

# Development of a Multi-Gas Microsensor Array for the Exploration Portable Life Support System

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# Introduction to Makel Engineering Inc.

## MEI Formed 1996

- HQ in Chico California

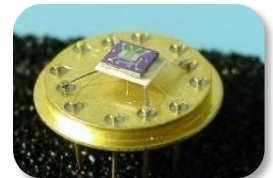
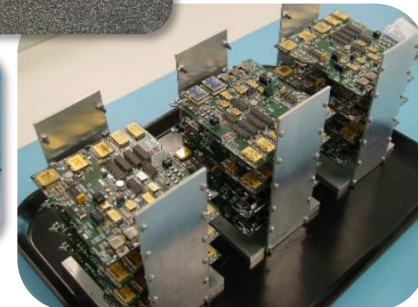
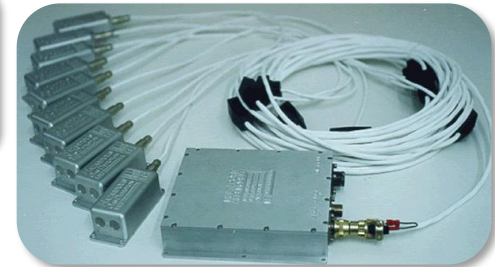
## Multidisciplinary Team

- Chemical, Mechanical and Electrical Engineering
- Technicians and support staff

## Facilities

- 16,000 ft<sup>2</sup>
- Office, laboratories, manufacturing

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# MEI Sensing Systems

## □ Aerospace

- Launch Vehicles and ISS
- Test Facilities
- Planetary Exploration



## □ Defense

- Aircraft Life Support Systems
- Advanced Propulsion System Sensors
- Warfighter Physiological Monitoring



## □ Industrial

- Emissions Monitoring
- Nuclear Systems Monitoring
- Biomedical



# Overview

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- Background and Motivation
  - Need for multi-gas measurement
- Sensor Operating Principles
  - Solid-state microsensors for O<sub>2</sub> and CO<sub>2</sub>
- Sensor Assembly Design
  - Size and power compared to existing sensors
- Prototype Testing and Performance
  - Measurement accuracy and range
- Conclusions and Next Steps

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# Background and Motivation

# Background and Motivation

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- ❑ New spacesuits are being developed to support exploration objectives to the Moon and beyond
- ❑ Technology gaps have been identified during the development of these spacesuits
- ❑ There is a need to monitor multiple species in the breathing gas stream (O<sub>2</sub>, CO<sub>2</sub>, Humidity)

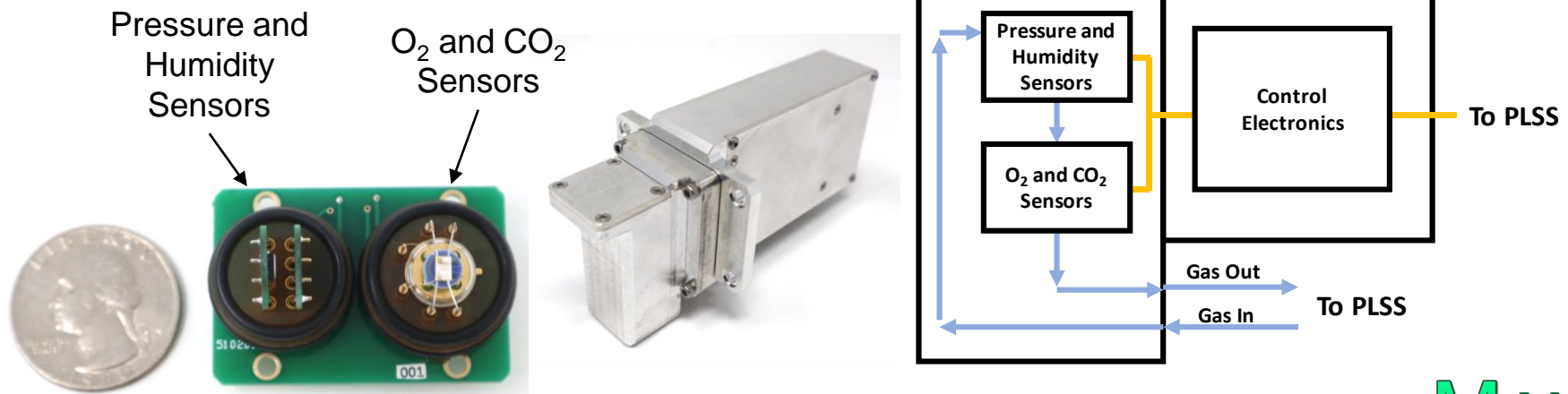
# Sensor Requirements

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- Current PLSS design only includes nondispersive infrared (NDIR) sensors for CO<sub>2</sub>
  - NDIR sensors outer mold is approximately 2.3 by 2.2 by 6.1 inches and power consumption is approximately 2 W
- Need to measure the major constituents of the breathing gas to provide general situational awareness
  - O<sub>2</sub> (20-100% ±1%)
  - CO<sub>2</sub> (0-30 torr ±0.3 torr)
  - H<sub>2</sub>O (5-90% Relative Humidity ±1%)

