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Forum 116

• Interplanetary Terminal Procedure Design Applied to Lunar Landing

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23 OCTOBER 2023

Interplanetary Terminal Procedure Design Applied to Lunar Landing

iTERPS

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Background and Purpose



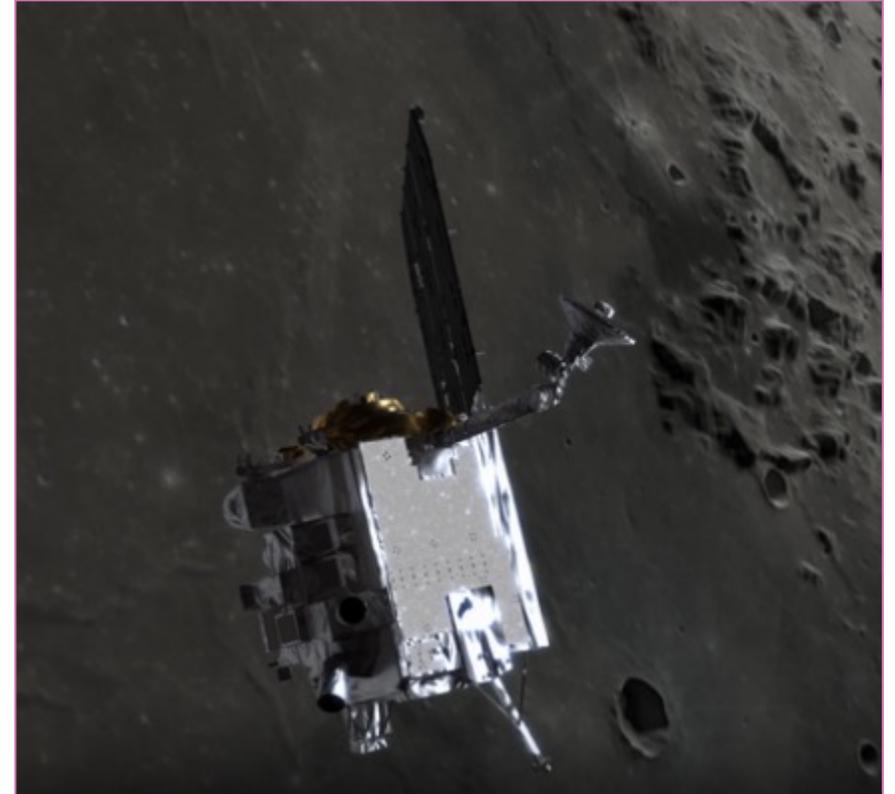
- Sea of Tranquility landing under Visual Flight Rules (VFR)
- Lunar south pole necessity for Instrument Flight Procedures (IFR)





iTERPS

A working theory to leverage current terminal procedure design (TERPS) criteria augmented with lunar physics in support of the exploration of the Moon, Mars and beyond





iTERPS Overview

- Lunar Procedure Innovation
- Entry Descent Approach and Landing
- Procedure Construction and Evaluation



Lunar Procedure Innovation

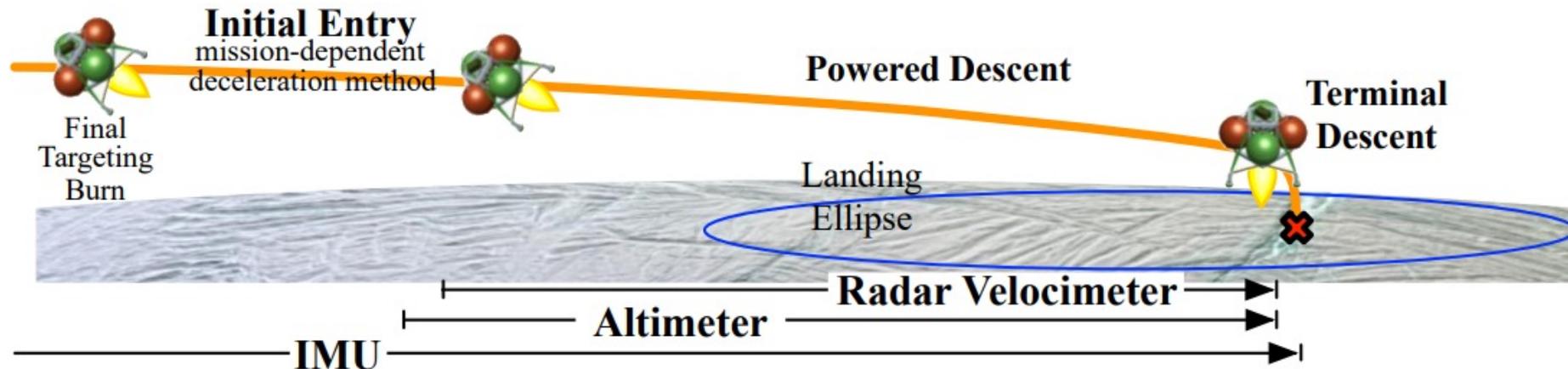
- Human factors-vetted approach / missed approach procedures
- Scalability of Point-in-Space (PinS) safety calculations
- Applicability to Advanced Air Mobility procedure development & training
- Innovative fixed altitude-variable entry PinS approach
- Dynamic approach plate with vehicle interoperability



