



Development and Validation of an Initial Electric Motor Rotor Vibration Model

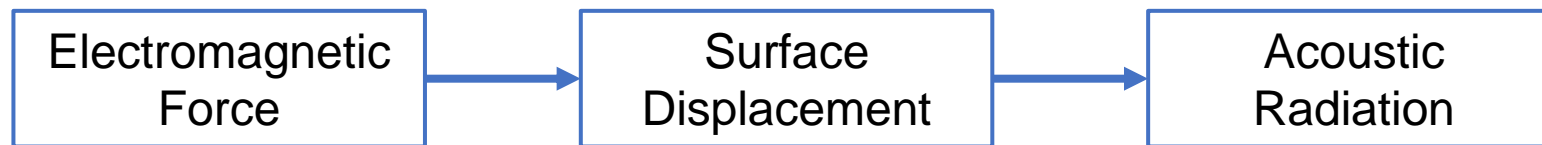
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*Spring 2023 NASA Acoustics Technical Working Group
NASA Langley Research Center*

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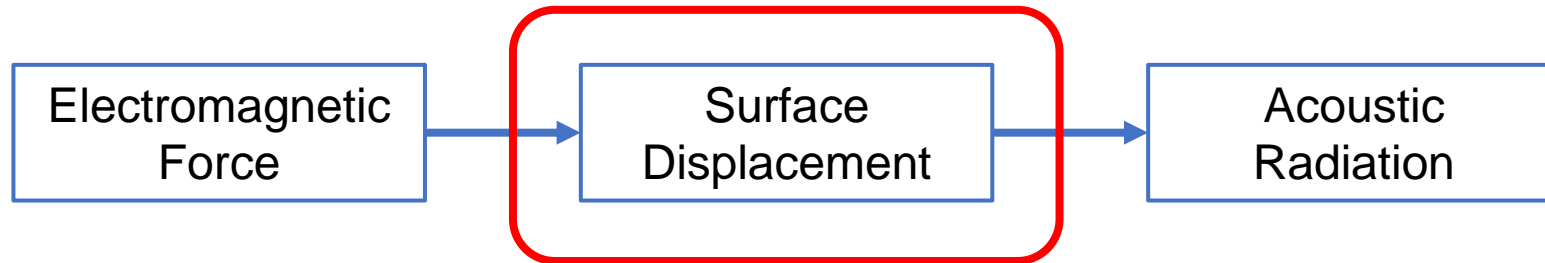


Motor Noise



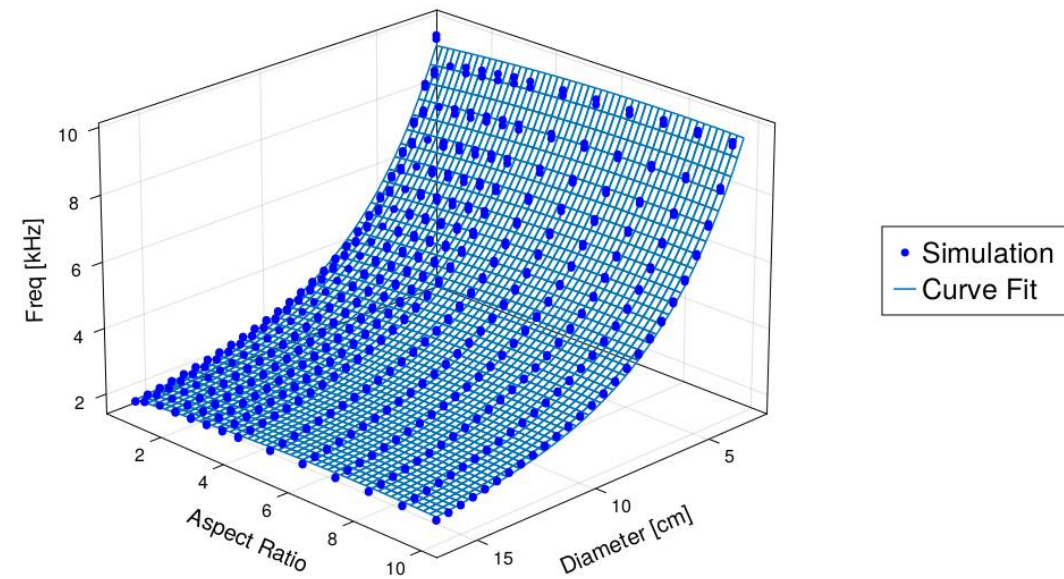
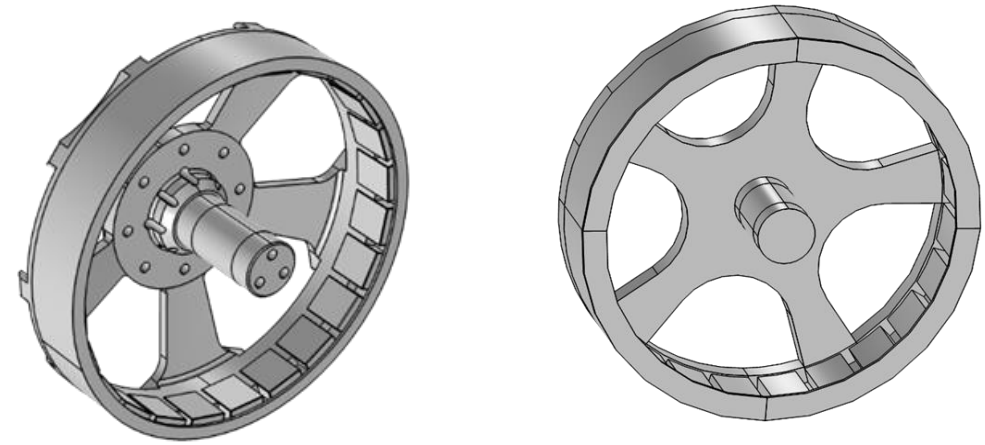


