



A relocatable lander to explore Titan's  
prebiotic chemistry and habitability

**Study of the thermal physical properties of insulating  
materials in a Titan environment**

**July 9-13, 2023 CEC/ICMC Honolulu, HI**

Richard Ottens, Peter Barfknecht, Kelly Burch, Steven Cale,  
Brian Comber, Matthew Francom, Hak Seung Lee, Paul Rueger



# Outline



- What is Dragonfly?
- What is DraMS?
- Why the need for insulators?
- Requirements for the Cold Zone
- Insulation Options
- Conductivity Rig
- Results
- Conclusions
- DraMS Cryo Team



# What is Dragonfly



- Dragonfly is a rotorcraft that will explore the chemistry of the surface of Saturn's Moon Titan.
- It will take advantage of the unique conditions of Titan's dense atmosphere (1.5 atm) and low gravity ( $1.4 \text{ m/s}^2$ ) to allow Dragonfly to fly from site to site.
- Dragonfly will be looking for prebiotic compounds on the surface, as well as studying the composition of the surface and atmosphere



John Hopkins APL



# What is DraMS



- Dragonfly Mass Spectrometer (DraMS) is a linear ion trap mass spectrometer consisting of two modes
  - Laser Desorption Mass Spectrometry: to measure the composition of surface samples targeting large organic molecules [1]
  - Gas Chromatography Mass Spectrometry: to separate and identify prebiotic molecules by volatilization. [1]
- DraMS is split into two thermal sections
  - Warm Instrument Zone (~Room Temp)
  - Cold Sample Zone (~150K, Titan temp ~94K)

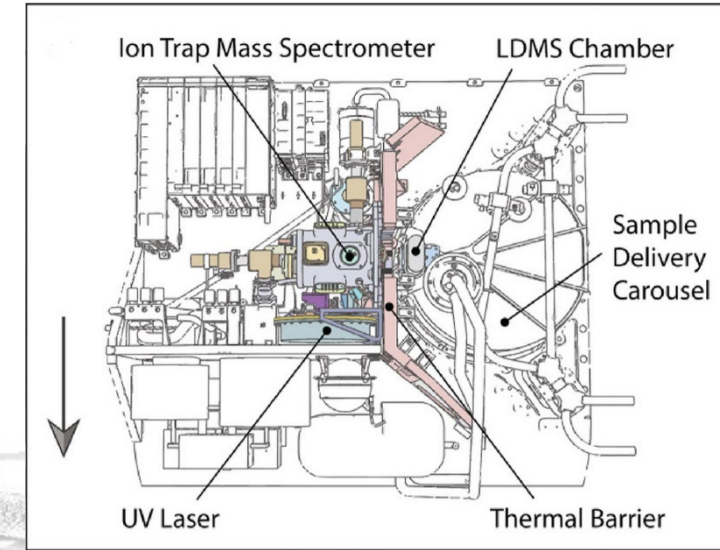
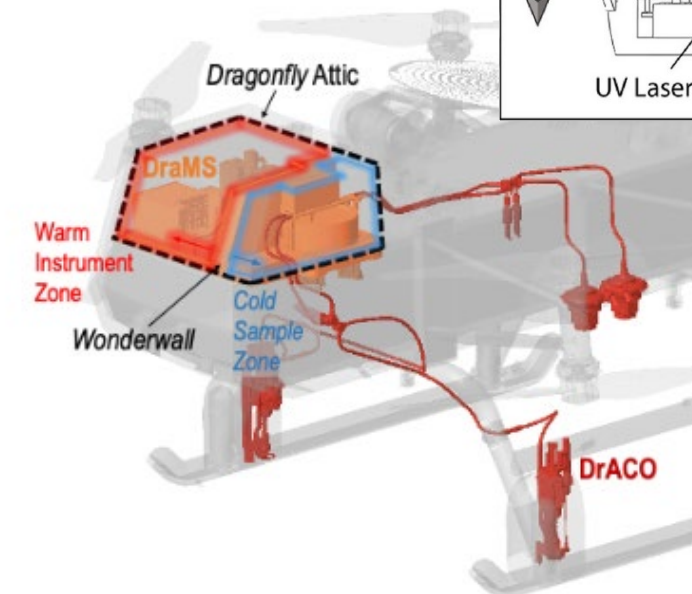


Figure 1. The Dragonfly "Attic" Cold Zone and Warm Zone. [1]



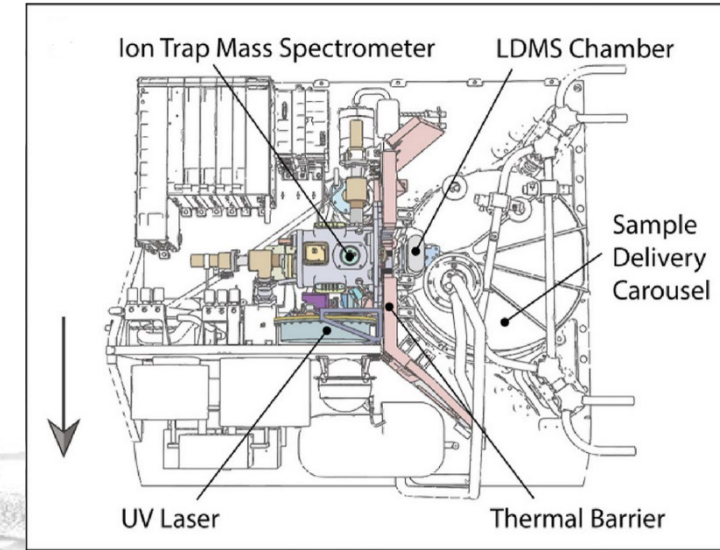
# Why the Need for Insulators



- The majority of Dragonfly's interior
  - Held at around room temperature
  - Semi-hermetic to the Titan atmosphere
- Dragonfly has a Multi-Mission Radioisotope Thermoelectric Generator (MMRTG) to supply limited electrical power and heat.
- Temperature differentials of  $\sim 150\text{K}$  only a few inches
- As a result, Dragonfly needs to be well insulated and standard cryogenic insulations (MLI, vacuum jackets) will not work due to convection and mass restrictions.



