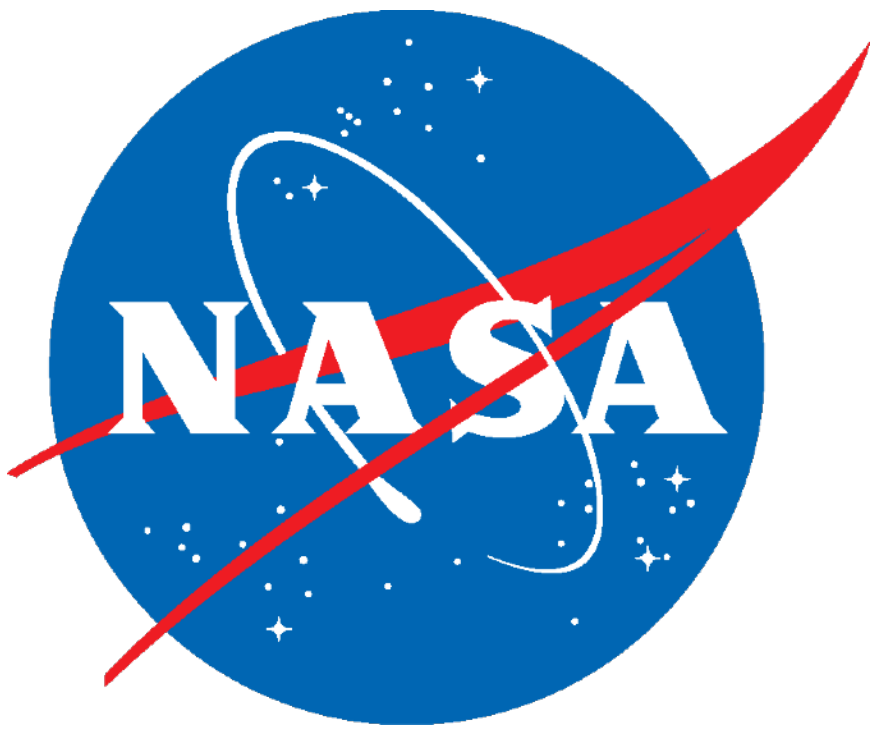


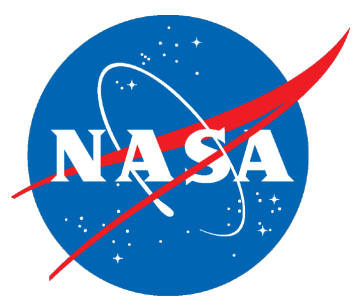


Multihierarchy Gaussian Process Models for Probabilistic Aerodynamic Databases using Uncertain Nominal and Off-Nominal Configuration Data

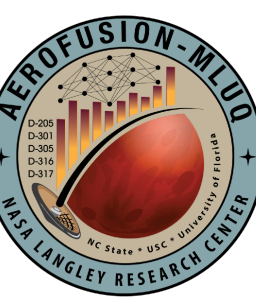
2023 AIAA SciTech Forum
January 23-27, 2023 - National Harbor, Maryland

James B. Scoggins, Thomas J. Wignall, Tenavi Nakamura-Zimmerer, Karen Bibb
NASA Aerofusion-MLUQ Early Career Initiative, NASA Langley Research Center



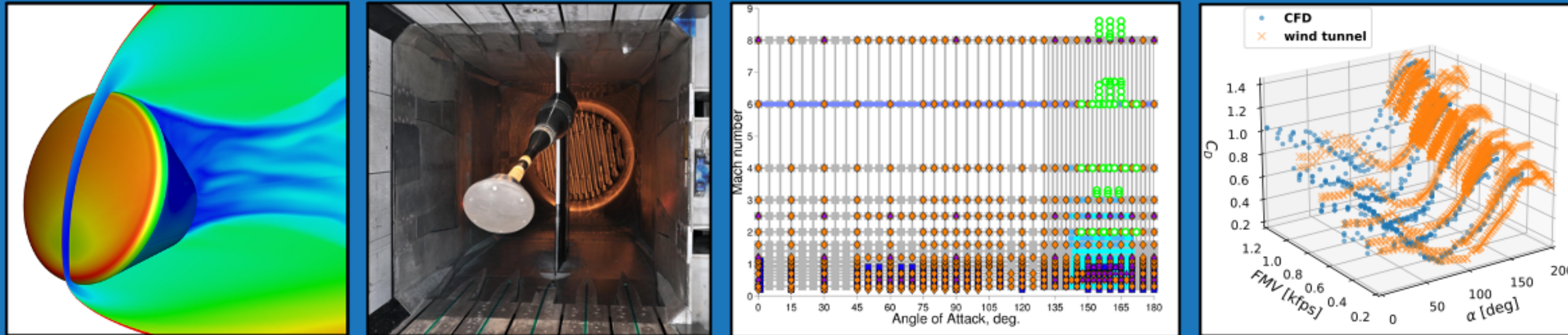


Outline



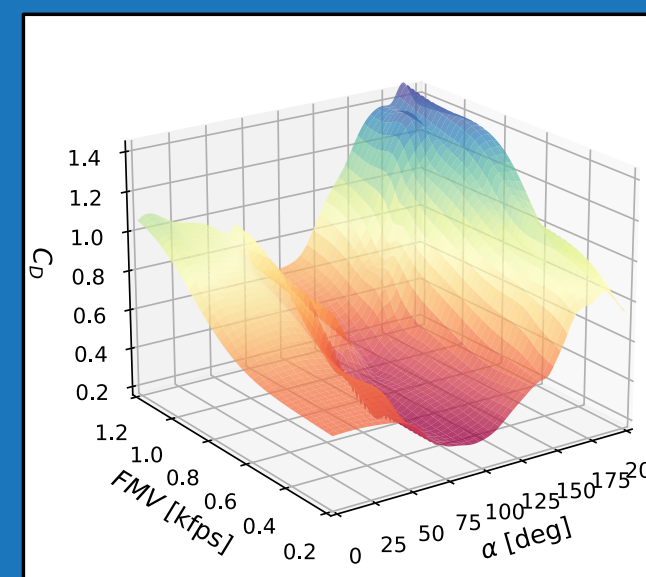
- Data fusion problem for aerodynamic databases
- Introduction to Gaussian Processes (GPs)
- Multihierarchy Gaussian Process Regression
- Example dataset based on Orion wind-tunnel data
- Results

CFD Solutions and Wind Tunnel Tests

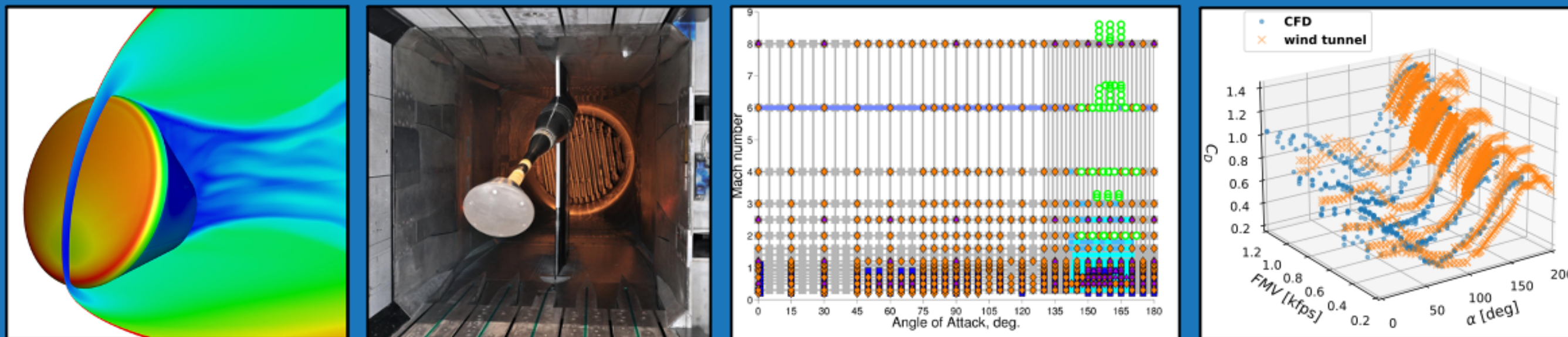


Aerodynamic Database Construction

$$\begin{bmatrix} \text{FMV} & \text{Mach or Velocity} \\ \alpha & \text{Angle of Attack} \\ \beta & \text{Side-slip Angle} \\ \text{Re} & \text{Reynolds Number} \\ \vdots & \end{bmatrix} \Rightarrow \begin{bmatrix} C_L & \text{Lift Force} \\ C_D & \text{Drag Force} \\ C_m & \text{Pitching Moment} \\ \vdots & \end{bmatrix}$$

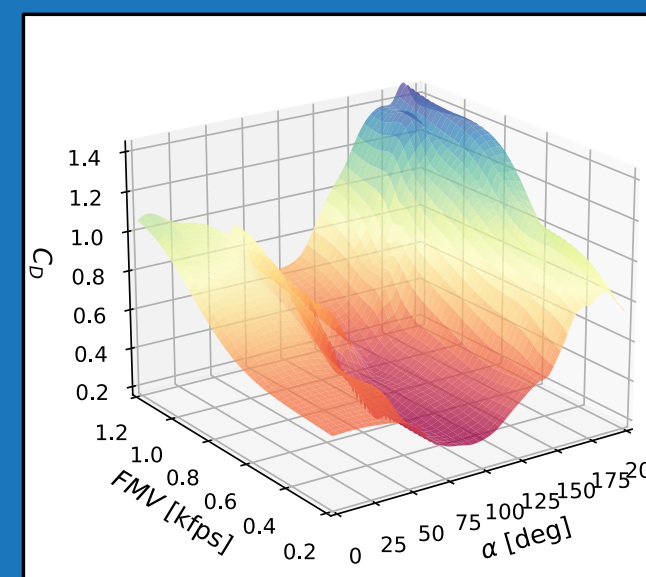


CFD Solutions and Wind Tunnel Tests

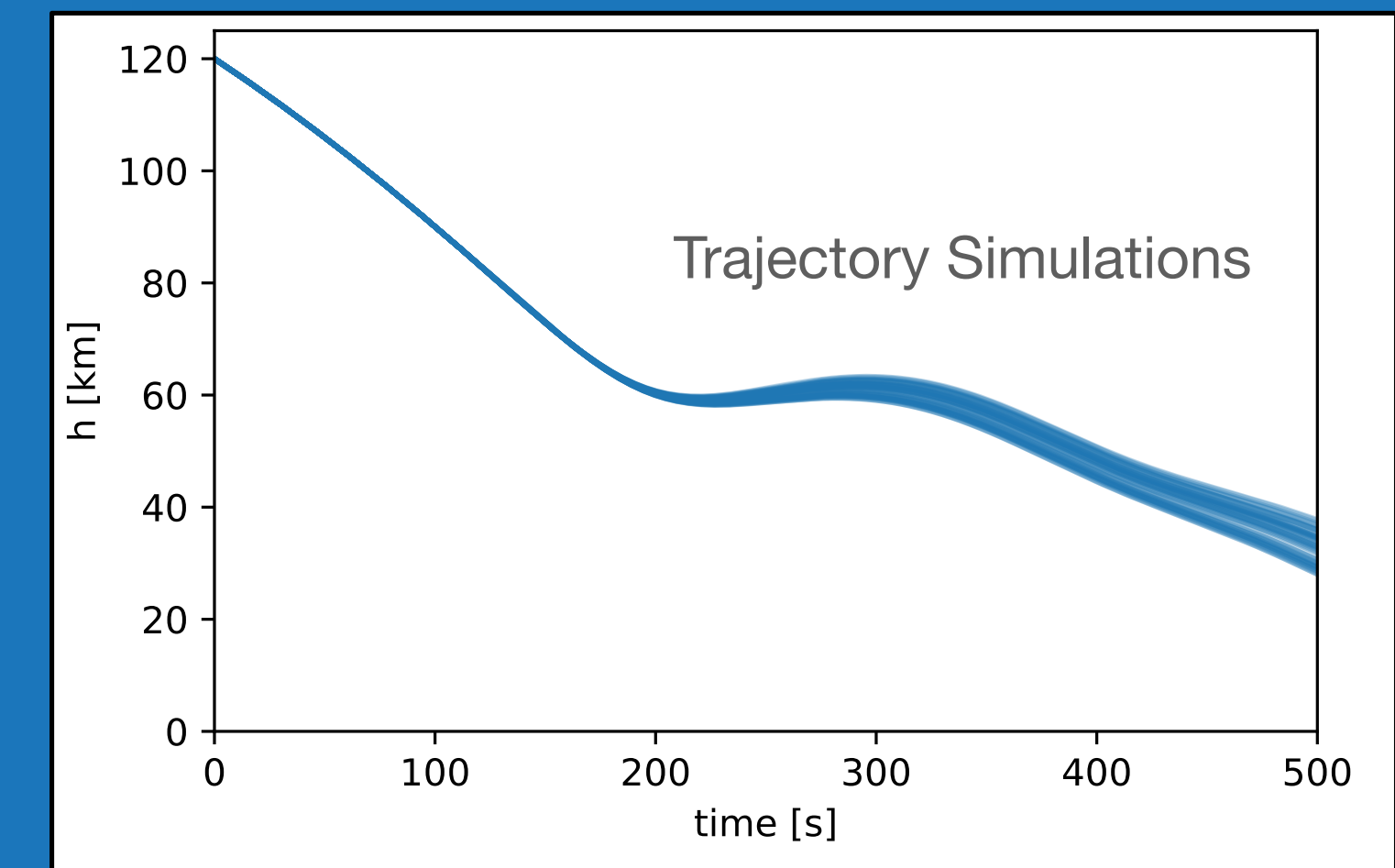
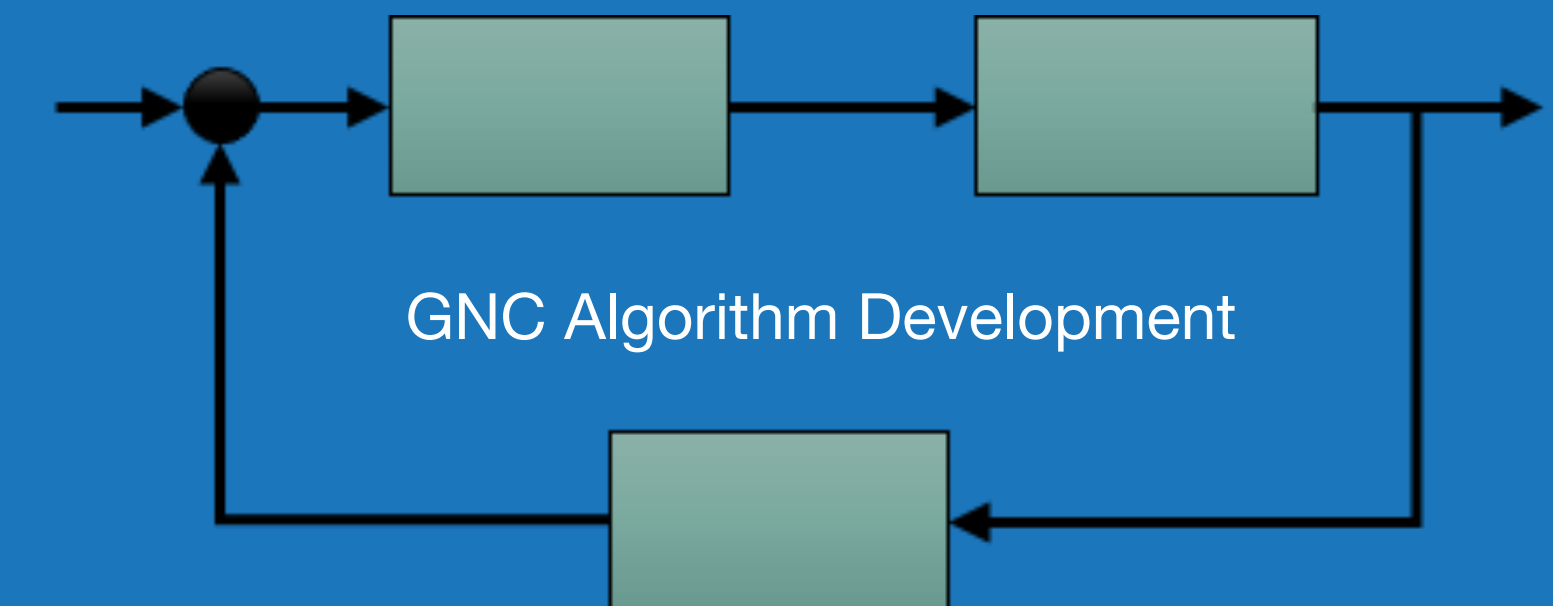


Aerodynamic Database Construction

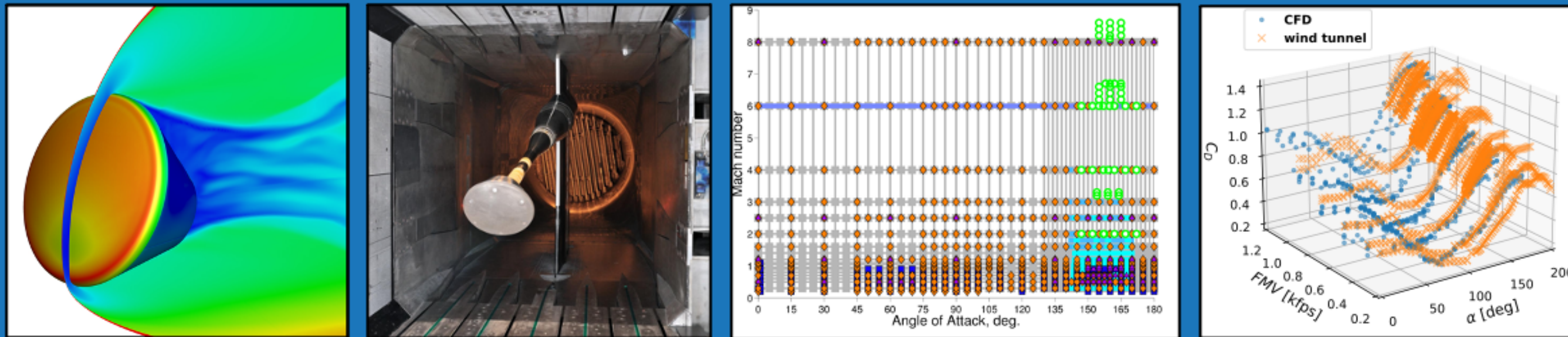
$\left[\begin{array}{l} \text{FMV} \\ \alpha \\ \beta \\ \text{Re} \\ \vdots \end{array} \right]$	$\Rightarrow$	$\left[\begin{array}{l} C_L \\ C_D \\ C_m \\ \vdots \end{array} \right]$	Mach or Velocity	Lift Force
			Angle of Attack	Drag Force
			Side-slip Angle	Pitching Moment
			Reynolds Number	:
			:	:



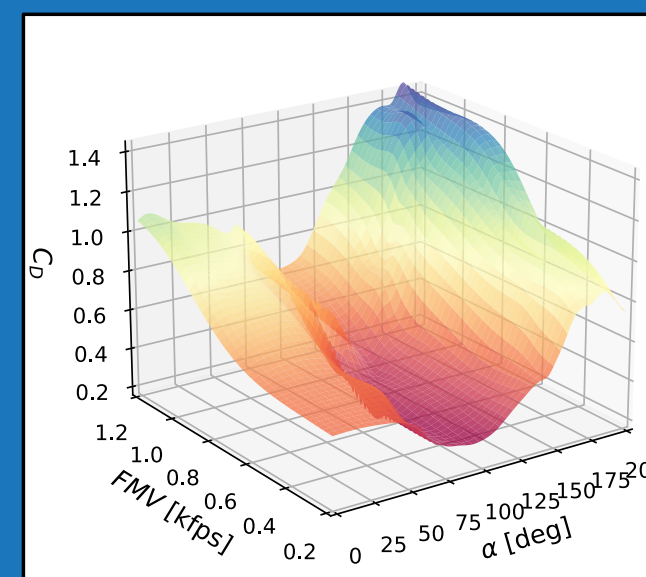
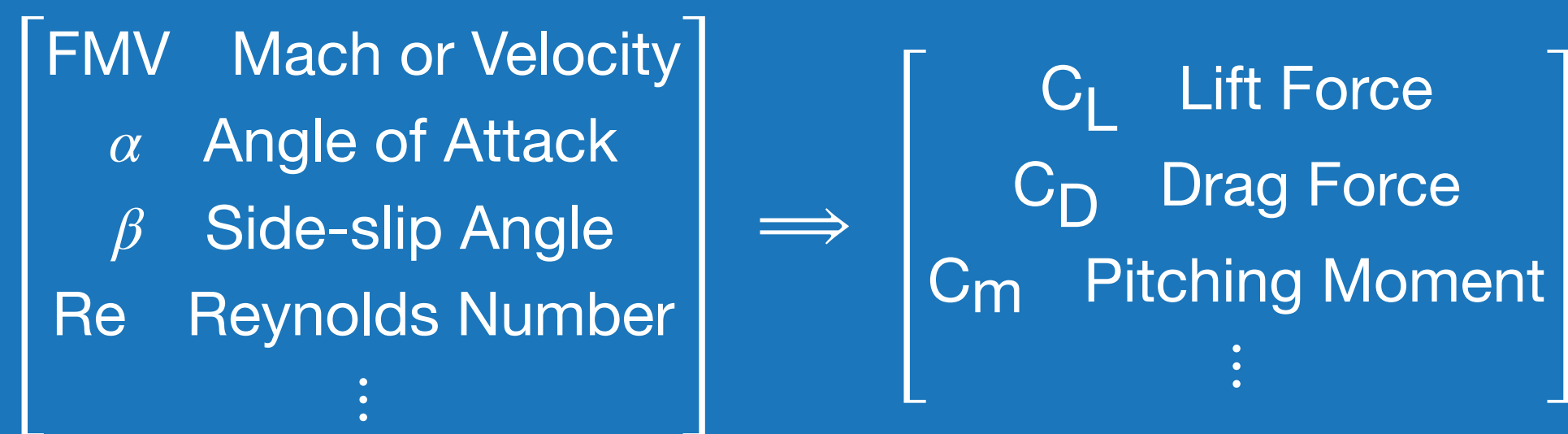
Engineering and Mission Design



