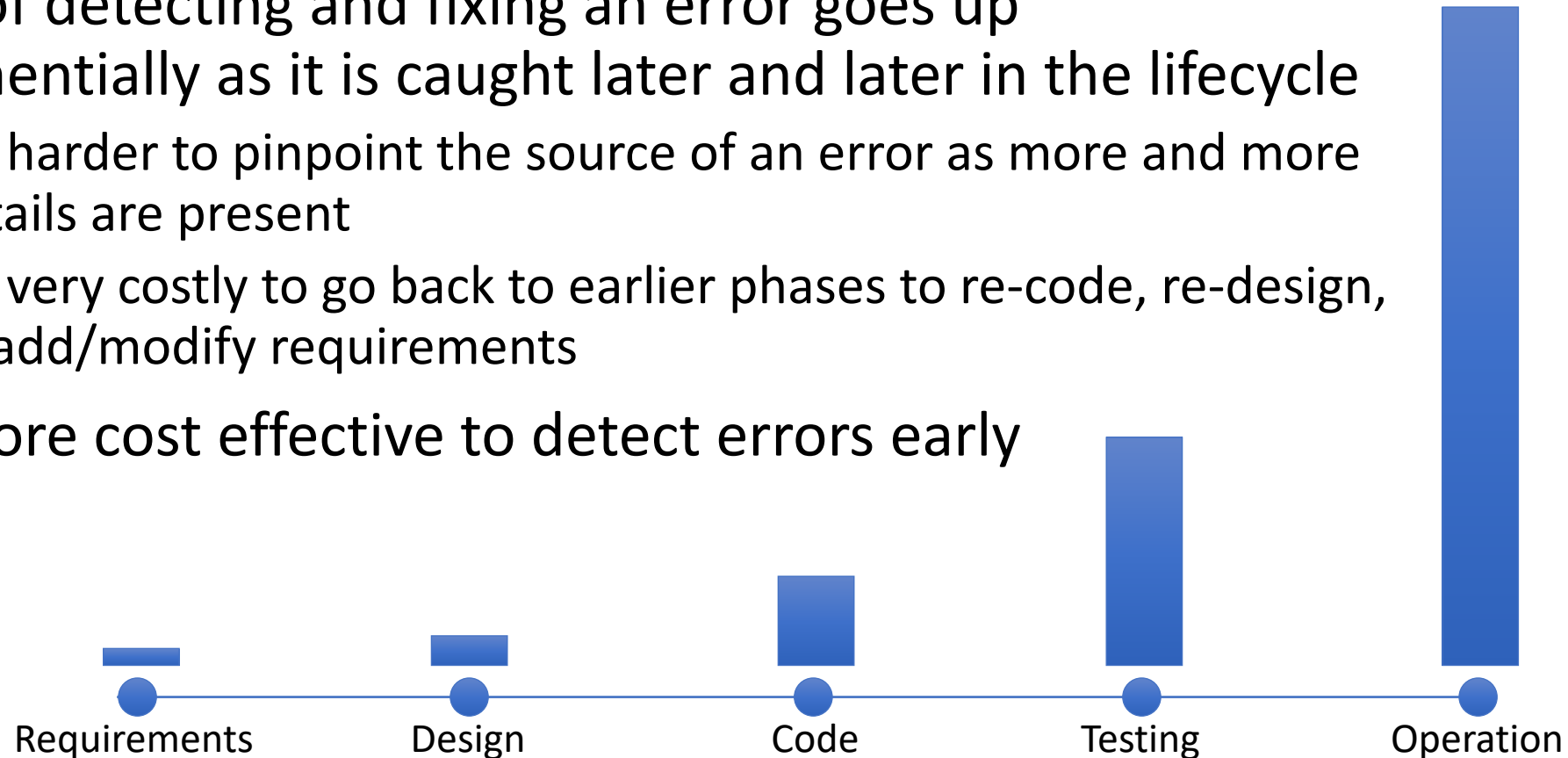
- This talk is summarizing the research done
 - under the Technical Challenge 3
 - in the System-Wide Safety project
 - in the Airspace Operations & Safety program
 - in the Aeronautics Research Mission Directorate

Motivation

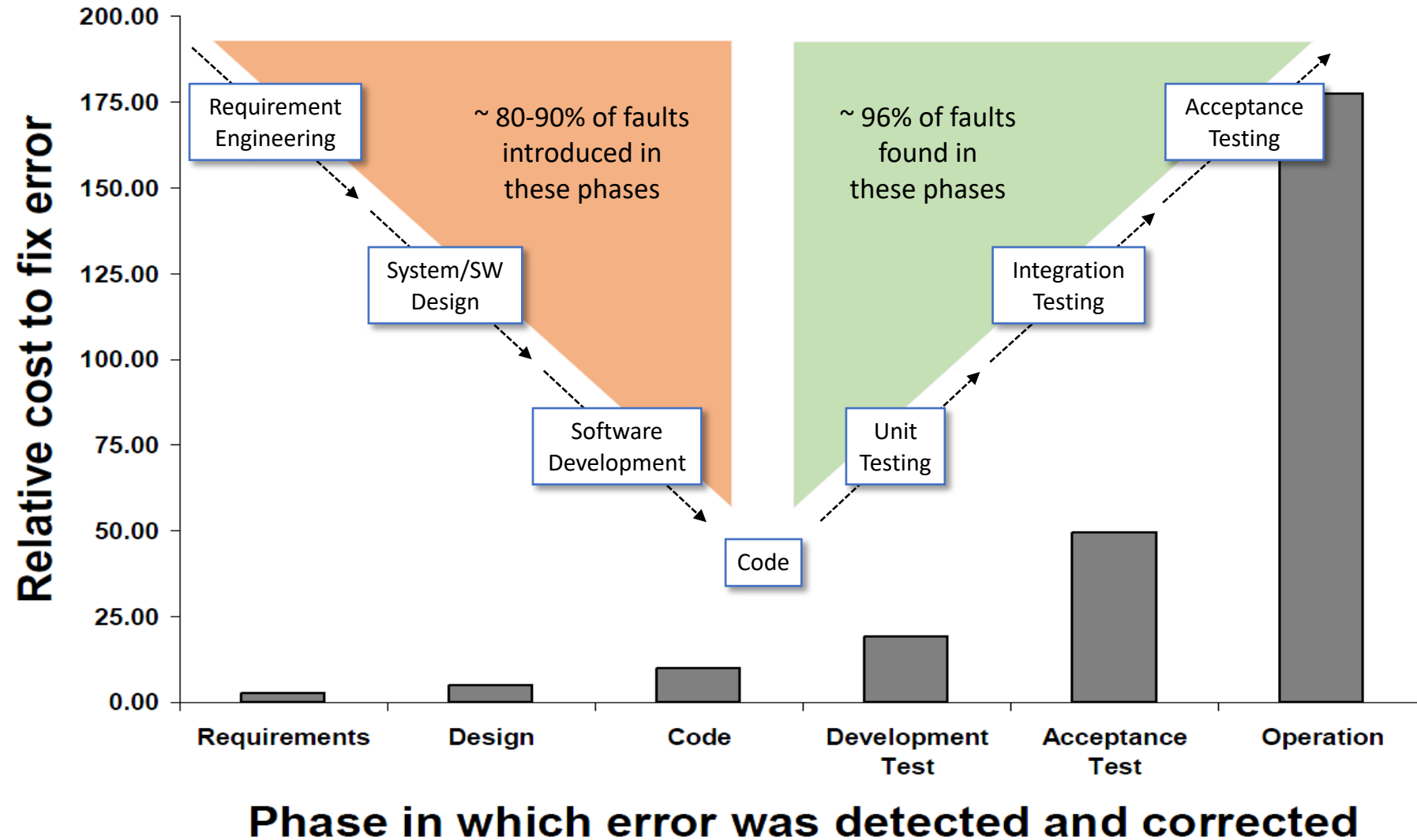
- What drive the V&V research in the first place?
 - Industry saying that the cost of V&V for certification is prohibitive
- Where is that cost coming from?
- How did this drive the NASA research strategy?

