Optical communications operations concept for the Artemis II crewed mission to the Moon

<u>Farzana I. Khatria</u>, Michael Bay^b, Jonathon King^c, Jessica Chang^a, Robert T. Schulein^b, Olga Mikulina^a, Jeffrey J. Zinchuk^b, Robert J. McGraw^b, Jacob Gregory^b, Terry Hudson^c, Paul W. Gramm^c, Nikki M. Desch^b, Steven J. Horowitz^b, Bryan S. Robinson^a

^aMIT Lincoln Laboratory, 244 Wood Street, Lexington, MA 02420
 ^bNASA Goddard Space Flight Center, 8800 Greenbelt Road, Greenbelt, MD 20771
 ^cNASA Johnson Space Center, 2101 E NASA Parkway, Houston, TX 77058



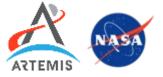


DISTRIBUTION STATEMENT A. Approved for public release. Distribution is unlimited. This material is based upon work supported by the National Aeronautics and Space Administration under Air Force Contract No. FA8702-15-D-0001. Any opinions, findings, conclusions or recommendations expressed in this material are those of the author(s) and do not necessarily reflect the views of the National Aeronautics and Space Administration. © 2022 Massachusetts Institute of Technology.

Delivered to the U.S. Government with Unlimited Rights, as defined in DFARS Part 252.227-7013 or 7014 (Feb 2014). Notwithstanding any copyright notice, U.S. Government rights in this work are defined by DFARS 252.227-7013 or DFARS 252.227-7014 as detailed above. Use of this work other than as specifically authorized by the U.S. Government may violate any copyrights that exist in this work.







- Background
- Optical comm system description
- Link operations plan
- Summary





- Artemis Program is NASA's pathway their Journey to Mars
- Artemis I uncrewed mission successful flight in 2022
 - Radio Frequency comm link, S-band downlink of
 ~ 1 Mbps at lunar ranges
- Artemis II crewed mission to launch in 2024
 - Optical communications capability provides 80 Mbps return link and 20 Mbps forward link from Lunar vicinity
 - Crew brings more mission video up/downloads, file transfers, and real-time video conferencing to the Mission
- Artemis III to land astronauts on Lunar surface



*https://www.nasa.gov/feature/nasa-s-first-flight-with-crew-important-step-on-long-term-return-to-the-moon-missions-to



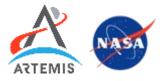


	RADIO	OPTICAL
Beam width	0.5 - 30°	1 - 100 μrad (<0.006°)
Point / Acquire Beam	Open loop	Scan or wide-field sensor (cooperative)
Stabilize / Track Beam	N/A	Optical-based tracking (cooperative), inertial sensors-based tracking, passive jitter rejection
Atmosphere Interactions	Rain can degrade > 20 GHz	Clouds can block lasers; turbulence must be considered
Regulatory / Permissions	Spectrum is regulated and licensed	Unregulated; need care with sensor satellites / aircraft

From: Khatri & Robinson, "Operational Considerations for Future Laser Communications Systems," *AIAA Improving Space Operations Workshop*, Pasadena, CA, May 5-6, 2015.



LLCD*: A Successful Operational Lasercom Demo



*Lunar Laser Communication Demonstration

- In 2013, NASA's LLCD showed operational utility of lasercom
 - 622 Mbps downlink, 20 Mbps uplink
 - Automated point/acq/track of 15 μrad beam
 - Daily downlink of entire 1 GB s/c buffer
 - File transfers up / down
- Demonstrated reliable data delivery over lasercom links through Earth atmosphere:
 - Link worked through thin cirrus clouds
 - Multiple ground sites and higher layer data protocols to combat clouds
 - Data interleaving and multiple apertures to mitigate turbulence
 - Aircraft/sensor sat avoidance easily achieved



LADEE Satellite Narrow beam (<0.006°)

OPTICAL – NOW KNOWN!



craft position information



Is space craft position informatio good enough to point?

Is it possible to stabilize if space craft jitter is >> beam?



Can clouds, weather, and turbulence be managed?