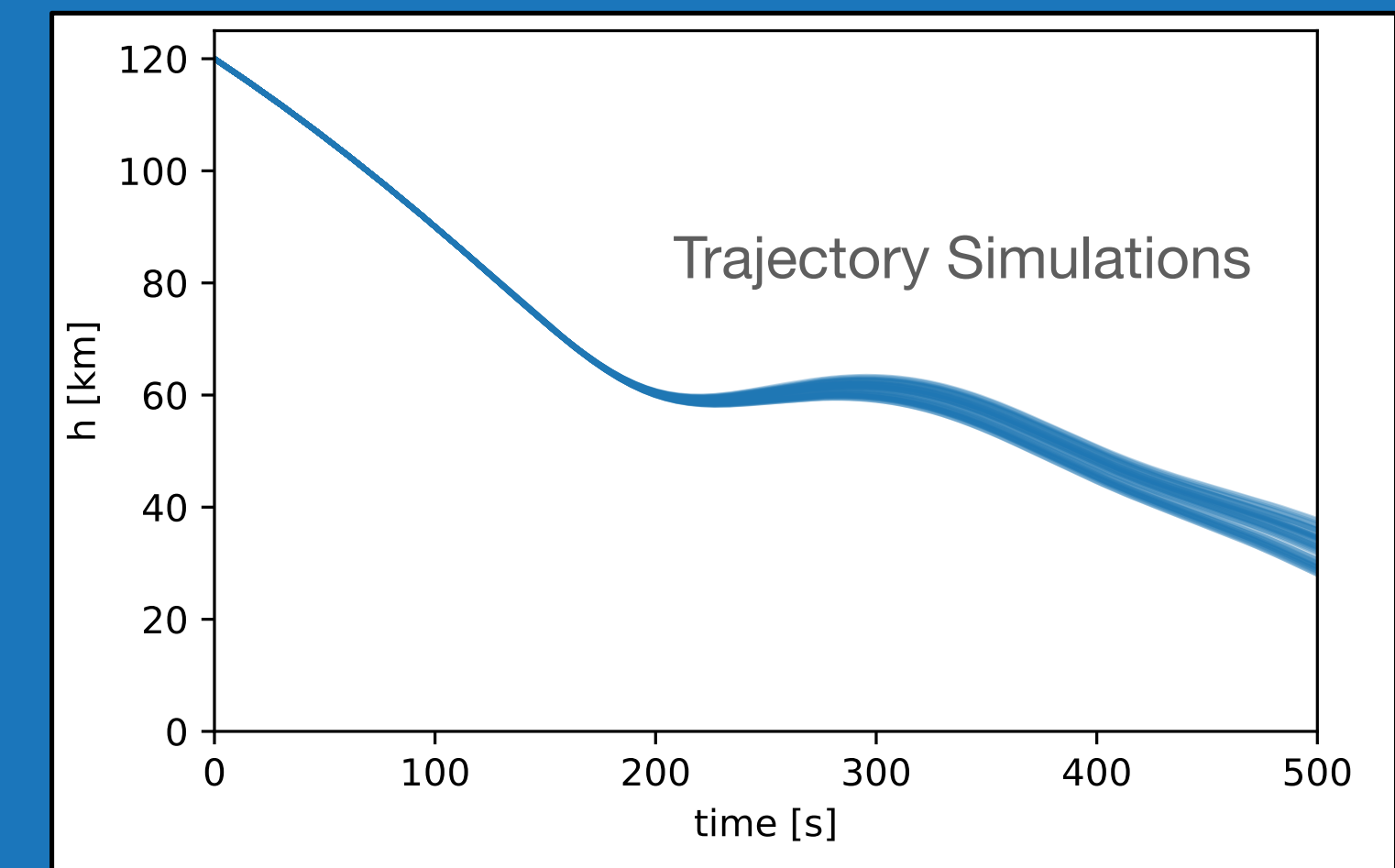
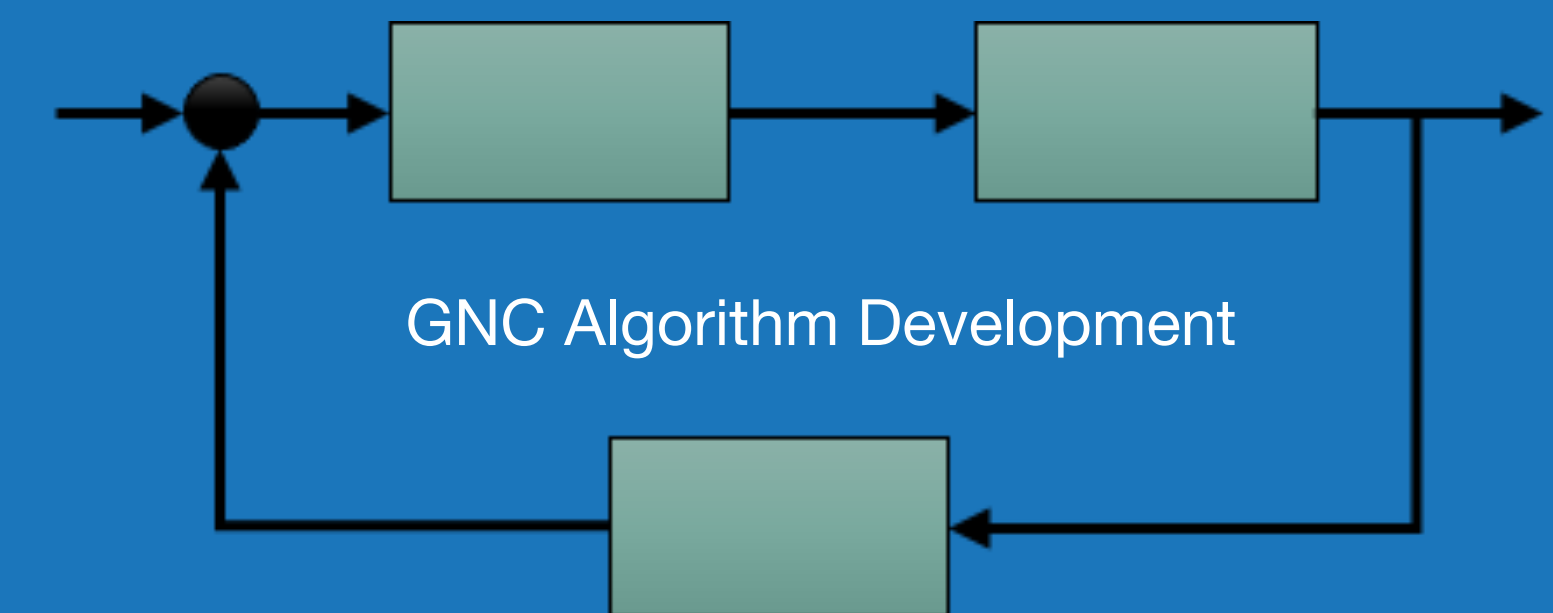
CFD Solutions and Wind Tunnel Tests



Aerodynamic Database Construction



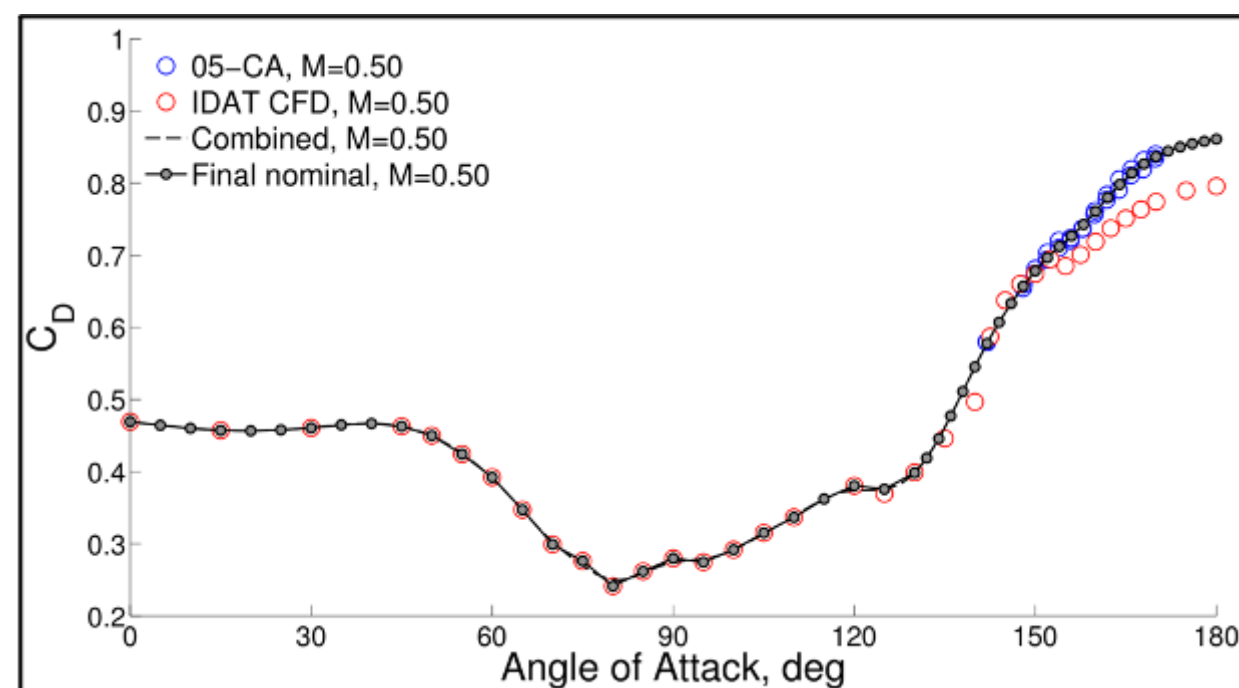
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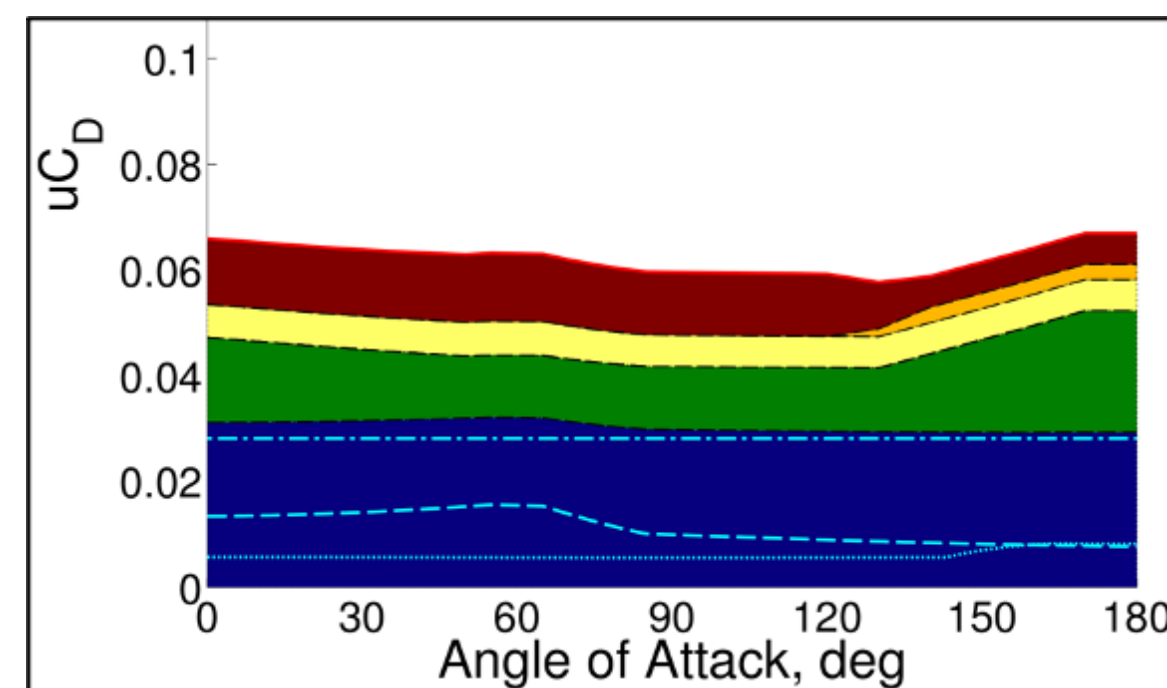
Typical data is noisy, with varying degrees of fidelity to flight vehicle

- Data continuously updated as design matures
- Different levels of fidelity in computational tools
- Wind tunnel models approximate vehicle geometry and roughness
- Wind tunnels cannot always reproduce flight conditions

Current state of the practice: “UQ by Inspection”



Nominal aerocoeficients constructed using expert judgment, given multiple sources of data.

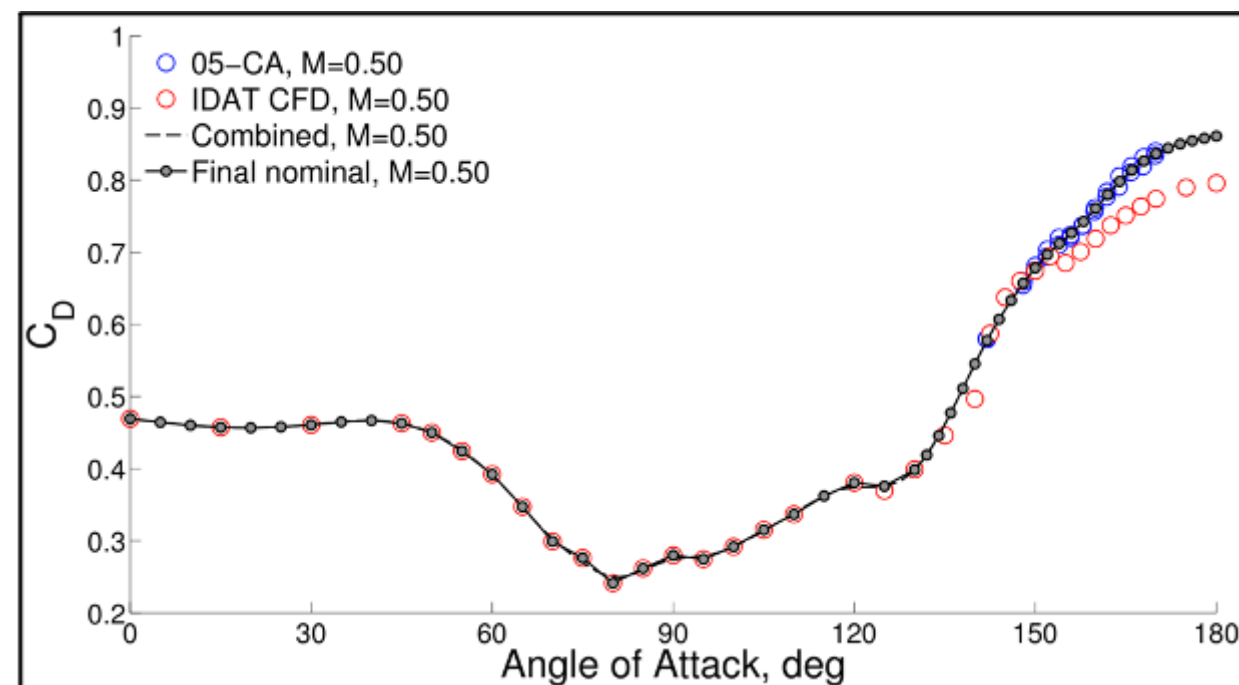


Uncertainty buildup based on dispersion factors, tuned to cover varying data sources.

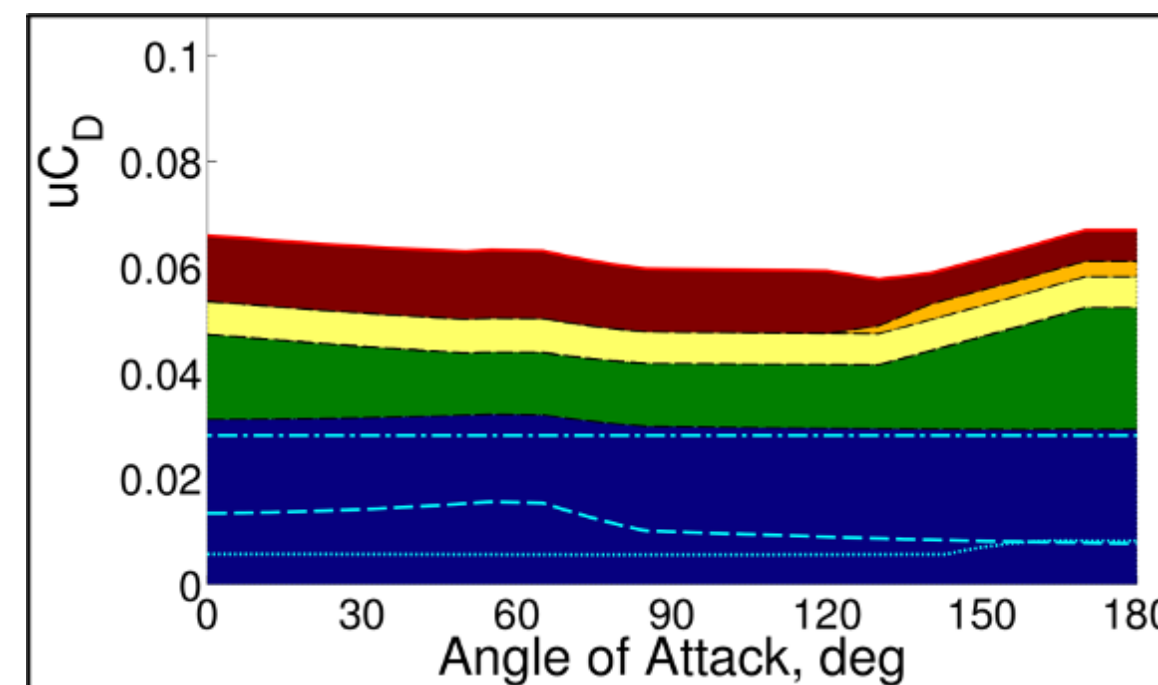
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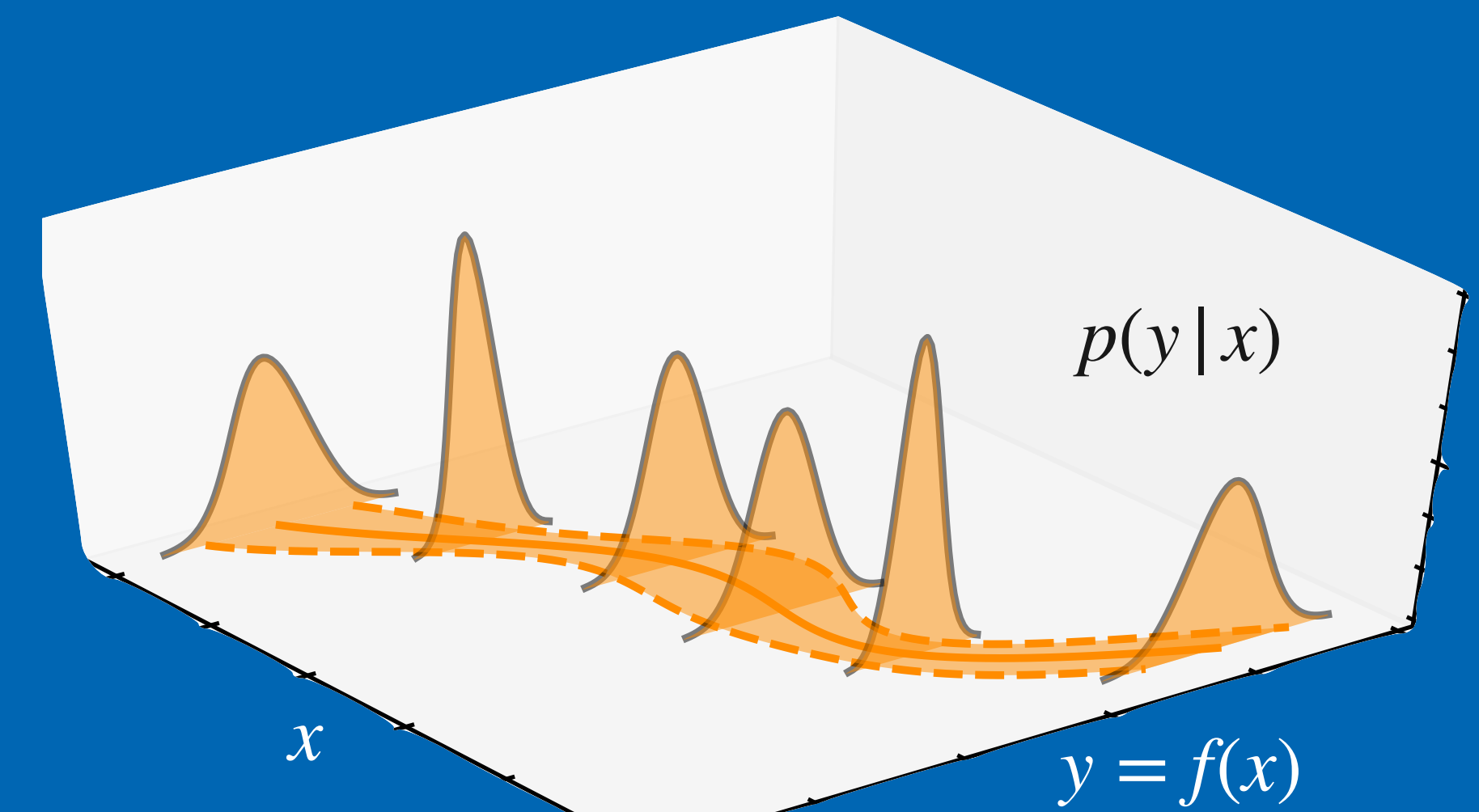


Nominal aerocoefficients constructed using expert judgment, given multiple sources of data.



