

# Space Nuclear Thermal Propulsion (SNTTP) Lessons Learned

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# Thank You

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Many thanks to Roger Lenard who was the Program Manager for SNTP  
and Timberwind

Without Roger's Leadership, this Program would have never happened

I am presenting a snapshot of work of over 400 people

All the credit for the accomplishments goes to them

# What SNTP was about

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- Develop and test a high thrust-to-weight nuclear thermal rocket engine
- Integrate with a launch vehicle
- Validate feasibility of performing ground-based boost-phase intercepts of long-burn Soviet ICBMs in boost and post-boost phases
- Required
  - High thrust-to-weight ratios:  $>20$  to 1
  - Rapid startup:  $< 10$  seconds to full power
  - $\sim 120$  seconds operation
  - High  $I_{sp}$ :  $\sim 1000$  seconds
  - 75,000 lbs Thrust
- Started: October 1987, Terminated 1992
- 2 National Labs, 4 Aerospace Contractors
- Approximately \$400 M spent\*

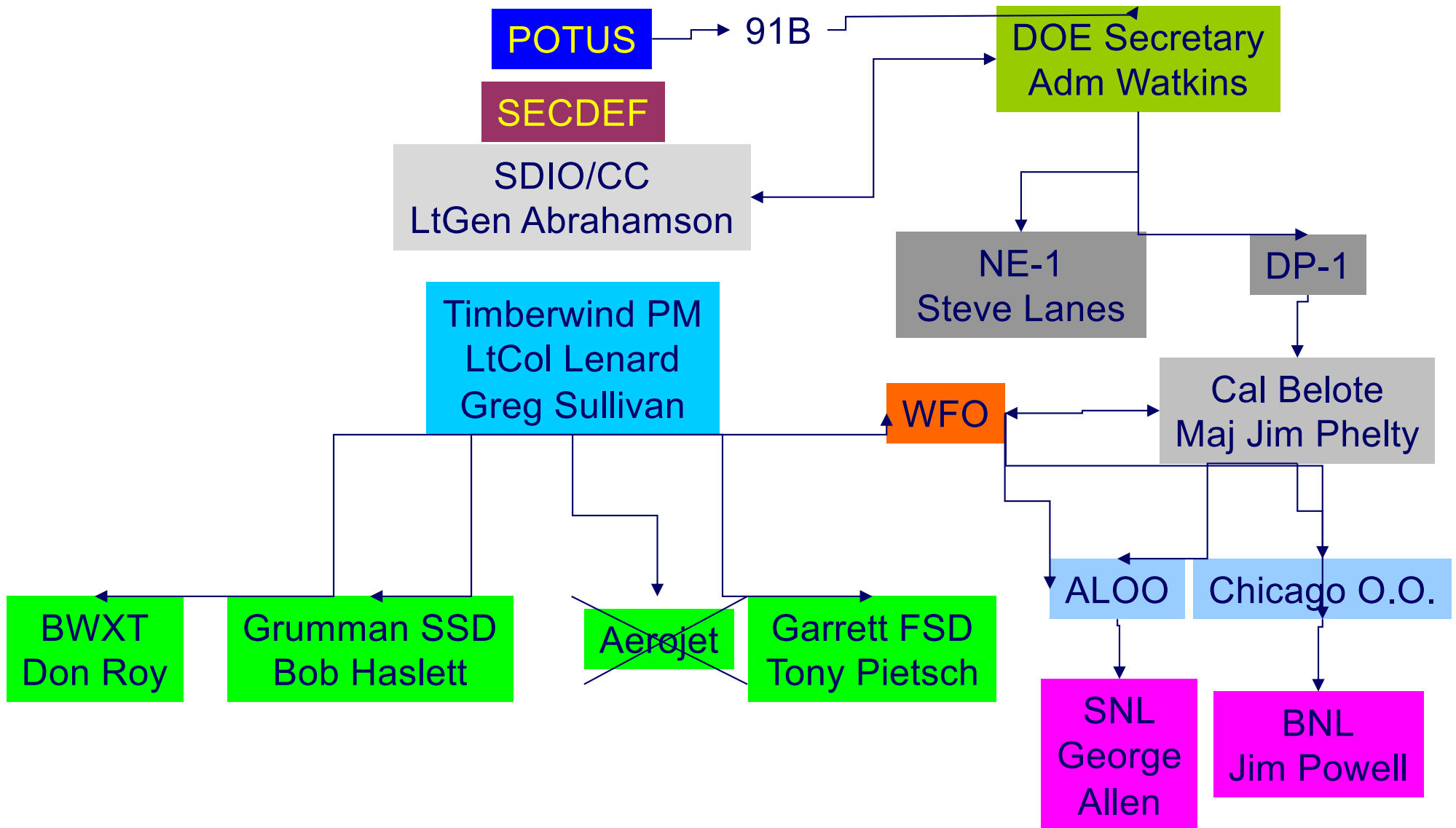
# The Beginning

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- July 1987: Team of Sandia, Brookhaven National Laboratories, XERAD, Grumman Space Systems, Allied Signal, Babcock & Wilcox, and Aerojet meet with LtGen Abrahamson
- Propose a high T/W\* ground-based system to perform boost-phase intercepts
- Requires extremely high  $\Delta V$ s and high accelerations
- Key technology: Particle-bed Reactor

\* In generic terms, the goal was to put the power of the Hoover Dam (2 GW) in the volume 2/3 the size of a 55 gallon drum

