

# Self-Aligned Focusing Schlieren (SAFS) Flow Visualization Technique

A compact, low-cost, easy-to-use flow visualization tool

## Challenge

- Understand how fluid flow interacts with bodies of interest
  - Simple wind tunnel models: Further understanding of fundamental physics
  - Aircraft/spacecraft: Measure flow field to evaluate full vehicle designs
- Requires non-intrusive, off-body measurement
  - Sensitive, compact, inexpensive, easy to align and adjust, quick setup time

## Expected Impacts

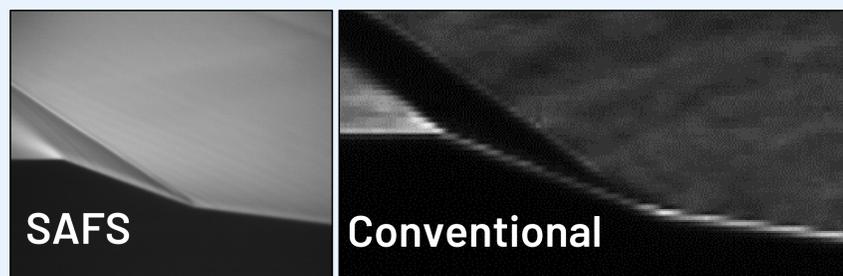
- CFD: High-quality flow visualization images for CFD simulation comparison/validation
- Researcher: Focusing schlieren now available as a tool for any researcher regardless of experience
- Flight testing: Simple flow visualization tool for flight testing and aircraft integration
- Big picture: May replace many existing conventional schlieren systems

## Solution

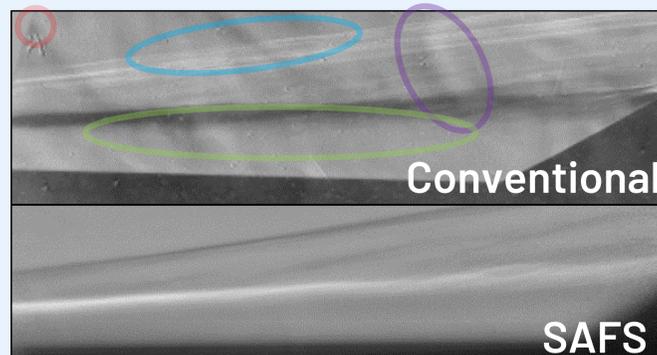
- SAFS system significantly reduces complexity of traditional focusing schlieren systems
  - Uses only one source/cutoff grid instead of two
  - Manipulates polarization of light for maximum light intensity throughput and window glare elimination

## Results

- Compact, with quick setup time (on the order of minutes)
- Insensitive to vibration
- Requires only single-sided optical access
- High-quality windows not needed
- Narrow depth-of-focus isolates only relevant flow features
- Resolution and sensitivity comparable to conventional schlieren
- Uses polarized light to eliminate window glare
- Non-orthogonal imaging for limited optical access facilities
- High-speed imaging possible (> 1 MHz)



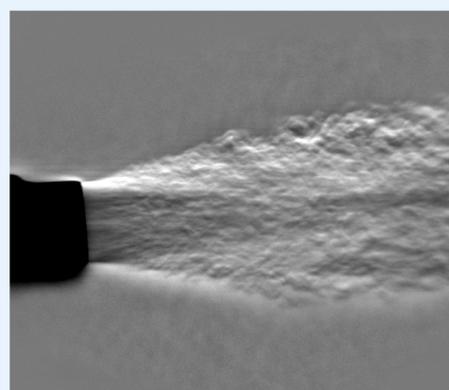
Comparison of SAFS (left) and conventional schlieren (right) images, cone-flare at Mach 3, with Mach waves from model's 3D printed layer lines visible in SAFS images



Comparison of conventional schlieren (top) and SAFS (bottom), demonstrating elimination of **window defects**, **path integration**, **HVAC thermals**, and **window boundary layers**



SAFS installed inside Beechcraft B200 King Air, with curved retroreflective material on engine cowling



Heat gun flow at propeller tip vortex position, demonstrating SAFS feasibility

## Ongoing Work

- Quantify sensitivity of system
- Develop software to aid in preliminary system design
- Implement system in supersonic flight vehicle (AFRC F-15)
- Aid other researchers in their own design applications

## Partners and/or Participants

- NASA Transformational Tools and Technologies Project
- NASA Advanced Air Transport Technology Project
- NASA Flight Demonstrations and Capabilities Project
- NASA Revolutionary Vertical Lift Technology Project

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