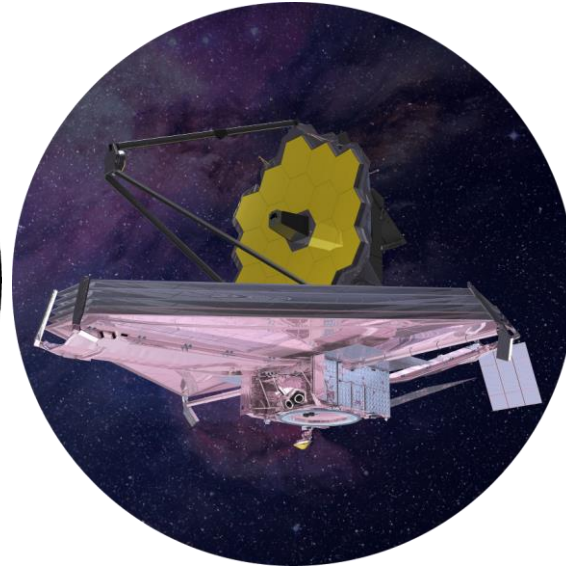
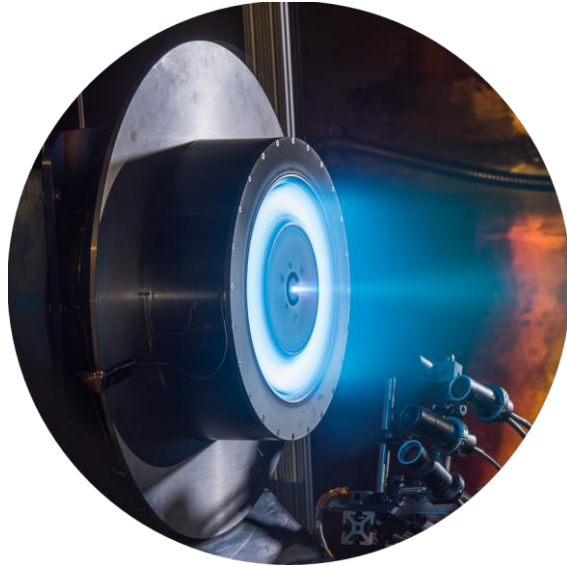