# Timberwind Organizational Structure

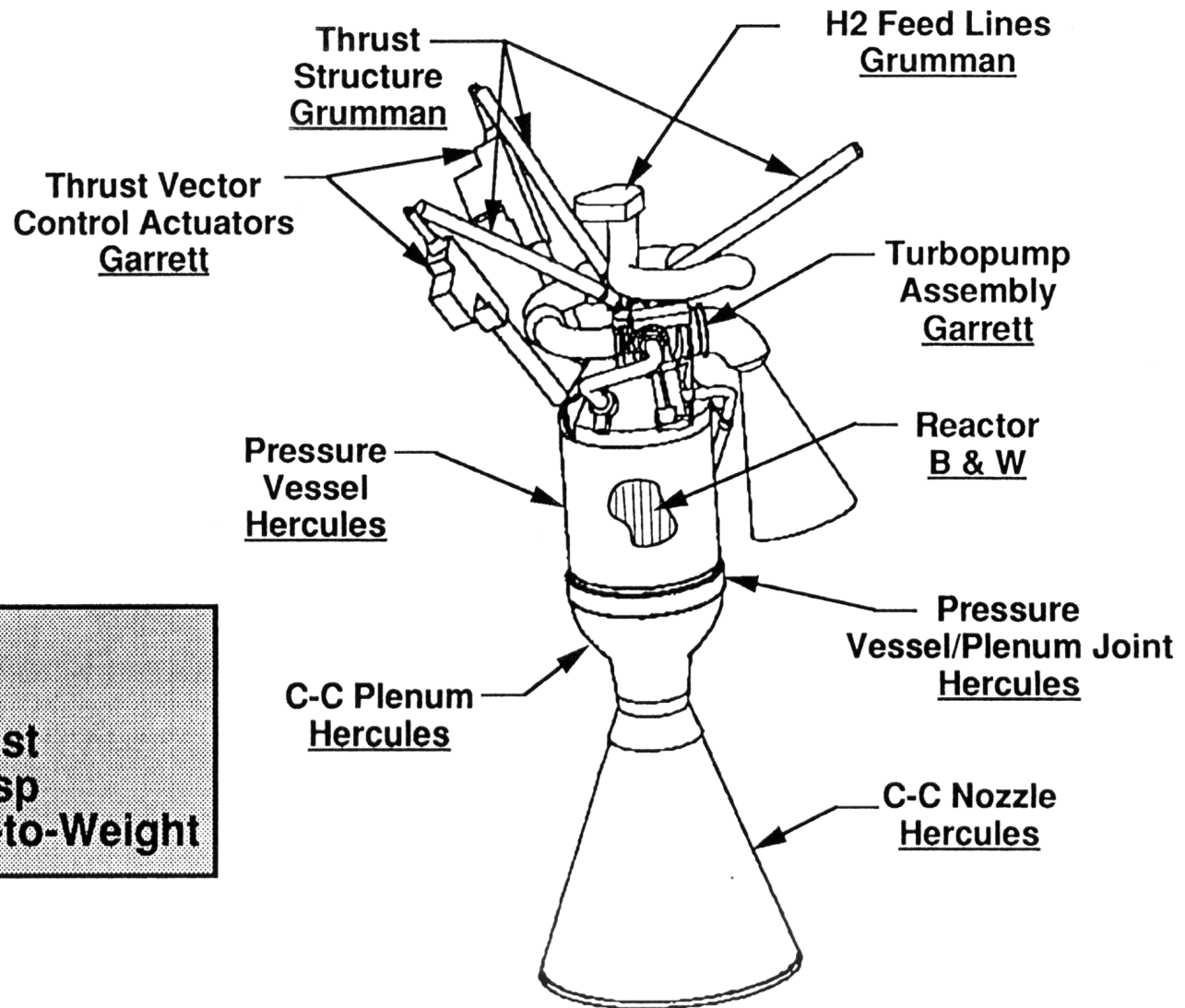


# Program Philosophy

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- Select the technology: Particle Bed Reactor
- Ascertain deficiencies: In the PBR case, almost everything, but greatest discrepancy was lack of proven NTP fuel
- Keep the team small and focused at all times
- Design rapid, hard-hitting tests to determine shortfalls and success/failure routes
- Use the safety/EIS process to guide project direction
  - Define objectives clearly
  - Don't change direction
  - Follow the process
- Make that decision: When you have 450 people charging the case, you don't want them marching in place
- Don't belabor successes – concentrate on problems that the technical team cannot solve

# SNTP PBR Engine Description



## • Design Goals:

- 20 - 80 klbf Thrust
- 900 - 1000 Sec Isp
- 25 - 35:1 Thrust-to-Weight

Typical Engine Height  $\approx$  10 Ft (40 klbf Class)

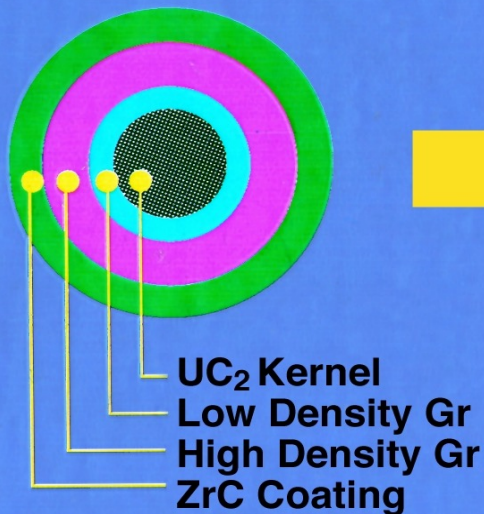
SPACE NUCLEAR THERMAL PROPULSION



# Particle Bed Reactor

## Enabling Technology—The Particle Bed Reactor

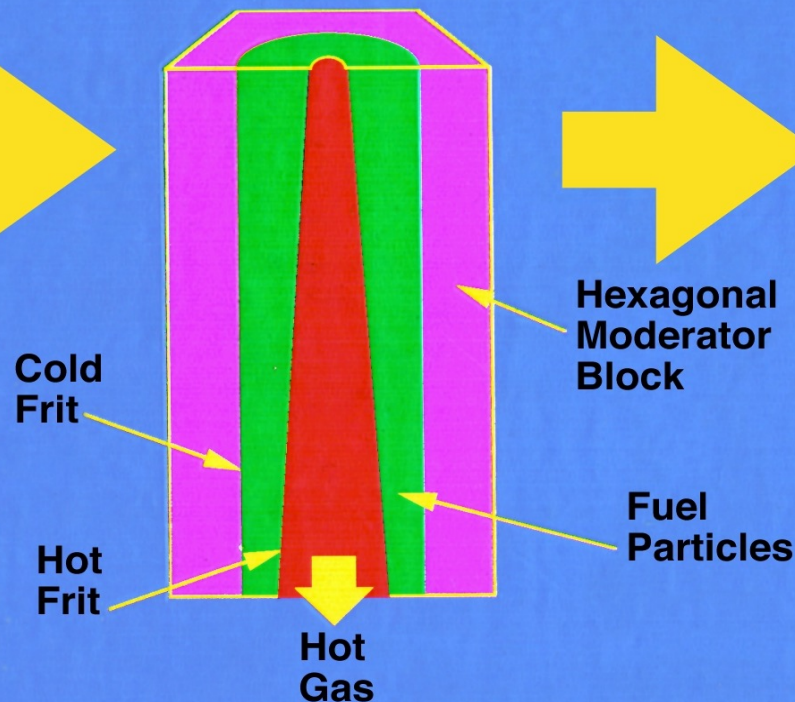
### Fuel Particle



#### Features:

- 400 $\mu$  diameter
- Melting Point  $\approx$ 3300 K
- Retains Fission Products

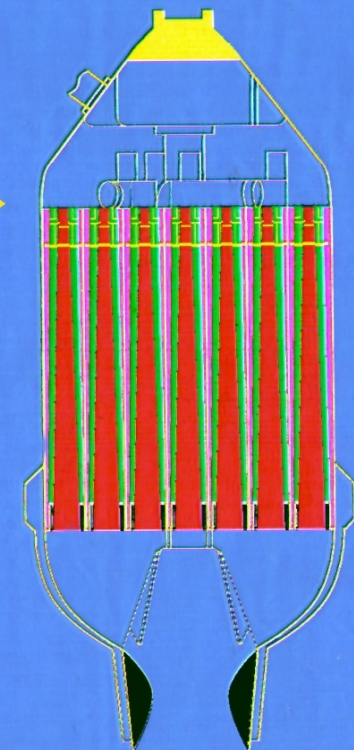
### Fuel Element



#### Features:

- Low Gas/Particle  $\Delta t$
- $\therefore$  Low Thermal Shock
- Gas Heated Directly
- High Power Density