# M-PALSS

- ❑ Multi-Parameter Astronaut Life Support Sensor
  - O<sub>2</sub>, CO<sub>2</sub>, Humidity, and Pressure
- ❑ Suitable for ground testing and compatible with existing NASA test equipment
- ❑ Does not meet all requirements for space flight



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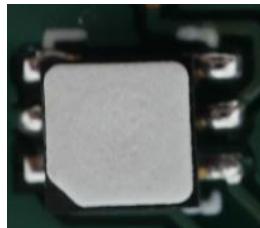
# Sensor Operating Principles

# Solid-State Sensing

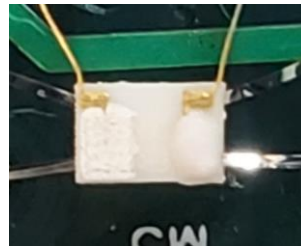
- ❑ Directly transduces a chemical signal to an electrical signal (resistance, current, or voltage)
- ❑ Small and low power
- ❑ Good match for PLSS situational awareness requirements



Pressure Sensor



Humidity Sensor



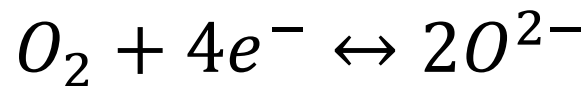
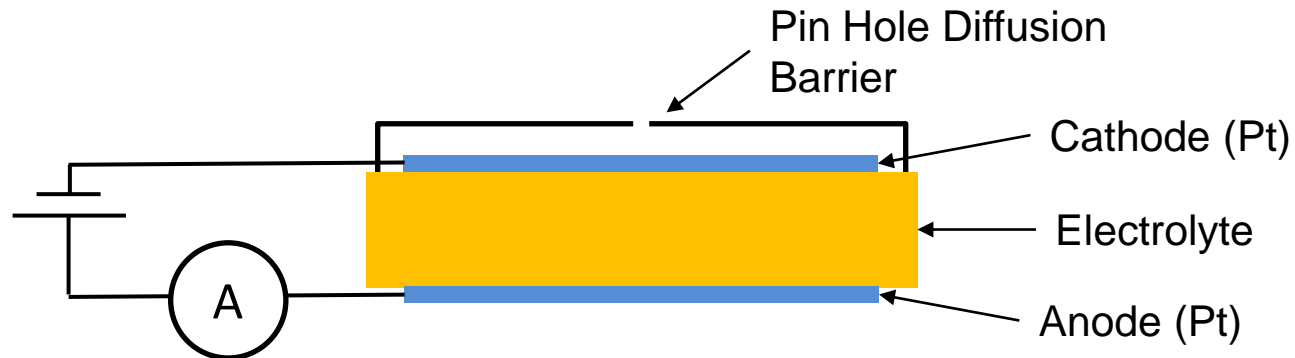
Carbon Dioxide  
Sensor



Oxygen Sensor

# Amperometric Oxygen Sensor

- Oxygen sensor produces electrical current when a voltage bias is applied across the electrodes



$$J_{O_2} = -D_{O_2}A \left( \frac{dC_{O_2}}{dx} \right) + J \left( \frac{C_{O_2}}{C} \right)$$

# Oxygen Sensor Calibration Derivation

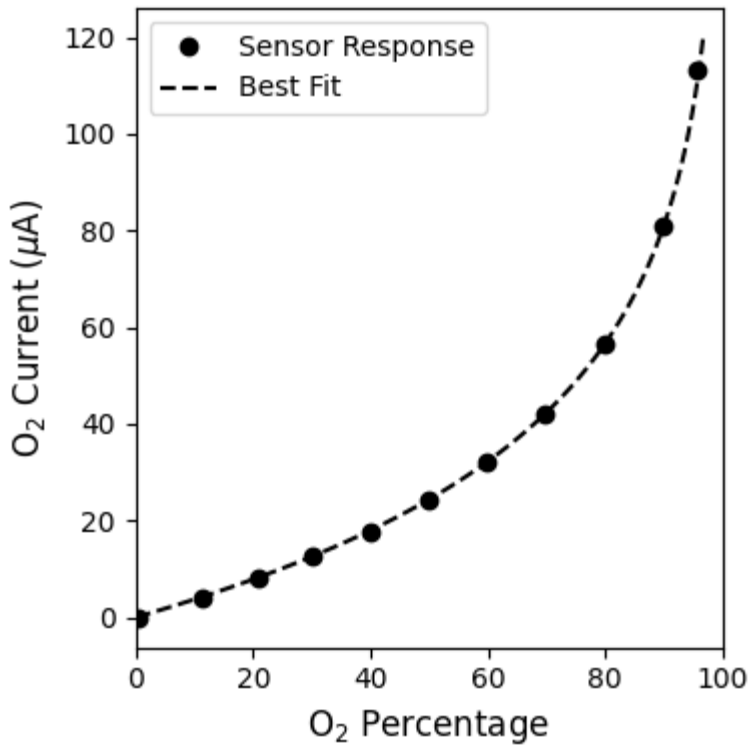
$$J_{O_2} = -D_{O_2}A \left( \frac{dC_{O_2}}{dx} \right) + J \left( \frac{C_{O_2}}{C} \right)$$

