

# JPSS-3 / 4 VIIRS Response Versus Scan Angle Characterization and Performance

Jeff McIntire<sup>1,\*</sup>, David Moyer<sup>2</sup>, Amit Angal<sup>1</sup>, and Xiaoxiong Xiong<sup>3</sup>

<sup>1</sup> Science Systems and Applications, Inc., Lanham, MD 20706, USA

<sup>2</sup> The Aerospace Corporation, El Segundo, CA 90245, USA

<sup>3</sup> NASA Goddard Space Flight Center, Greenbelt, MD, 20771, USA

Correspondence\*:

Jeff McIntire

jeffrey.mcintire@ssaihq.com

## 2 ABSTRACT

3 Scientific studies of the Earth's climate increasingly rely on high quality satellite observations.  
4 The Visible Infrared Imaging Radiometer Suite (VIIRS) is a key sensor onboard the Joint Polar-  
5 orbiting Satellite System 3 and 4 (JPSS-3 and JPSS-4) satellites that generates scientific data  
6 from land, ocean, and atmosphere used in climate models. Providing quality scientific data from  
7 space-borne sensors requires the instruments to be well calibrated, some aspects of which can  
8 best be measured prior to launch. One VIIRS parameter that needs to be measured pre-launch is  
9 the response versus scan angle (RVS). The RVS measures the relative change in the reflectance  
10 of the scanning optics as a function of angle of incidence. With the RVS, the gain calibration  
11 measured on-orbit can be transferred to any scan angle. The JPSS-3 and JPSS-4 instruments  
12 have undergone ground testing including the RVS measurements, which is the subject of this  
13 work. Results indicate that the measurements are comparable to previous VIIRS builds and are  
14 expected to contribute to the generation of high quality science data once JPSS-3 and JPSS-4  
15 are on-orbit.

16 JPSS, VIIRS, calibration, pre-launch, thermal emissive, reflective solar, RVS

## 1 INTRODUCTION

17 The response versus scan angle (RVS) is a key parameter Moyer et al. (2016); McIntire et al. (2017) used in  
18 the calibration of the Visible Infrared Imaging Radiometer Suite (VIIRS), which is currently on-orbit on the  
19 platforms Suomi National Polar-orbiting Partnership (SNPP) Cao et al. (2013); Xiong et al. (2014), Joint  
20 Polar-orbiting Satellite System 1 (JPSS-1) Xiong et al. (2016), and Joint Polar-orbiting Satellite System 2  
21 (JPSS-2) Oudrari et al. (2018). Additional VIIRS instruments are scheduled to fly on JPSS-3 McIntire et al.  
22 (2022) and JPSS-4 with launch dates set for 2032 and 2028 respectively. Each VIIRS build undergoes a  
23 rigorous ground test program to calibrate the sensor and prepare for the sensor calibration to be transferred  
24 to on-orbit operations. Some of the calibration parameters can be updated once the instrument is on orbit.  
25 However, there are some parameters that can only be measured prior to launch; one such parameter is the  
26 RVS. The RVS is the relative variation of the reflectance of the scanning optics as a function of the angle  
27 of incidence. Measurements of the RVS were made as part of a comprehensive ground test program for  
28 JPSS-3 and JPSS-4 VIIRS conducted by Raytheon Technologies in El Segundo, CA in 2019 and 2021

29 respectively. It is the analysis of these measurements and comparisons to earlier builds that is the focus of  
30 this work (preliminary results for JPSS-3 were published McIntire et al. (2021)).

31 On-orbit the calibration of VIIRS is maintained by measurements of the solar diffuser illuminated by  
32 the Sun (for the reflective bands) or measurements of a temperature controlled on-board blackbody (for  
33 the thermal bands). In order to transfer this calibration to any scan angle, the RVS must be measured with  
34 high fidelity. For the reflective bands, the RVS enters as a multiplicative factor; for the thermal bands, the  
35 contribution of the RVS is more complicated, but to leading order also scales as a multiplicative factor. On  
36 orbit, the RVS can be measured, but not with high fidelity, using ground targets for the reflective bands Wu  
37 et al. (2017) and a pitch maneuver for the thermal bands Wu et al. (2018); Wang et al. (2019, 2022). For  
38 SNPP, JPSS-1, and JPSS-2 VIIRS, on-orbit measurements have confirmed the pre-launch RVS (an error in  
39 a look-up-table caused some initial discrepancy for JPSS-1, but has subsequently been corrected). Thus,  
40 the RVS measurements prior to launch are critical to the maintenance of the calibration once the sensor  
41 reaches orbit.

42 The RVS was measured for all previous VIIRS builds, and the testing and analysis methodology have not  
43 substantially changed; as a result this work will focus not only on the results for JPSS-3 and JPSS-4 but  
44 also on the comparison to early VIIRS builds. Section 2 will give a brief overview of the sensor design  
45 and test architecture; Section 3 will provide a overview of the analysis methodology. Section 4.1 will  
46 discuss the reflective band RVS, while the thermal band RVS will be described in Section 4.2. Section 5  
47 will provide some conclusions.

