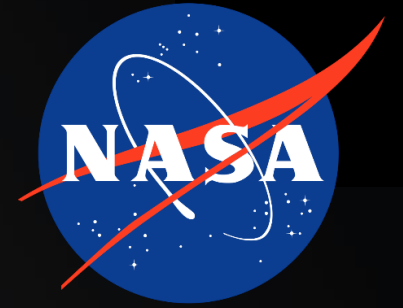


National Aeronautics and Space Administration



# Extreme Temperature Additively Manufactured GRX-810 Alloy Development and Hot-fire Testing for Liquid Rocket Engines

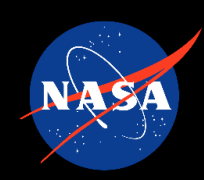
**Paul Gradl, Darren Tinker, Ben Williams**

NASA Marshall Space Flight Center

**Tim Smith, Christopher Kantzos**

NASA Glenn Research Center

9 January 2024

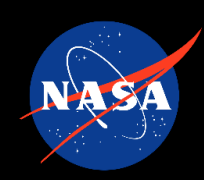


# The need for novel extreme environment alloys

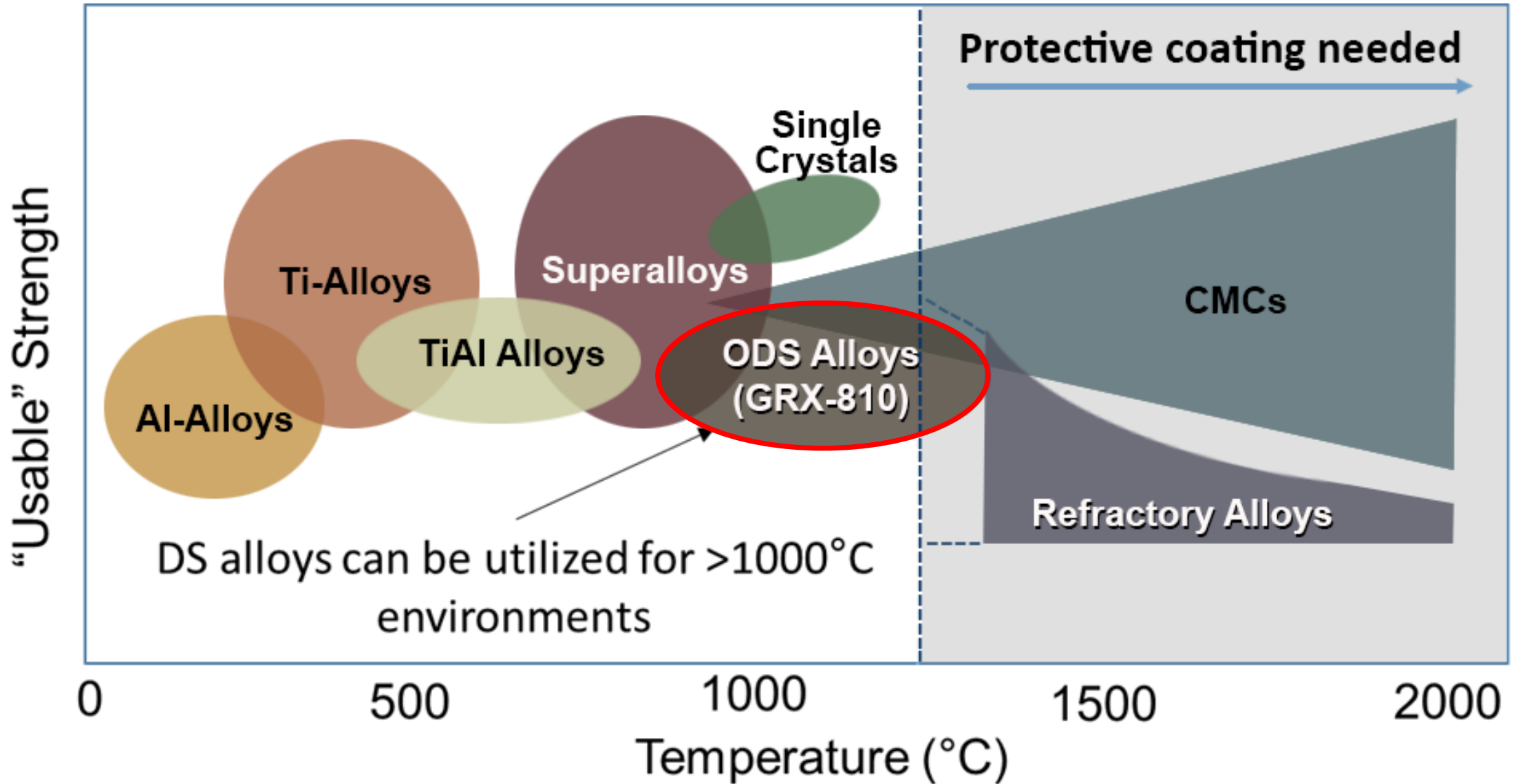


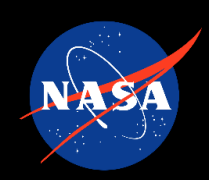
- Rocket engine components (injectors, nozzles, turbine static and rotating components, chambers) operate in extreme environments.
  - High pressure, high heat flux, high thermal gradients, long duty cycles, stress rupture (creep), oxidation
- Additive manufacturing (AM) has played a pivotal role in component development and production, but materials are still based on heritage metal alloys.
- New alloys using unique advantages of AM processing can provide a step change in properties to allow for higher performance.
- GRX-810 is an oxide dispersion strengthened (ODS) alloy for sustained use at 900-1200°C with high strength and 100x stress rupture capabilities.

**There is a need to advance metal alloys using additive manufacturing processes to enable higher performing components.**



# Landscape of Aerospace Materials and GRX-810

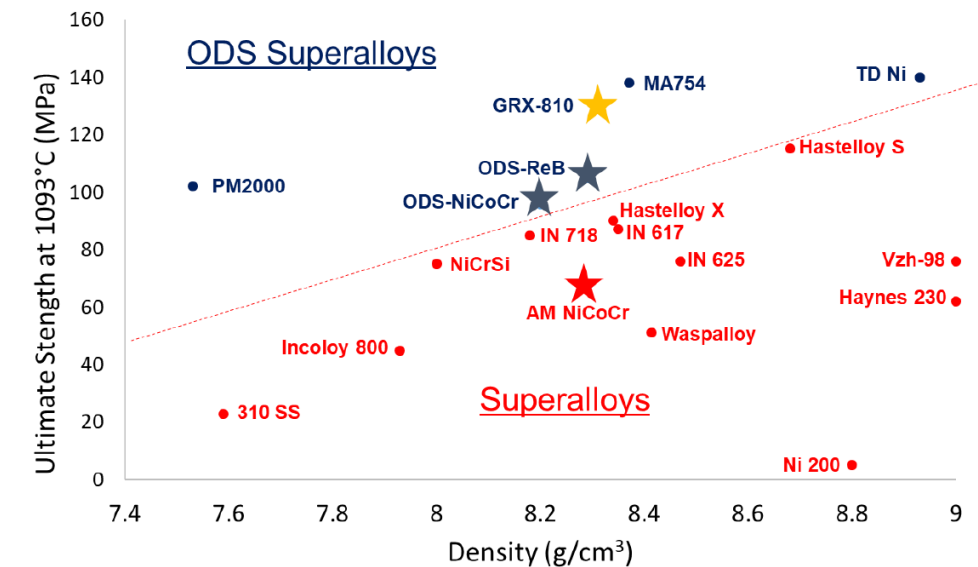
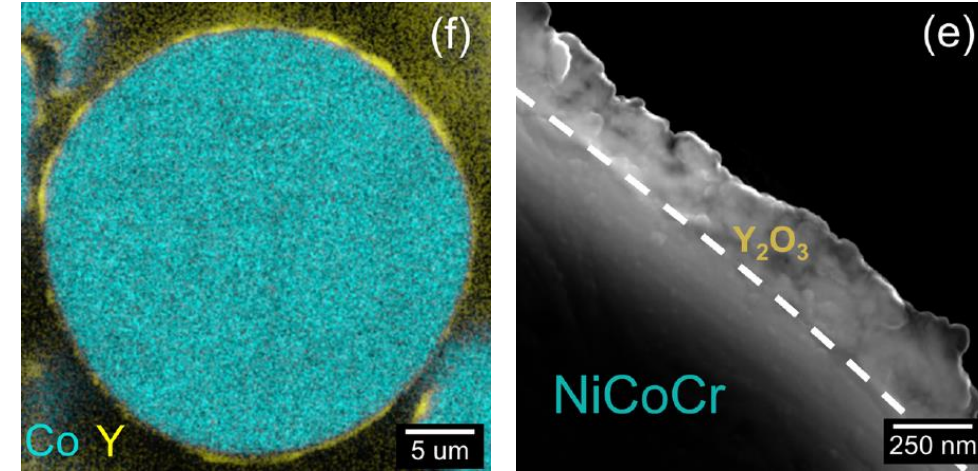