Motor Noise



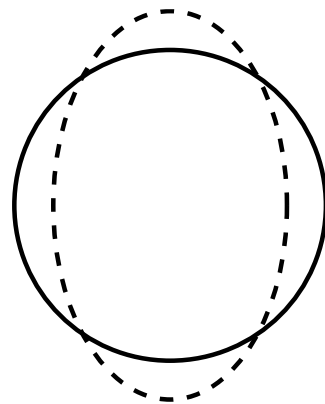
Objectives

- Electric motors on UAM aircraft can generate tones
 - Tones caused by resonance of motor rotor
 - Acoustically relevant even with propellor noise
- Need a low fidelity noise prediction tool for system level studies
 - Aim to predict the frequency and mode shapes of resonances
- Three stages of simulation and modeling
 1. High-fidelity geometry simulations
 2. Parameterized geometry simulations
 3. Parameter sweep and curve fit

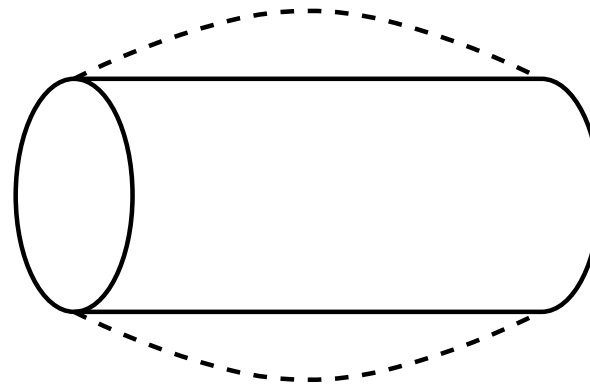


This supported Milestone RVL23.21.L4104

Mode Shapes

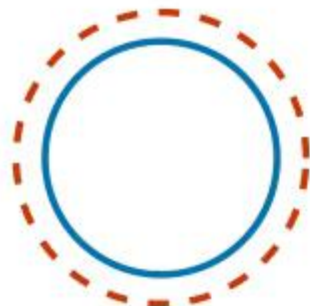


Azimuthal Mode
Number M

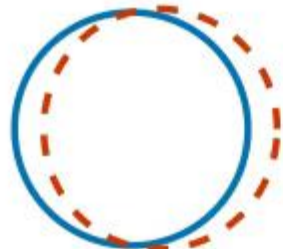


Axial Mode
Number N

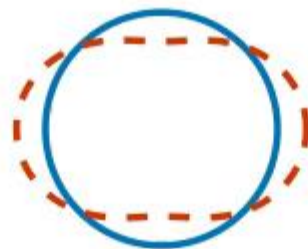
$M = 0$



