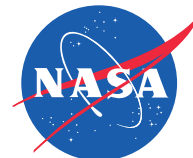


Predicting Maximum Temperatures of a Li-ion Battery on a Simulated Flight Profile using a Model-based Prognostics

Mohit Mehta (KBR, Inc.), Michael Khasin (NASA Research Center), Chetan Kulkarni (KBR, Inc.), and John Lawson (NASA Research Center)

Funded By

Presentation date: 11/17/2022



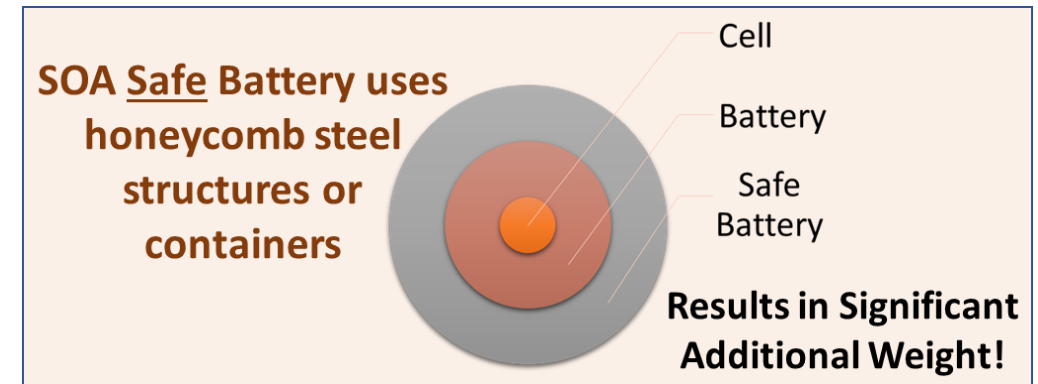
NASA Aeronautics Research Mission
Directorate (ARMD) Convergent Aeronautics
Solutions (CAS) Project, SPARRCI sub-project.

- ❖ Introduction
- ❖ Prognostics Architecture
- ❖ Results
- ❖ Challenges and Future Work

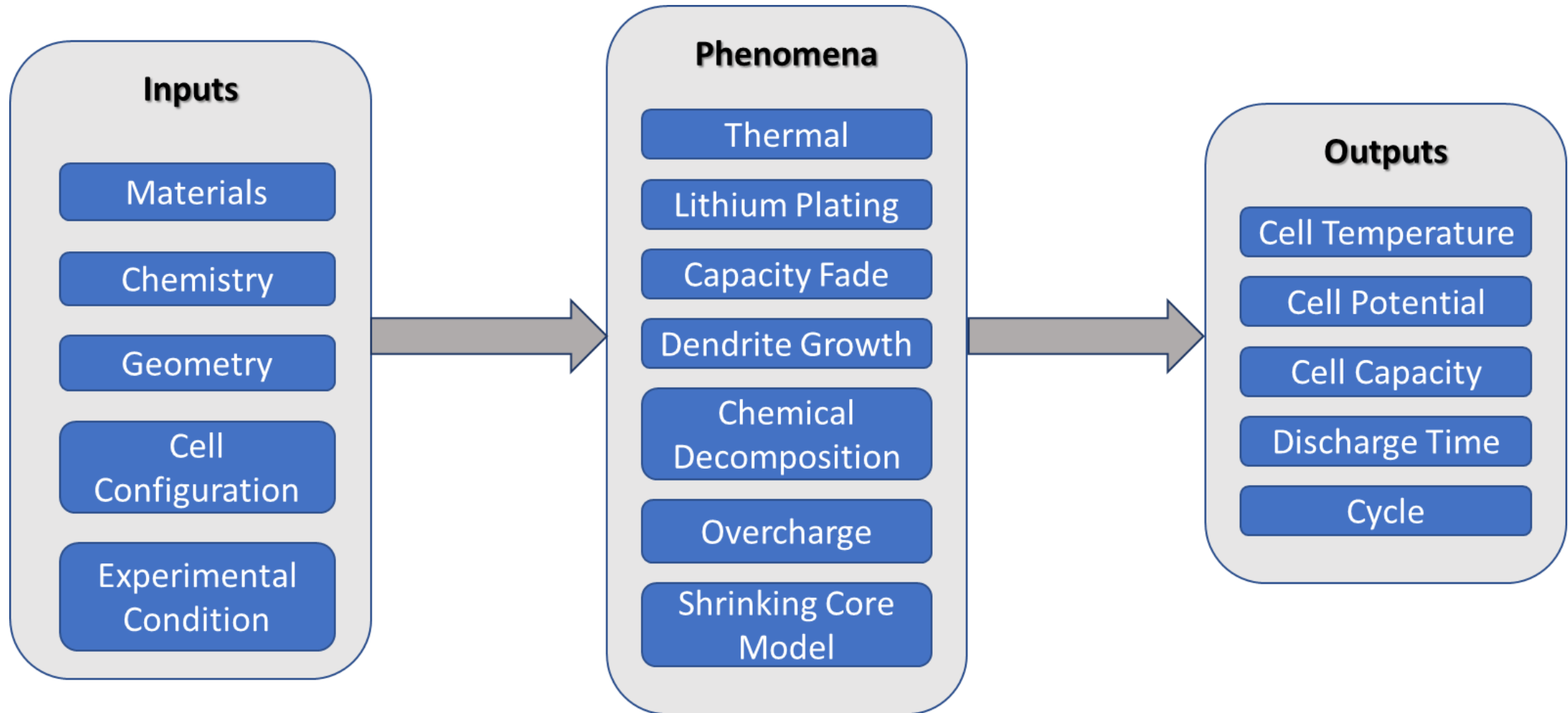
SPARRCI to Improve Specific Energy for a Safe Electric Aircraft Battery

Electric Aircraft need Better and Safer Batteries

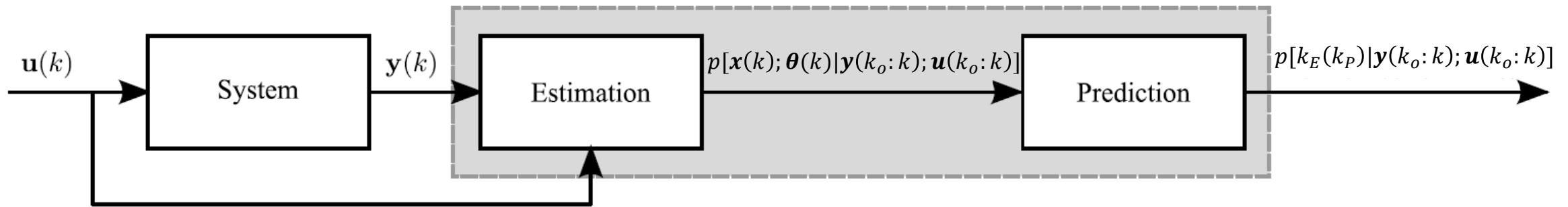
- **Consequence of Unmitigated Cell Thermal Runaway Events**
 - Fire
 - Explosion
 - Debris
- **Current Solution Results in Low Specific Energy and Specific Power for a Current Li-ion Battery**
- **Alternative Solution is reducing the non-battery chemistry mass with the development of:**
 - Better Internal Battery Monitoring Tools
 - Developing internal fault detection & mitigation strategies



Overview for Complexity in Modeling Nominal Cell Behavior



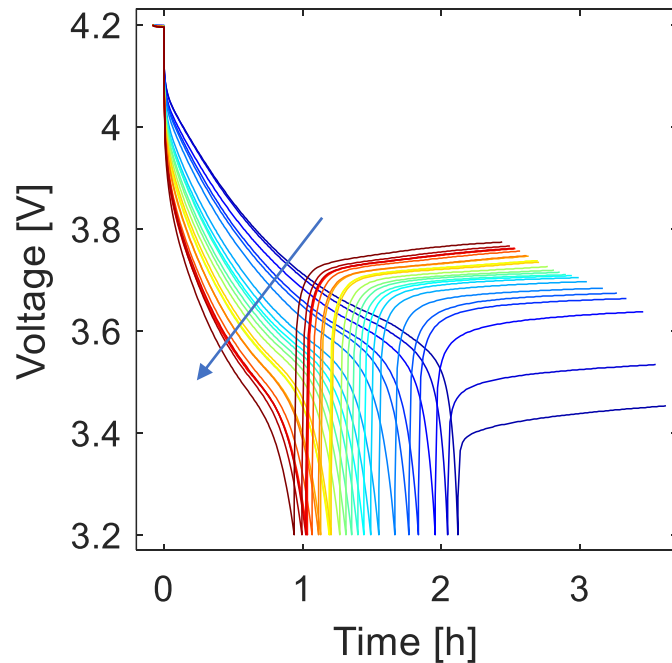
Simplified Prognostics Architecture



We use two such architectures 1: For Predicting Properties T, V, and EOD

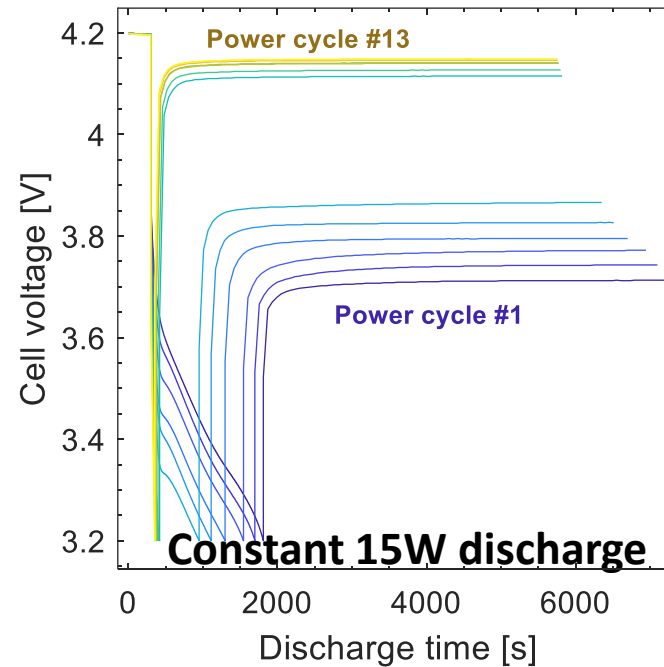
2: For Predicting Aging Parameters and hence, T and EOL