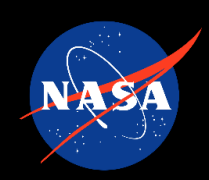


# GRX-810 Overview

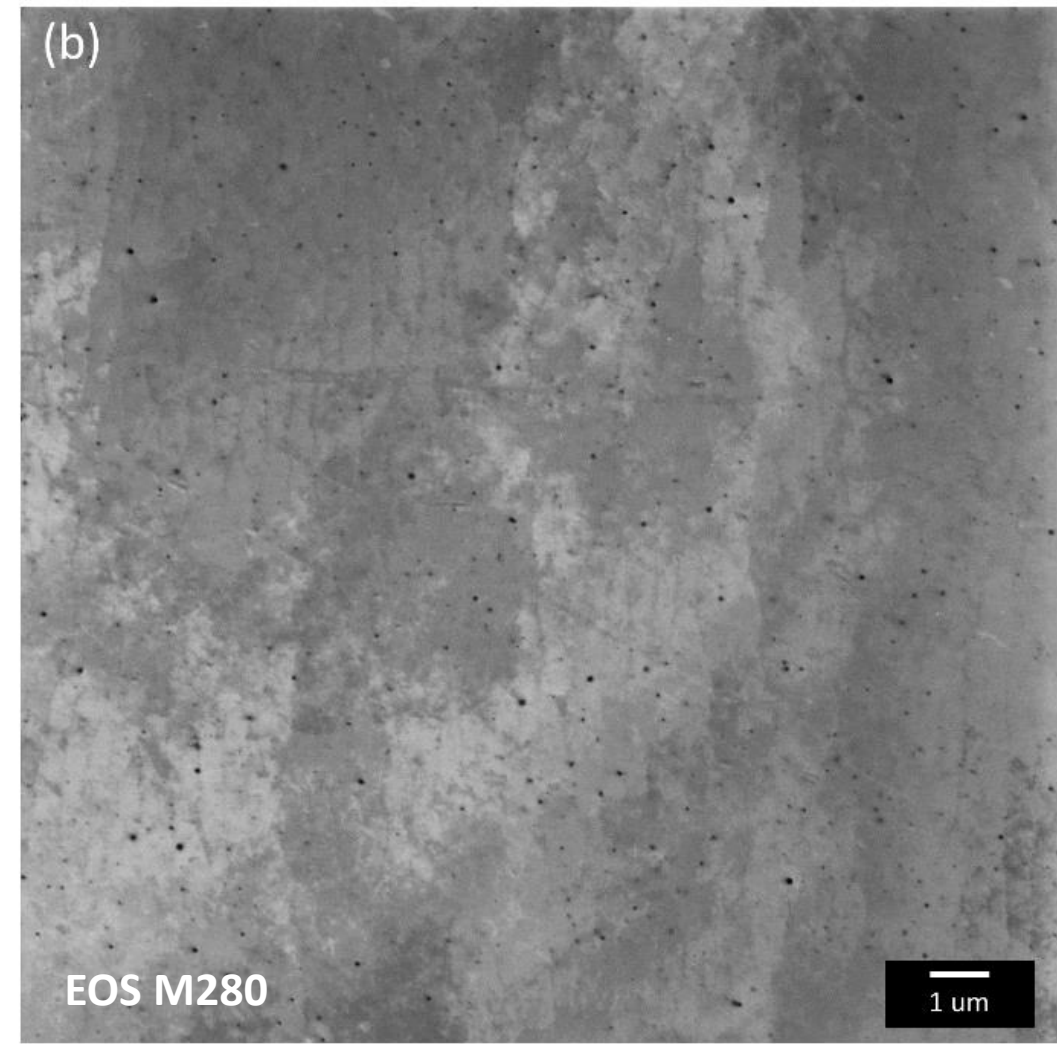
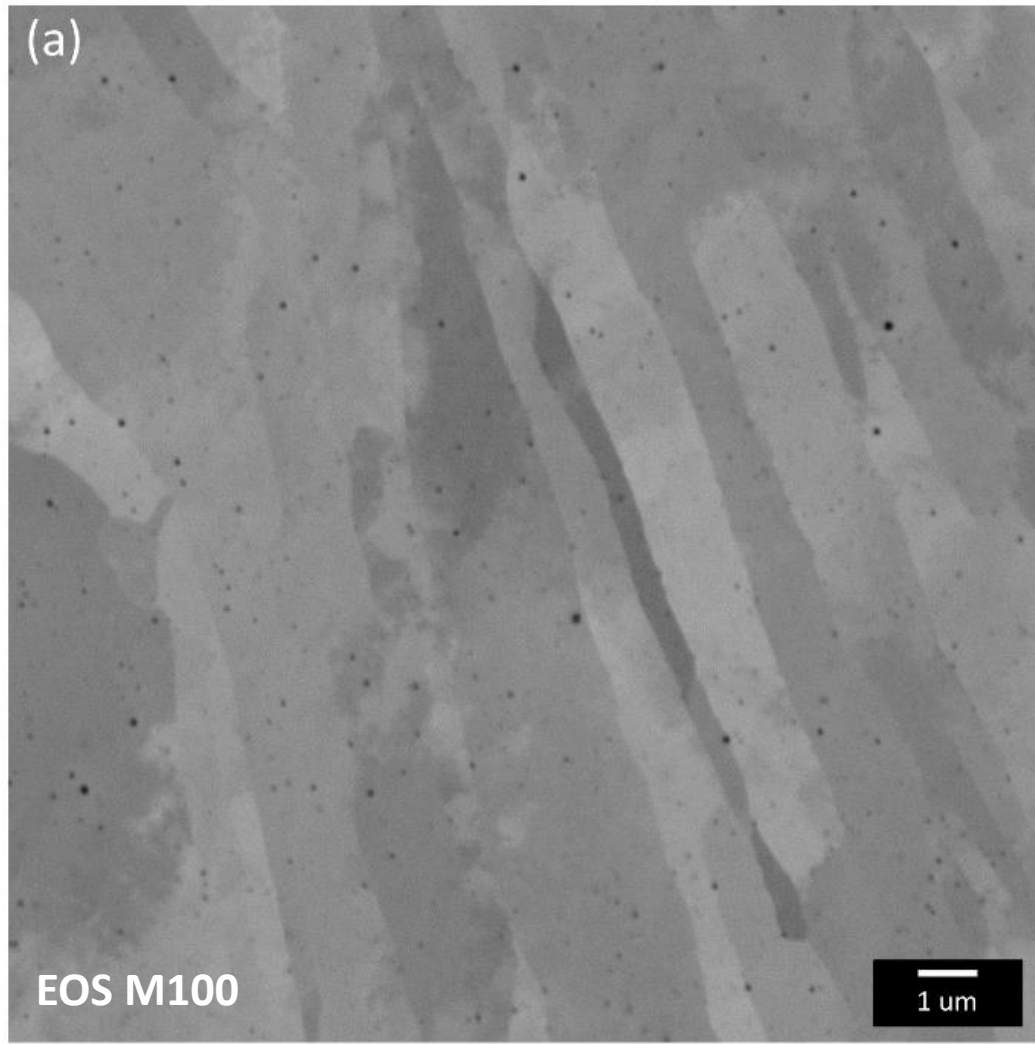


- Glenn Research Center eXtreme temperature alloy (GRX-810) was developed using Integrated Computational Materials Engineering (ICME) – significantly reducing development time.
- Ni-Co-Cr medium entropy alloy using Laser Powder Bed Fusion (L-PBF).
- Uses unique ODS powder coating process with nano-scale  $Y_2O_3$ .
- Significantly improved properties:
  - 2x strength at elevated temperatures ( $1100^\circ C$ )
  - 1,000x better creep rupture
  - 2x better resistance to oxidation
- Designed for simplified heat treatment – as-built or Hot Isostatic Pressing (HIP).