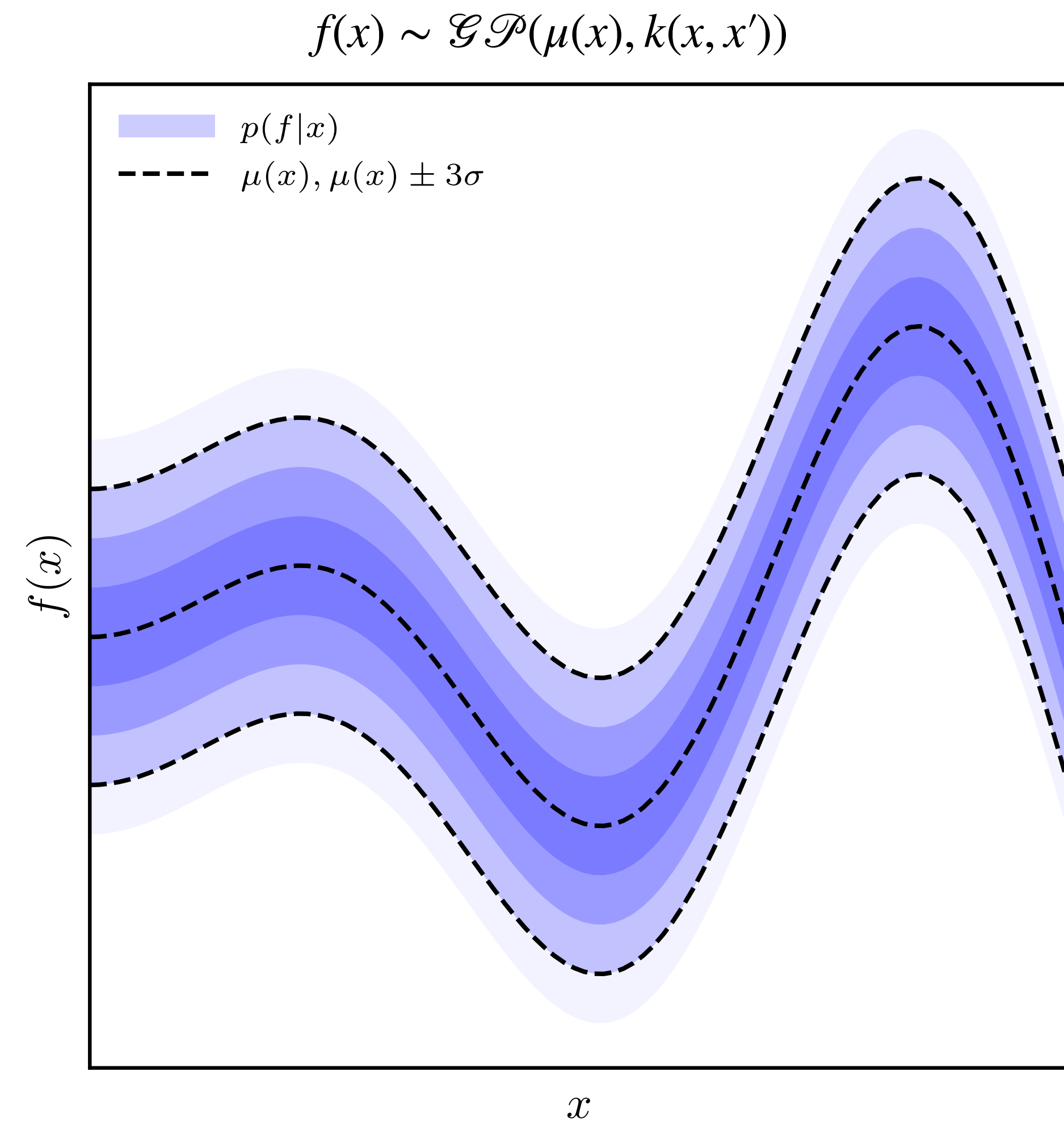
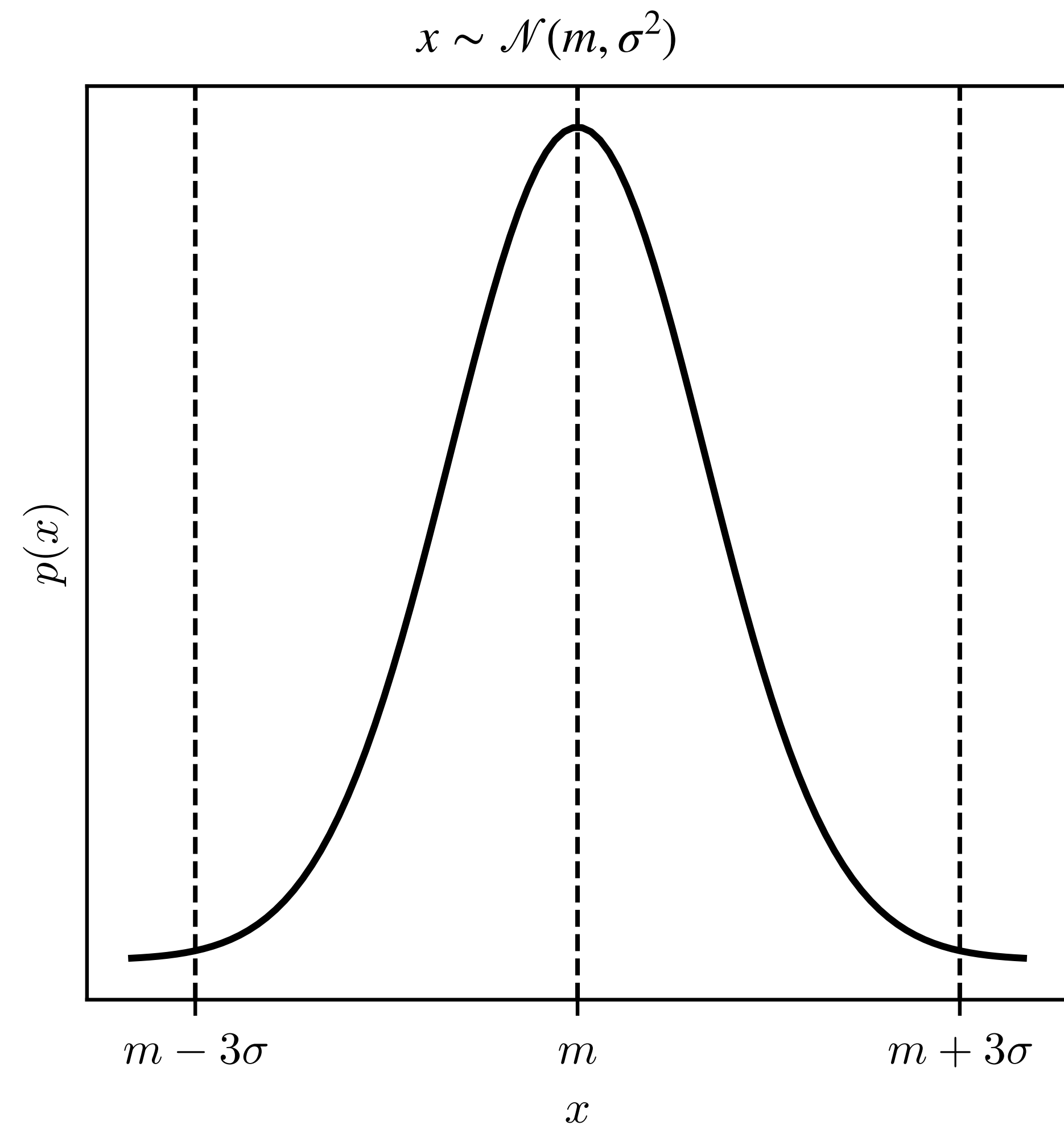
Uncertainty buildup based on dispersion factors, tuned to cover varying data sources.

Want to “learn” a surrogate conditional probability distribution, given all data sources

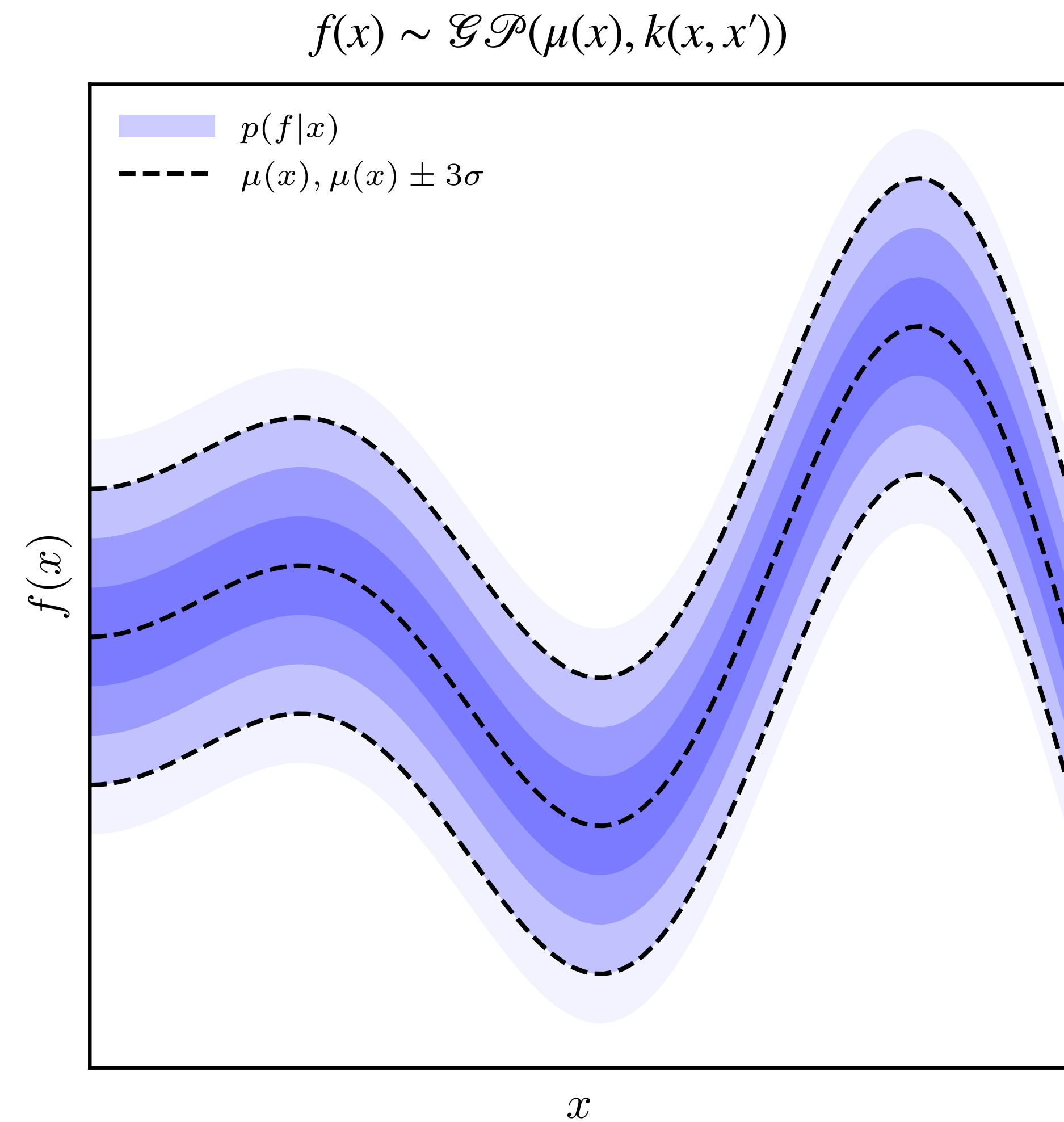
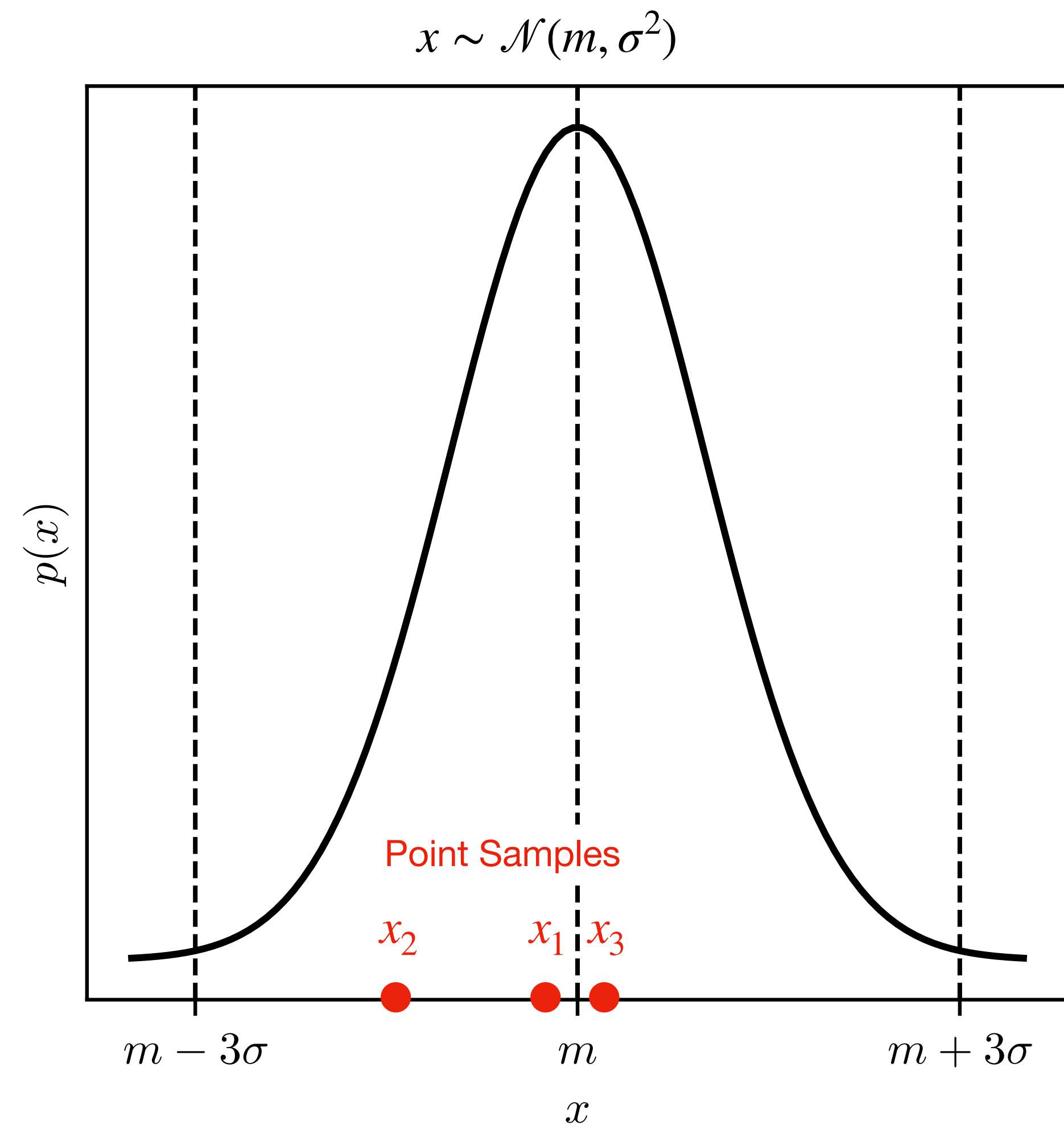


- $p(y|x)$ defines the “probability of outcome y given x ”
- surrogate model is “stochastic” but not “random”

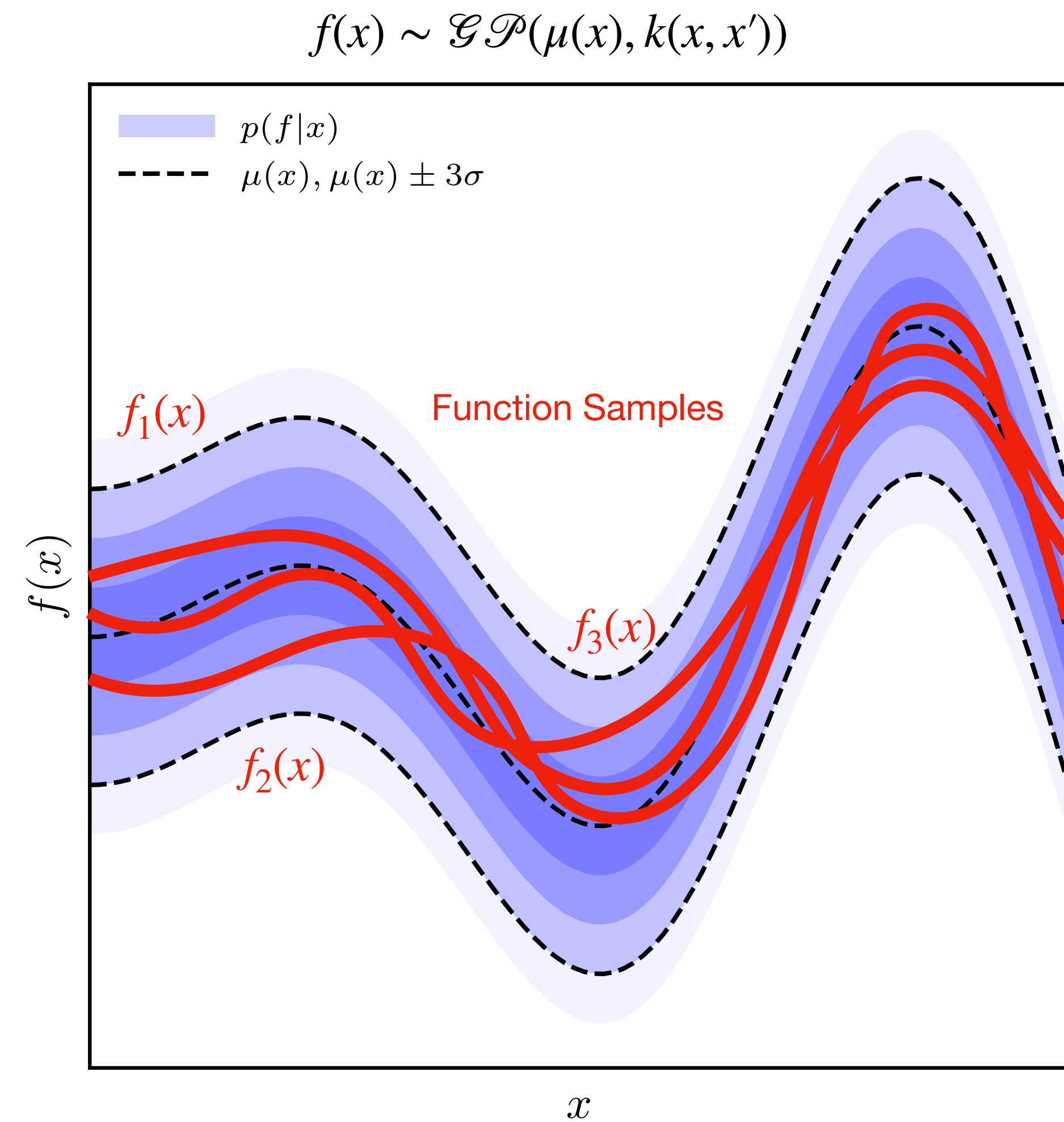
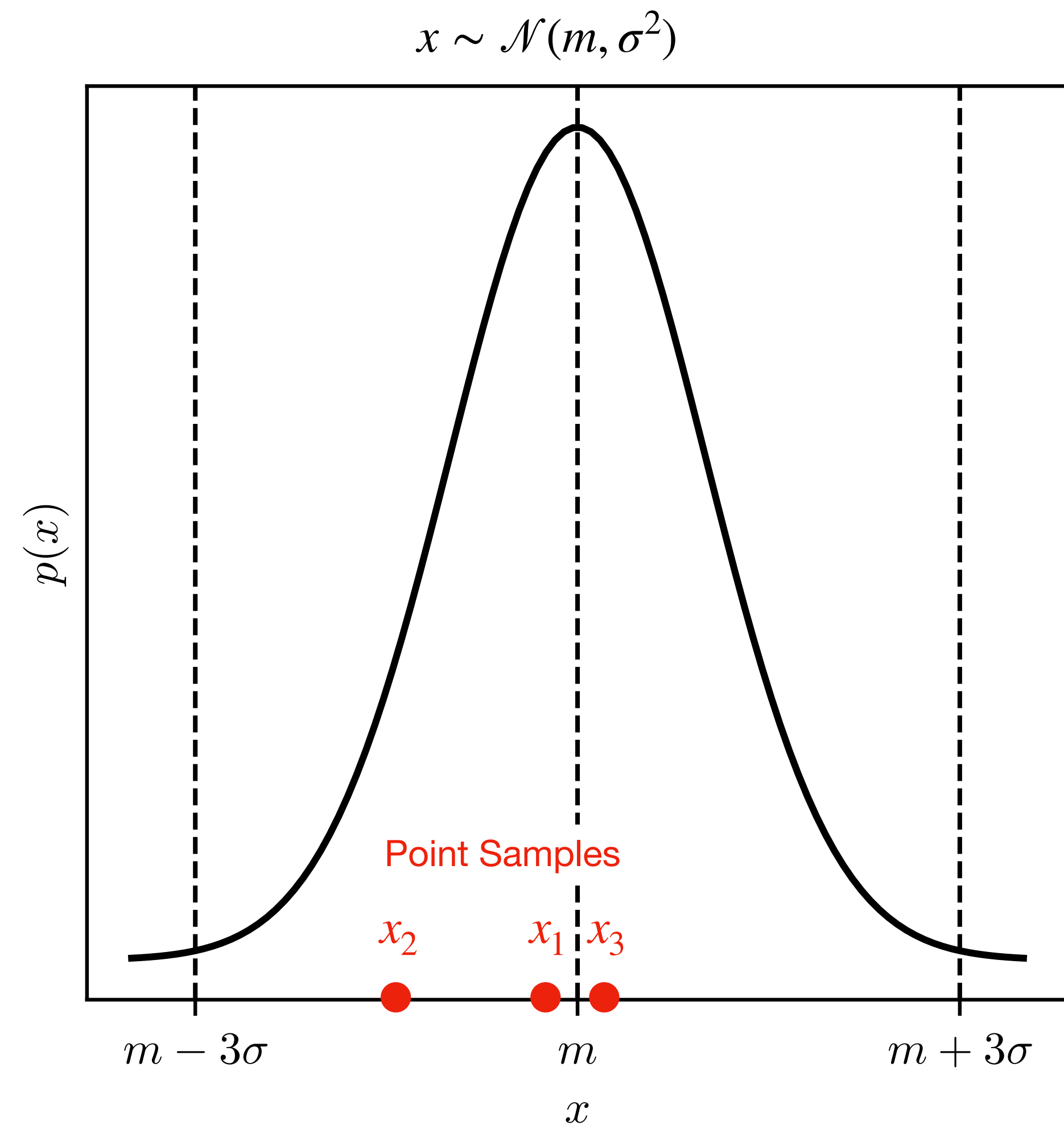
Generalization of the Gaussian distribution for random scalars to random functions



