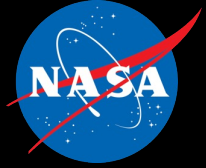




EXPLORE

Alison Park
Députy Tech Fellow – NASA Engineering and Safety Center
KITECH Mfg Tech Talk | Dec 4, 2023 | South Korea
CNU Student Mentoring Talk/ Dec 7 2023 | South Korea

Additive Manufacturing at NASA



Speaker Introduction

Alison Park – Deputy Tech Fellow at NASA



Works at NASA Chief Engineer's Office – NASA Engineering and Safety Center

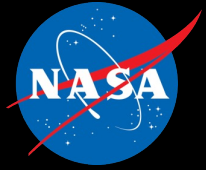
Responsible for safe implementation of Additively Manufactured Hardware for critical NASA missions – AM Qualification and Flight Certification Rationale (Higher TRL AM Techs)

Being involved in Additive Technology Innovation and Maturation activities around the agency (Lower TRL AM Techs)

B.S. Materials Science and Engineering, Purdue University

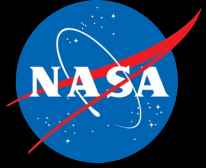
M.S. Aerospace Materials Engineering, UCLA

24 years of experiences working for both commercial (Rocket Propulsion) and Government (NASA) - 14 Years in Additive Manufacturing



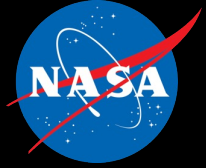
Executive Summary of Today's Presentation

Additive Manufacturing (AM) is certainly changing the space industry and providing new opportunities to travel to low earth orbit and explore our universe. New design opportunities – not previously possible – for new high performance metal alloys, light-weighting, managing thermal, structural, and dynamic loads are being enabled by AM. This presentation will showcase the vast portfolio of NASA's AM activities in the last 12 years; transportation from Earth to Destination, Habitat at Destination, Lander from Station to Surface, and Science mission spacecrafts. NASA's technical excellence is being leveraged heavily in the AM community thru collaborative projects, partnership agreements, tech transfer program. Challenges as well as opportunities will be discussed.



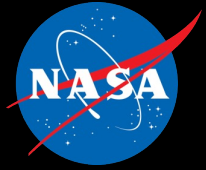
Today's Topics

- *Why* and *When* to use AM: Alison's perspectives
- AM at NASA and around the US Space Community
- Special Topics: How AM is being used/should be used
 - AM-enabled Alloy Development (GrCop Alloys, GRX-810)
 - AM Qualification and Certification
 - Classic Process Control-based approach (Current)
 - Digital Twin-based approach (Future)



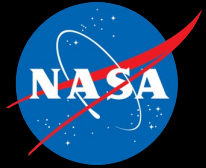
Alison's Perspectives on Additive Manufacturing

- Yes, It's a cool technology but use it with purpose
- Embrace the benefits and appreciate the challenges
- Do your homework before you try to implement on critical missions
- Exciting opportunities in adjacent technologies (Integrated Computational Materials Engineering, Digital Twin, Modeling and Simulation, in-situ Monitoring Sensors, Advanced Non-Destructive Evaluation, Design AI...) exist now and will continue to be advanced



When to Use (Metal) Additive Manufacturing

- ~~• You have a fully designed part~~
 - ~~• You need it to be good~~
 - ~~• You need it to be cheap~~
 - ~~• You need it quickly~~
 - You need to prototype and/or iterate a *lot*
 - You need an *extremely* optimized part (i.e., topology optimization)
 - You can't easily make the part using legacy "subtractive manufacturing"
 - You need a part with a high "buy to fly" ratio
 - You literally *can't* make it any other way
 - You want to decrease part count
 - Novel Alloys not feasible with traditional manufacturing
 - Sustainability/Local Manufacturing
- Recipe for Disappointment**



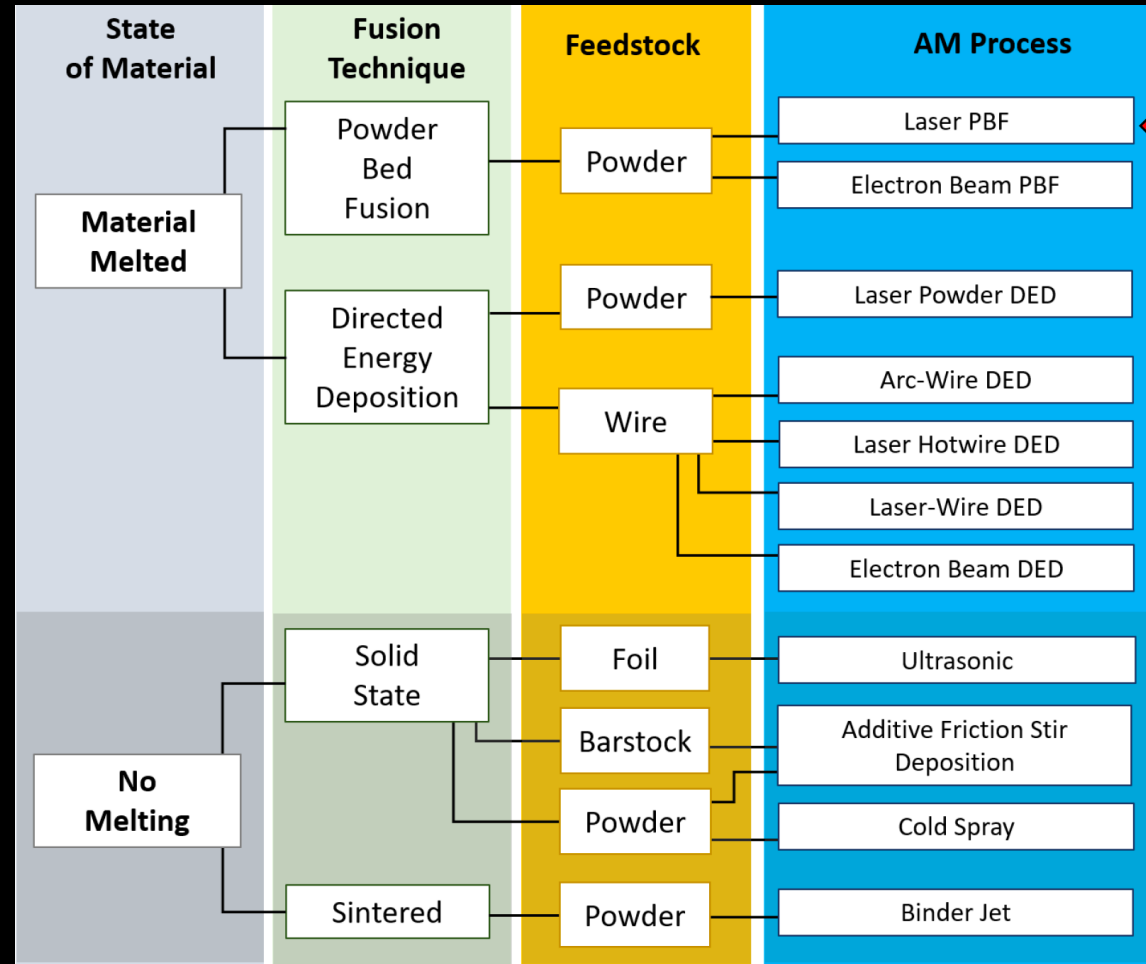
Additive Manufacturing (AM) – Which one??

Definition:

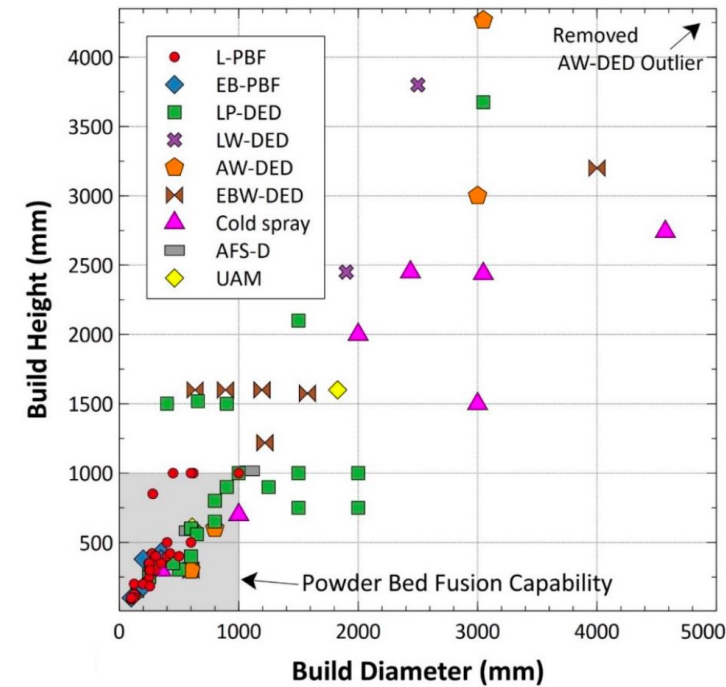
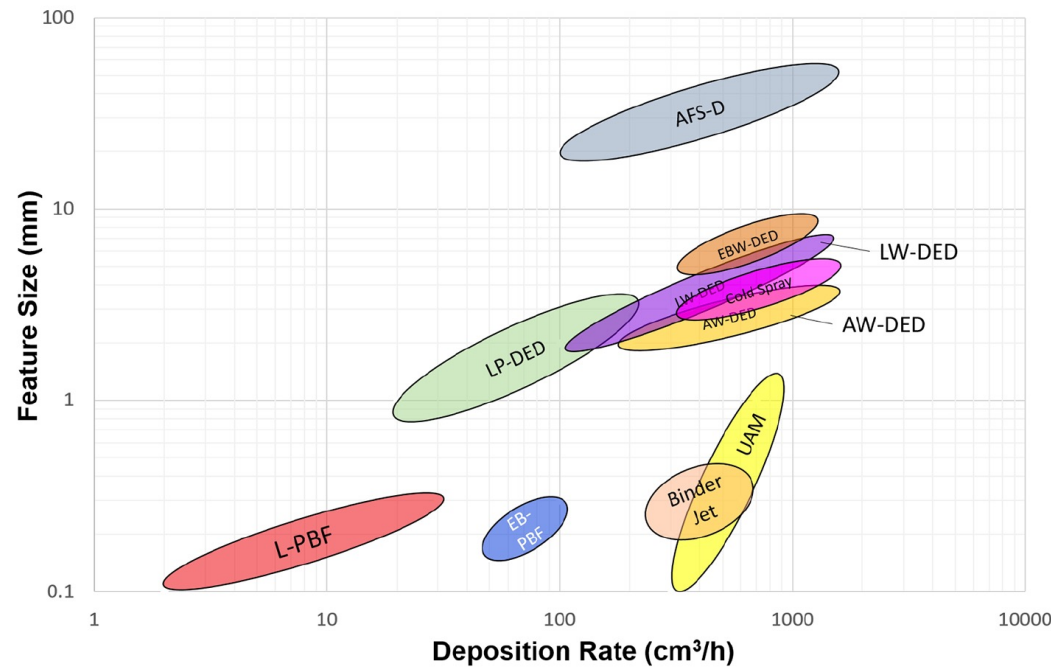
Process of joining materials to make parts from 3d model data, usually layer upon layer