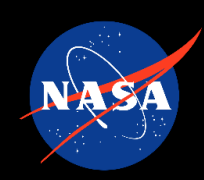




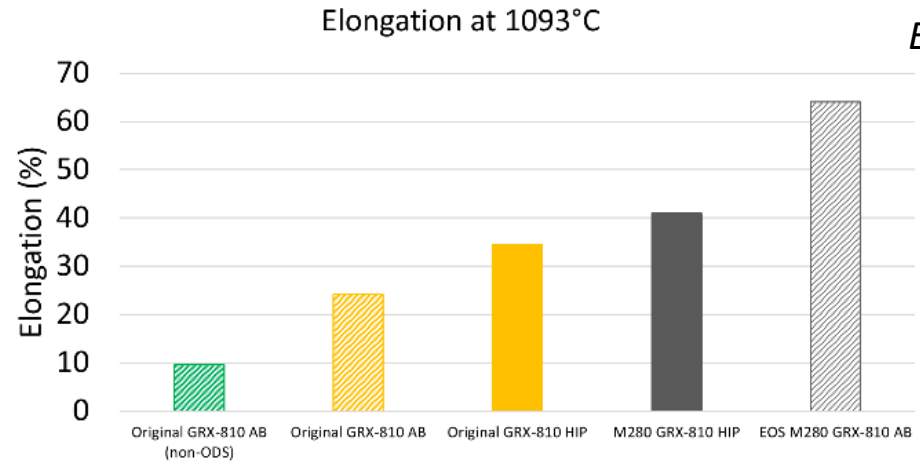
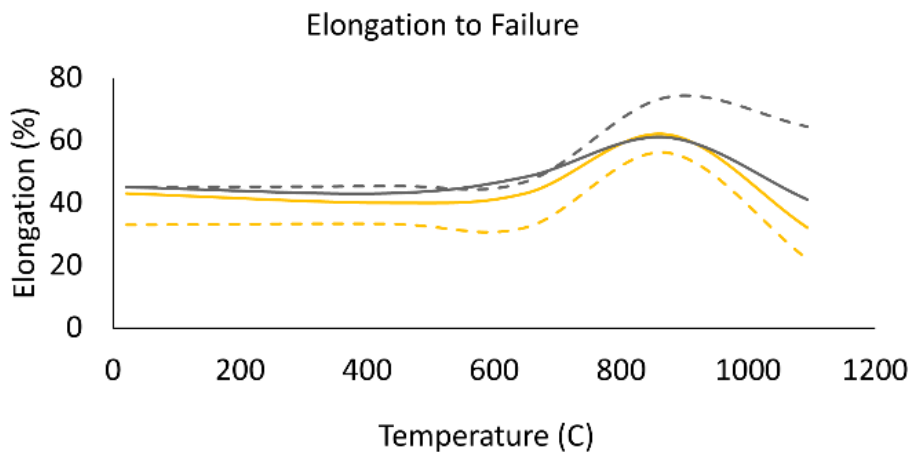
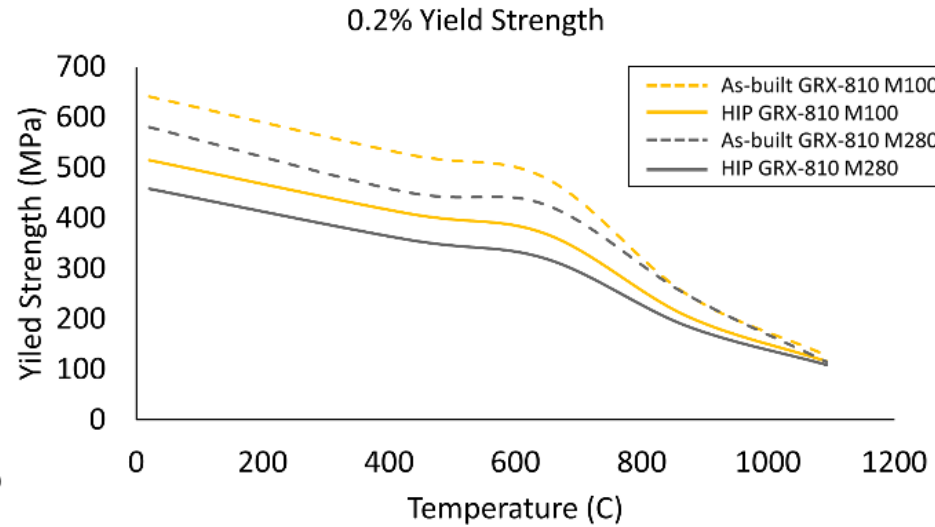
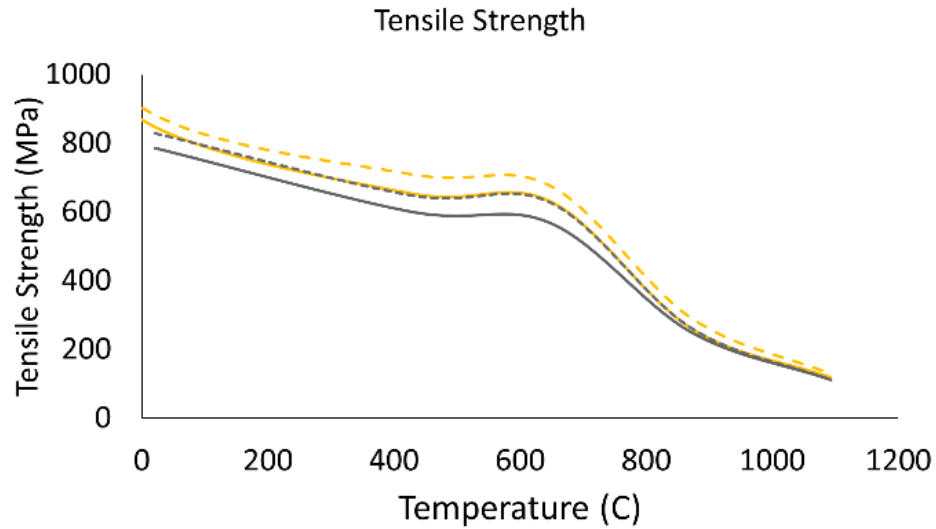
# SEM Images of GRX-810 using L-PBF



**Secondary electron SEM micrographs of (a) Lot 1 GRX-810 produced using an EOS M100 and (b) Lot 2 GRX-810 produced using an EOS M280. The fine circular dark features are nano-scale Y<sub>2</sub>O<sub>3</sub> particles.**

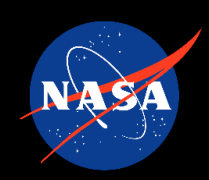


# GRX-810 Mechanical Properties



*EOS M100 (40 μm focus)  
EOS M280 (100 μm focus)*

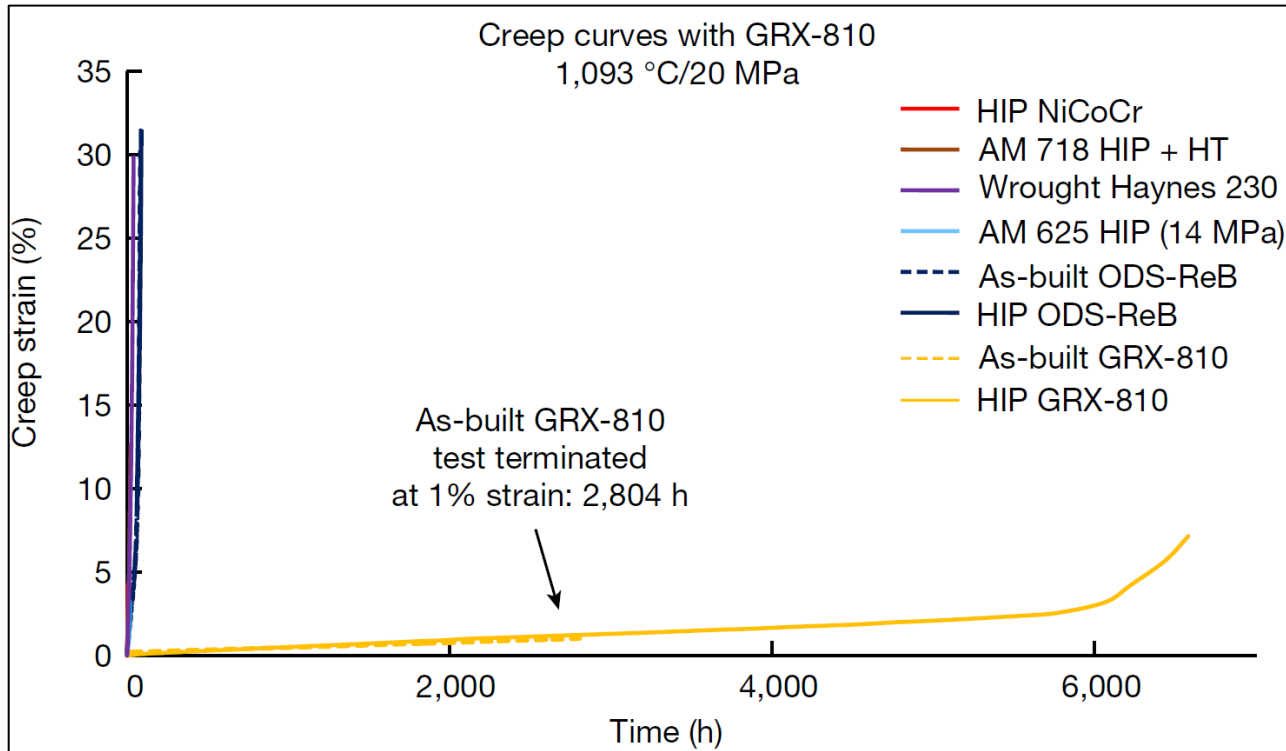
**Tensile properties of GRX-810 produced using an EOS M100 and using an EOS M280. Notable differences in room temperature strength and high temperature elongation were observed between the different lots.**