$$I_{sensor} = 4FJ_{O_2}$$

$$I_{sensor} = \underbrace{\frac{4FD_{O_2}AC}{L}}_{k} \ln \left( \frac{1 - \frac{C_{O_2}(L)}{C}}{1 - \frac{C_{O_2}(0)}{C}} \right) \quad \frac{C_{O_2}(L)}{C} \ll 1$$

$$I_{sensor} = -k \ln \left( 1 - \frac{C_{O_2}(0)}{C} \right)$$

# Oxygen Sensor Representative Data

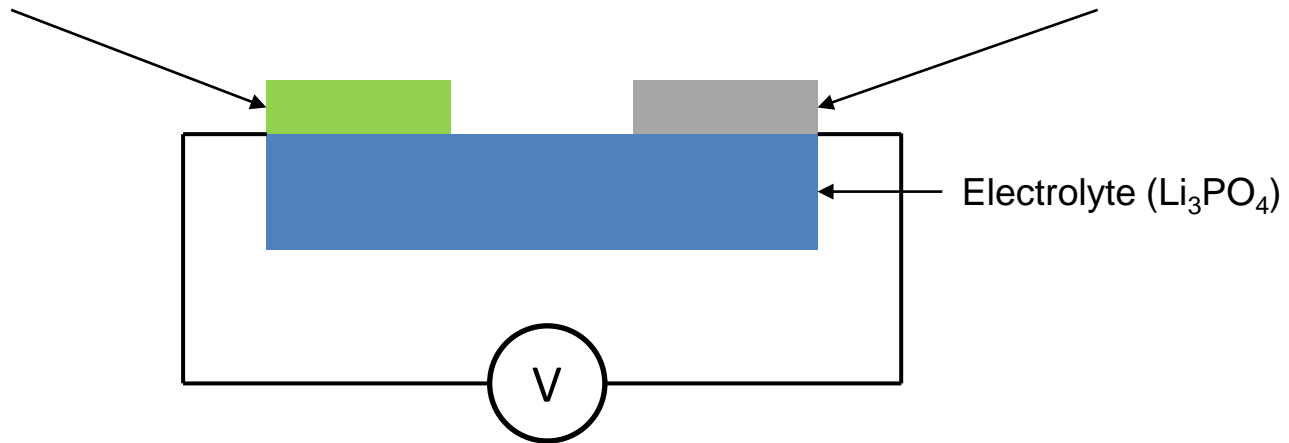
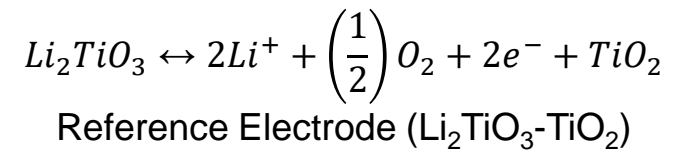
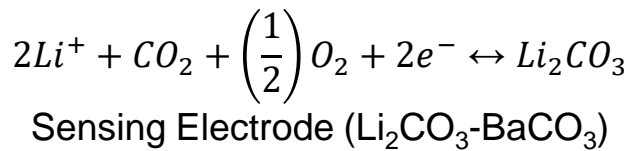
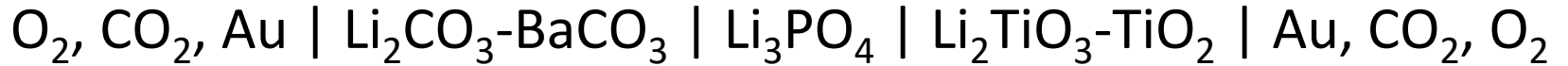