Entry Descent Approach Landing (EDAL)

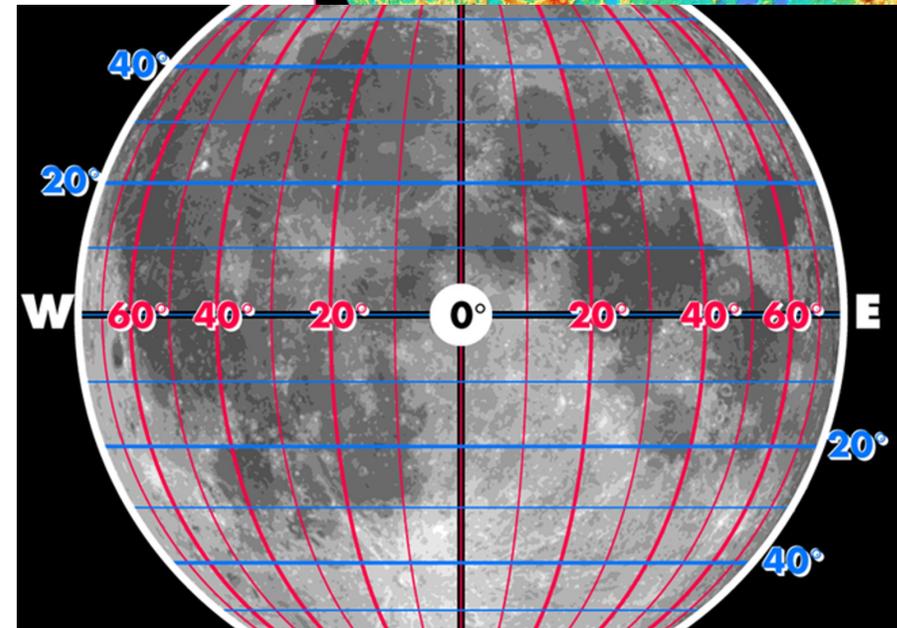
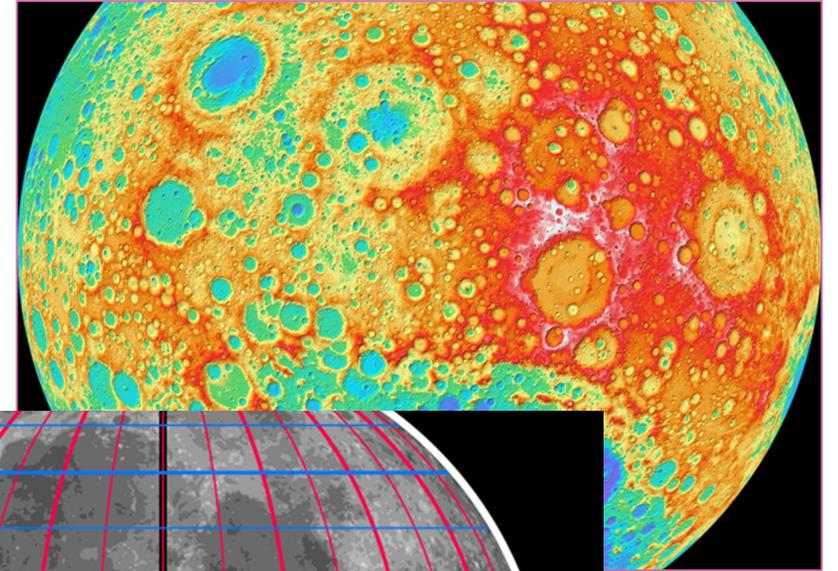
Entry Descent Landing (EDL) becomes Entry Descent *Approach* Landing (EDAL). Inclusion of approach procedures will provide innovative solutions to enable the safe execution of lunar landings:

- Construction
- Evaluation
- Validation
- Execution



EDAL Evaluation

- Lunar coordinates for waypoints via Selenographic coordinate system
- Satellite mapping
- Controlling obstacle identification and hazard mitigations





EDAL Construction

Landing Considerations

- **Obstructions-** Obstruction evaluation airspace analysis
- **Terrain-** Digital terrain elevation data
- **LiDAR-** Precision survey spatial data
- **Hazardous Conditions**