# GRX-810 Stress Rupture and Oxidation

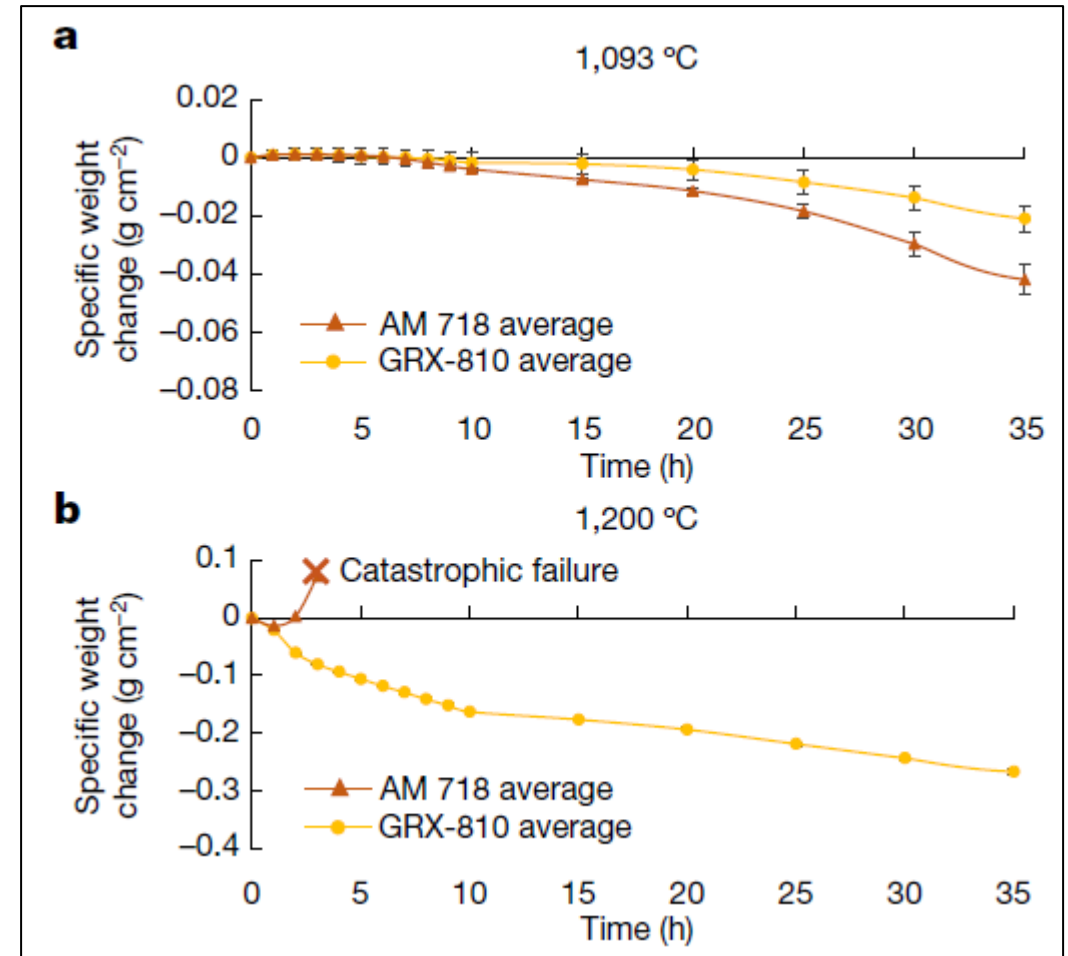


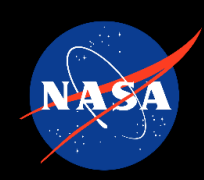
## Stress rupture shows 100x improvement over typical Ni-based superalloys



Creep results from M100 and M280 showing similar trends, although M280 performing better

## Cyclic oxidation at 1093°C and 1200°C

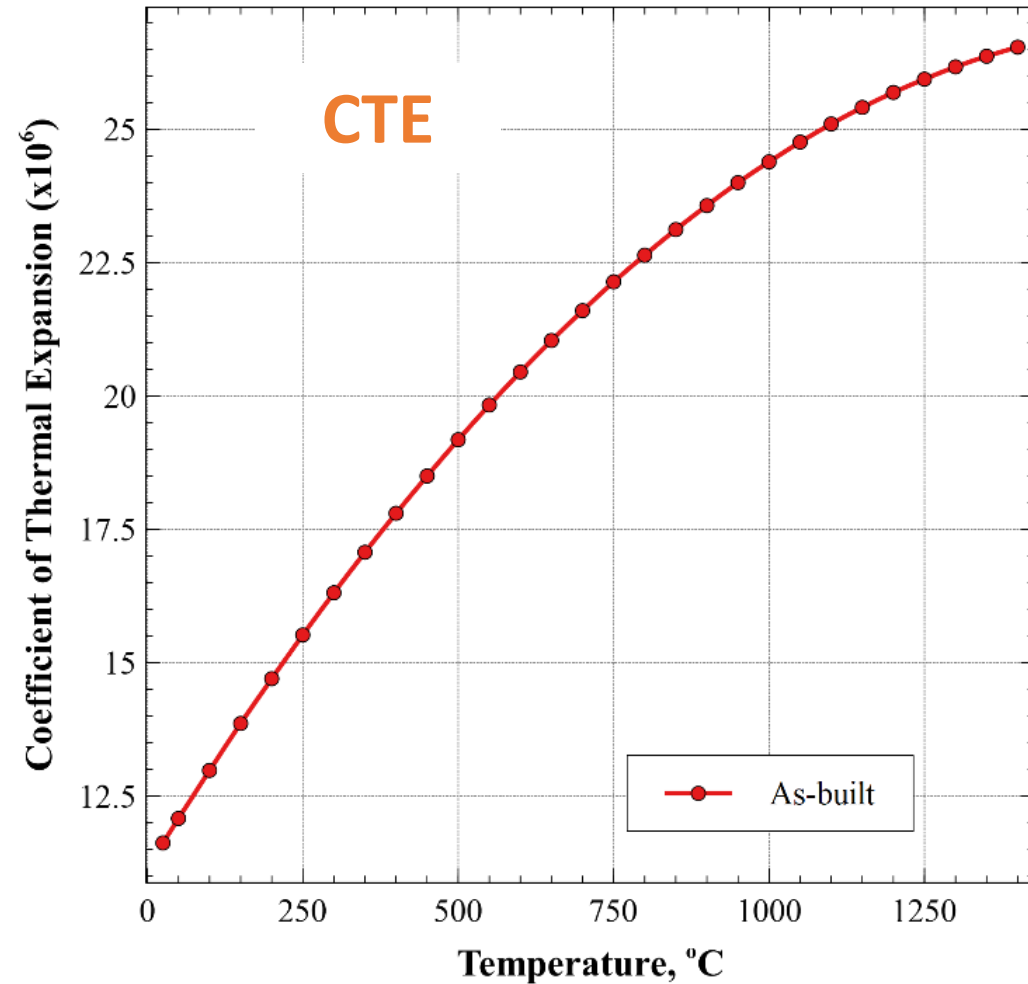
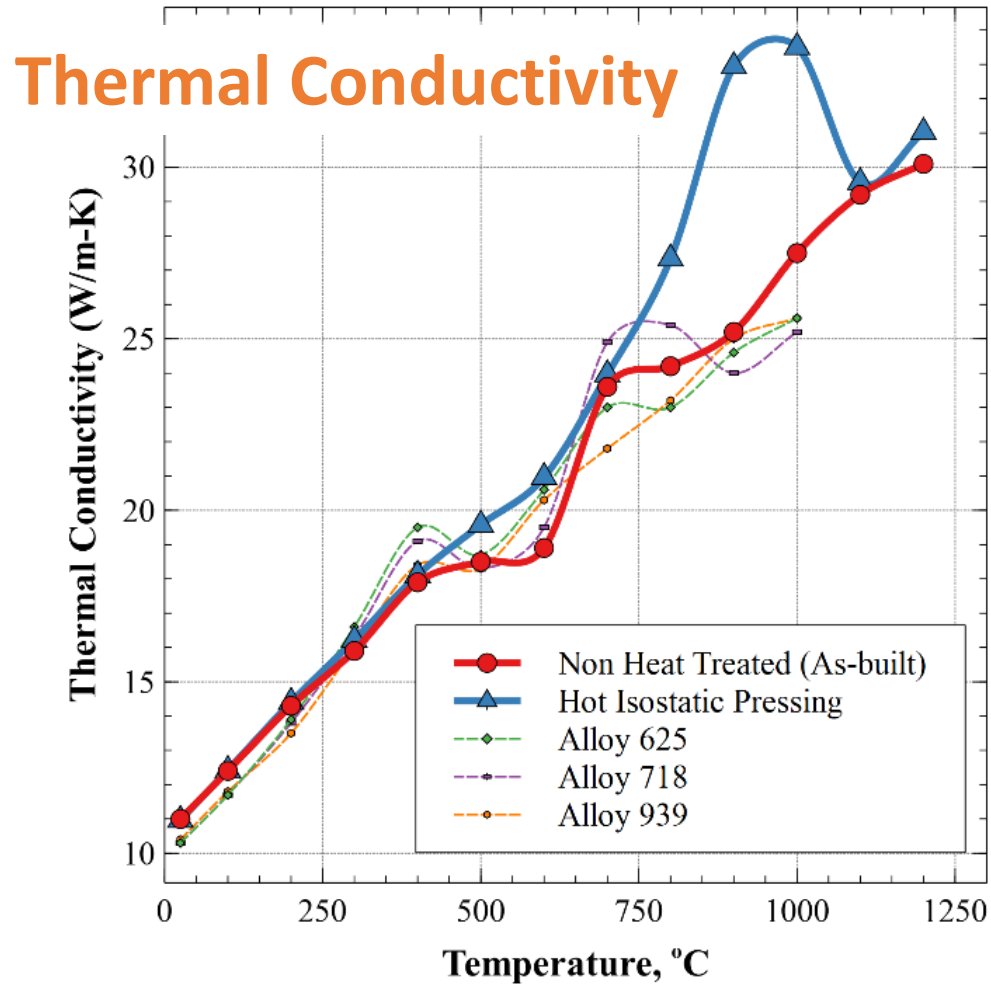