## 2 JPSS-3 / JPSS-4 SENSOR DESIGN AND TESTING PROGRAM

48 The VIIRS sensor design incorporates scanning optics to image a swath of the Earth perpendicular to the  
49 direction of spacecraft motion as it orbits the Earth Xiong et al. (2014). This includes a rotating telescope  
50 assembly (RTA) and a half-angle mirror (HAM). The RTA is a three mirror anastigmat that rotates once  
51 about every 1.7 seconds. The HAM is a double sided mirror rotating at half the speed of the RTA that  
52 de-rotates the light and passes the beam into the aft-optics. The only optic for which the angle of incidence  
53 (AOI) changes is the HAM. It is the variation in reflectance as a function of the HAM AOI that is captured  
54 in the RVS parameters Moyer et al. (2016); McIntire et al. (2017).

55 VIIRS is a filtered radiometer and has 22 spectral bands covering a wavelength range from roughly 0.4  
56  $\mu\text{m}$  to 12  $\mu\text{m}$  Xiong et al. (2014). These bands are listed in Table 1 with their respective nominal center  
57 wavelengths (note that the 12.013  $\mu\text{m}$  band has 2 entries; these are combined via TDI on-orbit, but their  
58 RVS is measured separately here). These bands are divided into two groups based on the main radiation  
59 source they will measure on orbit: reflective solar bands (RSB, I1-I3, M1-M11, and DNB) measure solar  
60 radiance reflected off the Earth and thermal emissive bands (TEB, I4-I5 and M12-M16) measure thermal  
61 radiation emitted by the Earth. These two groups of bands were measured separately in ground testing as  
62 different sources were required to generate sufficient radiance for measurement fidelity.

63 The testing for both sets of bands occurred under ambient conditions Moyer et al. (2016); McIntire  
64 et al. (2017). VIIRS was placed on a rotary table with the scan plane perpendicular to gravity. A source  
65 was placed in the Earth view and the instrument rotated to view this source at different scan angles  
66 (corresponding to different HAM AOI). The space view, which provides a dark offset measurement, had a  
67 ambient target attached to the rotary table such the instrument viewed this source in the space view for all  
68 measurements. For the RSB, a 100 cm integrating sphere was used, illuminated by four 200 W halogen  
69 lamps. Two external radiometers actively monitored the output of the sphere, and were used to remove any

70 source drift. Humidity monitors were also active for use in correcting the varying atmospheric transmission  
 71 for band M9. For the TEB, a cavity type blackbody was used as an extended source, whose temperature  
 72 was controlled at  $\sim 345$  K; an internal blackbody source (used for on-orbit calibration) was controlled  
 73 at  $\sim 312$  K. Data acquired from these tests was then analyzed via the methodology described in the next  
 74 section and in McIntire et al. (2017).

75 The list of measurements for JPSS-3 and JPSS-4 are provided in Tables 2 and 3 for RSB and TEB  
 76 respectively. For the JPSS-4 RSB test, one of the radiance monitors was not active for measurements 7 –  
 77 10; this resulted in the additional measurements 16 – 20. Unfortunately, this radiance monitor was also not  
 78 on for measurement 18. The radiance monitor in question was used to detrend bands I3 and M8-M11; as a  
 79 result, these bands did not use the affected data in the following analysis.

### 3 ANALYSIS METHODOLOGY

80 The analysis methodology was described here follows from McIntire et al. (2017). The HAM AOI is related  
 81 to the scan angle ( $\theta$ ) by the equation

$$\cos(AOI) = \cos(28.6) \cos\left(\frac{\theta}{2} - 23\right) \quad (1)$$

82 Here the AOI is a compound angle of the scan angle and an out-of-plane tilt angle (28.6 degrees) that folds  
 83 the light into the aft-optics.

84 The dark data is used as an offset subtraction for all pixels. Then a scan profile of the source is used  
 85 to generate a centroid, around which 20 pixels are averaged together over all scans. For the RSB, this  
 86 average  $dn$  is then corrected for drift using the appropriate radiance monitor. A humidity model described  
 87 in McIntire et al. (2017) was also used to correct band M9. For the TEB, there is a more complicated  
 88 equation relating the different thermal sources, including those inside the instrument which depend on scan  
 89 angle; this equation is given below

$$\frac{RVS_{ext}}{RVS_{int}} = \frac{dn_{ext}(L_{int} - L_{RTA})}{dn_{int}(L_{ext} - L_{RTA})} \quad (2)$$

90 where the  $L$  denotes the radiance for a given source ( $int$  and  $ext$  refer to the internal and external  
 91 blackbodies).  $RVS_{int}$  and  $RVS_{ext}$  represent the RVS at the view angles of the internal and external  
 92 sources.

93 Then the measured RVS is fit to a quadratic polynomial in the HAM AOI, or

$$RVS = a_0 + a_1AOI + a_2AOI^2 \quad (3)$$

94 The coefficients  $a_i$  are determined via a least-squares fit to the available data, which covers the expected  
 95 HAM AOI range on-orbit (28.6 – 60.5 degrees). The final RVS is normalized to the AOI at the SV (60.47  
 96 degrees). As there can be some differences in the reflectance of the two HAM sides, results were reported  
 97 independently for both (referred to as HAM A and B). The uncertainties are propagated through this  
 98 analysis, as described in McIntire et al. (2017).