Vehicle Considerations

- **Flight Characteristics-** Flight controls with lunar physics
- **Energetics-** Dissipation of energy to arrest entry velocity & deceleration to zero-zero landing
- **Slope / Tilt Limitations**



iTERPS Design

- Lunar Unit Conversion
- Fixed Altitude Variable Entry Approach
- Dynamic PinS Approach Evaluation



iTERPS Lunar Unit Conversion

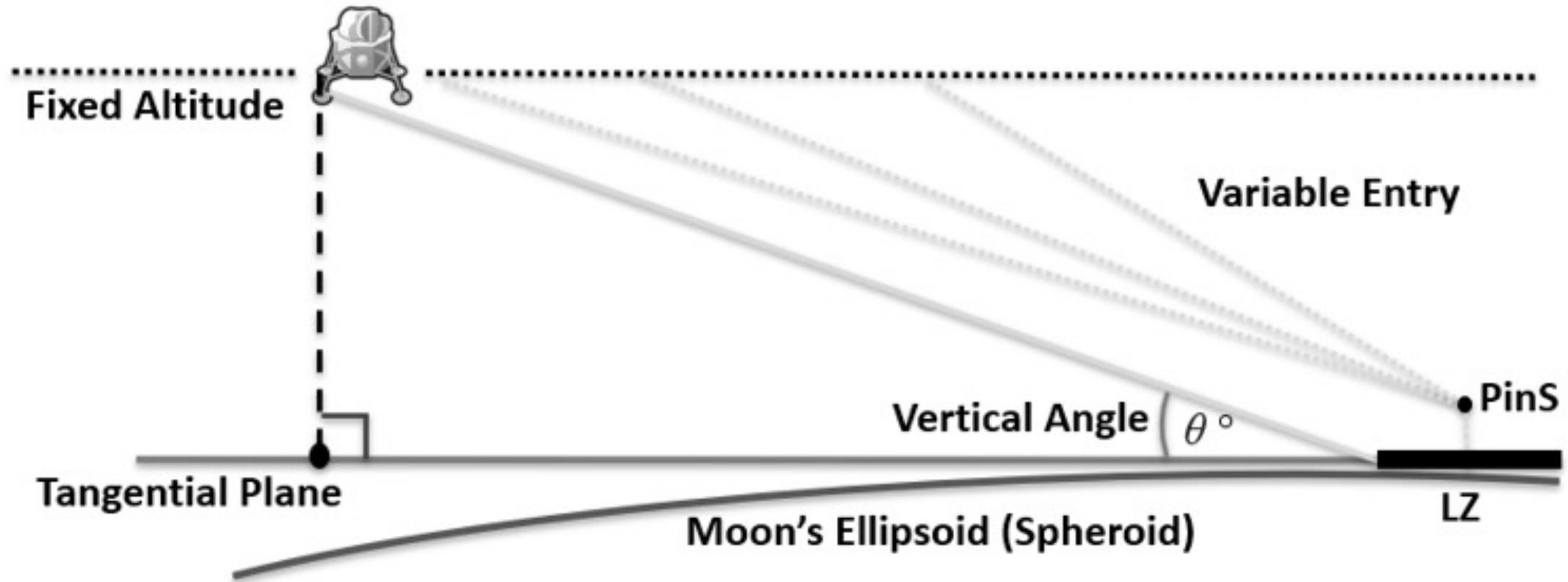
Apply existing PinS precision approach criteria with augmented lunar physics, radius and vehicle handling qualities/characteristics to generate a lunar approach plate and coding.

Formula	Earth	Moon
Distance Glidepath	<ul style="list-style-type: none">Barometric distanceEarth radius	<ul style="list-style-type: none">Vertical distance (ft or m)Moon radius
Turn Anticipation	<ul style="list-style-type: none">Heading change (degree)	<ul style="list-style-type: none">Vehicle limitation
Reaction and Roll	<ul style="list-style-type: none">Human + avionics (GNC)	<ul style="list-style-type: none">Autopilot + avionics + pilot monitor
Deceleration	<ul style="list-style-type: none">Earth gravity	<ul style="list-style-type: none">Moon gravity



EDAL Fixed Altitude - Variable Entry Approach Concept

Tailor **glideslope intercept** to any PinS zero-zero landing.