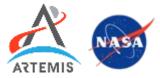
Can sensor satellites / aircraft be e avoided and not disruptive to lin

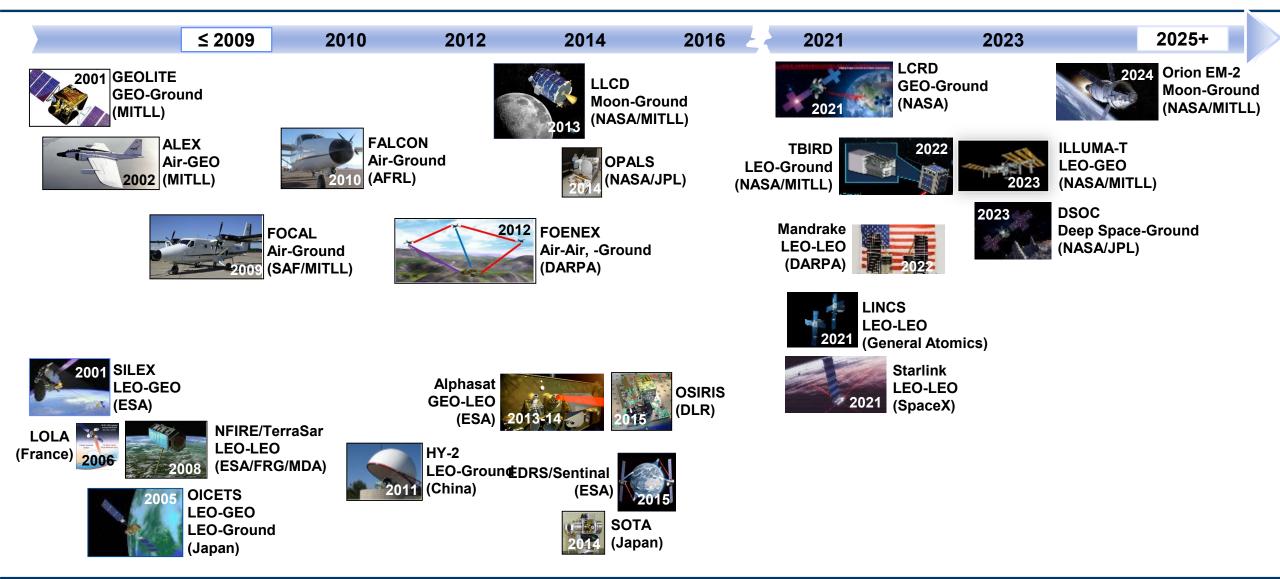
Can lasercom ever be routine and automated?

From: Khatri & Robinson, "Operational Considerations for Future Laser Communications Systems," AIAA Improving Space Operations Workshop, Pasadena, CA, May 5-6, 2015.



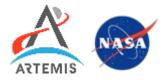
A Sampling of Space Lasercom Activities to Date