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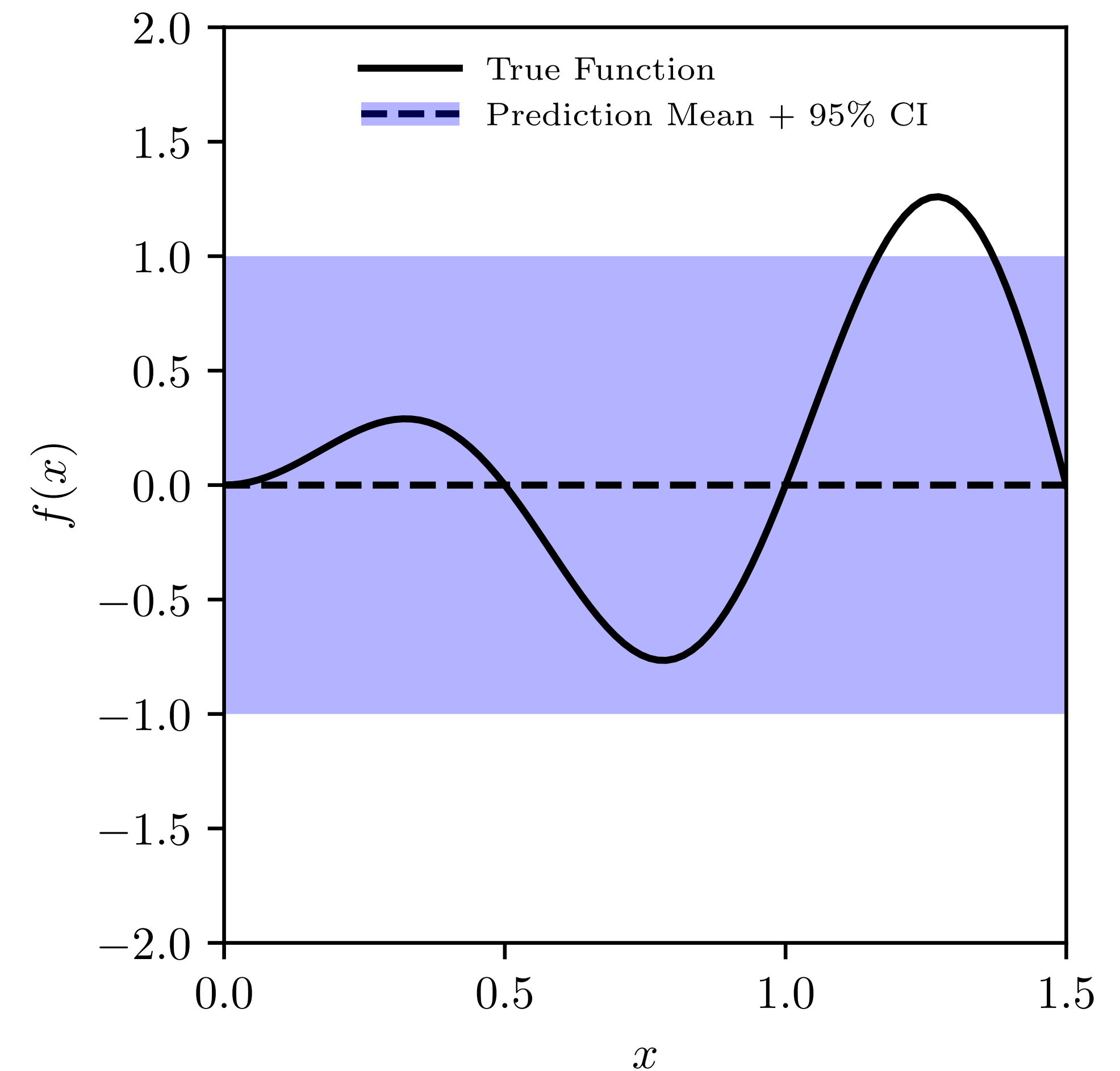


Any finite set of index points $\{x_1, \dots, x_n\}$ represents a multivariate Gaussian distribution for function values.

prior

$$f(x) \sim p(f|x) = \mathcal{GP}(\mu, k)$$

$$\Rightarrow \begin{bmatrix} f(x_1) \\ f(x_2) \\ f(x_3) \end{bmatrix} \sim \mathcal{N} \left(\begin{bmatrix} \mu(x_1) \\ \mu(x_2) \\ \mu(x_3) \end{bmatrix}, \begin{bmatrix} k(x_1, x_1) & k(x_1, x_2) & k(x_1, x_3) \\ k(x_2, x_1) & k(x_2, x_2) & k(x_2, x_3) \\ k(x_3, x_1) & k(x_3, x_2) & k(x_3, x_3) \end{bmatrix} \right)$$



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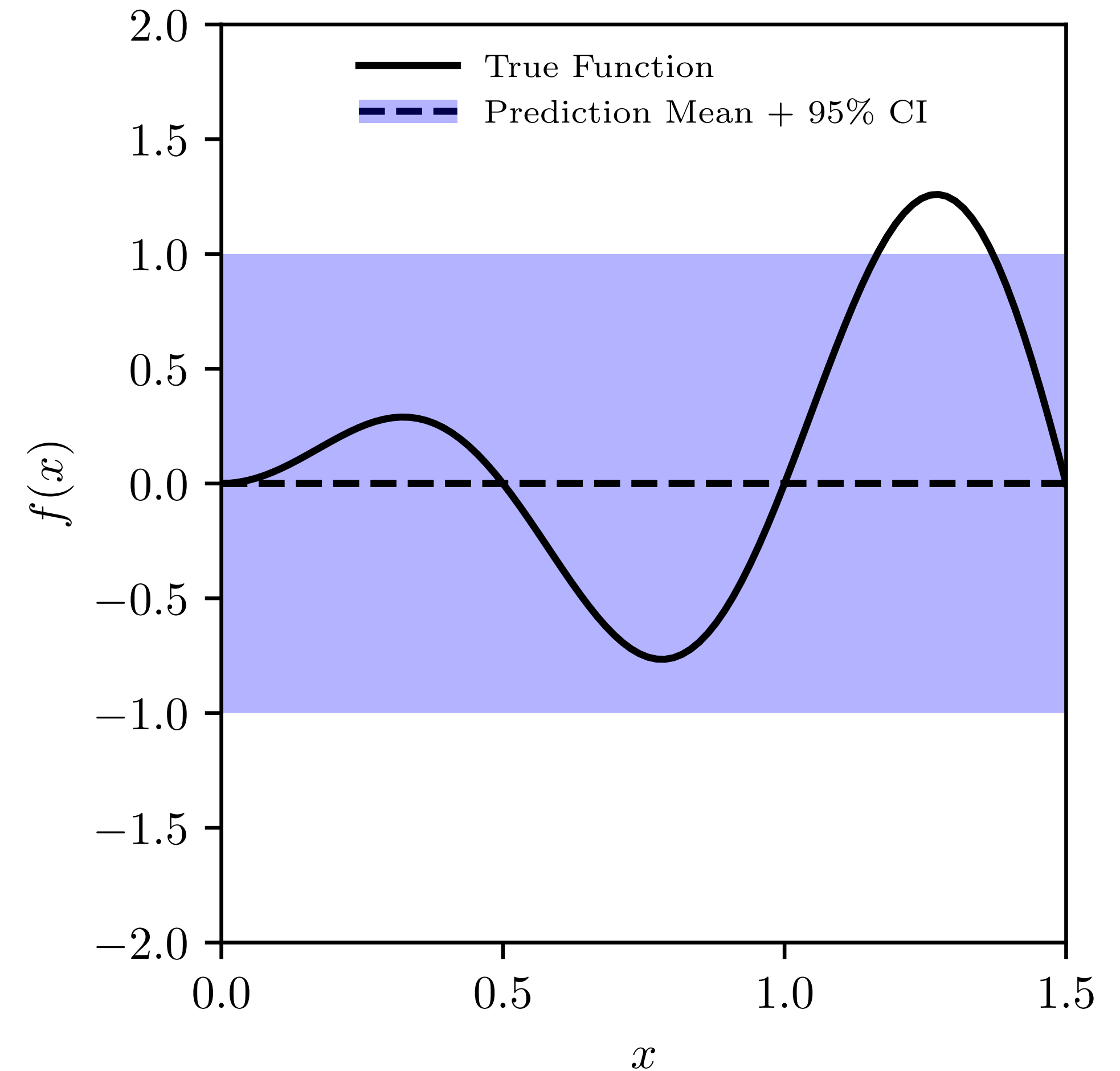
Regression performed by conditioning the distribution on data $\mathcal{D} = (X, y)$

posterior

$$\hat{f}(x) \sim p(f|\mathcal{D}, x) = \mathcal{GP}(\hat{\mu}, \hat{k})$$

$$\hat{\mu}(x) = \mu(x) + k(x, X) k(X, X)^{-1} (y - \mu(X))$$

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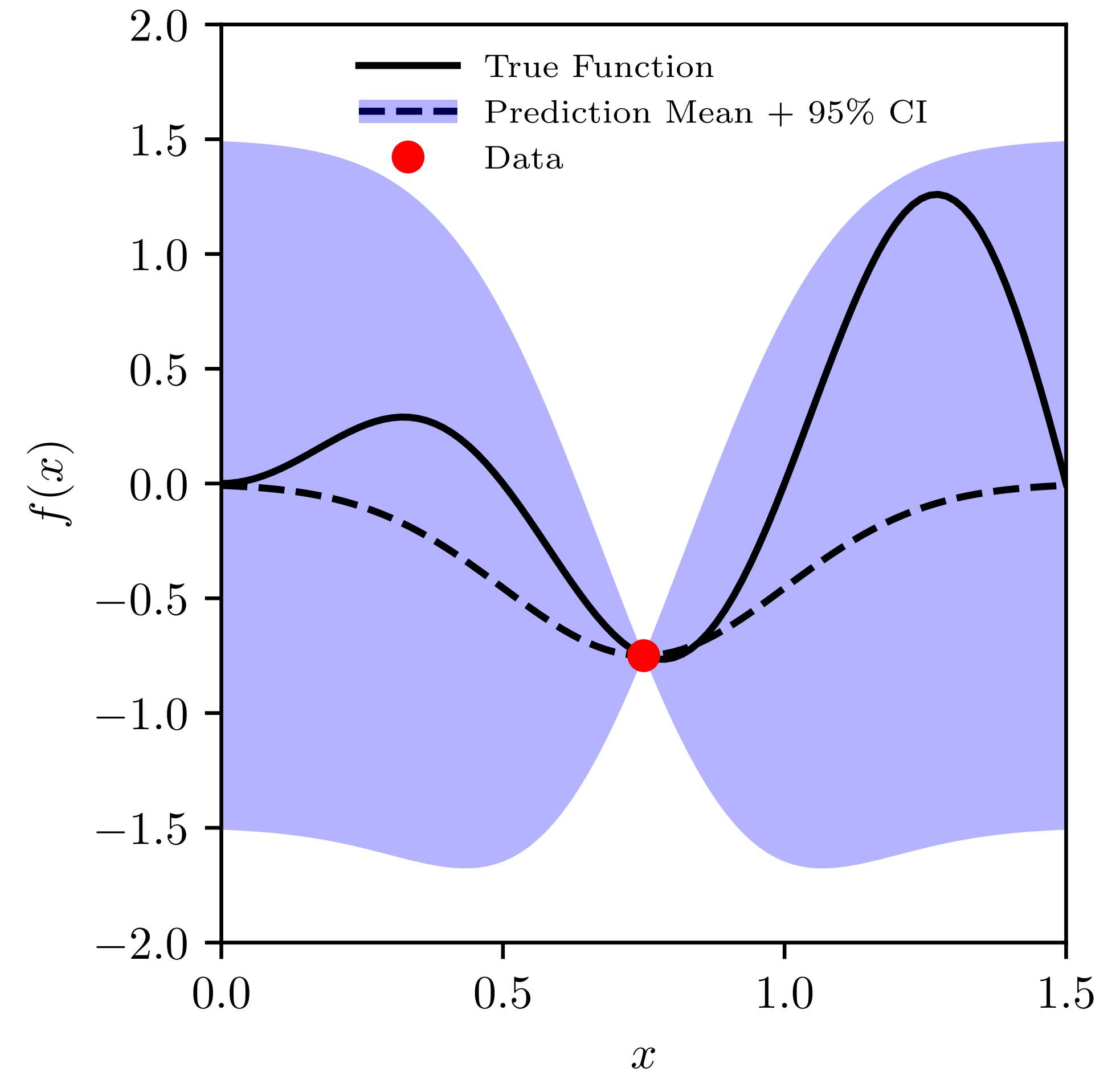
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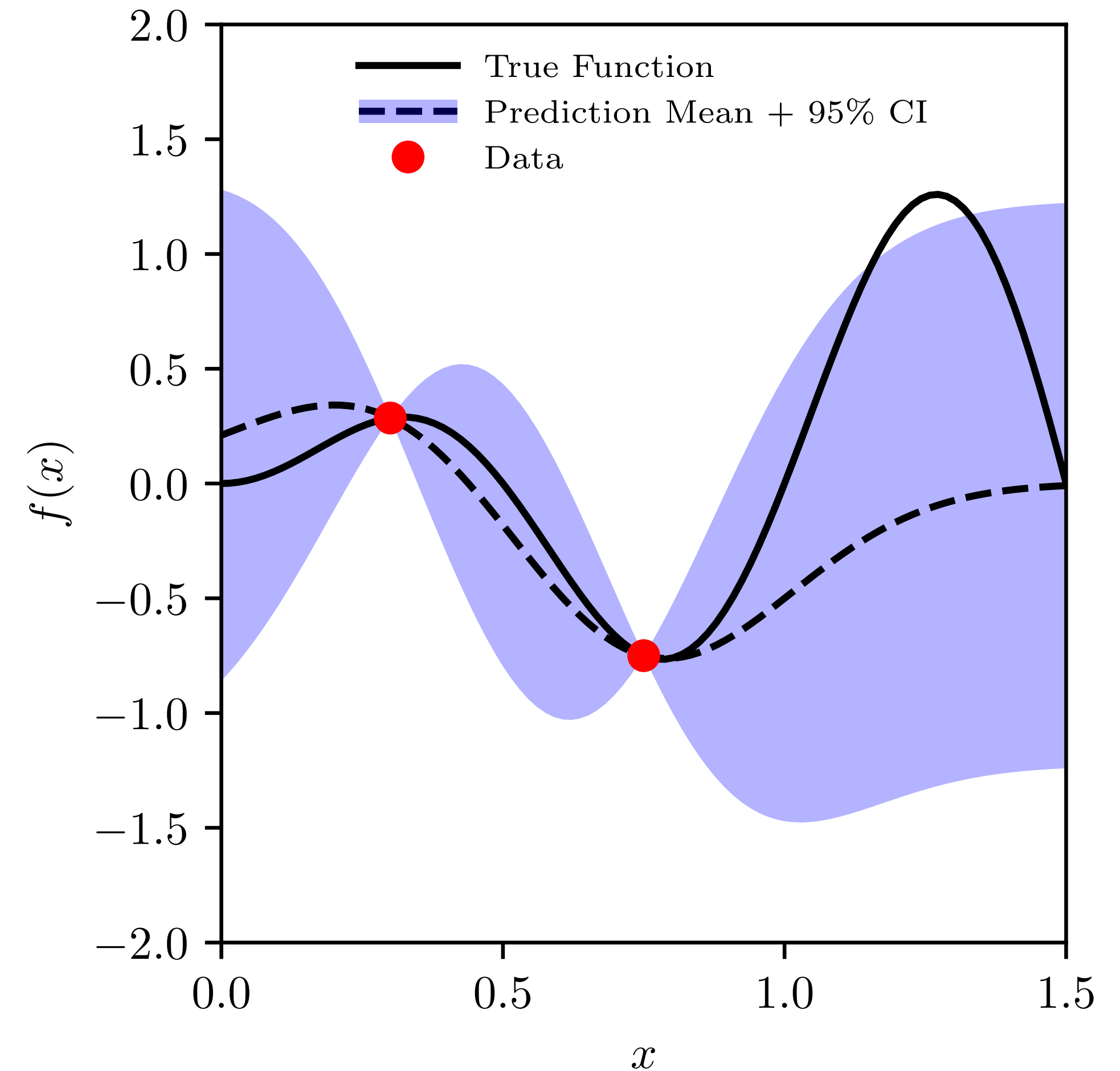
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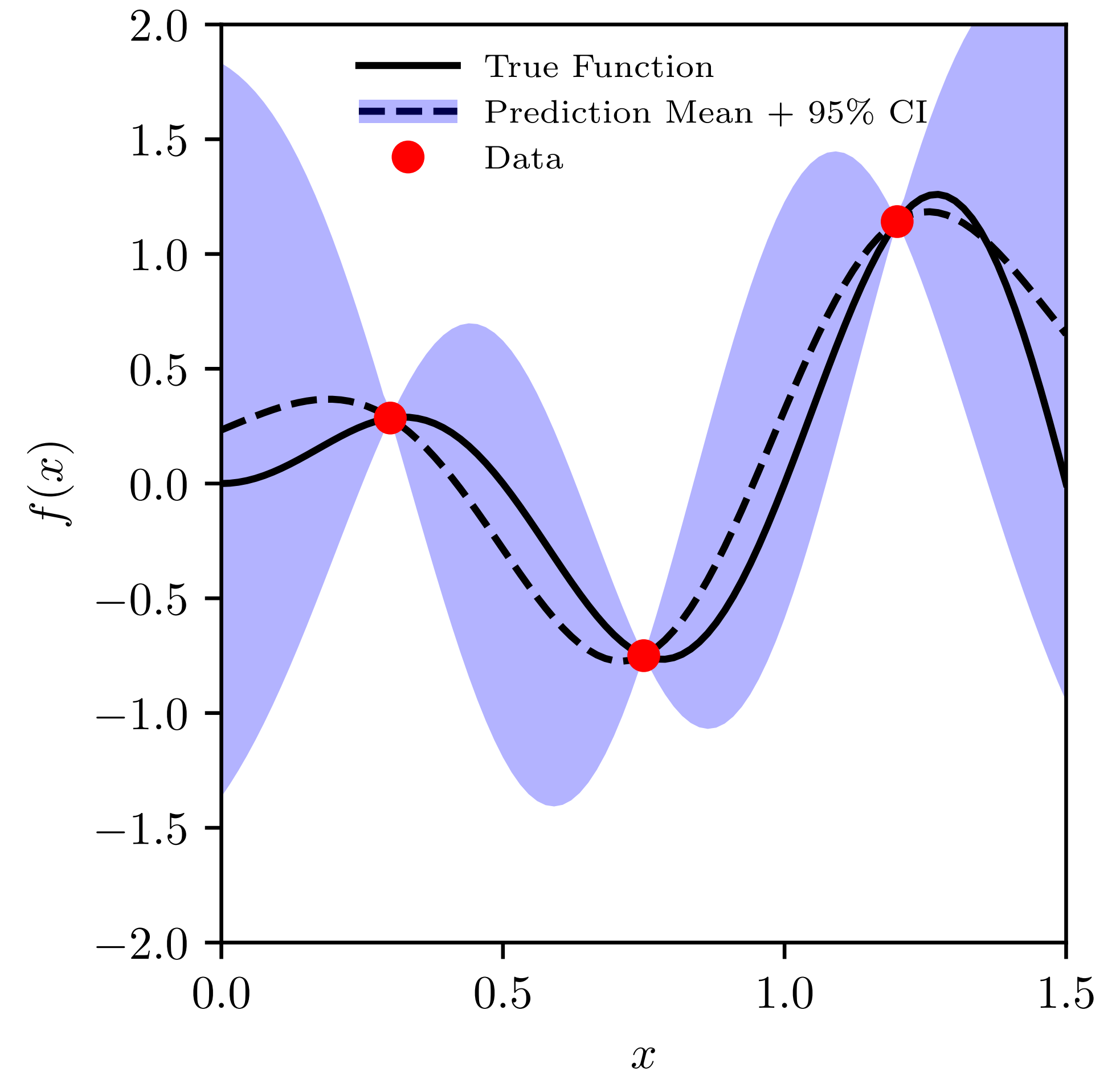
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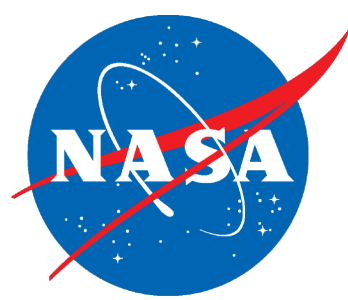
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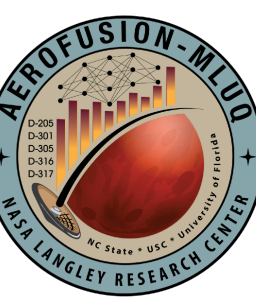
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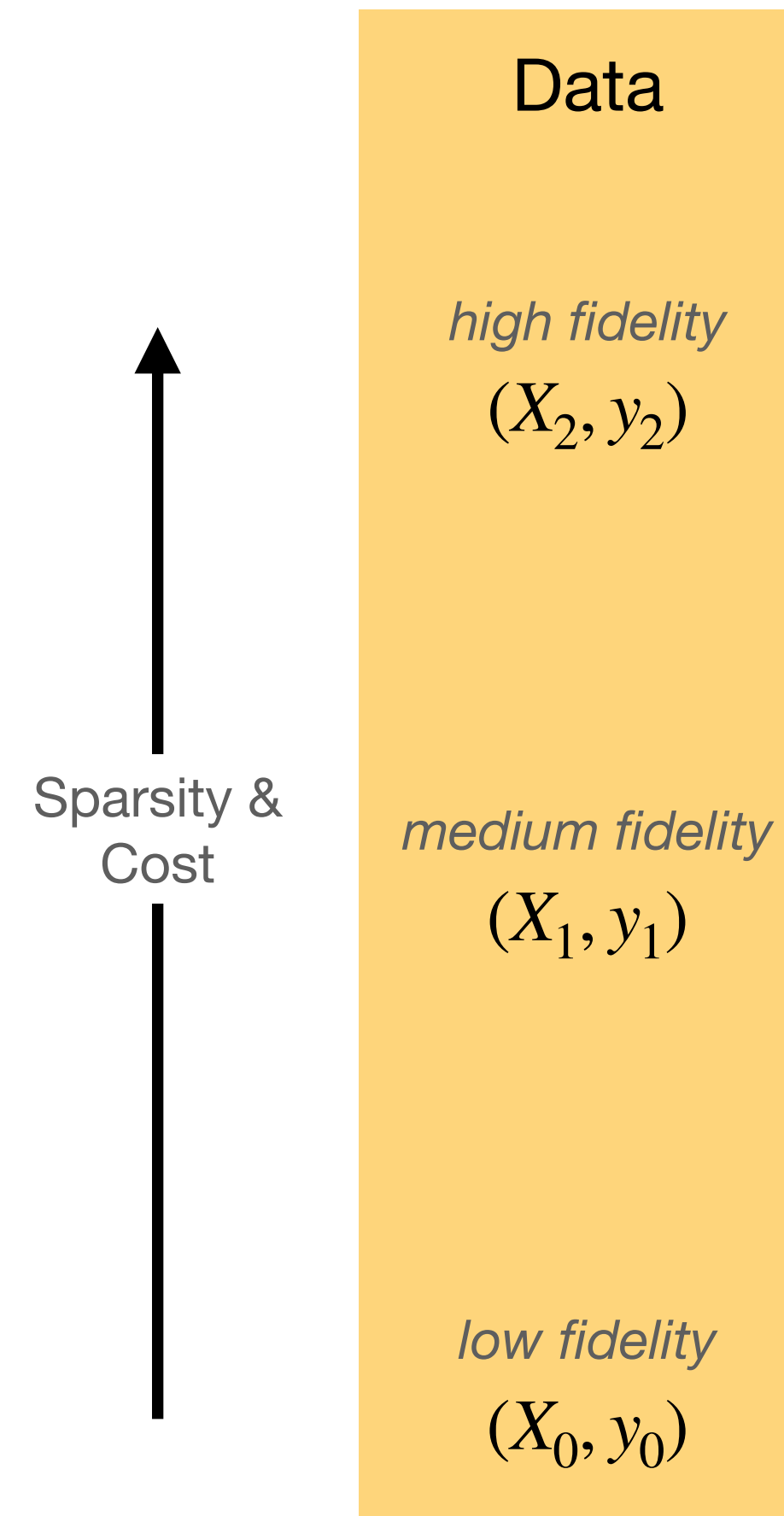