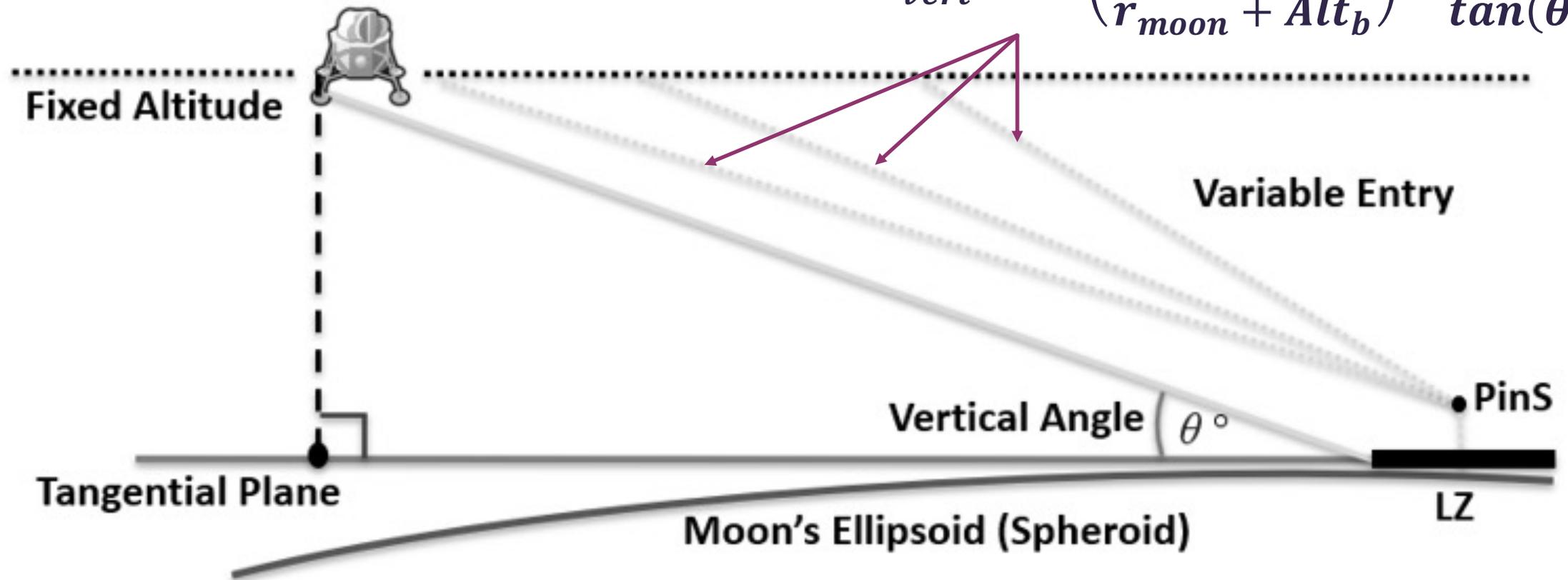




EDAL Fixed Altitude - Variable Entry Approach Concept

Vertical distance between two points on glidepath

$$d_{vert} = \ln \left(\frac{r_{moon} + Alt_e}{r_{moon} + Alt_b} \right) \times \frac{r_{moon}}{\tan(\theta)}$$

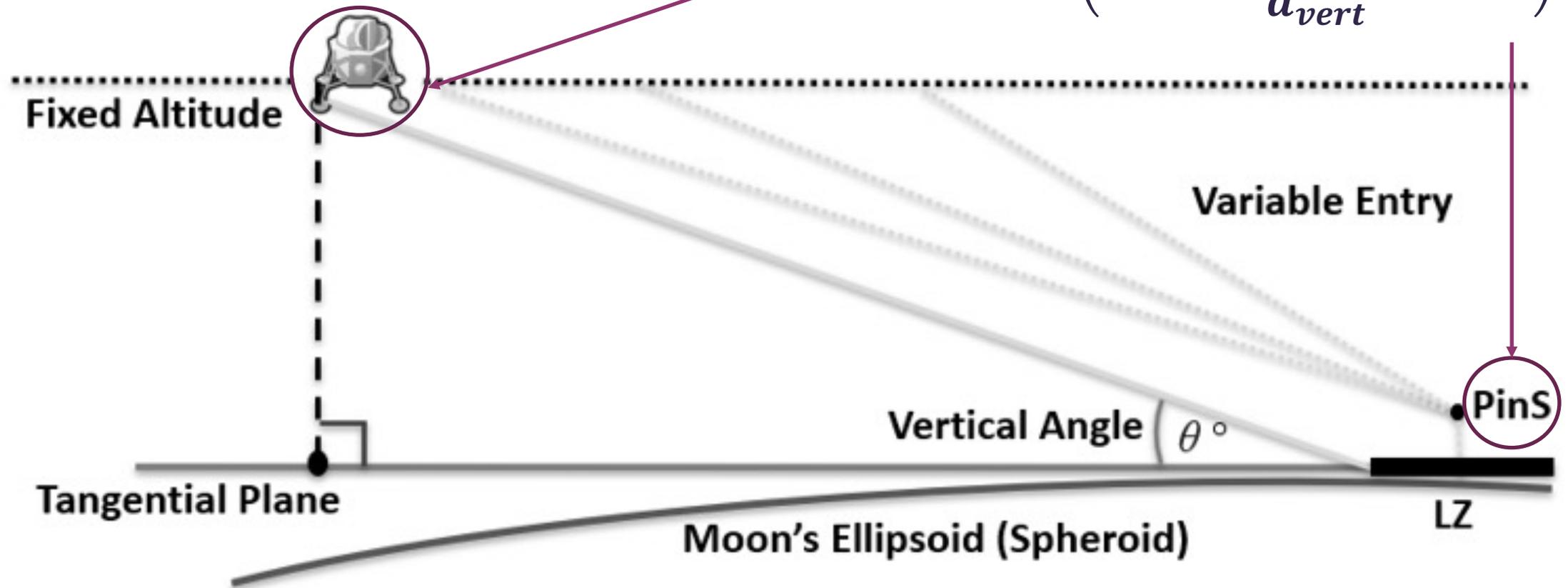




EDAL Fixed Altitude - Variable Entry Approach Concept

Descent angle capture

$$Capture_{\theta} = atan\left(\frac{Capture_{alt} - 300_{alt_e}}{d_{vert}}\right)$$