$$I_{sensor} = -k \ln \left( 1 - \frac{C_{O_2}(0)}{C} \right)$$

$$k \sim 35 \mu A$$

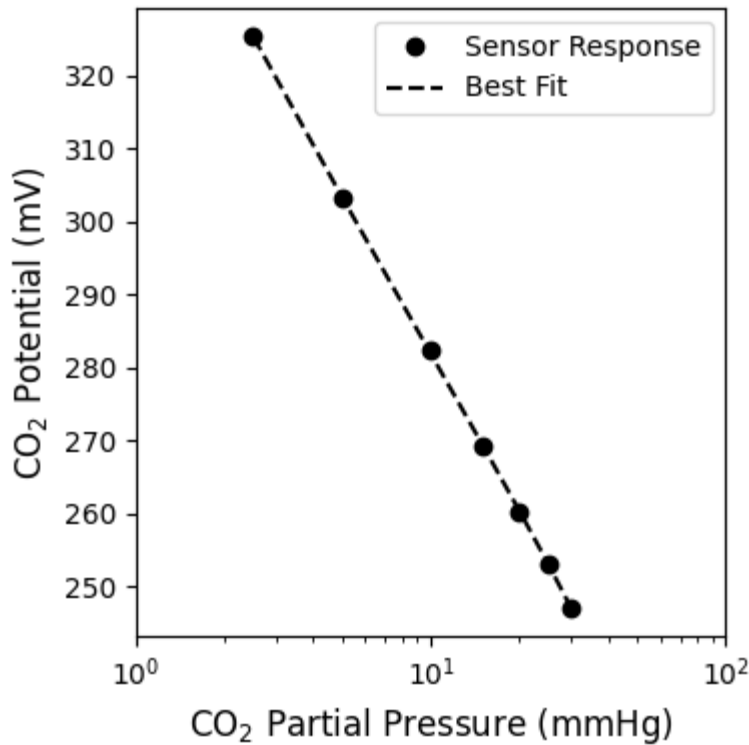
Saturation occurs at high oxygen concentration

# Potentiometric Carbon Dioxide Sensor



$$E_{cell} = E_{cell}^0 - \frac{RT}{2F} \ln \left( \frac{P_{CO_2}}{P_0} \right)$$

# Carbon Dioxide Representative Data



$$E_{cell} = E_{cell}^0 - \frac{RT}{2F} \ln \left( \frac{P_{CO_2}}{P_0} \right)$$

$$V = V_0 - m \log \left( \frac{P_{CO_2}}{P_0} \right)$$

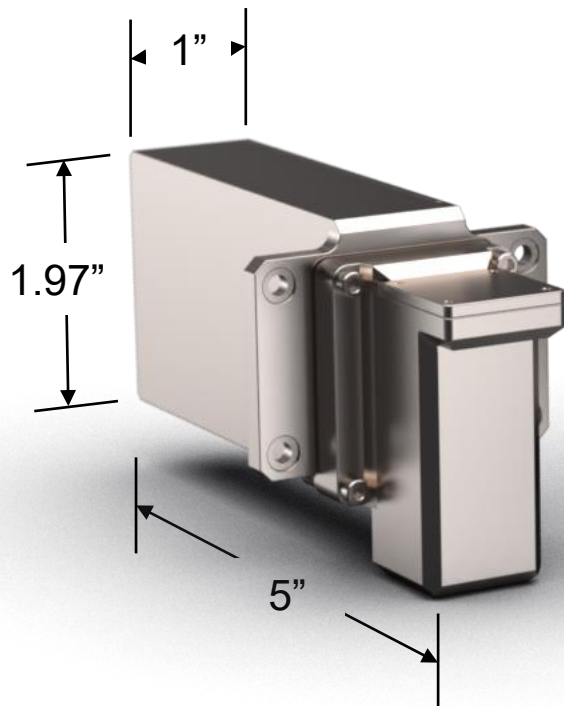
$$Slope (m) \sim 65 \frac{mV}{decade}$$

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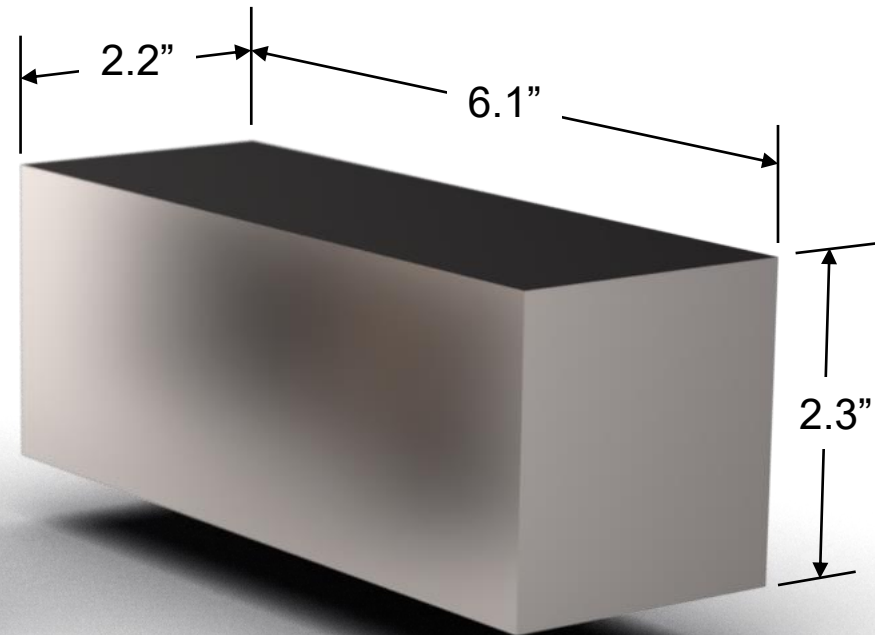
# Sensor Assembly and Design