# Two Decades of Aerospace Vehicle Analysis with HyperSizer and HyperX

National Aeronautics and  
Space Administration



Lloyd B. Eldred

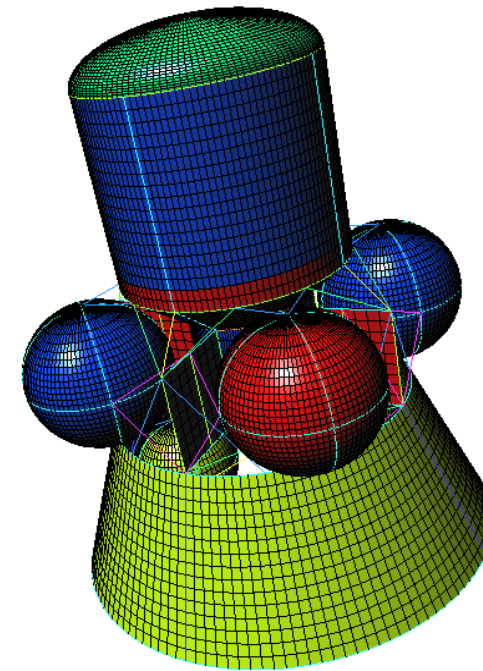
Vehicle Analysis Branch

NASA, Langley Research Center

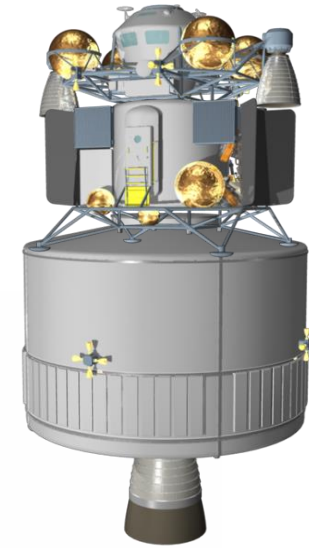
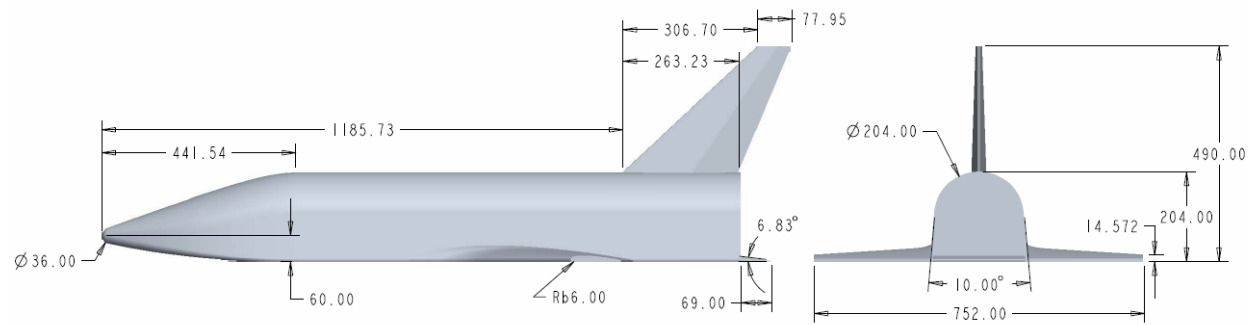
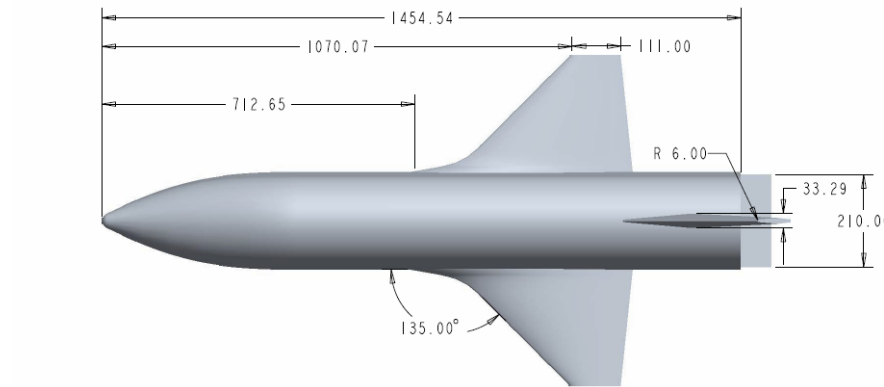
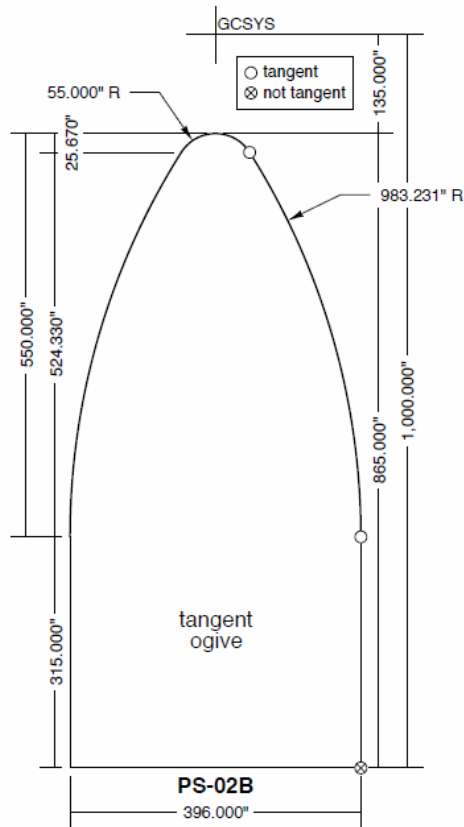
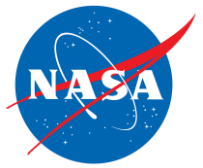
# Outline



- Multidisciplinary preliminary analysis process
- Twenty+ years of design
- Automation & Related Tools
- Conclusion