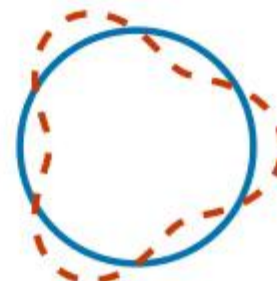
$M = 1$



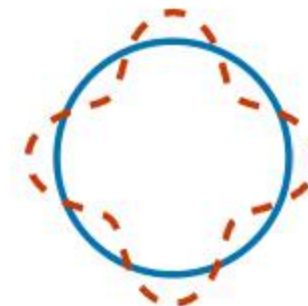
$M = 2$



$M = 3$

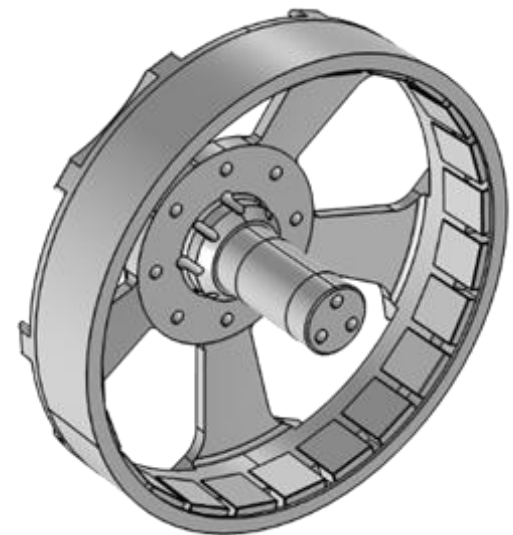


$M = 4$



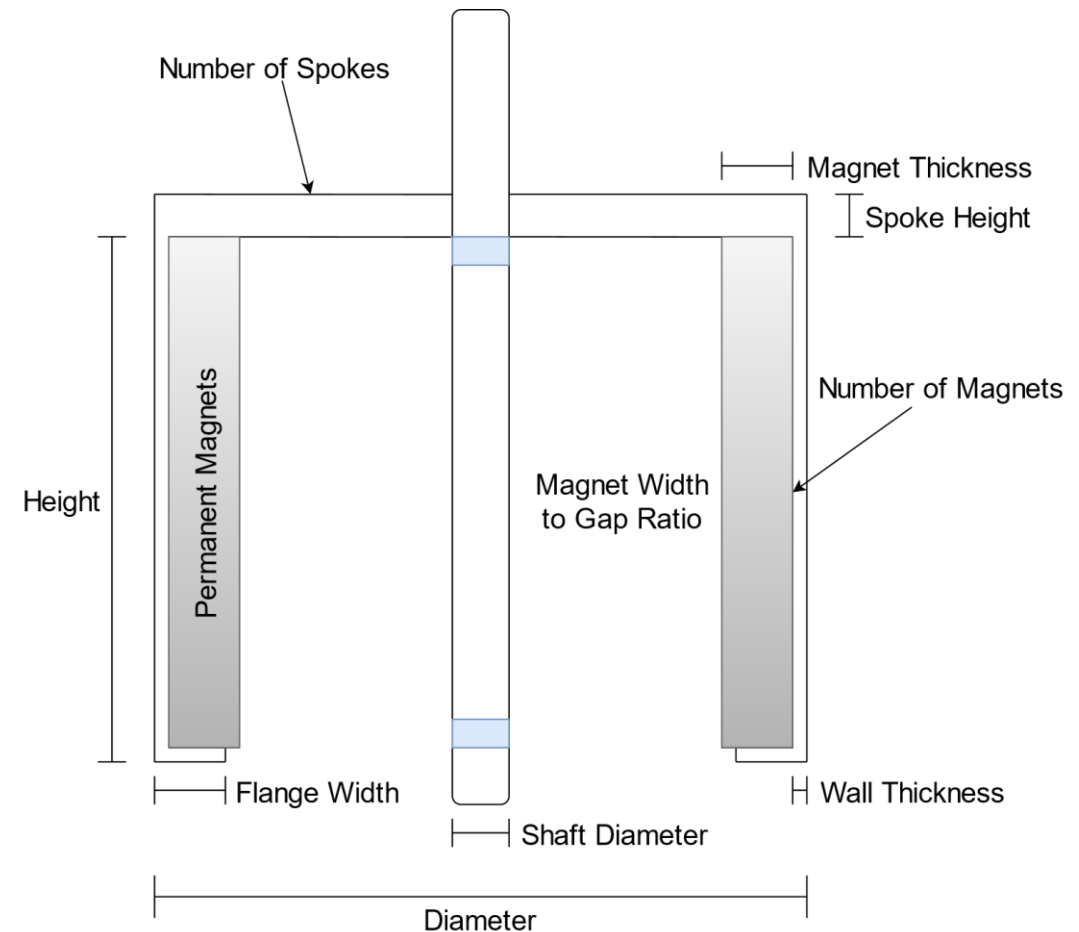
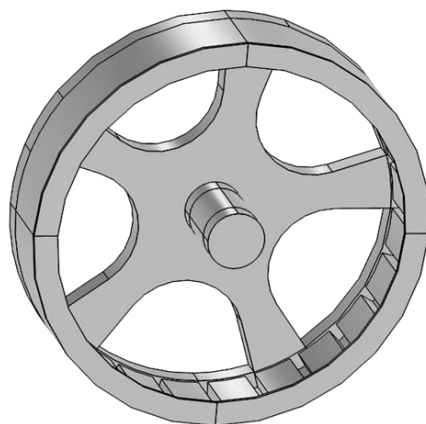
High-Fidelity Geometry Simulations

- Accurate geometry created of the rotor for two motors
 - T-motor U13
 - Scorpion SII 4020
- FEA eigenfrequency/mode simulations performed in COMSOL
- Experimental comparisons good
 - Acoustic tones
 - Experimental modal analysis



Parameterized Rotor Geometry

- Parameterized model defined by 10 key quantities
 - Most are non-dimensionalized by Diameter
- Parameters anticipated to be most critical
 - Diameter
 - Height (or Aspect Ratio)
 - Spoke Height
 - Wall Thickness
- Model tuned to experiments and high-fidelity models



