



Thermal Desktop Modeling of the RRM3 On-Orbit Cryogenic Methane Storage and Active Cooling Experiment

Erin Tesny, Daniel Hauser

NASA Glenn Research Center

Presented at The 30th Space Cryogenics Workshop

July 18, 2023

Outline



- Introduction & Background
- Thermal Desktop Model Set Up
- Previous CFD Modeling Results
- Current Zero G Modeling Results
- Conclusions & Future Work

Introduction & Background



- Cryogenic fluid storage & transfer systems are critical to future space missions in LEO and beyond
- Creating accurate models of these systems anchored to existing data is essential to developing predictive tools for future missions
- Several experiments have already been conducted on orbit to collect data on cryogenic fluid systems
- Robotic Refueling Mission – 3 (RRM3) microgravity experiment is one such dataset that can be used to create these models
 - Collected 4+ months worth of LEO cryogenic storage & transfer data

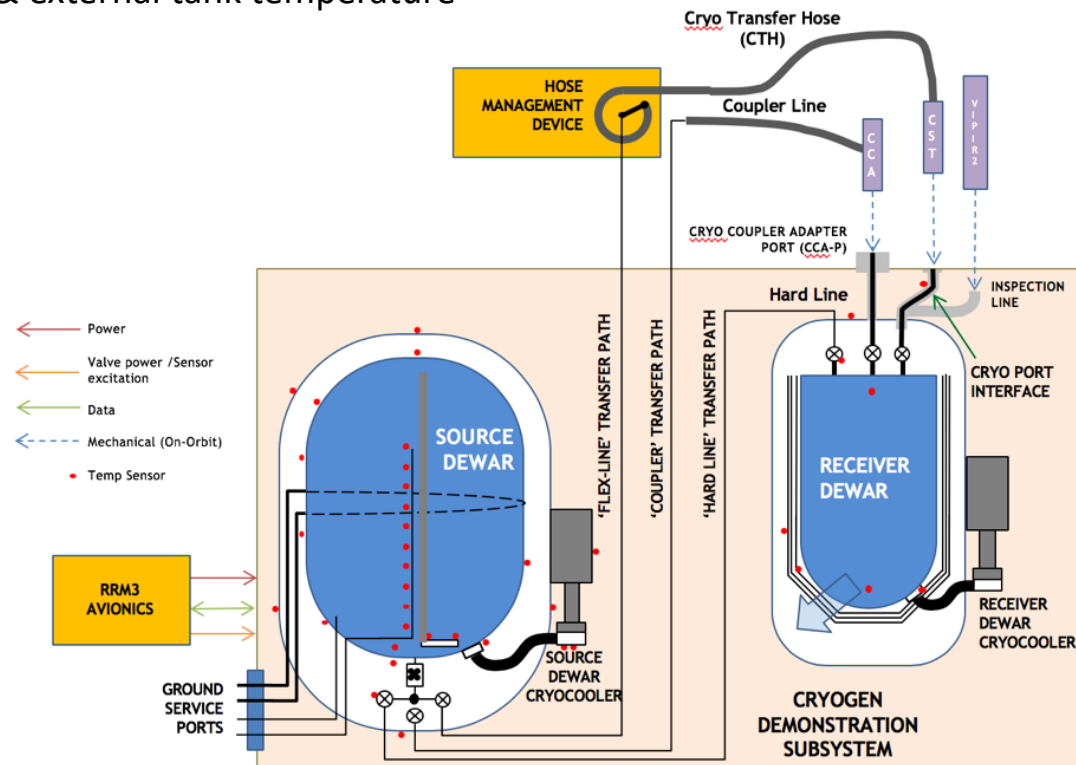


RRM3 Fuel Transfer Module on the ISS
from Breon et al (2020)

Introduction & Background Cont'd



- Current study: Self pressurization of the RRM3 source dewar
- 50-liter Aluminum 2219 tank
 - Fluid: Liquid Methane
 - Duration: 7 hour Cool to Reboost period
 - Thermal Desktop model created to measure:
 - Tank pressurization
 - Internal & external tank temperature

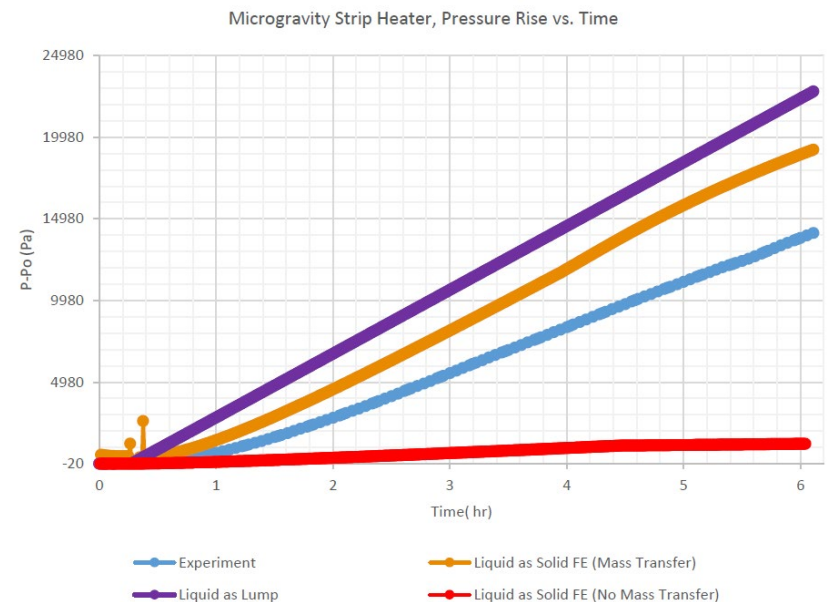
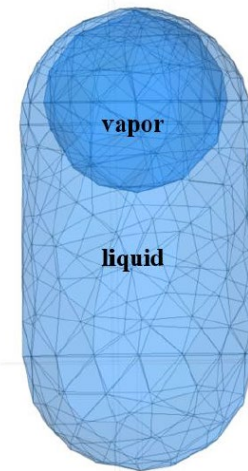