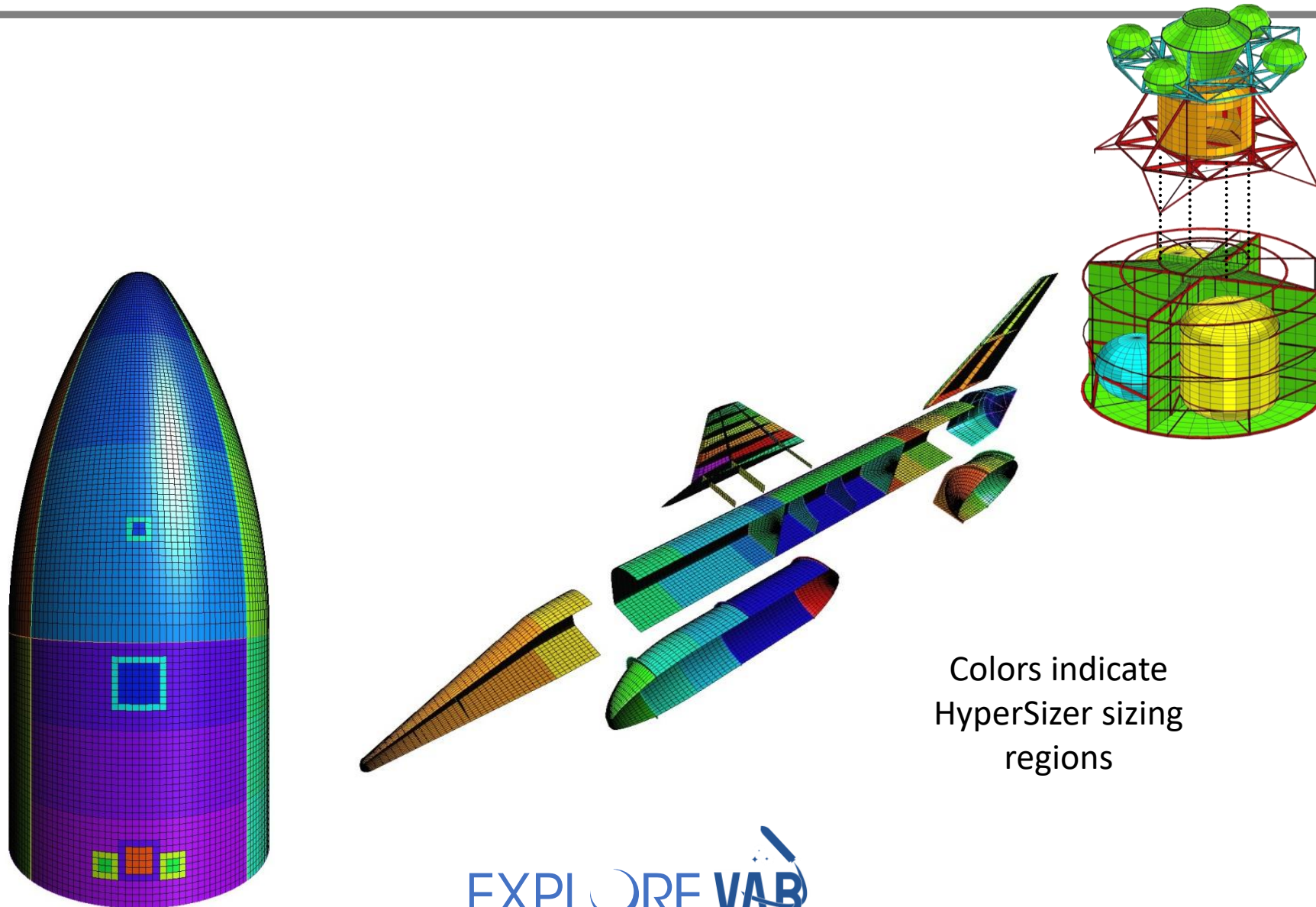
# Preliminary Design: Sketch



TSTO-2009-2B\_OML

SEF\_09/24/2009

# Preliminary Design: FEA Model

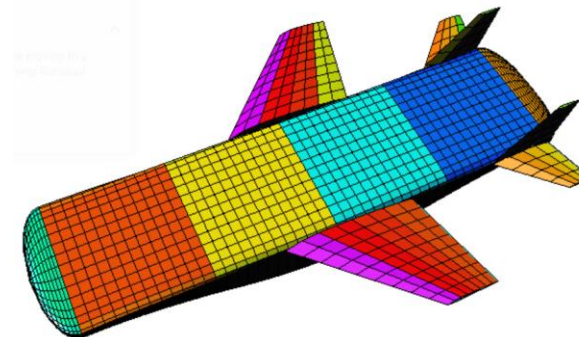


Colors indicate  
HyperSizer sizing  
regions

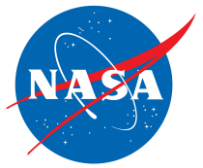
# Preliminary Analysis



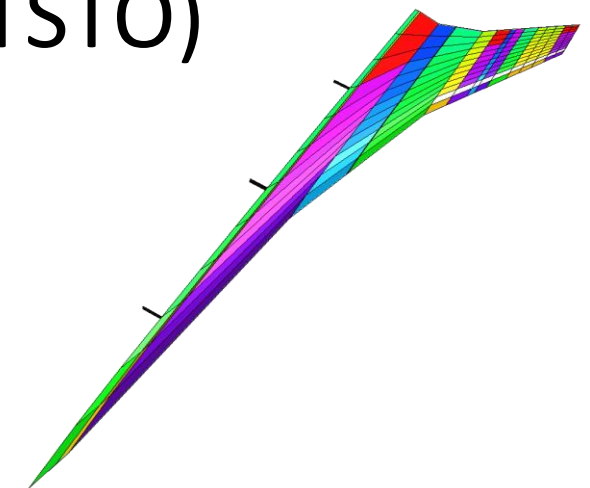
- Generate finite element model
- Identify critical load cases
  - Launch vehicles: MaxG, MaxQ $\alpha$ , etc.
  - Flight vehicles: 2g runway bump, Max pullup, etc.
- Create or approximate aero, thermal, propulsion, hydrostatic, etc. loads
- Solve in NASTRAN
- Size in HyperSizer/HyperX
- Iterate to converged mass
- Perform design trade studies



# Twenty+ Years of Design



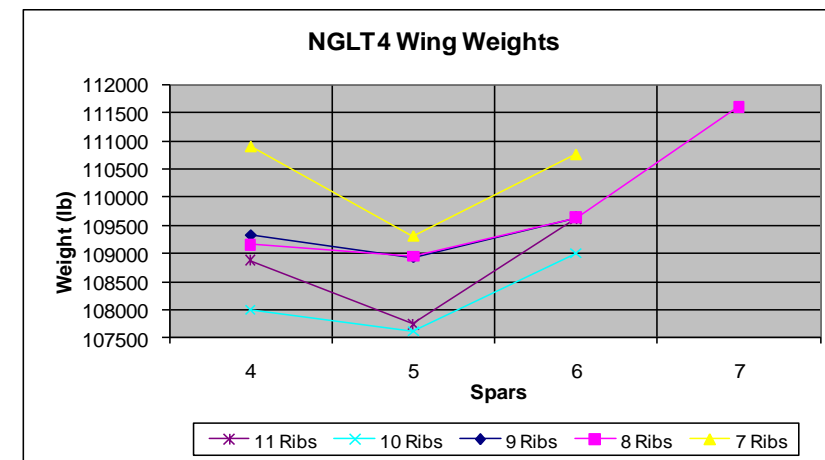
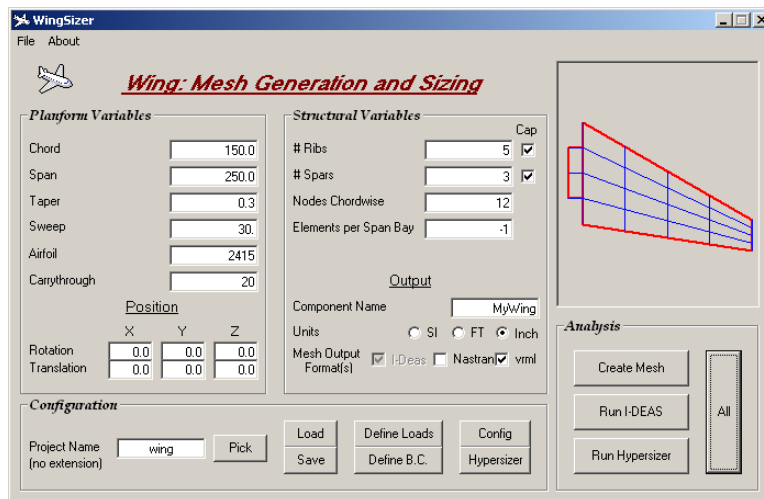
- Wingbox calibration
- Next Generation Launch Technology (NGLT) wing optimization
- Unusual tank concepts
- Lunar lander concepts
- Single and Two Stage to Orbit (SSTO and TSTO) hypersonic concepts
- Launch vehicle fairings
- Low boom supersonic aircraft