Cost of finding and fixing errors

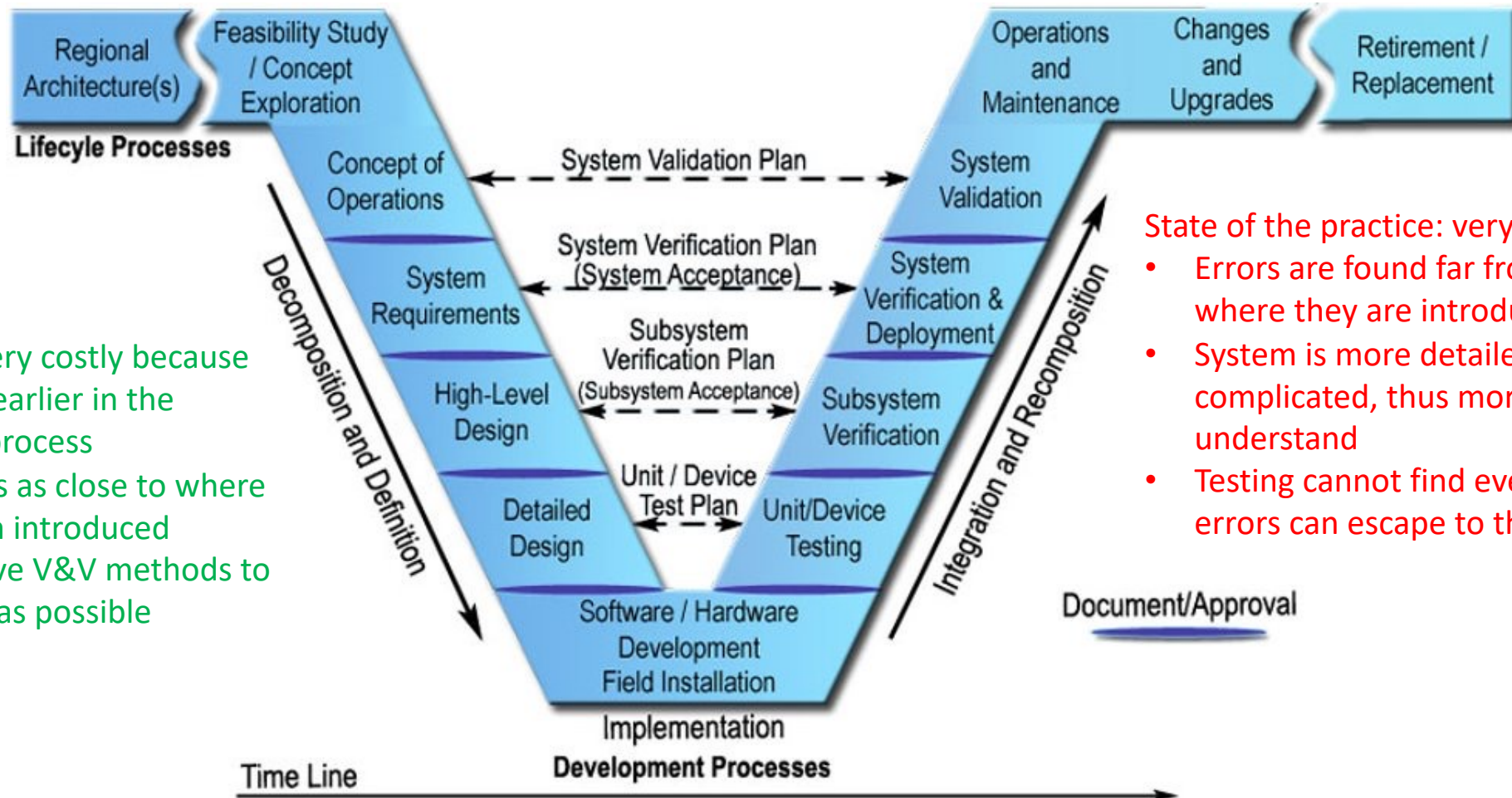
- Cost of detecting and fixing an error goes up exponentially as it is caught later and later in the lifecycle
 - it's harder to pinpoint the source of an error as more and more details are present
 - it's very costly to go back to earlier phases to re-code, re-design, or add/modify requirements
- It's more cost effective to detect errors early



Cost analysis



Motivation summary



TC3 philosophy: very costly because

- Push the V&V earlier in the development process
- Catch problems as close to where they have been introduced
- Favor exhaustive V&V methods to catch as much as possible

State of the practice: very costly because

- Errors are found far from the place where they are introduced
- System is more detailed, thus more complicated, thus more difficult to understand
- Testing cannot find every thing and errors can escape to the field