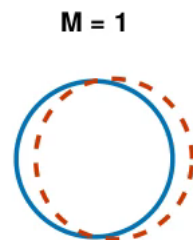
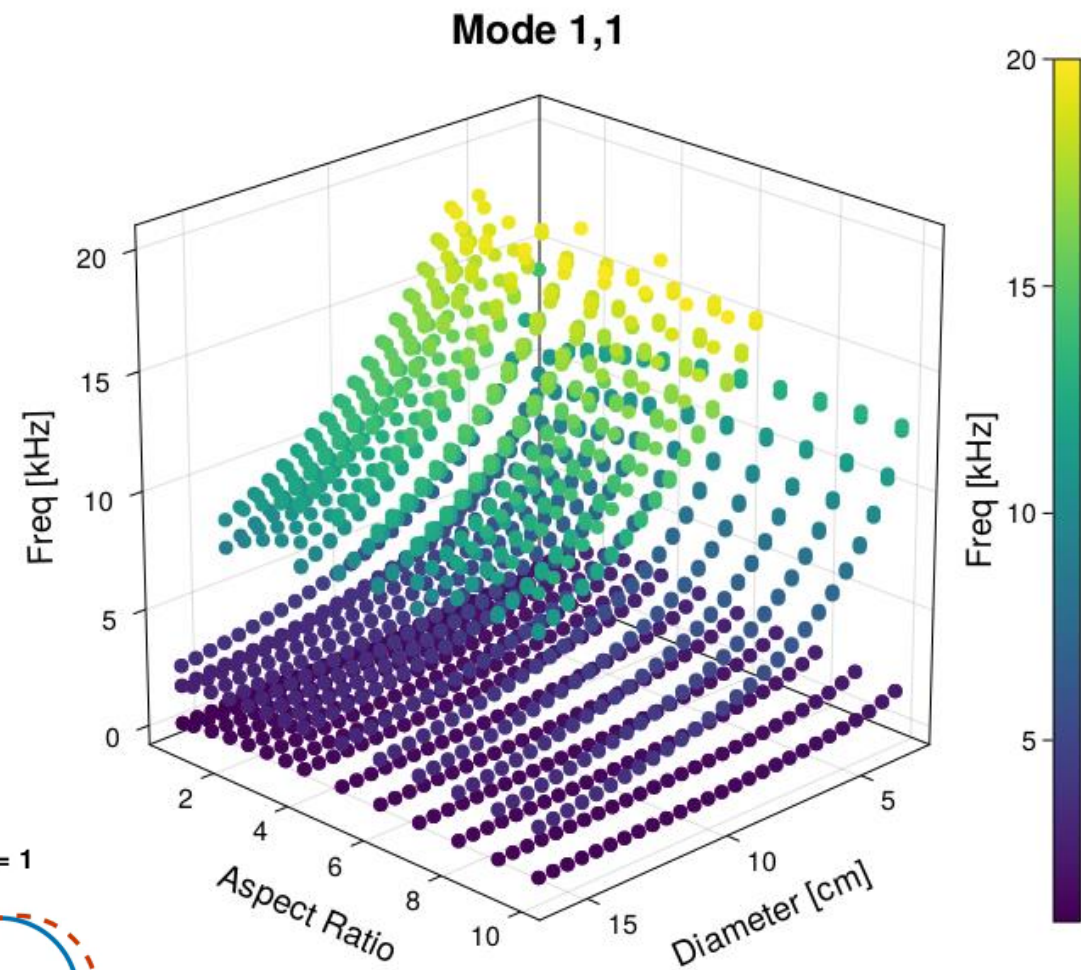


Parametric Simulation Sweep

- 35 eigenfrequencies/eigenmodes calculated for each simulation
- Four most important dimensions iterated over
 - Diameter: 30mm – 160mm
 - Aspect Ratio (Diameter/Height): 0.7 – 10
 - Spoke Height Ratio: 0.1 – 0.3
 - Wall Thickness Ratio: 0.02 – 0.04
- Other dimensions were given fixed values corresponding to off the shelf motors
- ~3400 total combinations
- Moderate computer requirements
 - Intel Xeon E5-2637 v4
 - 8 cores (7 threads used)
 - Used less than 64GB RAM
 - Less than 24 hours compute time