Reference Discharge Cycle



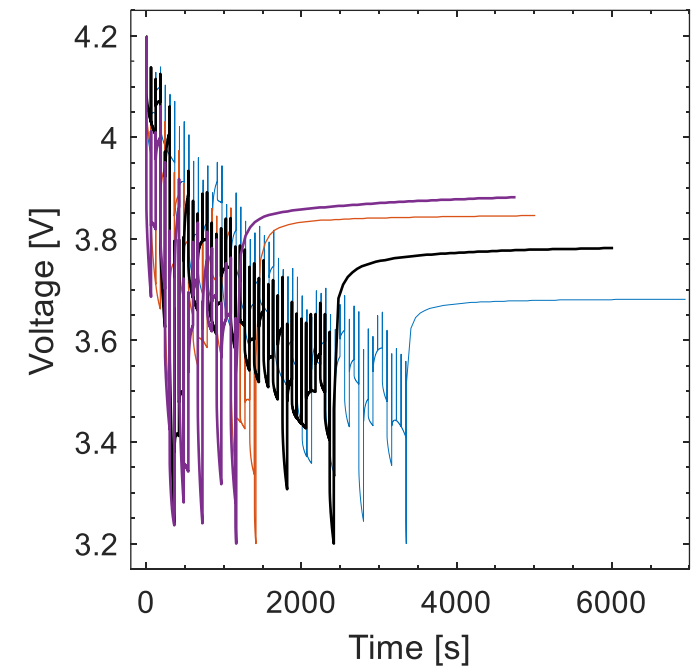
22 Ref. cycles with decreasing SOH

Power Discharge Cycle



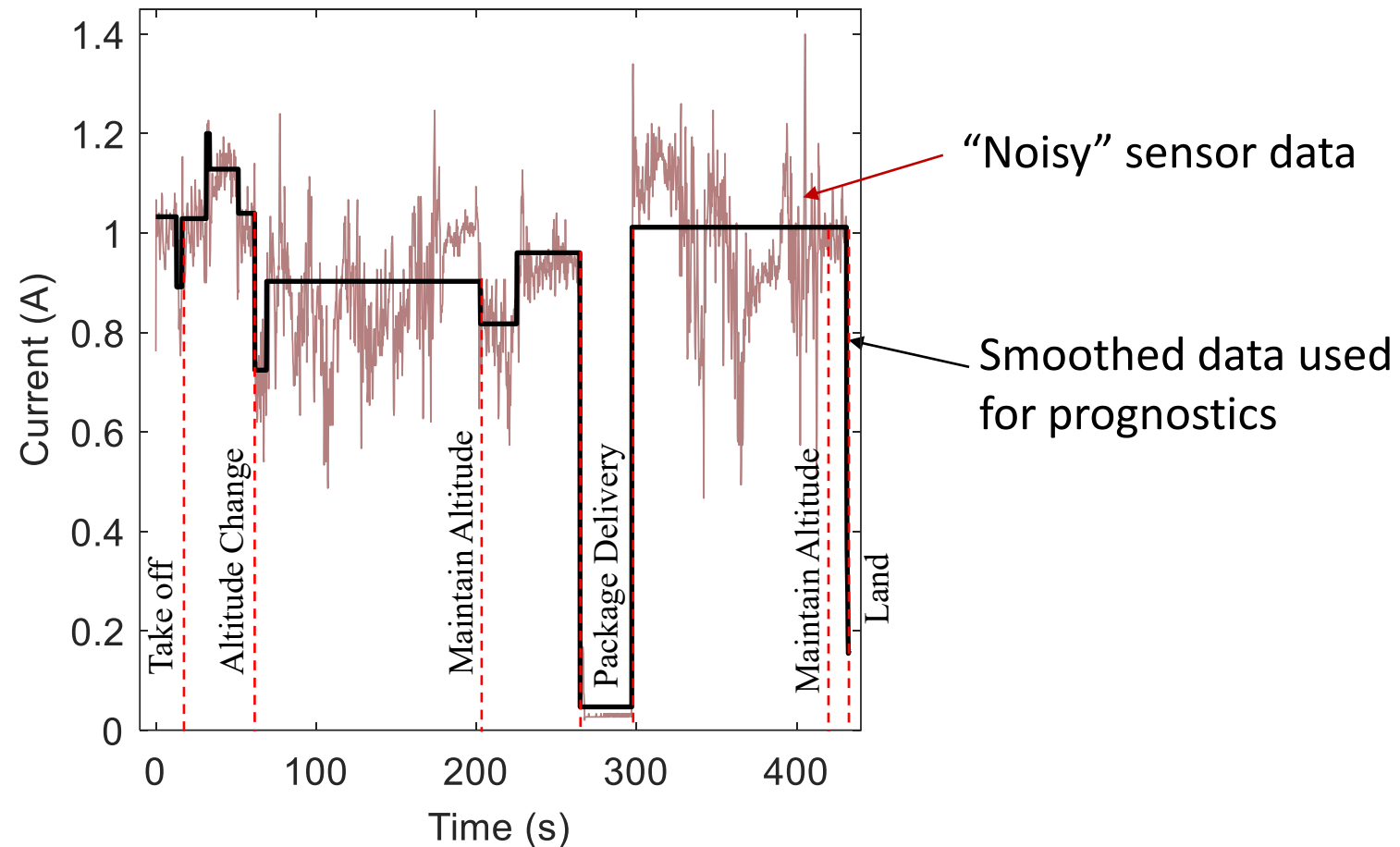
A Power cycle is measured after each Ref. cycle

Random Discharge Cycle



Approximate 50 RW cycles between any two Ref. cycles

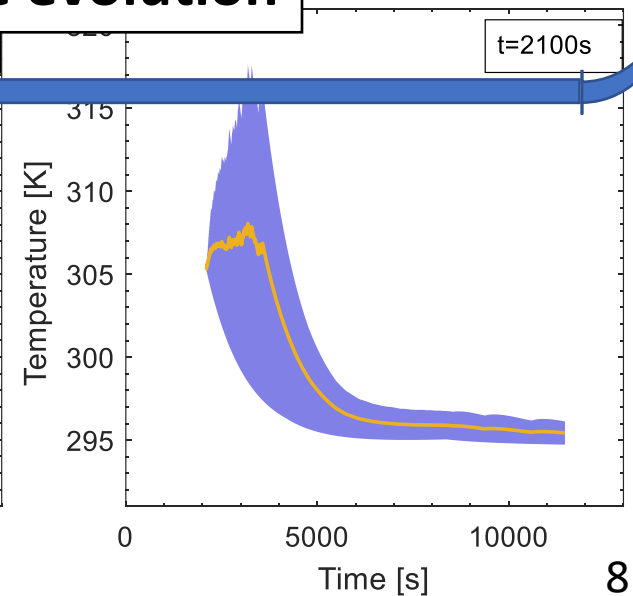
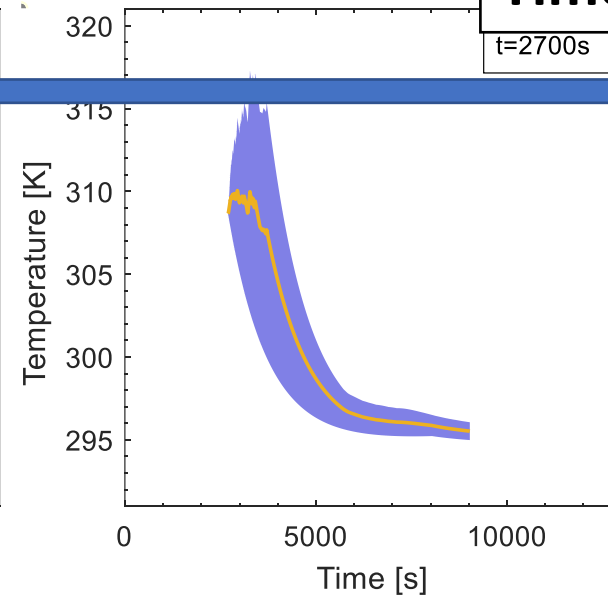
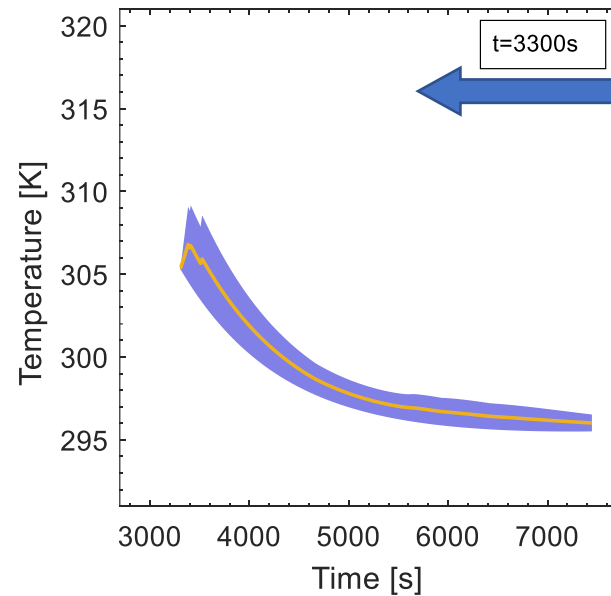
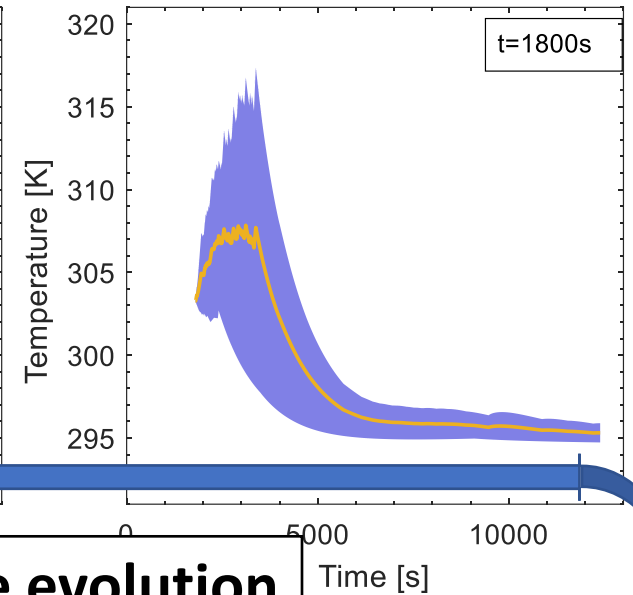
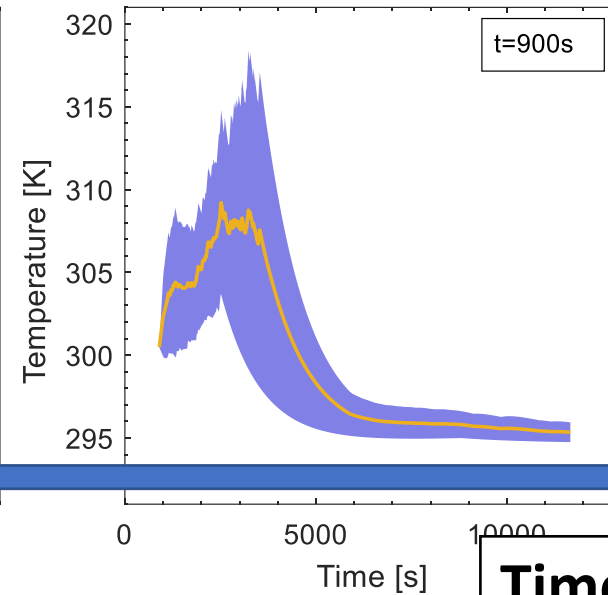
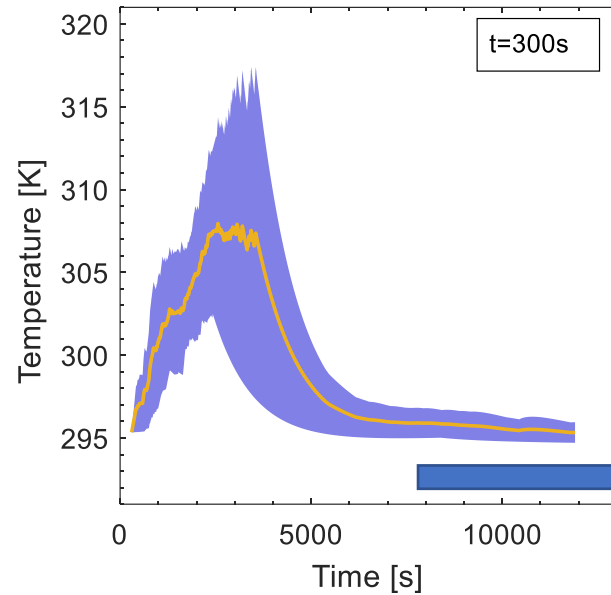
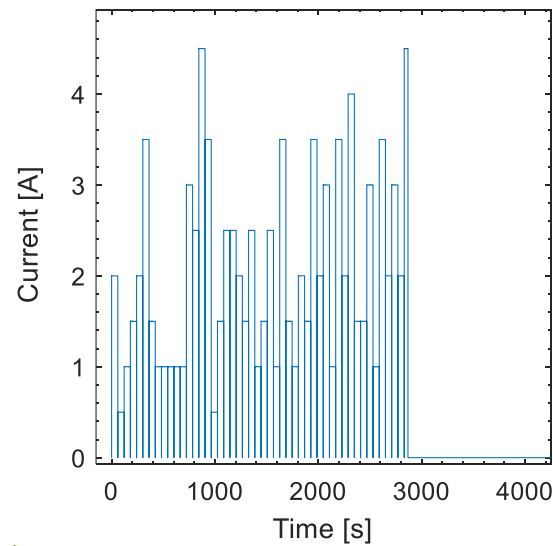
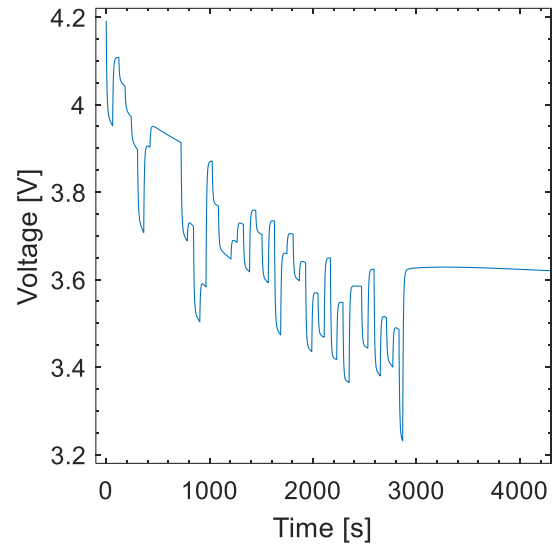
Data from a Short Test Flight