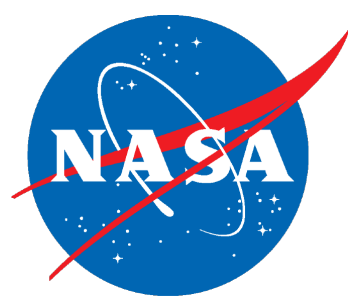


Multifidelity Gaussian Process Regression

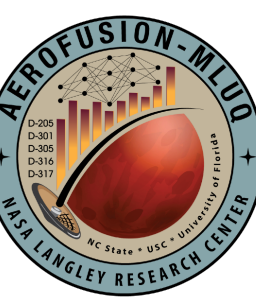


- Multiple sources of data of increasing fidelity
 - Increasing CFD mesh resolutions
 - Heat flux correlations and 3D CFD solutions
 - CFD solutions and wind tunnel data

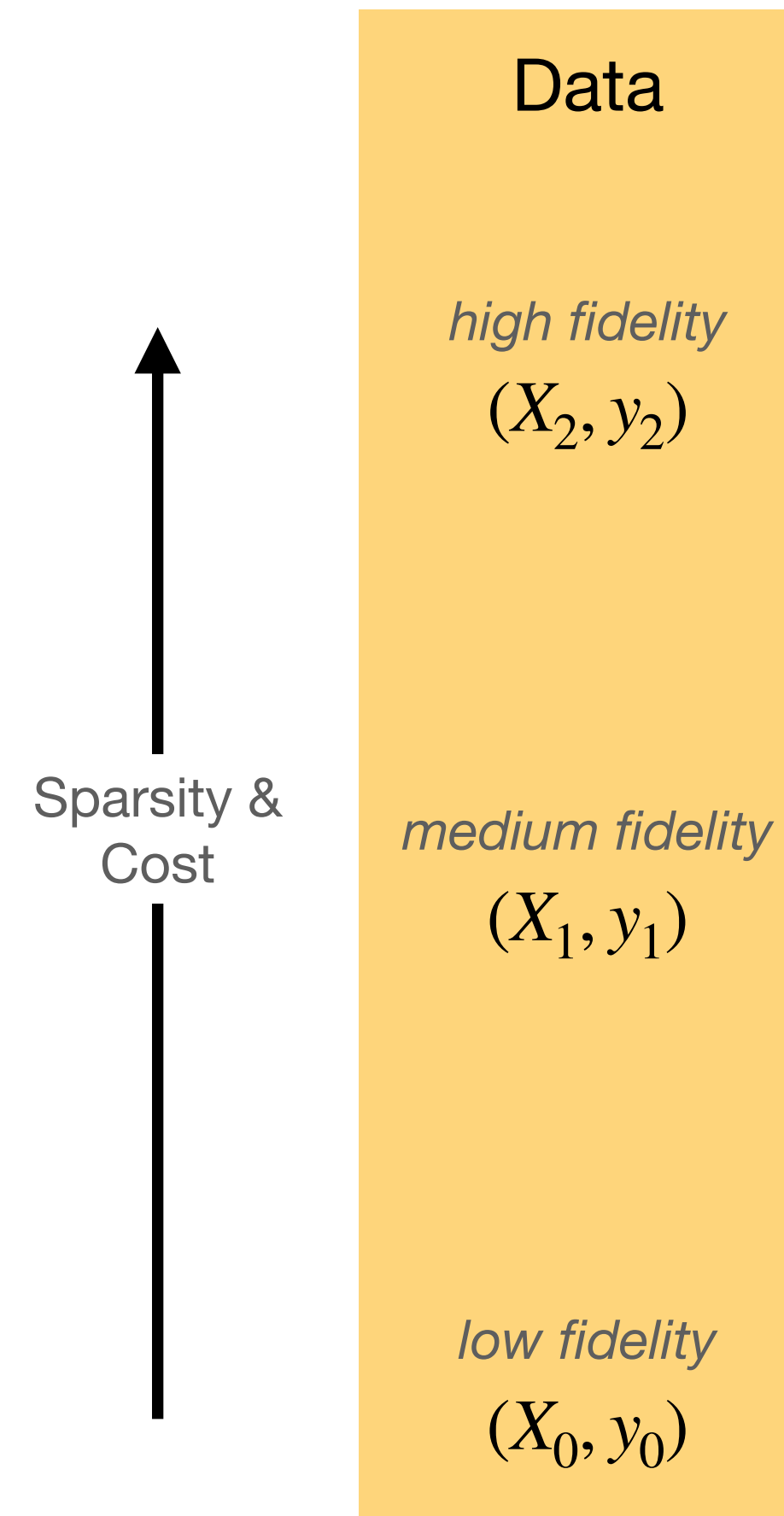




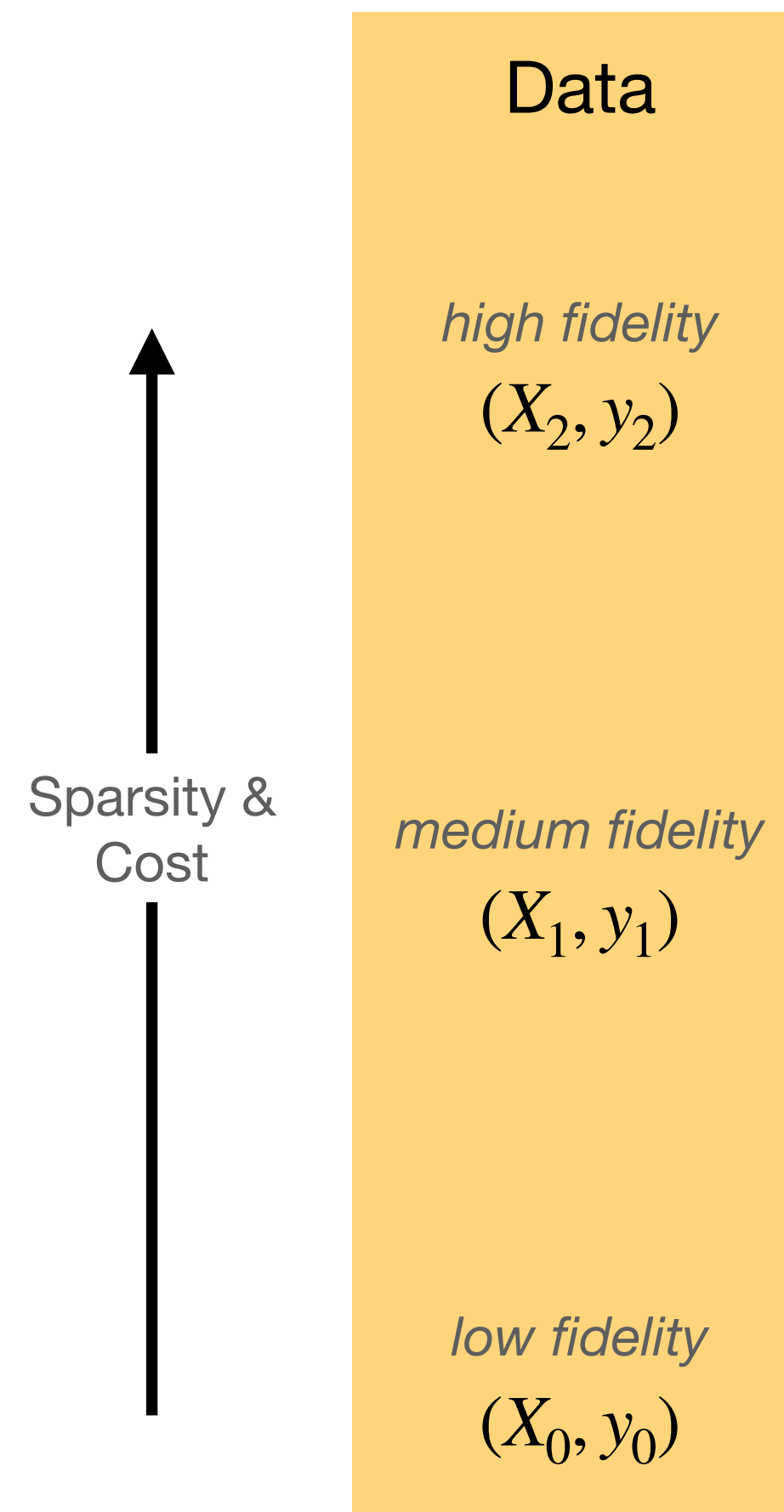
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- Low fidelity is dense and cheap to obtain, high fidelity is sparse and expensive



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$$f_k(x) = \rho_k f_{k-1}(x) + \delta_k(x)$$

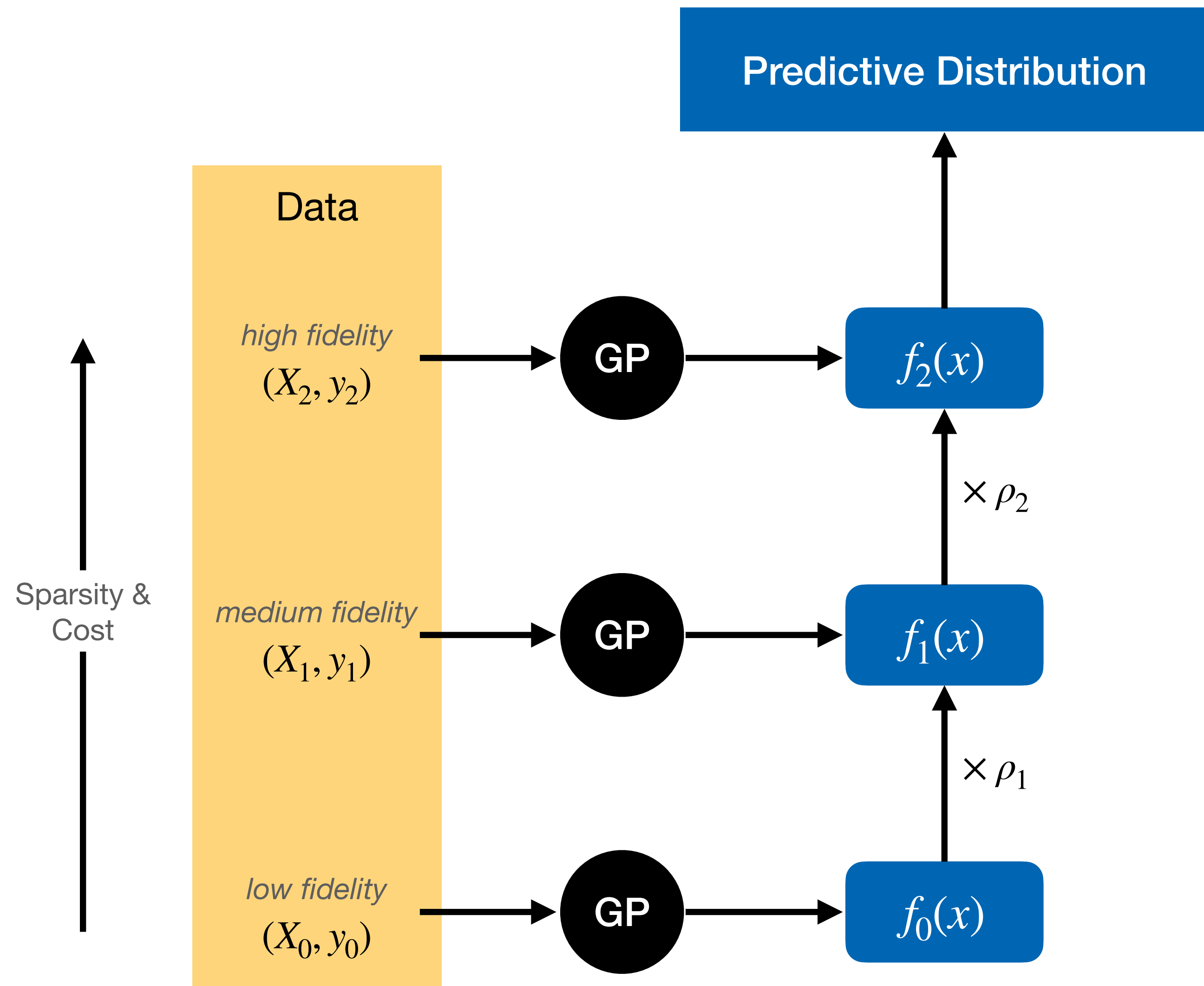


Diagram of the AR1 multifidelity GP model.

[1] Kennedy and O'Hagan. *Biometrika* 87:1-13, 2000.