# Wingbox & NGLT Wing



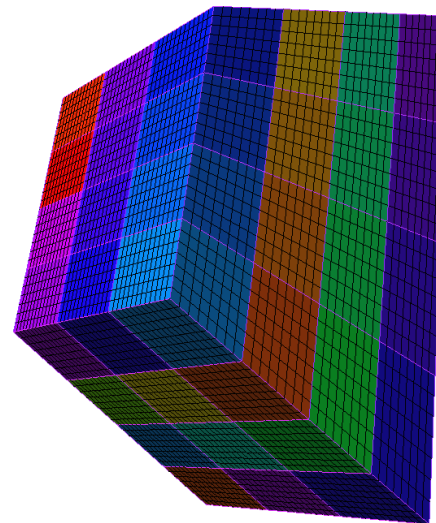
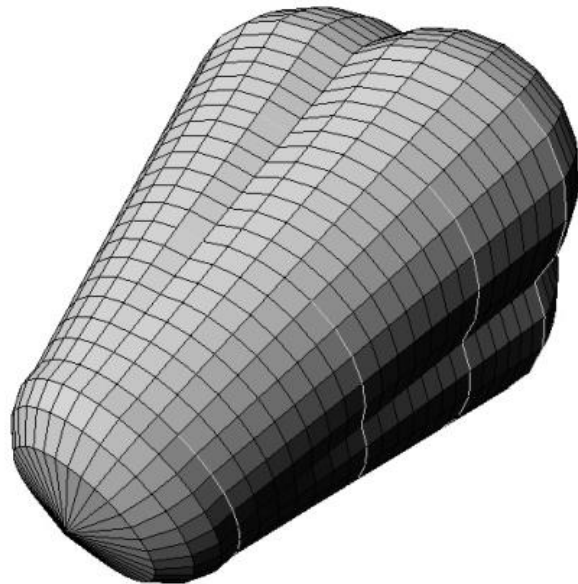
- Initial HyperSizer use (1999) predicted weight of NASA wingbox test article
  - As built weight was 1.5 times predicted weight
- Early 2000s automated study of Next Generation Launch Technology (NGLT) wing



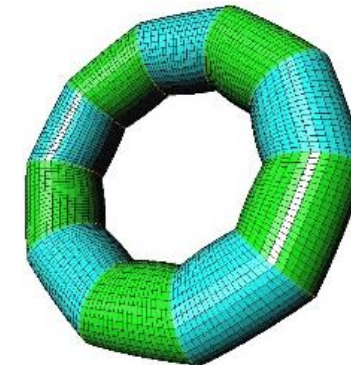
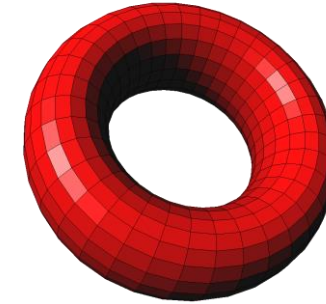
# Unusual Vehicle Components



**Multi-lobe tanks**



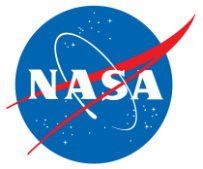
**Stiffened Mattress tanks**



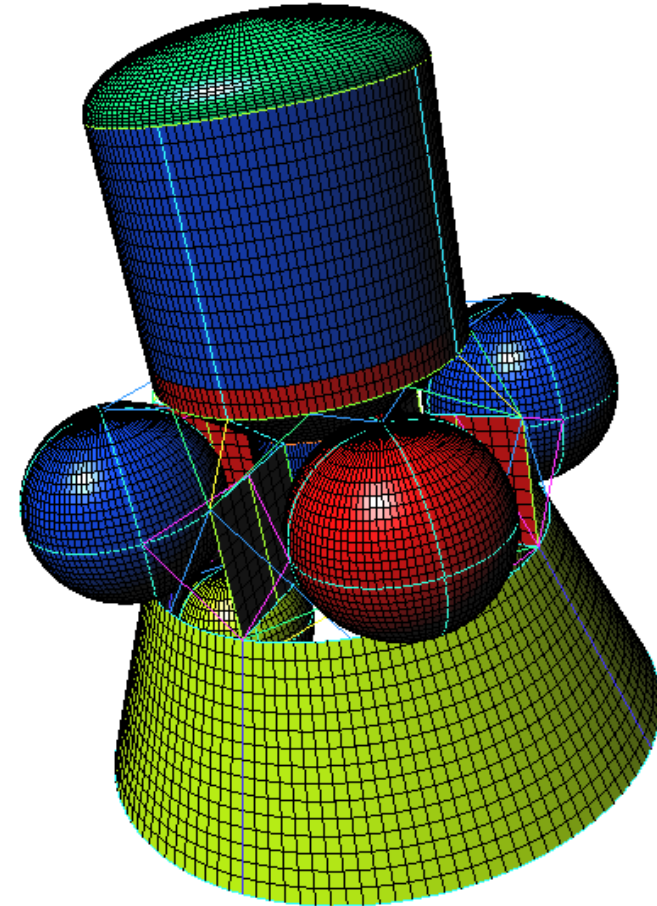
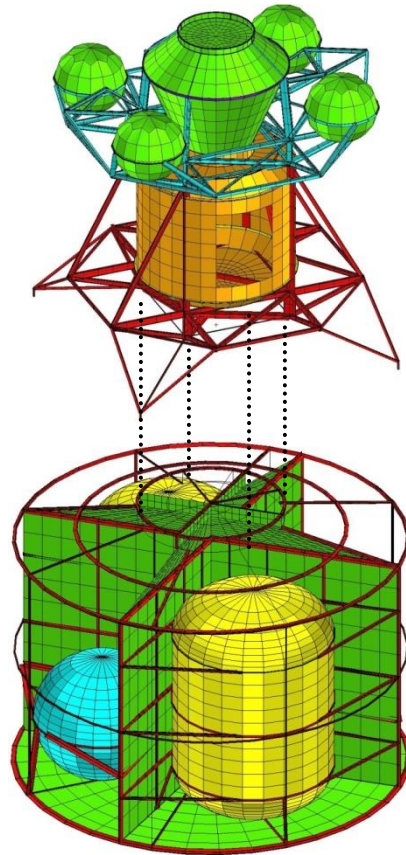
**Toroidal and  
segmented  
toroidal tanks**