John Hopkins APL



[2]

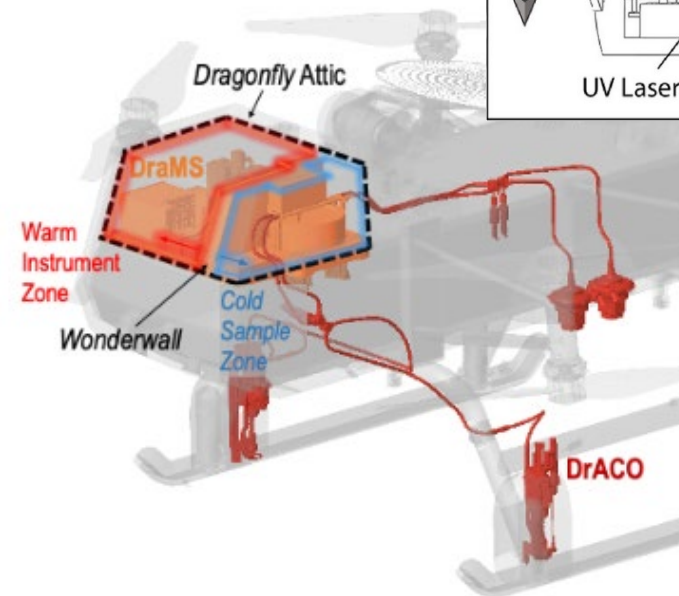


Figure 1. The *Dragonfly* "Attic" Cold Zone and Warm Zone. [1]

- Within the DraMS cold zone
  - 4 of 6 sides are near room temperature
  - Must be insulated to prevent excessive heat loss
- Within the Sample Delivery Carousel
  - Sample cups must stay cold to preserve sample integrity before test and during storage
  - The use of materials is limited to prevent overwhelming the mass spectrometer with reading of the lander itself