AM processes =

Material feedstock + Energy Source



Things to consider to pick the right AM Process



Complexity of Features

Scale of Hardware

Material Physics

Cost

Material Efficiency

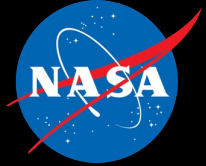
Speed of Process

Material Properties

Internal Geometry

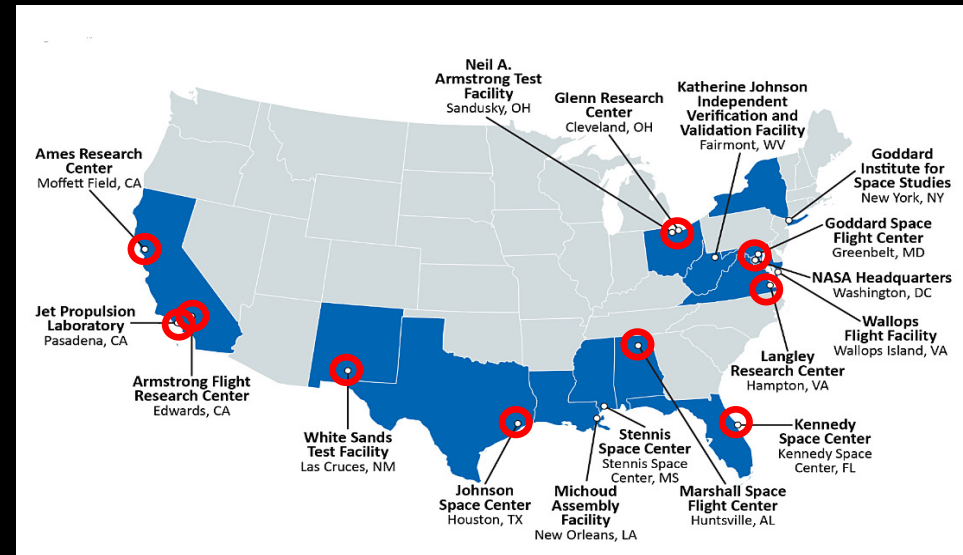
Availability

Post Processing



NASA is NOT monolithic

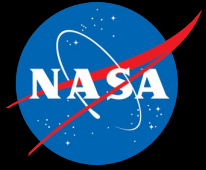
- Geographically different
- Flight Center vs. Research Center
- Mission Directorate Focused
- Each Center is supposed to be “mission-agnostic” but heritage center culture still exists and gets reflected in project execution and practices



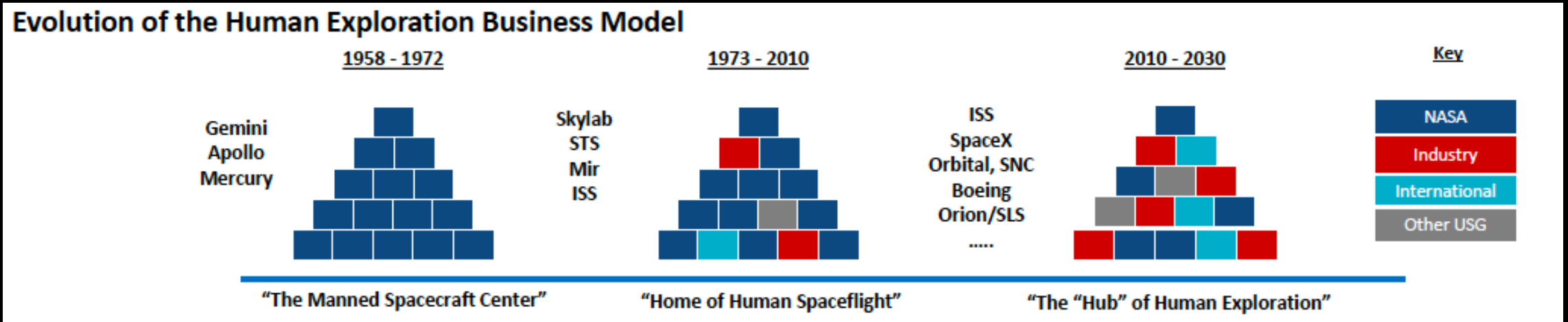
NASA Mission Directorates



Aeronautics	Space Technology	Science	Exploration Systems Development	Space Operations
NASA explores technologies that reduce aircraft noise and fuel use, get you gate-to-gate safely and on time, and transform aviation into an economic engine at all altitudes.	NASA technologies developed for spaceflight benefit our everyday life. The Artemis program proves and matures what those technologies can do and reduces risk for exploration of Mars and beyond.	NASA and the nation's science community use space observatories conduct scientific studies of the Earth from space to visit and return samples from other bodies in the solar system, and to peer out into our galaxy and beyond.	NASA's Artemis program is defining and creating the steps path from Earth back to the Moon and on to Mars, including the Orion capsule, the Space Launch System, Exploration Ground Systems, the Gateway, and Human Landing System.	NASA's work in beyond low-Earth orbit includes commercial launch services to the International Space Station, exploration systems, space transportation systems, and broad scientific research on orbit.

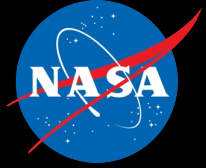


Space Business Model has been changing



Courtesy of Walt Ugalde (NASA JSC)

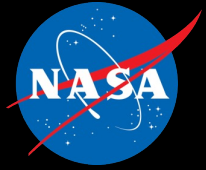
- *With NASA's newer procurement model (NASA being a buyer), NASA's need for production the flight hardware is shrinking*
- *But we deeply care about the underlying technologies that would advance us to beyond LEO exploration*
- ***Which technologies does NASA want to take a lead to own and further develop - AM***
- *Which technologies does NASA want Commercial Space market to lead*



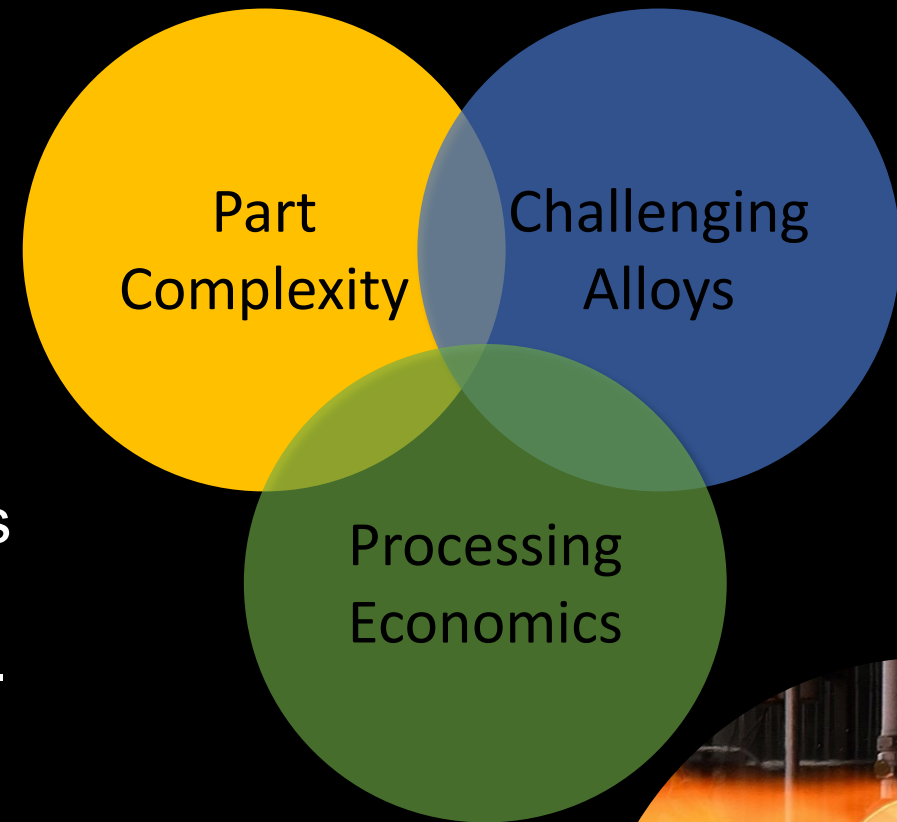
AM at NASA – past, present, and future

- Past (2010 ~ 2020)
 - AM Tech Development - Focused on the understanding AM processes through material characterization and testing, standards development, component design iteration, and **infusion into rocket propulsion development**
 - Having insights and oversights on other people’s AM design and hardware (Commercial Space companies)
- Present (2020~now)
 - AM-enabled Alloy Development (Refractory, Oxide Dispersion Strengthened, Cu-Cr-Nb)
 - AM Tech Development – Large scale for rocket nozzles, Multi-metallic and multi-AM process
 - Outreach – Technical Journal articles, Conference presentations, Webinars, AM 101, AIAA Book
 - Partnership with AM and Commercial Space Community
 - Certifying and Flying Commercial Space AM hardware on NASA missions
- *Future*
 - *NASA being the prolific user of AM (NASA being a maker for certain missions) leveraging Generative Design AI to evolve optimal design*
 - *NASA walking the talk*
 - *NASA Centers having its own AM User Group and integrated with each other*
 - *Digital Twin based AM Certification approach*
 - *AM enabling manufacturing ”for space, in space, and in-situ”*

The Case for AM in Rocket Propulsion

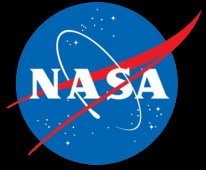


- Metallic AM can provide significant advantages for lead time and cost over traditional manufacturing for rocket engines.
 - Lead times reduced by 2-10x
 - Cost reduced by more than 50%
- Complexity is inherent in liquid rocket engines and AM provides new designs, part consolidation, and performance opportunities.
- Materials that are difficult to process using traditional techniques, long-lead, or not previously possible are now accessible using metal additive manufacturing.



Excerpt from "Metal Additive Manufacturing for Propulsion Applications"
AIAA Progress in Astronautics and Aeronautics Book Serie
P. R. Gradl, O. Mireles, C.S. Protz, C. Garcia. (2022).

NASA's earlier work – case study for AM Rocket Combustion Chamber



Category	Traditional Manufacturing	Initial AM Development	Evolving AM Development
Design and Manufacturing Approach	Multiple forgings, machining, slotting, and joining operations to complete a final multi-alloy chamber assembly	Four-piece assembly using multiple AM processes; limited by AM machine size. Two-piece L-PBF GRCo-84 liner and EBW-DED Inconel 625 jacket	Three-piece assembly with AM machine size restrictions reduced and industrialized. Multi-alloy processing; one-piece L-PBF GRCo-42 liner and Inconel 625 LP-DED jacket
Schedule (Reduction)	18 months	8 months (56%)	5 months (72%)
Cost (Reduction)	\$310,000	\$200,000 (35%)	\$125,000 (60%)