# Lunar Landers

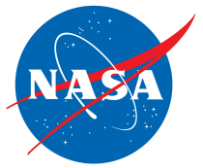


**Dash Lander Concept**

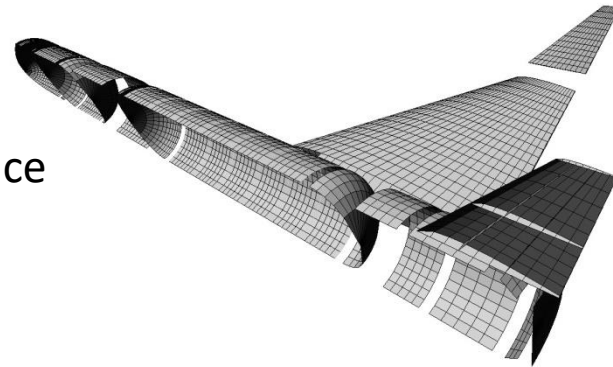


**NASA HLS Reference**

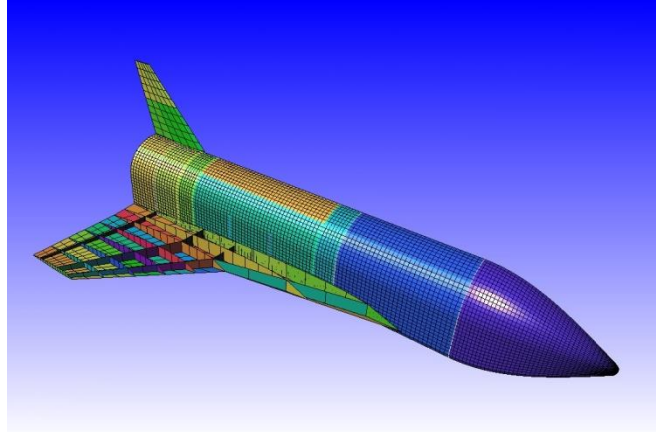
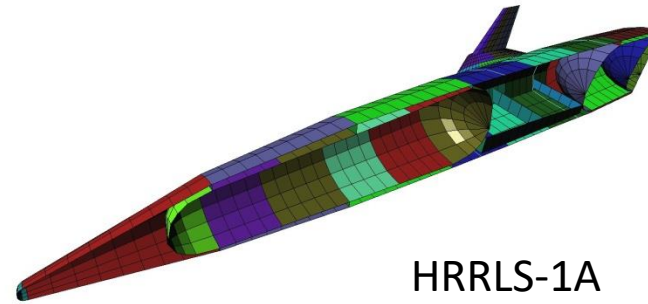
# Hypersonic Orbiters