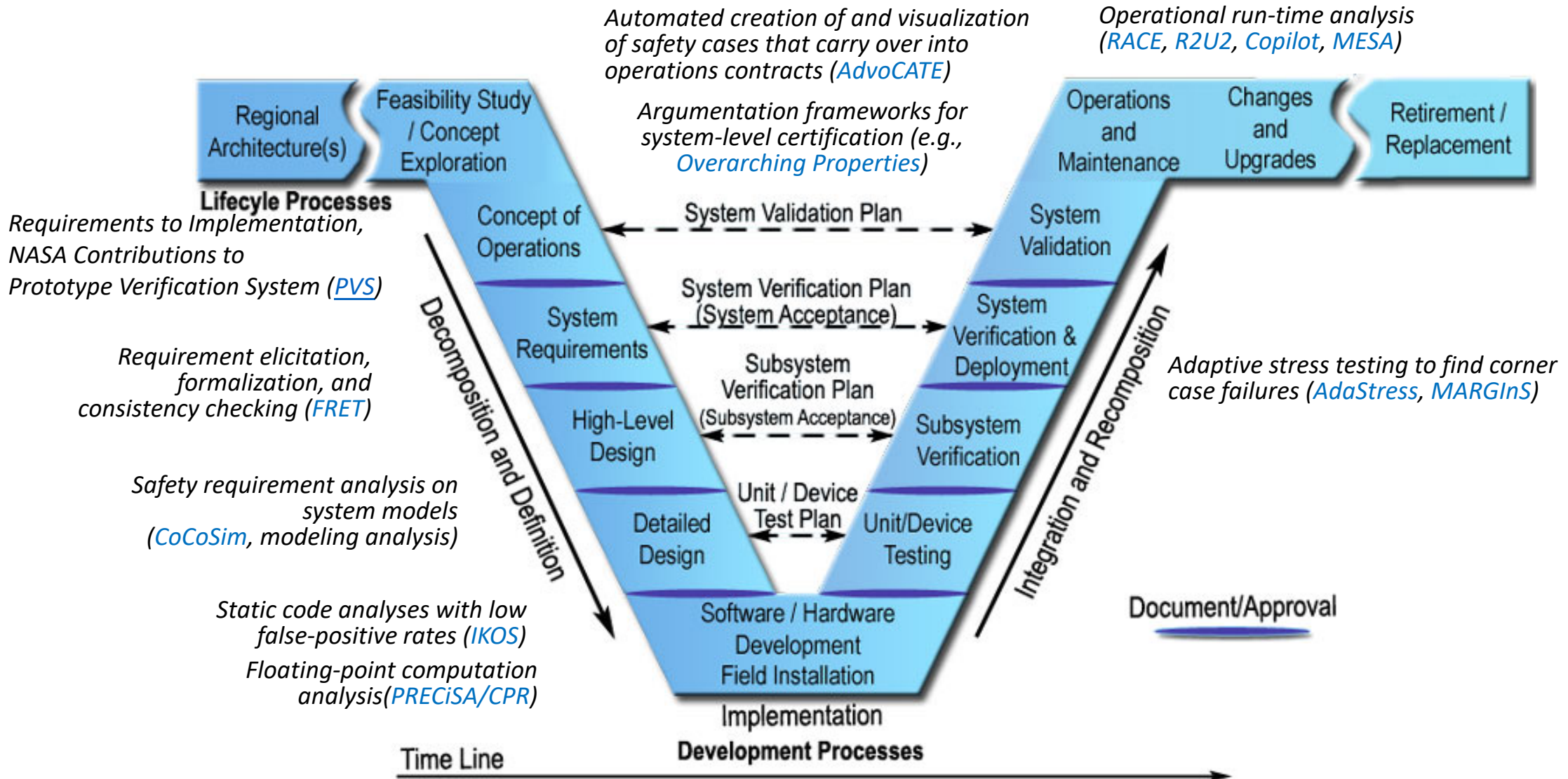
Overall strategy

- Formal methods and other advanced V&V techniques
 - Model complex systems as rigorous, mathematical entities
 - Make it possible to verify a system's properties in a more thorough fashion than empirical testing
- Focus on methods that reduce cost and reliance on testing
 - Applied earlier in the process on design models,
 - Avoid the high cost of detecting and fixing errors caught at testing
- Focus on methods that address complexity
 - Methods that work on models of the system where details are abstracted - complexity is reduced
 - Scalability challenges can be addressed such as with compositional V&V techniques
- Facilitate change at FAA to accommodate these techniques
 - Shift emphasis from testing for verification to testing for validation (of models used to verify system properties)
 - Case studies and other guidance for Industry and FAA
 - Industry has stepped up efforts to develop methods now that they see a path forward with the FAA

Research results

- What tools, techniques and processes were created?
- Have they been used outside NASA?
- What impact did they have?
- What is happening to them now?

Overview



Theorem Proving

- Formalization of semi-decision and decision procedures for non-linear arithmetic in the Prototype Verification System (PVS):
 - Branch and Bound: Solve global optimization problems using sound numerical enclosures.
 - Sturm Theorem: Satisfiability and validity of quantified univariate polynomial inequalities.
 - Tarski Theorem: Satisfiability and validity of systems of univariate polynomials inequalities.
- Implementation of automated strategies for proving properties involving non-linear arithmetic, including square root and transcendental functions, and for performing rigorous computations:
 - Interval arithmetic.
 - Affine arithmetic.
 - Bernstein polynomial basis.
 - Exact arithmetic.
- Included in the publicly available NASA PVS Library:
 - <http://github.com/nasa/pvslib>

PVSioChecker: Validation By Model Animation

- PVSioChecker is a PVS tool for validating numerical software against their functional specification using the following approach:
 - Formally verified functional models of algorithms, which use real arithmetic, are evaluated on a set of test vectors.
 - Implementation of these algorithms, which use floating-point computations, are evaluated on the same set of test vectors.
 - Outputs are compared using a user-specified precision.
- Released as part of NASA PVS Library.

FRET

Create tool (FRET) to create requirements (as close to English as possible) and formalize them for analysis

Requirement ID
FSM-001

Parent Requirement ID

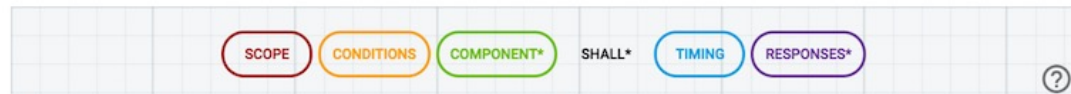
Project
LM_requirements

Rationale

Exceeding sensor limits shall latch an autopilot pullup when the pilot is not in control (not standby) and the system is supported without failures (not apfail).