RRM3 Cryogen Demonstration System Block Diagram from Kassemi et al. (2022)

Previous 0g Self-Pressurization Model in Thermal Desktop



- A previous modeling study¹ focused on creating a Thermal Desktop model of the Zero Boil-Off Tank Experiment (ZBOT) conducted onboard the ISS in 2017
- Pressurization of tank containing two-phase fluid in microgravity
 - Conduction (not convection) through fluid is the primary heat transfer method in microgravity
 - Liquid is modeled as a solid finite element nodes in TD
 - Vapor modeled as a single lump
 - Mass transfer across liquid-vapor interface (LVI) governed by Schrage Equation
 - Pressure rise using this method matches data within 25%



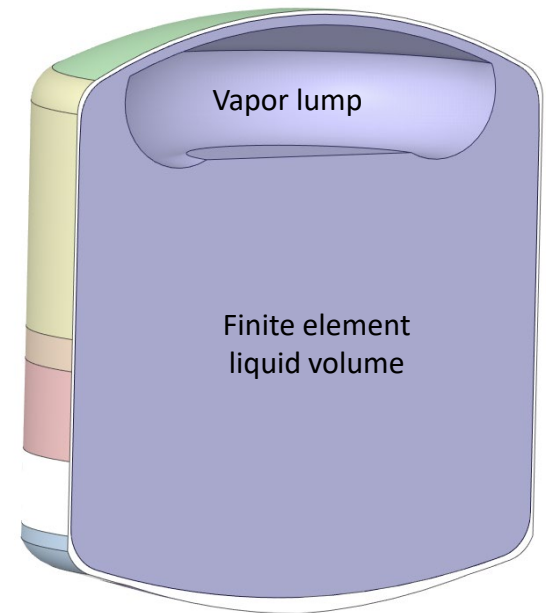
TD Model of ZBOT Fluid and resultant Pressure Rise in Tank from Tesny and Hauser (2019)

¹Tesny, E and Hauser, D (2019)

0g Thermal Desktop Model Setup



- Method used for ZBOT adapted for RMM3 Source Dewar
- Tank Wall simplified in SpaceClaim and imported into Thermal Desktop
- Tank Wall represented as a series of solid finite element nodes
 - All internal tank geometry removed from interior
- Liquid volume modeled as a finite element solid
- Ullage modeled as a single vapor lump
- Ullage shape and location approximated from CFD work that modeled the self-pressurization of the source dewar