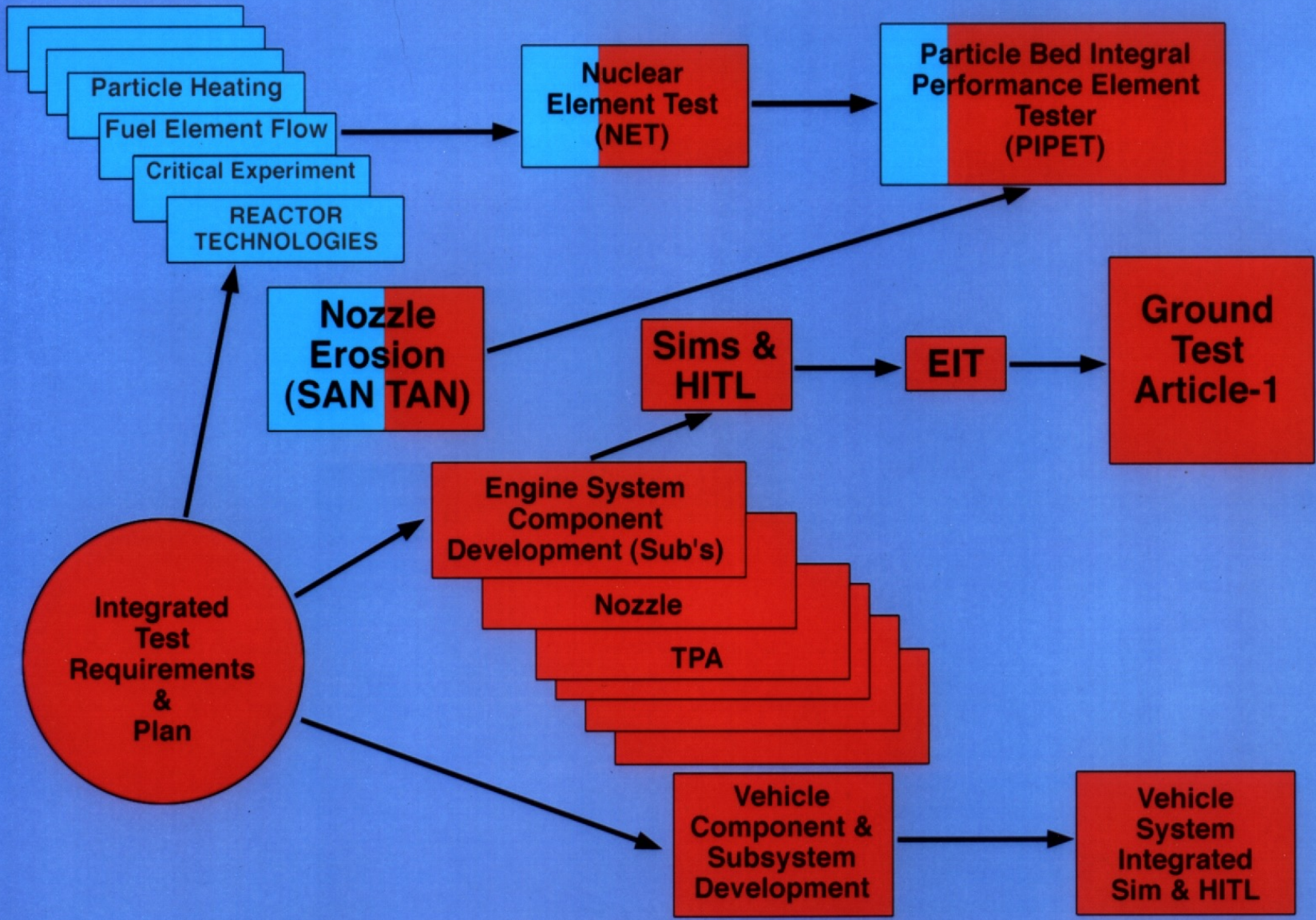
### Reactor



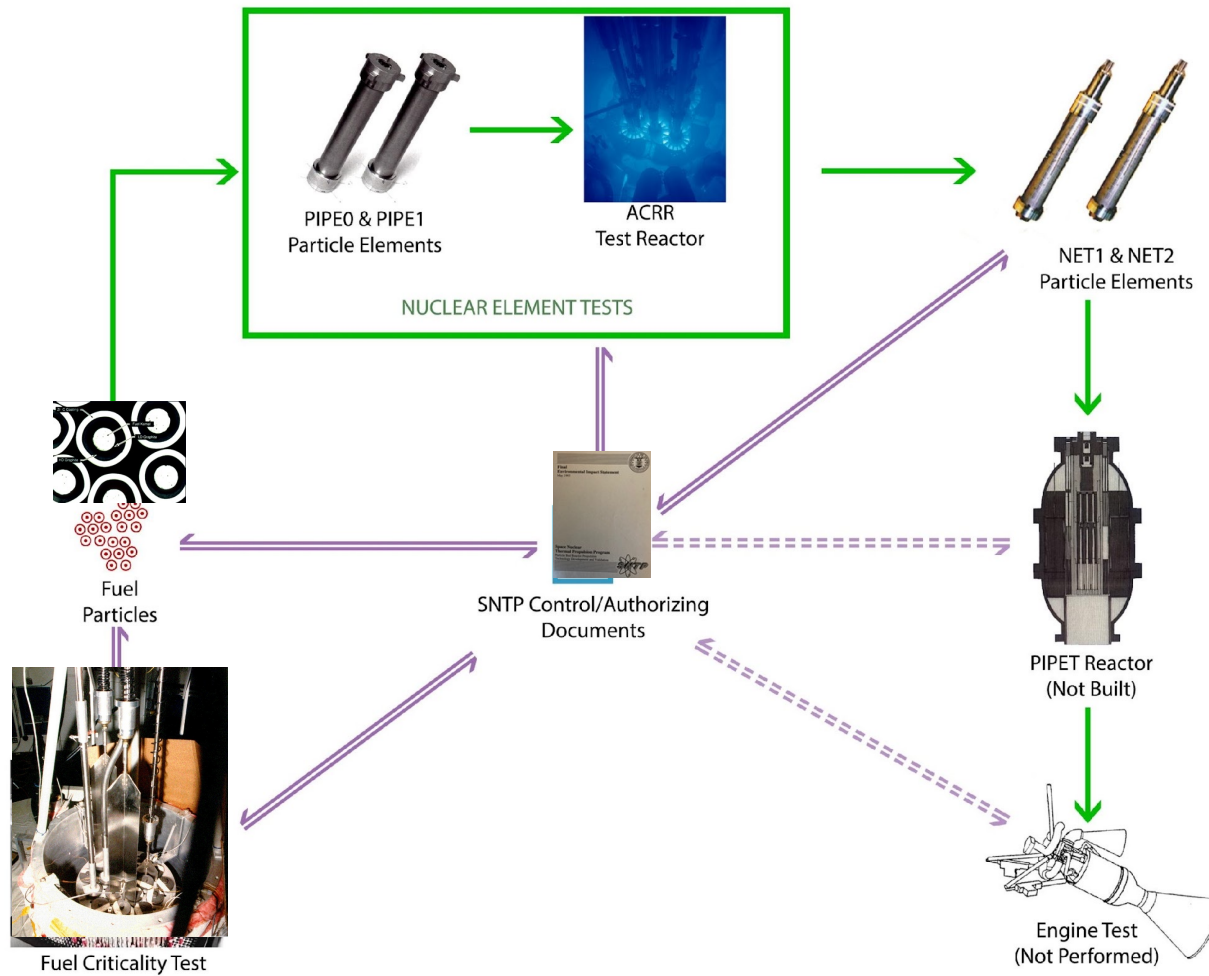
#### Features:

- Very Compact
- Low Pressure Drop
- Fast Start (<10 Sec)

# Integrated Test Plan



# Hardware Development Focused Around EIS and SARs



Prepared by A. X. Lenard

# Some Observations

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- The SNTP program had many firsts, developed substantial new hardware – focused on testing
- The critical experiment was of major importance both perceptually and in the determination of core performance
- CTE mismatches become extremely important when you have high temperature gradients – these are unavoidable with higher power densities
- New fuels can achieve NTP temperature requirements
  - UN appears to be a good selection for several reasons-works with LEU
  - Reticulated foams offer high power densities and an immobile geometry (no bed settling/reconfiguration during start stop cycles)
  - UNbZrC fuel (FSU development) is good fuel, but only with HEU
- New MCNP and hydrocodes will assist greatly in flattening power profiles

# Other Programmatic Accomplishments

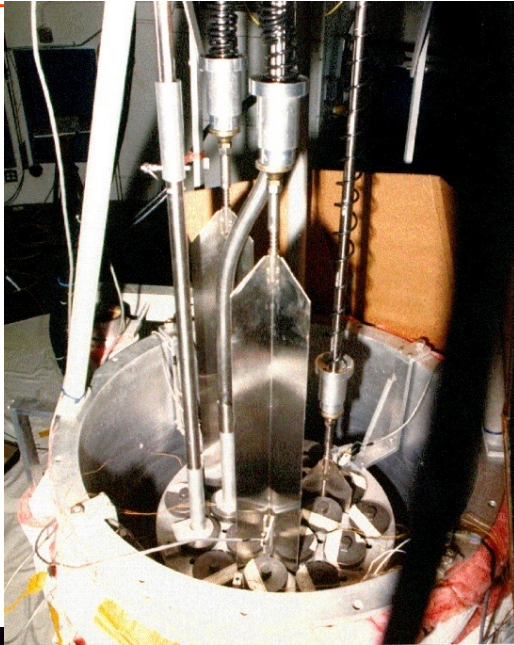
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- Program was reviewed by a DOD, SNTP specific, Defense Science Board (DSB), and briefed two other DSB's
- Program received a written commendation from the DSB for progress made for funds expended
- *Designed, licensed and began testing a new reactor (critical reactor) in less than 18 months.* First new reactor licensed in over 10 years
- Developed first ever EIS for conducting ground tests of nuclear thermal rocket (scrubbed exhaust)
- Performed complete process-Preliminary Draft EIS, Draft EIS, and Final EIS – Public Hearings
- Performed task for 1/35<sup>th</sup> the projected cost and 28% of the forecasted time