Simulation Sweep Results

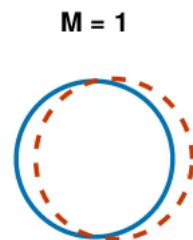
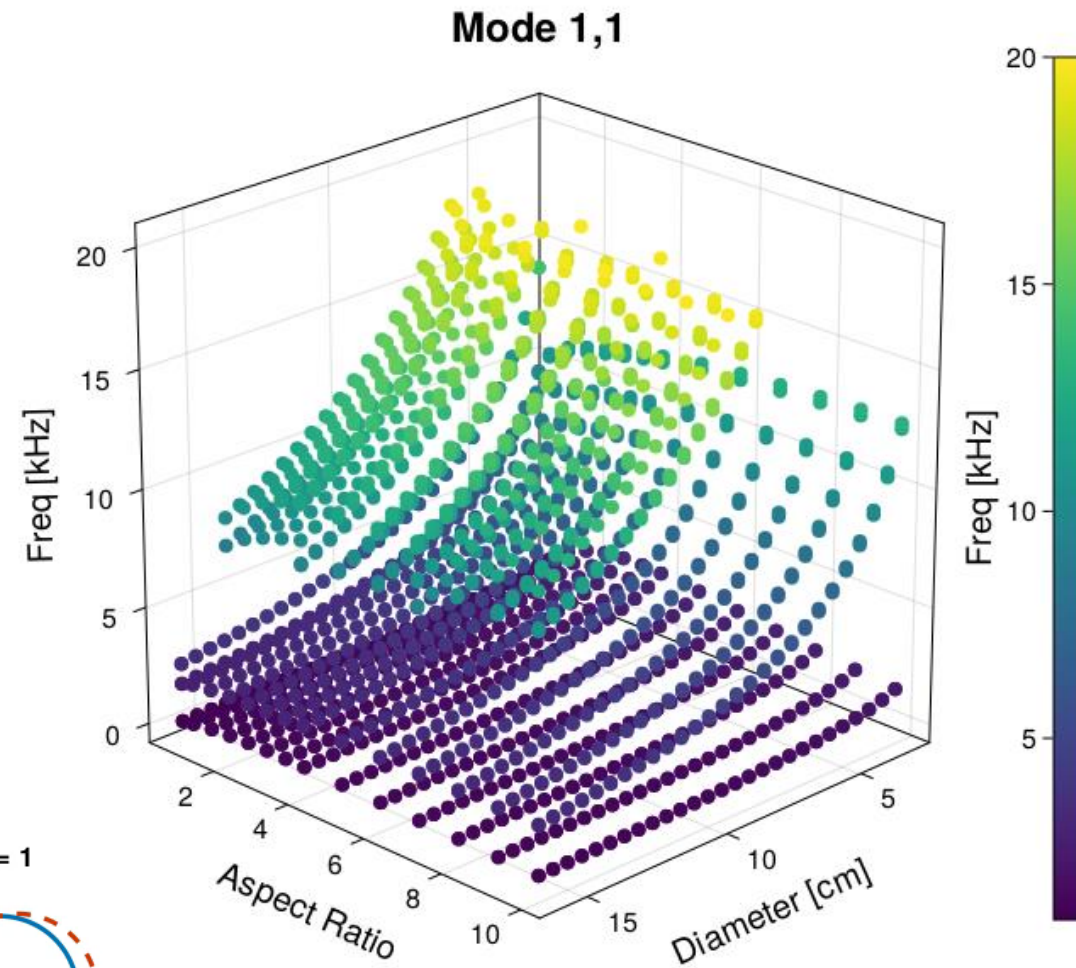
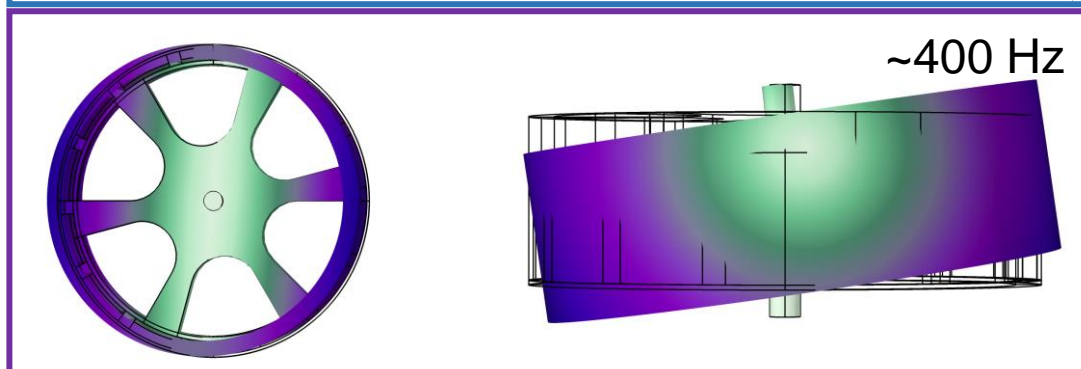
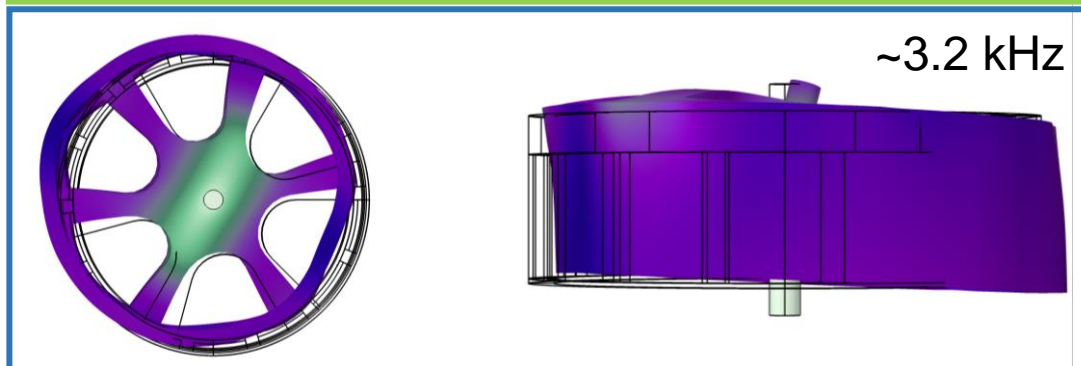
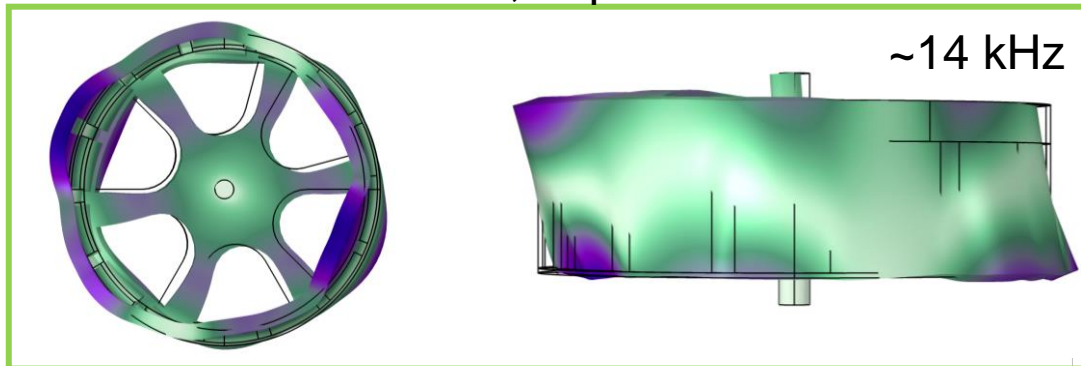
- 3400+ simulations
 - 35 results per simulation
 - Each result has M,N and frequency
- Dimensions not shown
 - Spoke Height
 - Wall Thickness
 - Other modes
- Three distinct groups are visible
 - Each group same mode shape
 - Differences in details of vibration