## 4 ANALYSIS RESULTS

### 99 4.1 RSB and DNB RVS Performance

100 Fits to the available data were made for all RSB and the DNB using the methodology outlined in Section  
101 3 and in McIntire et al. (2017). An example of the un-normalized fitting for JPSS-4 band M1 (HAM side A)  
102 is shown in Figure 1. The symbols represent the measured data and the lines denote the fit RVS functions.  
103 The fit lines show good agreement with the measured data, and this is also true for all other RSB. This is  
104 illustrated in Figures 2 and 3 for all RSB and the DNB (for a middle detector). Here HAM sides A and B  
105 are denoted by solid and dashed lines. The green and purple lines represent the JPSS-3 and JPSS-4 VIIRS  
106 RVS performance; black, red, and blue denote the performance for SNPP, JPSS-1, and JPSS-2 VIIRS  
107 Moyer et al. (2016); McIntire et al. (2017). In general, the RVS for the two HAM sides is very consistent  
108 for SNPP, JPSS-2, JPSS-3, and JPSS-4. Some HAM side differences were observed for JPSS-1 due to  
109 the two sides of the HAM mirror being coated at very different times Moyer et al. (2016). Additionally,  
110 the detector variation in a given band is small across all reflective bands, as seen in Figure 1. The RVS in  
111 Figures 2 and 3 are normalized to the space view angle ( $AOI_{SV} = 60.47$ ). The AOI used on-orbit ranges  
112 from about 28.6 to 60.5 degrees. The largest variation over this ranges occurs for the bluest bands (M1-M3)  
113 with up to about 1.5 % change. The variation is smallest for bands I2, M7, and M11 with less than 0.1  
114 % change observed. An atmospheric correction has been applied to band M9, as shown in Figure 4 for  
115 JPSS-4. The upper plot shows the M9 RVS without the correction and the lower plot shows the RVS after  
116 the correction was applied. Silimiar results were obtained for JPSS-3 band M9.

117 The comparison to earlier builds shows that the RVS has been consistent in the general shape. There was  
118 a change in coating from SNPP to JPSS-1 Oudrari et al. (2014, 2016) and the HAM side differences for  
119 JPSS-1 noted earlier. Nonetheless, the magnitude, shape, and detector variation are very similar, with some  
120 minor exceptions (HAM side differences in SNPP for M6 and in JPSS-1 for I2 / M7 as well as detector  
121 variation on JPSS-1 for the DNB edge detectors). The methodology employed to analyze the data has been  
122 consistent and the test setup has not changed markedly. This implies that the JPSS-3 and JPSS-4 RSB /  
123 DNB RVS will likely perform as well as the SNPP, JPSS-1, and JPSS-2 RVS have performed once on orbit.

124 The uncertainties on the measured RVS data were propagated through the fitting routines to provide an  
125 uncertainty estimate on the final RVS as a function of HAM AOI. As the RVS was normalized to the space  
126 view angle (about 60.5 degrees), the uncertainty is smallest at the normalization point and increases as  
127 the AOI decreases, where the maximum uncertainty occurs at the minimum AOI of  $\sim 28.6$  degrees. A  
128 detailed description of the uncertainty propagation and calculation is contained in McIntire et al. (2017).  
129 For both JPSS-3 and JPSS-4, the maximum RSB RVS uncertainties are listed in Table 4, along with JPSS-2  
130 uncertainties from McIntire et al. (2017). The values range from 0.046 % for band M4 to 0.007 % for  
131 band M10 and the DNB. Values were comparable to JPSS-2 uncertainty estimates. VIIRS RSB have a  
132 total uncertainty design requirement of 2 %, of which 0.3 % was allocated to RVS ref (2014). The values  
133 for all reflective bands and the DNB were well within this target value. The largest contributions to the  
134 uncertainty are the  $a_1$  and  $a_2$  terms. As the RVS enters into the calibrated reflectance as a multiplicative  
135 factor for the reflective bands, the maximum uncertainty on the reflectance due to RVS for each band is  
136 equivalent to the maximum RVS uncertainties listed in Table 4.

### 137 4.2 TEB RVS Performance

138 All of the TEB bands were analyzed according to the methodology outlined in Section 3 and in McIntire  
139 et al. (2017). An example of the un-normalized RVS is shown in Figure 5 for JPSS-4 band M14 (HAM side

140 A) for all detectors. The symbols represent the measured data and the lines denote the fit RVS functions.  
141 The RVS fits reproduce the measured data very well, as is the case for all other thermal bands. The RVS for  
142 a middle detector is shown for all thermal bands in Figure 6, where HAM sides A and B are denoted by  
143 solid and dashed lines. The JPSS-3 and JPSS-4 RVS are represented by the green and purple lines (SNPP,  
144 JPSS-1, and JPSS-2 are plotted in the black, red, and blue lines, respectively). Here the RVS has been  
145 normalized to the space view angle (AOI  $\sim 60.5$  degrees). Band M14 shows the largest variation, up to  
146 about 10 %, while the 3 - 4  $\mu m$  bands (I4, M12, and M13) show less than 0.5 % change over the full range  
147 of AOI. The HAM side differences are small in the thermal bands, and the detector dependence is also  
148 generally small.

149 While there is some variation with VIIRS build, the general shape, detector dependence, and magnitude  
150 of the RVS is very consistent. SNPP and JPSS-1 seem to have slightly smaller variation for the longer  
151 wavelength bands (8 - 12  $\mu m$ ), as well as larger HAM side differences. The mid-wavelength bands have  
152 consistently smaller RVS (below 0.5 % total variation) for all builds, with a little bit more curve to the  
153 shape in JPSS-3. Neither the analysis methodology nor the test setup has changed much over the series of  
154 VIIRS tested; this indicates that the level of performance will be comparable to the observed performance  
155 of SNPP, JPSS-1, and JPSS-2 on orbit.