Requirement Description

A requirement follows the sentence structure displayed below, where fields are optional unless indicated with "**". For information on a field format, click on its corresponding bubble.



FSM shall always satisfy (limits & autopilot) => pullup

user types
requirement

parser recognizes fields and
color codes them dynamically

Semantics

Always, the component "FSM" shall satisfy $((limits \ \& \ \text{autopilot}) \Rightarrow pullup)$.



Response = $((limits \ \& \ \text{autopilot}) \Rightarrow pullup)$.

Diagram Semantics

Formalizations

Future Time LTL

$G \ ((limits \ \& \ \text{autopilot}) \Rightarrow pullup)$

Target: *FSM* component.

Past Time LTL

$((limits \ \& \ \text{autopilot}) \Rightarrow pullup) \ S \ ((limits \ \& \ \text{autopilot}) \ \& \ \text{FTP})$

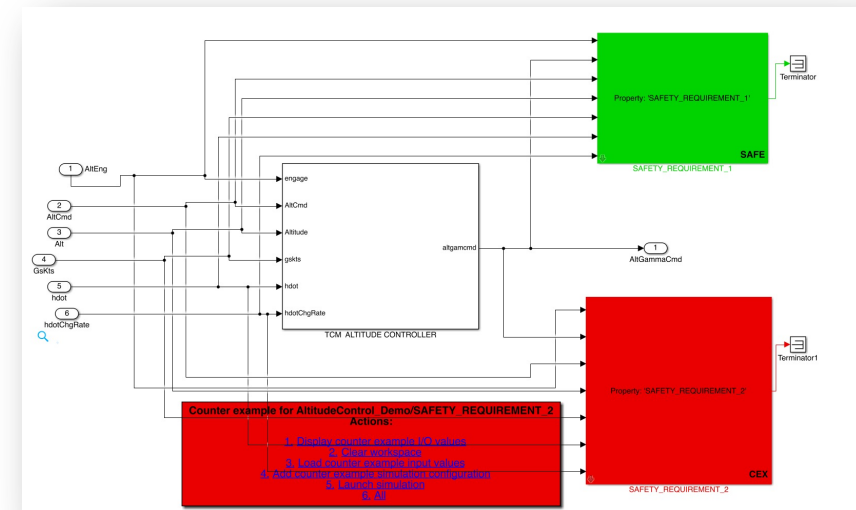
Target: *FSM* component.

FRET
generates
formal
semantics

CoCoSim

CoCoSim is applied directly to models; it can be used to verify safety requirements *before* the code is produced

- Integrated verification framework for Simulink models
- Fully automated analysis framework
- Shown effective in avionics applications



Limitations and future work

- Currently supports a subset of Simulink models described here <https://github.com/coco-team/cocoSim>
- Uses Z3 constraint solver – plan to integrate other solvers to extend the constraints that can be handled, for example trigonometric constraints
- Does support Stateflow but not embedded matlab

Kodiak: A Library For Solving Global Optimization Problems

- C++ Implementation of a formally verified generic branch and bound algorithm.
- Features:
 - Enclosure methods: Bernstein polynomial basis and interval arithmetic.
 - Parametric branching and bounding heuristics.
 - Support for symbolic computations.
- Tools:
 - Min/Max solver of non-linear functions.
 - Paving algorithm for computing under and over approximations of regions defined by system of non-linear inequalities.
 - Bifurcation analysis of systems of partial differential equations.
 - Boolean solver.
- Released under NASA's Open Source Agreement:
 - <http://github.com/nasa/kodiak>

IKOS

IKOS performs compile-time static analysis of software code and looks for errors that can happen at run-time

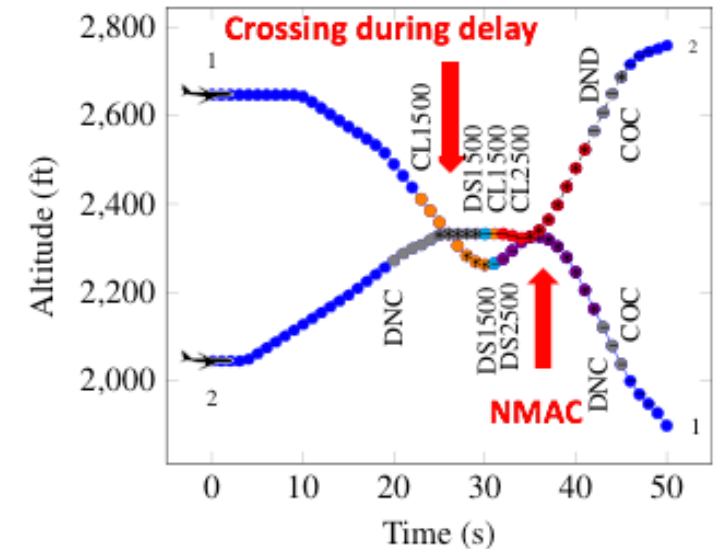
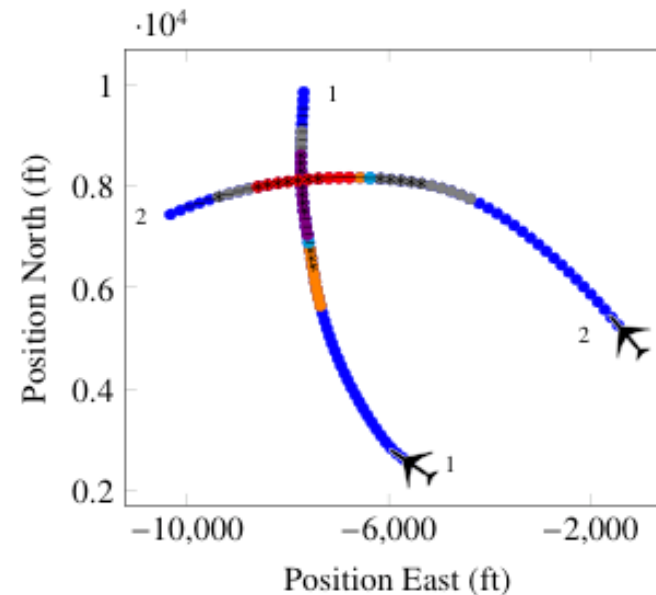
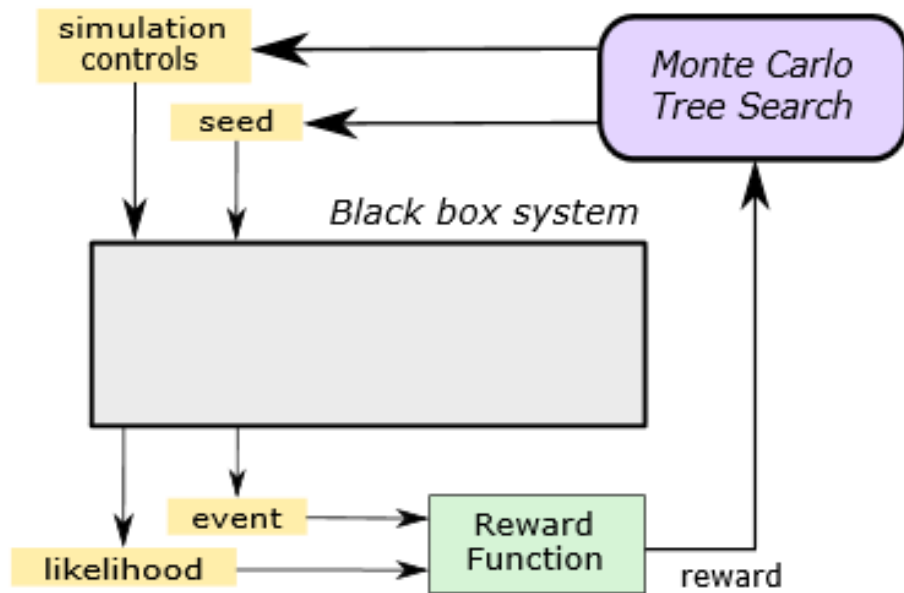
- Analyzes C code and C++ code
 - using the LLVM frontend
 - limited support for C++
- The code does not need to execute:
 - Libraries are not required, but knowledge on the environment and libraries is needed on a per case basis
- Available analyses:
 - Buffer overflow errors, Division by Zero, null pointer dereferences, use of uninitialized variables
- No size restriction, but so far applied to 500 KLOC max
- Available open source at