Data pre-processing is needed for accurate estimation

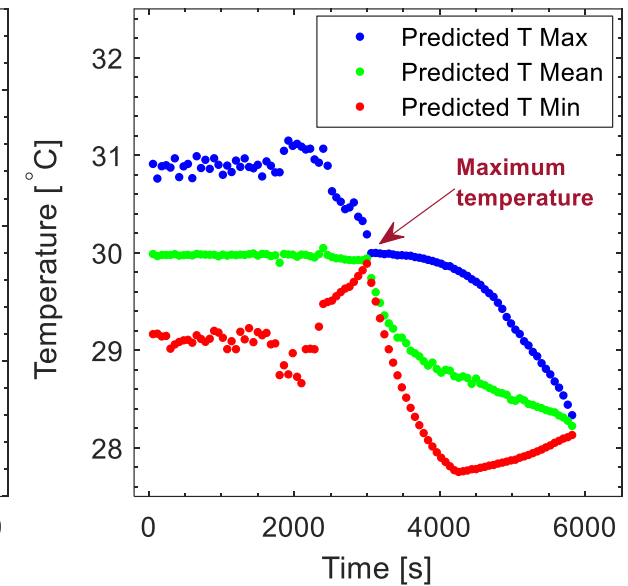
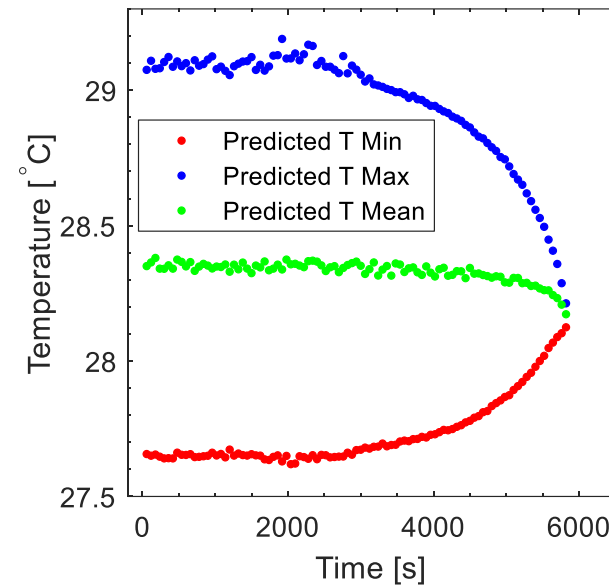
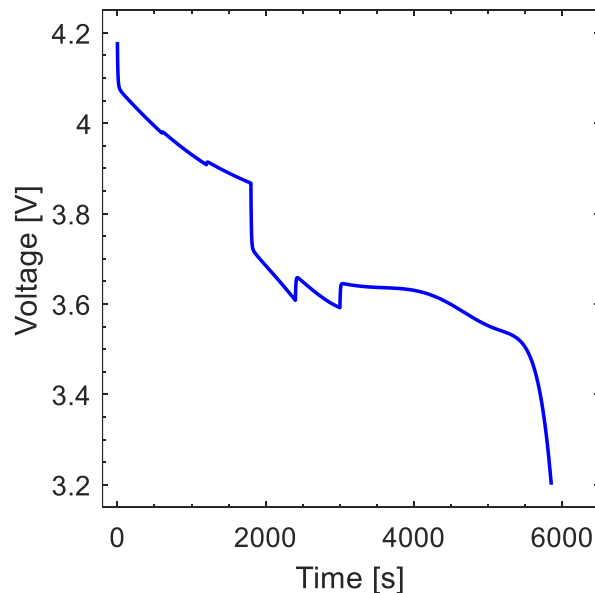
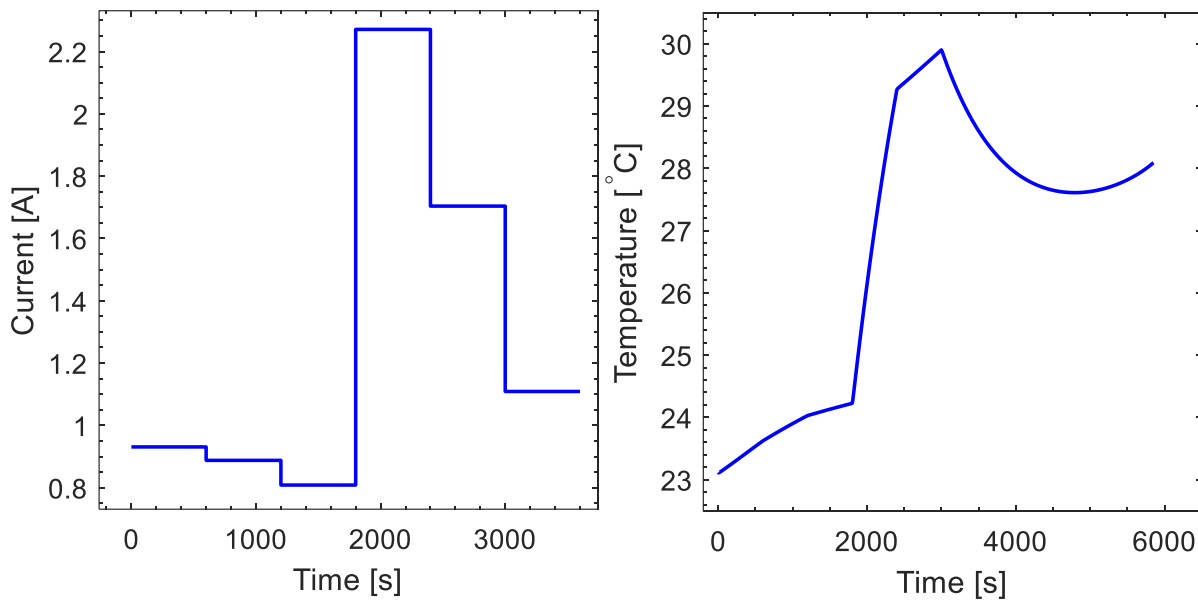
Random Walk is used as a substitute for Simulated Flight Profile

Prognostics on SFP



Time evolution

Two Temperature Metrics for SFP

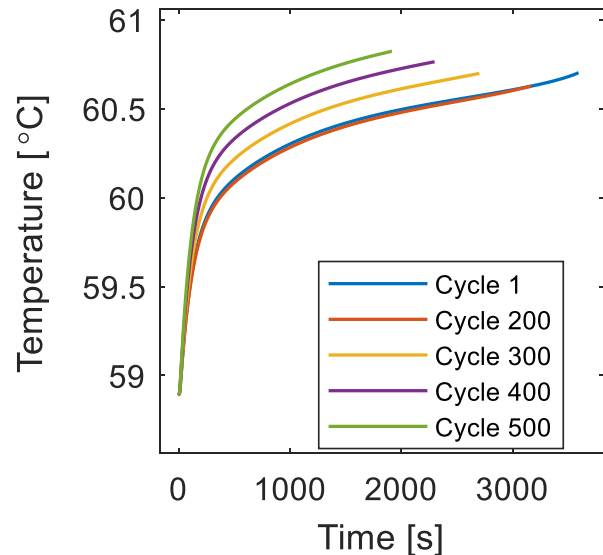
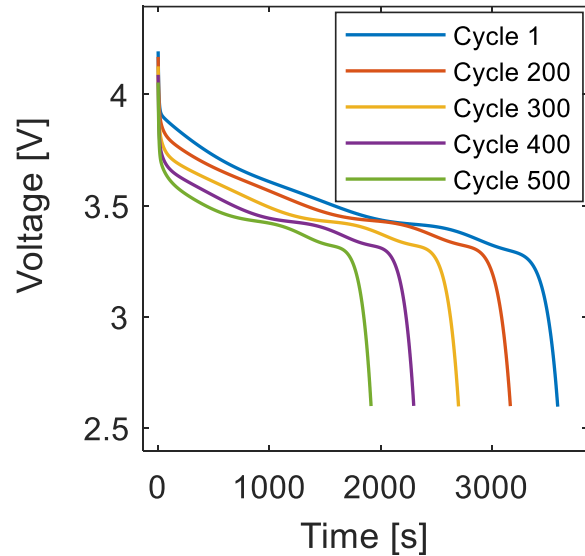