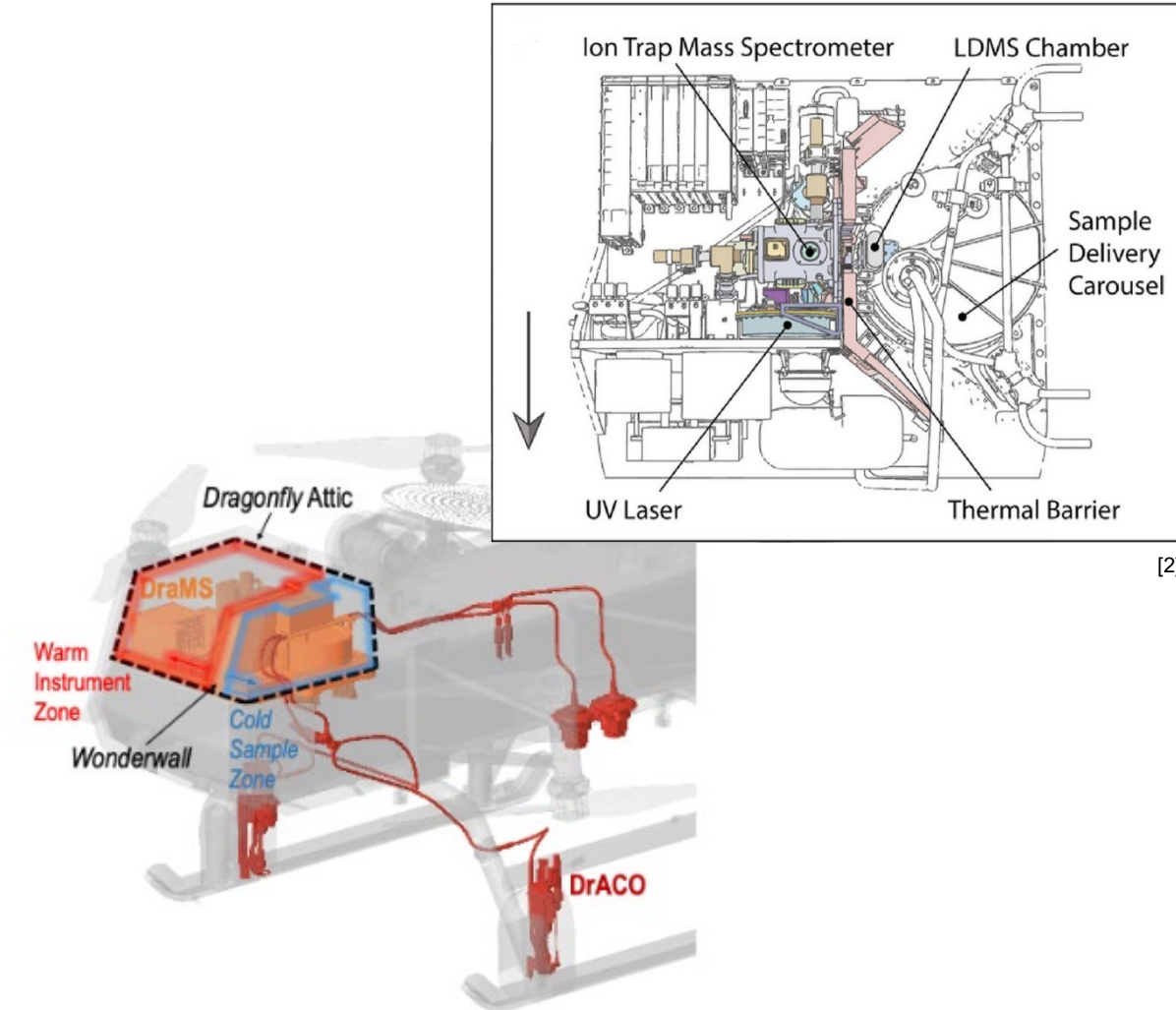


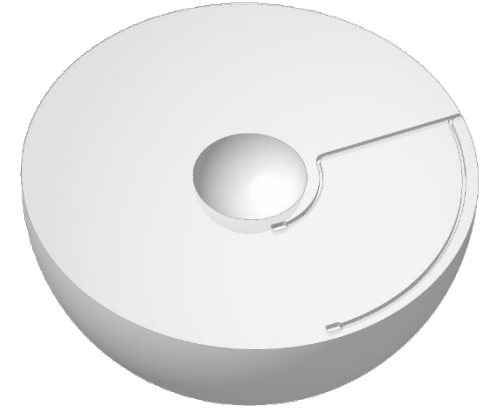
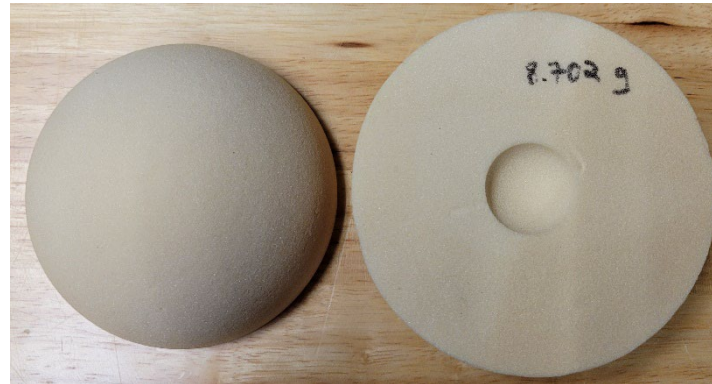
Figure 1. The *Dragonfly* "Attic" Cold Zone and Warm Zone. [1]



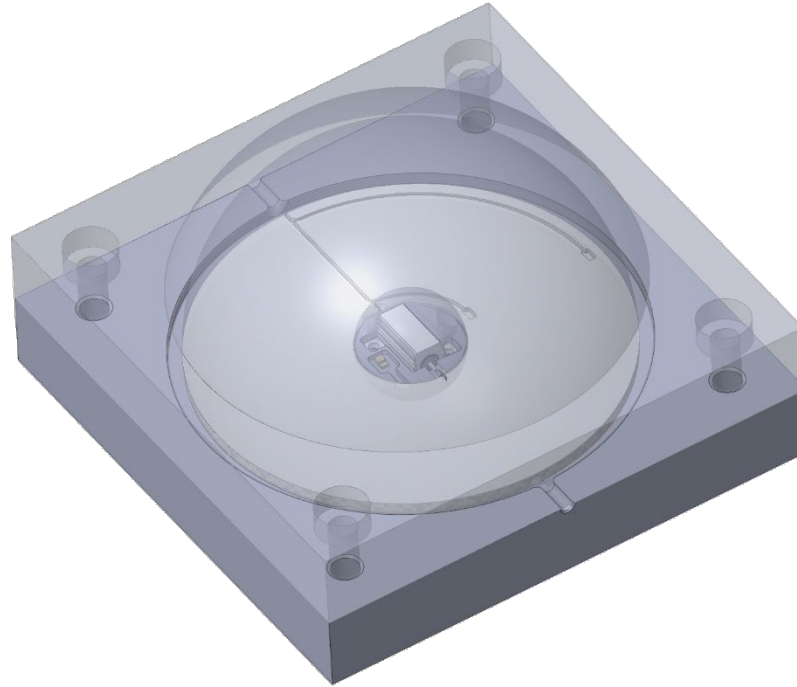
# Insulation Options (Published Room Temp Data)