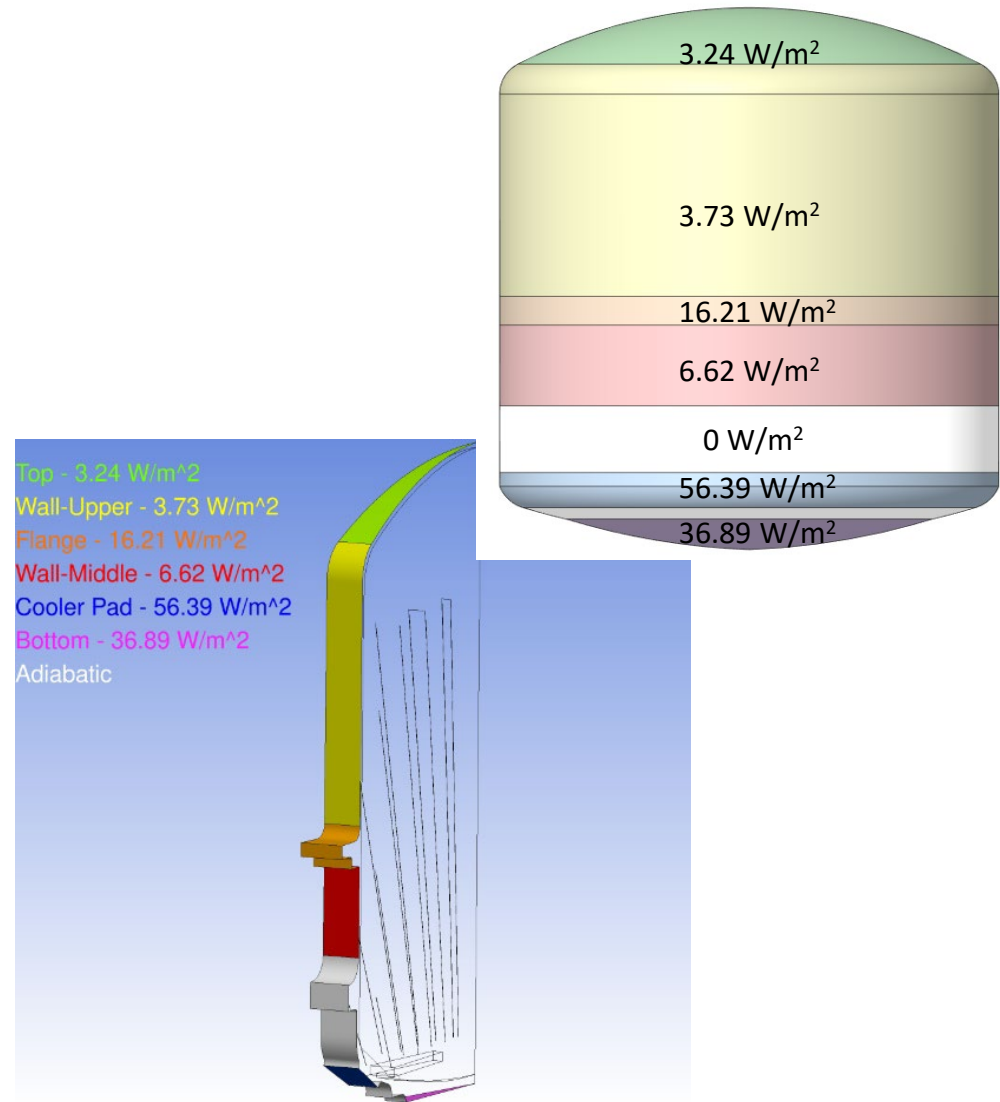


Source Dewar Thermal Desktop Model Cross-section

0g Thermal Desktop Model Setup



- Variable heat flux applied to outer tank wall to simulate heat loads on outside of source dewar
 - Finite elements split into different horizontal surfaces based on original tank geometry
- Conduction Coefficient of $10,000 \text{ W/m}^2/\text{K}$ between Tank Wall and Liquid Volume submodels
- Initial Temperature: 104.0 K
- Initial Pressure: 7.437 psia
- Test Duration: 7.0 hrs (Cool to Reboost period)



Heat Load Distribution from Kassemi et al. (2022)

0g Thermal Desktop Model Setup



- A similar method as in the previous ZBOT study was used to calculate heat and mass transfer across the Liquid-Vapor interface (LVI) Mass Transfer across LI:

– Schrage Equation

$$|\dot{m}| = \left(\frac{2\sigma}{2-\sigma} \right) \left(\frac{M_v}{2\pi R_u} \right)^{1/2} \left(\frac{P_i}{T_i^{1/2}} - \frac{P_v}{T_v^{1/2}} \right)$$

σ = accommodation coefficient

➤ Antoine Equation

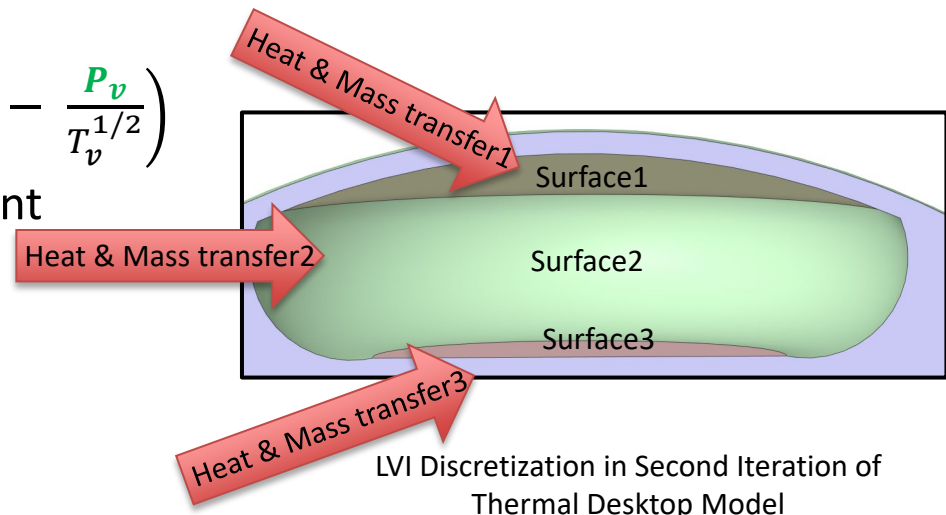
- $P = 10^{A + \frac{B}{T}}$

➤ Heat Transfer Across LVI

- $\dot{Q} = \dot{m}h_{vap}$

- Results presented for:

- Single Model: only 1 heat and mass transfer path across LVI surfaces
- Split Model: Heat and Mass transfer split up across 3 outer LVI surfaces (shown at left)