Predicting EOF
Temperature

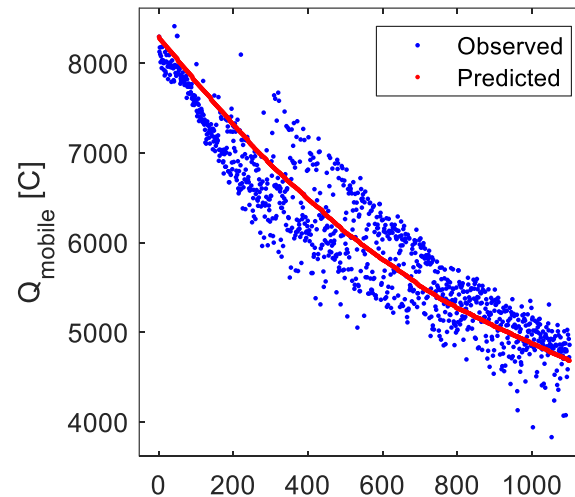
Predicting Maximum
Temperature

Empirical Aging Model

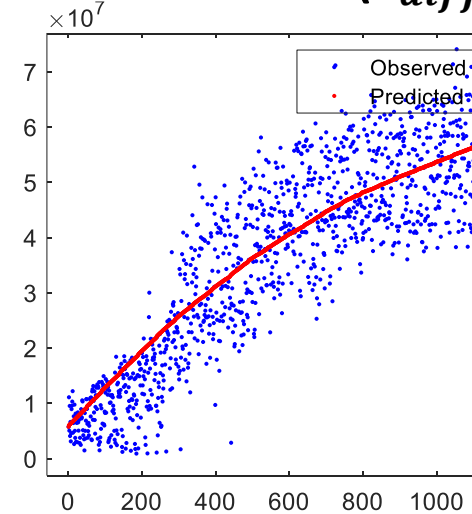
- Parameters are estimated using Simulated Flight discharge cycles
- C-rate ranges from 0.2-2.2C
- Each Simulated Flight Profile is stopped after V_{\min} is reached



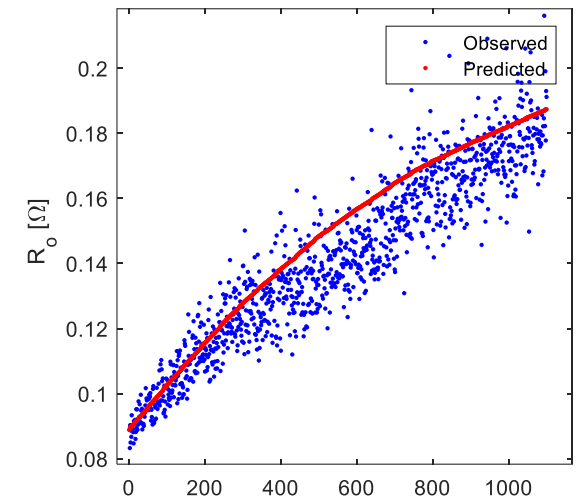
Usable Li-ion (Q_{mobile})



Diffusion time (τ_{diff})

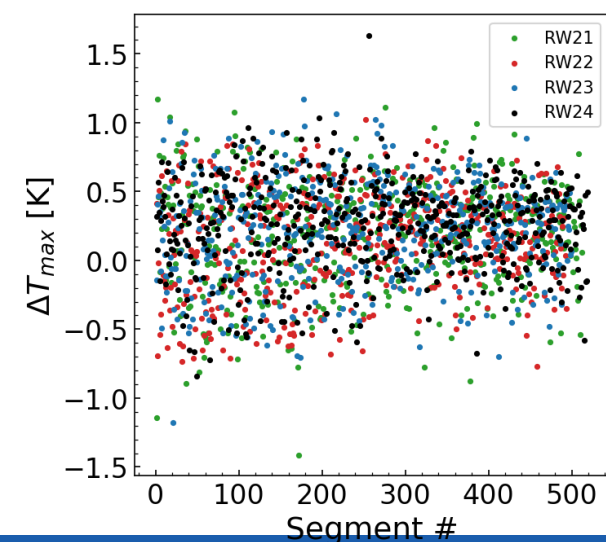
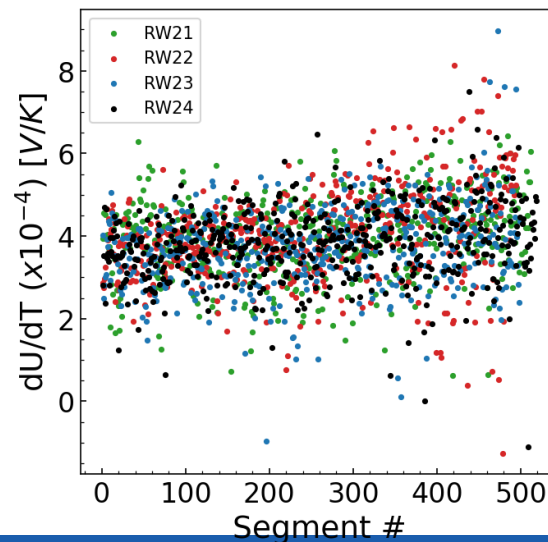
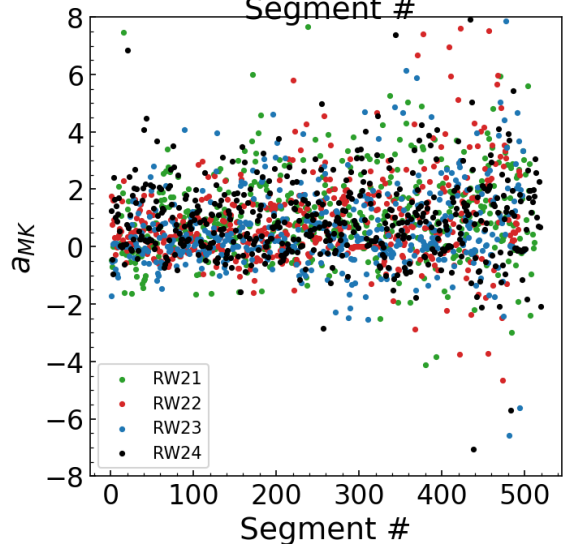
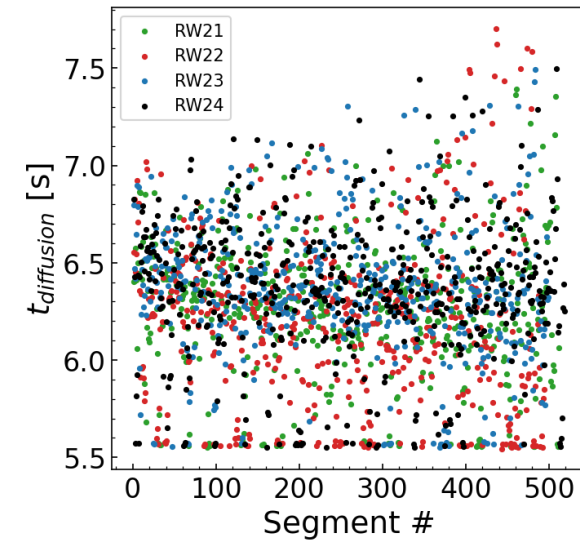
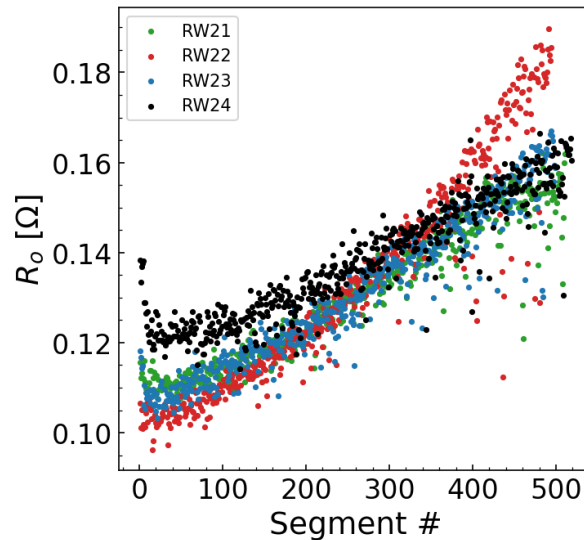
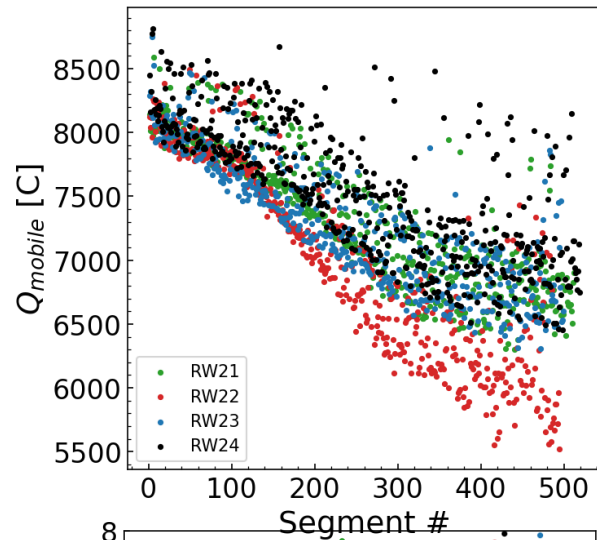


Ohmic Resistance (R_o)



Temperature increases qualitatively on voltage parameterized data

Challenge in Aging Related Temperature Prediction



Voltage Decoupled Thermal Models

$$\frac{dT}{dt} = I(t) C_b^{-1} \left(U - T \frac{dU}{dT} - V(t) \right) - \frac{T - T_a}{\tau}$$

Decoupled voltage
(sensor input or model output)

$$T(t) = T_a + [T(0) - T_a] e^{-t/\tau} + C_b^{-1} \int_0^t I(t') [V_0 - V(t')] e^{\frac{t'-t}{\tau}} dt'$$

ROM

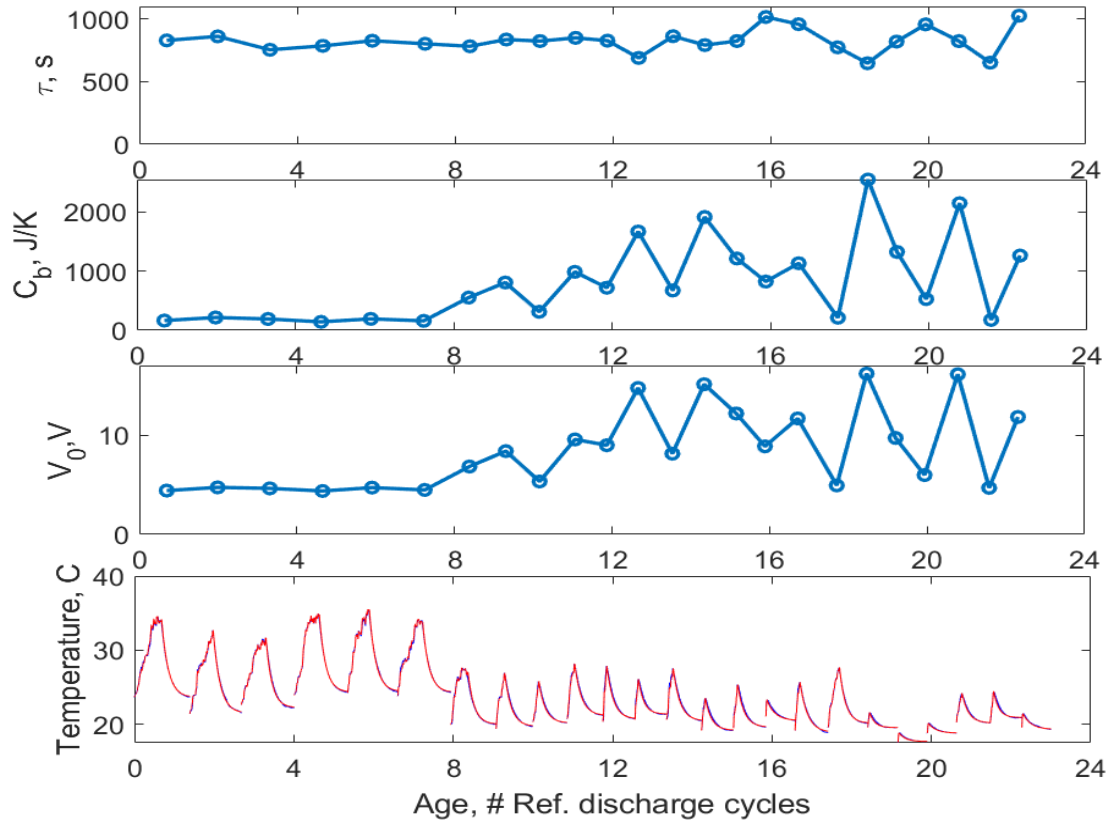
For realistic flight profiles with varying
values of the discharge currents

$$T(t) = T_a + [T(0) - T_a] e^{-t/\tau} + \frac{V_0}{C_b} \int_0^t I(t') e^{\frac{t'-t}{\tau}} dt'$$