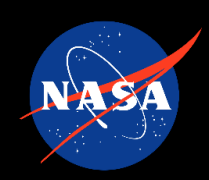
# GRX-810 Thermophysical Properties



**Thermal Conductivity and Coefficient Thermal Expansion (CTE) are similar to Ni-based superalloys**



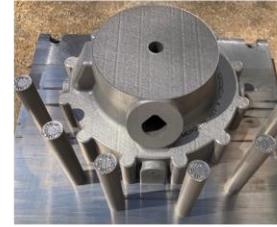




# Successful Hot-fire Testing



- Hot-fire testing of GRX-810 pentad injectors and channel-cooled nozzles using L-PBF GRX-810
- Propellants: LOX/LH2 and LOX/LCH4
  - Chamber Pressure  $\sim 750$  psig (52 bar)
  - LOX/LH2 Mixture Ratio = 5.3 – 7.0
  - LOX/LCH4 Mixture Ratio = 2.7 – 3.6
- Demonstrate successful component fabrication process
- Challenge material at elevated temperatures at MSFC TS115
- Increase TRL from 3  $\rightarrow$  5



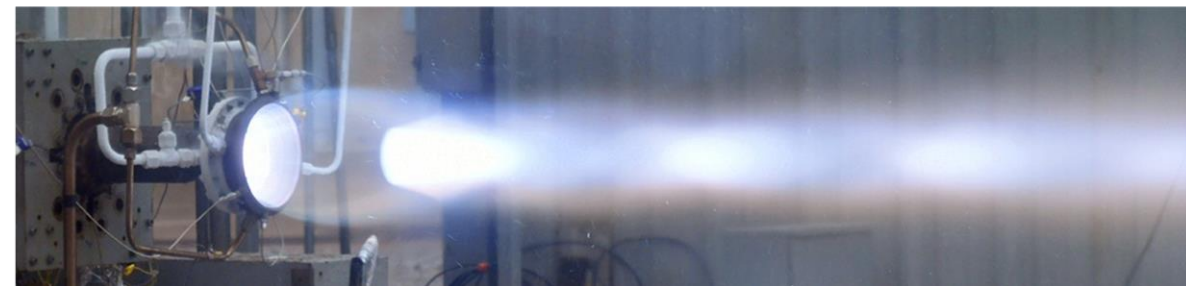
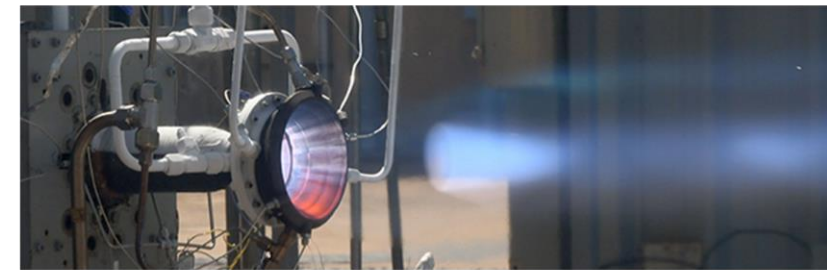
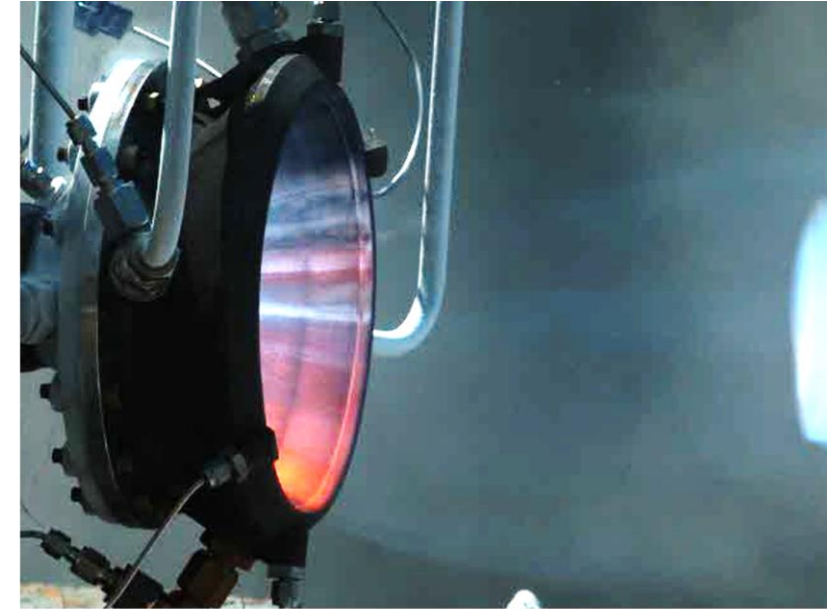
Injector