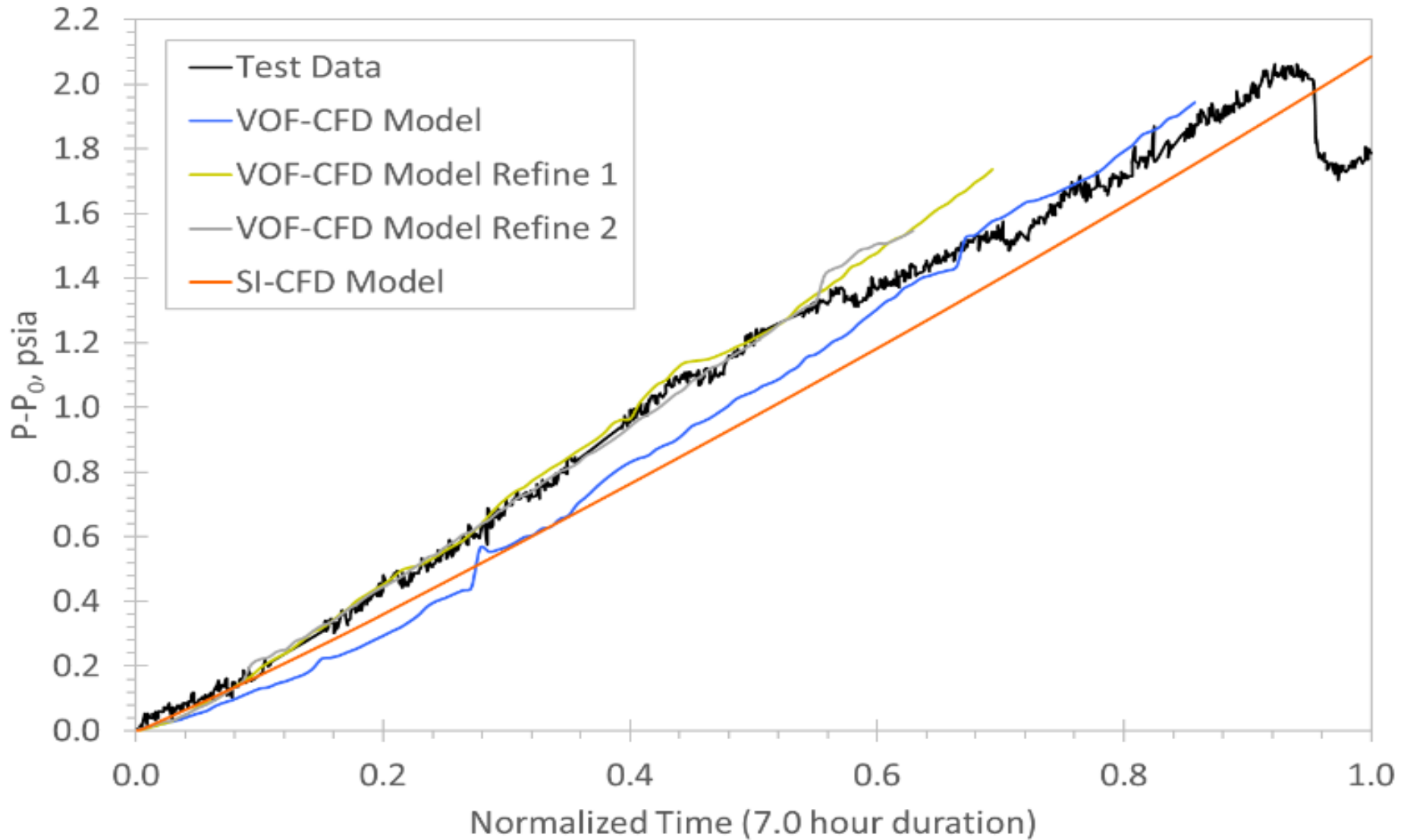


LVI Discretization in Second Iteration of Thermal Desktop Model

Previous CFD Modeling



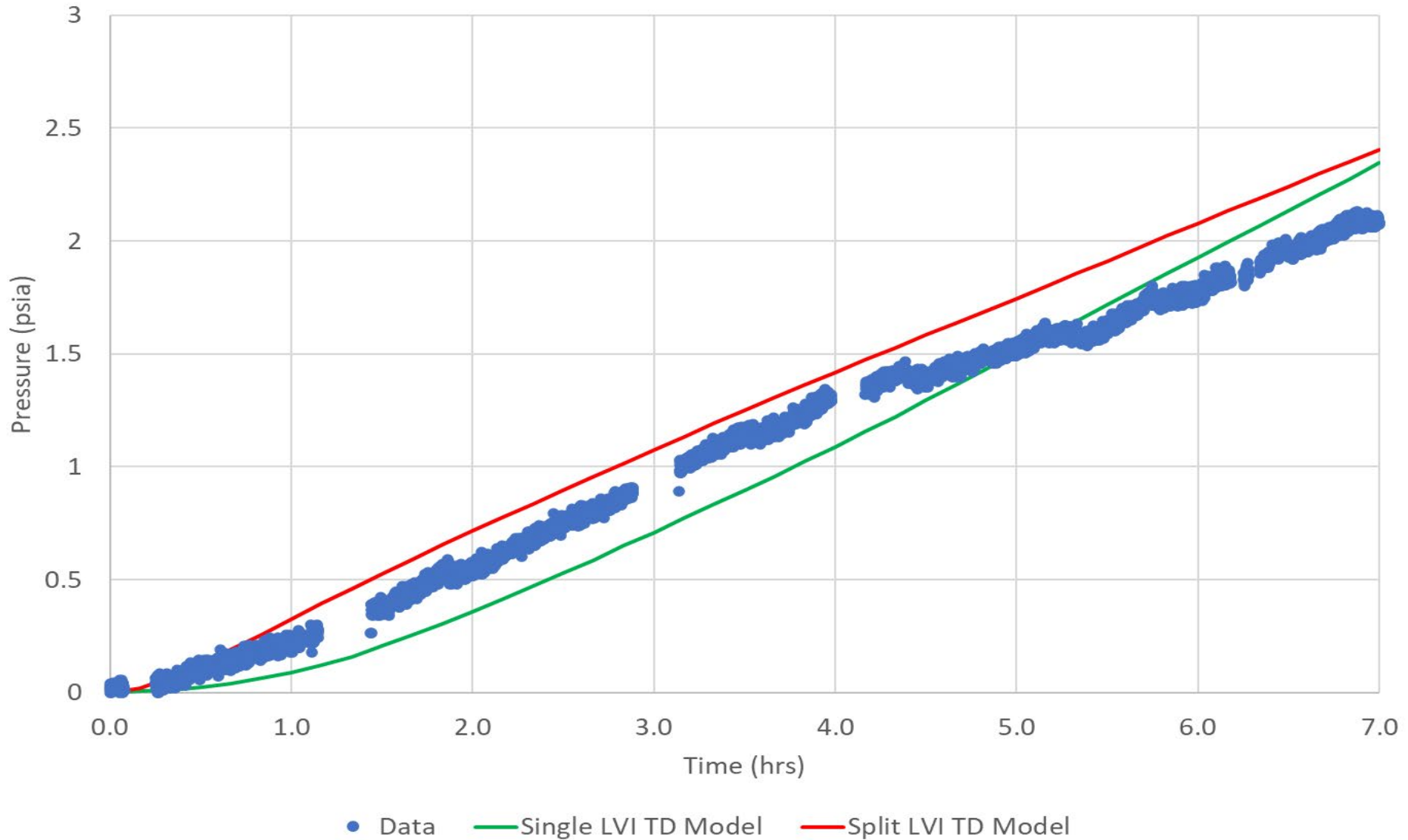
OG Pressure Predictions



Thermal Desktop Model



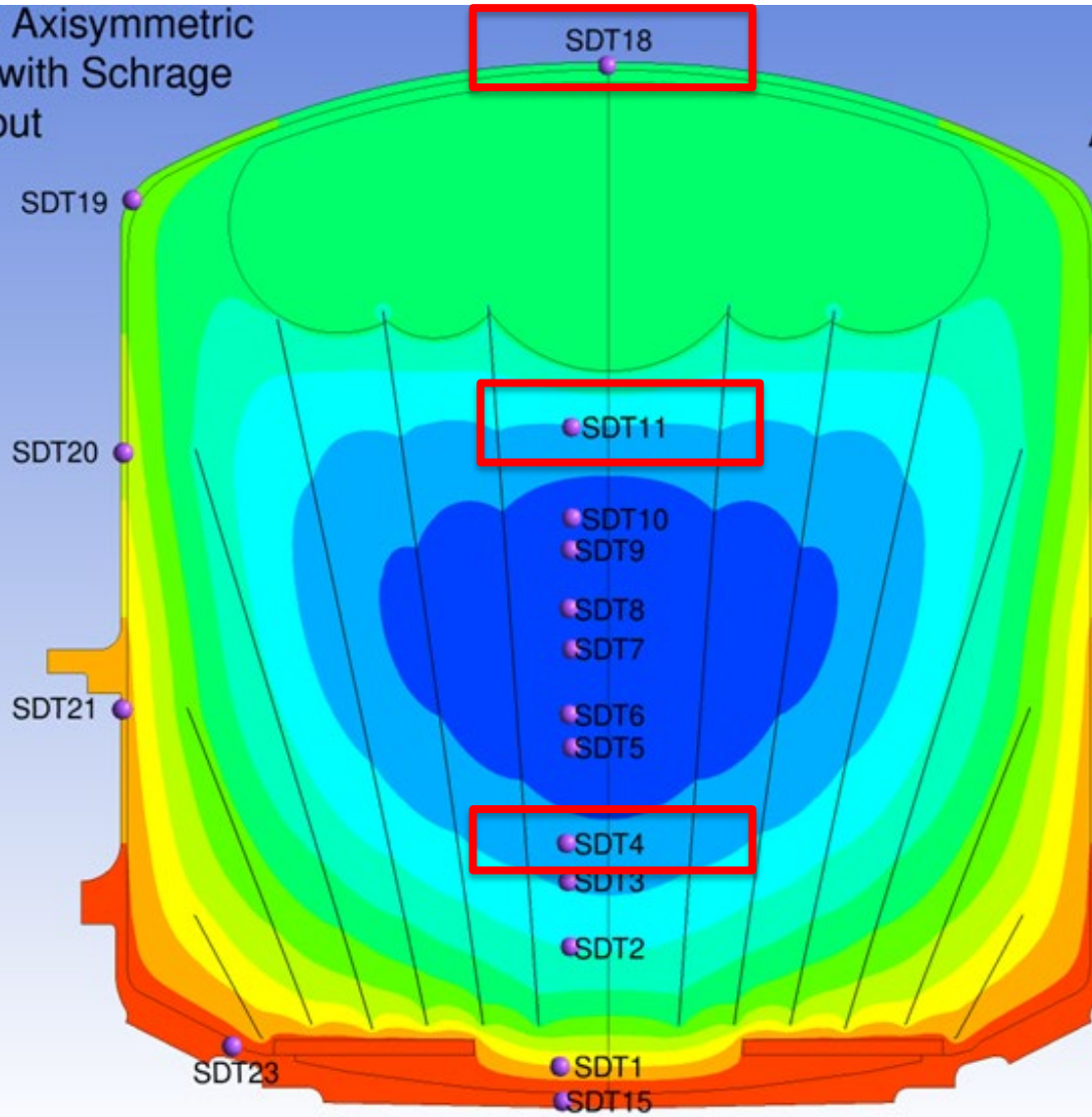
Thermal Desktop 0G Pressure Predictions



Previous 0g CFD Model Temperature Contours



RRM3 - 0G - 2D Axisymmetric
Sharp Interface with Schrage
8 Watts Heat Input
 $P_o = 7.437$ psia
 $T_o = 104.0$ K



ANSYS 2020 R1 ACADEMIC

Abs. Pres. = 9.521 [psi]

Sensor Temperatures:

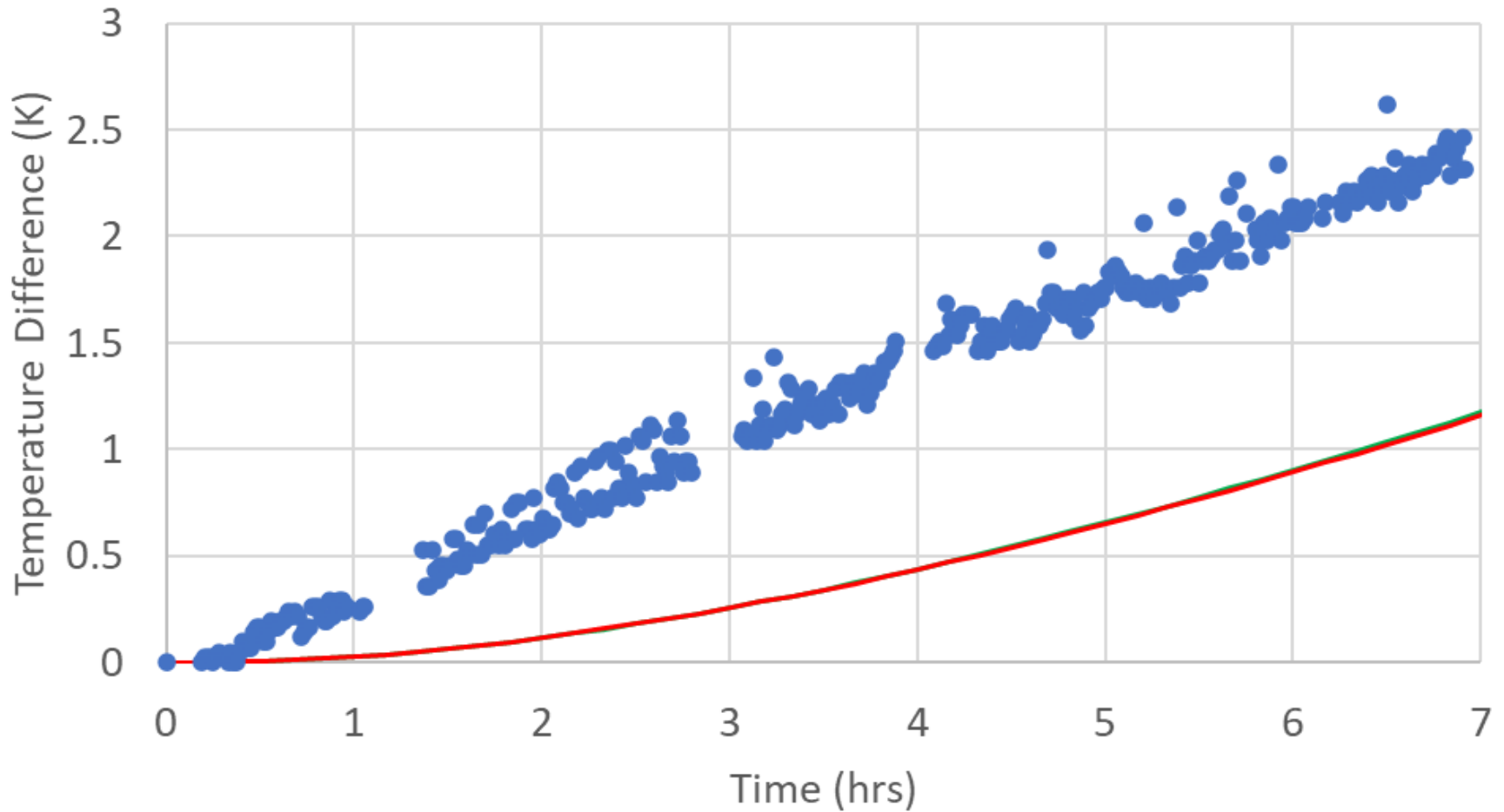
SDT1	= 108.398 [K]
SDT2	= 106.169 [K]
SDT3	= 105.692 [K]
SDT4	= 105.467 [K]
SDT5	= 105.056 [K]
SDT6	= 104.976 [K]
SDT7	= 104.892 [K]
SDT8	= 104.905 [K]
SDT9	= 105.005 [K]
SDT10	= 105.110 [K]
SDT11	= 105.717 [K]
SDT15	= 109.266 [K]
SDT18	= 106.921 [K]
SDT19	= 107.287 [K]
SDT20	= 107.804 [K]
SDT21	= 108.566 [K]
SDT23	= 109.082 [K]

Time = 420.0 min

Internal Fluid Temperature (SDT, Tank Bottom)