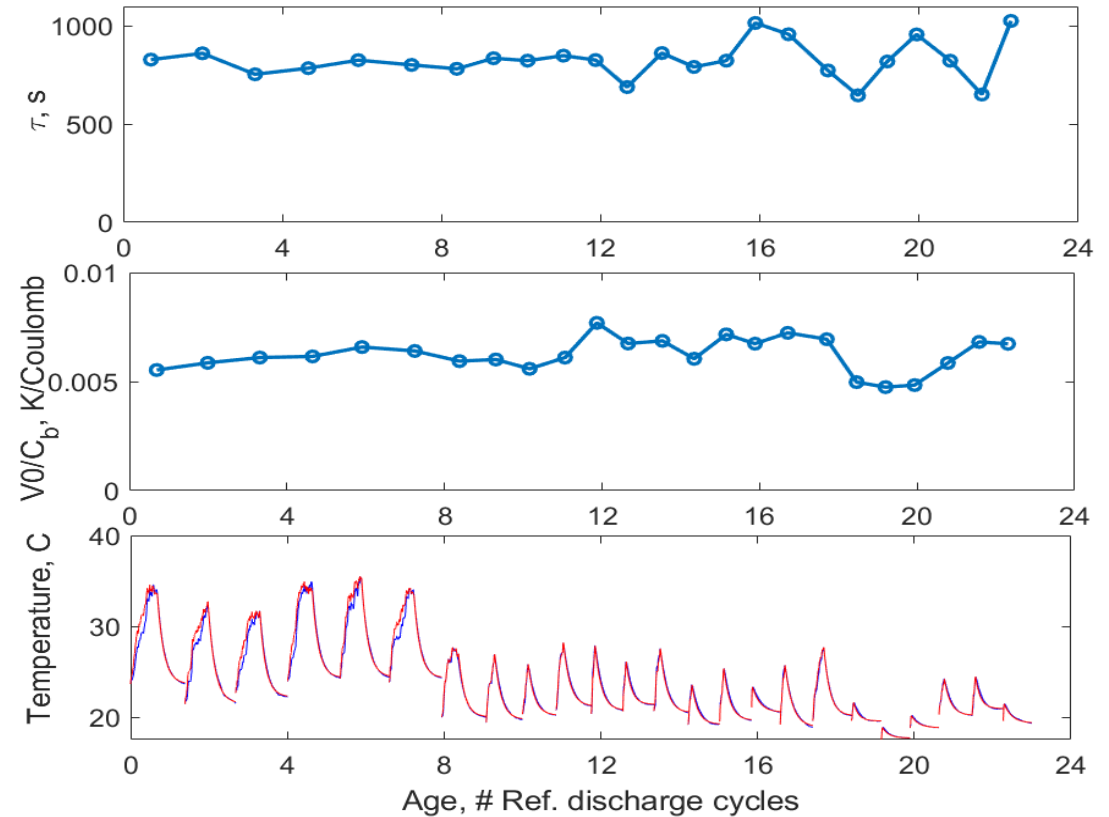
Overcompensation can be tested by decoupling voltage
and temperature parameter estimation

Fitting TM and ROM to SFP data

TM: $T(t) = T_a + [T(0) - T_a]e^{-t/\tau} + C_b^{-1} \int_0^t I(t')[V_0 - V(t')]e^{\frac{t'-t}{\tau}} dt' \rightarrow$ **ROM:** $T(t) = T_a + [T(0) - T_a]e^{-t/\tau} + \frac{V_0}{C_b} \int_0^t I(t')e^{\frac{t'-t}{\tau}} dt'$



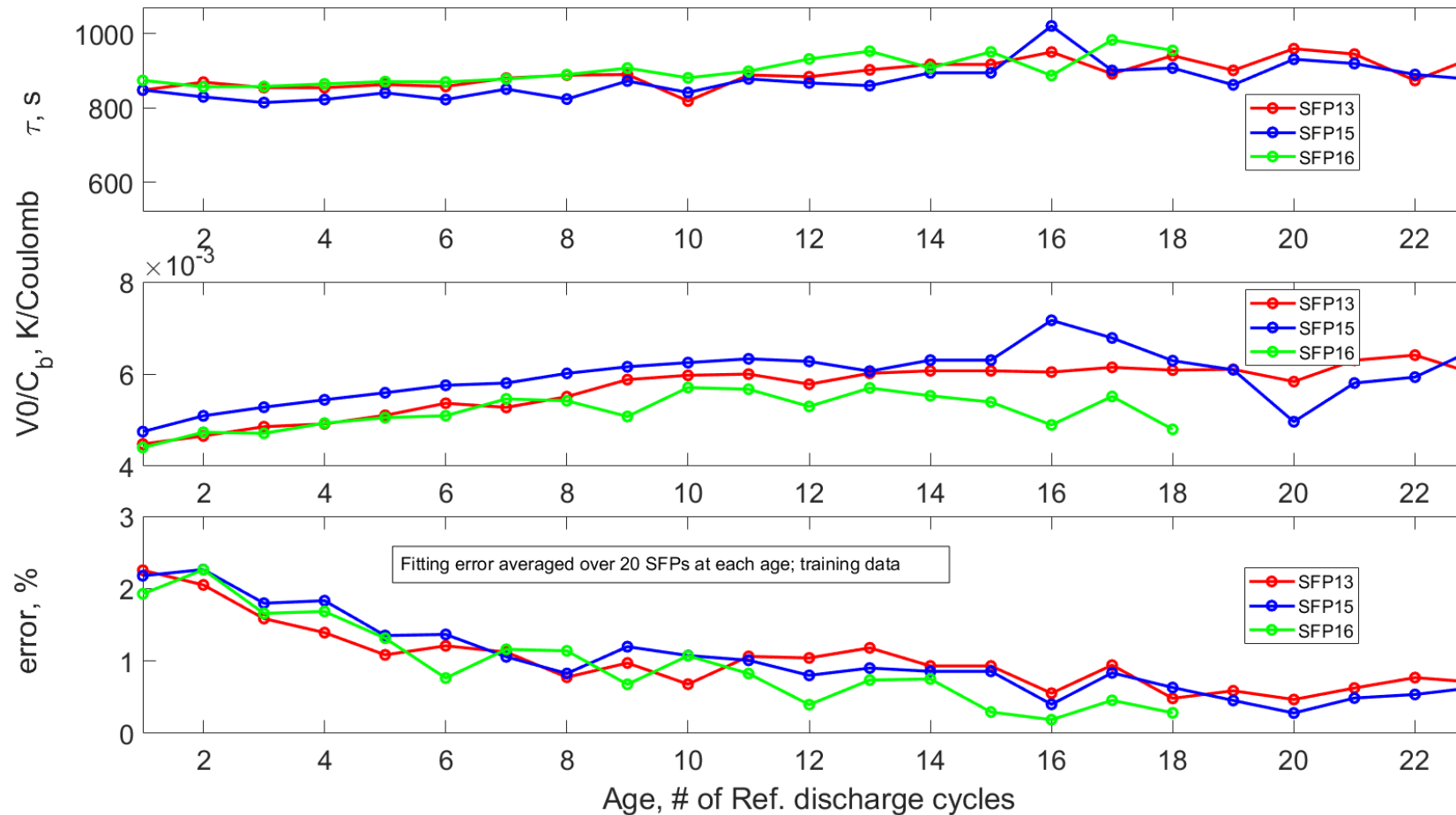
The 3-parameter TM is practically *non-identifiable* from SFP data at older age.



The 2-parameter ROM is *practically identifiable* from SFP data at each age.

Fitting ROM to SFP datasets

$$\text{ROM: } T(t) = T_a + [T(0) - T_a]e^{-t/\tau} + \frac{V_0}{C_b} \int_0^t I(t')e^{\frac{t'-t}{\tau}} dt'$$



The 2-parameter ROM is *practically identifiable* from SFP data at each age.

1. **Hybrid-ECM-based prognostics can predict temperatures for a “simulated” flight profile with noise** (assuming Poisson noise and Gaussian noise)
2. **Hybrid-ECM thermal model cannot be identified from SFPs alone, therefore it cannot be used for predicting aging parameters**
3. **Thermal ROM with two model parameters is identifiable and can be used to predict the aging parameters**
4. **Collecting experimental data with four datatypes (OCV, galvanostatic with range of “usable” C-rates, and “expected” loading profile) can provide better insights and provide a flexible path to model reduction for a targeted observable property**

Acknowledgments

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NASA Langley Research Center

Erik L. Frankforter
William C. Schneck
Peter W. Spaeth
Yi Lin
Daniel F. Perey

"We'll continue work to make flight even
safer ... to make it quieter ...
and through a healthy investment in
aeronautics, we'll reach new heights in
pursuit of making it cleaner and greener."

- NASA Administrator Charles Bolden



#StateOfNASA

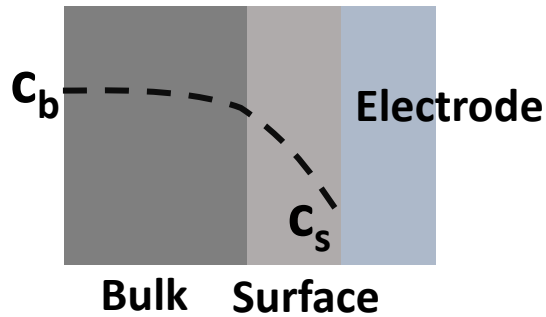
Funding

NASA Aeronautics Research Mission Directorate (ARMD) Convergent Aeronautics Solutions (CAS) Project, **SPARRCI** sub-project.