PREClS-A: Floating-Point V&V

- Development of static analysis tool for floating-point error analysis:
 - Input:
 - A set of functions involving floating-point arithmetic.
 - Range of input variables.
 - Outputs:
 - Bounds on round off errors of floating-point computations.
 - PVS proofs that those bounds are correct.
- Integration with Frama-C (under development)

AdaStress: ML for stress testing

- AdaStress is a software package for an accelerated simulation-based stress testing method for finding the most likely path to a failure event



Integrated by GE in their Continuous Testing framework for aviation products

MARGINs

MARGINs is a library of machine learning and statistical tools for system testing. MARGINs creates visualizations for aiding analyst understanding.

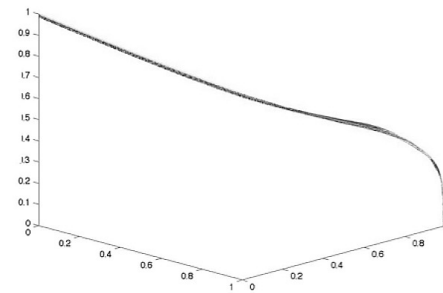
- Is an advanced test generation and test analysis toolset
- Implemented in MATLAB, C, and R – also runs in Octave
- Used for large and small scale space applications (Pad Abort, EFT-1, LADEE)

Pad Abort example



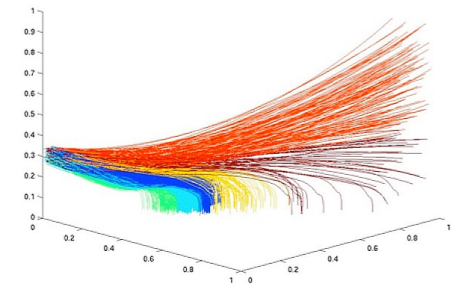
Limitations and future work

- Like any testing tool, MARGINs can increase confidence in the system but cannot guarantee the absence of errors
- Many of the results are visual, and require interpretation from domain experts
- Framework can be [extended](#) with additional components such as different machine learning techniques or test case generation schemes.



Standard testing
No automated analysis

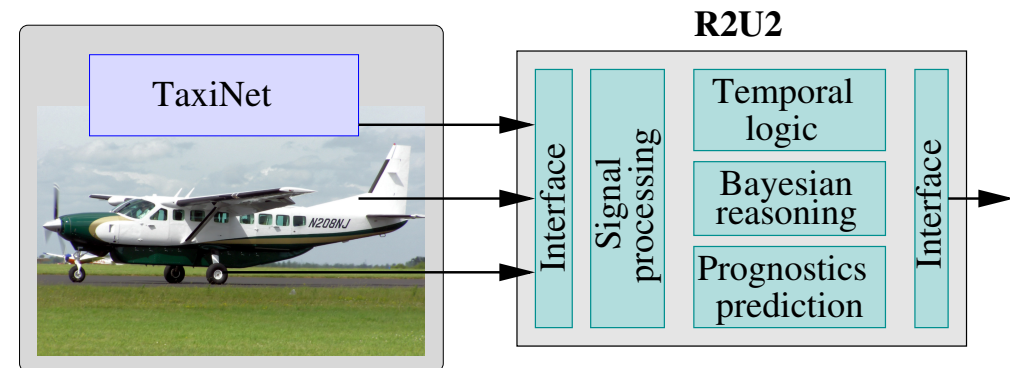
Vs.



Testing using MARGINs
Color indicates “risk classes”

R2U2: Run-Time monitoring

- R2U2 (Realizable, Responsive, Unobtrusive Unit), a hardware-supported tool and framework for the continuous monitoring of safety-critical and embedded cyber-physical systems.
- The R2U2 monitoring engine can be instantiated as a hardware solution, running on an FPGA, or as a software component.