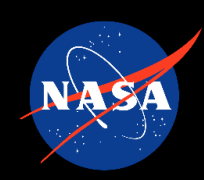


As-Built Nozzle

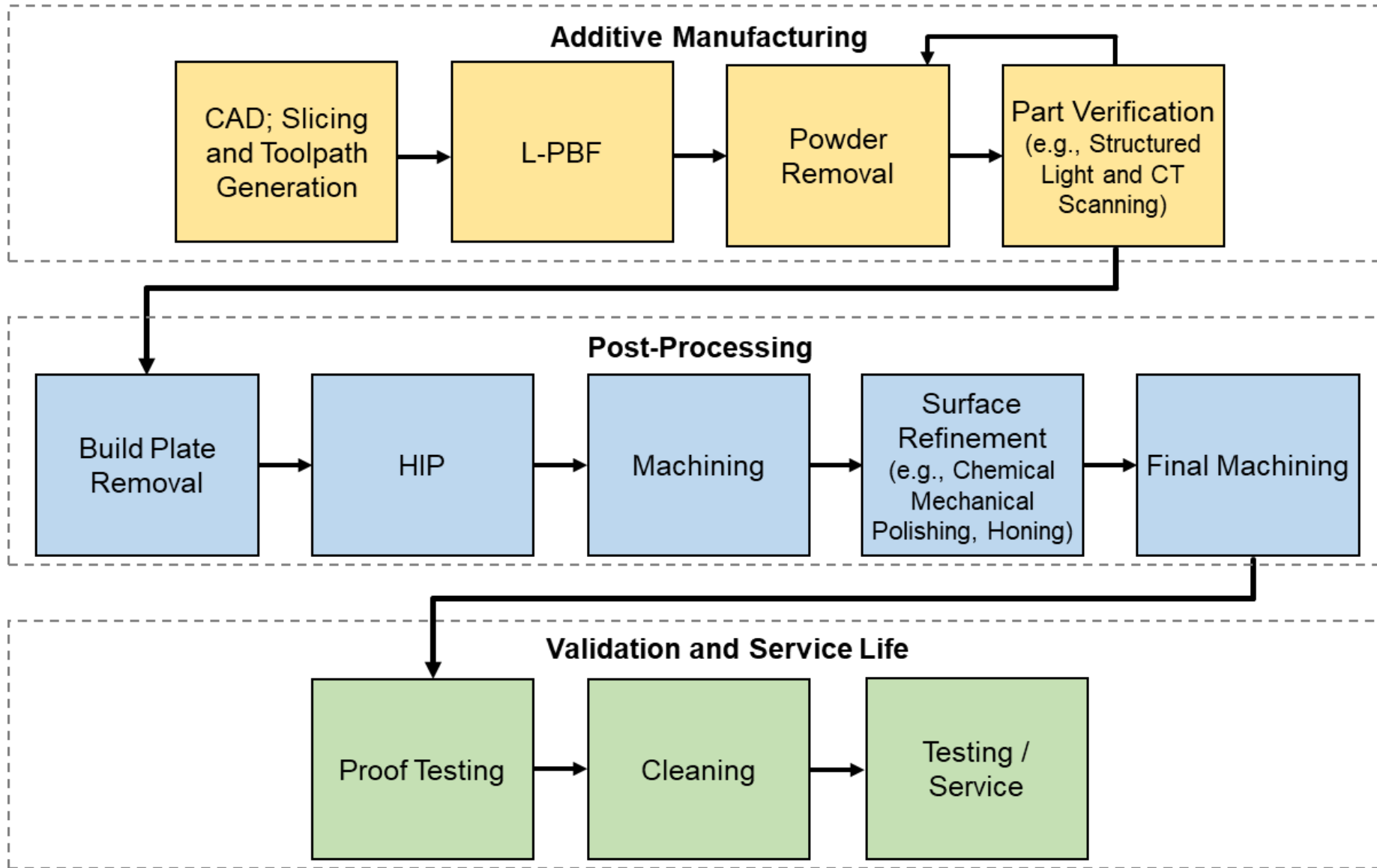


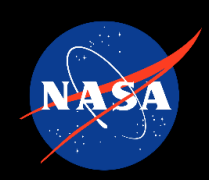
Post-HIP Nozzle



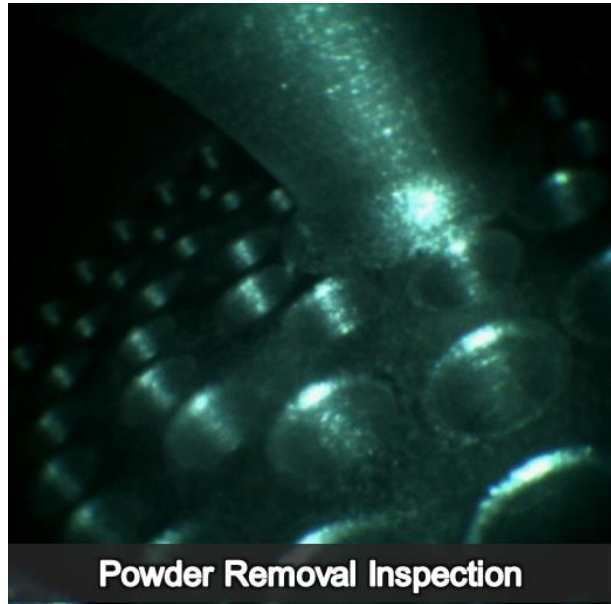


# Typical Process Flow of GRX-810 Hardware

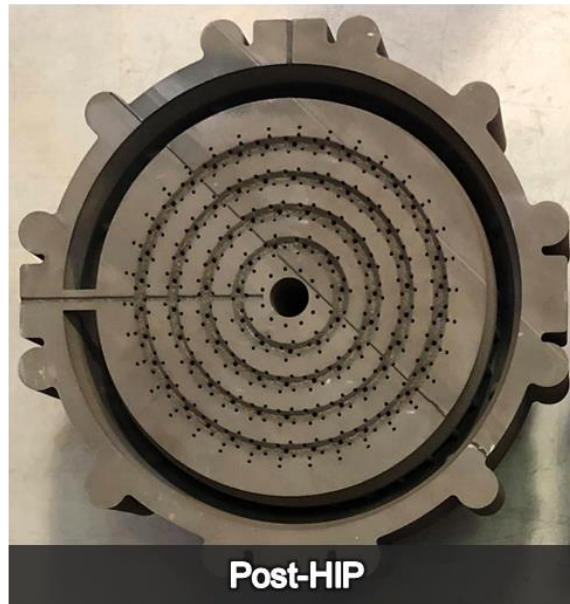




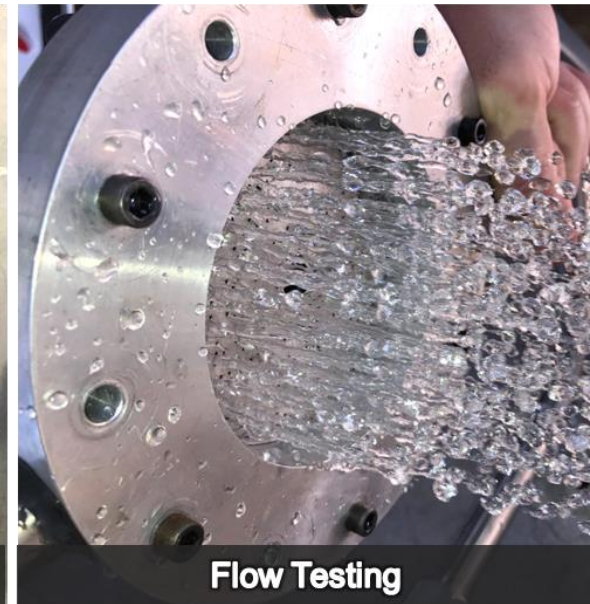
# Typical Process Flow of GRX-810 Hardware