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- **Requires an obvious hierarchy of fidelity levels!**

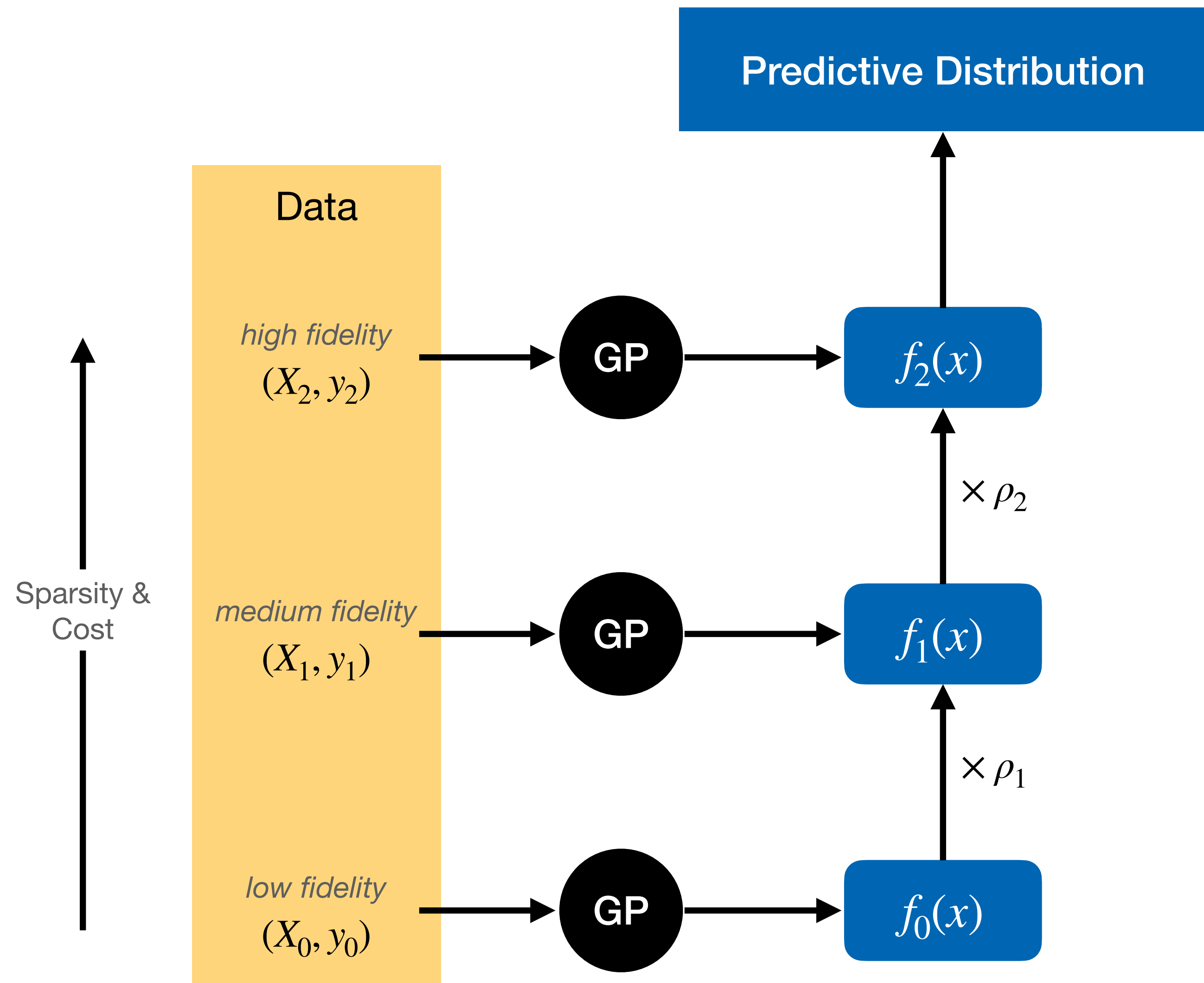
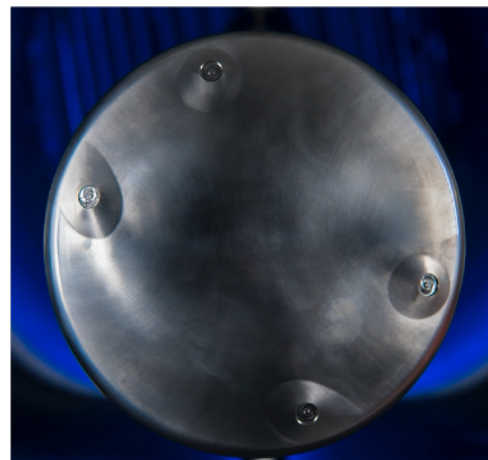


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- Easier to categorize “nominal” and “off-nominal” data

Asymmetric, Smooth



Data

+ *second effect*
 (X_2, y_2)

Symmetric, Rough



+ *first effect*
 (X_1, y_1)

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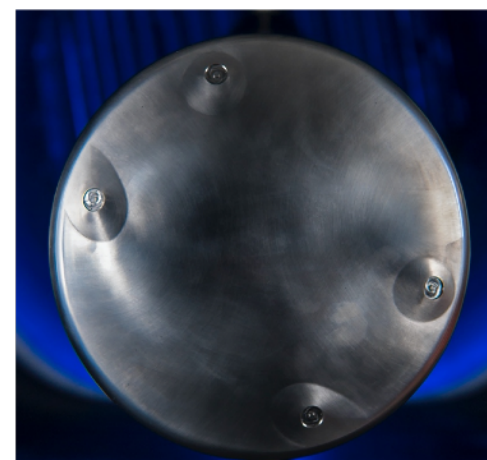
Orion heatshield models used in 133-CA test campaign in the National Transonic Facility

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Predictive Distribution

$$f(x) = f_0(x) + w_1 \Delta f_1(x) + w_2 \Delta f_2(x)$$

Asymmetric, Smooth



Data

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(X_2, y_2)

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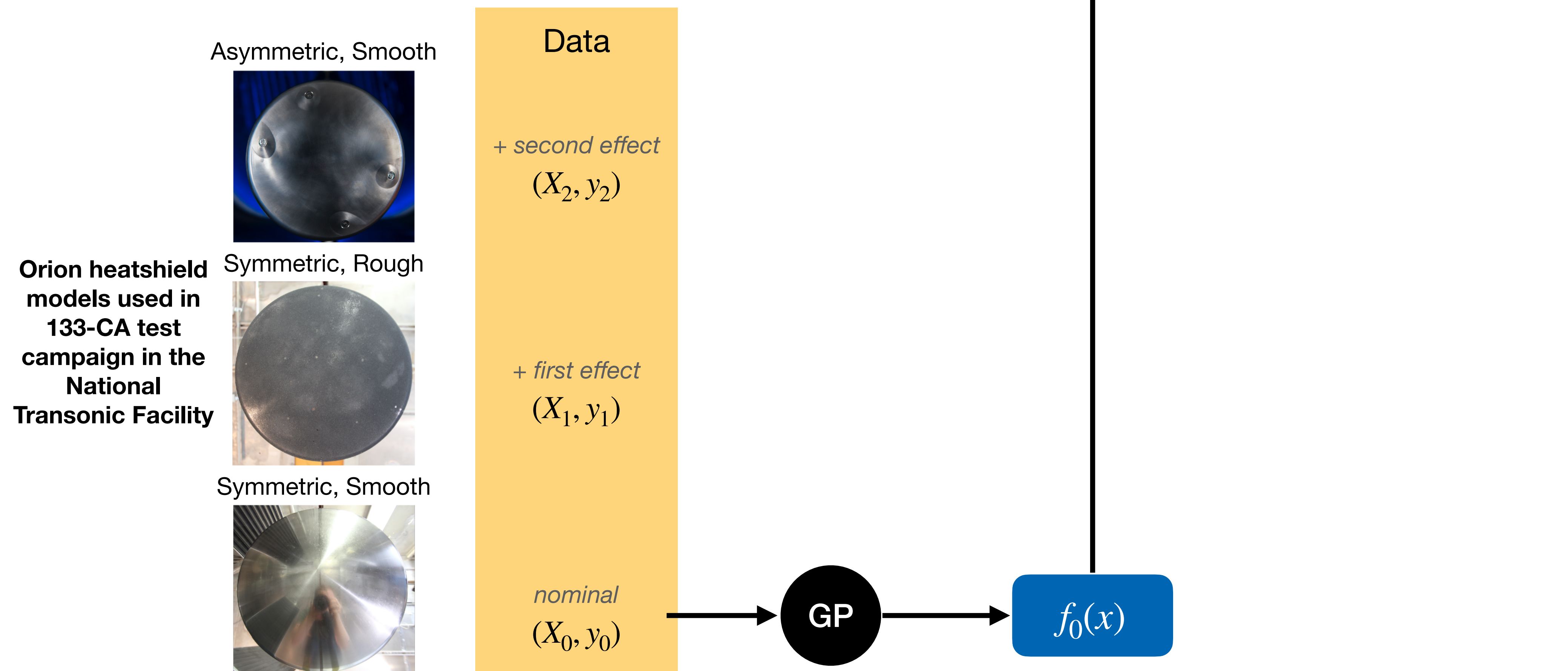
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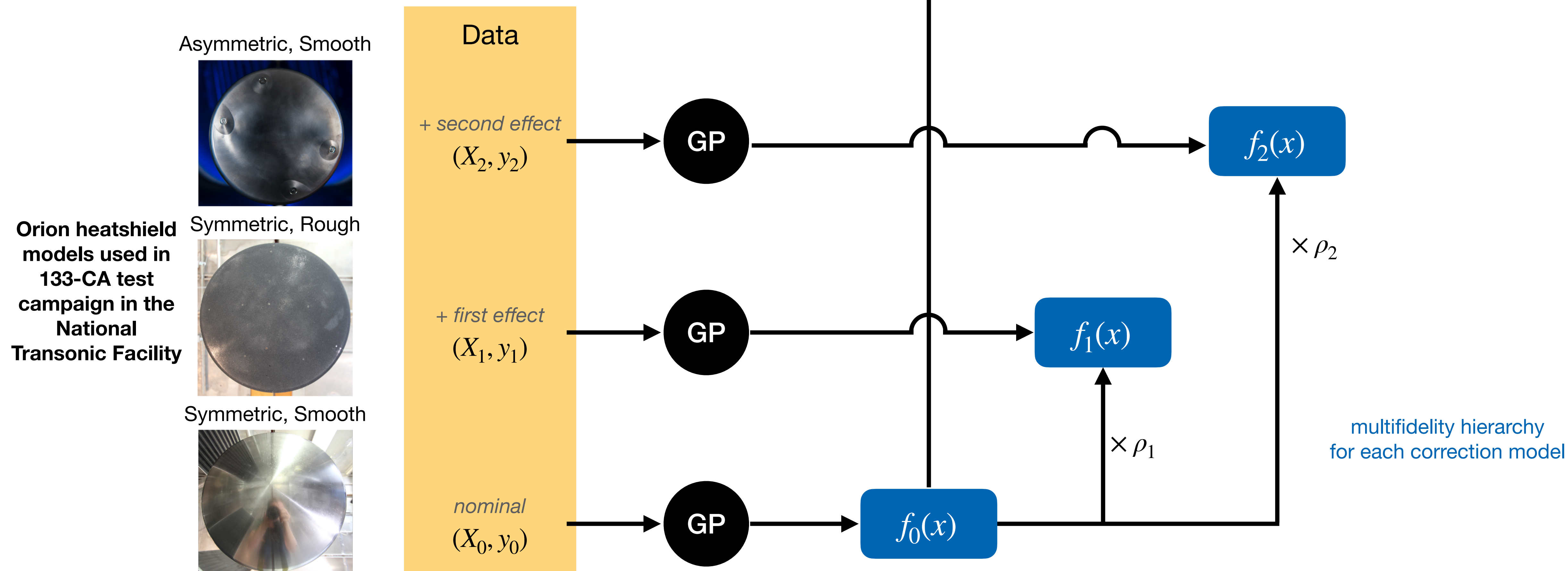
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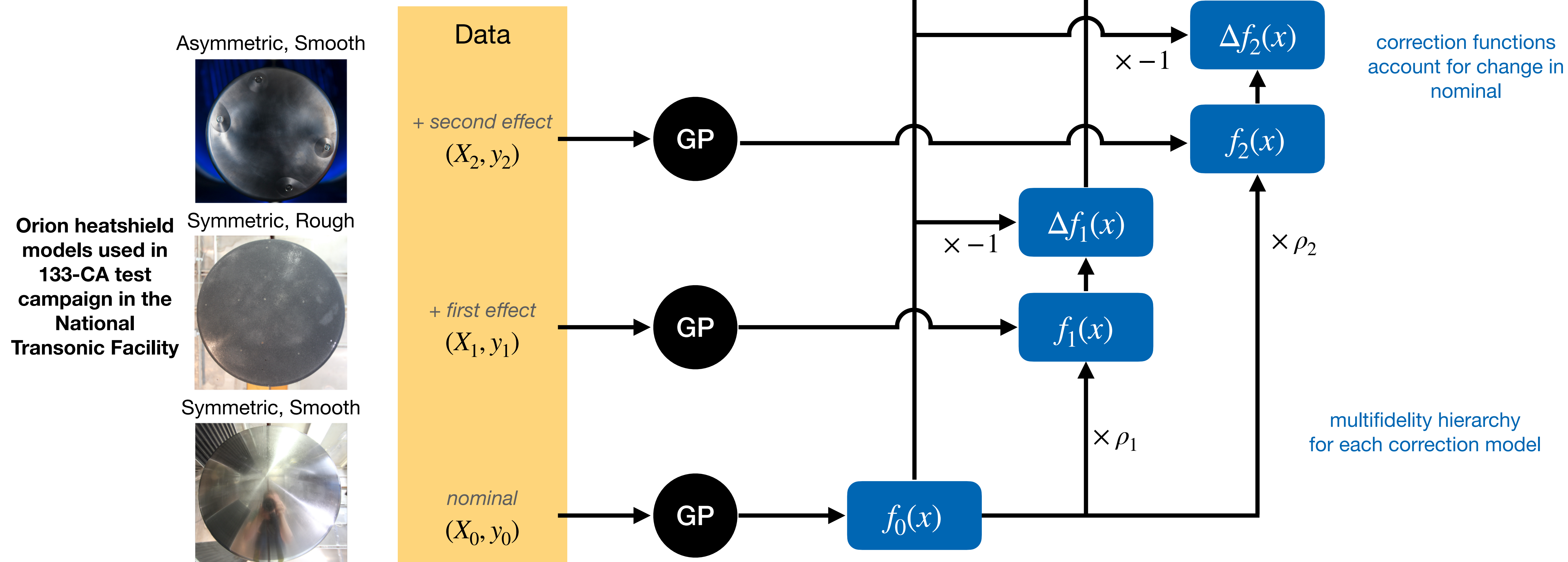
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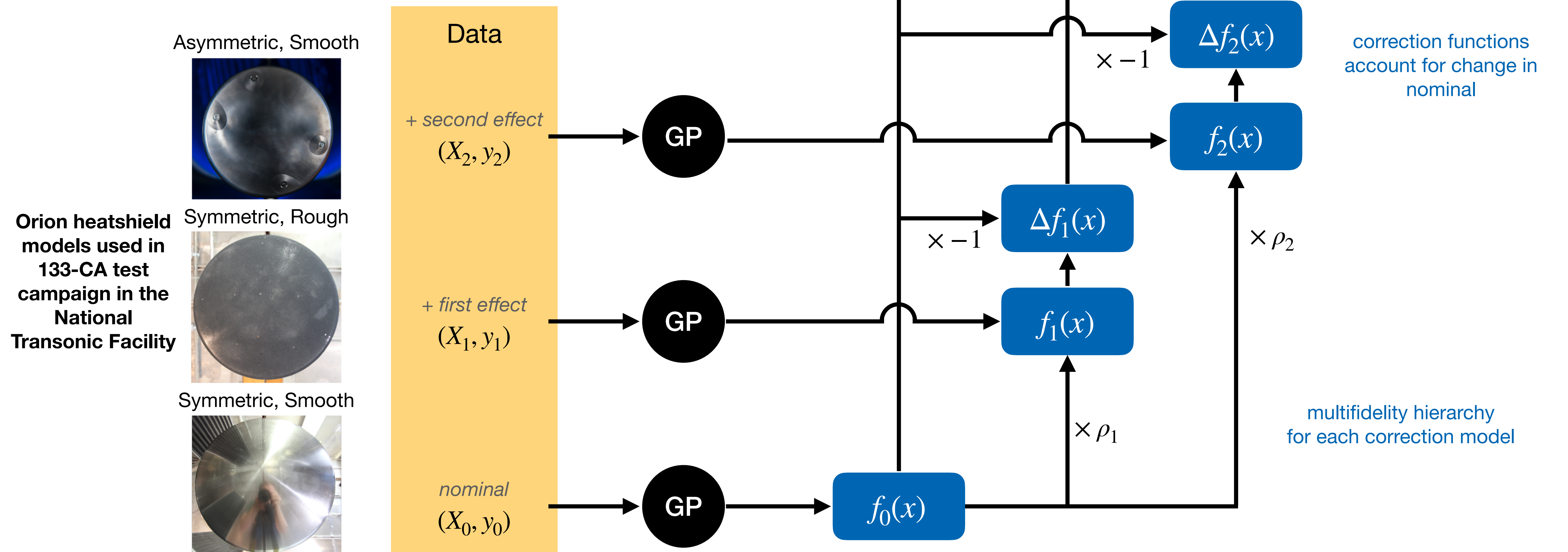
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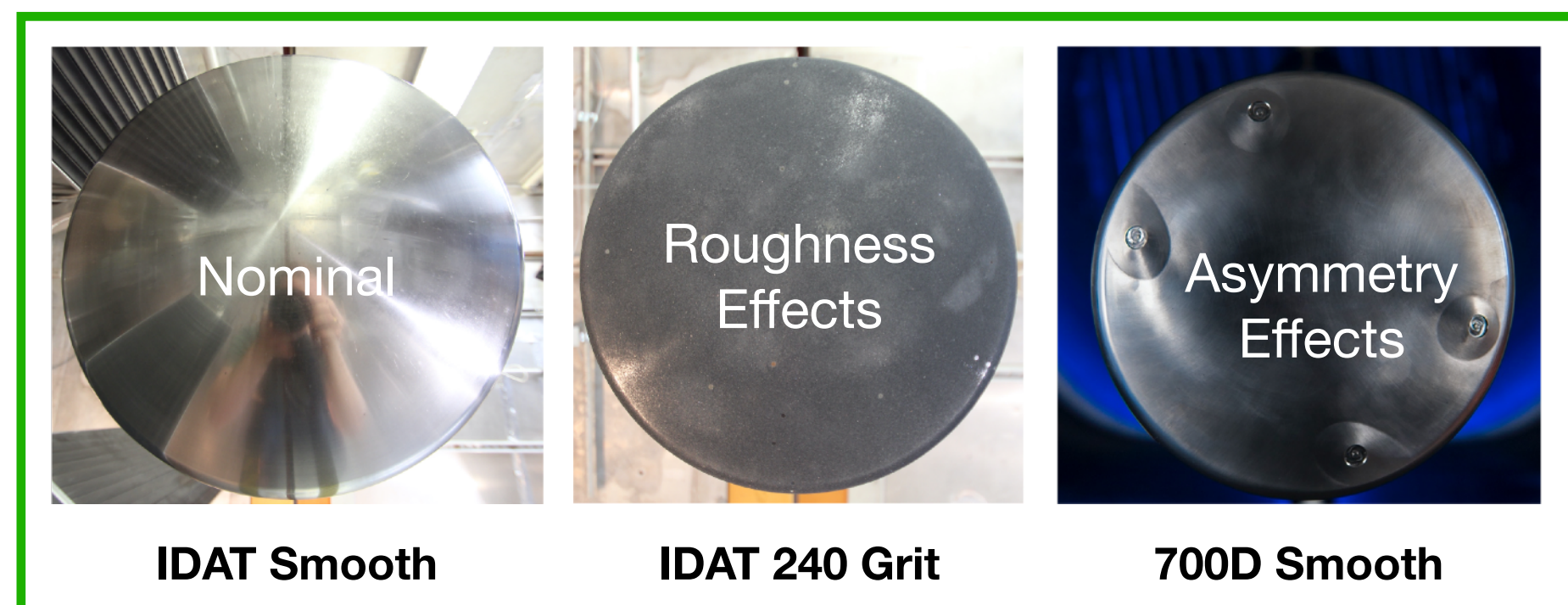
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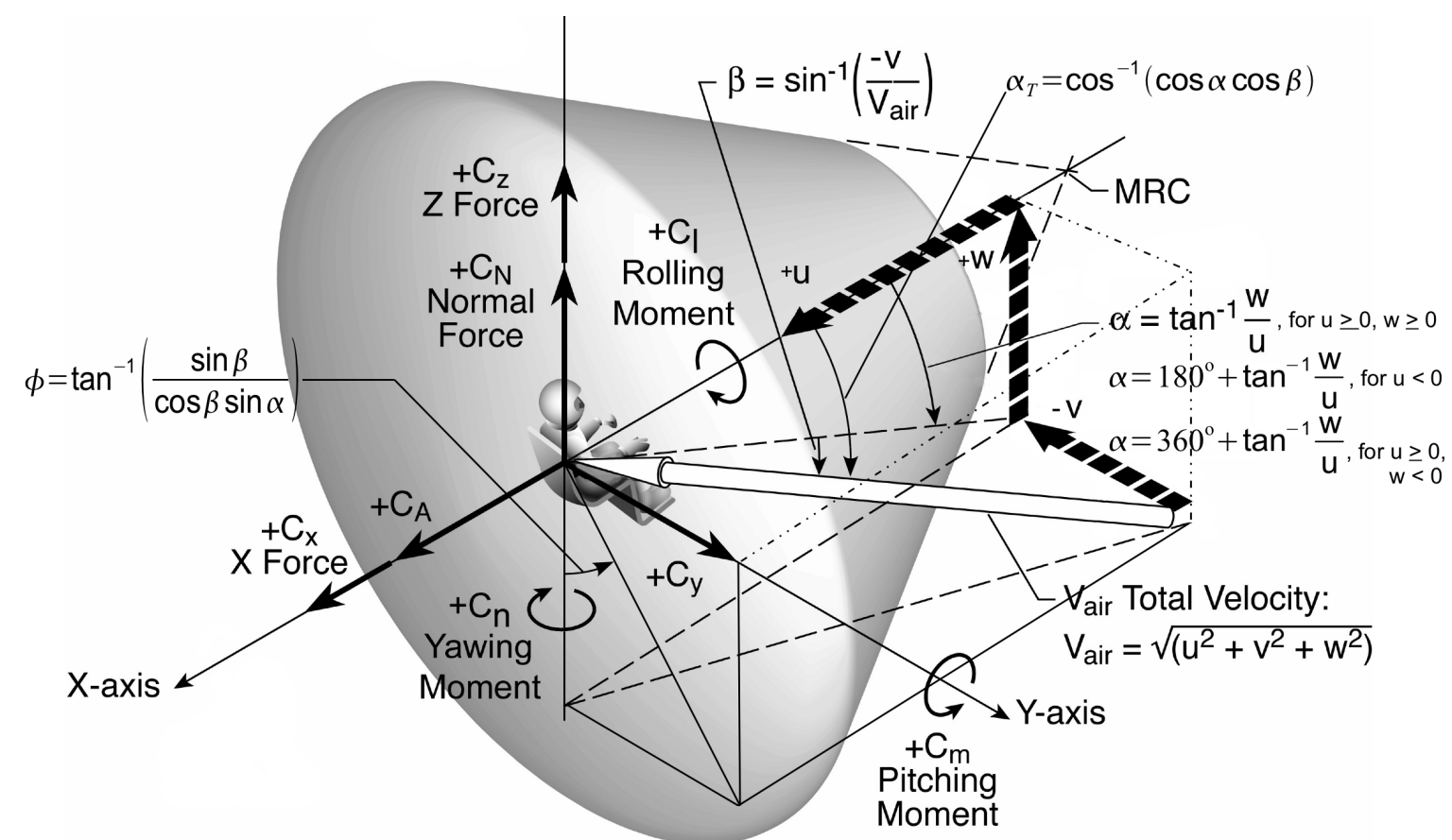
Example datasets taken from Orion NTF testing