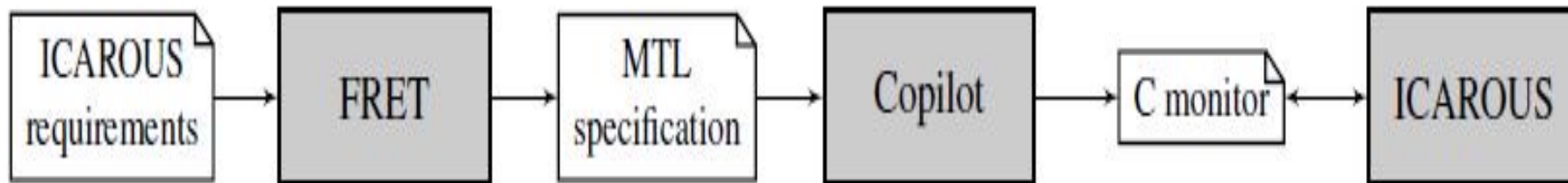
R2U2 is a run-time monitoring and V&V tool that combines *Metric Temporal Logic* observers, *Bayesian Network* reasoners, and *model-based prognostics*.

Co-Pilot

Goal: Develop run-time verification capabilities:

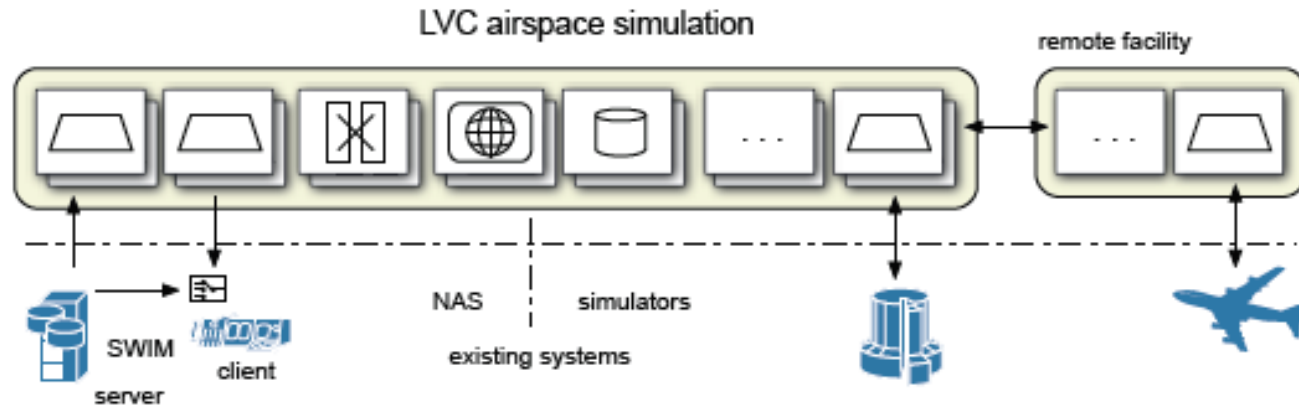
- ✓ Ultra-critical systems require high-level assurance
- ✓ They cannot always be verified during compile time
- ✓ Runtime verification enables monitoring of these systems during runtime
- ✓ RV Detects property violations early
- ✓ Limits the potential consequences of violations
- ✓ Facilitates writing RV monitors in high-level languages

- Co-Pilot 3.1 (Open Source)
 - ✓ Used in DARPA Assured Autonomy
 - ✓ Performs the translation to C
 - ✓ Stream-based runtime verification language and framework capable of generating portable, hard-real time C99 code
 - ✓ Suitable for embedded systems and architectures
- X-Plane
 - ✓ Co-Pilot has versatility to monitor flight sims in X-Plane



RACE

- started as a distributed LVC simulation framework in 2015



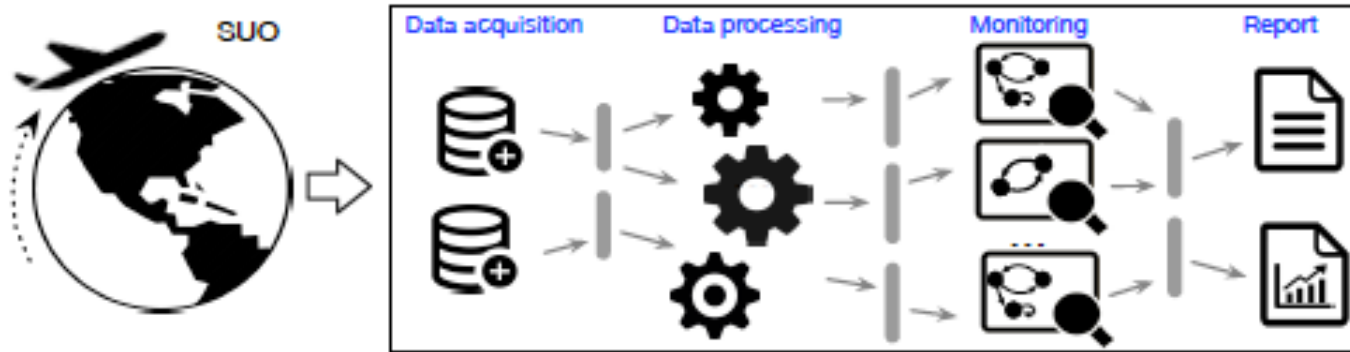
- evolved into a general event driven application framework:
 - can import/export from/to external systems - **connectivity**
 - can process high event rate and data volume - **scalability**
 - supports distributed and massively concurrent operation
 - has batteries included (except Java runtime, SBT build system)

Application: automatic detection of

- (1) temporal inconsistencies in SWIM messages
- (2) unsafe parallel approach trajectories
- (3) trajectory deviations between different reporting systems
- (4) loss of separation between live and simulated aircraft

MESA

MESA is a framework for building actor-based monitoring systems.



MESA (MessaGe-based System Analysis) enables using concurrent (runtime) monitors to check for properties specified in data parameterized temporal logic and state machines.

Application: Using MESA, we discovered violations at SFO of the following property:

PRSA : a flight shall cross inside a 1.0 NM radius around each waypoint in the assigned RNAV STAR route, in order.