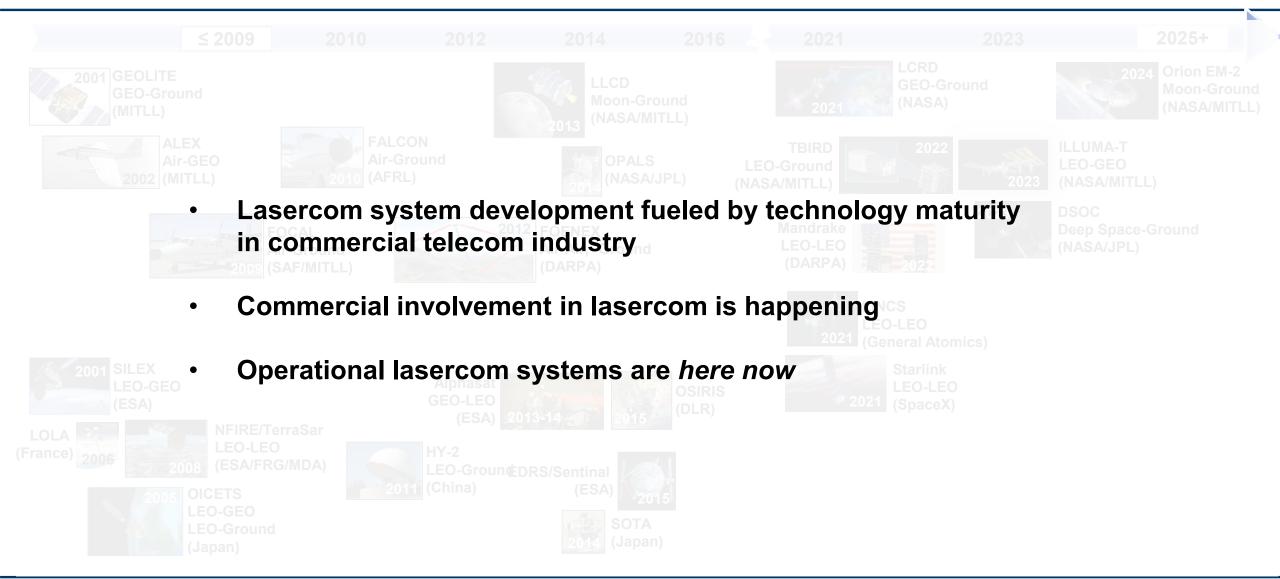






A Sampling of Space Lasercom Activities to Date







Orion

Operations

OpCom Mission Objectives

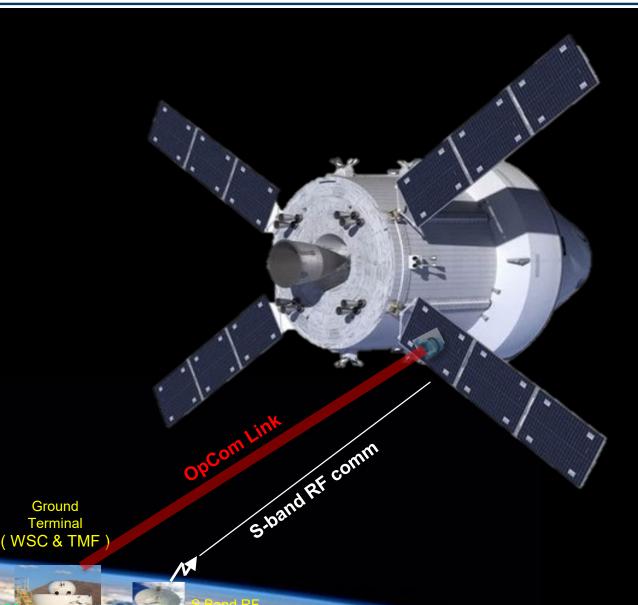
Optical Ground



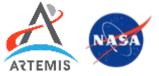
Development Test Objective (DTO):

Implement a laser communications capability for the Orion series of spacecraft, starting with Artemis II, in order to enhance its operational utility by

- 1. Transferring files from MCC to/from Orion
- 2. Transferring real-time video downlinks from Orion to MCC
- 3. Utilizing operational performance envelope by demonstrating optical comm during various conditions





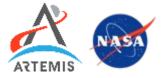


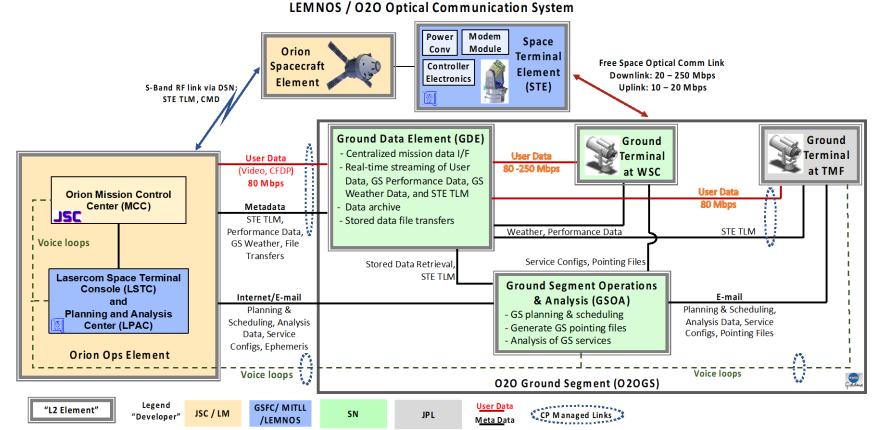
Orion Recorder Utilization Orion subsystems expected to generate ~250 GB of data by FD1 Lunar Obs / Perilune alone and ~300GB by end of Recorded Imagery 300 mission Storage with constant Using S-Band alone, Orion limited 1 MBps S-Band Downlink 250 to ~ 7GB of data downlink per day. Can not downlink all recorded data Utilization (G Bytes) 200 With just 1 hour/day of Optical Comm, Orion could downlink ~ **36GB of data per day, a 6x increase** 150 per day! Storage At the 260Mbps link capacity, Orion Storage with OpCom 100 could downlink 117GB per day 80 MBps Downlink almost 20x increase Storage Storage with OpCom with 2 Two 1 Hr 260Mbps contacts per 50 260 MBps Downlink OpCom day, Orion could downlink 234GB 260 MBps per day, and all of the recorded Downlinks data on the second day 2 5 6 7 8 1 3 4 q 10