156 The uncertainties in the thermal bands were more complicated as the individual sources had to propagate  
157 through Equation (2). Then the uncertainties were used in the fitting routine, and finally the error on  
158 the final RVS as a function of HAM AOI was calculated as per the methodology outlined in McIntire  
159 et al. (2017). As the RVS was normalized to the space view angle (about 60.5 degrees), the uncertainty is  
160 smallest at the normalization point and increases as the AOI decreases, where the maximum uncertainty  
161 is at the minimum AOI of  $\sim 28.6$  degrees. The maximum uncertainties for both JPSS-3 and JPSS-4 are  
162 listed in Table 5 for all thermal bands, along with the uncertainties for JPSS-2 for comparison McIntire  
163 et al. (2017). The largest was 0.29 % for band I4 and the smallest was 0.06 % for bands M15 and M16B;  
164 the thermal bands generally have larger uncertainties than the reflective bands due to the complexity of the  
165 uncertainty propagation. The thermal bands have a stratified scene requirement on the total uncertainty,  
166 but were all allocated 0.2 % for the RVS (except band M14 which was allocated 0.6 %). Some bands  
167 did not meet this threshold (bands I4 and I5), but previous builds have shown good performance on orbit  
168 with similar uncertainties. This was in part driven by larger noise in these two bands, which has been  
169 a feature since SNPP. The largest contributions to the total uncertainty for all thermal bands are the  $a_1$   
170 and  $a_2$  terms. The RVS enters the calibrated radiance as a multiplicative factor, but its influence on the  
171 brightness temperature is less straightforward. Tables 6 and 7 show the JPSS-3 and JPSS-4 uncertainties in  
172 the brightness temperature due solely to the RVS. Brightness temperature uncertainties are generally around  
173 0.1 K or lower, except at low scene temperatures (below 230 K for I4, M12 and M13; below 210 K for I5  
174 and M14–M16). I5 has slightly higher uncertainties for most scene temperatures. JPSS-4 uncertainties are  
175 generally lower than JPSS-3 uncertainties. Note that this is a worst case estimate near the end of a scan  
176 (where the HAM AOI is near its minimum).

## 5 CONCLUSIONS

177 Maintaining the calibration of the VIIRS instruments once on orbit is critical to ensure the continued  
178 delivery of high quality science data products for various science and climate studies. While some of the  
179 calibration can be performed on-orbit, some parameters can best be measured prior to launch. One such  
180 parameter is the response versus scan angle (RVS), which describes the relative variation in reflectance as  
181 a function of angle of incidence on the half-angle mirror. Measurements were made for the JPSS-3 and

182 JPSS-4 VIIRS instruments, the analysis results of which were discussed in this work. The results show  
183 that the RVS for all bands was within expectations, was comparable to previous builds (SNPP, JPSS-1,  
184 and JPSS-2), and had uncertainty estimates showing high fidelity for these measurements. Considering the  
185 consistency with previous builds and their continued delivery of high quality data products on orbit, it is  
186 expected that the measured RVS for JPSS-3 and JPSS-4 will contribute to the continuity of high quality  
187 science data for these missions.

## CONFLICT OF INTEREST STATEMENT

188 The authors declare that the research was conducted in the absence of any commercial or financial  
189 relationships that could be construed as a potential conflict of interest.

## AUTHOR CONTRIBUTIONS

190 Jeff McIntire wrote the manuscript and performed the analysis presented in this work. David Moyer and  
191 Amit Angal performed independent analyses to confirm the results presented here. Xiaoxiong Xiong  
192 oversaw the testing and analysis.

## FUNDING

193 Details of all funding sources should be provided, including grant numbers if applicable. Please ensure to  
194 add all necessary funding information, as after publication this is no longer possible.

## ACKNOWLEDGMENTS

195 The authors would like to thank the following: the Raytheon test team for conducting the performance  
196 tests and for developing much of the analysis methodology, and members of the government data analysis  
197 working group including James McCarthy for valuable comments. The above mentioned provided valuable  
198 information and support to the analysis presented in this work.

## REFERENCES

- 199 (2014). *Joint Polar Satellite System (JPSS) VIIRS Product Requirement Document (PRD)*. Tech. rep.,  
200 Goddard Space Flight Center, Greenbelt, MD, USA. Revision D
- 201 Cao, C., De Luccia, F. J., Xiong, X., Wolfe, R., and Weng, F. (2013). Early  
202 on-orbit performance of the visible infrared imaging radiometer suite onboard the  
203 Suomi National Polar-Orbiting Partnership (S-NPP) satellite. *IEEE Transactions on Geoscience  
204 and Remote Sensing* 52, 1142–1156
- 205 McIntire, J., Moyer, D., Angal, A., and Xiong, X. (2021).  
206 JPSS-3 VIIRS response versus scan angle characterization and performance. In *Earth Observing  
207 Systems XXVI* (SPIE), vol. 11829, 173–180
- 208 McIntire, J., Moyer, D., Chang, T., Oudrari, H., and Xiong, X. (2017).  
209 Pre-launch JPSS-2 VIIRS response versus scan angle characterization. *Remote Sensing* 9, 1300
- 210 McIntire, J., Xiong, X., Butler, J. J., Angal, A., Moyer, D., Ji, Q., et al. (2022).  
211 An overall assessment of JPSS-3 VIIRS radiometric performance based on pre-launch testing. *Remote  
212 Sensing* 14, 1999