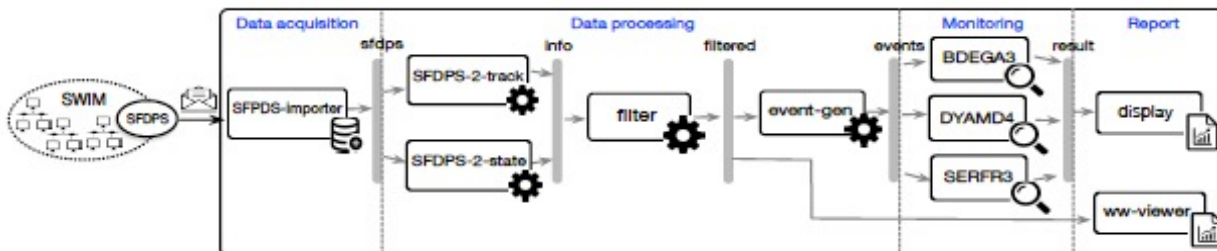
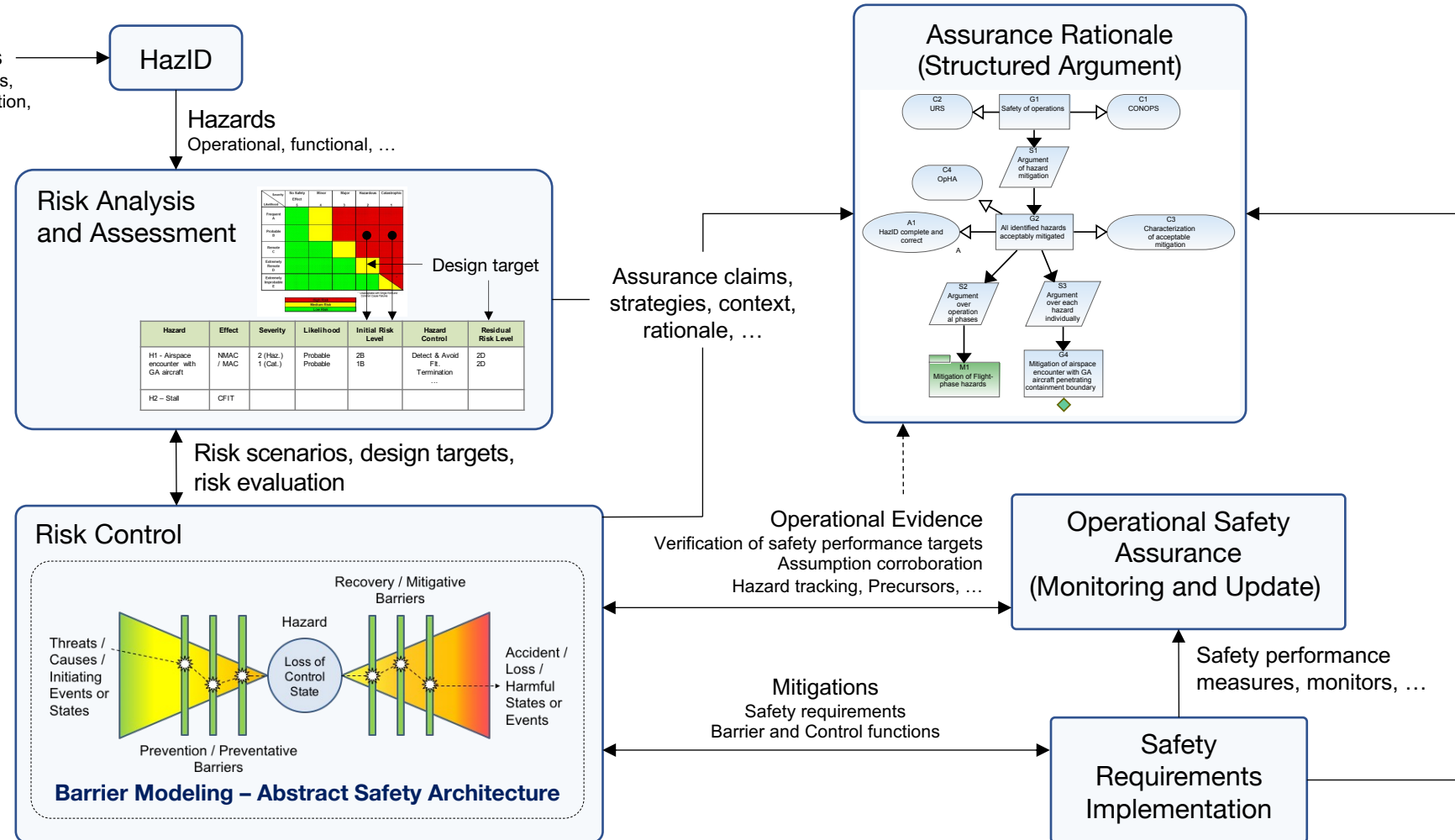


Fig. 6: Flight deviation from assigned RNAV STARs detected at SFO.

AdvoCATE

The Assurance Case Automation Toolset (AdvoCATE) supports the development and management of safety/assurance cases, providing novel capabilities in automating their production, with applicability to safety-critical applications in general, and especially aviation

- general, and especially aviation
- Hazard analysis and risk assessment:
- Capture of risk reduction and assurance requirements
- Safety architecture modeling using modules and hierarchy.
- Traceability and consistency between related artifacts
- Assurance analytics



Overarching Properties

Motivation and Objective

- Investigate the feasibility and efficacy of creating a viable path for certification / approval of airborne and ground-based systems based on the principles and methods of argumentation to show a system possesses the Overarching Properties (OPs)
- **Intent:** The *defined intended behavior* is correct and complete with respect to the *desired behavior*.
- **Correctness:** The *implementation* is correct with respect to its *defined intended behavior*, under *foreseeable operating conditions*.
- **Innocuity:** Any part of the *implementation* that is not required by the *defined intended behavior* has no *unacceptable impact*.