Examples of NASA's AM Dev Work for Rocket Propulsion



Main Fuel Valve

Part reduction: 5 to 1
Successfully tested



Fuel Turbopump

Schedule reduction by 45%
Part reduction: 40 to 22
Tested to 90,000 RPM



Combustion Chamber

Schedule Reduction: > 50%
Bimetallic L-PBF/DED
Tested to 100%

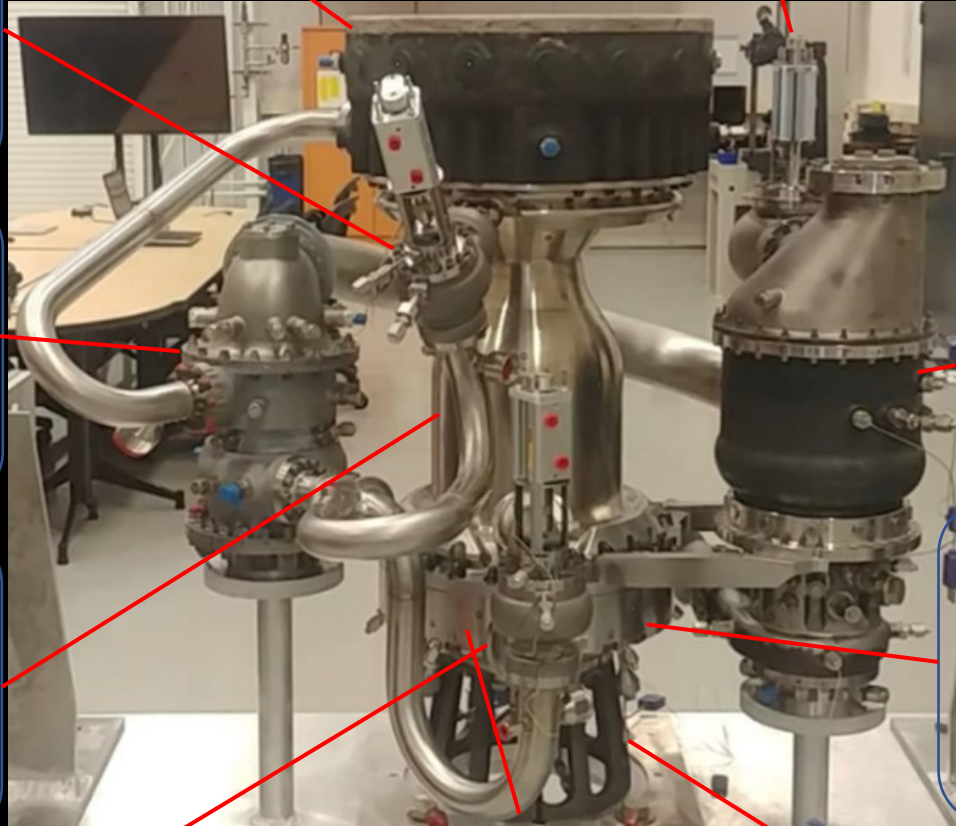


Regen Nozzle

Oxidizer Turbopump

Bypass Valve

Part reduction: 5 to 1



Main Oxidizer Valve

Part reduction: 6 to 1
Successfully tested



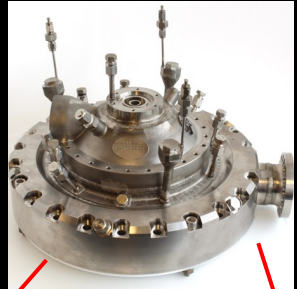
Oxidizer Turbopump

Part reduction: 80 to 41
Tested to 40,000 RPM.



Injector

Cost Reduction: 30%
Part reduction: 252 to 6
Eliminated braze joints
Tested to 100%



Mixer

Part reduction: 8 to 2

Coolant Control Valve

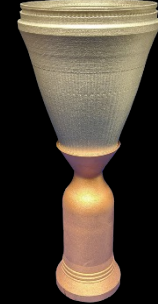
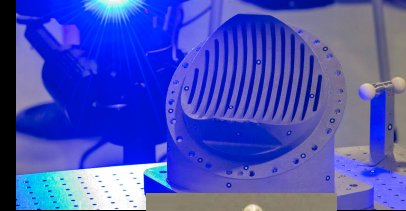
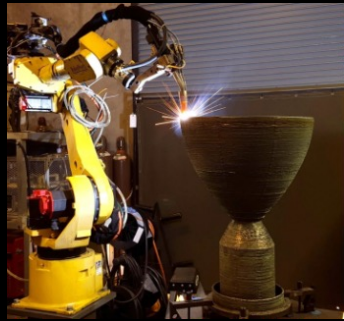
Part reduction: 5 to 1

Thrust Structure



Project lead – NASA Marshall Space Flight Center, 2013~2016

Examples of NASA's AM Dev Work for Rocket Propulsion



Laser Powder Directed Energy Deposition (LP-DED)

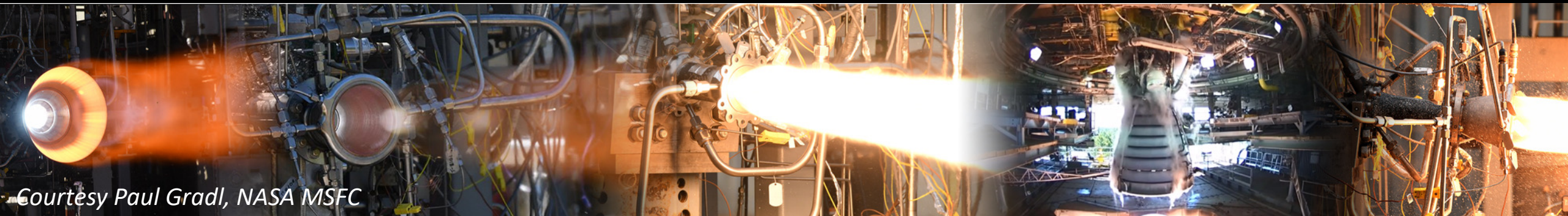


Laser Powder Bed Fusion (L-PBF) GRCop42. *Courtesy Dr. Dave Ellis (NASA GRC).*

Hybrid bi-metallic AM

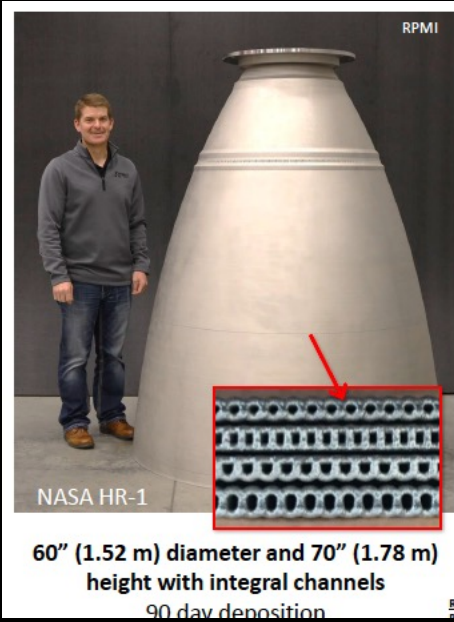
L-PBF AM turbomachinery.

GRX-810 combustor. *Courtesy Dr. Tim Smith (NASA GRC).*



Courtesy Paul Gradl, NASA MSFC

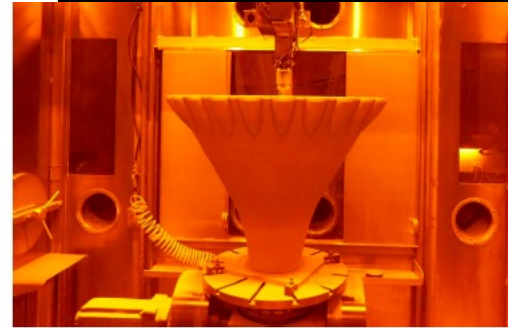
Current AM Development for Rocket Propulsion Components



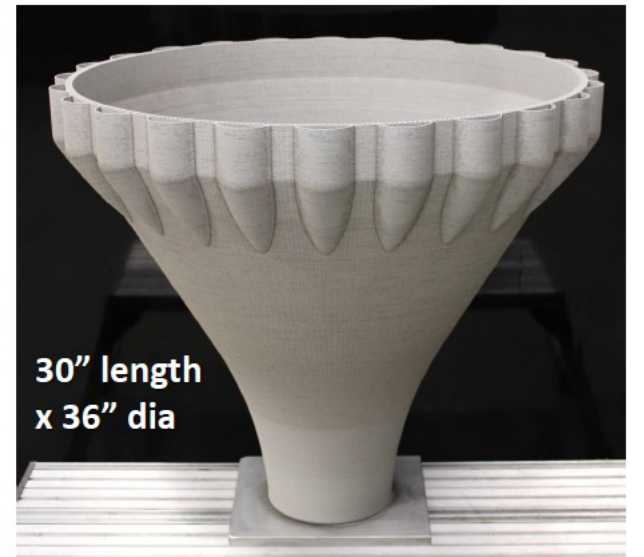
→ Laser Powder Directed Energy Deposition (LP-DED)
Large Scale Rocket Nozzle with integrally printed cooling channels

Reference: P.R. Gradl, T.W. Teasley, C.S. Protz, C. Katsarelis, P. Chen, Process Development and Hot-fire Testing of Additively Manufactured NASA HR-1 for Liquid Rocket Engine Applications, *14th Annual Meeting of the American Society of Mechanical Engineers*, 2021, 3011, <https://doi.org/10.1115/1.6035146>

Demonstrator Aerospike LP-DED Nozzle →



LP-DED Aluminum 6061-RAM2



Photos Courtesy Paul Gradl, NASA MSFC
Excerpts from Gradl's presentation
"Advancing Additively Manufactured Al6061
RAM2 Using Laser Powder Directed Energy
Deposition", Oct 2023, ICAM 2023

Multi-metallic and Multi-AM Process Rocket Chamber Dev



29

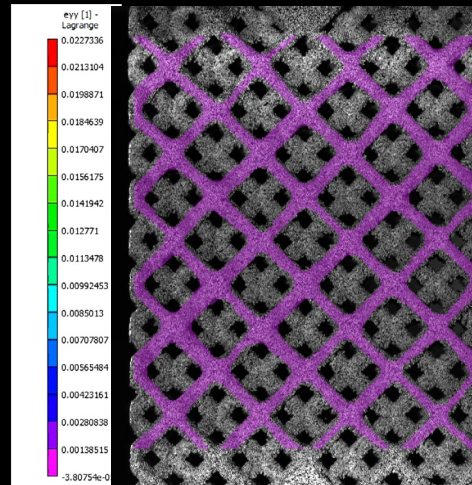
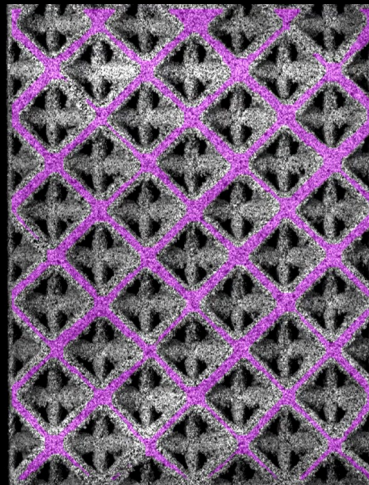


NASA's AM Dev Work for Cryogenic Fluid Management (CFM)

- Long-Duration Propellant Storage
 - LH₂ at 20 K.
 - LOX at 90 K.
- Component Optimization
 - Heat Exchangers
 - Augmented Injectors
 - Tanks with integral cooling channels
 - Galleries
 - Sponges
 - Lattice Struts

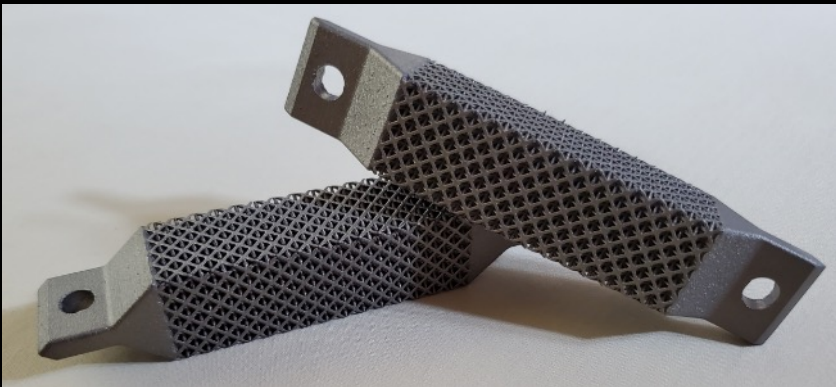


L-PBF AM IN718 TVS Augmented Injector. *Courtesy Travis Belcher (NASA MSFC).*

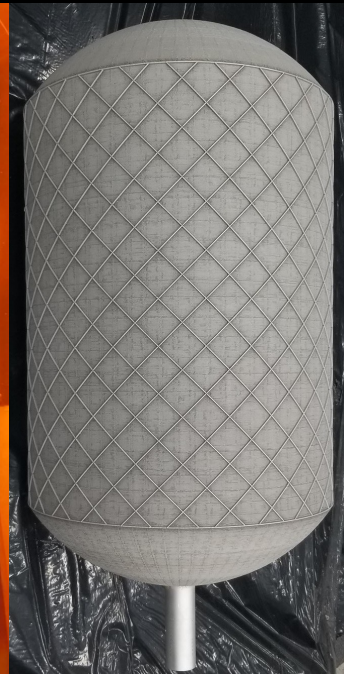
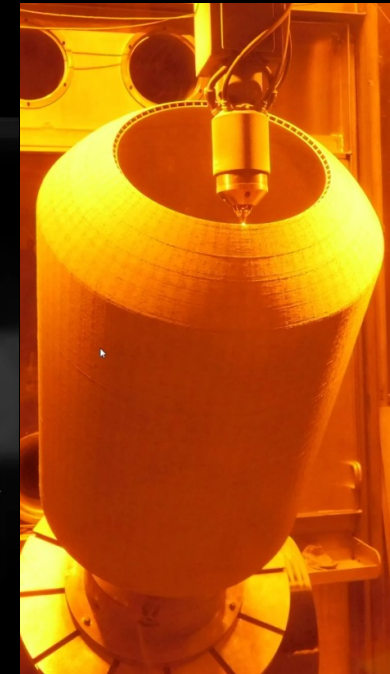


Node strain localization of Octet-Truss-30%RD (left) and Strut strain localization of Rhombic Dodecahedron (20 %RD), $a = 4$ mm, IN718 in SR+HIP+SA condition.