**Table 1.** VIIRS spectral bands along with their nominal center wavelengths (in nm).

Band	Center	Band	Center	Band	Center
M1	412	M7	865	I4	3740
M2	445	I2	865	M13	4050
M3	488	M8	1240	M14	8550
M4	555	M9	1378	M15	10763
I1	640	M10	1610	I5	11450
M5	672	I3	1610	M16A	12013
DNB	700	M11	2250	M16B	12013
M6	746	M12	3700		

- 213 Moyer, D., McIntire, J., Oudrari, H., McCarthy, J., Xiong, X., and De Luccia, F. (2016).  
 214 JPSS-1 VIIRS pre-launch response versus scan angle testing and performance. *Remote Sensing* 8, 141  
 215 Oudrari, H., McIntire, J., Xiong, X., Butler, J., Ji, Q., Schwarting, T., et al. (2016).  
 216 JPSS-1 VIIRS radiometric characterization and calibration based on pre-launch testing. *Remote*  
 217 *Sensing* 8, 41
- 218 Oudrari, H., McIntire, J., Xiong, X., Butler, J., Ji, Q., Schwarting, T., et al. (2018). An overall assessment  
 219 of JPSS-2 VIIRS radiometric performance based on pre-launch testing. *Remote Sensing* 10, 1921
- 220 Oudrari, H., McIntire, J., Xiong, X., Butler, J., Lee, S., Lei, N., et al. (2014).  
 221 Prelaunch radiometric characterization and calibration of the S-NPP VIIRS sensor. *IEEE Transactions*  
 222 *on Geoscience and Remote Sensing* 53, 2195–2210
- 223 Wang, W., Cao, C., and Blonski, S. (2019). A new method for characterizing noaa-20/s-npp viirs thermal  
 224 emissive bands response versus scan using on-orbit pitch maneuver data. *Remote Sensing* 11, 1624
- 225 Wang, W., Cao, C., and Blonski, S. (2022). Estimating the viirs thermal emissive band response versus  
 226 scan (rvs) and calibration offsets using on-orbit pitch maneuver data. *IEEE Transactions on Geoscience*  
 227 *and Remote Sensing* 60, 1–10
- 228 Wu, A., Xiong, X., and Cao, C. (2017). Assessment of stability of the response versus scan angle for the  
 229 S-NPP VIIRS reflective solar bands using pseudo-invariant desert and Dome C sites. In *Proc. of SPIE*  
 230 *Vol. vol. 10423, 1042322–1*
- 231 Wu, A., Xiong, X., and Chiang, K. (2018). Determination of the NOAA-20 VIIRS TEB RVS from  
 232 emissive radiation measurements during the pitch maneuver. In *Earth Observing Missions and Sensors:*  
 233 *Development, Implementation, and Characterization V* (SPIE), vol. 10781, 227–239
- 234 Xiong, X., Butler, J., Chiang, K., Efremova, B., Fulbright, J., Lei, N., et al. (2014).  
 235 VIIRS on-orbit calibration methodology and performance. *Journal of Geophysical Research:*  
 236 *Atmospheres* 119, 5065–5078
- 237 Xiong, X., Butler, J., Chiang, K., Efremova, B., Lei, N., McIntire, J., et al. (2016). VIIRS on-orbit  
 238 calibration methodology and performance. *Remote Sens.* 8, 84

239 **5.1 Tables**240 **5.2 Figures**

**Table 2.** JPSS-3 and JPSS-4 VIIRS measurements for the RSB bands including scan angle and AOI (both in degrees) in the order measured.

JPSS-3			JPSS-4		
Collection	Scan Angle	AOI	Collection	Scan Angle	AOI
1	-65.7	60.47	1	-65.70	60.47
2	-8.27	38.62	2	-8.66	38.74
3	-38.36	49.41	3	-38.82	49.59
4	5.69	34.49	4	5.17	34.63
5	-45.31	52.15	5	-45.81	52.34
6	-8.27	38.62	6	-8.73	38.76
7	-55.33	56.18	7	-55.73	56.35
8	-30.24	46.31	8	21.16	30.97
9	-8.27	38.62	9	-30.84	46.54
10	21.71	30.87	10	-8.71	38.76
11	37.82	28.87	11	-51.75	54.73
12	-17.41	41.67	12	37.41	28.89
13	54.70	28.90	13	-20.68	42.82
14	-8.31	38.63	14	54.27	28.87
15	-51.43	54.60	15	-8.64	38.73
			16	-8.64	38.73
			17	-55.71	56.34
			18	21.20	30.96
			19	-30.83	46.54
			20	-8.75	38.77

**Table 3.** JPSS-3 and JPSS-4 VIIRS measurements for the TEB bands including scan angle and AOI (both in degrees) in the order measured.