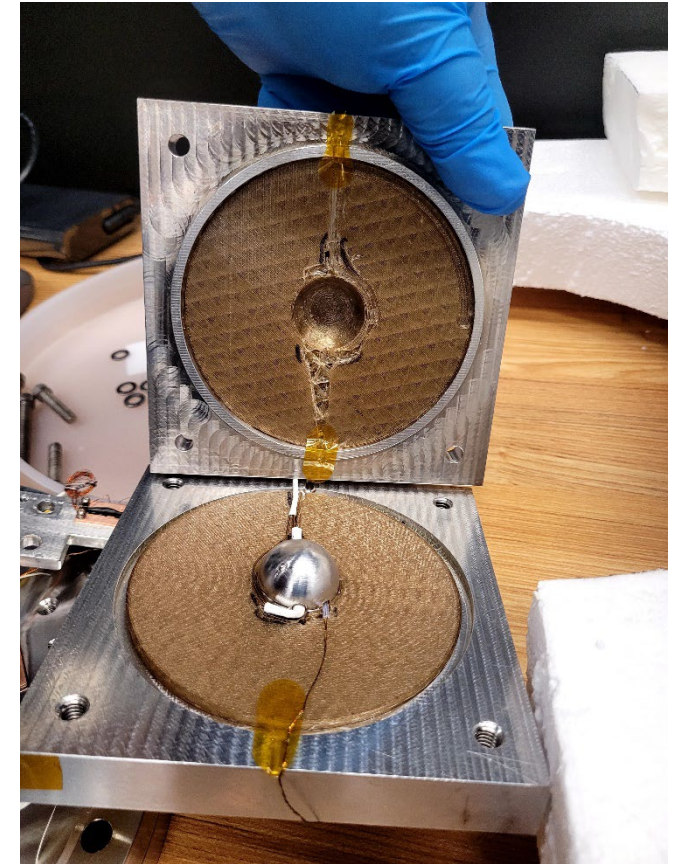
- Rohacell 31-HF (polymethacrylimide)
  - Density:  $32 \pm 7 \text{ kg/m}^3$  [3]
  - $31 \text{ mW/(m-K)}$  @  $20^\circ\text{C}$  [4]
- ELFOAM P200 (polyisocyanurate)
  - Density:  $32 \text{ kg/m}^3$  [5]
  - $24 \text{ mW/(m-K)}$  @  $24^\circ\text{C}$  [5]
- PEEK (polyetheretherketone)  
(3D Printed, with cubic cell infill)
  - Density:  $1300 \text{ kg/m}^3$  (bulk, printed density depends on infill) [6]
  - $300 \text{ mW/(m-K)}$  @  $25^\circ\text{C}$  [7]
- There is a need for testing these materials at cryogenic temperatures



- Uses concentric spheres with an insulation sample in between.
- This method is superior to the longitudinal method since all thermal paths must go through the material
- Samples are 4" OD, 1" ID hemispheres.
- The inner sphere ( $T_{\text{hot}}$ ) and outer sphere shell ( $T_{\text{cold}}$ ) are both independently temperature controlled



$$Q = \frac{4\pi k(T_{in} - T_{out})r_{in}r_{out}}{r_{out} - r_{in}}$$

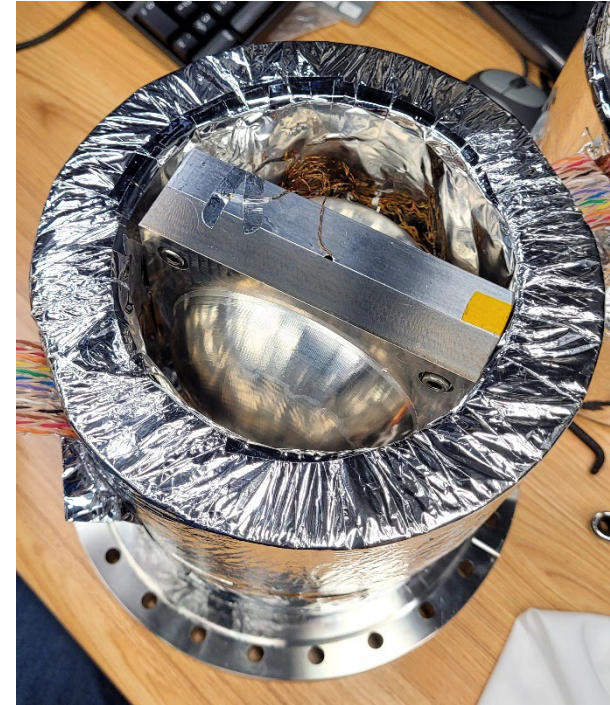
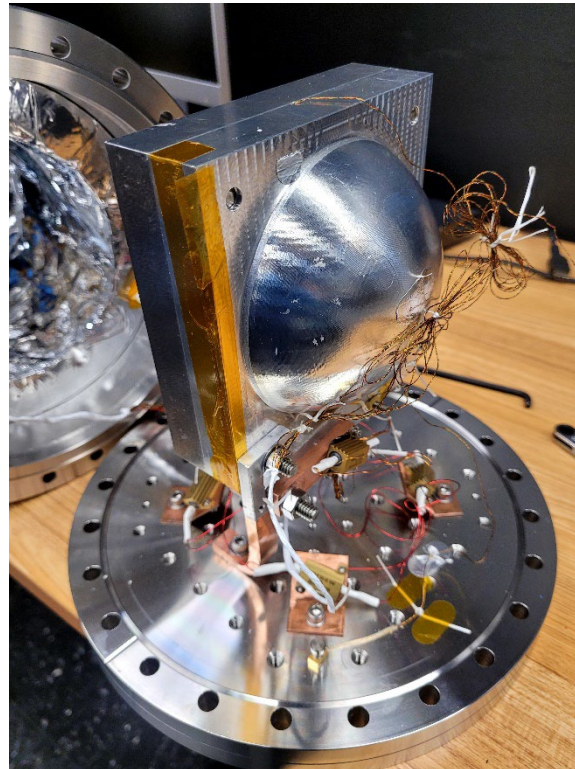