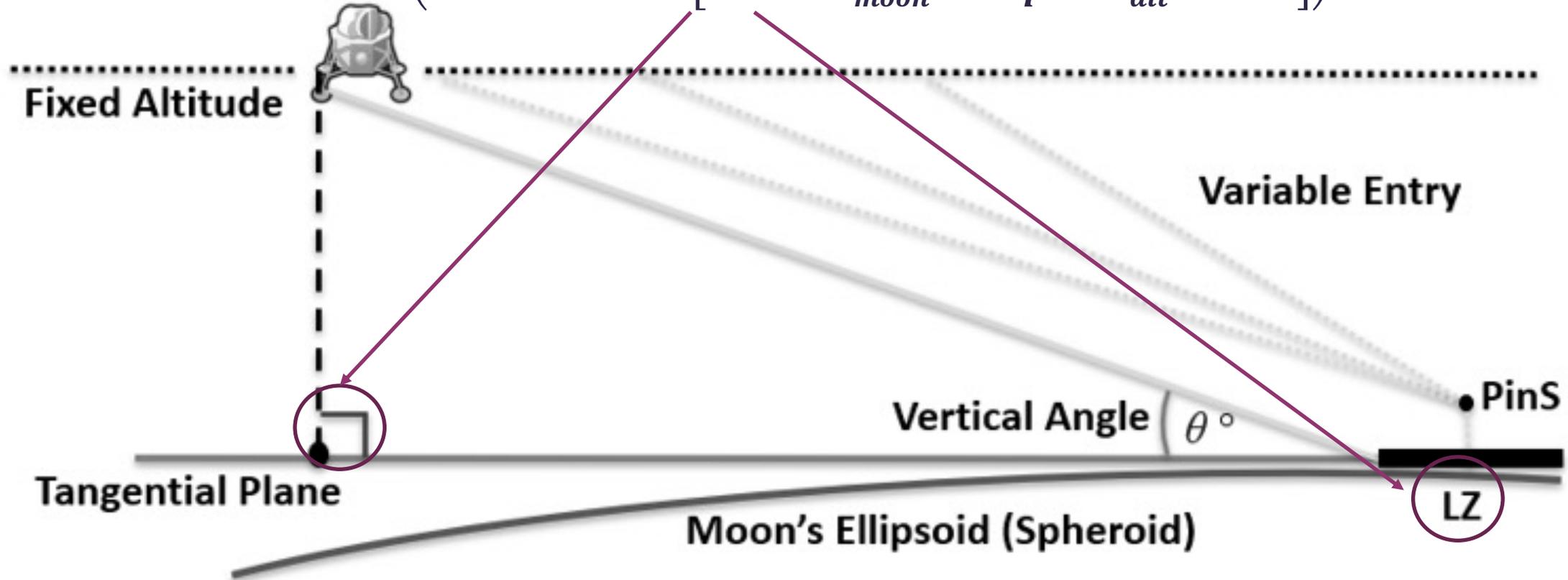




EDAL Fixed Altitude - Variable Entry Approach Concept

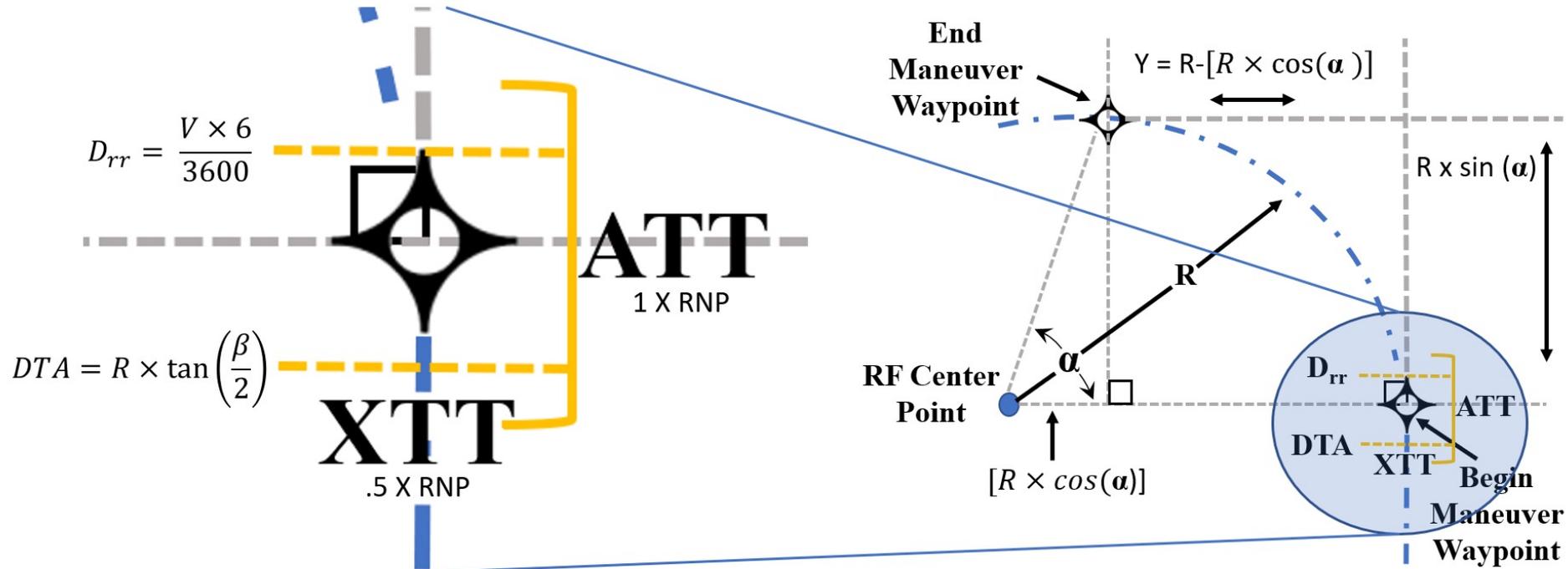
Minimum distance for descent angle capture