Courtesy Dr. Kavan Hazeli (University of Arizona).



L-PBF AM Ti6Al4V Cryo Lattice Struts. *Courtesy Travis Belcher (NASA MSFC).*



LP-DED AM A6061-RAM2 propellant tank with integral cooling channels. *Courtesy RPM Inc.*

NASA's AM Dev Work for Nuclear Power Systems



- Heat Pipes

- L-PBF AM Ni- or Nb- base
- Surface conformal
- Integrated wicking structures

- Stirling Alternator

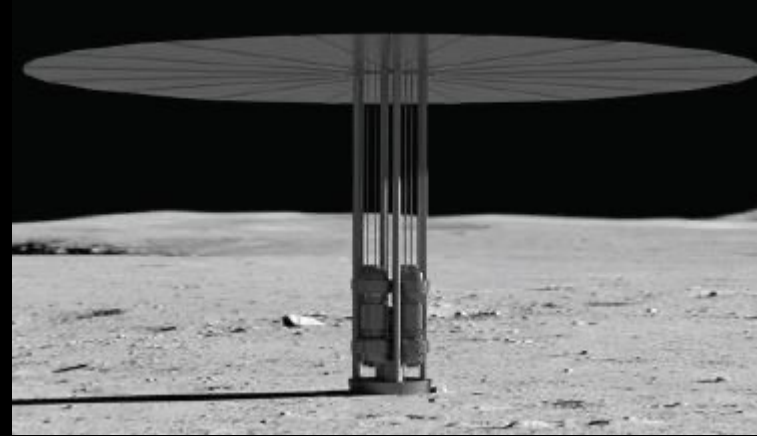
- Lattice Regenerator from L-PBF AM Tungsten

- Fuel

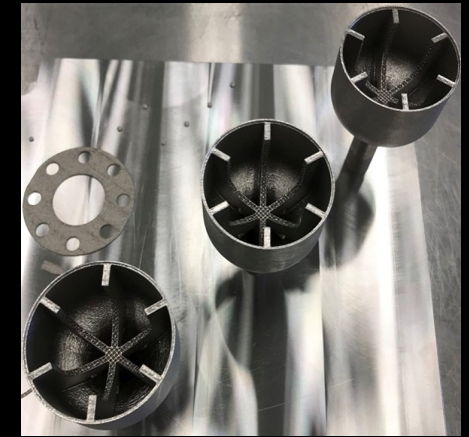
- Additive Friction Stir Deposition (AFS-D) of depleted U-8wt%Mo

- Shielding

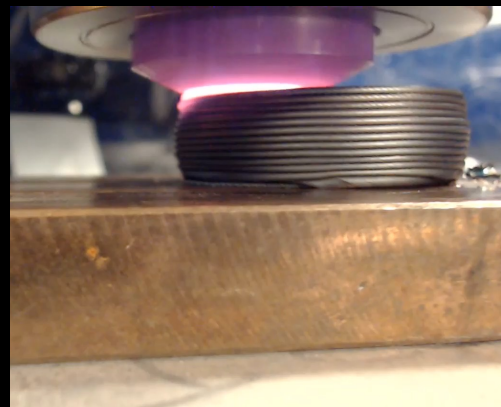
- Multi-material
- Hybrid AM
- Multi-physics topology optimization



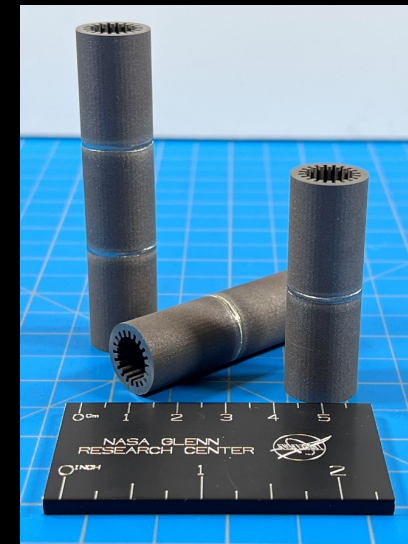
10 kW surface power reactor concept



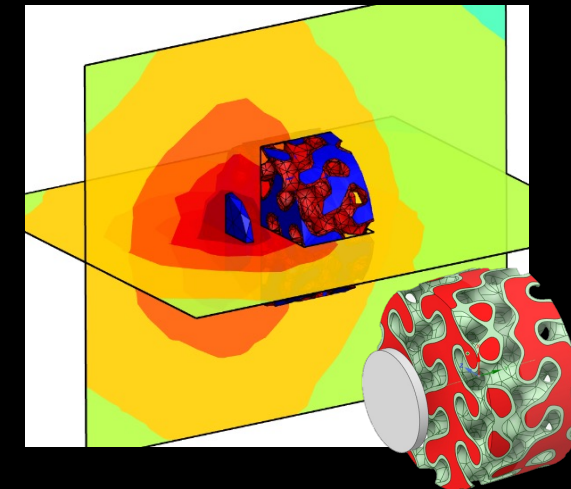
L-PBF AM IN718 integrated heat pipe funnels



AFS-D of dU-8Mo.



L-PBF AM C103 heat pipe segments. *Courtesy Dr. Justin Milner (NASA GRC).*

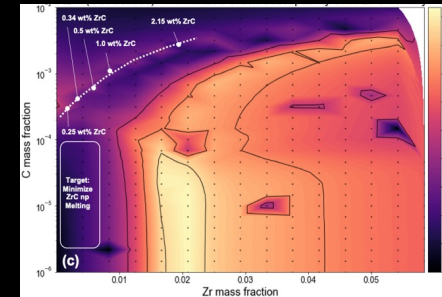


Hybrid AM shield topology optimized for neutron/ γ -ray radiation attenuation, loads, and active cooling. *Courtesy Dr. Jarvis Caffrey (NASA MSFC).*

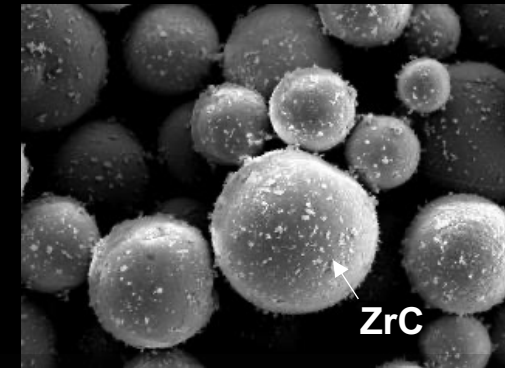
NASA's AM Dev Work in Refractory Metal AM



- Refractory metals for extreme high temperatures
 - Reaction Control System (RCS): chambers
 - Space Nuclear Power & Propulsion: clad, structure, heat pipes
 - Hypergolic / green propulsion: chambers and catalyst.
 - Electric propulsion: grids
 - Hypersonics: wing leading edges

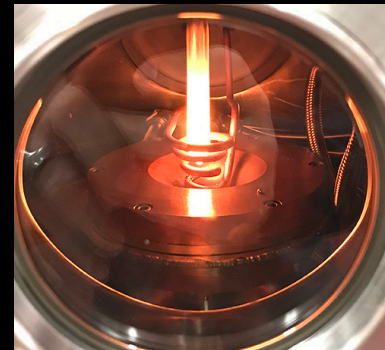


Crack susceptibility modeling. Courtesy Carly Romnes (Univ IL, Urbana-Champaign).



Mo with ZrC nano-powder. Courtesy Dr. Fernando Reyes (NASA MSFC).

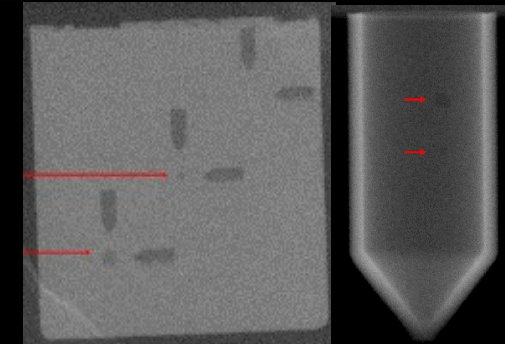
- Development Areas:
 1. ICME, Simulation, Modeling
 2. Feedstock Production
 3. Parameter Development / Part Prod.
 4. Heat Treatment & Joining
 5. Novel Non-Destructive Evaluation
 6. Surface Finishing & Coatings
 7. Property Characterization
 8. Component Test



EIGA of Mo44Re rod.



LP-DED AM C103 development. Courtesy Univ. of Texas, El Paso (UTEP).



Neutron radiograph of L-PBF W with engineered defects. Courtesy Phoenix.



L-PBF AM Mo RCS Chamber

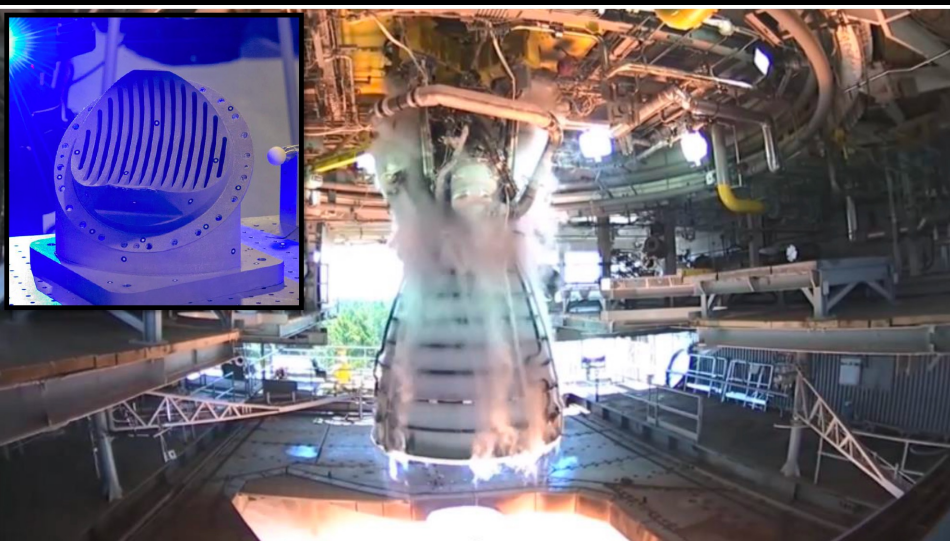


L-PBF AM W chamber test at 1900 °C. Courtesy Dr. Jamelle Williams (NASA MSFC).



L-PBF AM C103 Nozzle Extension. Courtesy Brandon Colon, UTEP.