MET (Day)

SPIE 2023 12413-9 - 9 F. Khatri 1/30/23







6 organizations

- Johnson Space Center (JSC)
- Lockheed Martin (LM)
- Goddard Space Flight Center (GSFC)
- Space Network (SN)
- MIT Lincoln Lab (MITLL)
- Jet Propulsion Lab (JPL)

5 major system elements:

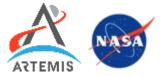
- Orion Spacecraft
- Space Terminal
- Ground Terminal
- Ground Data Element
- Orion Operations

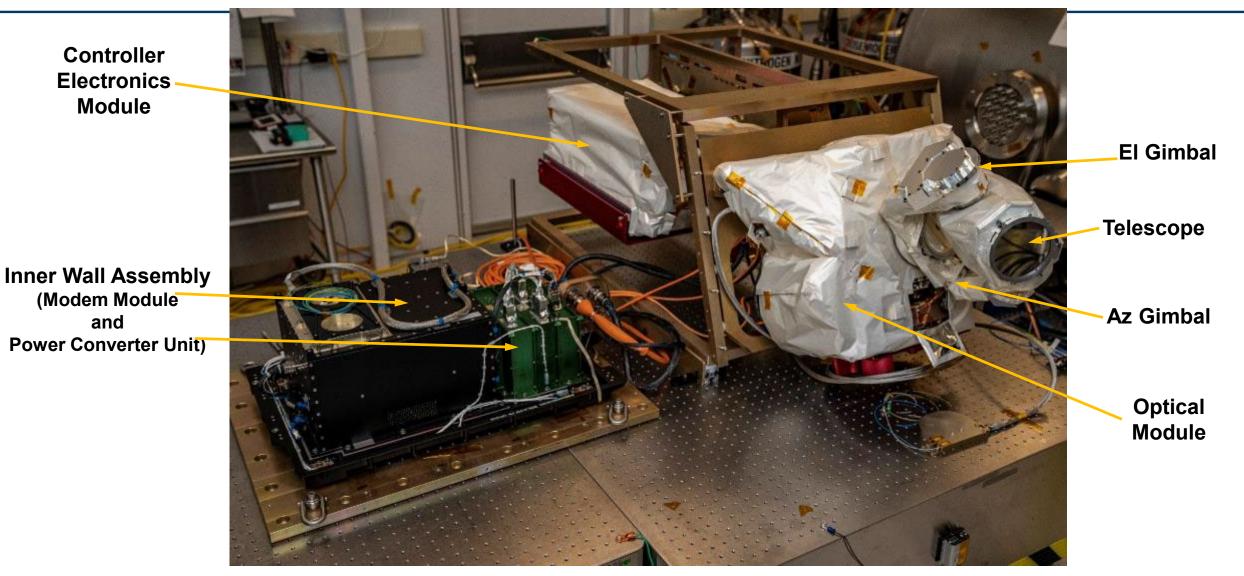
2 possible ground terminals

- White Sands
- JPL's Table Mountain Facility (TMF)