# Conductivity Rig – Sample Chamber



- The shell is attached above a temperature-controlled plate.
- The shell is then covered on all sides with Rohacell insulation.
- Everything is then inserted into a Stainless Steel chamber

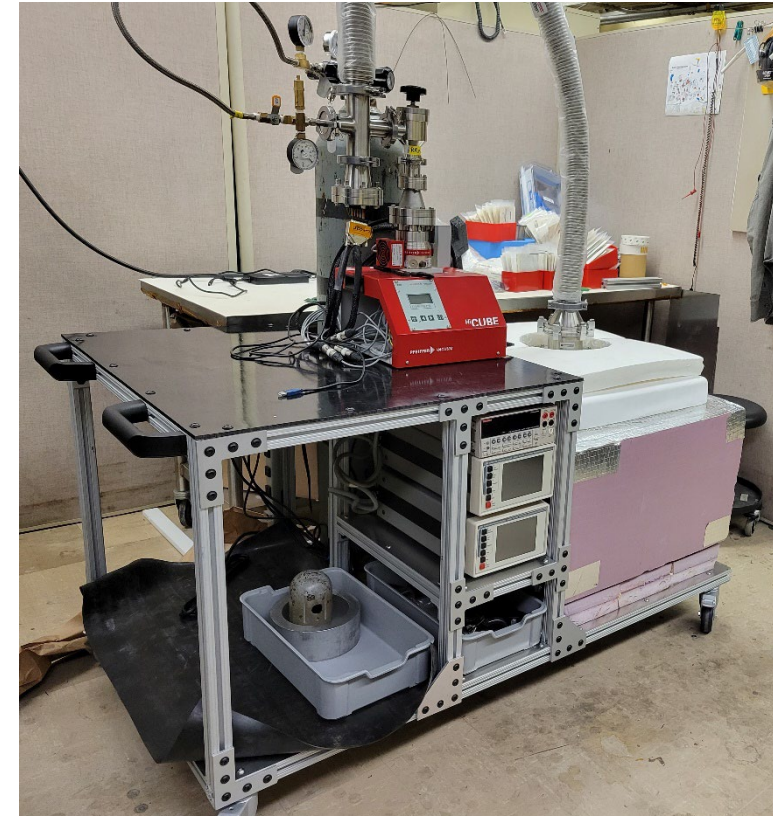
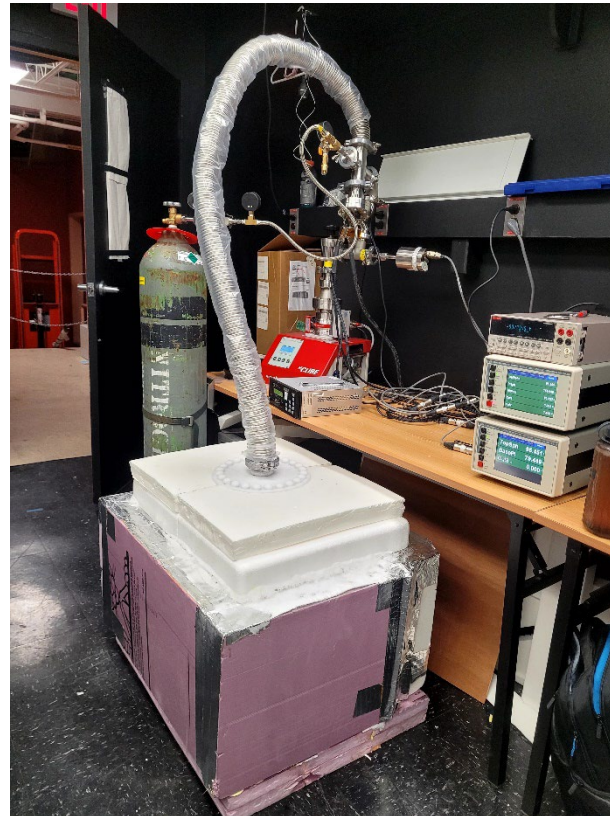




# Conductivity Rig



- The sample chamber is placed inside a LN2 bath
- The sample chamber has been designed to work from vacuum to 1.5 atm (absolute) to simulate Titan's environment
- Temperature readout and control done with two Stanford Research Systems CTC-100
- Heater output read by Keithley 2700