Collection	JPSS-3		Collection	JPSS-4	
	Scan Angle	AOI		Scan Angle	AOI
1	-7.95	38.51	1	-8.95	38.84
2	-66.22	60.69	2	-66.59	60.84
3	21.94	30.83	3	21.13	30.97
4	-45.12	52.07	4	-45.73	52.31
5	5.96	34.42	5	5.36	34.58
6	-8.04	38.54	6	-8.95	38.84
7	-55.58	56.29	7	-56.26	56.56
8	-20.16	42.64	8	-20.77	42.85
9	-7.97	38.52	9	-38.79	49.58
10	-50.99	54.42	10	-9.17	38.91
11	35.05	29.08	11	-51.56	54.65
12	-7.99	38.53	12	34.33	29.14
13	-61.16	58.58	13	-30.67	46.47
14	-35.13	48.16	14	-8.94	38.83
15	-27.19	45.18	15	-61.60	58.77

**Table 4.** JPSS-2 McIntire et al. (2017), JPSS-3, and JPSS-4 VIIRS RSB measured maximum uncertainties (in %).

Band	JPSS-2	JPSS-3	JPSS-4	Band	JPSS-2	JPSS-3	JPSS-4
M1	0.041	0.044	0.028	M7	0.016	0.014	0.013
M2	0.026	0.027	0.021	I2	0.017	0.016	0.014
M3	0.025	0.024	0.021	M8	0.013	0.012	0.010
M4	0.053	0.046	0.042	M9	0.021	0.037	0.031
I1	0.015	0.014	0.012	M10	0.009	0.007	0.007
M5	0.035	0.031	0.028	I3	0.026	0.030	0.028
DNB	0.010	0.007	0.007	M11	0.016	0.013	0.012
M6	0.023	0.022	0.019				

**Table 5.** JPSS-2 McIntire et al. (2017), JPSS-3, and JPSS-4 VIIRS TEB measured maximum uncertainties (in %).

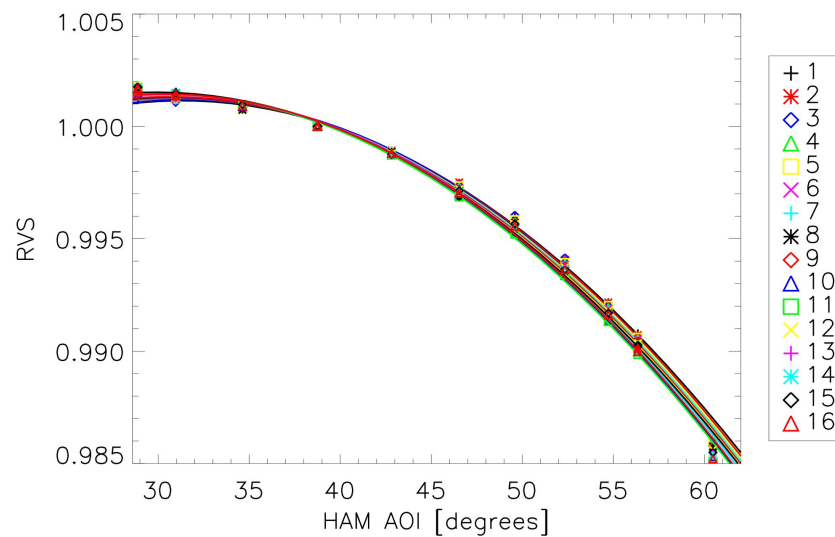
Band	JPSS-2	JPSS-3	JPSS-4
M12	0.17	0.19	0.10
I4	0.26	0.29	0.21
M13	0.15	0.18	0.10
M14	0.13	0.14	0.07
M15	0.11	0.12	0.06
I5	0.23	0.25	0.21
M16A	0.11	0.12	0.07
M16B	0.11	0.12	0.06

**Table 6.** Brightness temperature uncertainties due solely to the RVS maximum uncertainties listed in Table 5 (in K) for JPSS-3.

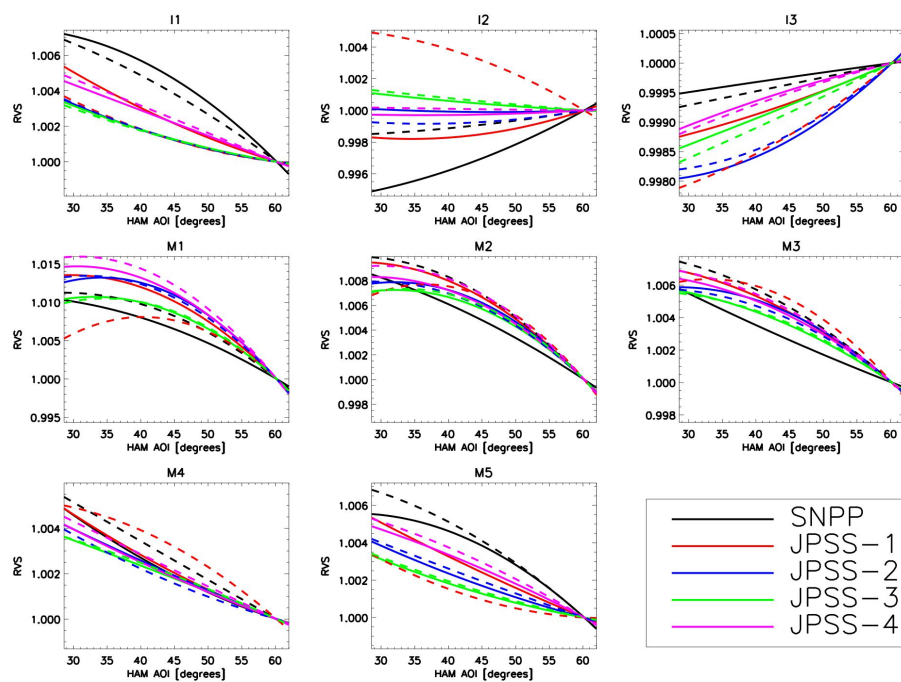
Scene T	M12	I4	M13	M14	M15	I5	M16A	M16B
190	0.33	0.55	0.38	0.42	0.23	0.60	0.23	0.22
210	0.35	0.59	0.41	0.18	0.12	0.31	0.12	0.12
230	0.28	0.47	0.24	0.08	0.07	0.18	0.06	0.06
250	0.07	0.13	0.08	0.04	0.03	0.09	0.03	0.03
270	0.03	0.04	0.03	0.03	0.03	0.07	0.03	0.03
280	0.04	0.06	0.04	0.04	0.04	0.10	0.04	0.04
310	0.05	0.08	0.05	0.06	0.05	0.15	0.06	0.06
330	0.06	0.10	0.06	0.08	0.07	0.19	0.08	0.08
345	0.07	0.11	0.07	0.09	0.08	0.23	0.09	0.09