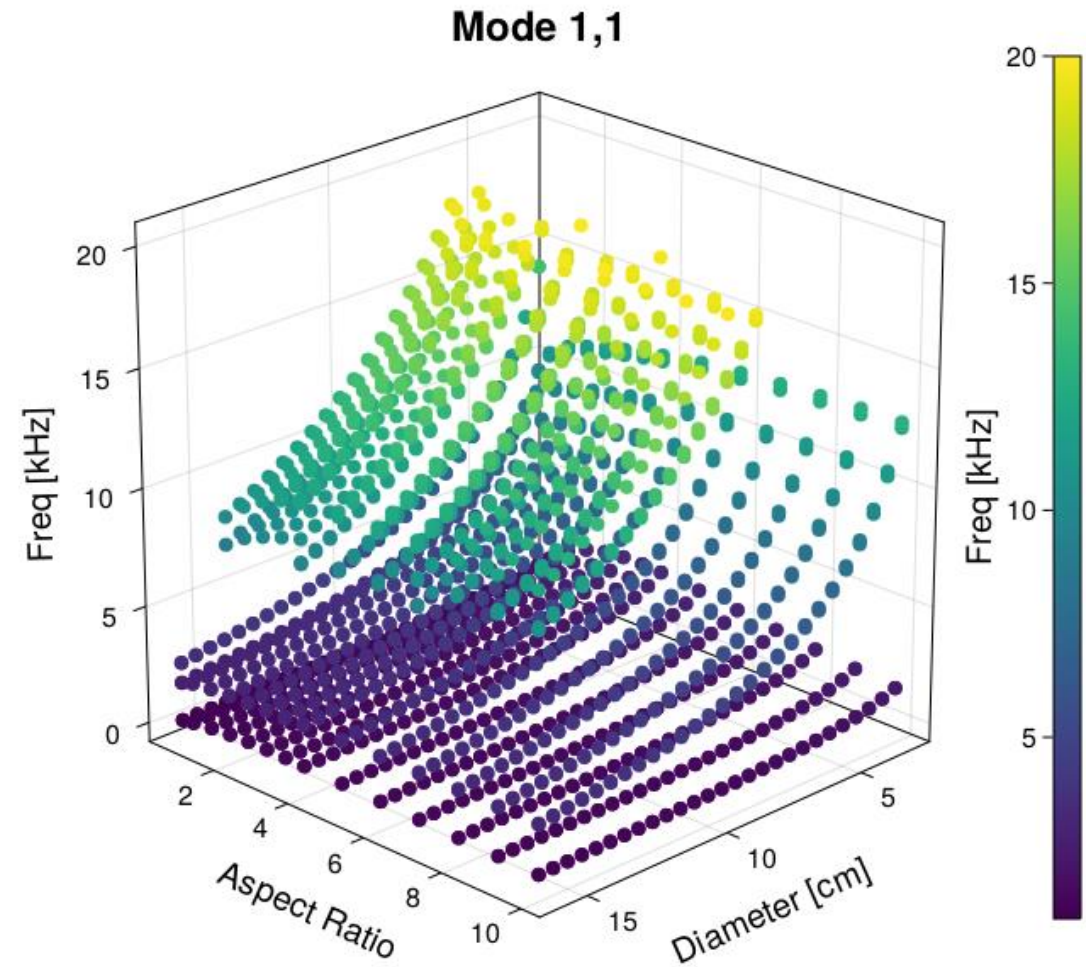
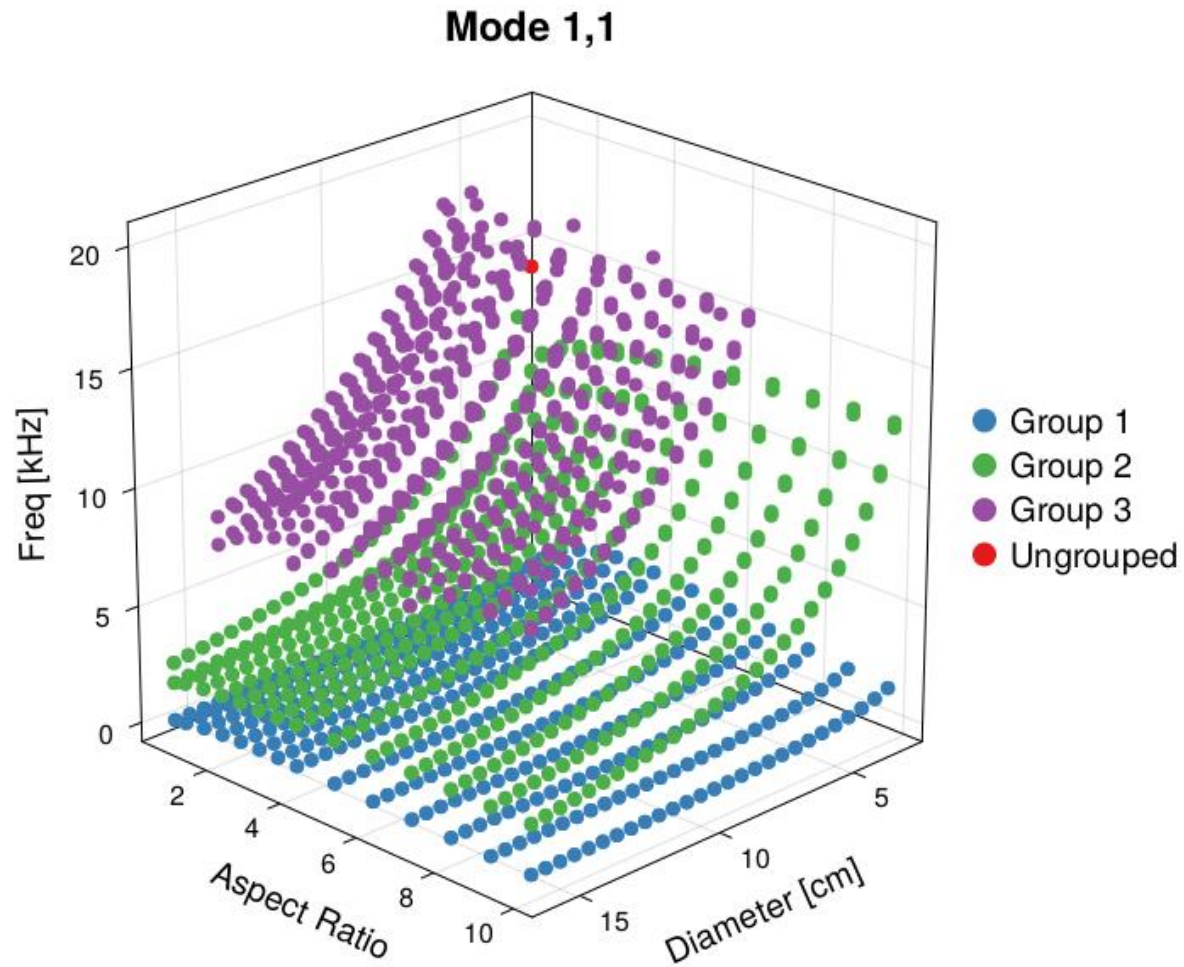
3D slice of 6D result space