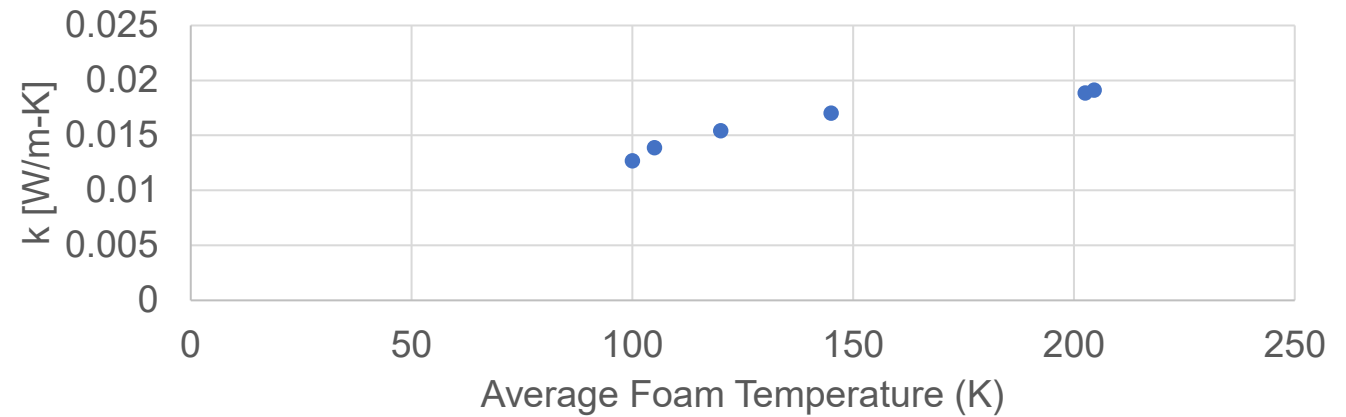


# Results - Rohacell

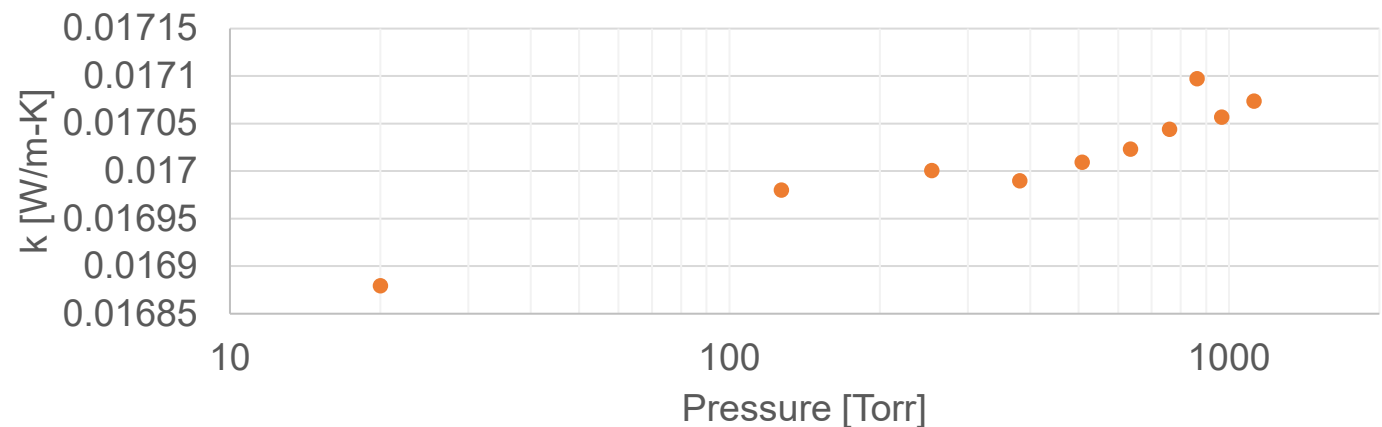


- Resulting conductivity is an effective conductivity that includes convective, conductive and radiative effects.
- Rohacell is temperature dependent
- Rohacell is a closed cell foam, thus the conductivity is minimally affected by pressure

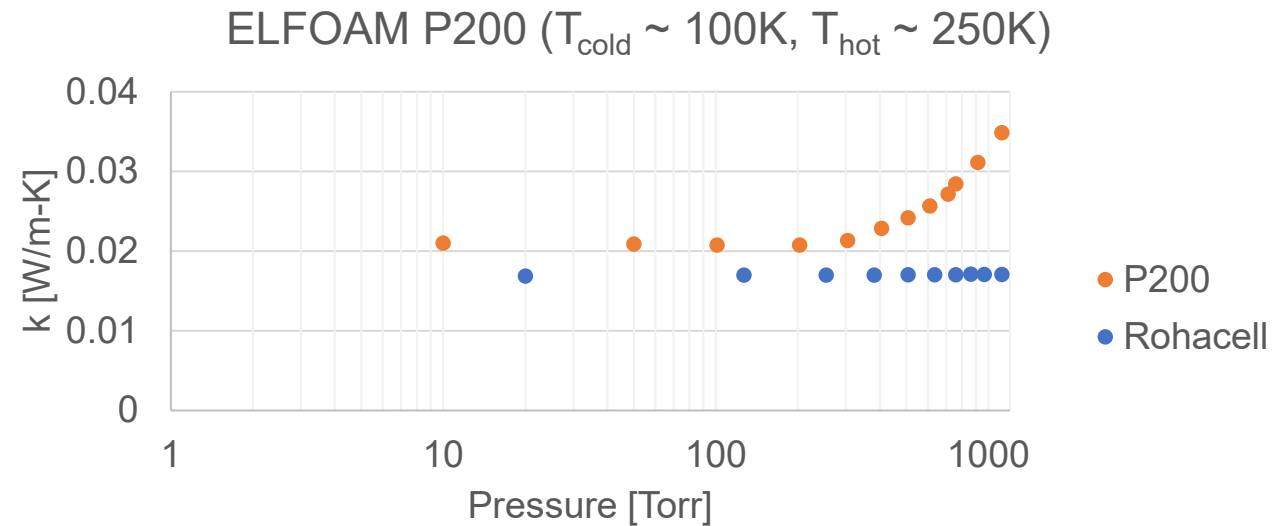
Rohacell ( $T_{\text{cold}} \sim 100\text{K}$ )