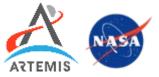


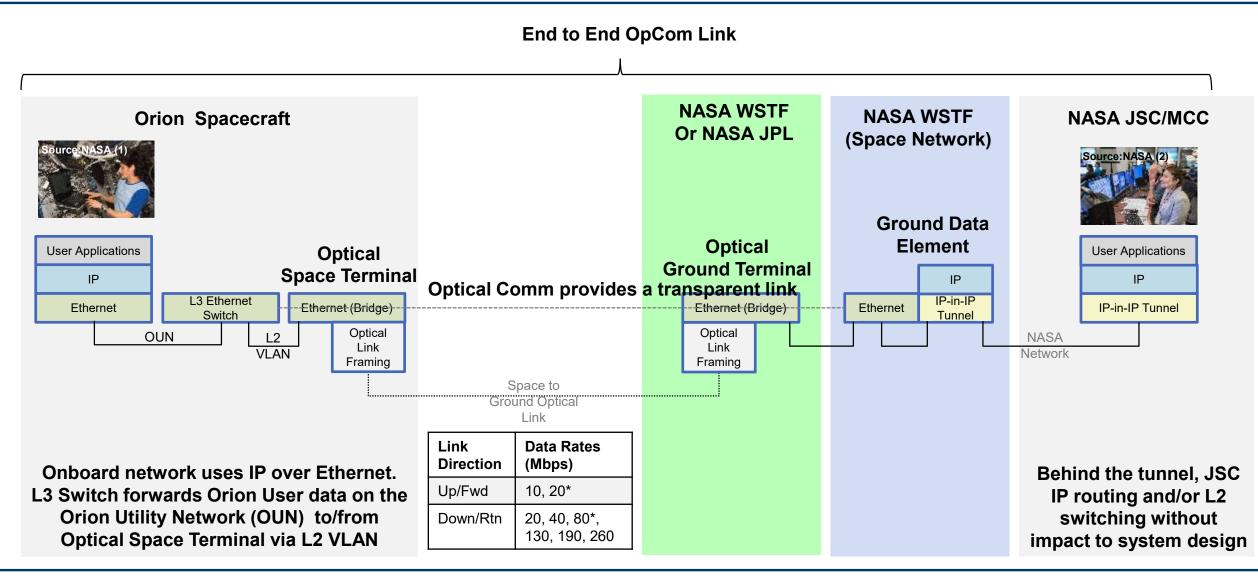
O2O Space Terminal Payload







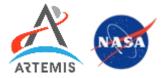


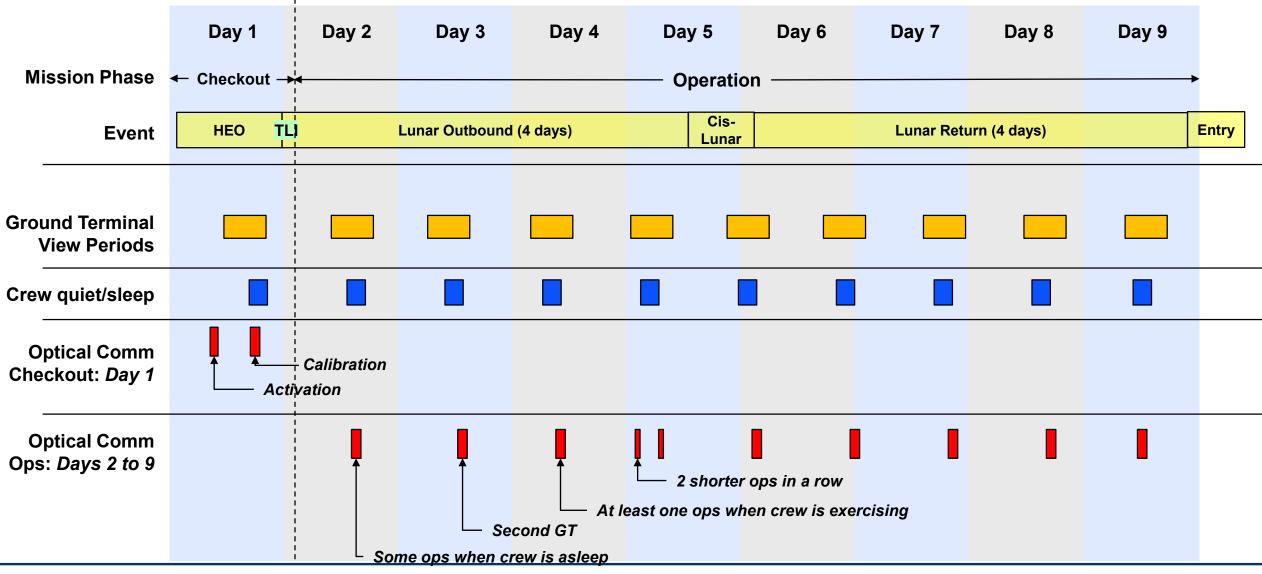


<u>https://www.nasa.gov/sites/default/files/thumbnails/image/iss015e09442crop.jpg</u>
 https://www.nasa.gov/sites/default/files/thumbnails/image/jsc2017e049159.jpg