Simulation Sweep Results

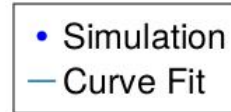
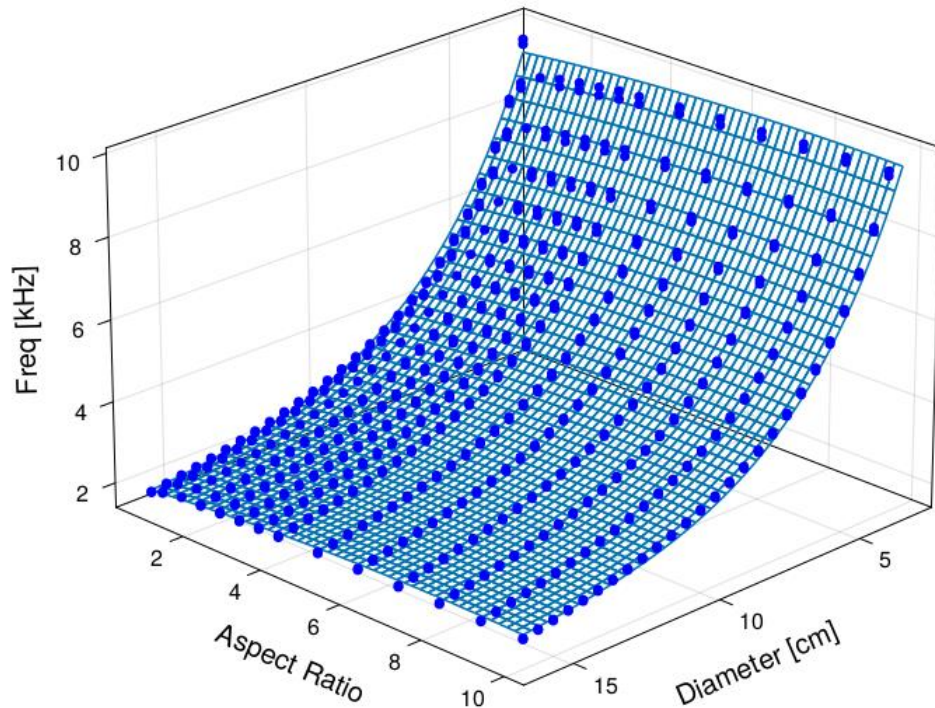
Diameter 10cm, Aspect Ratio 4



Simulation Sweep Results



Validation



Mode Shape	Predicted Freq [Hz]	Experimentally Measured Freq [Hz]
1,1	935	
2,1	3515	2971
1,1	6411	5942
3,1	6676	6919
0,1	7023	
0,1	8382	
4,1	8841	

- Each grouping can have a curve-fit applied
- Curves generate a list of mode shapes and frequencies at which resonance is predicted
- These predictions match what has been experimentally measured



Conclusions

- A small parameter simulation can capture the relevant physics of the resonant modes of outrunner brushless motors in the 1-5kW range.
- This model can be used to create a simple set of fitted functions that do not require even the FEA simulation to be performed.

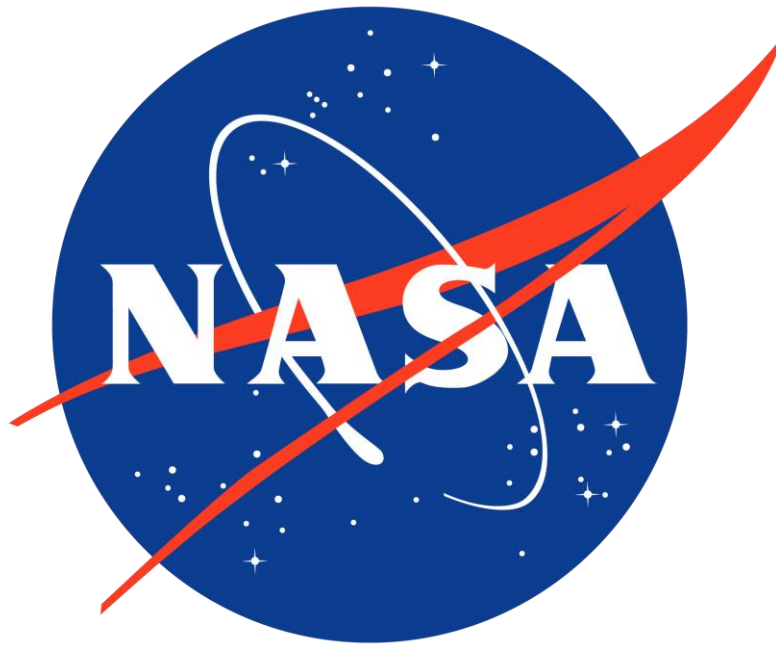


Future Work

- Find method to quantify important and unimportant parameters and modify the model accordingly
 - Run simulations to see impact of each parameter
 - Approximate some parameters as functions of others
- Improve mode shape categorization and omit modes that won't radiate acoustically
- Improve curve-fit equations
- Add amplitude predictions
- Continue experimental validation

- Integrate with the larger prediction tool (electro-magnetic forcing predictions and acoustic propagation calculations)

Questions?



Thank you!