# Supporting Slides

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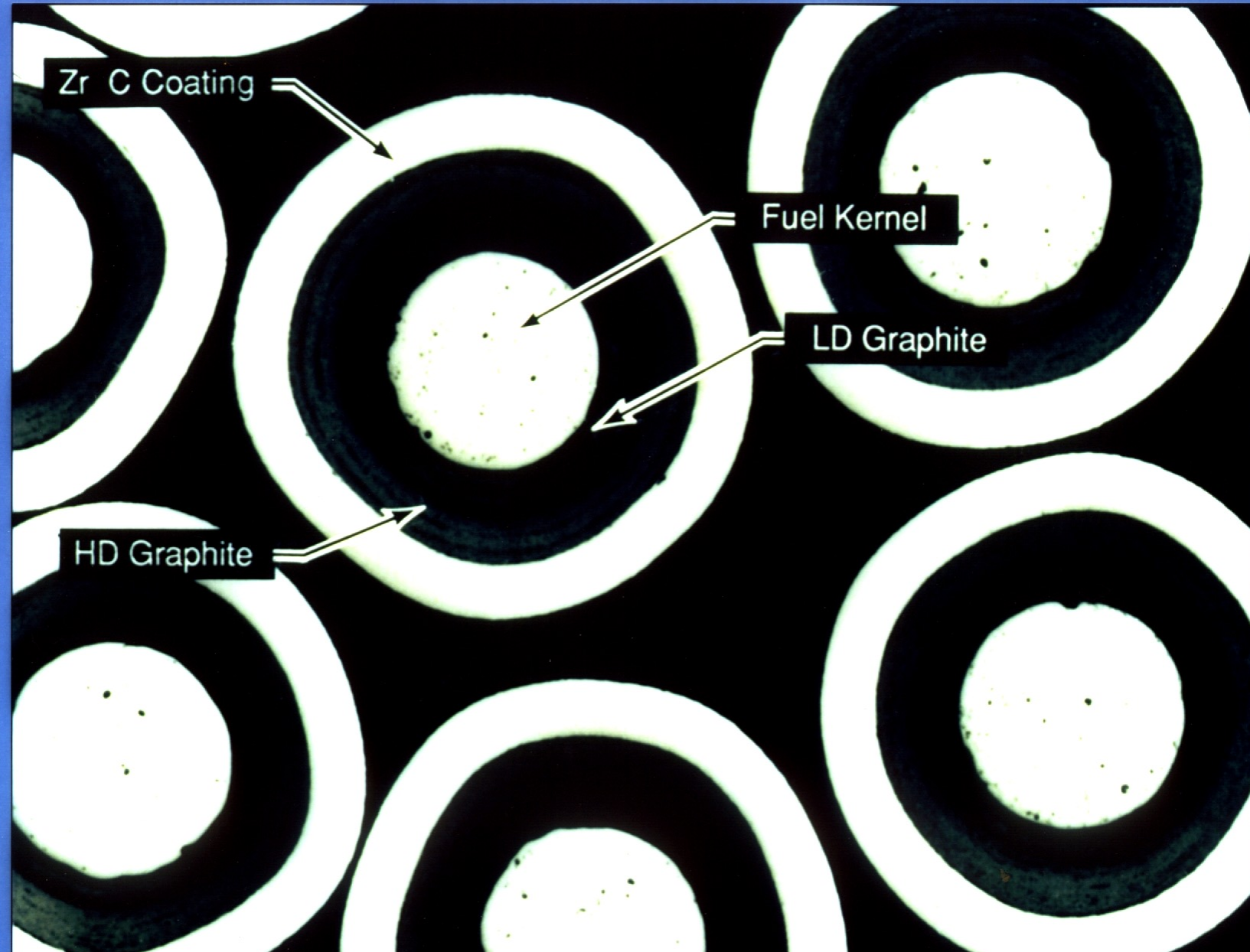
# SNTP PBR Critical Assembly



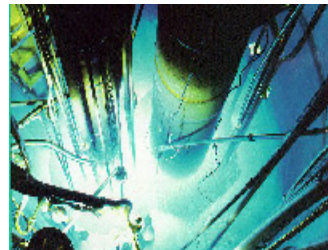
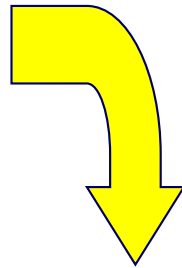
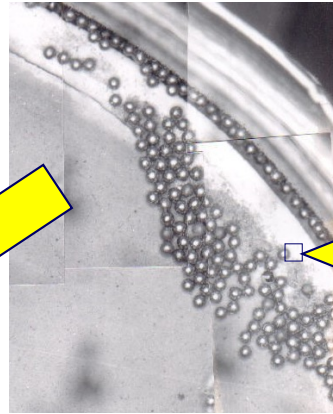
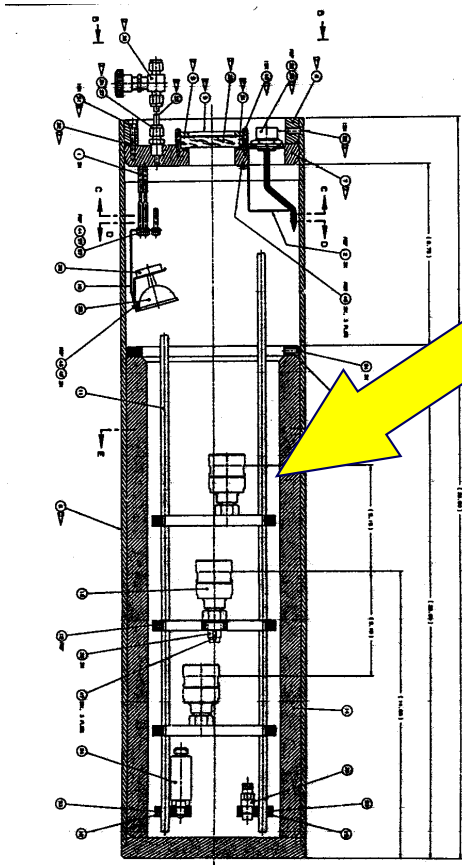
- Prototypic PBR assembly with fuel stalks
- Blended fuel to simulate ZrC coated TRISO
- Full test site SAR and independent review
- 18 months from start to first test
- \$19M for assembly, fuel and safety reviews
- \$6-7M for subsequent tests
- Tested at Sandia in SPR facility
- Test tank used for several other critical Assemblies – in use now



# Prototypic Fuel Particles



# In-Core Particle Testing



# PNT Accomplishments and Results

## Accomplishments

4 Capsule Tests

12 Fuel Holders

200,000 Fuel Particle

1800 - 3100K

150 - 600 Seconds

1 - 4 Cycles

Fission Product  
Release Measure

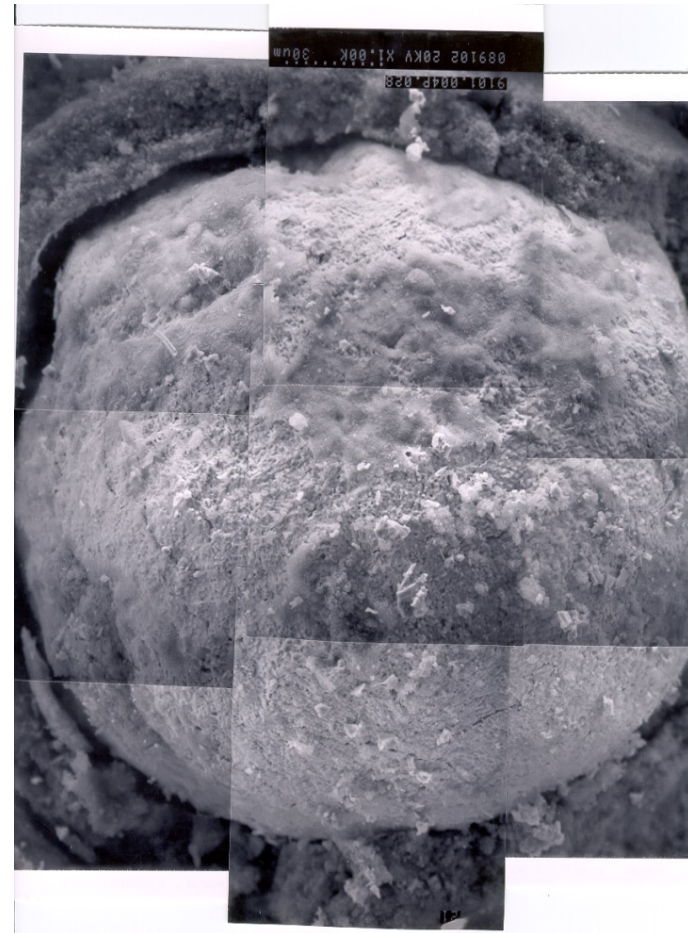
## Results:

Performance Limits Determined For Baseline Fuel  
Integrated Fission Product Release Model Operational

# Lessons Learned

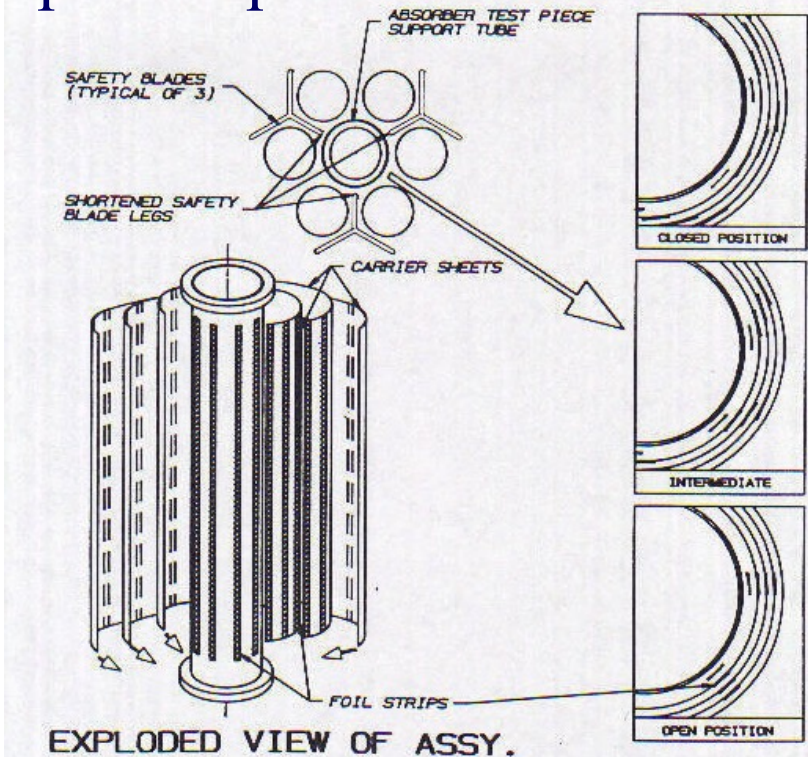
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- If you don't have the fuel ready-you don't have bupkis
- Fission product retention will be exceptionally difficult in any material at 3000K, a reasoned discussion needs to occur to ascertain the virtues and liabilities of it as a requirement
- There are significant differences between in-core and ex-core testing; internal versus external heating
- Process knowledge and control is exceptionally important, you need to know what you did wrong as well as right
- *Toward the end of the program, we had a fuel particle that would operate satisfactorily for the original mission duration*

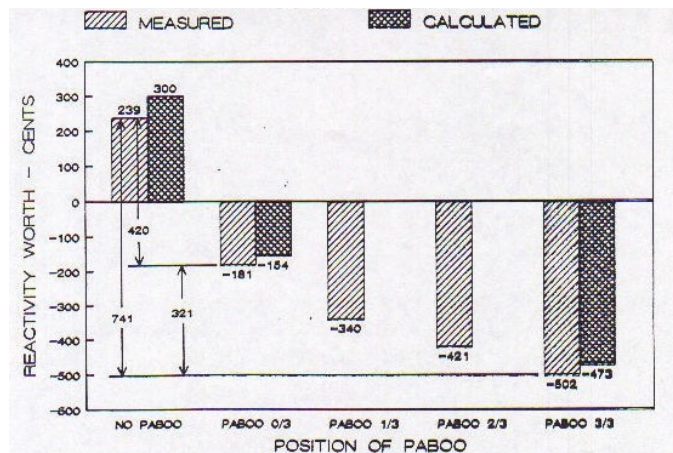


# Some Significant Results

## A Rapid Response Internal Controller Series 1: 10/23/89-10/27/89

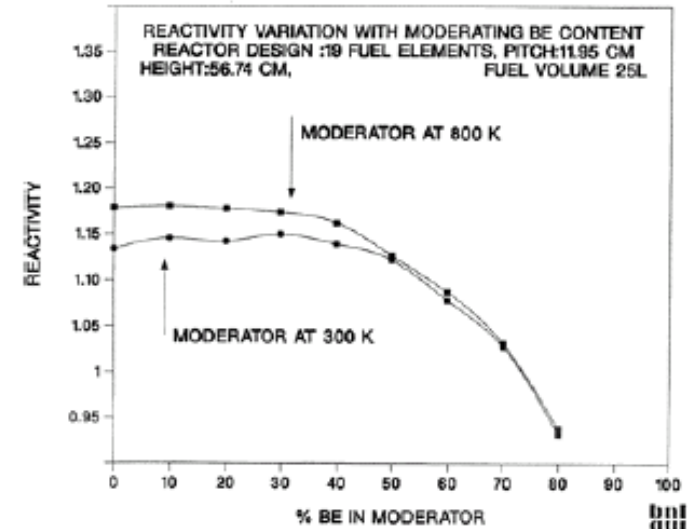


- 1<sup>st</sup> critical, safety and control
- Reactivity worth; boron worth
- Series 2: 12/11/89-12/15/89
  - Pulse neutron reactivity
  - Moderator temperature coefficient 10-20C
  - Flux monitor tests
- Series 3: 1/15/90-1/19/90
  - Peek-a-Boo Experiment
  - Poly-filled void reactivity worth
- Series 4: 10/4/90-10/12/90
  - Fission density/power calibration
- Series 5: 10/29/90-11/2/90
  - Stalk swap
  - Moderator temperature coefficient 20-80C
- Series 6: 3/91
  - Warm rho
- Series 7: 4/91
  - Hot rho



# Lessons Learned

- Rapid prototyping is extremely important – we got a new reactor running
- Exceptional team-building experiment
- Tremendous external appeal, particularly to review and safety boards
- It doesn't need to take a long time
- It doesn't need to cost a lot
- You'd be surprised at the new things you learn by doing real hardware



SNTP CX

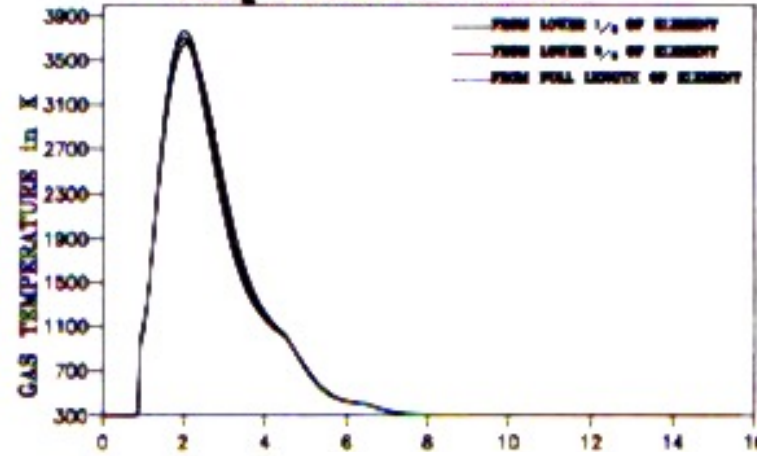
What do you mean it  
Has a positive temperature  
Coefficient ##\$%^&\*\*()!!!