**Table 7.** Brightness temperature uncertainties due solely to the RVS maximum uncertainties listed in Table 5 (in K) for JPSS-4.

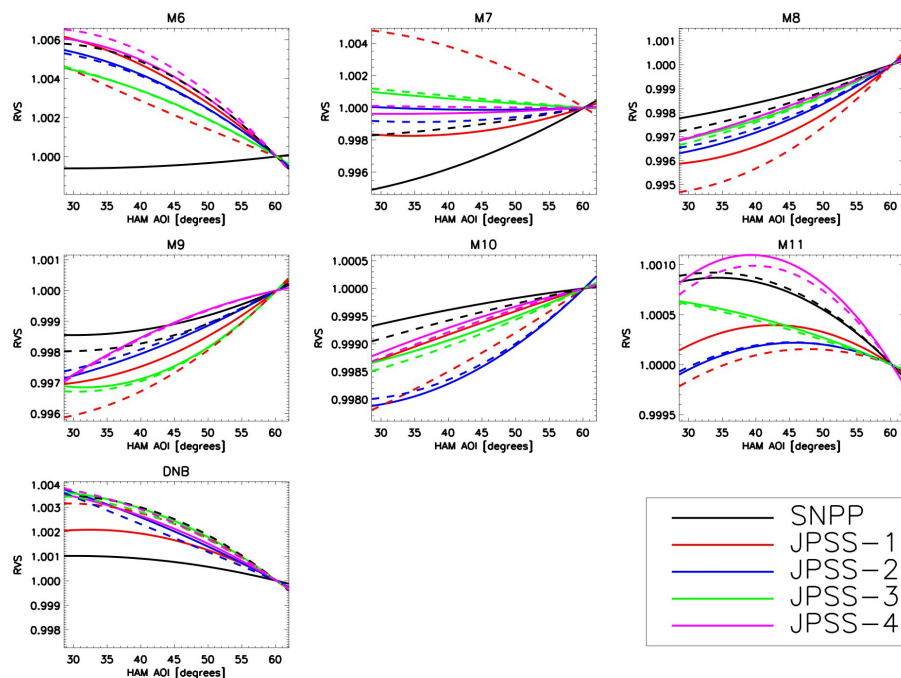
Scene T	M12	I4	M13	M14	M15	I5	M16A	M16B
190	0.18	0.41	0.21	0.21	0.12	0.49	0.12	0.12
210	0.20	0.45	0.23	0.09	0.06	0.25	0.07	0.07
230	0.16	0.36	0.13	0.04	0.03	0.15	0.04	0.04
250	0.04	0.10	0.04	0.02	0.02	0.07	0.02	0.02
270	0.02	0.03	0.01	0.01	0.01	0.06	0.02	0.02
280	0.02	0.05	0.02	0.02	0.02	0.09	0.02	0.02
310	0.03	0.06	0.03	0.03	0.03	0.12	0.03	0.03
330	0.04	0.07	0.03	0.04	0.04	0.16	0.04	0.04
345	0.04	0.08	0.04	0.04	0.04	0.19	0.05	0.05



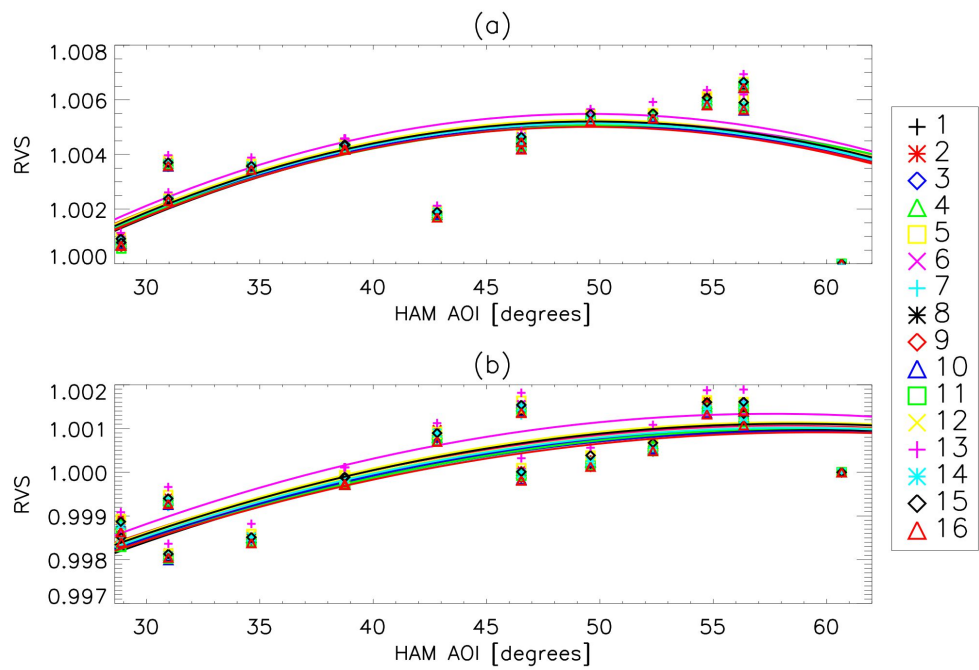
**Figure 1.** JPSS-4 band M1 RVS fitting (HAM side A). Symbol / color combinations listed in the legend denote different detectors in band M1.