SDT4 Temperature Rise

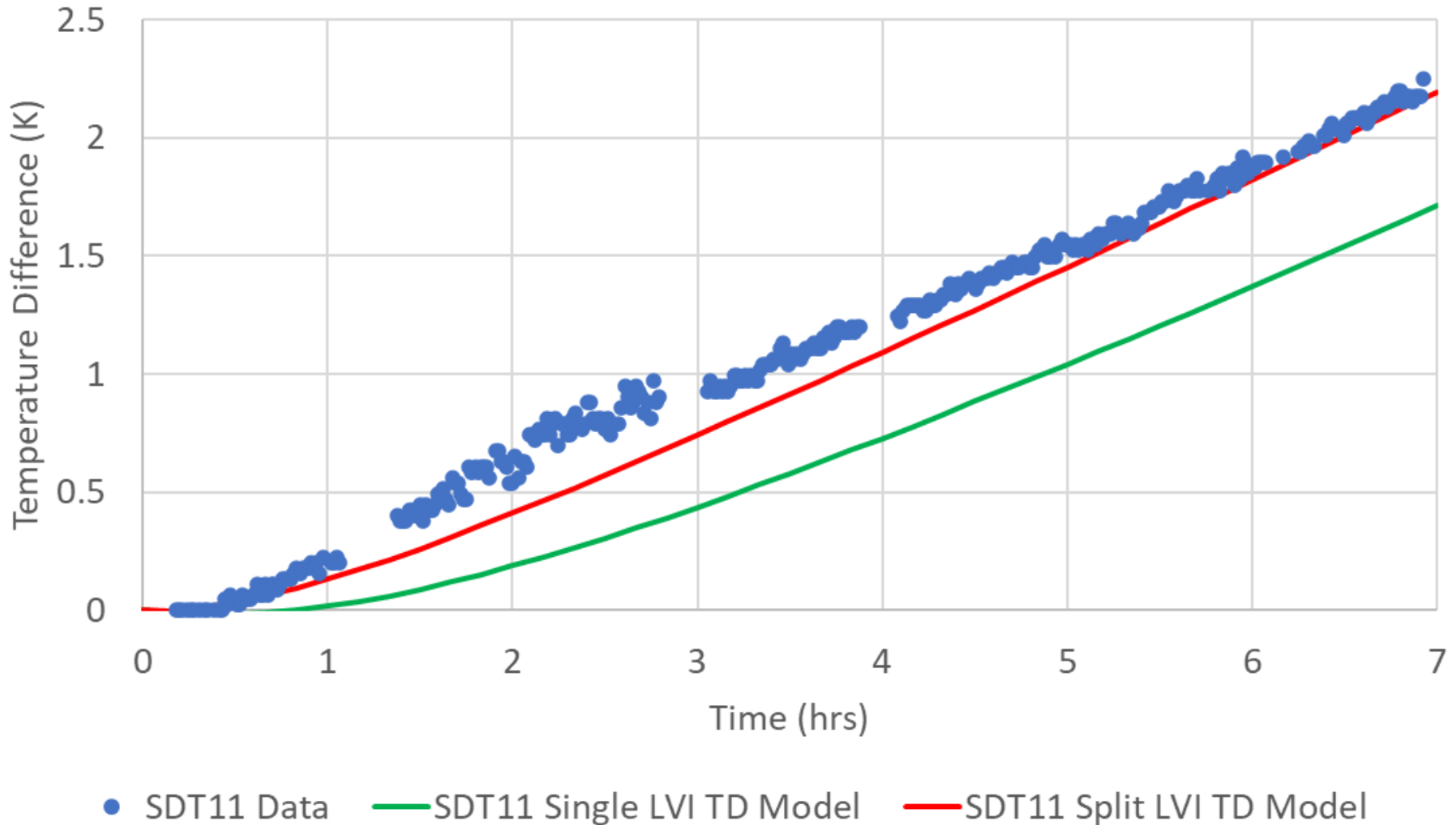


● SDT4 Data — SDT4 Single LVI TD Model — SDT4 Split LVI TD Model

Internal Fluid Temperature (SDT11, near LVI)