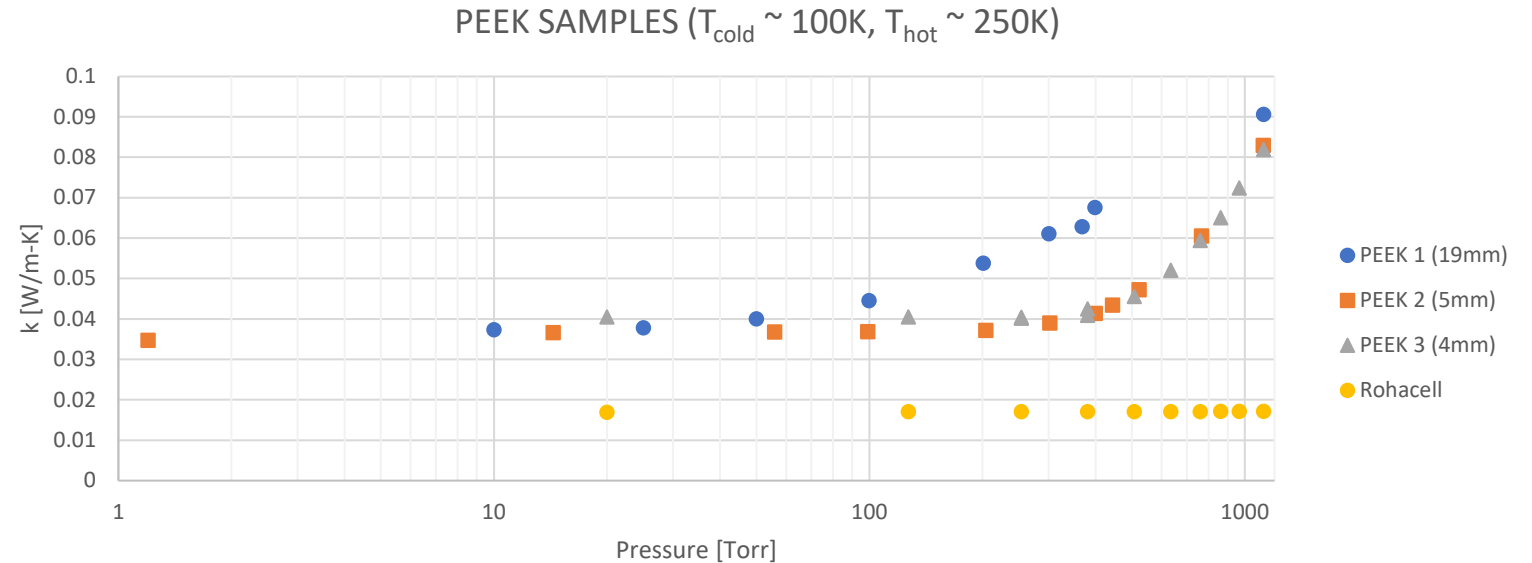
Rohacell ( $T_{\text{cold}} \sim 100\text{K}$ ,  $T_{\text{hot}} \sim 250\text{K}$ )



- ELFOAM P200 is less expensive and less volatile than Rohacell
- It is closed cell foam
- It is about 25% more conductive than Rohacell
- It does have a pressure dependence



- Tested three infills
  - 19mm cell wall (~5% infill), 0.4mm wall thickness
  - 5mm cell wall (~9% infill), 0.3mm wall thickness
  - 4mm cell wall (~10% infill), 0.3mm wall thickness
- Poor compared conductivity but has the advantage of being any shape and minimal volatility issues.





# Results - PEEK



- As cell size decreases

- Convective flow occurs at higher pressure, knee at  $Ra \sim 500$
- Conductive contribution from the PEEK increases

- Converting P to Ra

$$Ra = \frac{g \beta \Delta T l^3 \rho}{\alpha \mu}$$

g: accel. of gravity

$\beta(T,P)$ : volumetric expansion coeff. of  $N_2$

$\Delta T$ : Temp. difference of a cell =  $\Delta T$  \* (cell size/full radial distance)