Backup Slides

Hybrid Echem Model (SPM + ECM):

concentration distribution assumption



Cell Voltage:

$$V = E_p^0 - E_n^0 - \dot{V}_o - \dot{\eta}_p - \dot{\eta}_n$$

V_o Voltage across ohmic resistance

η Butler-Volmer Kinetics

$$\dot{c} = \frac{c_b - c_s}{\tau_{diff}} \quad \text{Li}^+ \text{ concentration in a electrode}$$

$$\{\dot{\cdot}\}^k = \frac{\{\cdot\}^k - \{\cdot\}^{k-1}}{\tau} \quad \text{State Evolution}$$

M. Daigle, C.S. Kulkarni, Electrochemistry-based Battery Modeling for Prognostics, in: Annual Conference of the Prognostics and Health Management Society, 2013: p. 13.

Lumped Thermal Model:

Non-linearity in EOD temperature can be attributed to the 5th phase

$$\frac{\partial T}{\partial t} = \underbrace{\frac{Q}{m_{cell} C_p}}_{\text{Heat generated}} - \underbrace{\epsilon_{conv}(T - T_{amb})}_{\text{Heat loss}}$$

Modified Scaling Parameter based on Birk's model

$$x_{NE}(E_{NE}^{OC}) = \frac{\Delta x_{NE}}{1 + e^{\frac{\alpha(E_{NE}^{OC} - E_{0,NE,5})}{V_T}}}$$

$$\epsilon_{conv} = \frac{A_{surf}}{C_p m_{cell}} h_{conv}$$

$$Q = I \left(U_c - \alpha U_a - V - T \frac{dU}{dT} \right)$$

- Li-ion Chemistry

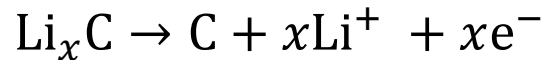
Cathode (LCO/NMC/LFP)

Anode (Graphite/Lithium)

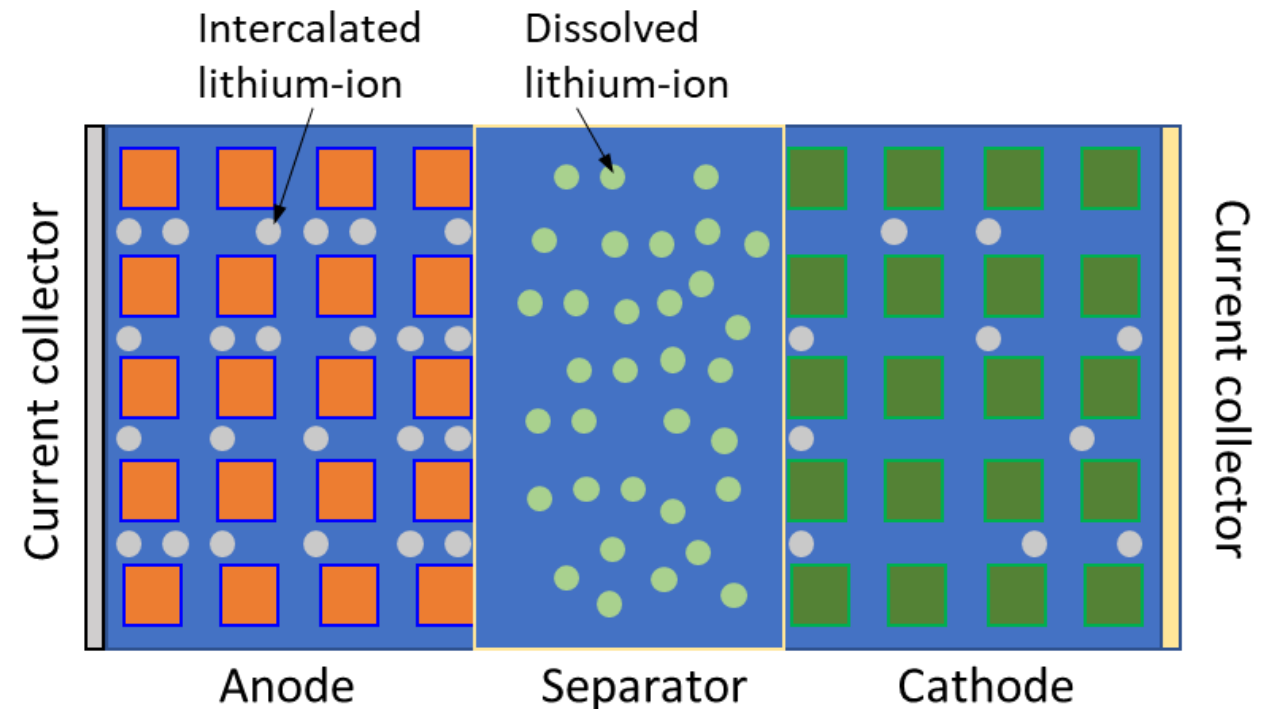
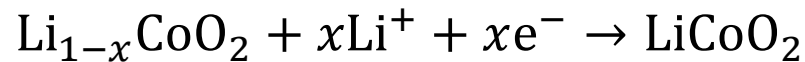
Electrolyte (LiCF_3SO_3 , LiPF_6 in EC/DMC)

Separator (PP/ Al_2O_3)

Anode reaction

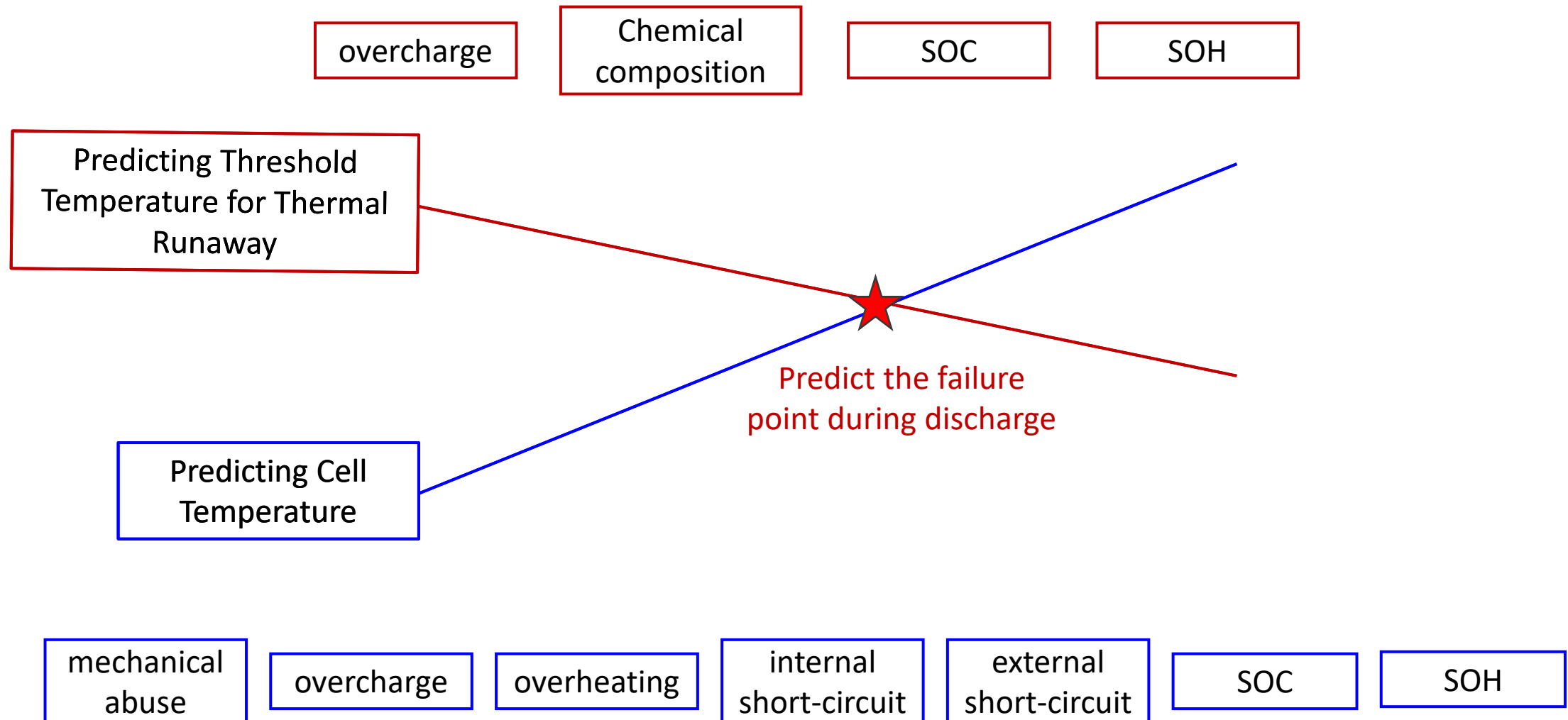


Cathode reaction

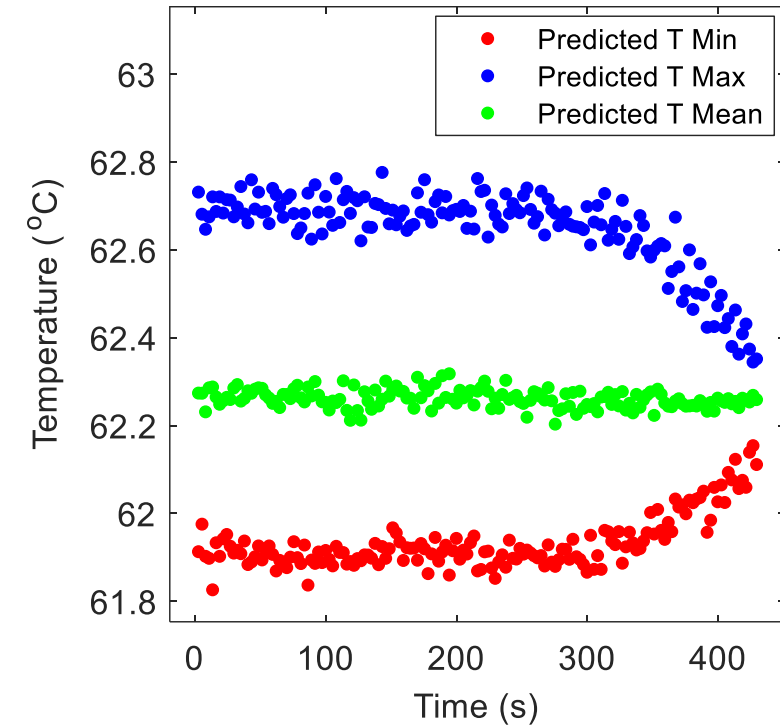
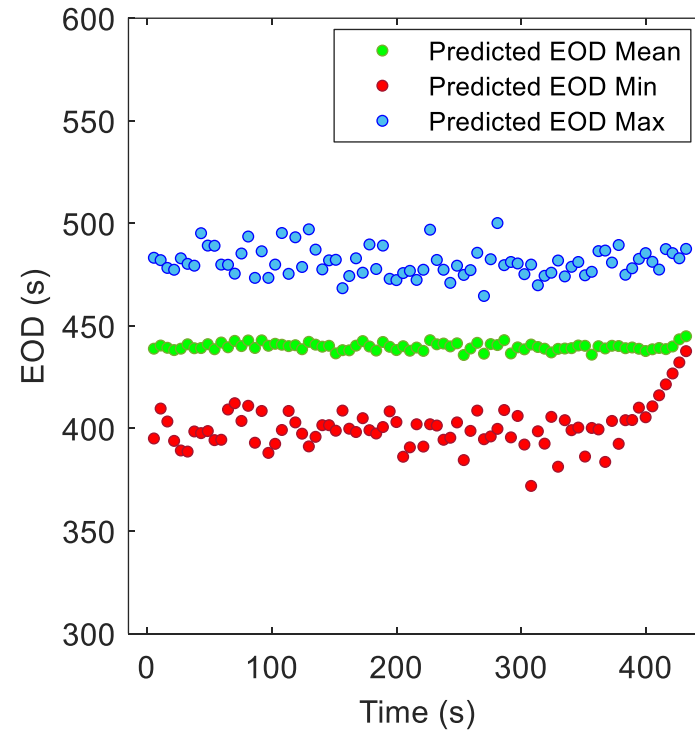
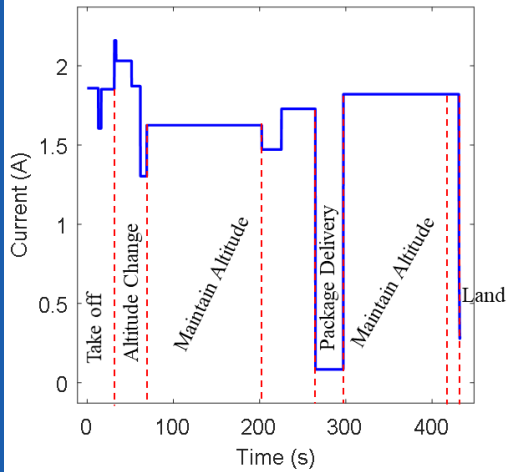