# Size and Power

- Phase I Prototype is smaller than current NDIR sensors
  - Power consumption of approximately 1.8 W (5 V)



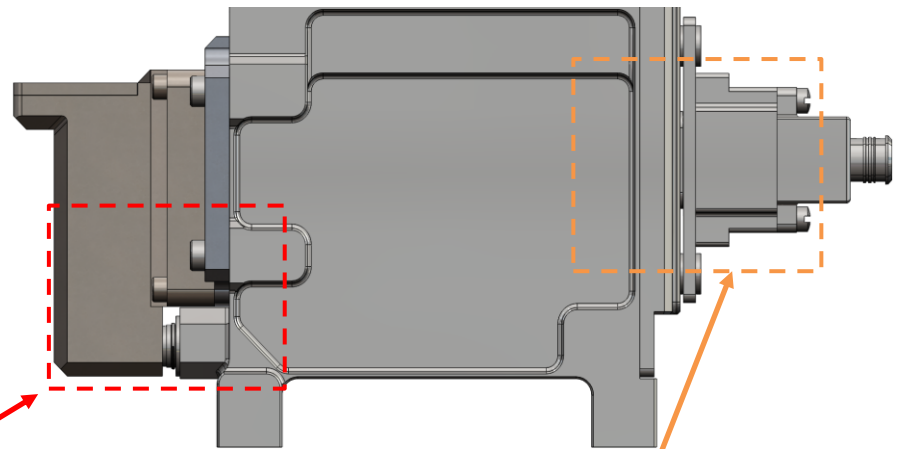
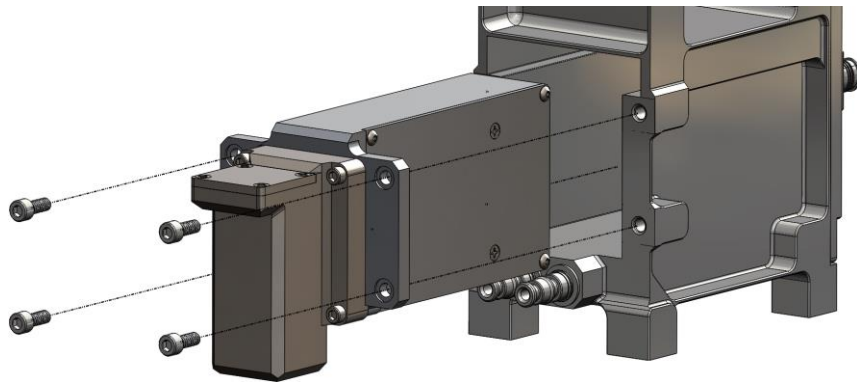
Phase I Prototype