**Powder Removal Inspection**



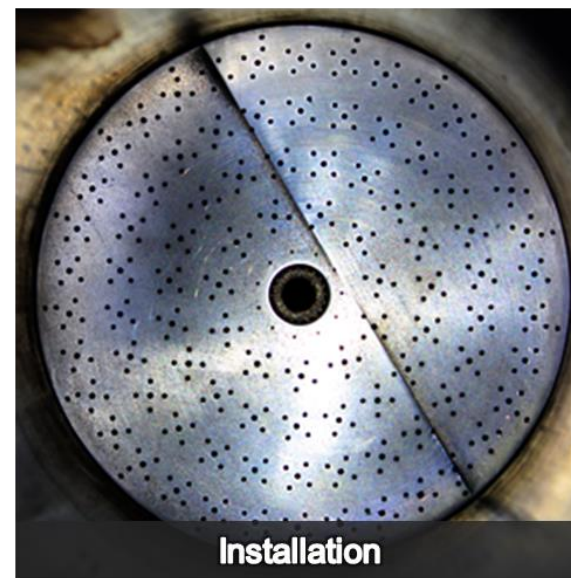
**Post-HIP**



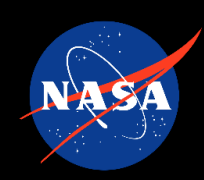
**Flow Testing**



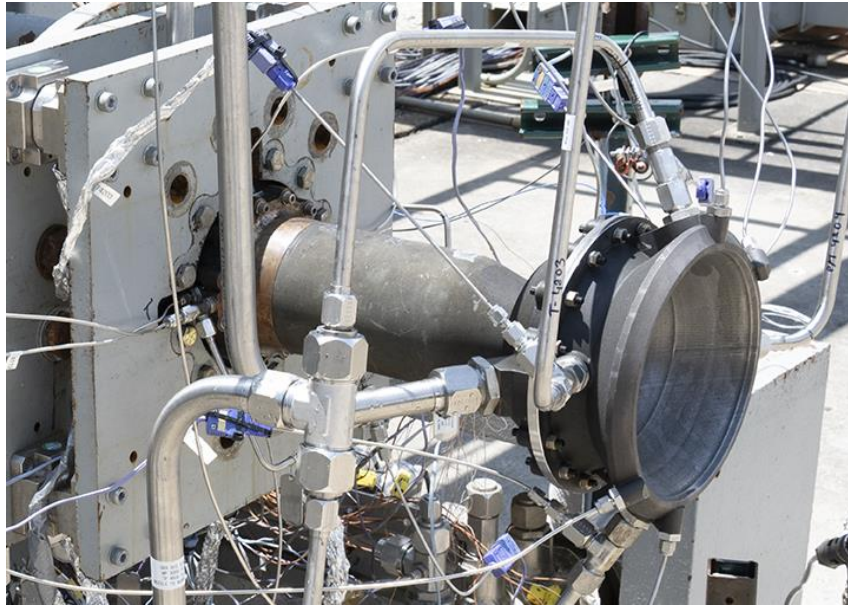
**Face Polish**



**Installation**



# Hot-fire Testing of L-PBF GRX-810 Injector and Nozzle

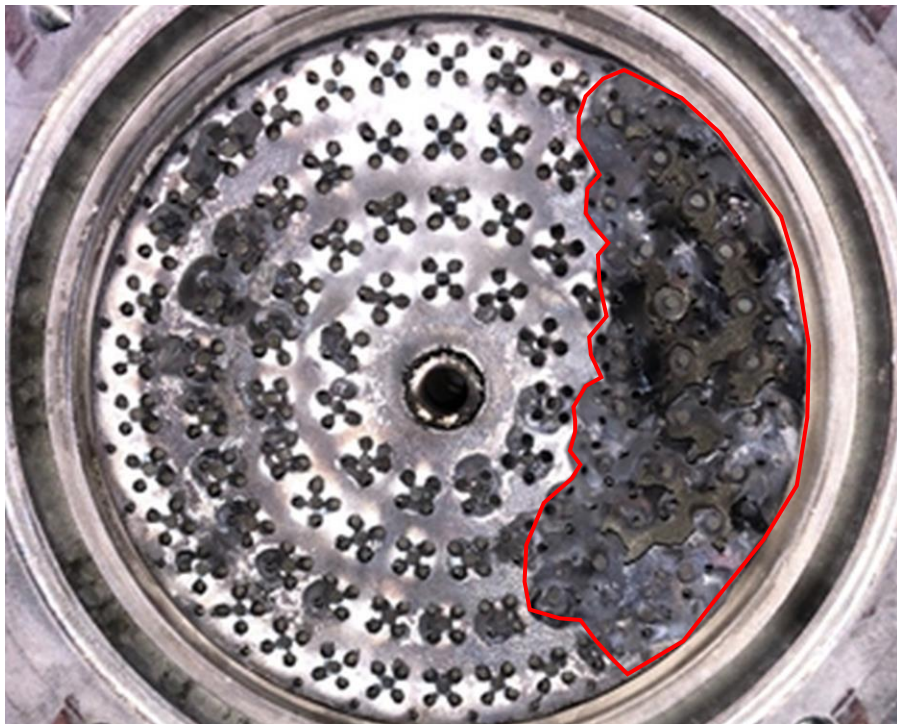


Component ( - )	Starts ( - )	Duration (s)	Pc		MR ( - )	
			(bar)	(psia)		
H2 Injector	SN01	9	302.8	49.5 - 57.2	718 - 829	5.33 - 7.02
CH4 Injector	SN02	29	586.5	37.4 - 52.4	542 - 760	3.03 - 3.65
CH4 Injector	SN03	84	2,227.9	43.2 - 52.1	626 - 756	2.68 - 3.19
Nozzle	SN04	91	2,309.4	37.4 - 52.1	542 - 756	2.68 - 3.11
Nozzle	SN05	8	149.1	49.0 - 50.5	711 - 732	3.00 - 3.19

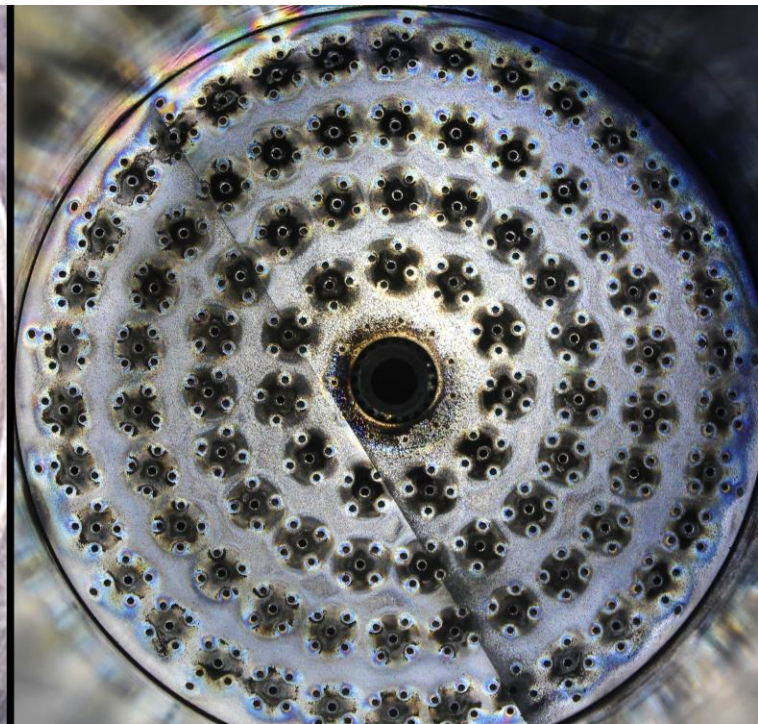
# Injector Test Results



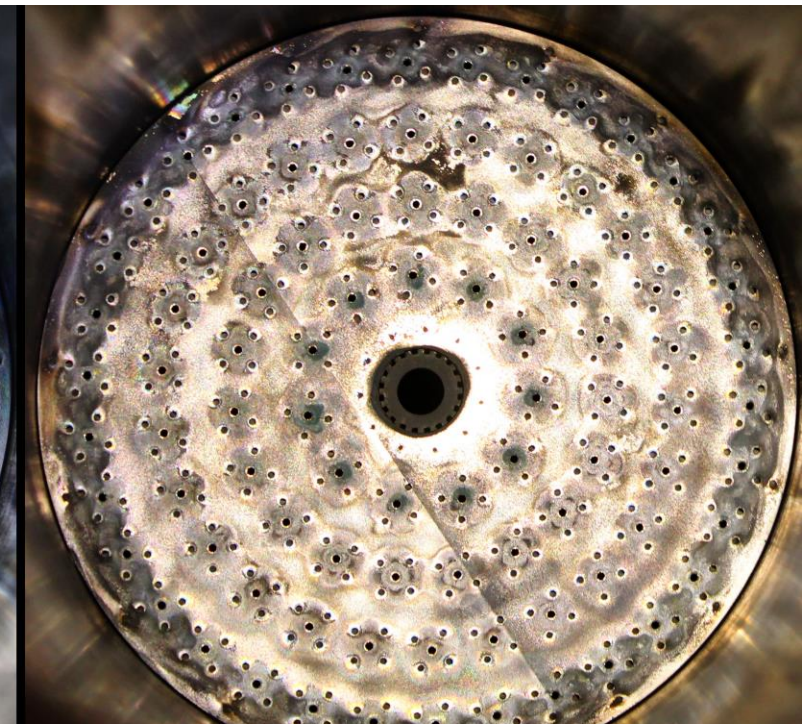
- GRX-810 Injector A achieved 84 starts and 2,228 sec (LOX/LCH4)
  - Demonstrated greater life than Inconel 625/718 equivalent
- GRX-810 Injector B achieved 30 starts and 591 sec (LOX/LCH4)
- GRX-810 Injector C achieved 9 starts and 303 sec (LOX/LH2)



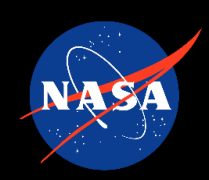
Inconel 625 Injector after 10 starts (Erosion)



GRX-810, LOX/LCH4 – 13 starts



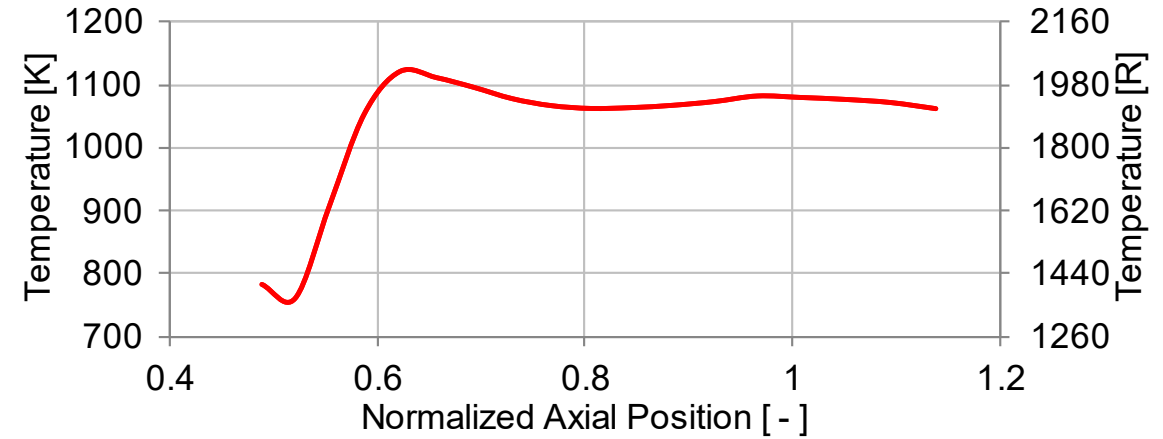
GRX-810, LOX/LCH4 – 84 starts



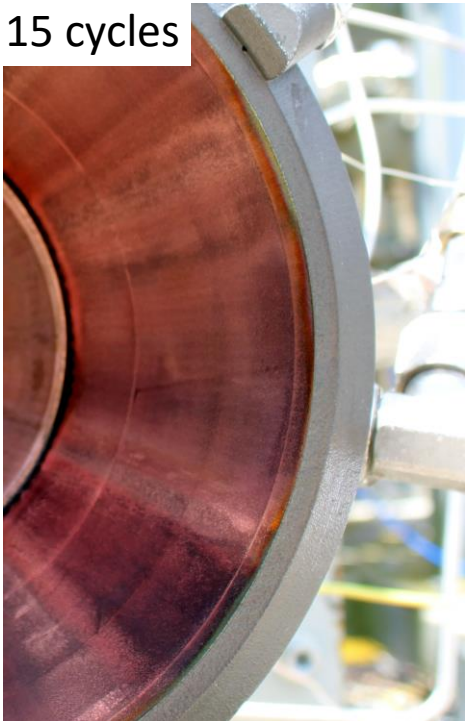
# GRX-810 Nozzle Test Results



- Regeneratively-cooled (LCH4) nozzle accumulated 90 starts and 2,309 seconds.
- Local wall temperature  $>1,000^{\circ}\text{C}$