But it's only over  
a narrow range

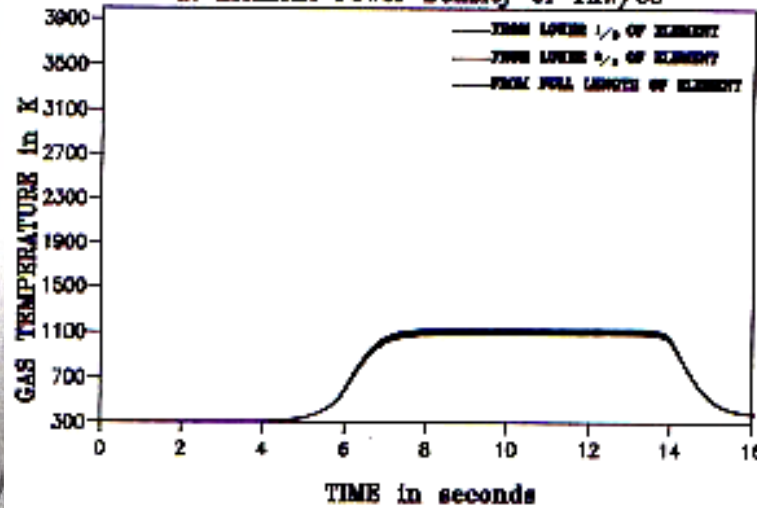
# In-Core Element Testing



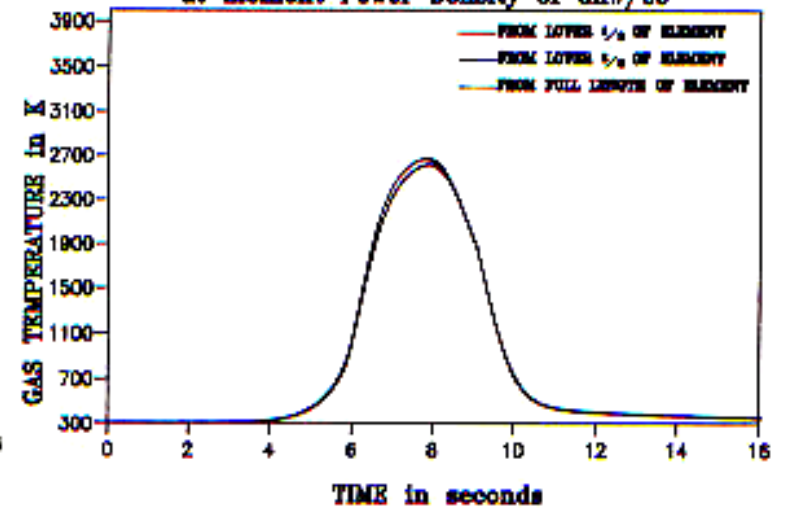
PREDICTED ELEMENT COOLANT TEMPERATURES during Accidental Rod Withdraw



PREDICTED ELEMENT COOLANT TEMPERATURES at Element Power Density of 1KW/cc



PREDICTED ELEMENT COOLANT TEMPERATURES at Element Power Density of 3KW/cc



# SNTP Progress in High Temperature Resistant Coatings

## HOT FRIT DEVELOPMENT

**1990 Goals**

- 2750 K
- 500 sec.
- 5 cycles
- No Fuel/Frit Interactions

**1992 Goals**

- >3000K
- > 2000 sec.
- > 10 cycles
- NoFuel/Frit Interactions
- Minimum FP Leakage

Development of Different Coatings/Process Options on Hot Frit Samples

Testing in High Temperature (3000+ K) Hydrogen (1 to 70 atm)

Coating Materials  
TaC, NbC, ZrC

Coating Process  
CVD & CVR

Hot Frits  
Graphite & C/C

**RESULTS**

- 1992 Goals Appear Achievable
- Variety of Options Possible
- Best-to-Date Performance TaC on Graphite (>2hour life)

Development of Different Coatings on Particles Using CVR & CVD Processes

Development of New Coating Process (Liquid Bismuth Carrier)

Fabricate Baseline & Surrogate Particles with ZrC & NbC & TaC Coatings

Test in High Temperature Hydrogen (up to 3200 K) both Nuclear & non-Nuclear

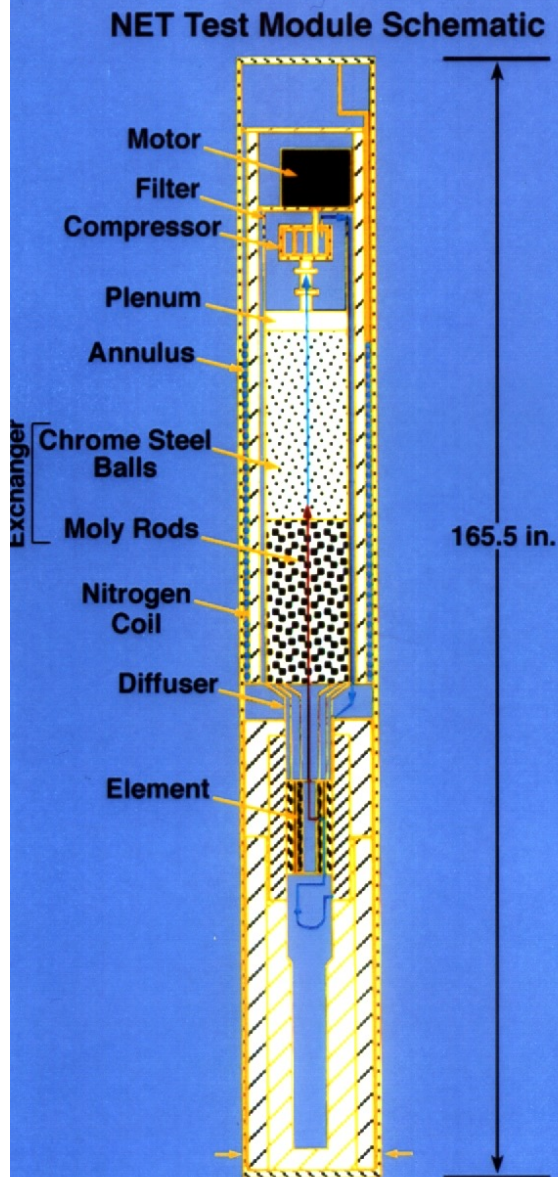
**RESULTS**

- Baseline Particles Limited by Kernel to ~2600 K
- Coatings for 3500 K Appear Achievable

## PARTICLE DEVELOPMENT

# Nuclear Element Test (NET)

*Closed loop, in reactor test of a complete fuel element in cryogenic hydrogen*



## GOALS:

- Demonstrate integration of fuel element technologies
- Test to full temperature capability
- Validate fuel element designs
- Support
  - PIPET/GTA development
  - Fuel development
  - Model verification
  - Safety analysis