NDIR Sensor Approximate  
Outer Mold Line

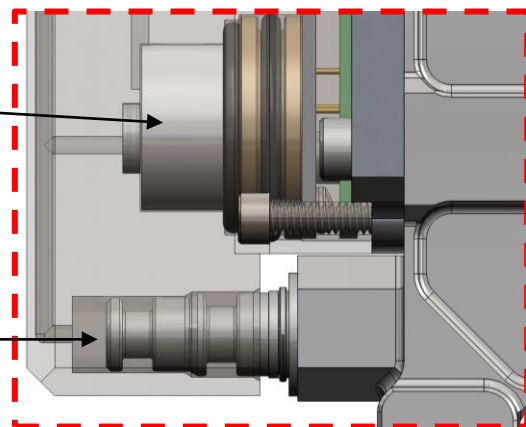
# Preservation of Sensor Interfaces

**Mechanical Interface**



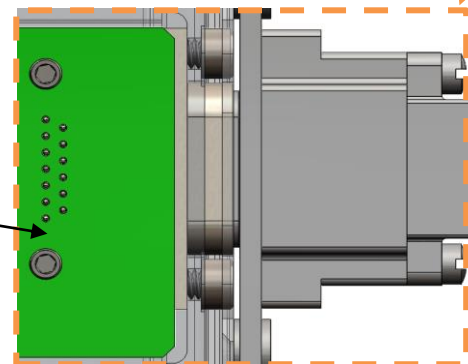
Sensor

Fluidic Port



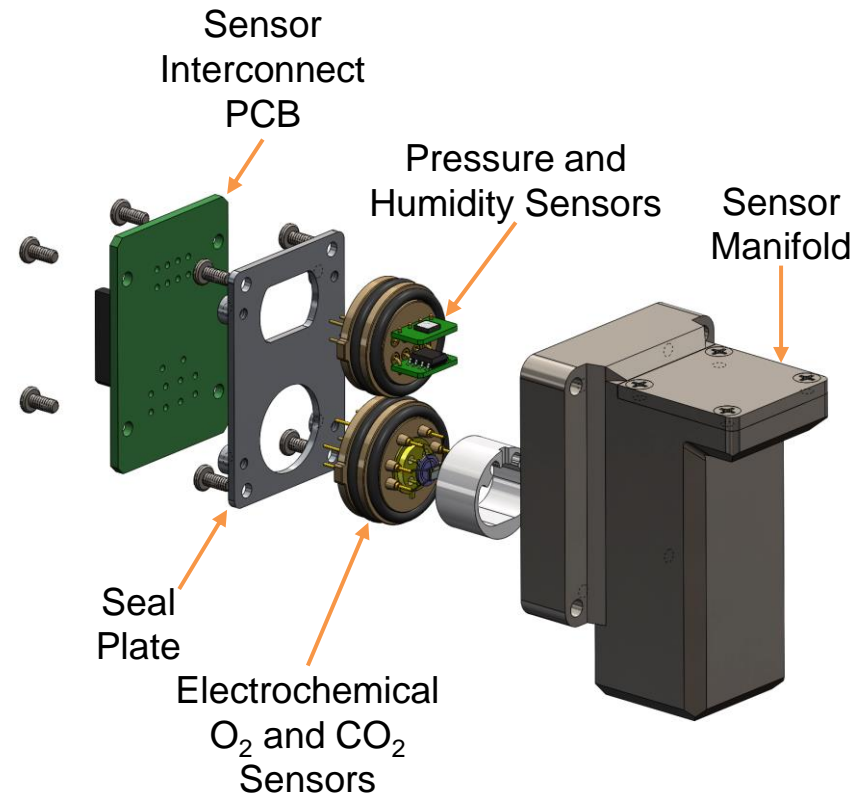
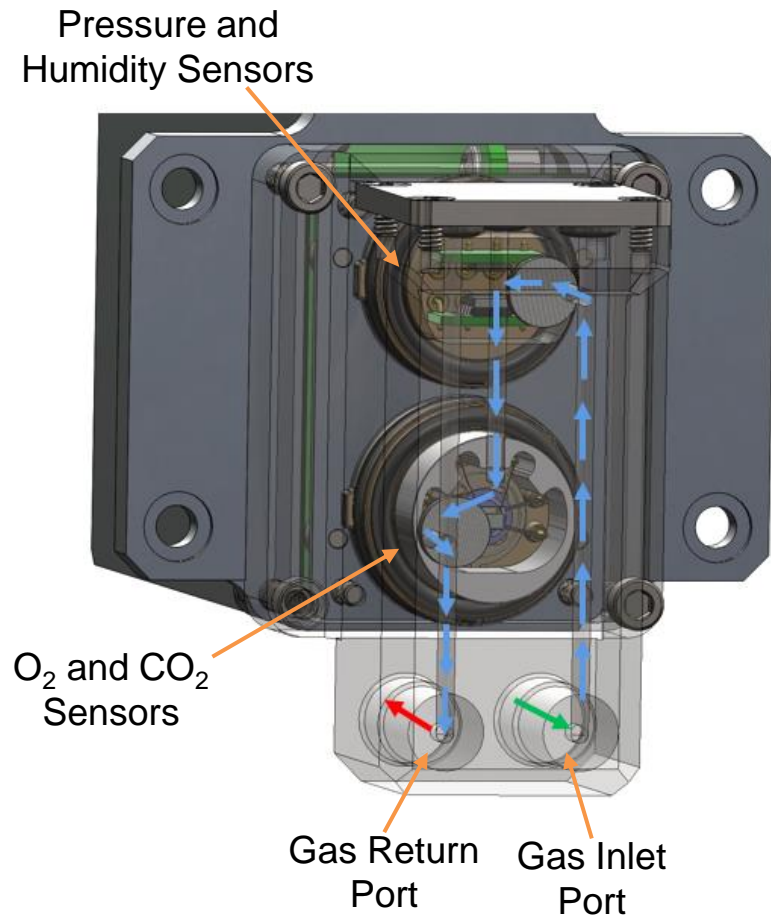
**Fluidic Interface**