Great Insertion of AM in Spaceflight – Commercial Space



GE Aviation fuel nozzle (L-PBF)

Titanium pylon rib for F-15 replacements using DED (2003)



Ti6Al4V propellant tank using electron beam wire DED



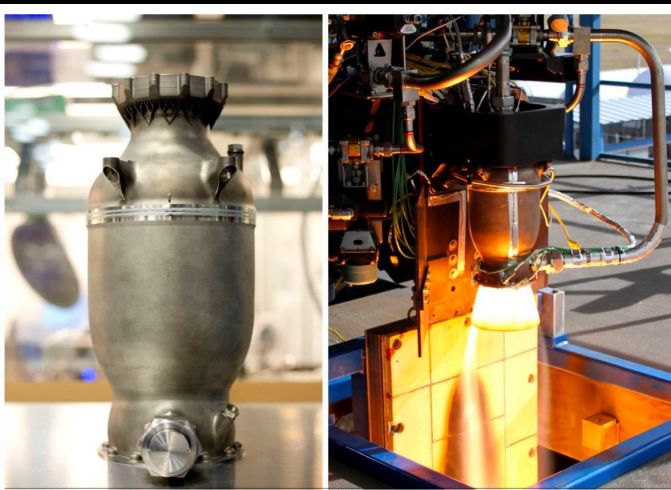
J-2X Rocket Engine Gas Generator Duct:
a) Traditional tooling, b) L-PBF part



Credit: Lockheed Martin

Excerpt from Online AM Training "Metal Additive Manufacturing for Propulsion Applications"

AIAA Progress in Astronautics and Aeronautics Book Serie
P. R. Gradl, O. Mireles, C.S. Protz, C. Garcia. (2022).



Credit: SpaceX

SpaceX SuperDraco Inconel 718 combustion chamber using Inconel 718 L-PBF

Aerojet Rocketdyne RL-10CX All-additive thrust chamber assembly

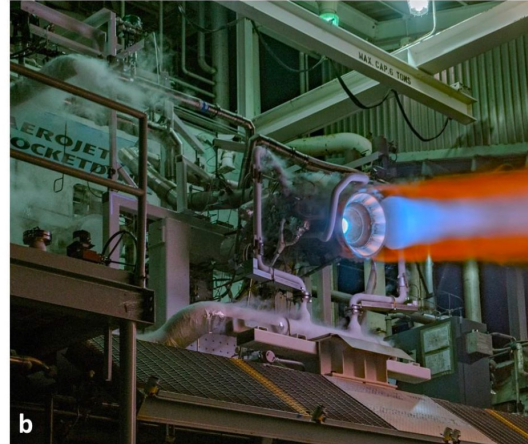


L-PBF Inconel 625 Injector

L-PBF C-18200 Chamber

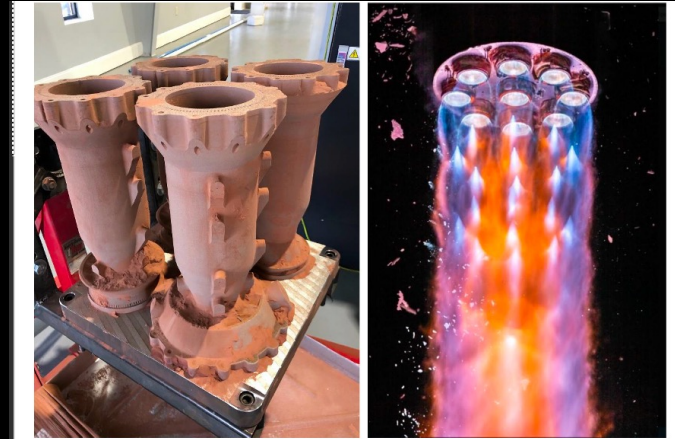
L-PBF Multi-piece Inconel 625 Actively-cooled Nozzle

a



b

Credit: LG Harris Aerojet Rocketdyne



Credit: Relativity Space

Relativity Space – DED for tanks and L-PBF engine components, including NASA alloy GRCop-42

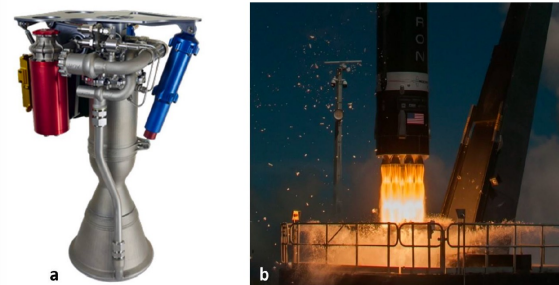
Turbomachinery components using L-PBF



Credit: Barber Nichols

Excerpt from Online AM Training "Metal Additive Manufacturing for Propulsion Applications"

AIAA Progress in Astronautics and Aeronautics Book Series
P. R. Gradl, O. Mireles, C.S. Protz, C. Garcia. (2022).



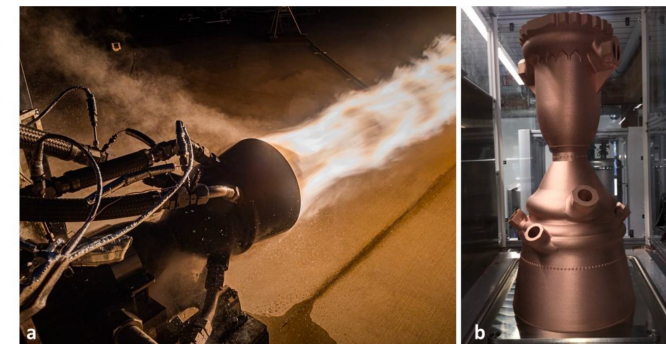
a

b

Credit: Rocket Lab / Andrews Burns and Simon Moffatt

Rocket Lab Rutherford engine using L-PBF Inconel 718 combustion chamber

Launcher Space L-PBF CuCrZr 1 meter height combustion chamber



a

b

Credit: Vast / Launcher



In-Space Manufacturing (ISM)

On-Demand Manufacturing & Recycling of Plastics

- 3D printing and recycling system designed to repeatedly recycle plastic materials into feedstock for manufacturing in the microgravity environment of the ISS.

On-Demand Manufacturing of Metal

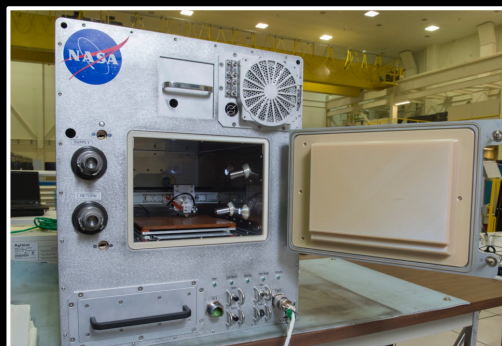
- Additive and subtractive manufacturing systems for creating metal parts on demand.

On-Demand Manufacturing of Electronics

- Ability to fabricate electronics during missions, such as crew and structural monitoring systems



*First 3D printer in space, 2014
ISS Demonstration*



*Refabricator ISS
Demonstration*



*Redwire Commercial ISS Additive
Manufacturing Facility (AMF)*



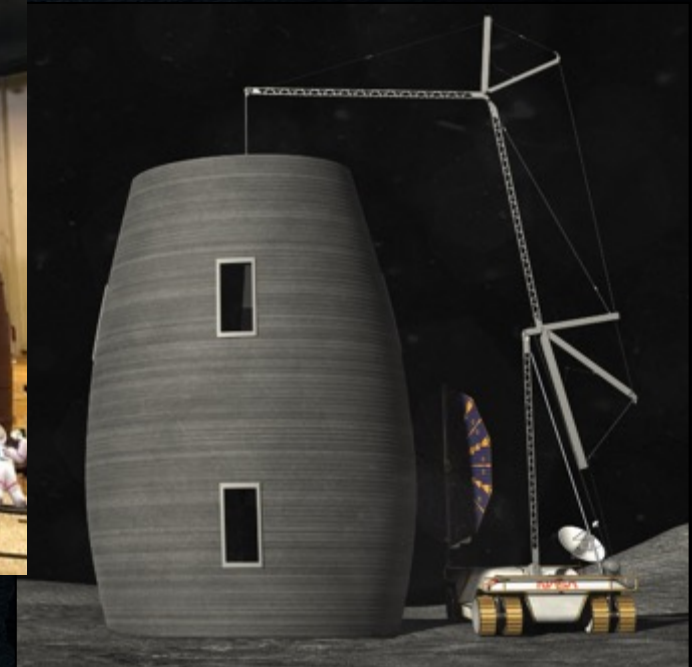
*Techshot Fabrication
Laboratory*

Moon-to Mars Planetary Autonomous Construction Technology (MMPACT)

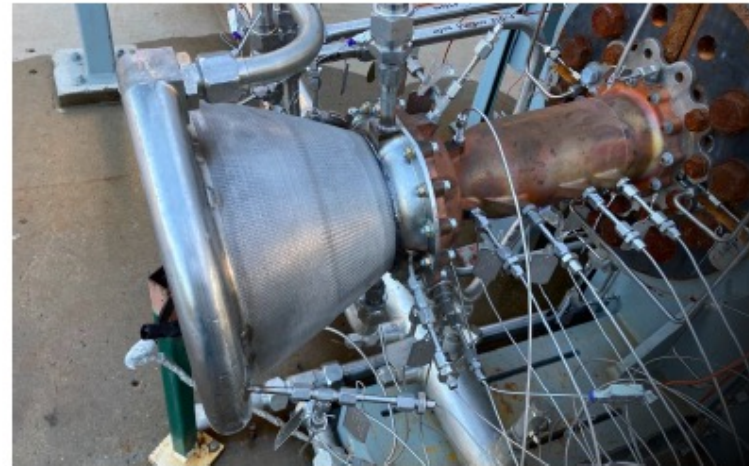
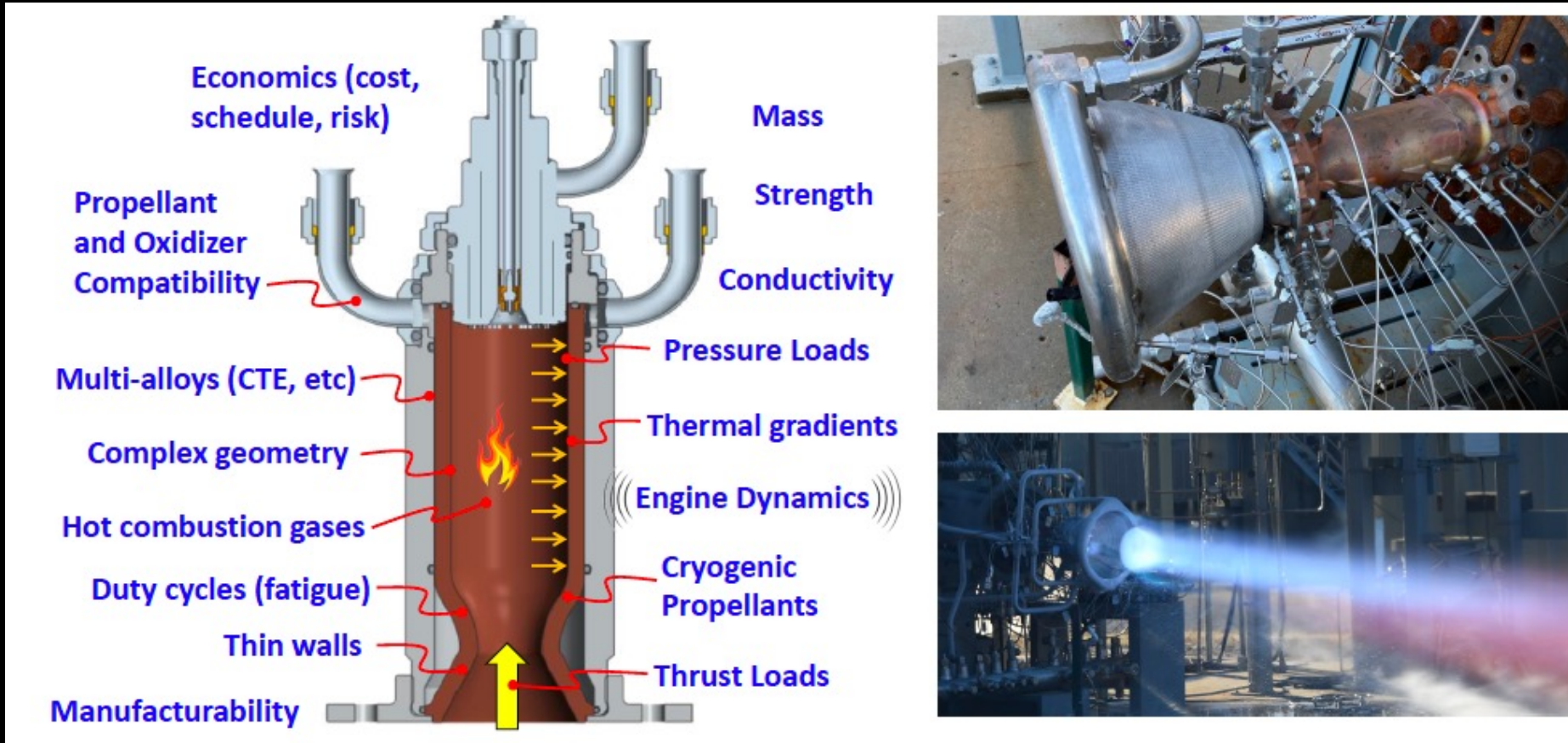
Develop, deliver, and demonstrate capabilities to:

- Protect crew and hardware
- Build infrastructure
- Construct landing pads, habitats, shelters, roadways, berms and blast shields using lunar regolith-based materials

Partners: ICON, SEArch+, USAF, Defense Innovation Unit, Texas Air National Guard

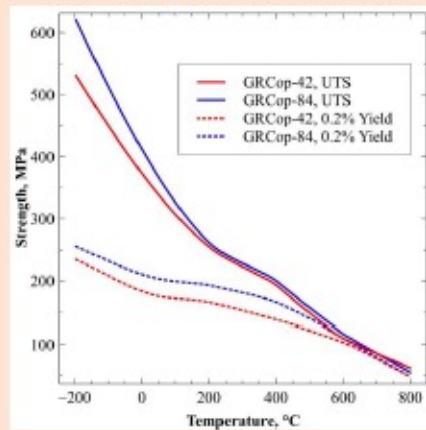


Materials Consideration in Extreme Rocket Engine Environment

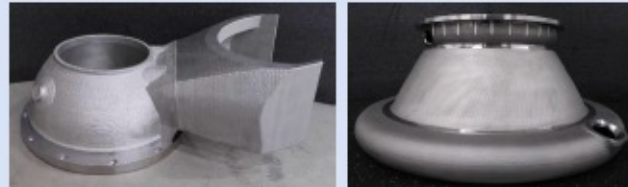
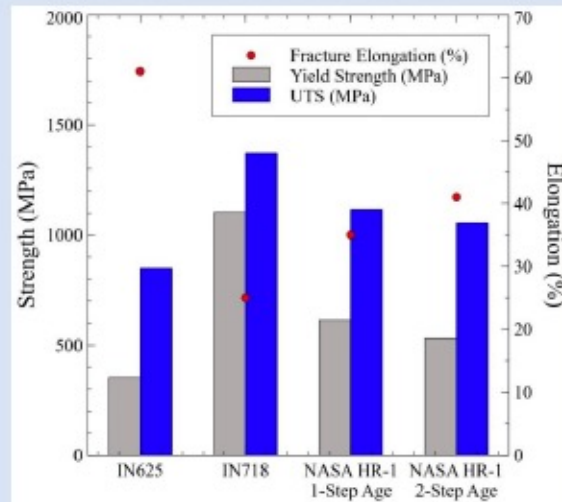


AM enabling novel Alloy Systems

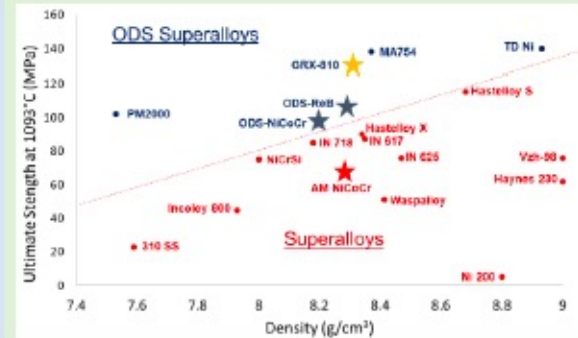
GRCop-42, High conductivity and strength for high heat flux applications



NASA HR-1, high strength superalloy for hydrogen environments



GRX-810, high strength, low creep rupture and oxidation at extreme temperatures



Ref: Tim Smith, Christopher Kantzos / NASA GRC 46

NASA's motivation for AM alloy development

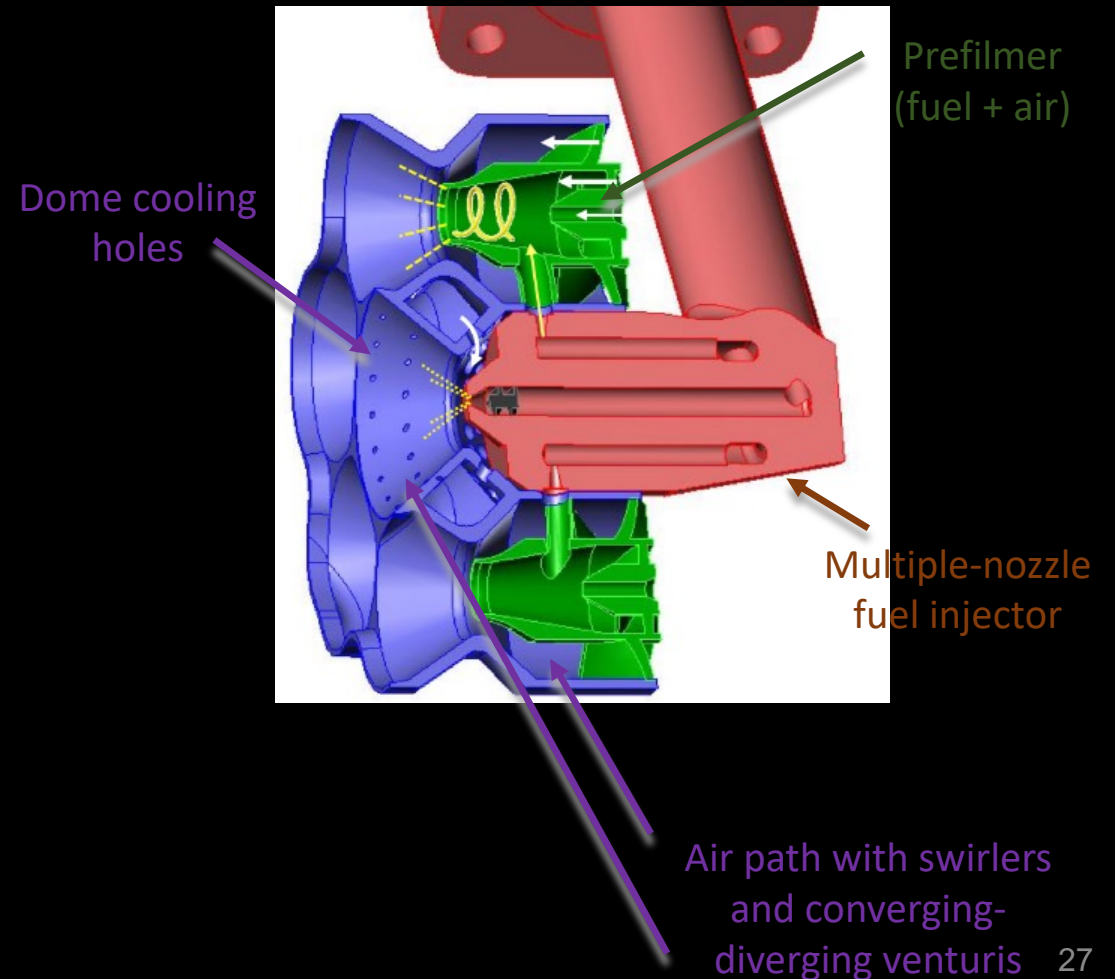
- NASA Missions typically require operation in extreme conditions
- Conventional materials and processing limit design and operation
- Success Case Study—GRX-810

Problem: Optimized Combustor Dome requires AM fabrication and higher than SOA temperatures

Proposed Solution: Develop a high ductility, high temperature material for an additively-manufactured (AM) combustor fuel nozzle and dome for supersonic aircraft (>1093°C (2000°F) operating temperature).

New Alloy and AM fabrication Enabled:

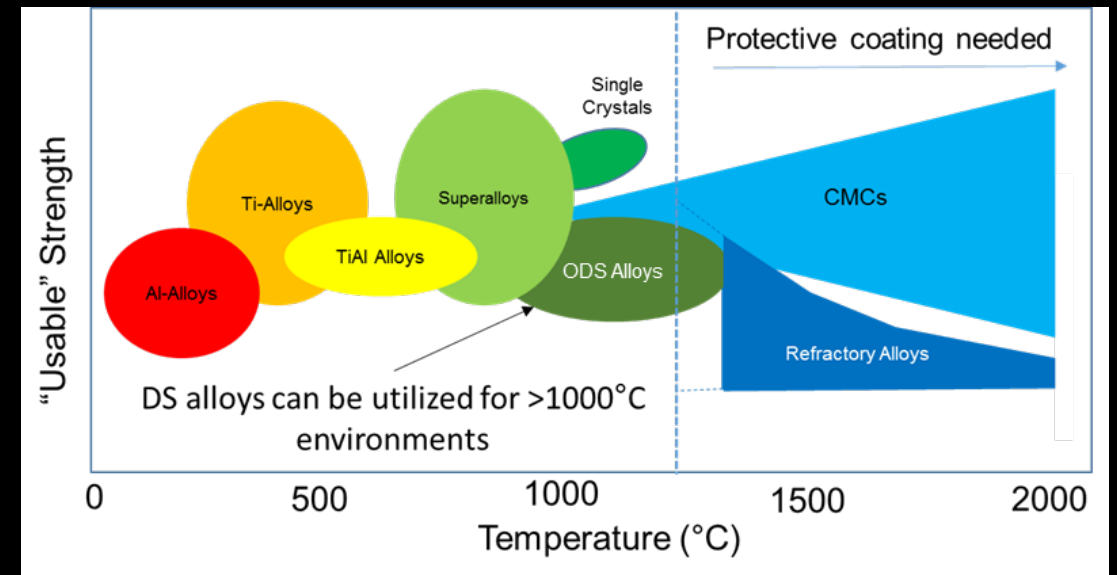
- Improvements to the turbine combustor design reducing NOx pollution and lowering weight
- Lean-front-end small-core combustors



High Temperature AM Compatible Materials

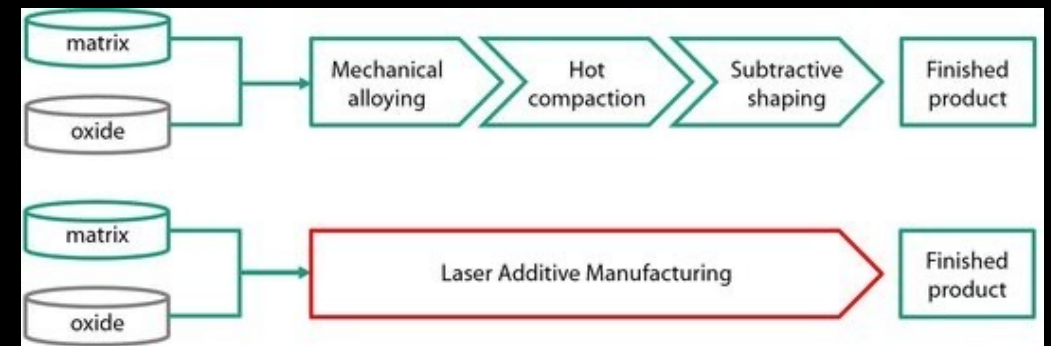
High Temperature Material Options:

- Refractory metals
- Carbon-Carbon composites
- CMC's
- Ni-base superalloys
- Oxide Dispersion strengthened (ODS) alloys



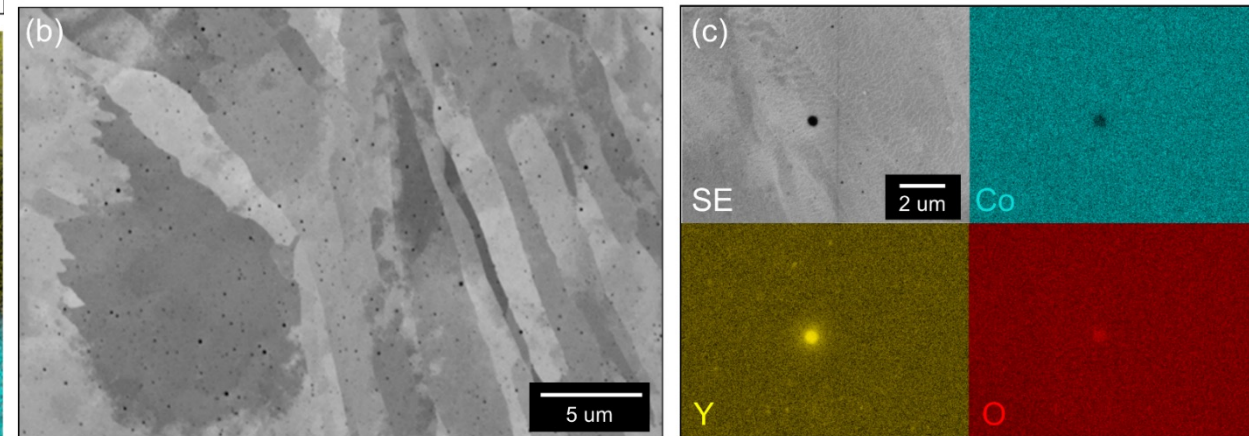
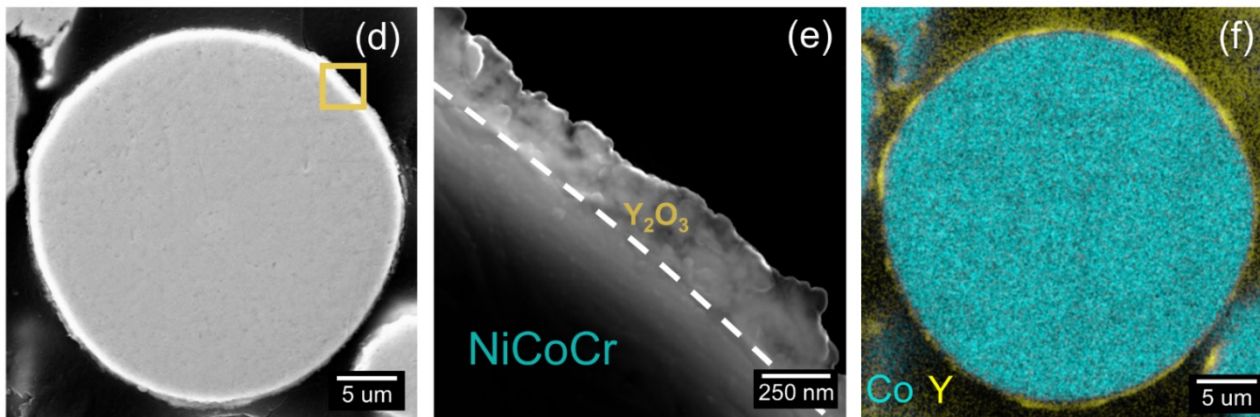
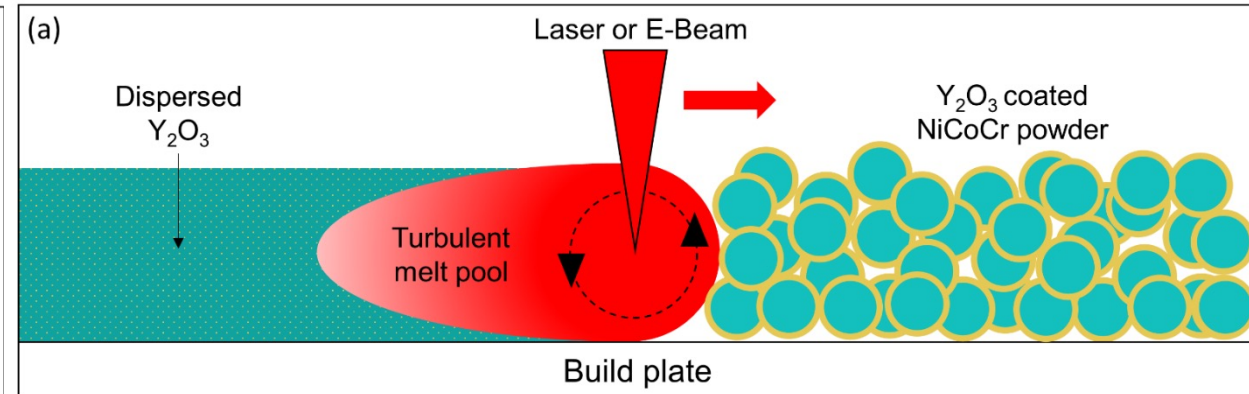
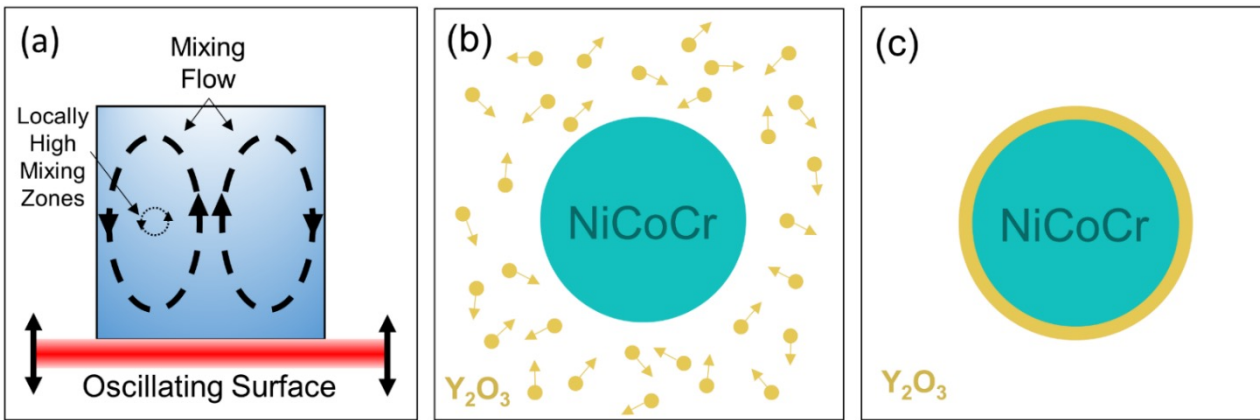
(ODS) alloys offer higher temperature capabilities compared to Ni-base superalloys. However, it has been a challenge to produce ODS alloys through conventional manufacturing methods.

ODS Conventional Manufacturing vs AM



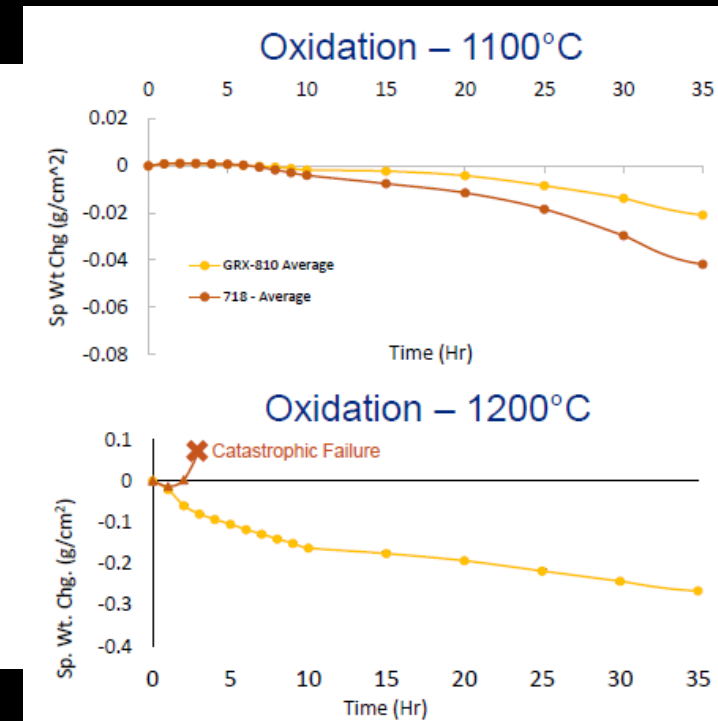
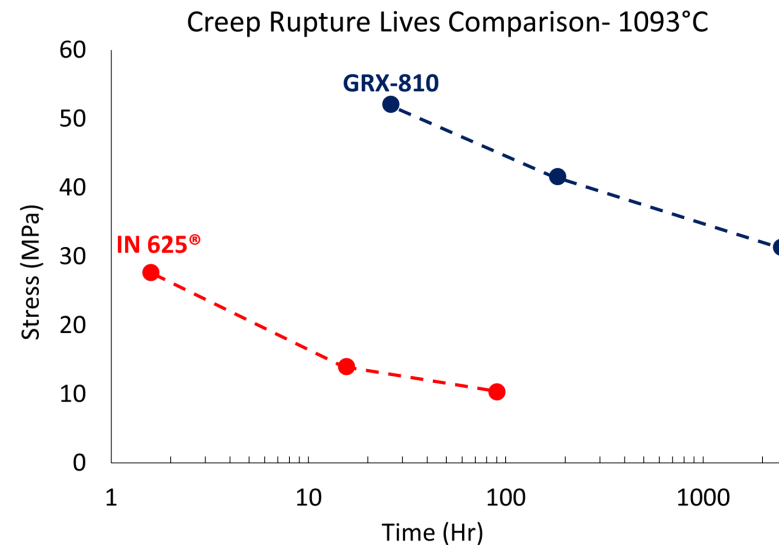
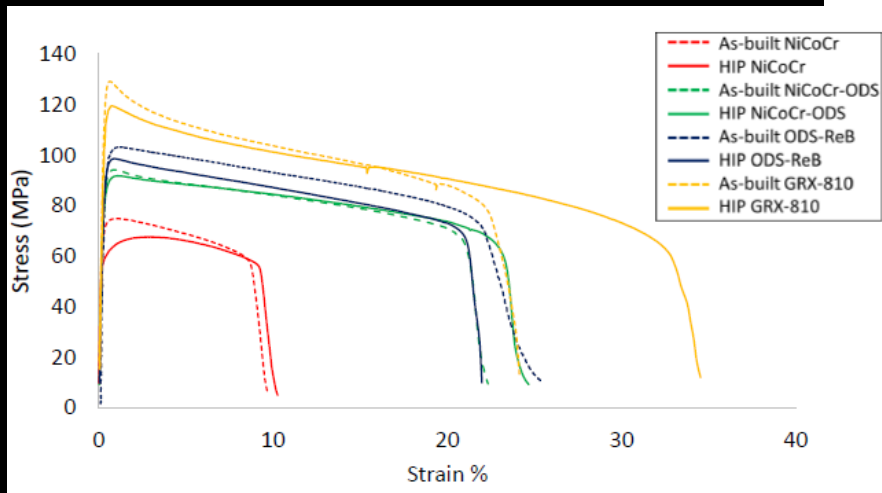
Radical New Approach Developed for ODS AM