- All reported results based on Orion 133-CA test campaign performed in the National Transonic Facility at NASA LaRC [1]

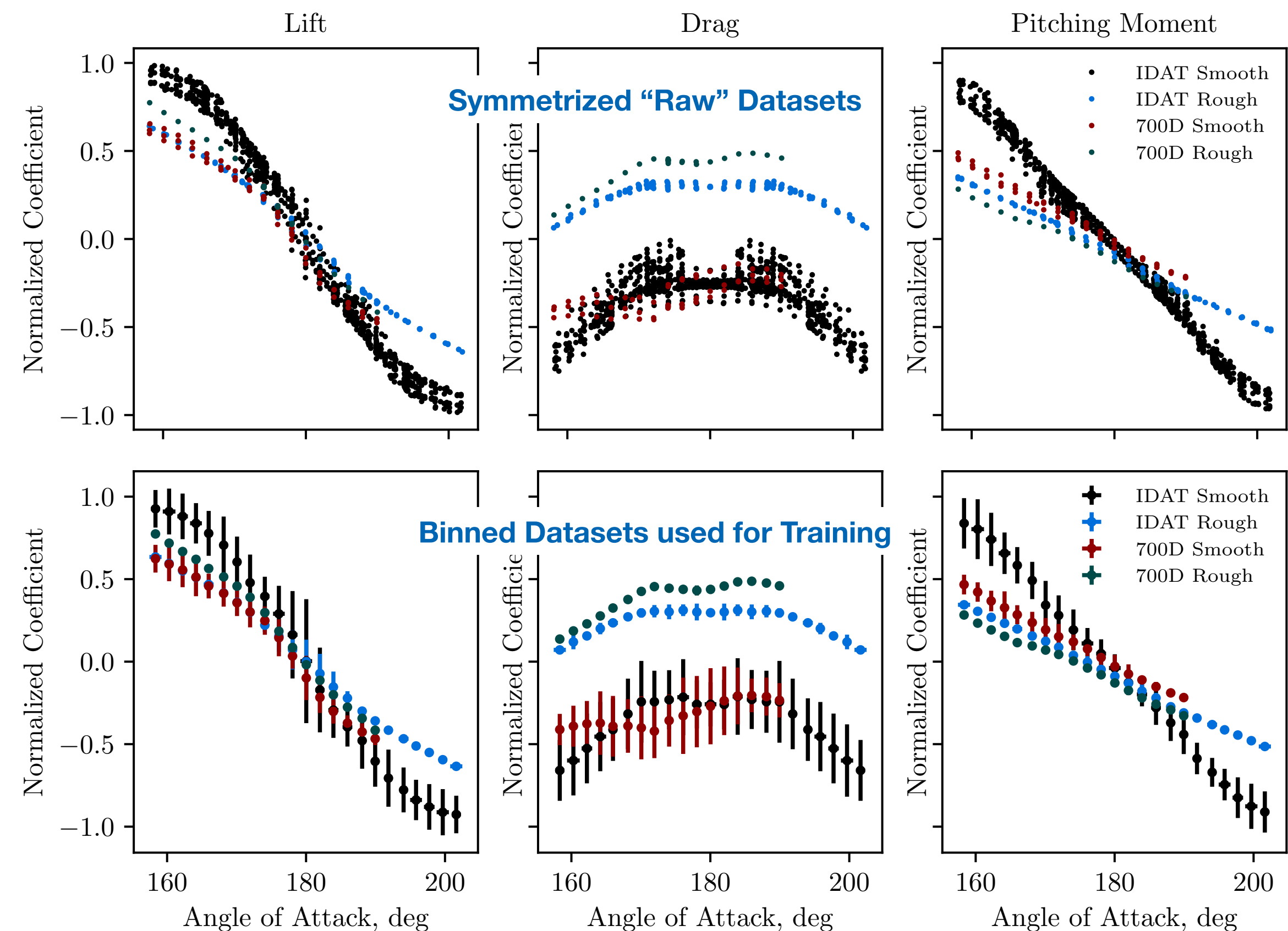
Train



Test

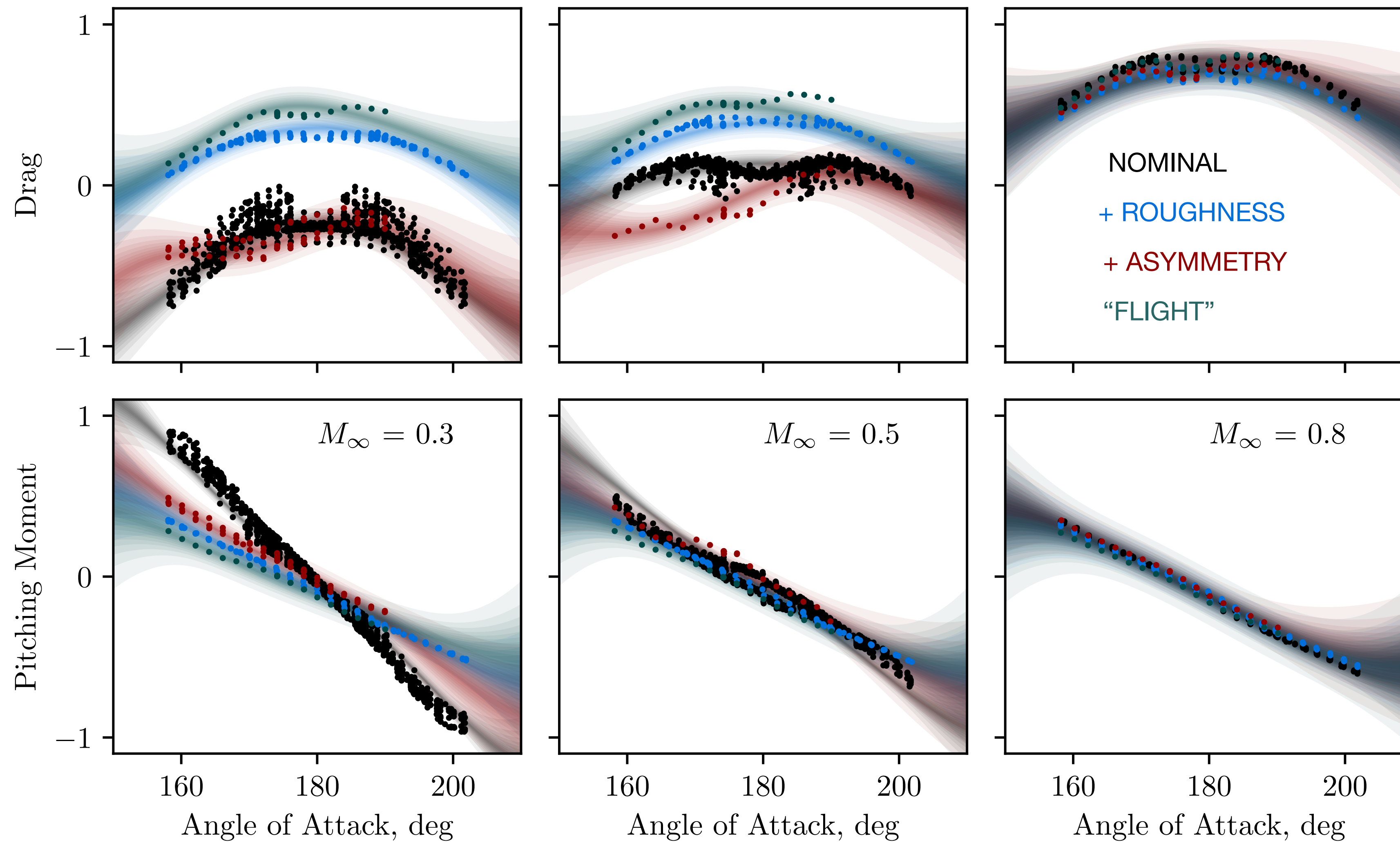


Orion "IDAT" Geometry with coordinates, forces, and moments.



Slices of data around Mach 0.3 and Reynolds 7.5×10^6 .

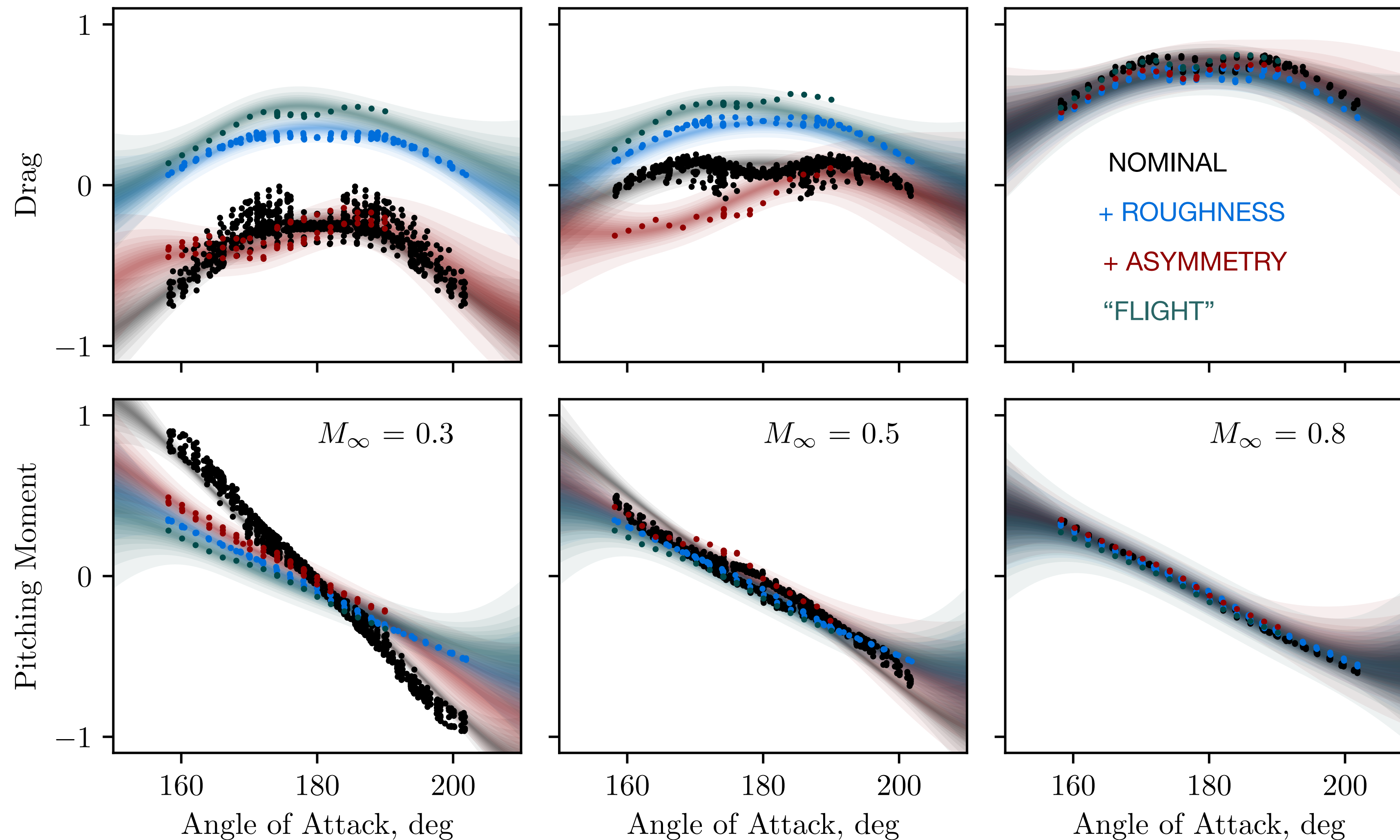
[1] Brauckmann. CAP WTT Report EG-CAP-12-65, NASA LaRC, 2022 (under preparation).



Aerodynamic coefficient function distributions at 3 Mach numbers and Reynolds 7.5×10^6 .

Model distributions compared to data

excellent agreement
with “noisy” data

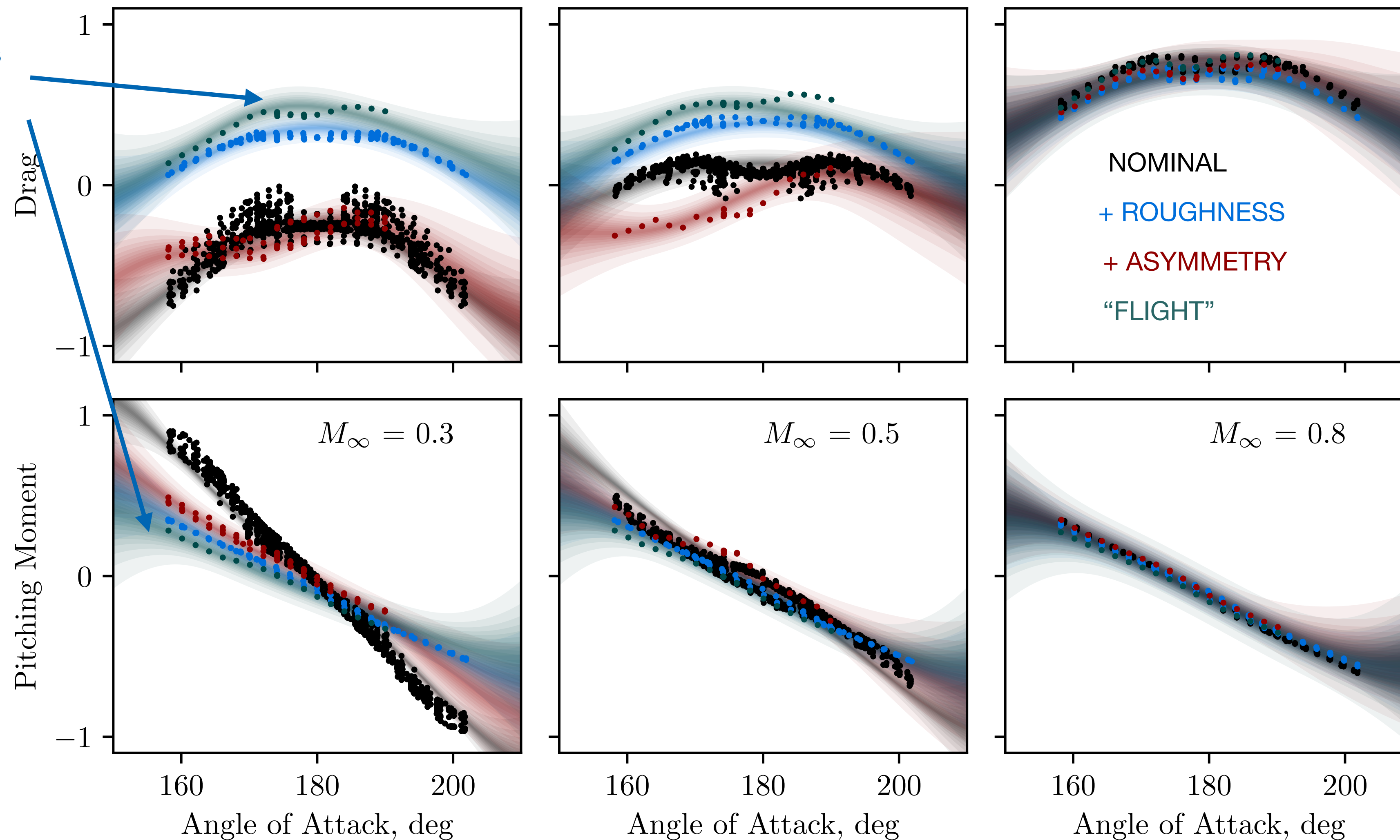


Aerodynamic coefficient function distributions at 3 Mach numbers and Reynolds 7.5×10^6 .

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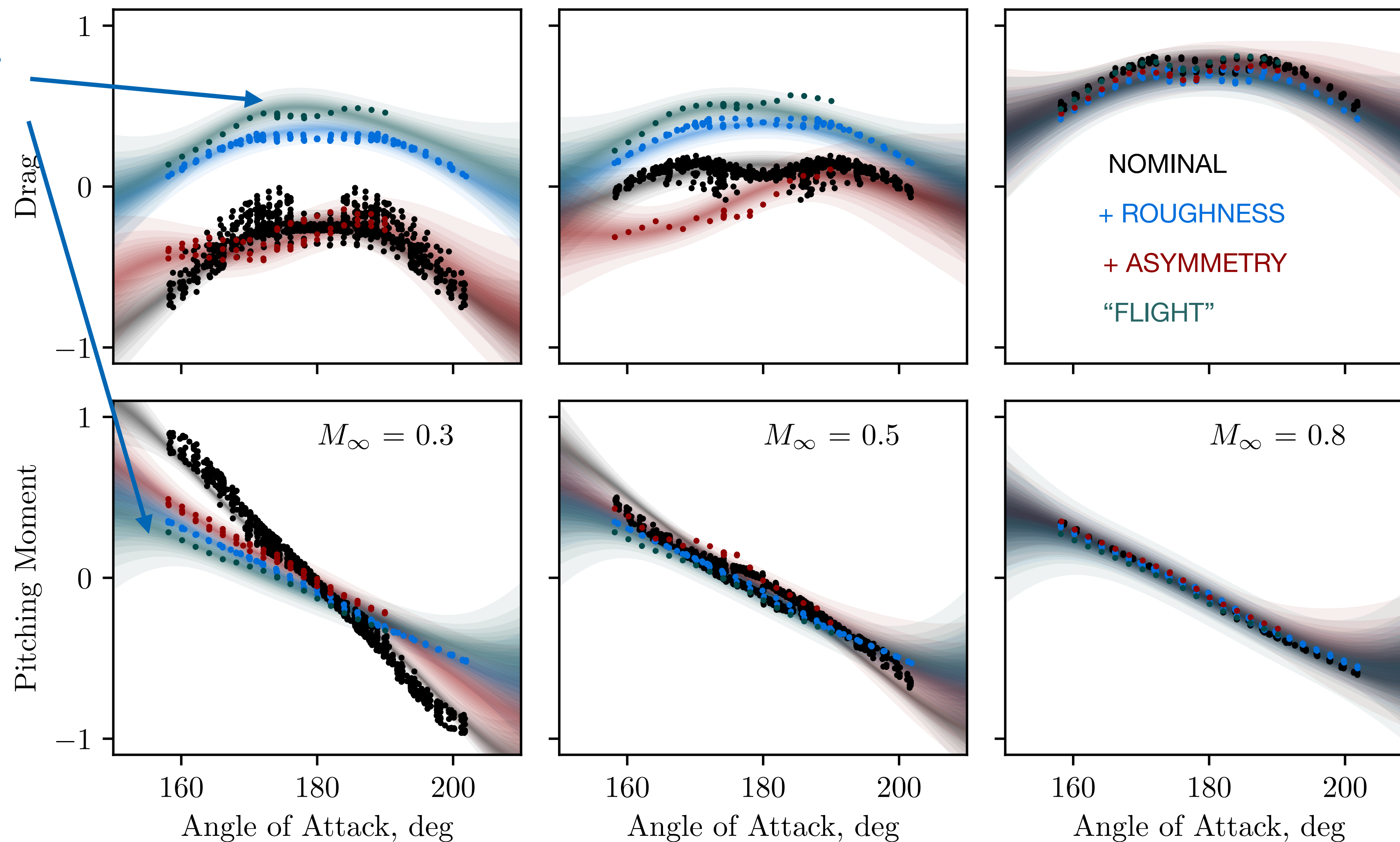