Control Electronics



**Electrical Interface**

# Sensor Module

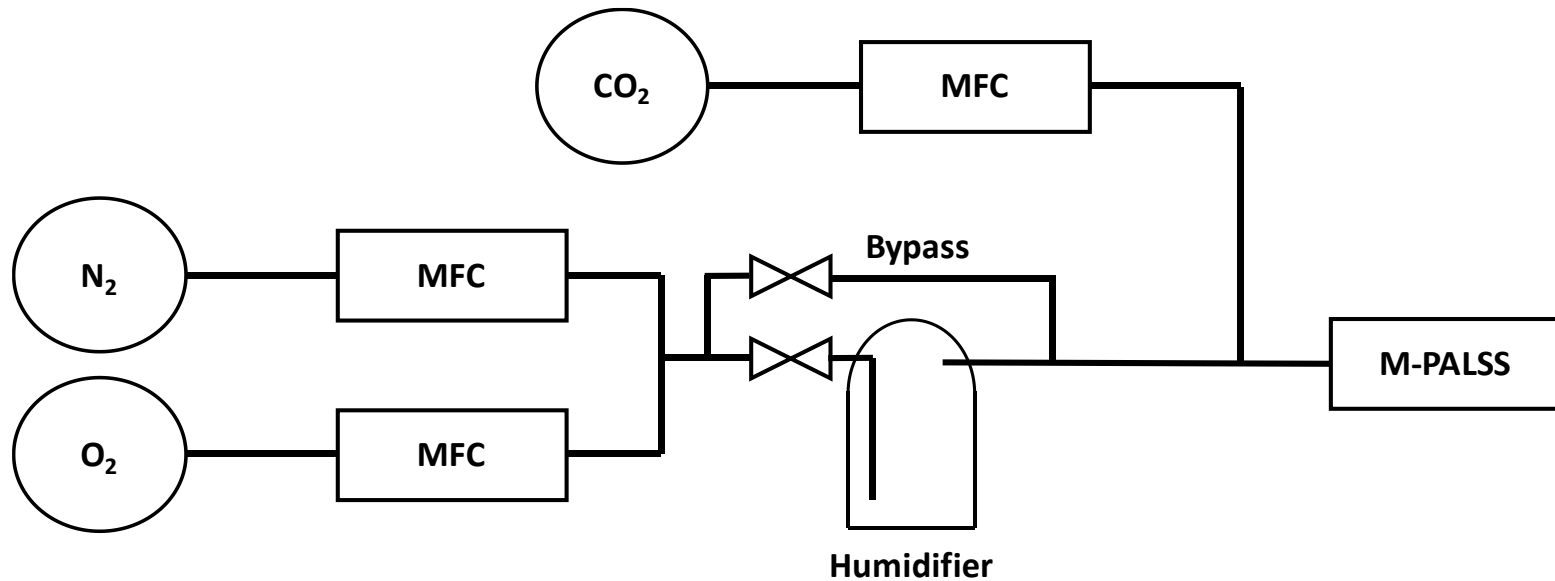


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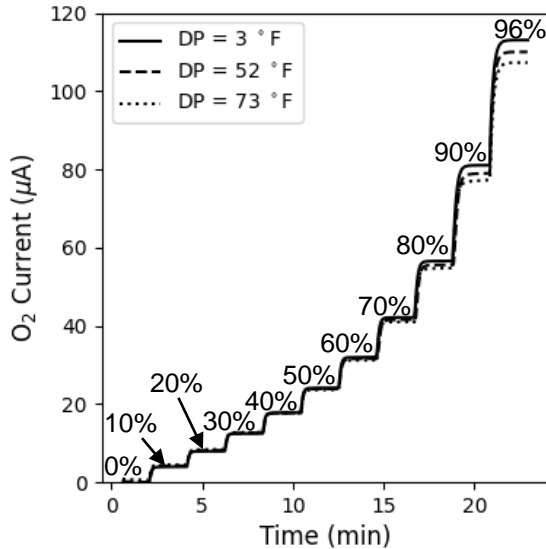
# Prototype Testing and Performance

# Gas Testing Configuration

- ❑ Certified gas mixtures are mixed using mass flow controllers (MFCs)
- ❑ Inline bubbler used to control the humidity