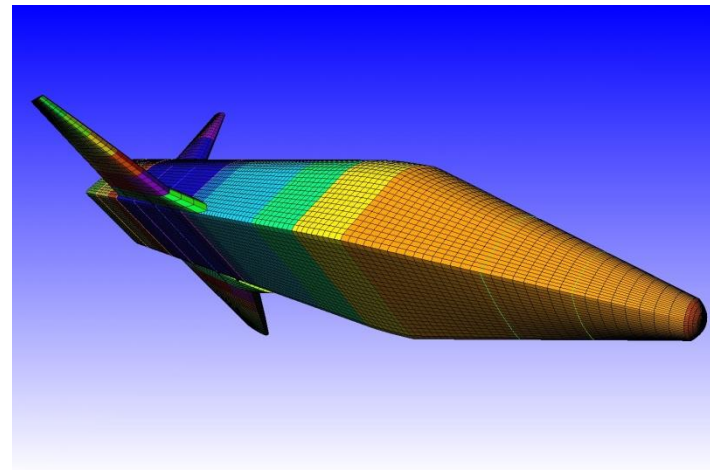
ISAT reference  
design



HRRLS-1A



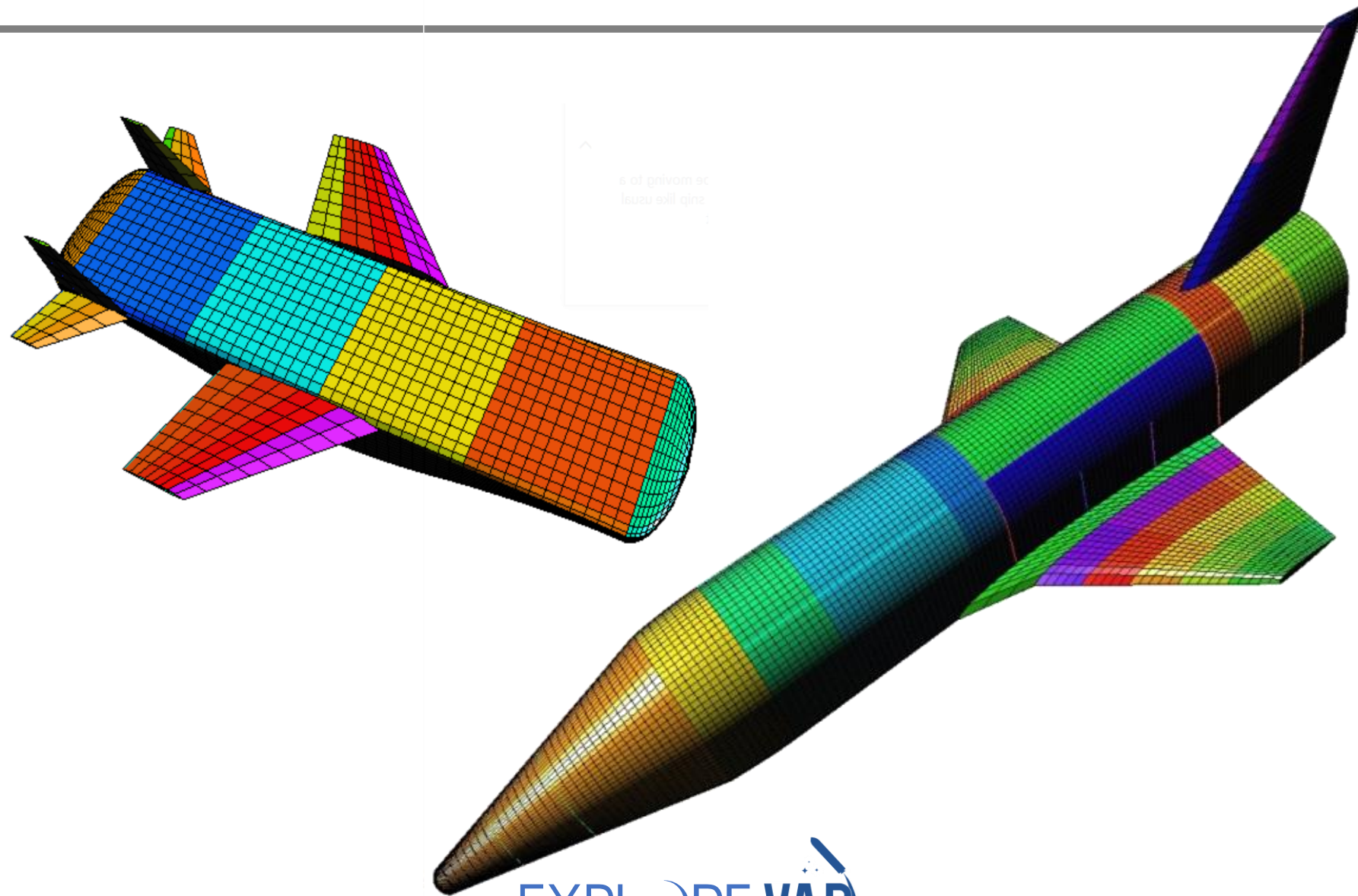
RALV-A



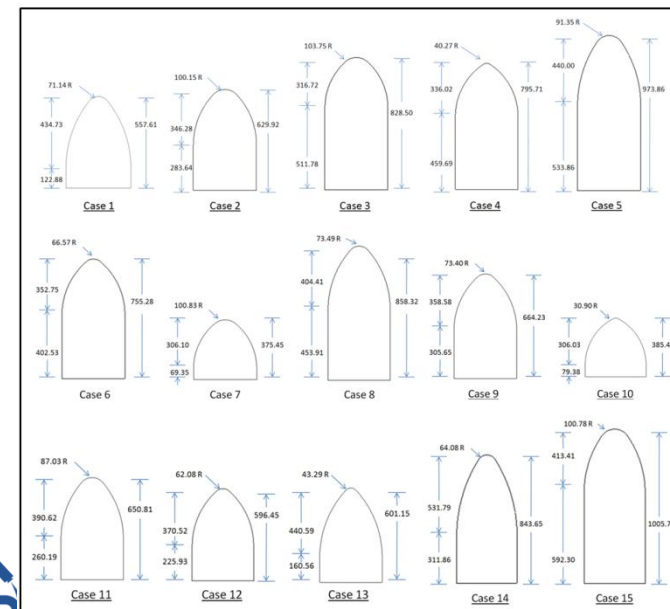
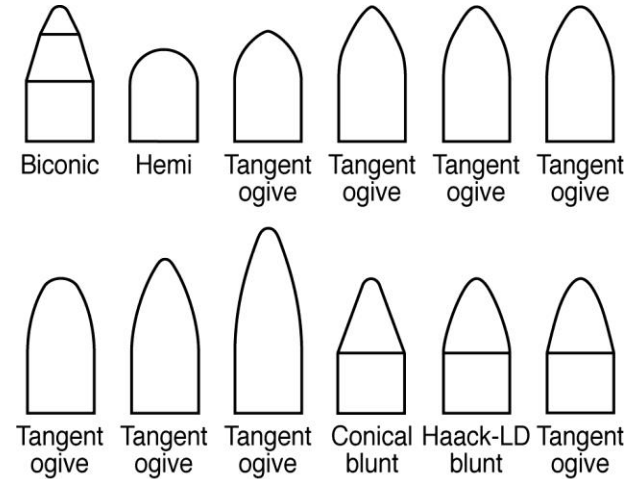
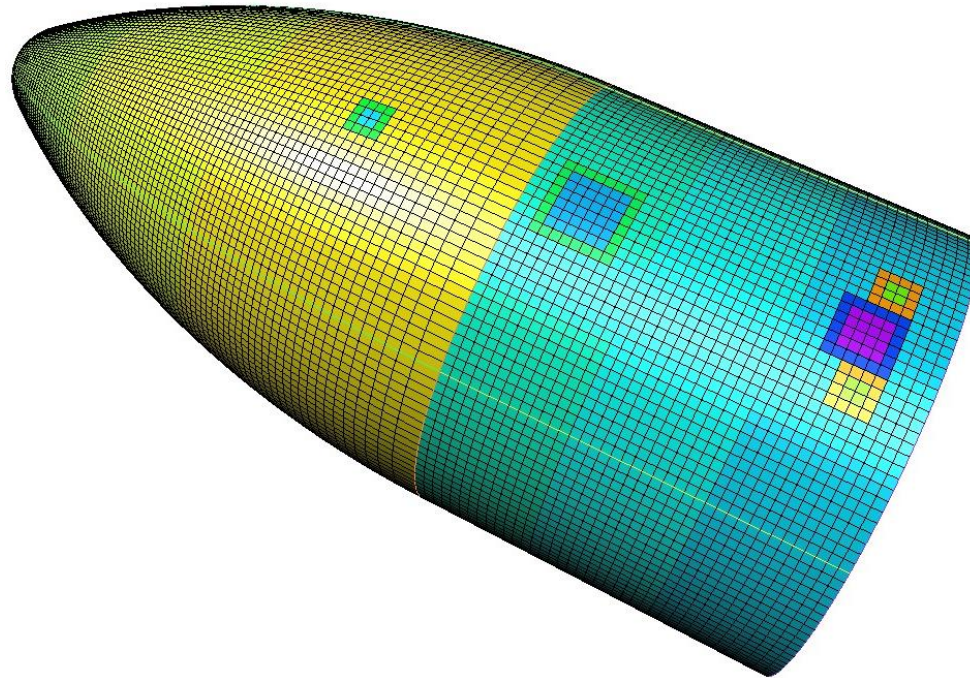
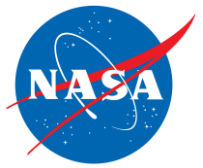
RALV-B