Case studies and Impact

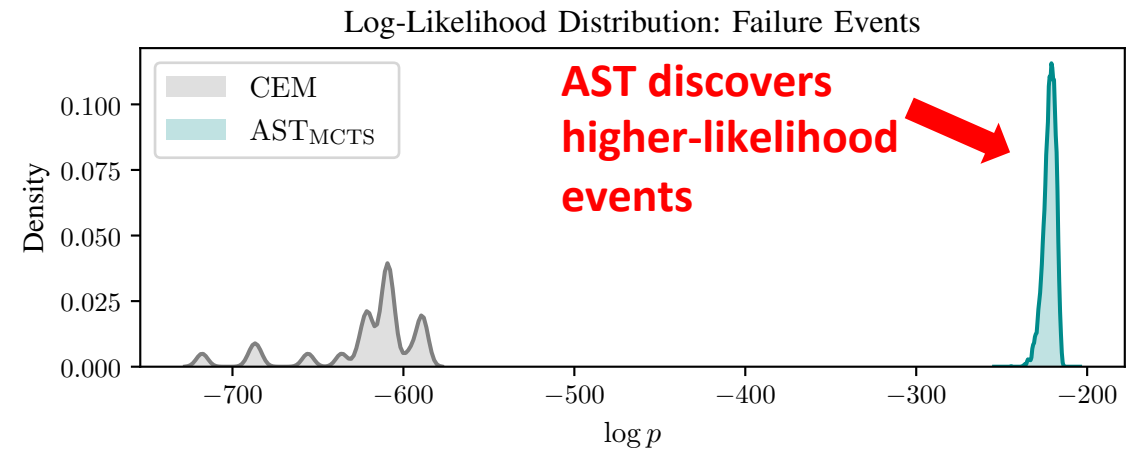
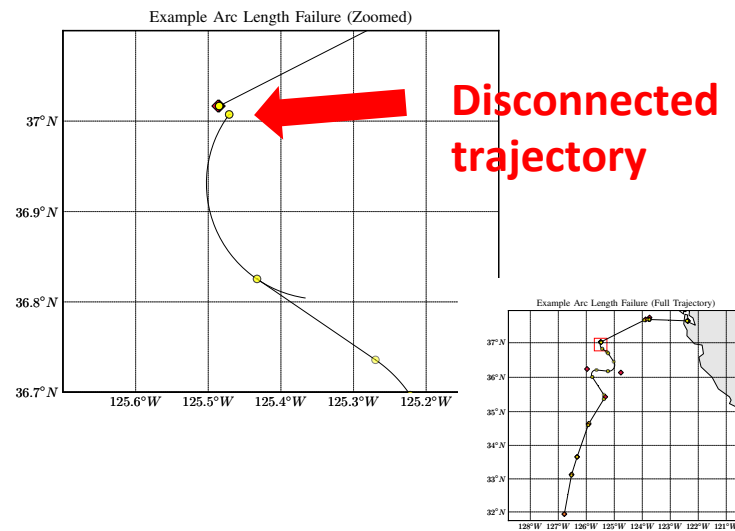
- Industry-led case studies
 - Collins Aerospace / Raytheon: formal methods, CoCoSim
 - GE: AdvoCATE, AdaStress
 - Boeing: many tools presented to large audience
- In-house case studies
 - CoPilot used in TC2
 - Troupe project integrated many tools in development process

Collins / Raytheon Experience

- Use of formal methods to do fault analysis for an AutoThrottle example.
 - Approach shows that more faults could be found that way over traditional approaches.
 - The formal approach also proves less costly (time savings)
- Use of CoCoSim and Design verifier to prove formal properties on FADEC model in Simulink.
 - The experience proved that formal methods are useful and can find interesting problems.
 - It also showed that scalability and usability improvements are still needed.
 - Also provided feedback on FRET, the requirement formalization tool.

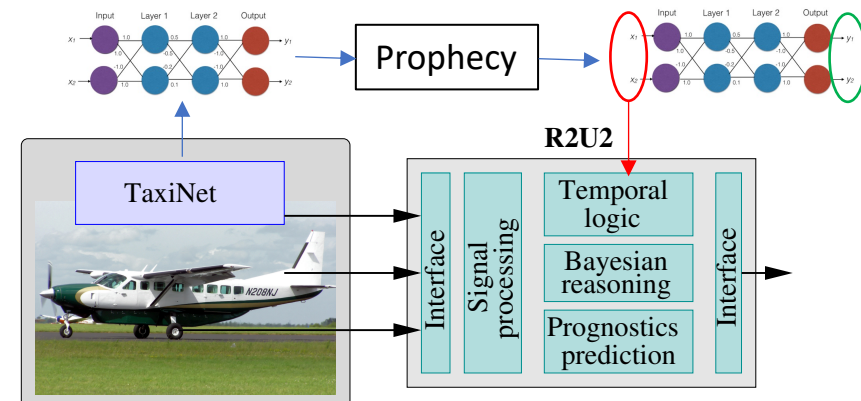
GE Experience

- Application of adaptive Stress Testing (AdaStress) to a flight management system
- GE decided to fold AdaStress into their continuous test framework for aviation products



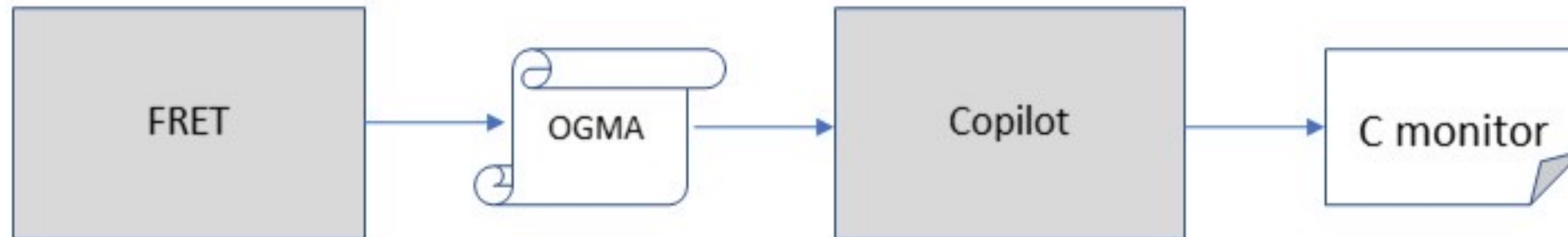
Boeing presentations

- Boeing: taking a hard look (with their evaluation) of many NASA-develop tools
 - Reaches across many Boeing orgs
 - Could revolutionize the way Boeing does V&V
- NASA gave a series of virtual talks to Boeing, which were very well attended (80+ attendees)
 - IKOS, CoCoSim, FRET, MARGInS, RACE, MESA
- This is resulted in an on-going deeper and tighter collaboration on assurance for autonomy under a Space Act Agreement.
- Example: Using Prophecy with R2U2 on Taxinet



CoPilot used in TC2

- The goal is to develop run-time verification capabilities:
 - Ultra-critical systems require high-level assurance
 - They cannot always be verified during compile time
 - Runtime verification enables monitoring during runtime
 - RV Detects property violations early
 - Limits the potential consequences of violations
 - Facilitates writing RV monitors in high-level languages
- An associated tool, OGMA, takes requirements capture files from FRET & Collins Lustre and transforms them to Copilot specs



Conclusions

- TC3 was a very successful project. It produced many tools and techniques for the assurance of critical aviation software systems
- We have covered all aspects of the V diagram
- Some of the tools have been used in industry and have shown promises to reduce assurance cost.
- These tools and techniques are still needed as we turn our focus towards autonomous systems
 - Many are being extended to deal with autonomy
 - New tools are also being developed