Full Multiphysics Models Allows us to identify Important Mechanisms for capturing Thermal and Battery Performance with Aging over High C flight profiles

Simplified Picture of Complexity in Modeling Thermal Runaway

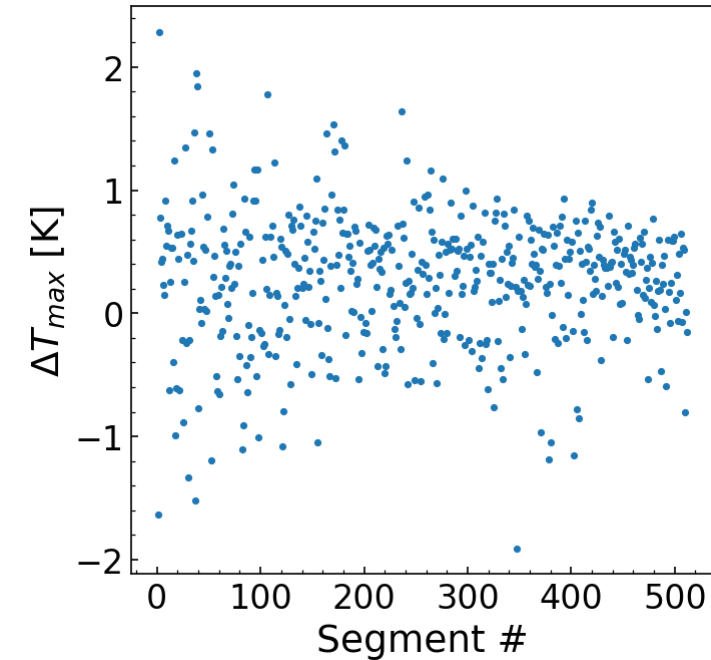
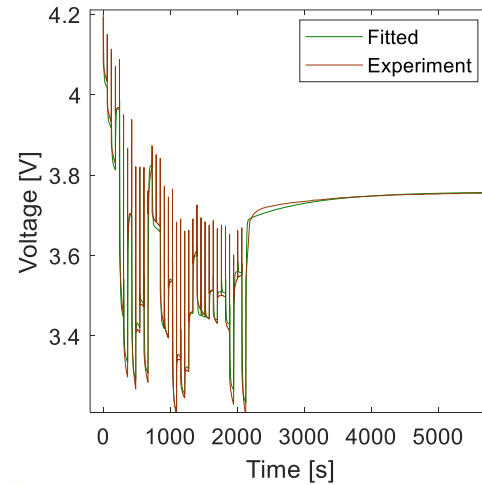
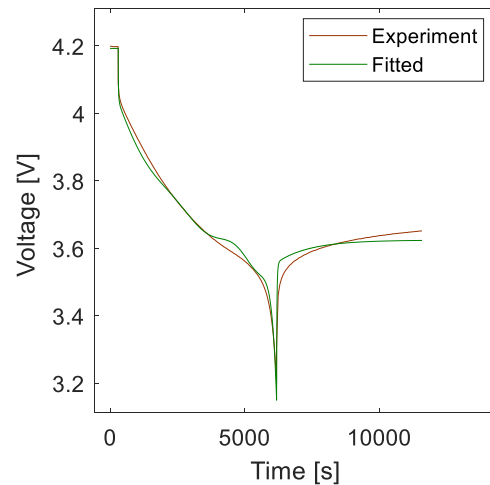
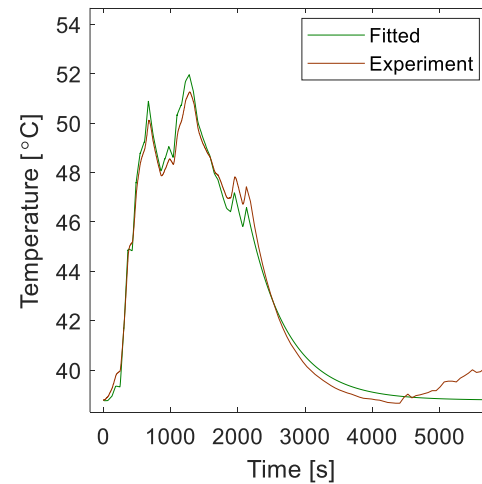
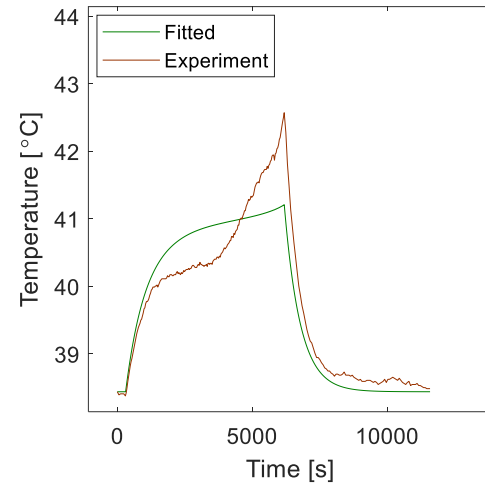


Applying Prognostics Algorithm on a Short Test Flight Profile



The Distribution of the End-of-Discharge Estimation shows that the Algorithm is working correctly.

Part I: Effect of Discharge Protocol



Estimating entropic term (dU/dT) in addition to Q_{mobile} , R_o , τ_{diff} improves temperature prediction for RW but not Reference discharges

What are ROMs?

- A ROM is a simplified model of the system which interpolates in a subset of data.
- Different subsets of data will be associated with different ROMs. For example, ROM1 may predict a battery's voltage while ROM2 may predict its temperature.
- A ROM can be physics-based or purely data-driven.

Advantages of ROMs:

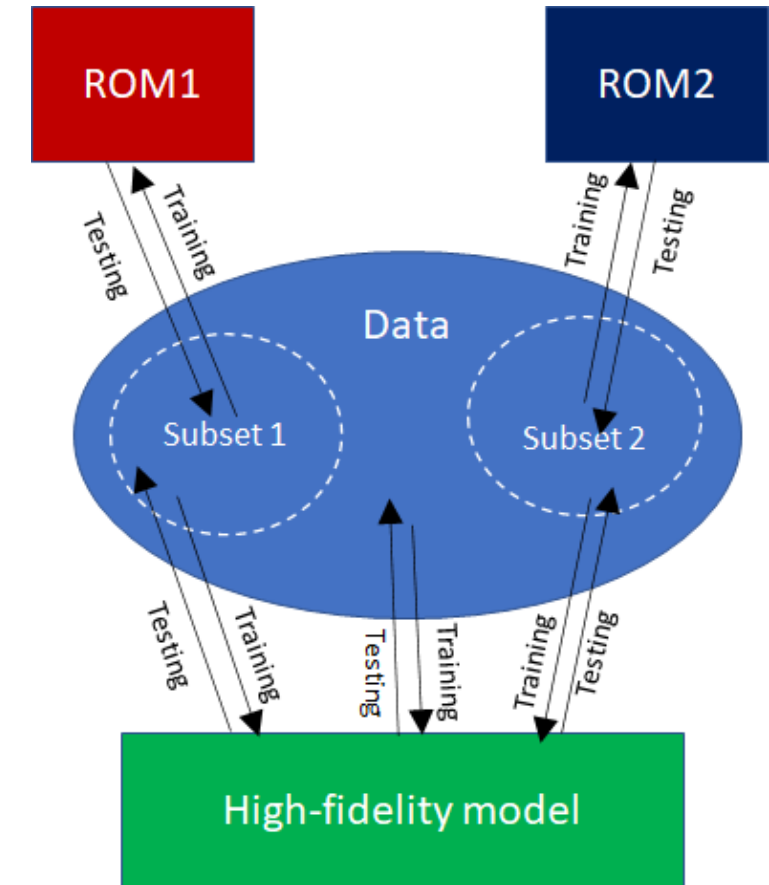
- The computational complexity of a ROM is lower than that of a high-fidelity model.
- A ROM can be practically identifiable, i.e., its parameters can be uniquely fit to data.

Disadvantages of ROMs:

- Limited range of validity compared to a high-fidelity model.