Technique for coating metal powders with fine dispersoids was developed and demonstrated first for simple NiCoCr powders and then transitioned to ICME developed alloy Glenn Research Extreme-Temperature above 810°C (GRX-810)

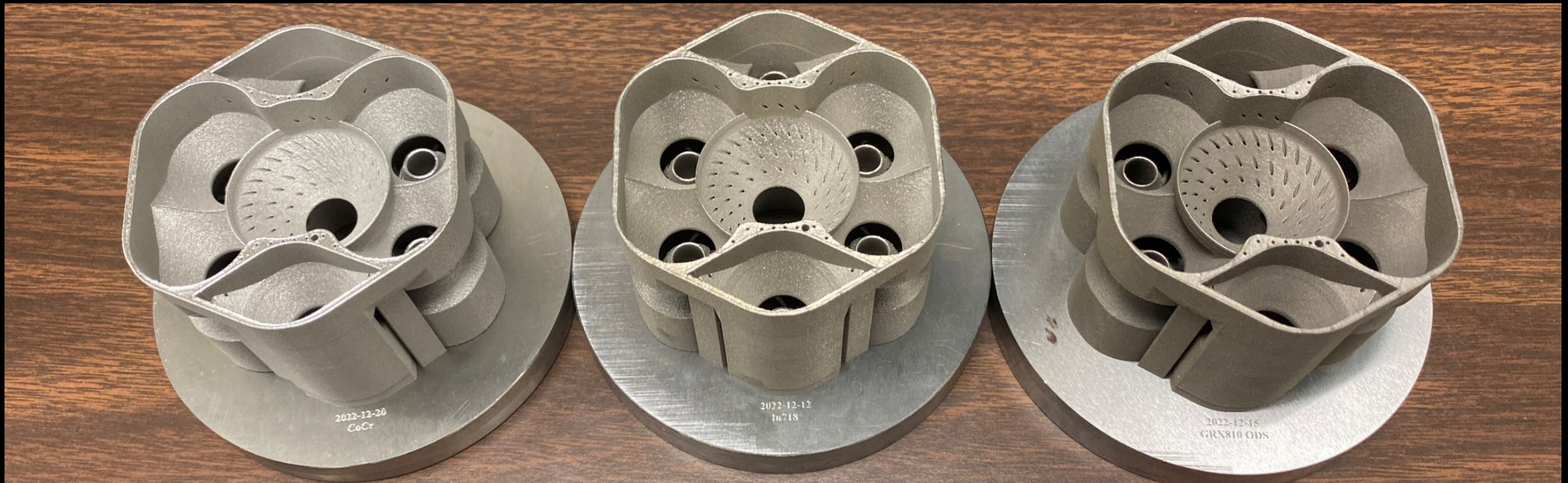


ODS and AM Provides Substantial Improvements

NASA alloy GRX-810 is an example of what can be achieved when a matrix is designed for optimum printing and solid solution strengthening then combined with dispersion strengthening for high temperature stability



Component Demonstration: Combustor Dome



Only GRX-810 dome survived without crack formation

Component Demonstration: Injector / Nozzle



- Nozzle achieved 90 starts and 2,309 seconds of accumulated time
- Injector achieved 84 cycles and 2,228 seconds of accumulated time
- Injector provided at 5x improvement in erosion life compared to previously tested Alloy 718 and Alloy 625 injectors
- Separate design and testing of GRX-810 as a LOX/H₂ injector also completed without erosion seen in conventional alloys



NASA is moving from Low Earth paradigm to Deep Space paradigm

Additively manufactured parts are already being used for NASA programs in critical applications – “ready to fly”

- Transportation from Earth to Destination
- Stations at Destination
- Lander from Station to Surface
- Spacecraft for Science Missions
- Sample Return from Mars

Human exploration of space, especially deep space, requires extreme reliability

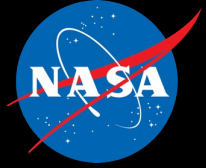


AM technology is demonstrating incredible benefits in wide-spread applications, design flexibility, affordability, schedule, performance, sustainability, and production on demand.

- **Certification is the #1 challenge:** complexity, unknowns, and delays are severely negating the benefits.
- ***Certification of remotely produced products is an unresolved technology gap.***



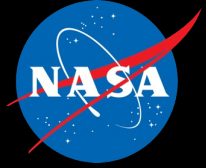
Certification focus in Advanced Manufacturing



- NASA continues to be a leader in the advancement of policies for the Certification of AM processes and parts
 - NASA-STD-6030, Agency standard for additive manufacturing
 - ASTM CoE projects
 - Agency-wide AM Certification Support Team (AACT) activities
 - NASA University Leadership Initiative (ULI)
 - *Development of an Ecosystem for Qualification of Additive Manufacturing Processes and Materials in Aviation*
 - NASA Space Technology Research Institute (STRI)
 - *Institute for Model-Based Qualification & Certification of Additive Manufacturing (IMQCAM)*



Certification focus in Advanced Manufacturing

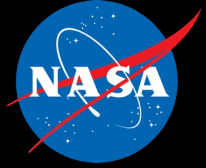


To meet the certification goals of the future, the role of computational materials (CM) will have to grow to meet expectations

- Shortened certification schedules
- Cost pressures
- Rapidly evolving technologies and capabilities in advanced manufacturing

We must acknowledge that the certification process is a human endeavor, often relying on judgement in accepting new materials and processes

- Engineers and managers holding the certification responsibilities are frequently not experts in CM
- The introduction of CM into the certification process must be **measured and pragmatic**
- This requires **planning and a strategy** to lead to well-substantiated acceptance of new roles for CM
- Do **NOT** expect the certification community to readily accept an instantaneous revolution
 - There are strong reasons the community is notoriously conservative
 - Incremental progress must start **NOW**.



Certification focus in Advanced Manufacturing

Two aspects of qualification & certification to consider:

1. Design Certification

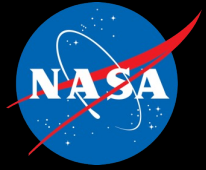
- Demonstration that design meets all requirements of the defined mission

2. Hardware Certification

- Demonstration that hardware meets all requirements of the certified design

Opportunities for computationally-assisted certification

- ***To Begin, focus primarily on augmenting the existing cert. processes, NOT replacing them***
- NASA AM requirements neither endorse nor prohibit such methods
- Tools used in certification require verification and validation
- Best approach to leverage opportunities:
 - Support incremental progress in addition to revolutionary tools
 - Focus on beneficial tools with tractable validation strategies
 - Identify areas of emphasis for industry tool development
 - Strongly leverage partnerships across government, academia, and industry



Today's Certification Approach

- Current approaches for certification rely heavily on empirical methods where the outcomes from numerous combinations of parameters are evaluated experimentally.
- Experimental iteration is the accepted method—Costly in both time and resources
- Endless variations arise due to the sensitive nature of the metallurgical process and individual aspects of AM machines.
- Under current paradigms, once qualified, the **AM process is not extensible**, limiting the ability to make design or manufacturing changes.

Example Laser Powder Bed Fusion Process



Iteration required on Steps 2-6 to Mature/Qualify Part and Processes

The Goal: Digital Twin based Cert

- The ideal “digital twin” (DT) concept bridges both design certification and hardware certification.
- This is the “keep your eyes on the prize” goal
 - See presentation on the NASA STRI scope
- The implementation of the DT concept in certification is best done as an evolution as opposed to a revolution
- The DT lends itself to an evolutionary implementation because its development is inherently incremental and when complete the twin exists as a collection of digital entities that combine to form a substantiation of the process and part.
- We need to watch for opportunities, even if small, to augment the certification process with CM while the DT concept matures.

Design Certification Opportunities

Examples of Design Certification opportunities to leverage CM

- Modeling for build success and build material quality
 - Optimal orientation, thermal control, support strategies
 - Coupon to part geometry correlations based on flaw population and microstructure
 - ***Prediction-based planning for pre-production article assessment***
- Assistance in definition of AM Process Box for process qualification
 - Computational validation of parameters
 - Prediction of process box boundaries
 - ***“Challenge Part” design – prove process box reliability through geometry and thermal history***
- Intelligent risk-based part zoning through integrated assessment tools
 - Prediction of process quality: flaw populations / microstructure
 - Prediction of inspection capability

Hardware Certification Opportunities

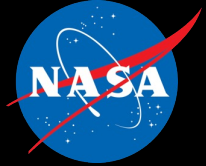
Examples of Hardware Certification opportunities to leverage CM

- Efficiencies in routine part acceptance
 - Machine learning tools
 - Microstructural evaluation
 - Powder feedstock characterization and acceptance
- Validation of in situ monitoring techniques and data
 - Causal relationship must be established between defects and monitored response
 - Modeling of process and monitoring methods may reduce empirical burden
 - ***Enable in situ monitoring to serve a quantitative NDE role***
- Computational non-destructive evaluation (NDE)
 - Understanding detection capability in complex geometries
- Hardware acceptance activities are ripe for near-term implementation of CM

Challenges for CM Implementation

- Acceptance of Computational Materials in the Certification Environment
 - Generate acceptance through gradual implementation
 - Overcome the extensibility issue through broad-based verification and validation
 - Anticipate the same limitations that drive cost in the empirical method will exist for a CM-based approach to certification
 - Most any uniqueness or change in process currently requires empirical evidence to re-certify
 - The same will be true for CM approaches
 - Modeling and simulation approaches in CM will face the same V&V scrutiny whenever the application extends beyond the initial scope of CM tool development and V&V
 - We must plan for ongoing V&V efforts for CM tools as they extend to prediction objectives for various material systems and processes
 - Extended use of CM technologies will not be “free”
 - Keep the human element in mind as we communicate the intend to leverage the use of CM
 - Use well-substantiated, limited scope, incremental advances
 - Start now

NASA STMD – Space Tech Research Institute 2023



Institute Overview



What's an STRI?

Long-term sustained investment in research and technology critical to NASA's future. Highlights include:

- **Empowered** university-led team
- Specific research objectives with **credible expected outcomes** in 5 years
- Talented, **diverse**, cross-disciplinary and fully **integrated** team
- Low to mid TRL

Award Details

- Expected duration: **5 years**
- Award amount up to **\$3M per year**
- Institutes expected (and *empowered*) to implement their own review internal processes
- NASA oversight – annual reviews and brief quarterly status reports

Research Areas

1. A furthered understanding of **materials-processes-structure-property relationships** for AM through advanced computational toolsets coupled with innovative experiments, laying the foundation for accelerated product certification.
2. Development of **uncertainty-quantified, physics-based models and simulations** to understand the factors that can affect the formation, distribution, and the effects of process-induced defects and to address other principal sources of variability.
3. Integration and application of methodology, software tools, artificial intelligence (AI), machine learning (ML), and/or databases, etc., based on computational and experimental tools to a **model-centric certification approach**.

itters

Awarded to Team led by A.D. (Tony) Rollett (Carnegie Mellon's U), PI & Somnath Ghosh (Johns Hopkins U)

NASA led book on Metal Additive Manufacturing



Metal Additive Manufacturing for Propulsion Applications

Edited by
Paul R. Gradl, Omar R. Mireles,
Christopher S. Protz, and Chance P. Garcia



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Timothy C. Lieuwen, Editor-in-Chief
Volume 263

P. R. Gradl, O. Mireles, C.S. Protz, C. Garcia. (2022).

Metal Additive Manufacturing for Propulsion Applications.

AIAA Progress in Astronautics and Aeronautics Book Series

<https://arc.aiaa.org/doi/book/10.2514/4.106279>

Online version and hardcopy available

AM processes are an enabling technology and revolutionizing the propulsion industry. AM-related advancements in new industries, supply chains, design opportunities, and novel materials are increasing at a rapid pace.

The goal of this text is to provide an overview of the practical concept-to-utilization lifecycle in AM for propulsion applications.