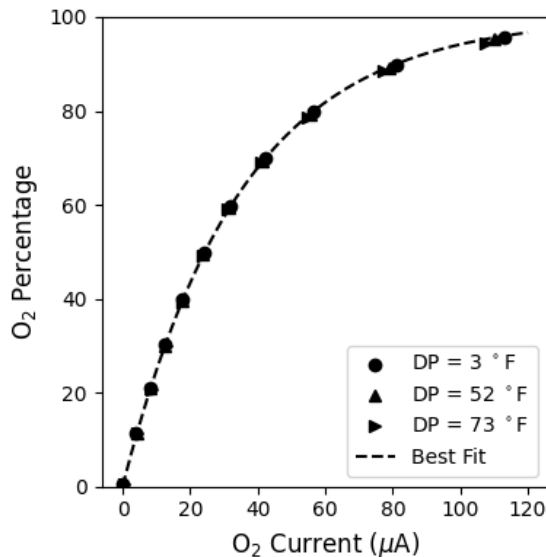
# Oxygen Sensor Performance



Testing conducted at 14.7 psia

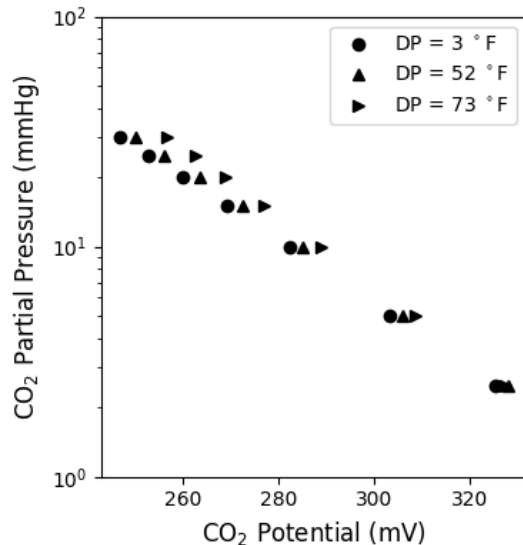
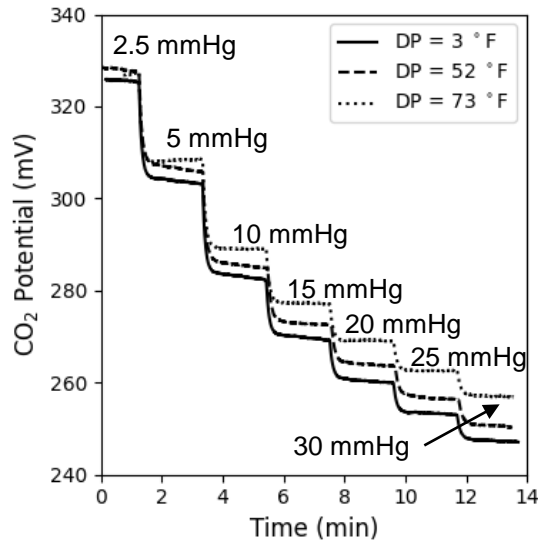
Response time  $t_{90} \sim 3s$

Annotated percentages are prior to humidification



$$O_2(\%) = 100(1 - e^{-\frac{I}{k}})$$

# Carbon Dioxide Sensor Performance



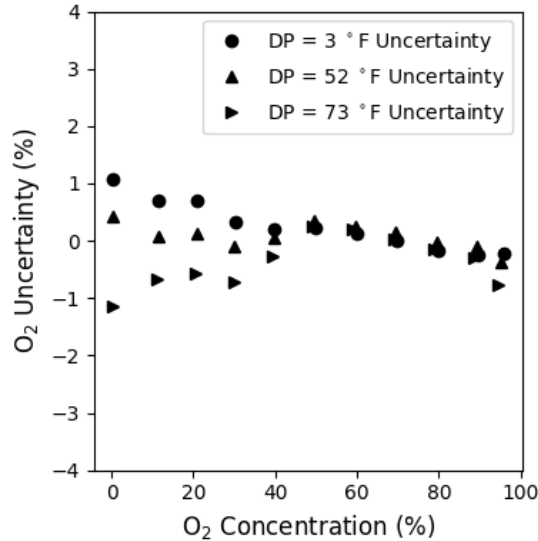
Testing conducted at 14.7 psia

Response time  $t_{90} \sim 12s$

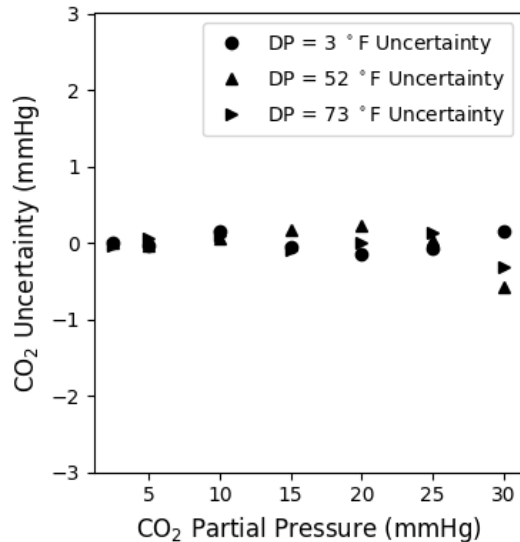
CO<sub>2</sub> sensor humidity interference is compensated by using the independent humidity sensor

$$P_{CO_2} = P_0 \left( 10^{-\frac{V-V_0}{m}} \right)$$

# Measurement Uncertainty



O<sub>2</sub> sensor meets target requirements of ±1% with exception 0% O<sub>2</sub>



CO<sub>2</sub> sensor meets target requirement of ±0.3 mmHg with exception of 30 mmHg after humidity correction

$$P_{CO_2} = P_0 \left( 10^{-\frac{V-V_0(RH)}{m(RH)}} \right)$$

# Conclusion and Future Work

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- Work presented here completed as part of NASA Phase I SBIR Contact 80NSSC22PB085
  - Phase II work is ongoing
- Solid-state microsensors packaged into a prototype that preserves interfaces of current NDIR CO<sub>2</sub> sensors and meets target measurement accuracy
- Next steps of development are focused on meeting all the requirements for spaceflight and integration with the PLSS