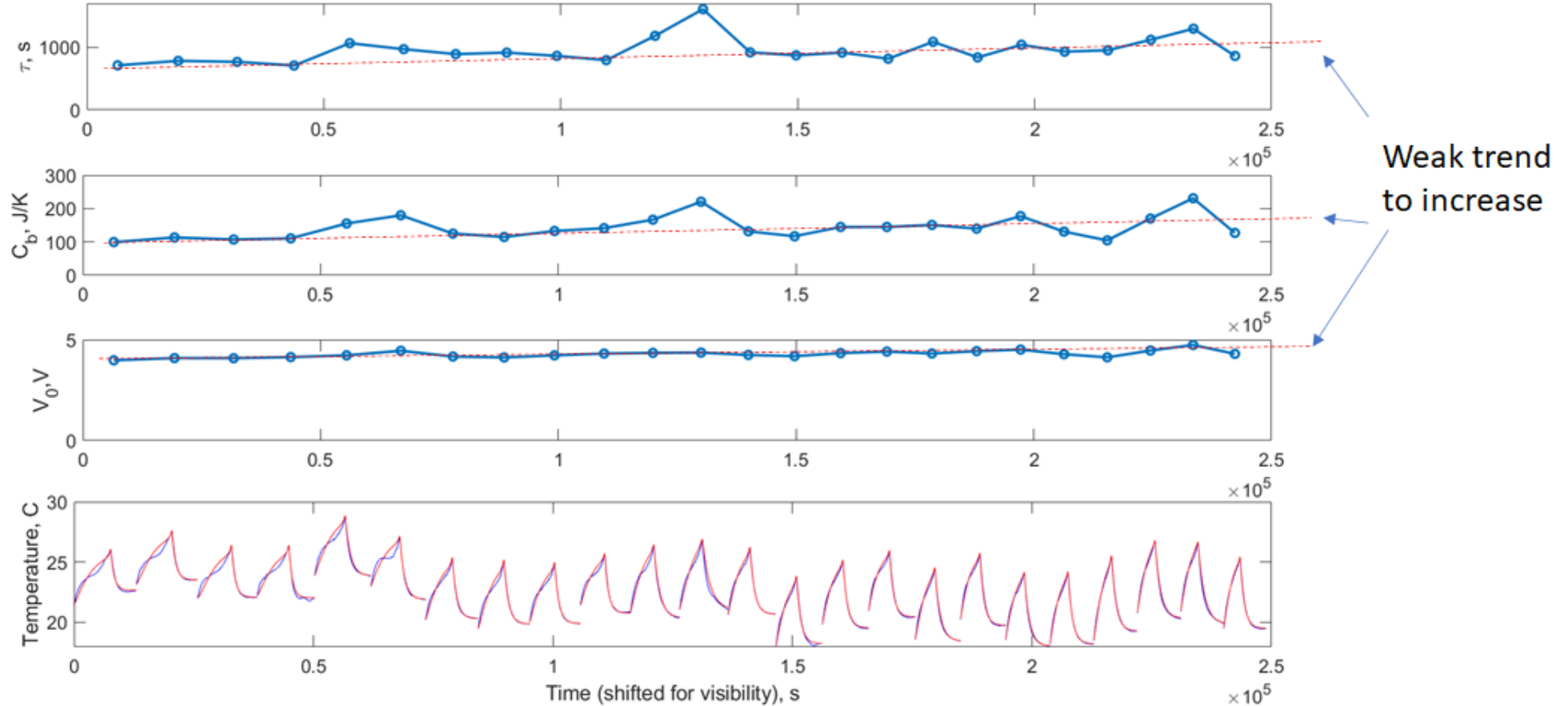
How to derive ROM?

- Our approach is inspired by [Manifold Boundary Approximation Method](#):
 - Parameter sensitivity applied to high-fidelity model is used eliminate some parameters from the model
 - The resulting ROM is fitted to the data. If it's not completely identifiable, the reduction is repeated, until the final ROM is completely identifiable.



Fitting the TM to galvanostatic discharge data

$$\text{TM: } T(t) = T_a + [T(0) - T_a]e^{-t/\tau} + C_b^{-1} \int_0^t I(t') [V_0 - V(t')] e^{\frac{t'-t}{\tau}} dt'$$

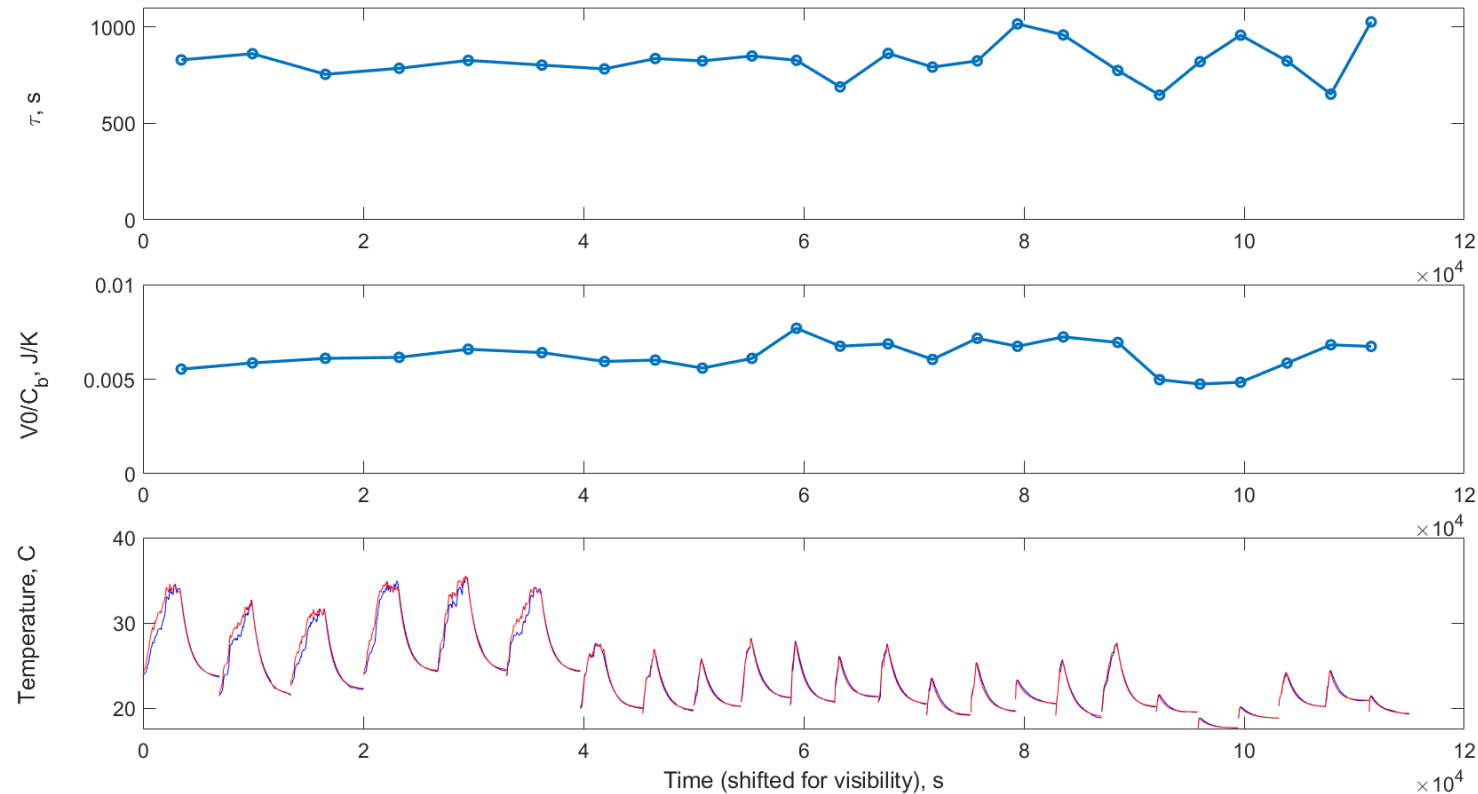


The TM is identifiable from galvanostatic discharges data.

Thermal ROM for RW data

$$\text{TM: } T(t) = T_a + [T(0) - T_a]e^{-t/\tau} + C_b^{-1} \int_0^t I(t') [V_0 - V(t')] e^{\frac{t'-t}{\tau}} dt'$$

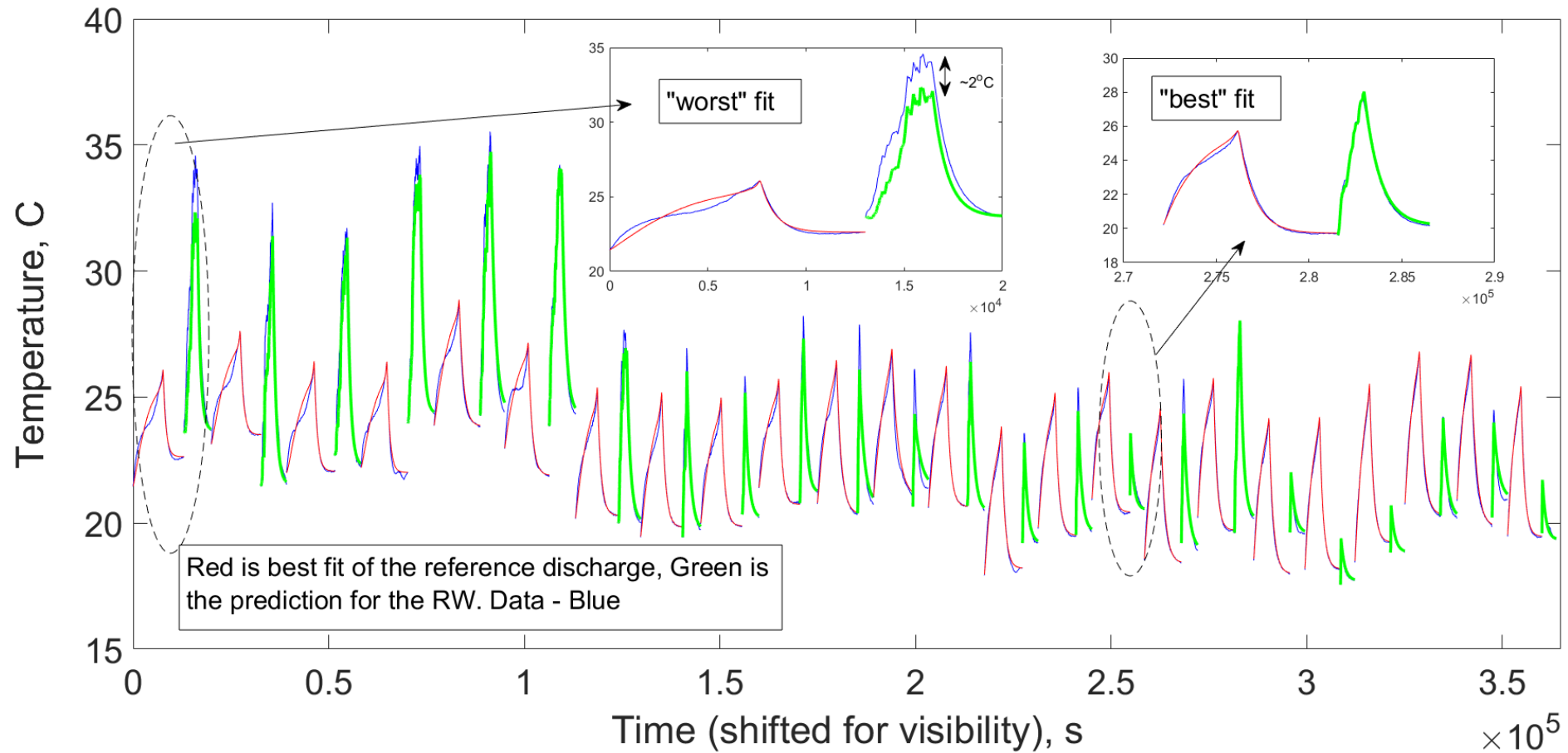
$$\rightarrow \text{ROM: } T(t) = T_a + [T(0) - T_a]e^{-t/\tau} + \frac{V_0}{C_b} \int_0^t I(t') e^{\frac{t'-t}{\tau}} dt'$$



- The ROM is accurate for and is identifiable from the RW data;
- The temperature evolution decouples from the voltage evolution.

Using the TM to predict RW data.

$$\text{TM: } T(t) = T_a + [T(0) - T_a]e^{-t/\tau} + C_b^{-1} \int_0^t I(t') [V_0 - V(t')] e^{\frac{t'-t}{\tau}} dt'$$



Fitting the TM to galvanostatic discharge data gives a decent prediction for RW data.