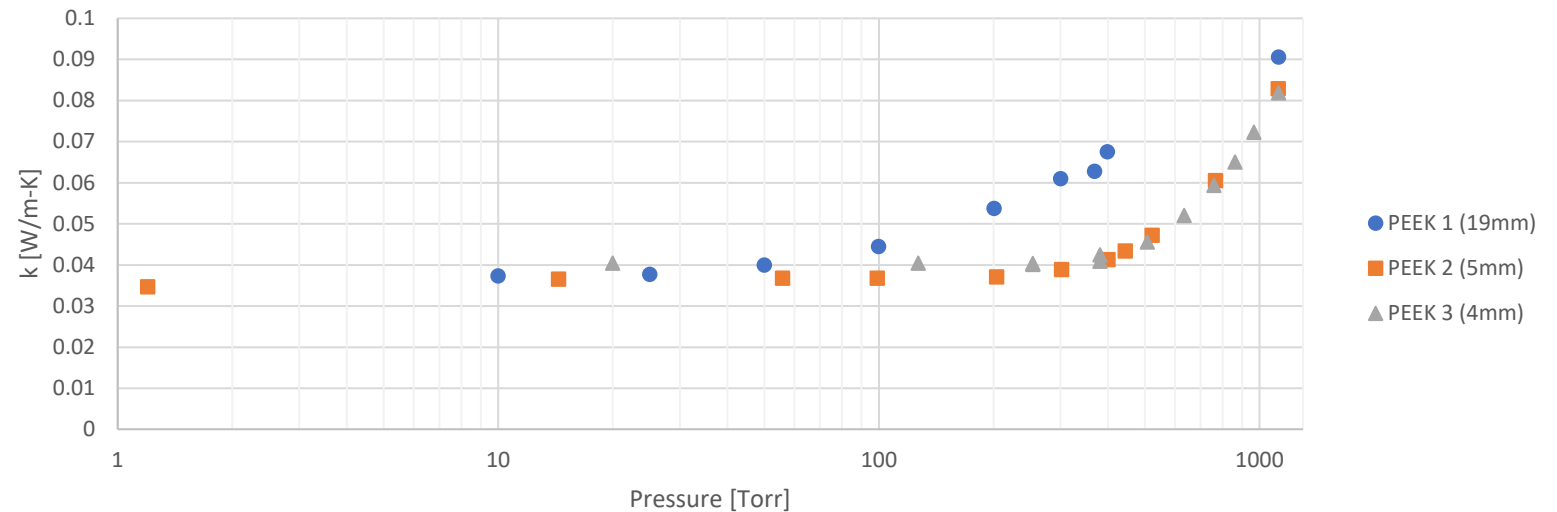
l: cell size

$\rho(T,P)$ : density of  $N_2$

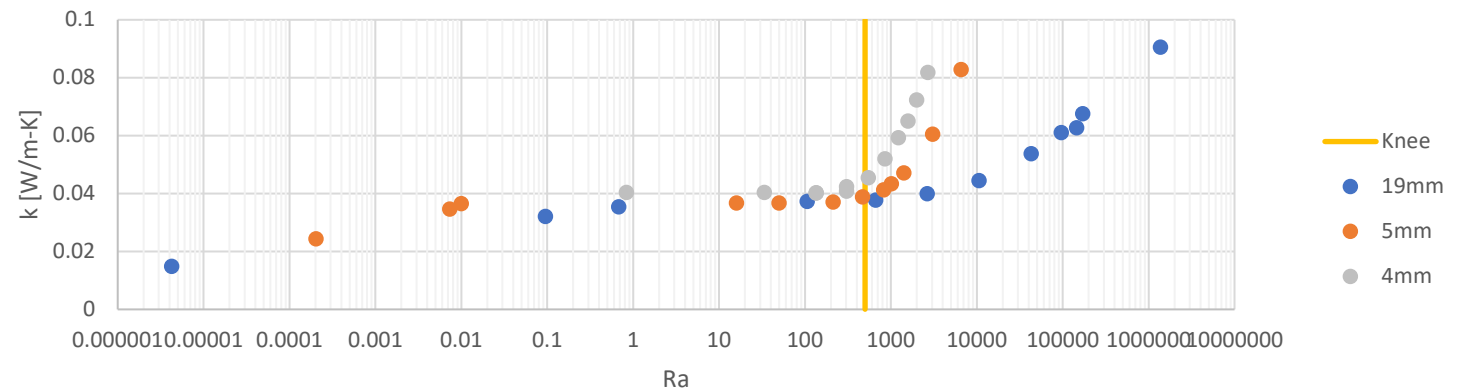
$\alpha(T,P)$ : Thermal diffusivity of  $N_2$

$\mu(T,P)$ : viscosity of  $N_2$

PEEK SAMPLES ( $T_{\text{cold}} \sim 100K$ ,  $T_{\text{hot}} \sim 250K$ )



PEEK Ra





# Conclusions



- Rohacell has the lowest thermal conductivity
- However, because Rohacell outgases, PEEK is a better option for the sample carousel. Covered or encapsulated Rohacell is a possible option
- P200 can be used as a less expensive replacement for Rohacell for testing purposes.



Mike Yakovlev, John Hopkins APL



# DraMS Cryo Team



Peter Barfknecht

Kelly Burch

Steven Cale

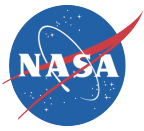
Brian Comber

Paul Rueger

Matt Francom

Richard Ottens

Hak Seung Lee



# References



1. Trainer, M. G., et al. "Development of the Dragonfly Mass Spectrometer (DraMS) for Titan." *52nd Lunar and Planetary Science Conference*. No. 2548. 2021.
2. Grubisic, Andrej, et al. "Laser desorption mass spectrometry at Saturn's moon Titan." *International Journal of Mass Spectrometry* 470 (2021): 116707.
3. Evonik Corporation. "Product Information Roahcell HF", <https://performance-foams.evonik.com/en/products-and-solutions/rohacell/attachment/140034?rev=794f98c8bd5fde83bf7a48c583636425>
4. Emkay Plastics. "Rohacell Foam Technical Product Manual", <https://www.emkayplastics.co.uk/wp-content/uploads/2019/10/Thermal-Properties.pdf>
5. Elliott Company. "ELFOAM P200: Polyisocyanurate Foam", <https://www.elliottfoam.com/pdf/ELFOAM%20P200%20TDS%20%20May2016.pdf>, 2016
6. MatWeb. "Overview of materials for Polyetheretherketone, Unreinforced", <https://www.matweb.com/search/DataSheet.aspx?MatGUID=2164cacabcde4391a596640d553b2ebe&ckck=1>
7. Rule, D. L., et al. "Low-Temperature Thermal Conductivity of Composites: Alumina Fiber/Epoxy and Alumina Fiber/PEEK", NIST, MISTIR 89-3914, 1989.