# Airbreather & TSTO Orbiter

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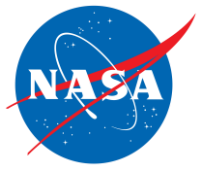


# NASA Ares V/SLS Payload Fairing

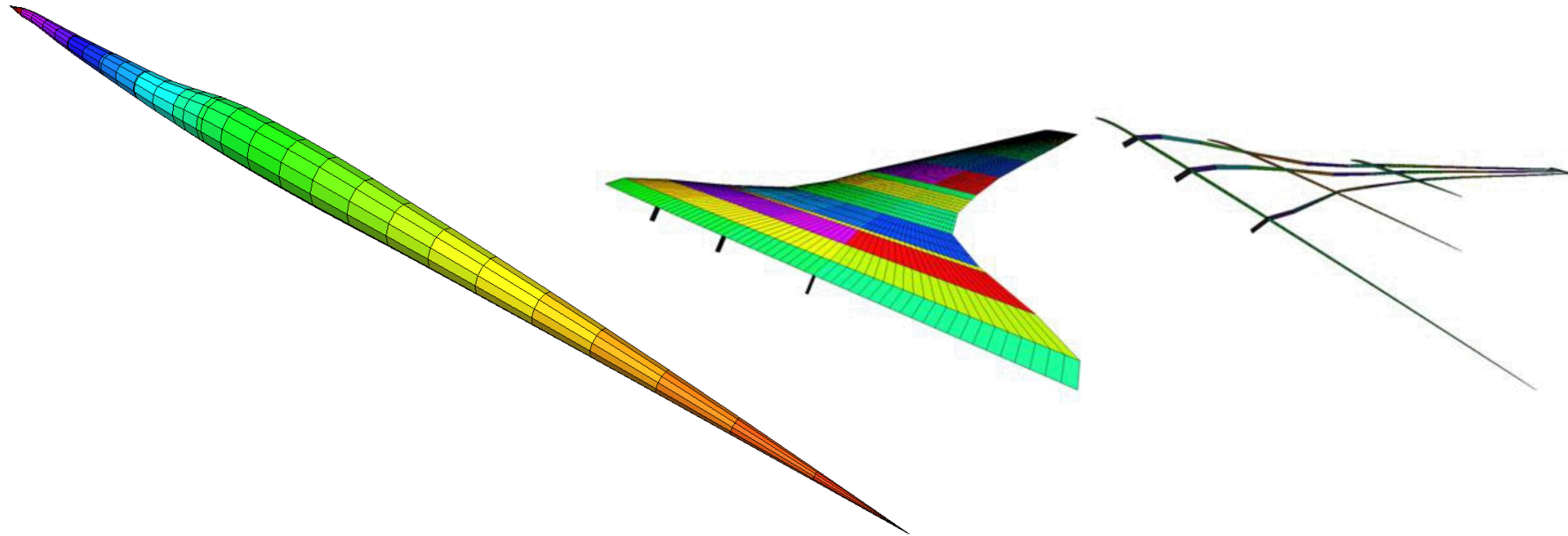


- Created and sized dozens of different shaped shroud FEA models for the shroud team over 5.5 years
- Drove NASA change to composite fairing baseline