$$Capture_{Dmin} = \frac{\pi r_{moon}}{180} \left(90 - \theta - \sin^{-1} \left[\frac{\cos(\theta)(r_{moon} + LTP_{elev} + 300_{alt_e})}{r_{moon} + Capture_{alt}} \right] \right) - FAS_{Length}$$





EDAL Fixed Altitude - Variable Entry Concept

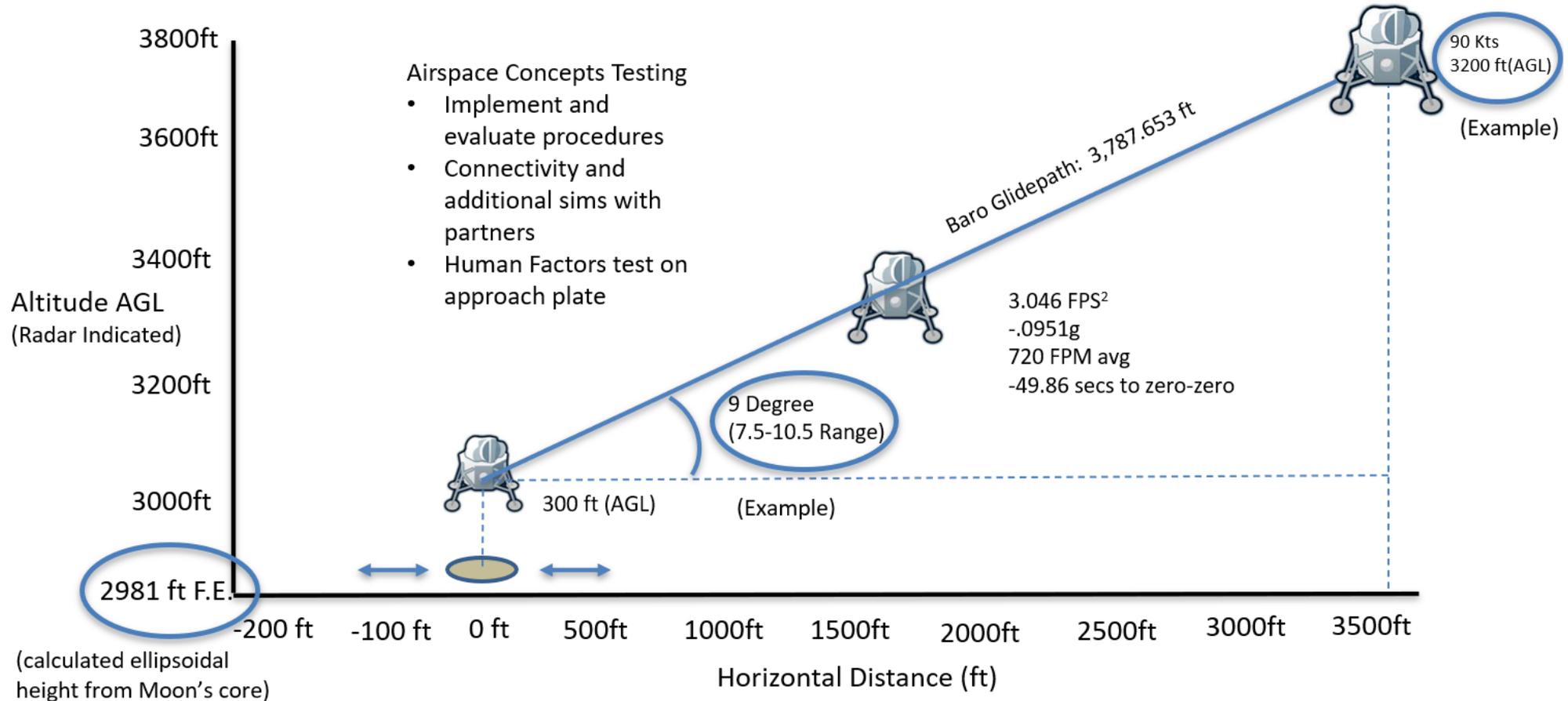


- β = magnitude of heading change in degrees
- D_{rr} = distance reaction and roll
- DTA = distance turn anticipation
- ATT = along track tolerance
- R = arc radius
- XTT = cross track tolerance