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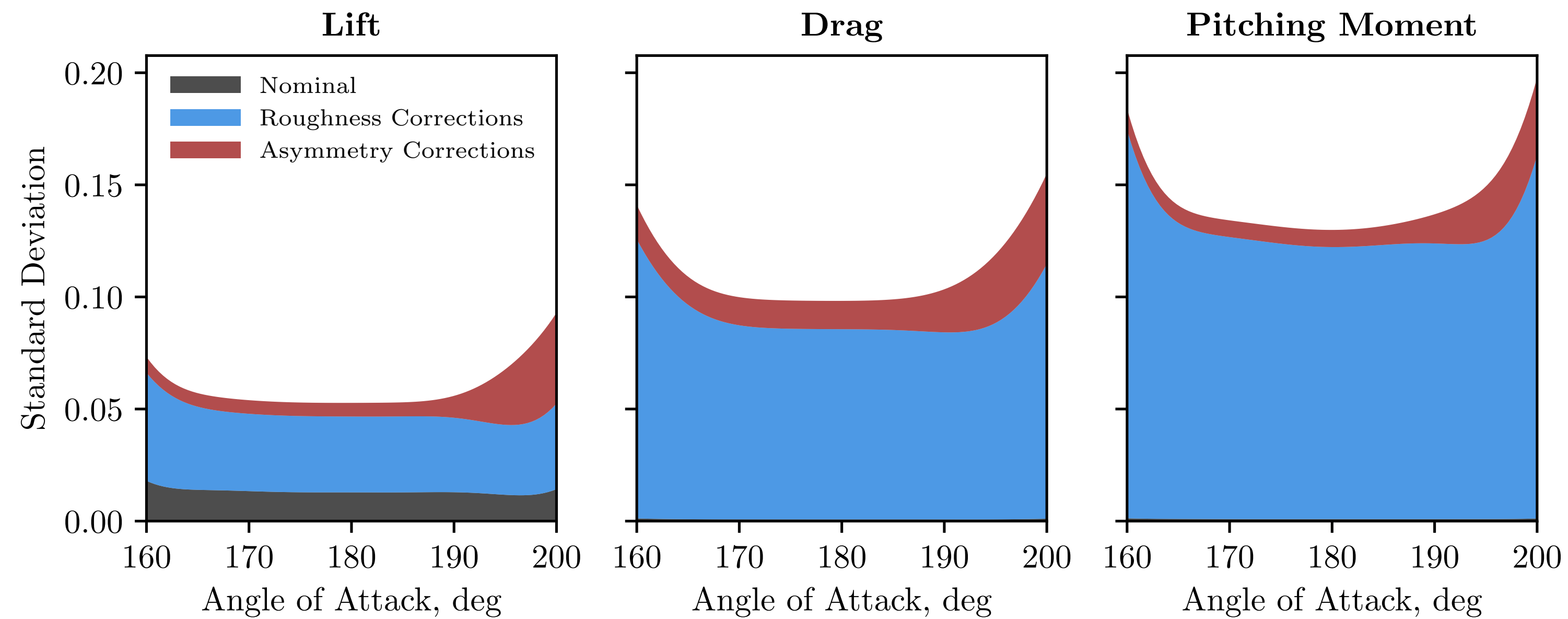
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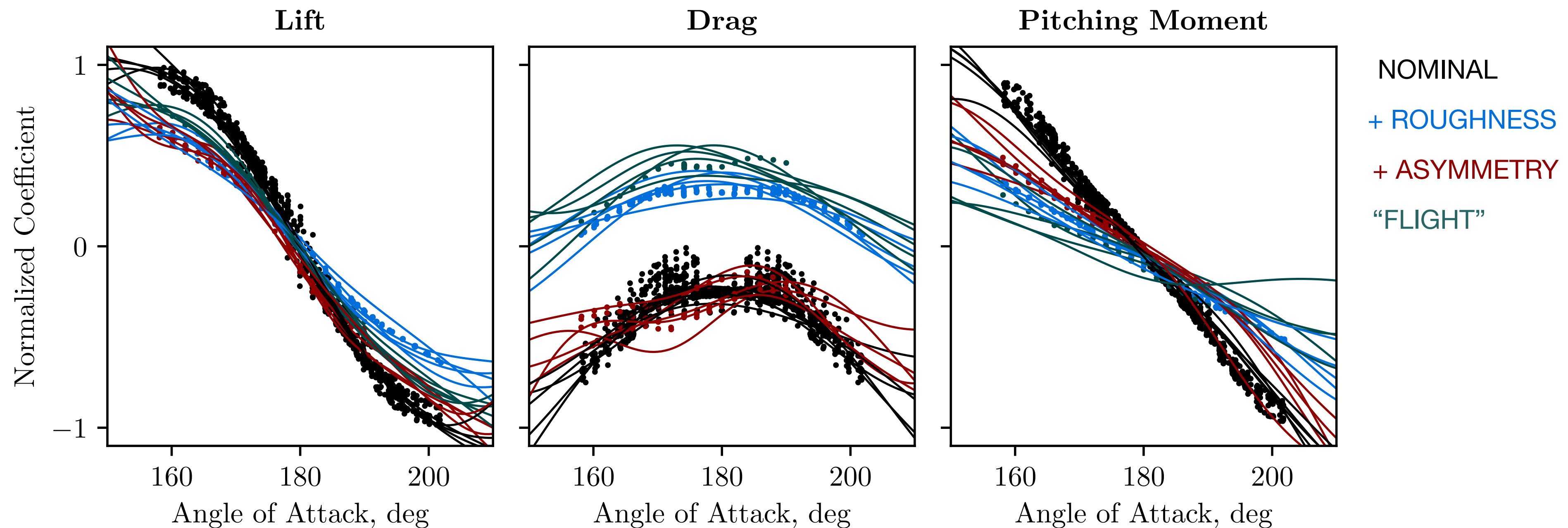


Aerodynamic coefficient function distributions at 3 Mach numbers and Reynolds 7.5×10^6 .

- Can attribute uncertainty from different sources to overall uncertainty in predictive distribution
- Roughness corrections account for significant uncertainty in drag and pitching moment
- Only uncertainties present in data, does not include those associated with factors not considered here

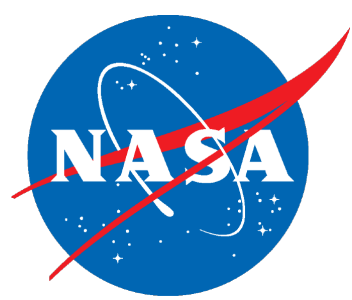


Uncertainty buildup from different sources at Mach 0.5 and Reynolds 7.5×10^6 .

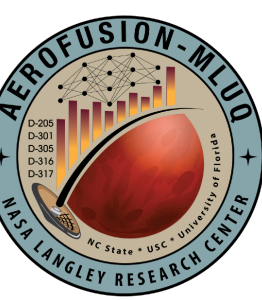


Aerodynamic coefficient function samples at Mach 0.5 and Reynolds 7.5×10^6 .

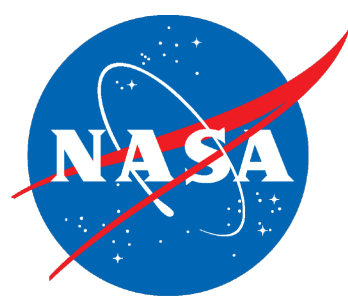
- State-of-the-practice uses random dispersion offsets from the nominal for Monte Carlo trajectory simulations
- Proposed approach allows for function sampling that is more consistent with underlying conditional distribution
- **Each function is a plausible explanation of the data, reproduces conditional distribution in aggregate**



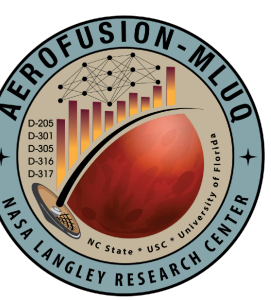
Probability distributions for derived quantities



Model distributions can be used to obtain conditional distributions on derived quantities of interest



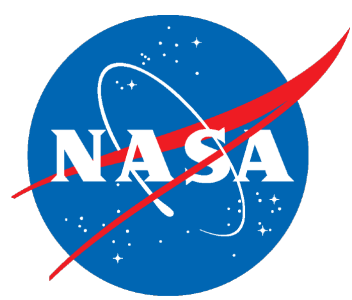
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- **Example:** trim angle of attack found using *Bayes' Theorem*

$$p(\alpha = \alpha_{\text{trim}}) = p(\alpha | C_{m,\text{cg}} = 0) \propto p(C_{m,\text{cg}} = 0 | \alpha) p(\alpha)$$



Probability distributions for derived quantities

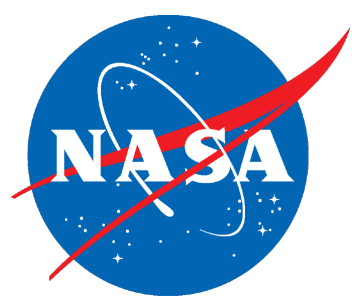


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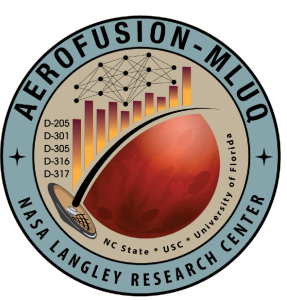
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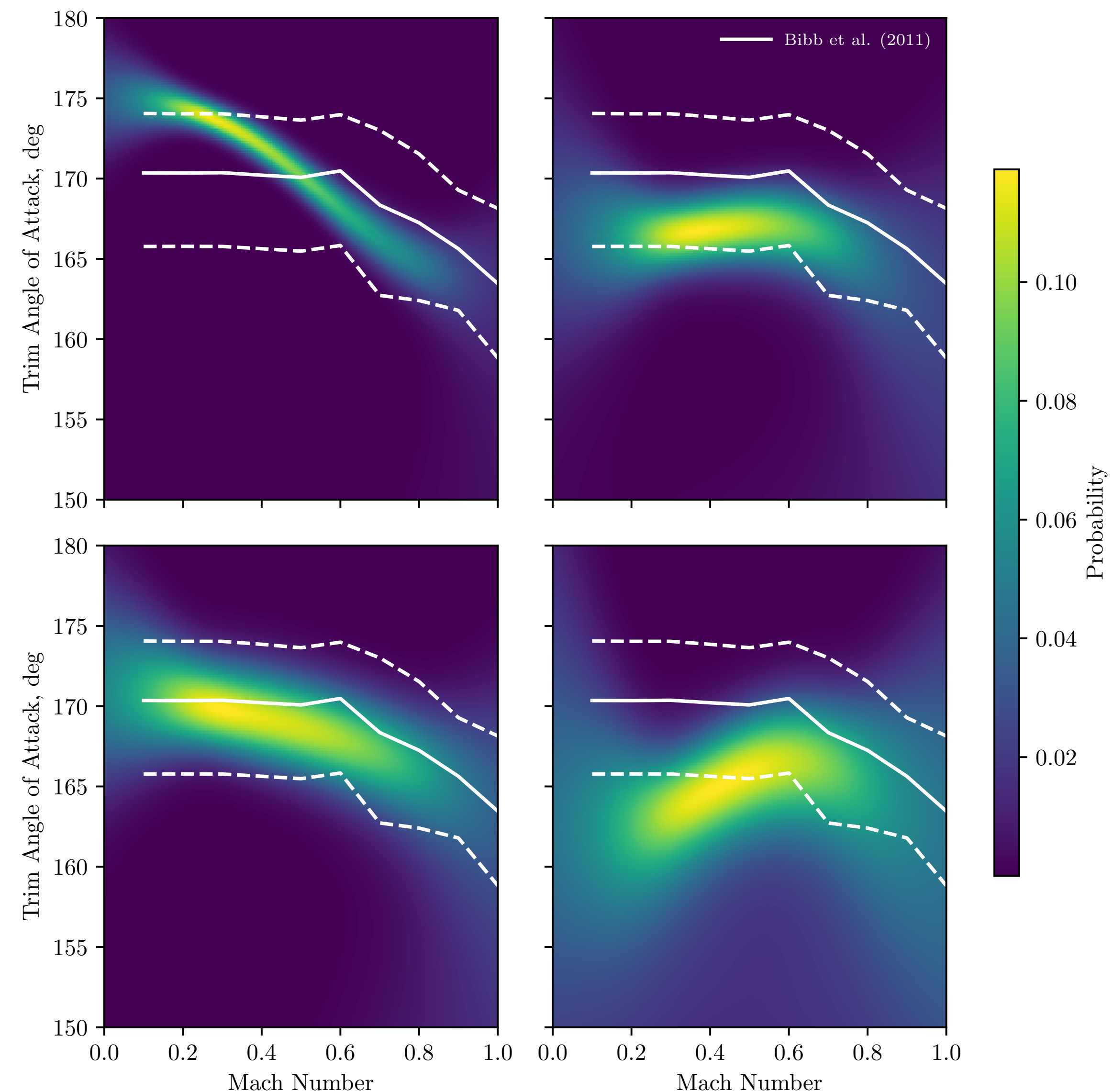
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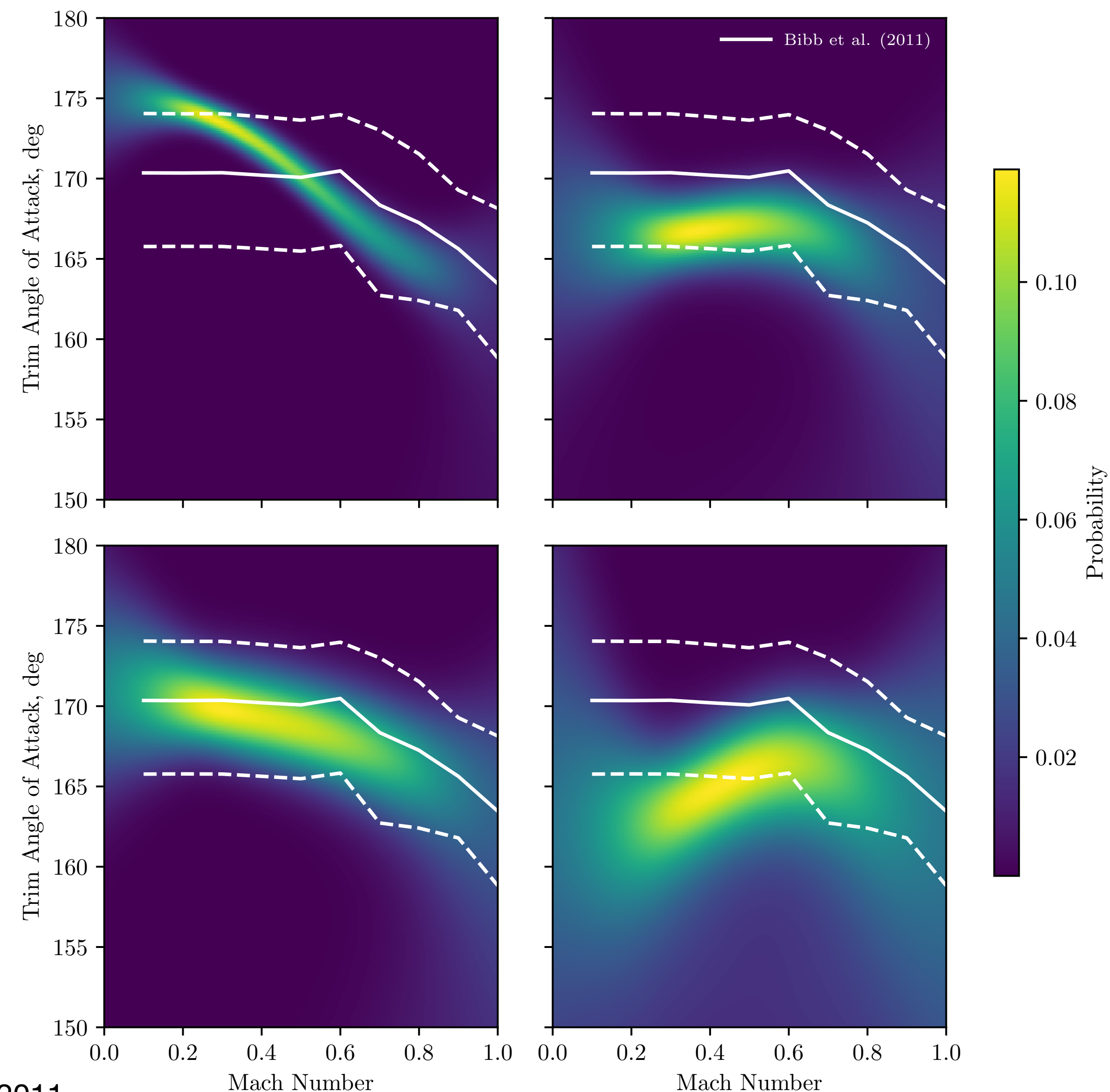


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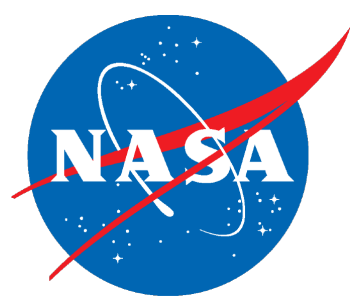
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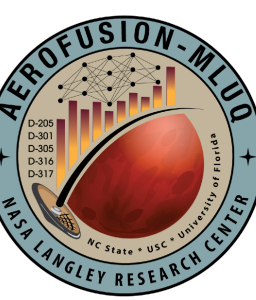
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- Orion v0.60 DB nominal and 100% CI bounds [1] provided for comparison

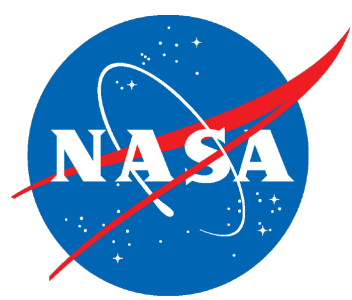


[1] Bibb, Walker, Brauckmann, Robinson. *29th AIAA Applied Aero. Conf.*, No. 2011-3507, 2011.



Concluding Remarks

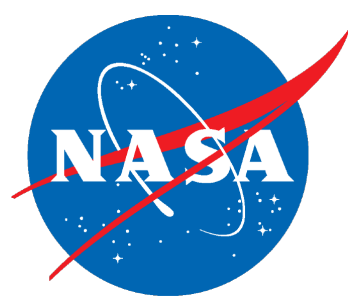




Concluding Remarks



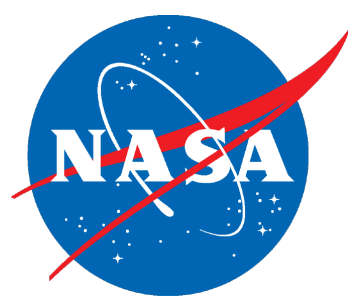
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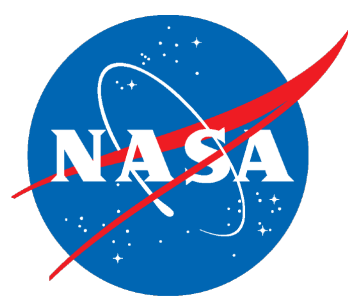
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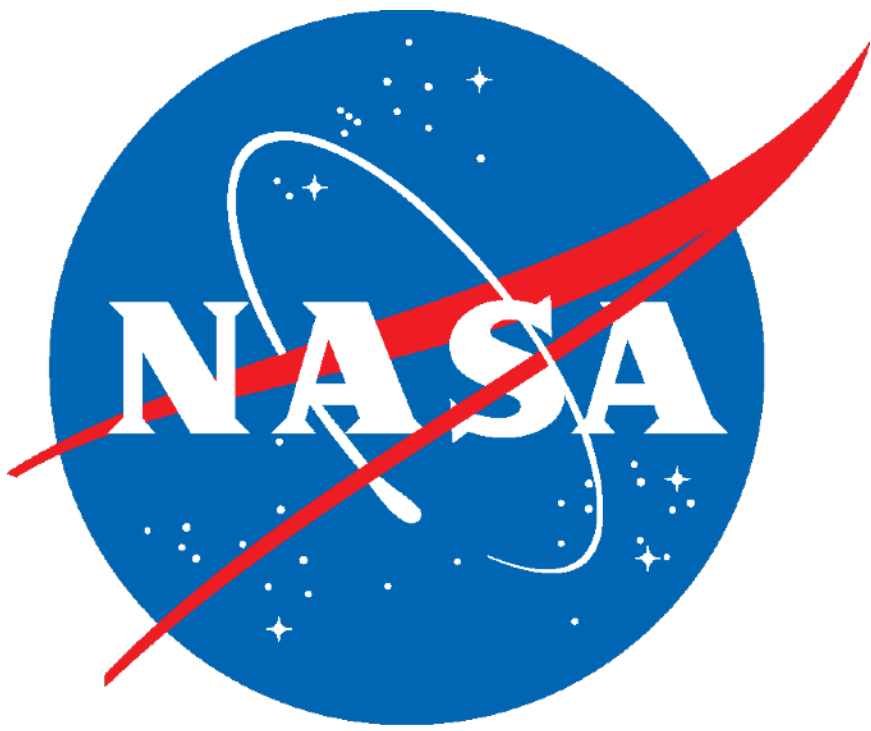
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 - Improve model accuracy by incorporating nonlinear basis functions into correction weights
 - Incorporate nonlinear autoregressive GPs



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- Increased flexibility for aerodatabase designers to incorporate expert judgement with a consistent treatment of uncertainty
- Future improvements:
 - More complex fidelity structures, incorporating more corrections
 - Improve model accuracy by incorporating nonlinear basis functions into correction weights
 - Incorporate nonlinear autoregressive GPs
- When no flight data is available, could attempt to use previously flown vehicles as surrogate for correction weights



Multihierarchy Gaussian Process Models for Probabilistic Aerodynamic Databases using Uncertain Nominal and Off-Nominal Configuration Data

2023 AIAA SciTech Forum
January 23-27, 2023 - National Harbor, Maryland

James B. Scoggins, Thomas J. Wignall, Tenavi Nakamura-Zimmerer, Karen Bibb
NASA Aerofusion-MLUQ Early Career Initiative, NASA Langley Research Center