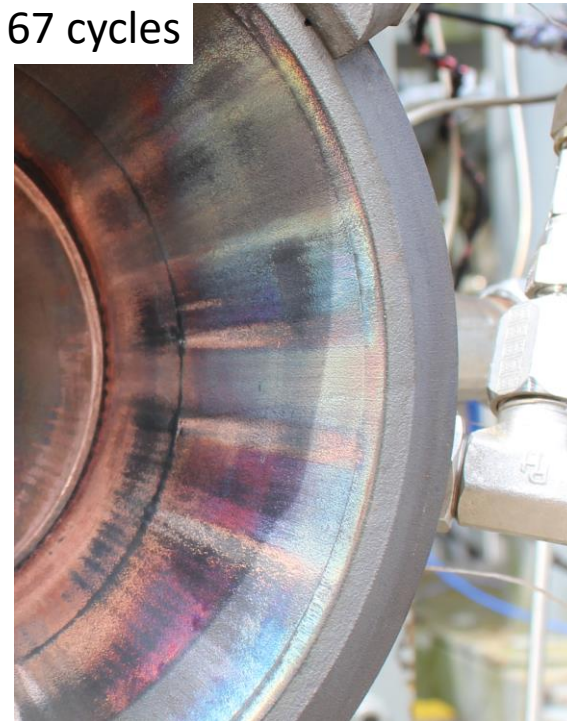
15 cycles



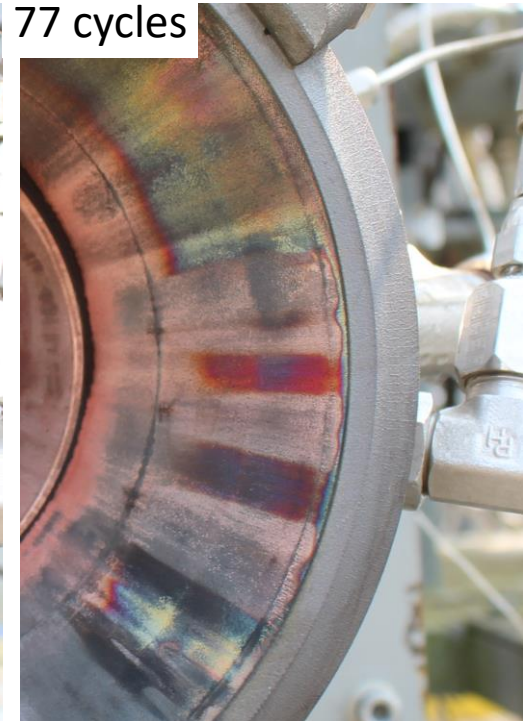
50 cycles

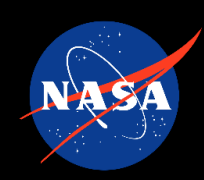


67 cycles



77 cycles





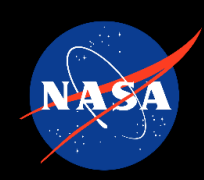
# Hot-fire Testing



GRX-810 Nozzle and Injector Testing



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# L-PBF GRX-810 Component Examples



Turbine Blisk  
Demo



Inducer with flow passages



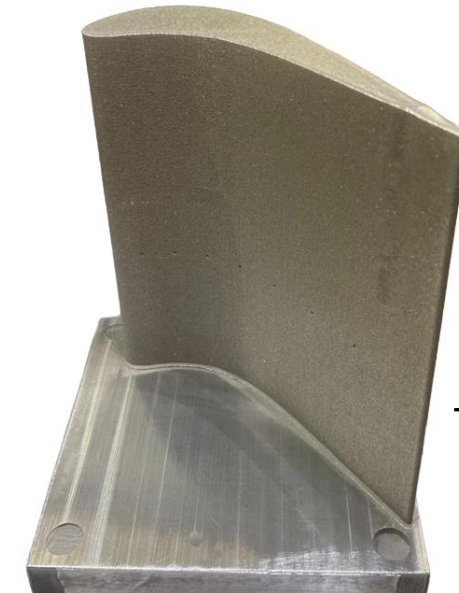
Pentad Injector



Regen Nozzle



Shrouded  
Blisk



Turbine Blade  
with integral  
ports

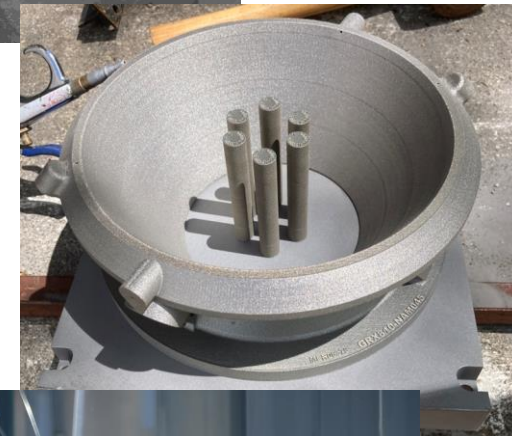
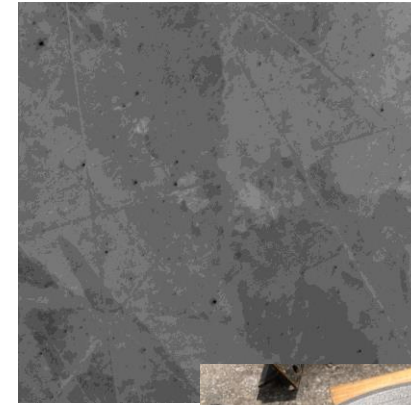




# Summary and Future Work

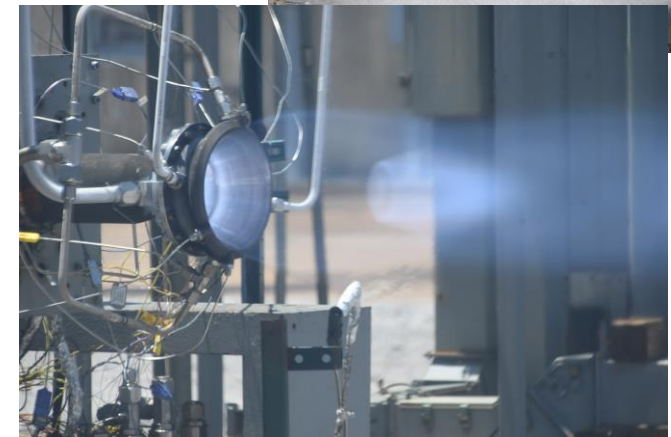


- Demonstrated successful formulation, powder processing and coating, microstructure characterization, mechanical properties, component development and hot-fire testing of GRX-810 alloy.
- Successful scale up from M100 to M280 platform and properties.
- Elevated temperature tensile (1100°C) is 2x Alloy 718.
- 1,000x improved creep rupture compared to Ni-based superalloys.
- Demonstrated successful injectors and nozzle L-PBF builds, processing, and hot-fire testing.
- Accumulated 221 starts across components, increased TRL = 5



## Ongoing Development

- Commercialization of the powder processing and coating.
- Continued mechanical testing and characterization ongoing.
- Scale up to M400 and larger machine platforms (L-PBF, DED).
- Properties and data will be available to industry partners.



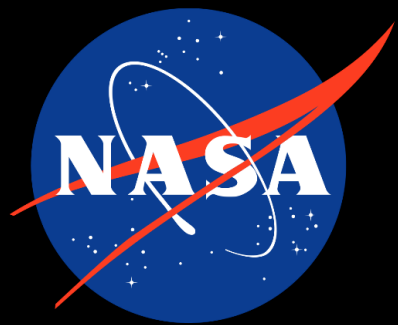


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# Acknowledgements



- Optimized and Repeatable Components using Additive (ORCA) Project
  - Under STMD Game Changing Program
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- Michael J. Mills
- John W. Lawson
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- Praxair
- RPM Innovations
- TS115 test crew
- And many others involved