EDAL PinS Approach

Lunar Quad-Zero PinS Approach



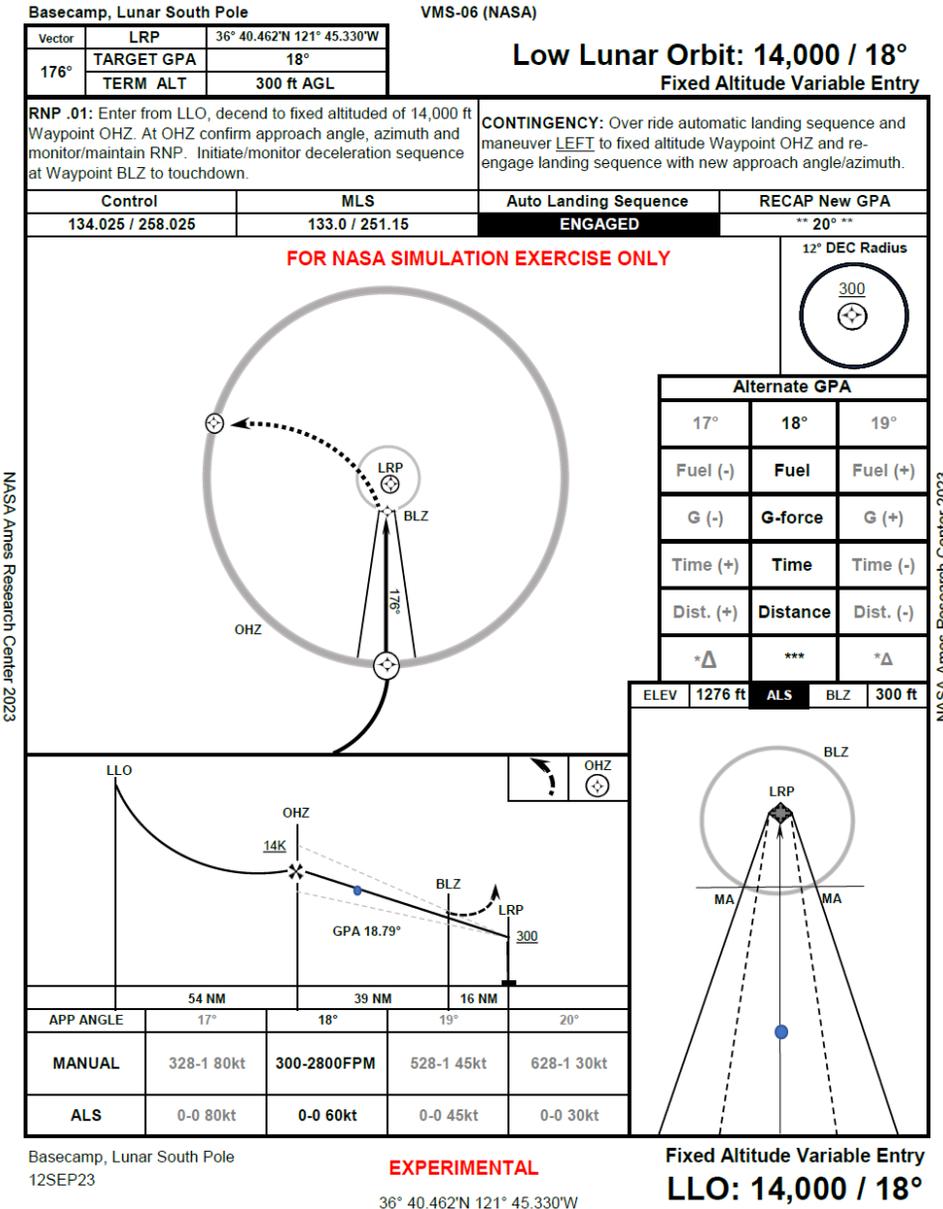


iTERPS Approach Plate

- Lunar Approach Plate Design
- Pilot Training

iTERPS Approach Plate

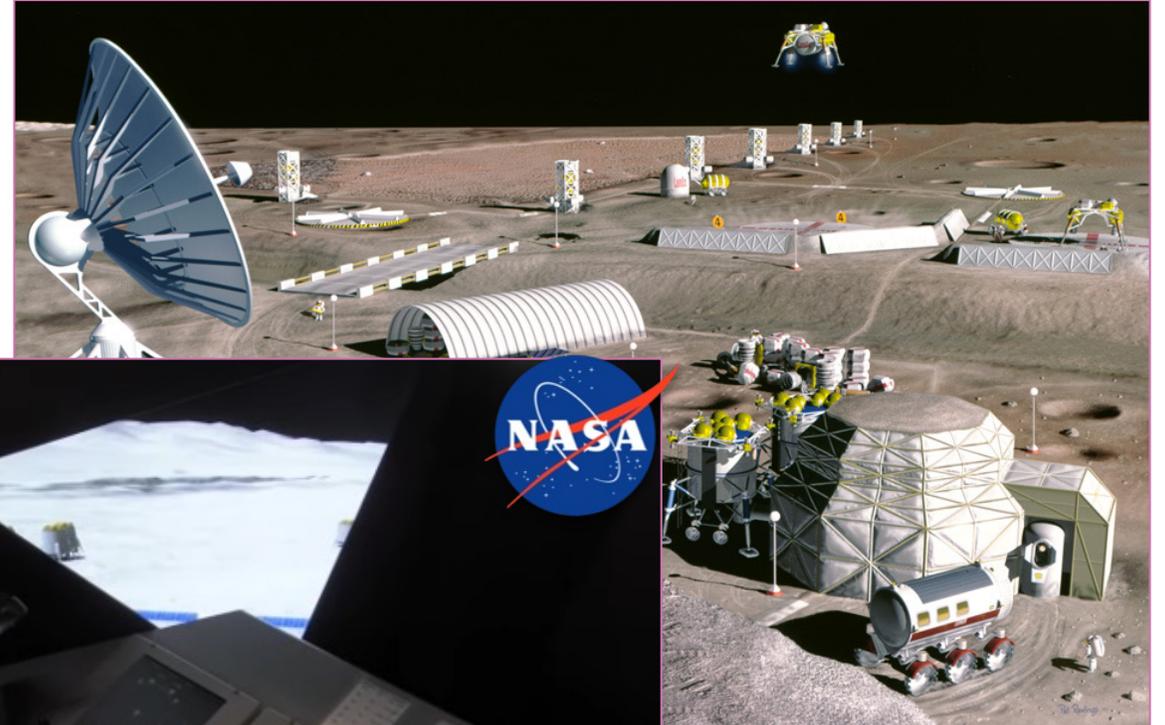
- Cross-monitor pilot/autopilot performance
- Human factors-vetted format
- Novel fixed altitude variable entry approach
- Contingency management through dynamically-generated approach