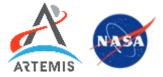


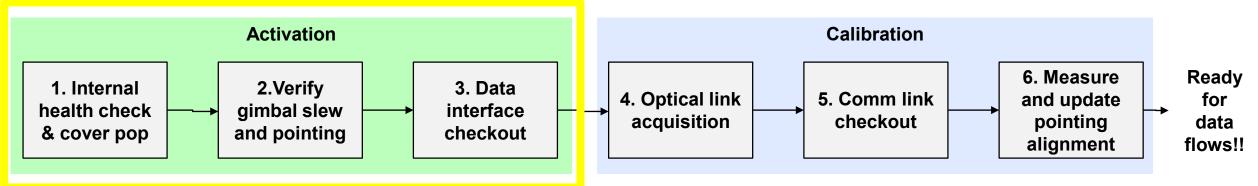
Notional Artemis II OpCom Mission Timeline











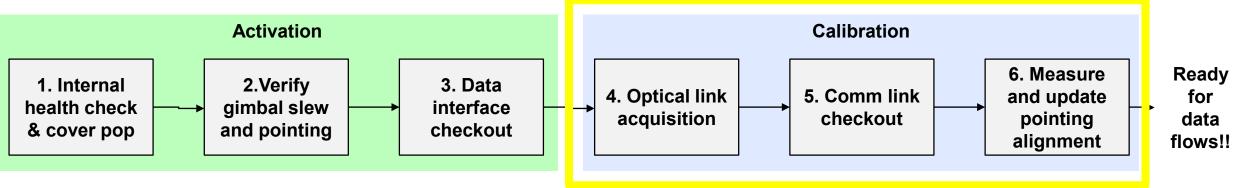
- 1. Internal health check & cover pop
 - Power up payload controller
 - Send "no op" to validate commanding
 - Verify telemetry packet timing
 - Verify operational heaters
 - Verify on-board star tracker
 - Power up modem
 - Perform "closed cover" self test to assess health of optical module
 - Activate HOPAs to un-latch (pop) cover
 - Repeat "closed cover" self test

2. Verify gimbal slew and pointing

- Explore gimbal operational envelope
- Point gimbal using live spacecraft position and attitude
- Stow and hold gimbal to point at latch
- 3. Data interface checkout
 - Put modem in data loopback mode and count incoming and outgoing Ethernet frames







4. Optical link acquisition

- Grossly point gimbal to ground terminal using live spacecraft position and attitude (ground terminal points to space terminal and transmits wide divergence beacon)
- Automatic scan of acquisition sensor until beacon is detected → system pulls in and tracks on beacon
- Transmit downlink comm signal (ground terminal receives and tracks → improves its pointing and narrows beam)
- Additional beacon signal detected, system can high bandwidth track

5. Comm link checkout

 Verify comm link for various comm link pairs by monitoring codework error rates (at the optical link layer)

6. Measure and update pointing alignment

- Identify difference between "open loop" and "fine tracked" pointing
- Compute and update alignment matrix

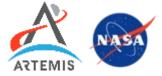




- Provision optical link and run the following applications:
 - File transfer
 - PDF files: crew files, news, email, text
 - Recorded data
 - Procedures
 - Op Nav files
 - Recorded Video files
 - Recorded JPEG
 - Real-time video
 - Solar array wing cameras
 - Op nav video
 - Crew Module high rate cameras
 - HD cameras
 - Crew cabin cameras, GoPro
 - Video conference

- Utilize operational performance envelope:
 - High and low elevation, day/night
 - Acq/track optical comm signal during
 - Solar array track/step/slew
 - Crew activity/exercise
 - Thruster fires
 - Different ground terminals, handover
 - Thermal system effects
 - Atmospheric effects

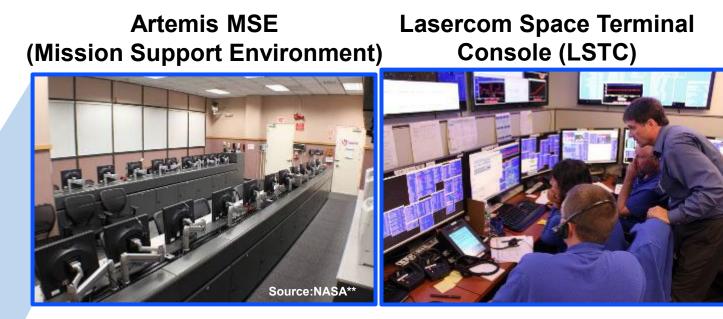






Artemis MCC (Mission Control Center)