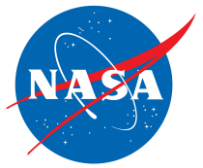
# Low Boom Supersonics



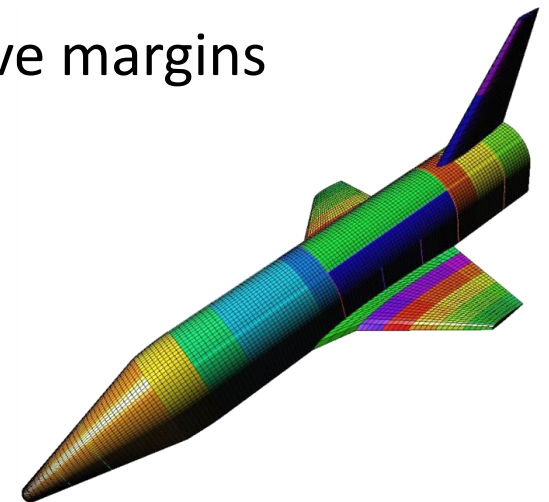
- Simple fuselage and piecewise trapezoidal wings meshes generated automatically, glued in NASTRAN, and sized in HyperSizer



# Batch Integration Tools



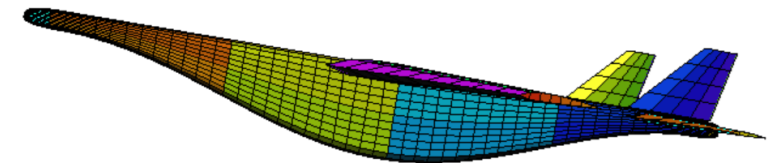
- *HSLoad*: tool that reads a text input file for instructions to drive HyperSizer's API
  - Opens database, loads model
  - Creates components and assemblies
  - Applies templates
  - Sizes
  - Post processes: final mass, list of components with negative margins
- *HXLoad*: HyperX update of HSLoad in work



# Batch Mesh Generation: *Loft*



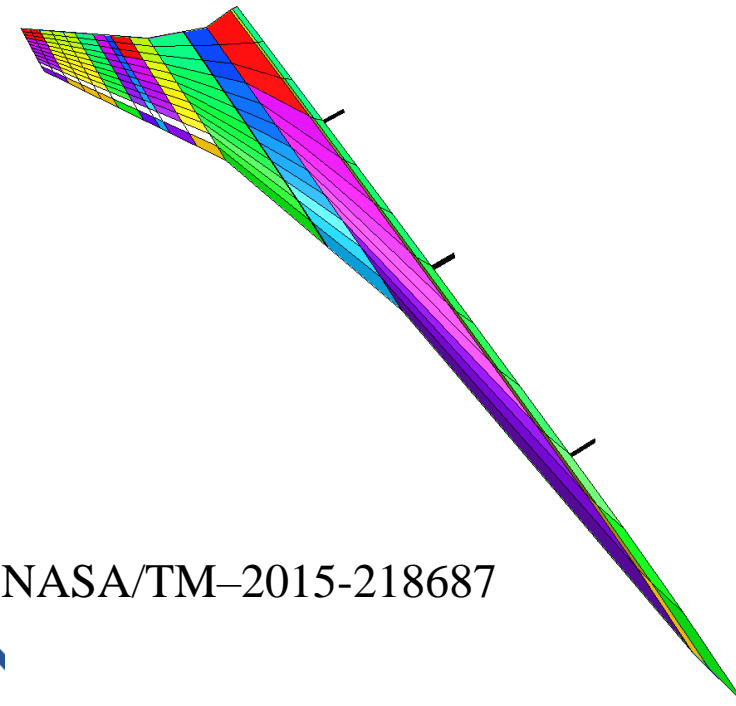
- Every mesh in this presentation was generated by Loft
- Built for easy HyperSizer/HyperX import
  - User specified object names create meaningful properties & components in HyperSizer
  - Easy setup and post processing
  - Consistent normal and material directions
- Text file input
- Outputs NASTRAN, Abaqus, Tecplot, VRML, STL
- Ideal for rapid mesh generation, parametric studies, and batch integration
- Approved for general public release by NASA



# Batch Integration: Supersonics



- Design of Experiments study for low-boom supersonic design using HSLoad, Loft, NASTRAN, and HyperSizer



Ref: NASA/TM-2015-218687



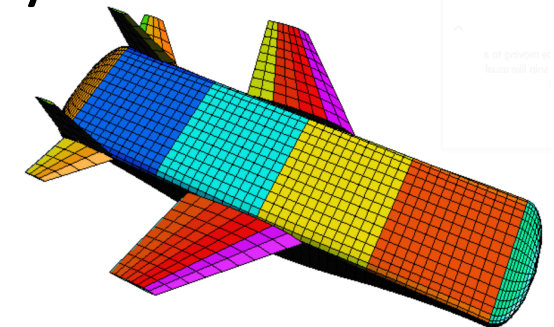
# Batch Applications: **SPARC**

*Structural Preliminary Analysis for aeRospaCe vehicles*

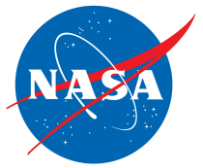
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Space Administration



- Current effort to build a low-maturity design environment where non-structural inputs are approximated and mapped to a *Loft* generated FEA model
- Uses our high fidelity analysis tools: NASTRAN and HyperX with simpler models and loads
- Application to aircraft, rockets, landers, and components like wings and tanks
- Allows for structural design insight and input in a system before detailed aero, thermal, propulsion, and trajectory analyses are available



# Twenty+ Years of Sizing



- Multidisciplinary vehicle design has been performed for
  - Vehicle components: wings and tanks
  - Lunar landers
  - Payload fairings
  - Supersonic aircraft
  - Hypersonic vehicles
- Utilities were created to drive HyperSizer and generate FEA meshes