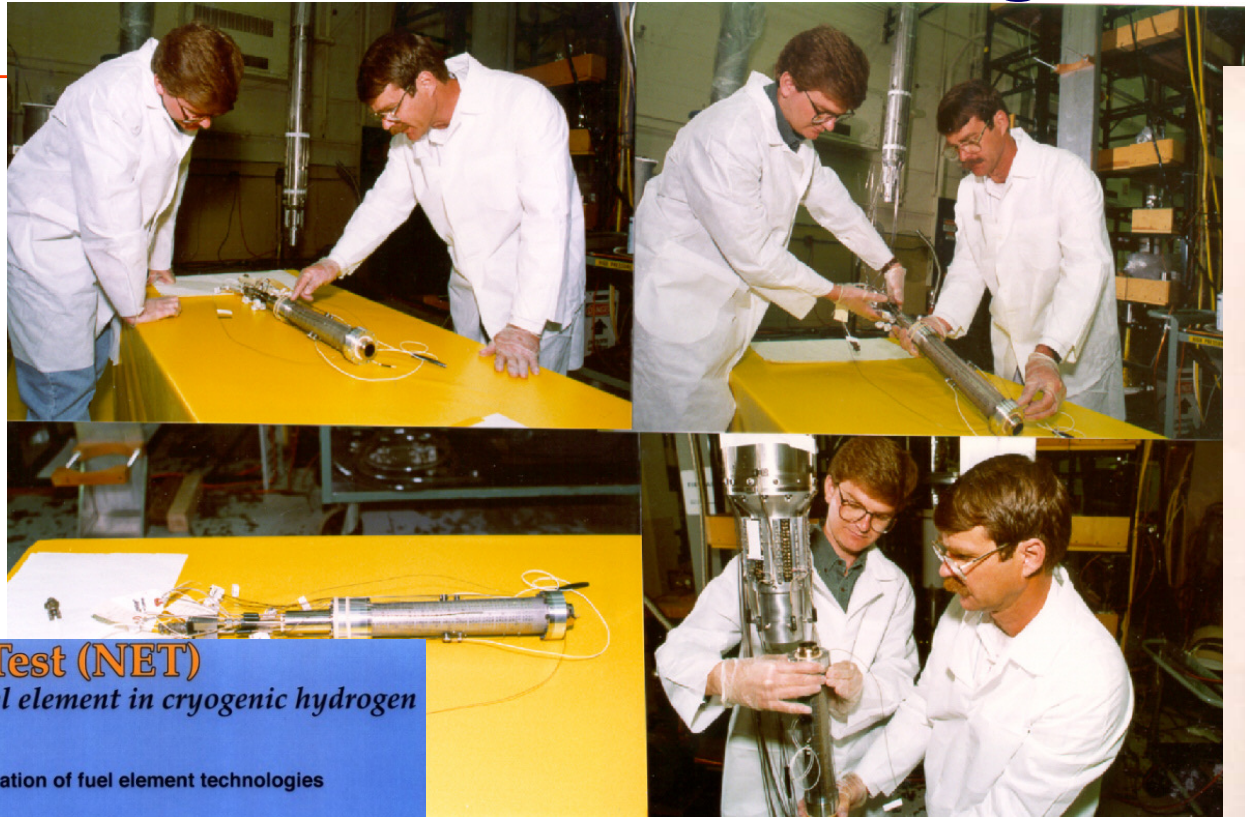
## ACCOMPLISHMENTS:

- Test hardware designed and fabricated
- Unfueled experiment assembly (NET-0) tested to cryogenic temperature and flow performance characterized in helium
- Test reactor (ACRR) control capabilities demonstrated

## STATUS:

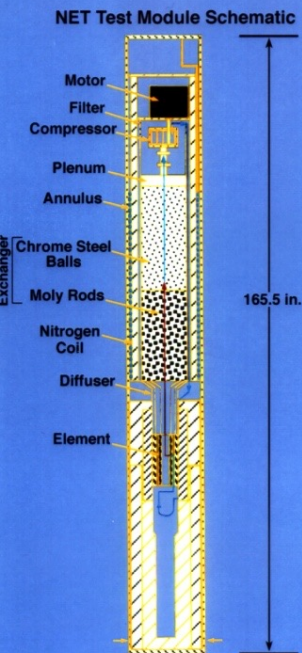
- NET-0 testing in cryogenic hydrogen summer 1992
- First fueled test (NET-1) early CY 1993

# In-core Fuel Element Testing



## Nuclear Element Test (NET)

*Closed loop, in reactor test of a complete fuel element in cryogenic hydrogen*



### GOALS:

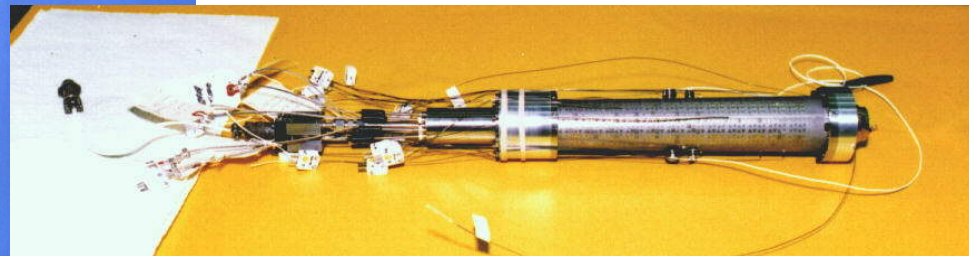
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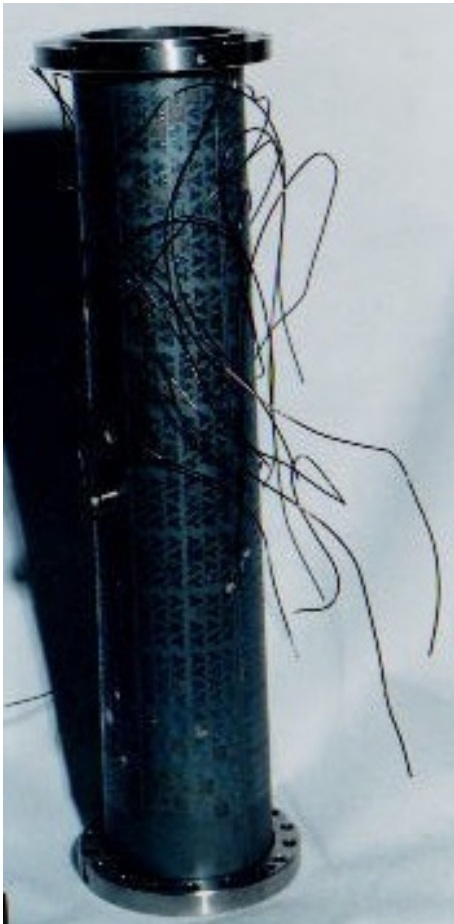
### STATUS:

- NET-0 testing in cryogenic hydrogen summer 1992
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# Post-test Hardware

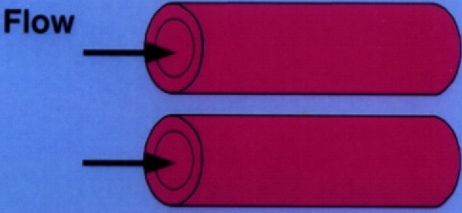
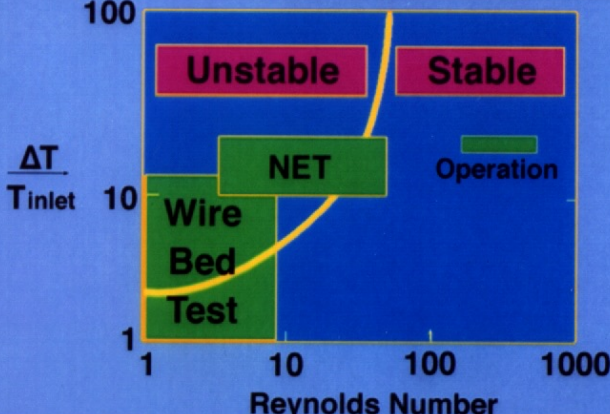

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## Post-Irradiation Evaluation