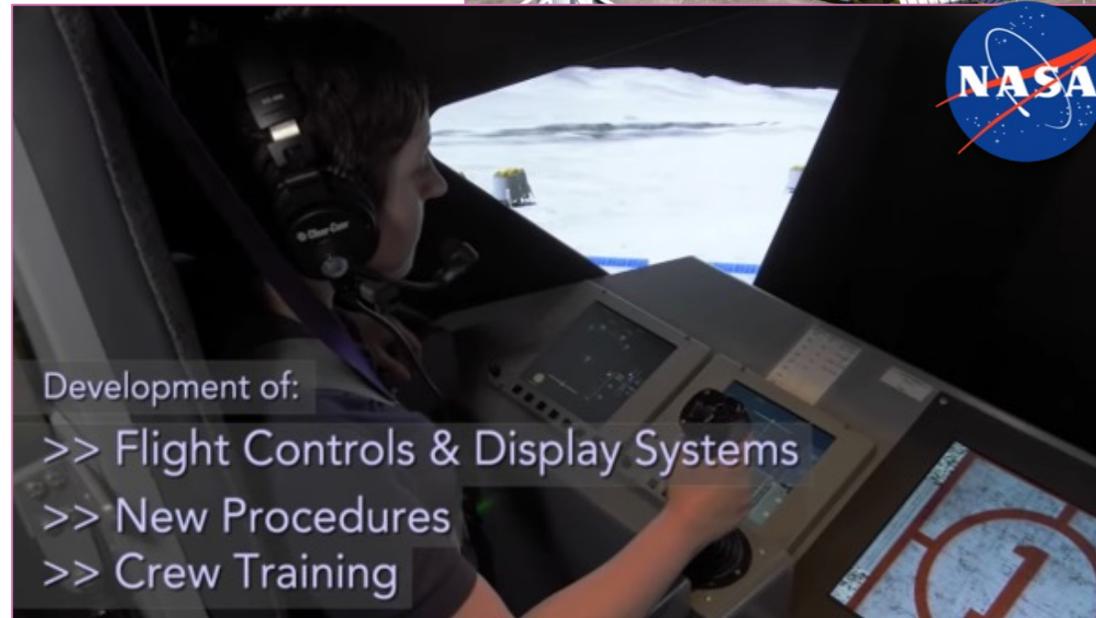


iTERPS Implementation

- Safety
- Standardized Training
- Scalability



Envisioned Lunar Base



NASA Ames Research Center Vertical Motion Simulator (VMS)



Conclusion

- Interplanetary Terminal Procedure Design
- Construction, Evaluation, Validation, and Execution
- Lunar Approach Plate
- Entry Descent “Approach” and Landing (EDAL)



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