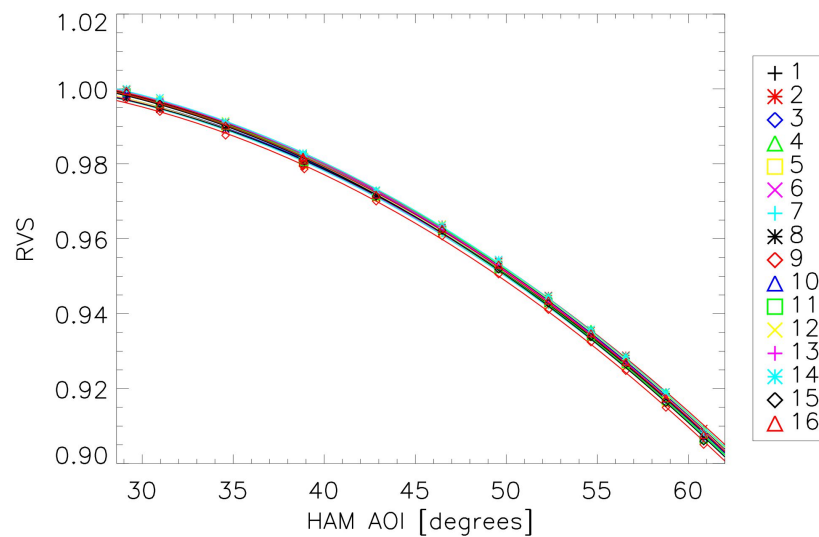
**Figure 2.** VIIRS RVS for bands I1-I3 and M1-M5. HAM sides A and B denoted by solid and dashed lines.



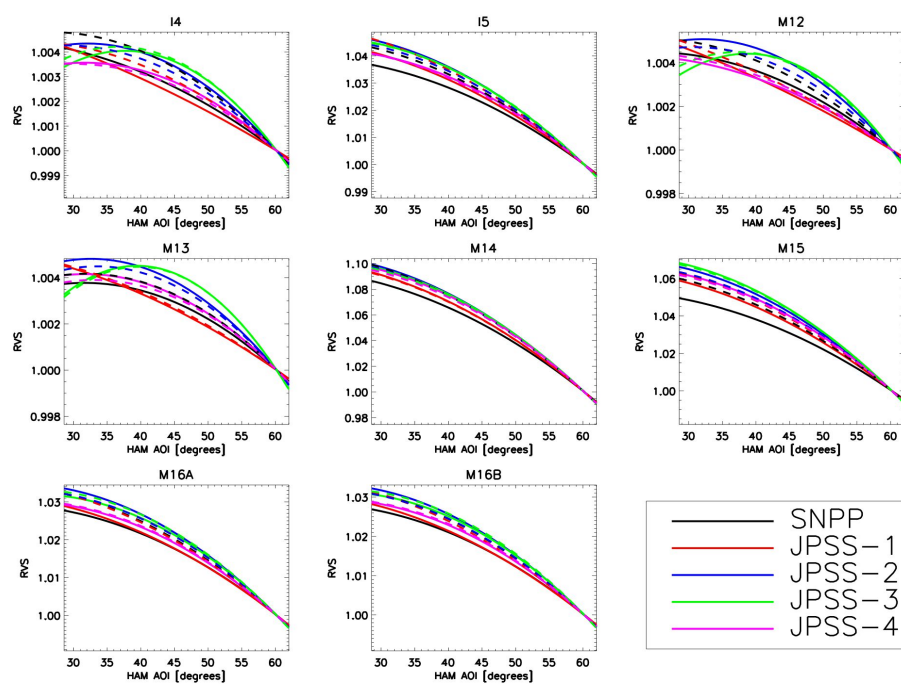
**Figure 3.** VIIRS RVS for bands M6-M11 and the DNB. HAM sides A and B denoted by solid and dashed lines.



**Figure 4.** JPSS-4 VIIRS RVS for bands M9, with and without a humidity correction (upper and lower plots) for HAM side A. Symbol / color combinations listed in the legend denote different detectors in band M9.



**Figure 5.** JPSS-4 band M14 RVS fitting (HAM side A). Symbol / color combinations listed in the legend denote different detectors in band M14.



**Figure 6.** VIIRS RVS for TE bands. HAM sides A and B denoted by solid and dashed lines.