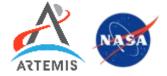
*https://www.nasa.gov/feature/artemis-i-mission-control-at-a-glance



- Payload will be operated by the Instrumentation and Communications Officer (INCO) in the Artemis MCC
- MSE / LSTC offer backup support and trouble shooting

**https://www.nasa.gov/centers/dryden/capabilities/CodeM/CodeMR/multimedia/imagegallery/MCC2_5.html





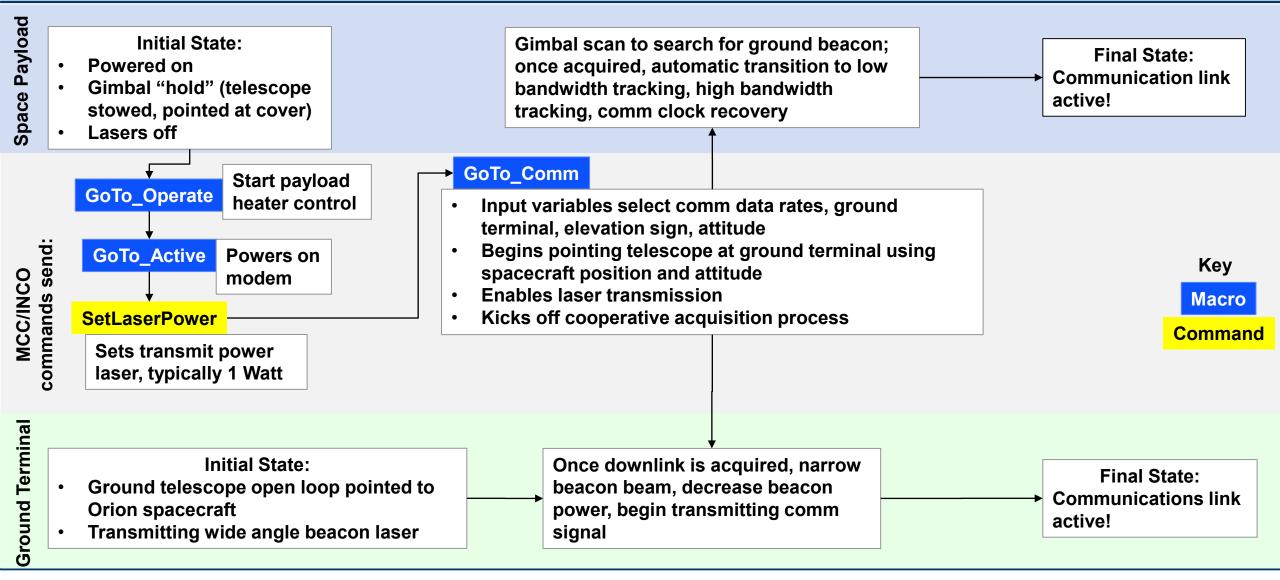
3 methods to command OpCom payload are available for MCC/INCO:

- 1. Native commands:
 - Low level commands understood by OpCom controller electronics
 - A subset of these native commands have been given Compact Unique Identifiers (CUI) so that they can be run directly by MCC/INCO
- 2. Macro commands:
 - "Programs" or scripts written in STOL¹ that employ native commands and simple logic; macros can be nested as well
 - Each macro has a unique identifier number
 - A subset of these macros have been given Compact Unique Identifiers (CUI) so that they can be run directly by MCC/INCO
 - Macros can also be run using a native command that runs macro its by identifier number
- 3. Block commands:
 - The most generic transfer of information from MCC/INCO to OpCom controller electronics
 - Used for file transfer (CFDP) or for running any non-CUI native commands



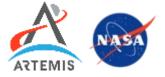


With animations









- Artemis II mission will utilize OpCom to provide operational laser communications service
- OpCom will enable video up/downloads, file transfers, and real-time video conferencing, for the 10-day mission
- OpCom operated by MCC/INCO using simple pre-defined commands and macros that are highly automated