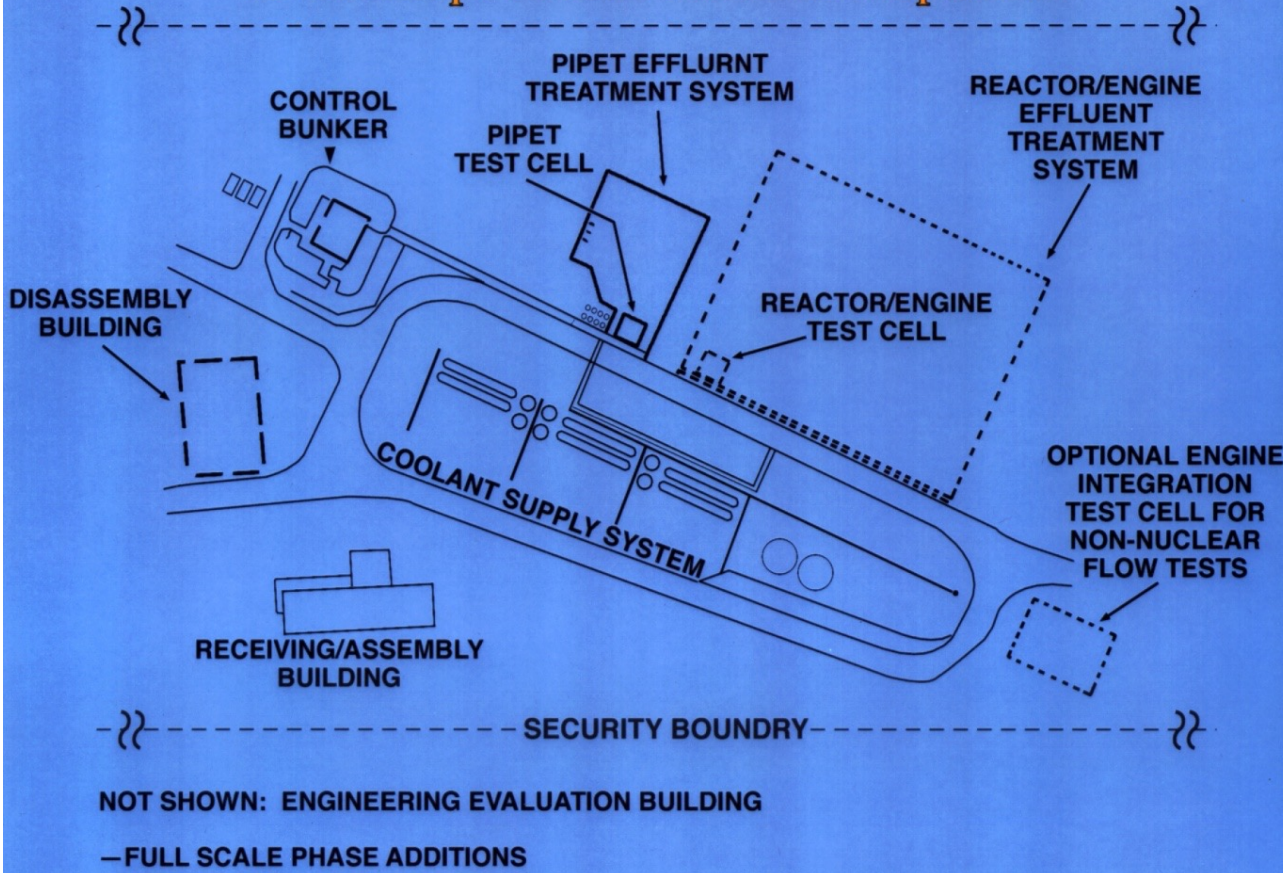
- The hot frit stress fractured
  - Un relieved CTE mismatch
  - Error in engineering design
- Design fixes were identified
  - Termination of many (Clinton) SDI space programs prevented further hardware development
- Nuclear portions of program were decommissioned

# Thermal Hydraulic Design Validation Process

Task	Approach	Results
<ul style="list-style-type: none"> <li>Scoping Analysis (Conservative 1-D)</li> </ul>	<ul style="list-style-type: none"> <li>Model Flow through parallel heated tubes</li> </ul> 	
<ul style="list-style-type: none"> <li>2D &amp; 3D Analysis</li> </ul>	<ul style="list-style-type: none"> <li>SIMBED code (Bnl)</li> <li>TEMPEST code 9PNL/B&amp;W)</li> <li>FLOW 2D code (SNL)</li> <li>PBRFLOW code (Grumman)</li> </ul>	<ul style="list-style-type: none"> <li>Flow through bed is stable</li> <li>Must prevent back conduction to cold frit</li> </ul>
<ul style="list-style-type: none"> <li>Modify Cold Frit Design</li> </ul>	<ul style="list-style-type: none"> <li>Insulate platelet frit from particle bed</li> </ul>	
<ul style="list-style-type: none"> <li>Benchmark Codes</li> </ul>	<ul style="list-style-type: none"> <li>Electrically Heated Wire Bed Test</li> <li>NET series</li> <li>PNT-3 series</li> <li>Special Effects Tests</li> </ul>	<ul style="list-style-type: none"> <li>To be in June, 1992</li> <li>NET start at end of 1992</li> <li>Ongoing</li> <li>Ongoing</li> </ul>
<ul style="list-style-type: none"> <li>Code Validation</li> </ul>	<ul style="list-style-type: none"> <li>NET - Low Flow</li> <li>PIPET - All prototypical conditions</li> </ul>	<ul style="list-style-type: none"> <li>NET - End of 1992</li> <li>PIPET - Approx. 1995</li> </ul>

# Completed Title-I Design EIS PBR Test Facility

**A New Ground Test Complex is Needed to Develop Nuclear Thermal Propulsion**



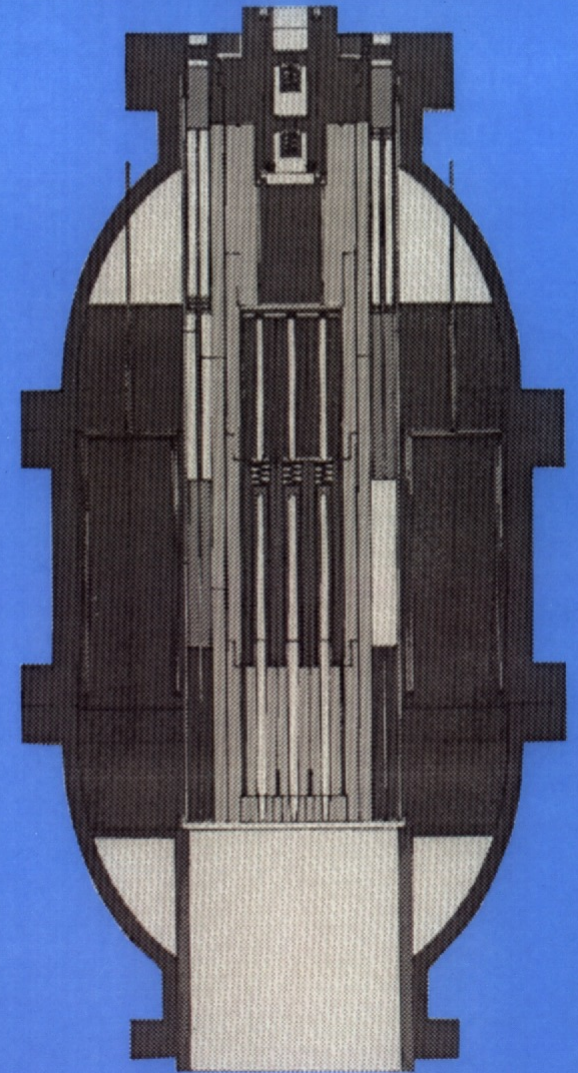
## Saddle Mountain Test Station

- Phase I 550 MW<sub>th</sub> PIPET
  - 2000-3000 s run times
  - ~ 50 MW/l power density
  - Confined test; numerous filters and scrubbers modest release fractions
- Phase II 2 GW<sub>th</sub> Engines
  - 1000 s run times
  - 2-week intervals
  - 75,000 lb<sub>f</sub> thrust
  - Confined test
- Completed Ground Test EIS and all supporting Documents
- Completed all public hearings
- Safety Analysis Reports
  - \$200M Phase I Test
- Phase II Facility Cost: TBD

# PIPET

## Fuel Element and Test Reactor Objectives

- **Demonstrate Fuel Element Performance and Operability Under Prototypical Conditions.**
- **May be Used as a Test Bed to Help Resolve Other SNTP Program Technology Issues**
- **Support for Testing of Alternative Fuel Concepts**



# Summary

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- Design – build – test: The only method to have a good final concept
- A tested fuel form is the foundation of an NTR – without it you have no program
- SNTP accomplished multiple hardware tests that provided clarity to many unknowns
- A ground-test EIS completed - major programmatic milestone
- May have available facility to test nuclear rocket engine

# POST SNTP Advances

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- Since SNTP, there have been some technical advances that NTR development easier
- But, the physics for a NTR have not changed
- Operational multi-use NTR does not require the very high power densities of the Timberwind concept-reduced risk
- Lessons-learned fostered some changes
  - High Assay Low Enriched Uranium (HALEU)
- Building an NTR is easier than before