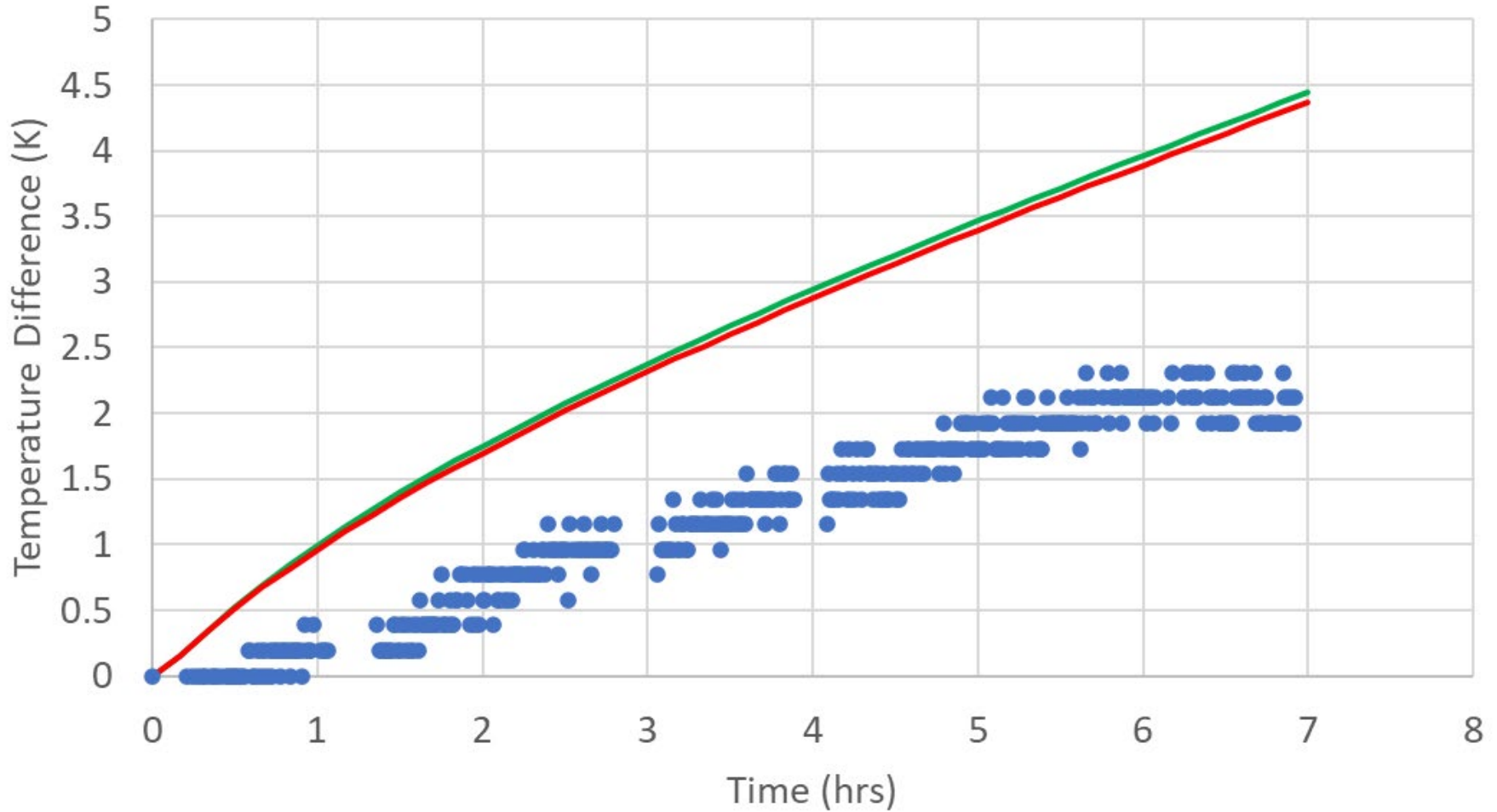
SDT11 Temperature Rise



External Tank Temperature

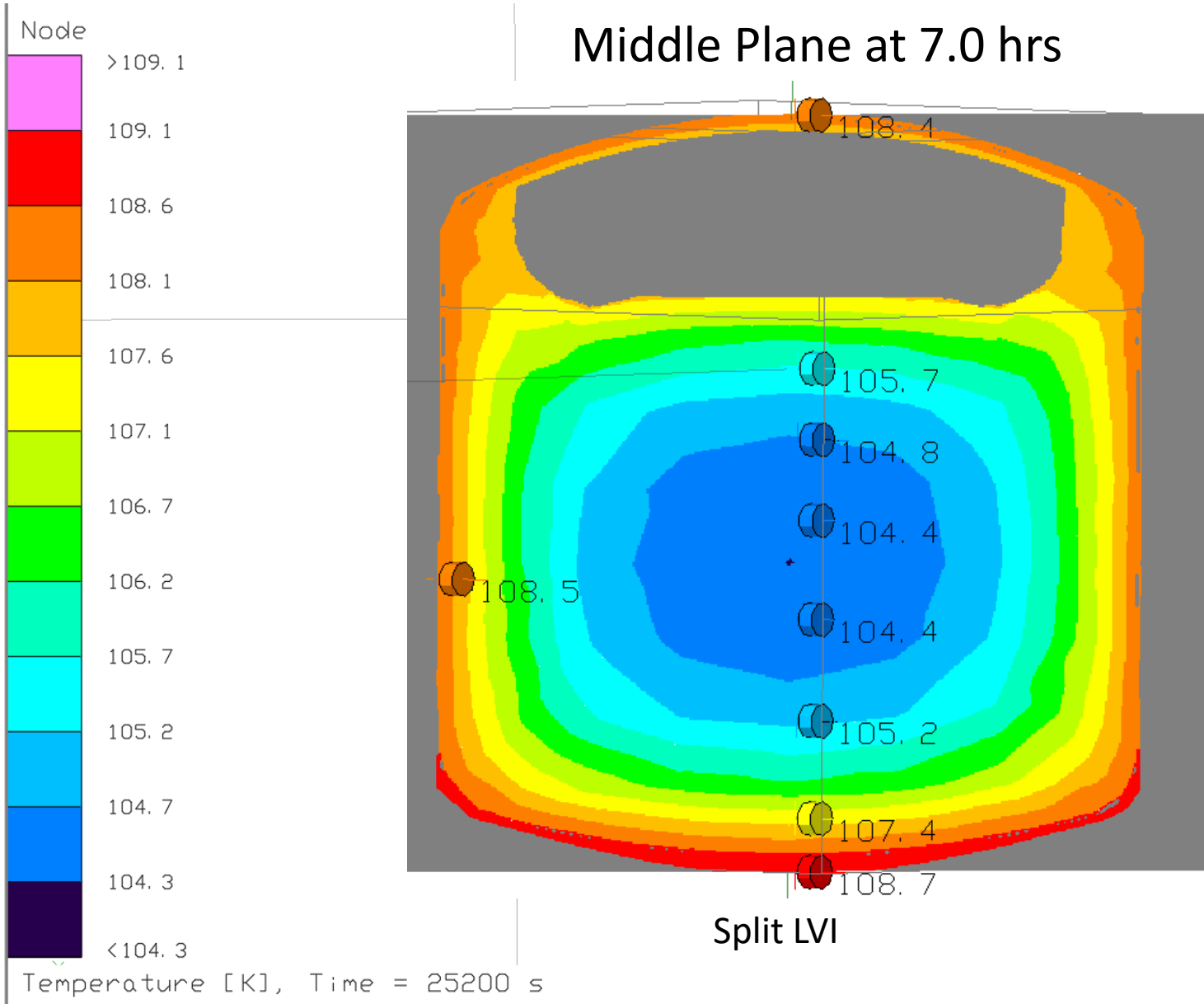


SDT18 Temperature Rise



● SDT18 Data — SDT18 Single LVI TD Model — SDT18 Split LVI TD Model

0g Thermal Desktop Model Temperature Contours



Conclusions



- Split LVI Thermal Desktop Model was able to more accurately predict the pressure rise inside the ullage and the liquid and wall temperatures over the Single LVI model
 - Split LVI model predicts pressure rise within ~15%, whereas Single LVI predicts within ~25%
- Split model also gives better agreement with data temperature rise close to interface due to refined heat & mass transfer prediction
 - Split LVI predicts temperature rise within ~15%, whereas Single LVI predicts within ~50%
 - Difference between models less apparent farther away from the interface
- Both models do similarly well predicting outside wall temperatures and overpredict the temperature rise by ~90%
- The Split LVI model can be used to inform design of future tank self-pressurization in microgravity



Questions?

References



1. Tesny, E and Hauser D. “Thermal Modeling of Zero Boil off Tank Experiment.” AIAA Propulsion and Energy 2019 Forum, 2019, <https://doi.org/10.2514/6.2019-4280>.
2. Kassemi M, White D, Hauser D. (2002) *CFD Analysis of the RRM3 On-Orbit Cryogenic Methane Storage and Active Cooling Experiment* [Paper presentation]. JANNAF In-Space Chemical Propulsion TIM, Huntsville, AL
3. Baylor, Michael. “Sn4 Becomes First Full-Scale Starship Prototype to Pass Cryogenic Proof Test.” NASASpaceFlight.Com, 27 Apr. 2020, www.nasaspaceflight.com/2020/04/starship-sn4-set-for-test/.
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5. Sempstrott, Danielle. “Kennedy Plays Critical Role in Liquid Hydrogen Tank Development.” NASA, 1 Nov. 2021, www.nasa.gov/feature/kennedy-plays-critical-role-in-large-scale-liquid-hydrogen-tank-development.
6. Tesny E and Hauser D. Validation of ZBOT Experiment Using SINDA/FLUINT Multi-node Analysis Tool. eCryo-RPT-0161. 2019